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Mechanics Research Communications xxx (2006) xxx-xxx

MECHANICS RESEARCH COMMUNICATIONS

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Trigonometric function used to formulate a multi-nodal finite tubular element

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10 Abstract

11 It is presented an alternative formulation to solve the problem of the deformation analysis for tubular element under 12 pinching loads. The solution is based on a new displacement field defined from a total set of trigonometric functions. The 13 solution is developed in a multi-nodal finite tubular ring element with a total of eight degrees of freedom per section con-14 sidered. The purpose of this paper is to provide an easy alternative formulation when compared with a complex finite shell 15 element or beam element analysis for the same application. Several case studies presented have been compared and dis-16 cussed with numerical analyses results reported by other authors and the results obtained with a shell element from a Cos-17 mos/M[®] programme. 18 © 2006 Published by Elsevier Ltd.

19 Keywords: Multi-nodal; Finite tubular ring element; Pinching loads; Trigonometric functions 20

21 1. Introduction

Structural design methodologies for pipelines systems start to be used, especially when it is expected that 22 enhancements in the design process will lead to more cost effective and durable designs. These piping elements 23 exhibit complex deformations fields given their toroidal geometry and the multiplicity of the configuration of 24 external loads. Mathematical and experimental models which permit to verify the flexibility of pipe systems 25 26 were published for a long time by some researchers working in this scope: (von Kárman, 1911; Vigness, 1943; Cheng and Thailer, 1970) and more recently (Thomson, 1980; Wilczek, 1984; Bathe and Almeida, 27 1982). News models used for stress and displacements fields determination, under mechanical or thermal 28 loads, were recently formulated using numerical techniques with new finite elements by Melo and Castro 29 (1992), Fonseca et al. (2002, 2005). Studies on circular cylindrical shells under concentrated forces have been 30 31 presented by many researchers using curved beam, shell and solid elements (Kouhia and Stenberg, 2000; Gob-

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^{0093-6413/\$ -} see front matter @ 2006 Published by Elsevier Ltd. doi:10.1016/j.mechrescom.2006.06.008

24 June 2006 Disk Used

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E.M.M. Fonseca et al. | Mechanics Research Communications xxx (2006) xxx-xxx

etti and Nascimbene, 2001). Shell element formulations based on the so-called degenerate models introduced 32 by Ahmad et al. (1970) have been used due their ability in moderately thick and thin plate or shell elements A 33 long report about the development of finite shell or plate elements with emphasis on innovations was pre-34 35 sented by MacNeal (1998). News formulations of solid element were presented for the linear and nonlinear analysis of thin-walled structures by Kim et al. (2004). The particular case of the formulation of cylindrical 36 37 shells using ring elements has a straightforward contribution from Oñate (1992), where this author used a 38 combined formulation for the displacement field dealing with fourier series along the circumference and alge-39 braic shape functions along the axial direction. Analysis of a circular ring is reasonable for the structural design of most buried pipes when they can withstand the three most basic loads; internal pressure, transpor-40 41 tation/installation and external pressure (Watkins and Anderson, 2000). The ring deflection is a potential per-42 formance limit for transportation/installation of pipes. The aim of this study is to develop an alternative numerical method and present an experimental work for pipes analysis when used in underground sewerage 43 without internal pressure or when submitted to mechanical loads imposed during transportation or an instal-44 lation. Satisfactory results for circular rings can be obtained by several different methods. This work presents 45 46 an alternative formulation based on one finite element with two nodal sections used to simulate pinching loads 47 in the referred situations and analyse the structural mechanics of pipes in linear bending conditions.

48 2. Finite element formulation: assumptions and deformation model

The geometric parameters considered for this element definition are: the length of the pipe s, the wall thickness t and the mean section radius of the pipe r. Fig. 1 shows the essential parameters and the degrees of freedom used to define the finite tubular ring element.

The basic kinematics assumptions refer to the deformation of a thin shell as used in a small-defection analysis. The assumptions in the problem formulation of in-plane bending are: the shell is thin, this meaning that the normal to the shell surface does not distort and the transverse section is inextensible, not including pressure effects. The strain analysis and curvature geometry field of a non-symmetrically deformed cylindrical shell may be founded on the behaviour of plates and shells of revolution. Kinematics expressions relating the midsurface strains are expressed by (1-3) and the curvatures and twist by (4-5).

$$\varepsilon_{ss} = \frac{\partial u}{\partial s}, \quad \varepsilon_{\theta\theta} = \frac{1}{r} \left(\frac{\partial v}{\partial \theta} + w \right) \quad \text{and} \quad ma_{s\theta} = \frac{1}{r} \frac{\partial u}{\partial \theta} + \frac{\partial v}{\partial s} \tag{1-3}$$

63
$$\chi_{\theta\theta} = \frac{1}{r^2} \left(-\frac{\partial v}{\partial \theta} + \frac{\partial^2 w}{\partial \theta^2} \right) \text{ and } \chi_{s\theta} = \frac{2}{r} \left(-\frac{\partial v}{\partial s} + \frac{\partial^2 w}{\partial s \partial \theta} \right)$$
 (4-5)

64 where ε_{ss} is the longitudinal membrane strain, $\varepsilon_{\theta\theta}$ is the meridional curvature from ovalization, $\gamma_{s\theta}$ the shear 65 strain, $\chi_{\theta\theta}$ is the meridional curvature variation from ovalization and $\chi_{s\theta}$ the twist variation.

A simplified theory is useful under certain conditions and is applicable to a variety of shells forms. However, we shall deal only with the inextensional deformation of circular cylindrical shells. This theory is often

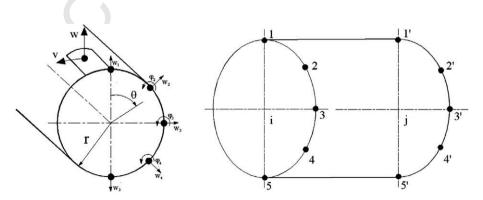


Fig. 1. Geometry and degrees of freedom of the finite tubular ring element.

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E.M.M. Fonseca et al. | Mechanics Research Communications xxx (2006) xxx-xxx

for preferred when shell structures resist loading principally through bending action. Such cases include a cylinder subjected to loads without axial symmetry and confined to a small circumferential portion and there is considerable bending caused by changes in curvature, but no stretching of midsurface length. Deformations of these types are thus described as inextensional referred by Ugural (1981). In this shell theory, the midsurface in-plane strain components given by Eqs. (1–3) are taken to be zero. We realise from Eq. (1) that u depends on $\frac{23}{24}$ θ and Eq. (2) leads to:

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$$w = -\partial v / \partial \theta$$

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77 3. The displacement field for the finite tubular ring element

The displacement filed proposed characterises a uniform ovalization in the tube and a variation along the shell length. To obtain the shape functions a displacement field has been considered in the normal radial direction. As represented in Fig. 1 the element presents five nodes under symmetric conditions make possible studies in-plane bending. So, only a half of a ring section is needed to consider when trigonometric even functions are used. The initial radial displacement used to define the ovalization effect must be calculated using the trigonometric polynomial approximation. A formulation based on trigonometric functions is used and eight parameters are necessary to define the transversal displacement field approximated by:

86
$$w(s,\theta) = \sum_{i=2}^{9} a_{i-1}i\cos i\theta + S\sum_{i=2}^{9} b_{i-1}i\sin i\theta$$
(7)

87 Using simple differential equation from beam bending theory the transverse displacement can be calculated:

$$v(s,\theta) = -\sum_{i=2}^{9} a_{i-1} \sin i\theta - S \sum_{i=2}^{9} b_{i-1} \cos i\theta$$
(8)

90 The rotation field is considered using the derivative function of the transverse displacement field:

92
$$\varphi(s,\theta) = \frac{1}{r} \frac{\partial w}{\partial \theta} = -\frac{1}{r} \sum_{i=2}^{9} a_{i-1} i^2 \sin i\theta - \frac{S}{r} \sum_{i=2}^{9} b_{i-1} i^2 \cos i\theta$$
(9)

The unknown parameters a_i are determined by imposing boundary conditions according to the ring element section *i* and *j* considered, resulting a system of equations to be solved. The degrees of freedom considered in the proposed element are: node 1 and 5 are the transversal displacement and their derivatives vanish, for nodes 2 at 4 have a radial displacement and one derivative function. A system with 16 equations is solved for *i* and *j* section of the ring element and shapes functions are obtained:

$$\{\delta\} = [B']\{a_i\} \tag{10}$$

101 with $\{\delta\}^{T} = \{W_{1i} \ W_{2i} \ \varphi_{2x} \ W_{3i} \ \varphi_{3i} \ W_{4i} \ \varphi_{4i} \ W_{5i} \ W_{1j} \ W_{2j} \ \varphi_{2j} \ W_{3j} \ \varphi_{3j} \ W_{4j} \ \varphi_{4j} \ W_{5j}\}^{T}$ 102 which represents the global displacement field for transversal and rotation degrees of freedom in the semi no-103 dal ring and [B'] is constant matrix that results from the imposed boundary conditions. For node 1 the trans-104 versal displacement is equal to one and all others equal to zero. The first shape function appears and is called 105 N_{1i} . With imposed Eq. (6) we determine the shape function BN_{1i} . The same has been used for all others nodes. 106 The unknown constants are determined inverting the system equation (10). The first shape function is repre-107 sented by Eq. (11). With these conditions, a new shape functions are determined and the generic local displace-108 ments field for in-plane finite element formulation are given by Eqs. (12) and (13).

$$N_{1i} = \left[\frac{3}{16}\cos(2\theta) + \frac{5}{32}\cos(3\theta) + \frac{1}{8}\cos(4\theta) + \frac{3}{32}\cos(5\theta) + \frac{1}{16}\cos(6\theta) + \frac{9}{64}\cos(7\theta) + \frac{1}{8}\cos(8\theta) + \frac{7}{64}\cos(9\theta)\right] - s\left[\frac{3}{16}\frac{\cos(2\theta)}{L} + \frac{5}{32}\frac{\cos(3\theta)}{L} + \frac{1}{8}\frac{\cos(4\theta)}{L} + \frac{3}{32}\frac{\cos(5\theta)}{L} + \frac{1}{16}\frac{\cos(6\theta)}{L} + \frac{9}{64}\frac{\cos(7\theta)}{L} + \frac{1}{8}\frac{\cos(8\theta)}{L} + \frac{7}{64}\frac{\cos(9\theta)}{L}\right]$$
(11)

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(6)

E.M.M. Fonseca et al. | Mechanics Research Communications xxx (2006) xxx-xxx

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$$v(s,\theta) = BN_{1i}W_{1i} + BN_{2i}W_{2i} + BN'_{2i}\varphi_{2i} + BN_{3i}W_{3i} + BN'_{3i}\varphi_{3i} + BN_{4i}W_{4i} + BN'_{4i}\varphi_{4i} + BN_{5i}W_{5i} + BN_{1j}W_{1j} + BN_{2j}W_{2j} + BN'_{2j}\varphi_{2j} + BN_{3j}W_{3j} + BN'_{3j}\varphi_{3j} + BN_{4j}W_{4j} + BN'_{4j}\varphi_{4j} + BN_{5j}W_{5j}$$
(12)

$$w(s,\theta) = N_{1i}W_{1i} + N_{2i}W_{2i} + N'_{2i}\varphi_{2i} + N_{3i}W_{3i} + N'_{3i}\varphi_{3i} + N_{4i}W_{4i} + N'_{4i}\varphi_{4i} + N_{5i}W_{5i} + N_{1j}W_{1j}$$

$$+N_{2j}W_{2j} + N'_{2j}\varphi_{2j} + N_{3j}W_{3j} + N'_{3j}\varphi_{3j} + N_{4j}W_{4j} + N'_{4j}\varphi_{4j} + N_{5j}W_{5j}$$
(13)

118 All other shape functions are determined in the same form. This formulation presents a simple formula for 119 calculation of the displacement field for the tangential and transversal displacement under a shell element:

121
$$\{v \ w\}^{\mathrm{T}} = [N] \times \{\delta\}$$

(14)

(15 - 16)

(17)

As referred previously, the mechanical deformation model considers that the pipe undergoes a semi-membrane strain field by Eqs. (4–5). The typical use of the principle of virtual work leads to the system of algebraic equations. Having solved the system of algebraic equations, the displacement field is obtained for all the nodes of the finite tubular ring element.

125 the limite tubular ring element.

126 4. Deformation models on pipes design

One step of design is resistance to loads imposed on the pipe during transportation and installation. The most common load is the diametral F-load, see Fig. 2. This load appears when pipes are stacked or when soil is compacted on the sides or on top of the pipe (Watkins and Anderson, 2000). If yield strength of the pipe material is exceeded due the F-load, either the pipe wall will crack or the cross section of the pipe will permanently deform. Either of these deformations, a crack is a deformation but unacceptable. The yield strength may possible be a performance limit even though the ring does not collapse.

The limit of the F-load for some manufactures pipes is based on the maximum allowable ring deflection based on Eq. (15). For the design process of the pipe transportation/installation loads, the maximum allowable F-load for plain pipes is calculated by the expression 16 (Watkins and Anderson, 2000).

138
$$d' = d/D$$
 and $F = (\pi \sigma_c D/3)(t/D)^2$

139 The ring defection for mechanical analysis of thin-walls rings with symmetrical loads is given by:

141
$$d' = 0.0186(FD^2/EI)$$

where d' represents the ring deflection, *d* the decrease in diameter, *D* the mean diameter of the pipe, *t* the wall thickness for plain pipe, σ_c is the yield strength and *EI* the wall stiffness per unit of length of the pipe.

144 5. Case studied 1: pinched cylinder

The pinched cylinder problem tests the ability of simulation to model complex bending and membrane states. In the pinched cylinder, the shear-locking phenomenon is more detrimental to the solution accuracy than membrane-locking. A severe problem for most finite elements is the locking which can appear as shear or membrane locking, as reported by Andelfinger and Ramm (1992). Transverse shear locking can occur in

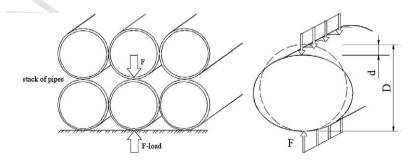


Fig. 2. Cylinder submitted to axisymmetric compression load: common transportation/installation on pipes.

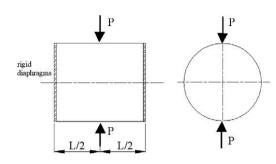


Fig. 3. Geometry of a pinched cylinder loaded.

149 shear deformable beam, plate and shell elements or solid elements if these are applied to the analysis of thin-150 walled structures. Transverse shear locking is one of the most important locking effects because it can essen-151 tially exclude an analysis with a reasonable amount of numerical effort in practical applications. This effect is 152 significant only if there is a certain in-plane bending deformation of the structure. Shear locking occurs where 153 the shear stiffness is a lot higher than the bending stiffness. Membrane locking can occur in thin shells or 154 curved elements where the membrane stiffness is a lot higher than the bending stiffness. Researchers use the 155 pinched cylinder problem as a benchmark test to assess the performance of curved beam or shells elements. The problem here presented, concerns a pinched cylinder with rigid end diaphragms, subjected to diametrical 156 load as shown in Fig. 3, proposed by MacNeal and Harder (1985) and studied by a several authors. The geo-157 158 metric configuration of the cylinder is the length (L = 600 in.), the radius (r = 300 in.) and wall thickness (t = 3 in.). The elastic modulus was taken $(E = 3 \times 10^6 \text{ psi})$ and Poisson's ratio (v = 0.3). Only a half of the 159 cylinder is modelled due the geometry and load symmetry. The point load applied is equal (P = 1/2 lb). 160 161 The reference displacement, as reported by Dvorkin and Bathe (1984), Kouhia and Stenberg (2000) and the exact deflection at the load point is equal (0.18248E-4 in.), using $w_{\text{exact}} = 164.24(P/Et)$. This is known 162 163 as a benchmark result where there is a bending dominant behaviour in the thin shell limit. Thus transverse 164 shear and membrane locking occurs as the shell becomes thinner. Fig. 4 represents the vertical displacement along length of cylinder obtained with our formulation with two different meshes and shell element from Cos-165 166 mos/M^{\otimes} .

Table 1 shows the normalized vertical displacement under the load of the pinched cylinder modelled using our proposed element and other references. Fig. 5 shows the vertical displacement using $Cosmos/M^{\ensuremath{\mathbb{S}}}$. The solution obtained by the ring element does not exceed the analytical solution because it does not include transverse shear deformation. Thus, the numerical results of the present element are also compared with another solution, using a shell element from $Cosmos/M^{\ensuremath{\mathbb{S}}}$ which allows transverse shear deformation. The result of the displacement using ring element performs well even in the coarse mesh configurations. The small slenderness

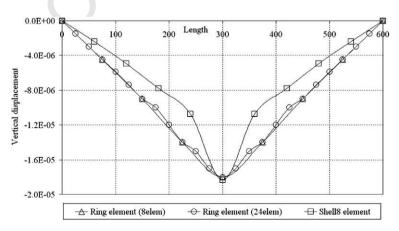


Fig. 4. Displacement with our formulation and Cosmos/M[®].

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E.M.M. Fonseca et al. | Mechanics Research Communications xxx (2006) xxx-xxx

Table 1		
Normalized displa	cement under the loa	ıd

Meshes		$w_{\text{FEM}}/w_{\text{exac}}$
$5 \times 5^{\mathrm{a}}$		0.5100
10×10^{a}		0.8300
20×20^{a}		0.9600
$4 \times 4^{\mathrm{b}}$		1.1769
8×8^{b}		0.9912
16×16^{b}		1.0059
Ring element (8 or 24 elements)		0.9864
Shell8 element Cosmos/M [®] (400 elements)		1.0060
 ^a Dvorkin and Bathe (1984). ^b Kouhia and Stenberg (2000). 	D1ep_Y	
	1.83580E-005	
	1.37680E-005	
	9.17890E-006	
	4.58950E-006	
	0.00000000	
	-4.5895E-006	

Fig. 5. Vertical displacement using Cosmos/M[®].

.1789E-006 .3768E-005 .8358E-005

173 ratio (t/r = 1/100) of the cylinder is chosen to demonstrate the capability of our ring element to overcome 174 shear and membrane-locking phenomena.

175 6. Case studied 2: ring deformation

The studied case of a ring deformation is shown in Fig. 6, representing a tubular steel ring structure submitted to a F-load along the length (L = 170 mm) and free ends. The tubular section presents an external diameter (DE = 165 mm) and an internal diameter (DI = 155 mm). The elasticity modulus is (E = 2.1 GPa) and the Poisson coefficient is equal to (v = 0.3). The theoretical value of the decrease diameter (d) on the top section and the lateral displacement at middle of section (u) can be obtained using the Castiglino's theorem:

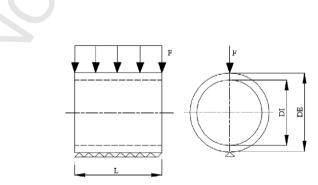


Fig. 6. Parameters for ring structure.

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(19)

E.M.M. Fonseca et al. | Mechanics Research Communications xxx (2006) xxx-xxx

$$d = -\left[\frac{\Pr^3(\pi^2 - 8)}{8\pi EI} + \frac{\Pr\pi}{8GAm} + \frac{\Pr\pi}{8EA}\right]$$
(18)

184
$$u = \frac{\Pr^3(4-\pi)}{4\pi EI} + \frac{\Pr}{4GAm} - \frac{\Pr}{4EA}$$

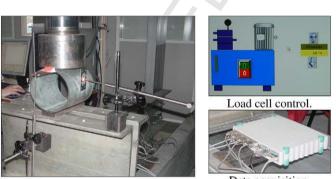
In this case: $I = Lt^3/12$, A = tL, G = E/2(1 + v) and m = 9/10 is the correction factor. The transversal dis-185 placement at the top and the lateral displacement will be analysed using the experimental method, analytical 186 187 solutions and numerical results obtain with the developed finite ring element. The experimental setup used is represented in Fig. 7. The loading system is implemented by a hydraulic jack and load cell, and both displace-188 189 ments are obtained with a data acquisition system using two LVDT's.

190 Fig. 8 represents the decrease displacement in diameter function of the F-Load, obtained with five exper-191 imental tests. The elastic part of the experimental results may be compared with the results of Eq. (15) or (18) 192 and numerical results from the finite ring element. Fig. 9 represents the lateral displacement measured for the 193 same five experimental tests. The elastic part of the experimental results may be compared with the results 194 provided by Eq. (19) and the numerical results obtained from the finite ring element.

195 The experimental tests present satisfactory results when compared with the numerical and analytical solution. Increasing loading test will gradually increase scattered data results. Numerical results approach the the-196 197 oretical curve, while the experimental results are positioned below this curve. The experimental results present

198 a non-linear trend for higher load values. Linear data can be useful to determine pipe stiffness, defined as resis-

199 tance to ring deflection.



LVDT's (max.linearity deviation 0.12%).

Data acquisition.

Fig. 7. Experimental assembly for ring structure.

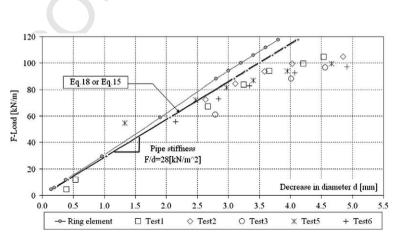


Fig. 8. Decrease displacement in diameter d.

8



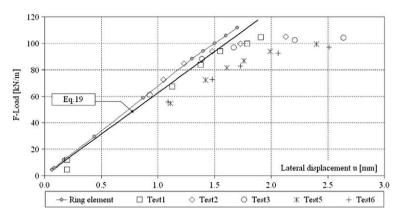


Fig. 9. Lateral displacement u.

200 7. Conclusion

201 A computer program based on the linear theoretical formulation described was developed for the analysis 202 and determination of the transverse displacement field along ring section of tubular pipes. To validate the element accuracy, numerical results were presented with a good agreement with other authors, analytical solu-203 tions and numerical analysis using Cosmos/M[®]. The presented solution has performed well even with 204 coarse element meshes, and appears as a simple and easy-to-handle alternative to the use of shell or beam finite 205 206 elements due the capability and effectiveness of this element. The presented formulation using a multi-nodal 207 element under symmetric conditions makes possible to determine a displacement field under shell surface for in-plane bending problems. A pre-processor for generate one-dimensional meshes with two sections (i and j) 208 per element is only necessary. The Cosmos/M[®] shell element presents eight nodes per element, and a more 209 refined three-dimensional mesh is necessary. The ring element exhibits excellent behaviour for all configura-210 211 tions in linear bending conditions like all shell elements, when using a small slenderness ratio between thick-212 ness and radius.

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