

A compliance feedforward scheme for a class of LTV motion systems

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Abstract—The implementation of lightweight high-performance motion systems in lithography applications imposes among others lower requirements on actuators, amplifiers, and cooling. However, the decreased stiffness of lightweight designs brings the effect of structural flexibilities to the fore especially when the so-called point of interest is not at a fixed location. This is for example the case when exposing a silicon wafer. To deal with structural flexibilities, a feedforward controller is proposed that combines two concepts: (a) continuous compliance compensation control and (b) snap feedforward control. Expanded to a subclass of LTV motion systems, the resulting controller compensates for the position-dependent and time-varying compliance of a flexible structure. The compliance function used will be derived using partial differential equations (PDE). The method is validated by simulation results.

I. INTRODUCTION

In the semiconductor industry, the focus on ever improving throughput, overlay, and imaging of exposed silicon wafers traditionally lead to more aggressive motion profiles, i.e. higher accelerations, and structural designs with higher stiffness and mass. The required forces to be applied during operation are therefore increasingly higher, which increases the demands on actuators, amplifiers, and cooling. The resulting force density and heat generation necessitated to accelerate the mass therefore becomes increasingly infeasible, which prompts for more flexible lightweight designs [14].

As a consequence, when the performance location changes with time which typically occurs during wafer exposure, the dynamics of the system are expressed differently due to the different contributions from structural modes at that specific location. This leads to the requirement of taking the time-varying aspect of the plant into account when calculating feedforward compensation forces, which are key in achieving position accuracy.

The traditional approach toward the positioning problem would be an acceleration (or mass) feedforward controller.

But this is not sufficient to compensate for the flexible dynamics. Alternatively, snap feedforward control [9] can only account for structural flexibilities to a certain extent, and cannot cope easily with time or parameter-varying dynamics. In [4] and [5] a collection of feedback and feedforward control methods are summarized, applicable to non-minimum phase and flexible motion systems. Among them, the work in [7] proposes a model inverse-based feedforward control signal for nonlinear plants, which requires the feedforward signal be known a priori and also pre-actuation, i.e. non-causal control effort. More recently, [8] addresses the problem of regulating the plant output when the disturbance is not known a priori, and [12] proposes a model-based non-causal feedforward scheme for double integrator-based Linear Time-Varying (LTV) systems, a class of systems also considered in the present work. In [6], work has been done on the discrete-time control of a stage, considering the plant as a Linear Time-Invariant (LTI) system. In [11] a feedforward method for flexible systems with time-varying performance locations is presented. The method utilizes a lifted feedforward (discrete-time) representation, however, it does not take the plant variation in-between the time-intervals into account. Spatial feedforward control [10] has been developed in order to prevent excitation of the structural modes of the positioning system. However this method uses over-actuation. Hence, the number of structural modes to be suppressed should equal the number of additional actuators.

Different from the current approaches in the literature, this work introduces a compliance compensating feedforward control scheme for motion systems with time-varying performance locations. That is, the same system class as addressed in [13]. The feedforward controller proposed here will be combined with the control objectives of classical snap feedforward, but for a time-varying performance location. Therefore this feedforward controller can account for position-dependent dynamics, which does not merely attempt at masking possible internal chuck deformations (ICD) caused during acceleration and deceleration, but instead generates appropriate control effort (force) to counteract such deformations. This renders the controller capable of tackling more arbitrary tracking set-points and control objectives than in [13]. Furthermore, as low as second-order set-points are compatible with this control scheme, which is the minimum required to be followed by a mass. Moreover, third order set-points produce a continuous control effort while fourth-order or higher set-points produce a smooth control effort which doesn't excite higher order dynamics as much as snap feedforward control. This feedforward controller does not require pre-actuation, i.e. is causal, which allows for set-

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points not given a priori. The spatially continuous dynamics of the plant used as an example in this work (a flexible Euler-Bernoulli beam) are derived from the partial differential equation (PDE) representation, and so is the position-dependent compliance function of the beam. The method is validated by continuous-time simulation.

The remainder of this paper is organized as follows. Section II introduces the problem statement. Section III proposes the novel feedforward control scheme, along with its mathematical derivation, notions of stability and performance. Section IV discusses the simulation environment and the results which validate the method. Finally, in Section V, some concluding remarks are given.

II. PROBLEM STATEMENT

During the production of chips, a silicon wafer is positioned atop the wafer stage of a lithographic system. A source emanating (extreme) ultraviolet (EUV) light passes through the reticle, which is part of the reticle stage, and which contains a blueprint of the integrated circuits (ICs) to be processed. Beyond the reticle, light passes an optical column with projection lenses before it exposes the photo-sensitive layers of the wafer's surface. An illustration of the wafer stage during exposure is shown in Fig. 1. Assuming that it is a lightweight structure, i.e. its dynamic behavior is substantially dependent on position, it follows that during exposure the time-varying performance location is subjected to position-dependent dynamics.

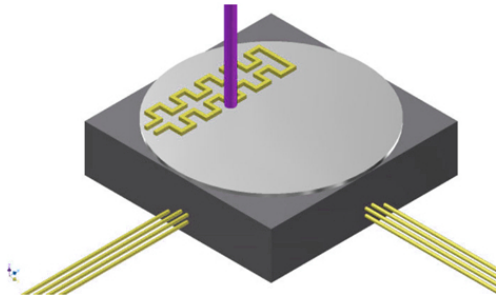


Fig. 1: Schematic representation of a wafer stage of a lithographic system, where the sensors lie at the edges of the stage, and where during exposure of the silicon wafer to the laser beam, the performance location changes over time.

Consider a straightforward snap feedforward control scheme [9], illustrated in Fig. 2, and an LTI double-integrator based motion system P , that is,

$$P(s) = \underbrace{\frac{1}{ms^2}}_{P_{rb}} + \underbrace{\sum_{n=1}^l \frac{b_{0,n}c_{0,n}}{m_n s^2 + d_n s + k_n}}_{P_{nrb}}, \quad (1)$$

where m is the total mass, $b_{0,n}$ and $c_{0,n}$ the input and output coefficients, m_n , d_n and k_n the modal mass, damping and stiffness respectively, P_{rb} the rigid body (RB) mode, $l \in \mathbb{N}^+$ the number of non-rigid body modes (NRB), and s the Laplace variable. Aiming to satisfy the control objective of perfectly compensating for the low-frequency properties

of the plant, i.e. mass and compliance, the snap feedforward controller follows from the principle of plant inversion, or

$$FF_{\text{snap}}(s) = \frac{U_{ff}(s)}{Y_d(s)} = K_{fa}s^2 + K_{fs}s^4 = ms^2 - m^2Cs^4, \quad (2)$$

with $Y_d(s)$, $U_{ff}(s)$ the Laplace transforms of the desired output and the control effort respectively, and C the compliance, which equals the DC contributions of all the NRB modes of P_{nrb} , that is,

$$C = \sum_{n=1}^l \frac{b_{0,n}c_{0,n}}{k_n}. \quad (3)$$

A straightforward attempt to extend this control scheme for

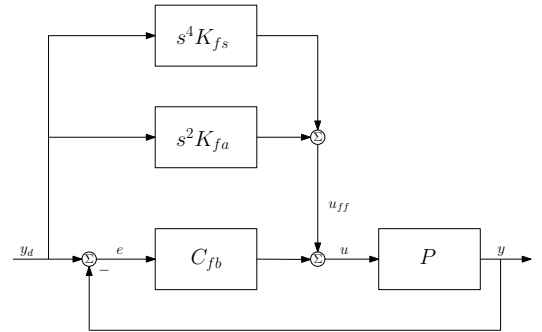


Fig. 2: Snap feedforward control scheme.

plants with a now time-varying compliance function, defined as $C(t)$, is to straightforwardly use the feedforward controller

$$FF_{\text{snap}}(p, t) = \frac{u_{ff}(t)}{y_d(t)} = mp^2 - m^2C(t)p^4, \quad (4)$$

where $p = d/dt$ is the time differential operator¹. It will be shown however that in general, feedforward control schemes developed for LTI systems such as (2) cannot be directly applied to position-dependent and time-varying systems.

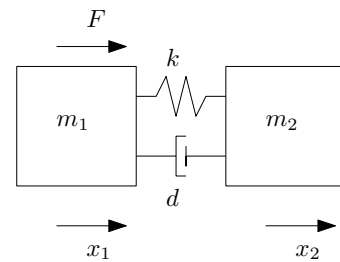


Fig. 3: Mass-spring-damper (MSD) system.

To see this, consider the example of a mass-spring-damper LTI Single-Input Multiple-Output (SIMO) system, shown in Fig. 3. Given no damping, i.e. $d = 0$, the non-collocated (from force F to displacement x_2) response of the system is given by

$$P_{\text{msd}}(s) = \frac{k}{s^2(m_1m_2s^2 + k(m_1 + m_2))}. \quad (5)$$

¹Use of p instead of s is required due to the time-varying nature of the controller, which does not admit a straightforward Laplace transform.

The total mass of the system equals $m_{\text{msd}} = m_1 + m_2$ and from (5) it can be found that the compliance is

$$C_{\text{msd}} = -\frac{m_1 m_2}{m_{\text{msd}}^2 k}. \quad (6)$$

As follows from (2) and (5), the snap feedforward controller

$$FF_{\text{snap-msd}}(s) = m_{\text{msd}} s^2 - m_{\text{msd}}^2 C_{\text{msd}} s^4 \quad (7)$$

perfectly accounts for the plant dynamics in the sense that $P_{\text{msd}}(s) FF_{\text{snap-msd}}(s) = 1$.

Let us now consider by means of example a smooth time-varying stiffness, i.e. $k(t) \in \mathcal{C}^2$. This introduces a time-varying compliance in (5), while the mass of the system remains the same. The equations of motion are given by

$$\begin{aligned} F(t) - k(t)(x_1(t) - x_2(t)) &= m_1 \ddot{x}_1(t) \\ k(t)(x_1(t) - x_2(t)) &= m_2 \ddot{x}_2(t). \end{aligned} \quad (8)$$

Assuming a desired trajectory for the second mass, $y_d(t)$, eliminating $x_1(t)$ from (8), the force $F_{\text{msd}}(t)$ required to account for the plant dynamics is given as follows,

$$\begin{aligned} F_{\text{msd}}(t) &= m_{\text{msd}} \ddot{y}_d(t) + \frac{2\dot{k}^2(t) - k(t)\ddot{k}(t)}{k^3(t)} m_1 m_2 \ddot{y}_d(t) \\ &\quad - \frac{\dot{k}(t)}{k^2(t)} 2m_1 m_2 y_d(t)^{(3)} + \frac{m_1 m_2}{k(t)} y_d(t)^{(4)}. \end{aligned} \quad (9)$$

The first and last term of (9) correspond to the mass and snap terms in (7), however as it can be seen there are two additional terms required in order to account for the time-varying plant dynamics. In (9), the first two time derivatives of the time-varying stiffness appear, along with the third time-derivative of $y_d(t)$. Therefore, the structure as suggested by (4) is inherently too limited to cope with the time-varying stiffness.

To account for time-varying compliance, we first adopt the more general (possibly infinite-dimensional) motion system P , described by the LTV state-space model,

$$\begin{aligned} \dot{x}(t) &= A'x(t) + B'u(t), \\ y(t) &= C'(t)x(t) + D'(t)u(t), \end{aligned} \quad (10)$$

where $A' \in \mathbb{R}^{n \times n}$, $B' \in \mathbb{R}^{n \times 1}$, $C'(t) \in \mathbb{R}^{1 \times n}$, $D'(t) \in \mathbb{R}^{1 \times 1}$. The output matrices can be time-dependent, due to a time-varying point-of-interest.

In terms of (1), the symbolic transfer function from u to y of (10) can be written as

$$P(p, t) = \frac{y(p, t)}{u(p, t)} = \frac{1}{mp^2} + \sum_{n=1}^l \frac{b_{0,n} c_{0,n}(t)}{m_n p^2 + d_n p + k_n}. \quad (11)$$

III. COMPLIANCE FEEDFORWARD CONTROL FOR A CLASS OF LTV MOTION SYSTEMS

The proposed control scheme is illustrated in Fig. 4 and consists of the following components,

- 1) *Desired output signal*; the signal y_d can be a second-order or higher setpoint, which typically has a continuous first derivative (velocity), that is $y_d \in \mathcal{C}^1(0, \infty)$;

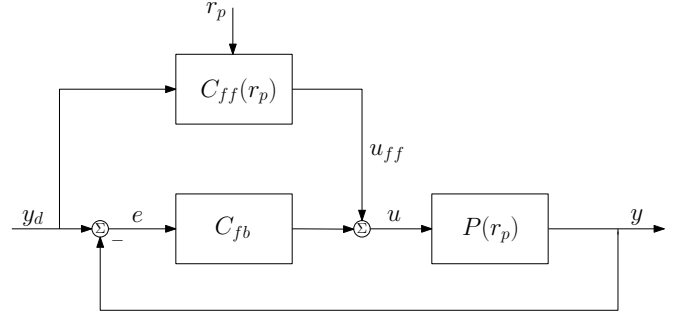


Fig. 4: Block diagram of the proposed LTV feedforward control scheme.

note that for the lithographic industry the scanning interval of constant velocity (in-between the dashed lines) is the interval in which the tracking error is required to be sufficiently small;

- 2) *Performance location function*; in the case of position-dependent dynamics, a real-valued function $r_p = r_p(t)$ is required, which indicates the point of interest (POI) as a function of time $t \in \mathbb{R}$; for a distributed parameter system, r_p is continuously differentiable at least once, i.e. $r_p \in \mathcal{C}^1$; note that in some applications $r_p = y_d$, e.g. during wafer stage operation, in the x and y directions, which are parallel to the stage;
- 3) *Plant*; the plant $P(r_p)$ is a Single-Input Single-Output (SISO) flexible motion system, defined by (10), or "equivalently" by (11); its performance location can be static or time-varying in nature indicated by r_p ;
- 4) *Feedback controller*; the LTI feedback controller C_{fb} acts on the error e between the setpoint and the plant output, i.e. $e = y_d - y$;
- 5) *Feedforward controller*; the feedforward controller C_{ff} accounts for the mass and time-varying compliance of P ; it connects the setpoint y_d to the output signal u_{ff} .

A. Controller derivation

Consider a system as in (1). A low-frequency approximation can be derived by preserving the RB mode, i.e. $P_{rb}(s)$, plus the total compliance, the latter is given in in (3). Assuming the input and output signals admit Laplace transformations $U(s)$, $Y(s)$, the low-frequency approximation is given by

$$\begin{aligned} P_{0,1}(s) &= \frac{Y(s)}{U(s)} = P_{rb}(s) + C \\ &= \frac{1}{ms^2} + \sum_{n=1}^l \frac{b_{0,n} c_{0,n}}{k_n}. \end{aligned} \quad (12)$$

Given a time-varying compliance function $C = C(t)$, the input-output relation in time-domain can be represented by

$$\begin{aligned} P_{0,2}(p) &= \frac{y(t)}{u(t)} = \frac{1}{mp^2} + C(t) \Leftrightarrow \\ y &= \frac{1}{m} \int \int^t u(\tau) d\tau + C(t)u(t) \Leftrightarrow \\ m\ddot{y}(t) &= u(t) + m(C(t)\ddot{u}(t)). \end{aligned} \quad (13)$$

The plant in (13) provides an accurate plant approximation in the low frequency interval only. Plant inversion of (13) would (if possible) produce a model-based feedforward controller which can account for low-frequency output disturbances, i.e. tracking setpoints commonly used in such systems. Assuming y_d a desired output trajectory and u_{ff} the desired control input, yields the feedforward control differential equation

$$\ddot{u}_{ff}(t) = -\frac{2\dot{C}(t)}{C(t)}\dot{u}_{ff}(t) - \frac{1+m\ddot{C}(t)}{mC(t)}u_{ff}(t) + \frac{1}{C(t)}\ddot{y}_d. \quad (14)$$

As it can be seen, a critical issue in solving (14) is the division by the compliance function $C(t)$. This results in the possibility of division by zero, namely if the compliance function becomes zero, as found in [13]. Furthermore, even if $C(t)$ is non-zero and slowly varying, it can be shown that the solution to (14) is only marginally stable if $C(t) < 0$, causing undesirable undamped oscillations.

To resolve the possible non-feasibility of the inversion, a plant modification is introduced which allows for inversion and simultaneously maintains a fairly accurate low frequency plant description. This is achieved by pre-filtering the time-varying compliance function $C(t)$ with a second-order low-pass filter, as shown in Fig. 5. Note that a second (or arbitrarily higher) order low-pass filter is required to prevent division by zero when solving for the signal $v(t)$. The plant

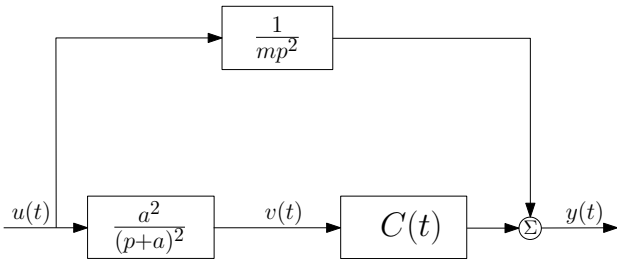


Fig. 5: Plant approximation $P_2(p)$ underpinning the proposed feedforward controller, consisting of a RB mode, and second-order low-pass filter cascaded with a time-varying gain $C(t)$ which equals the time-varying compliance of the system.

to be inverted as illustrated in Fig. 5 is governed by the equations

$$y(t) = C(t)v(t) + \frac{1}{m} \iint^t u(\tau) d\tau \Leftrightarrow \quad (15)$$

$$m\ddot{y}(t) = u(t) + m(C(t)\ddot{v}(t)),$$

and

$$a^2u(t) = \ddot{v}(t) + 2a\dot{v}(t) + a^2v(t), \quad (16)$$

where $a > 0$ denotes the cut-off frequency of the low-pass filter. Given a desired trajectory $y_d(t) \in \mathcal{C}^1$, solving (16) with respect to $v(t)$ and after substitution in (15), gives the

differential equation,

$$\ddot{v}(t) = -\underbrace{\frac{2a(am\dot{C}(t)+1)}{a^2mC(t)+1}}_{\xi_1(t)}\dot{v}(t) - \underbrace{\frac{a^2(m\ddot{C}(t)+1)}{a^2mC(t)+1}}_{\xi_2(t)}v(t) + \underbrace{\frac{a^2m}{a^2mC(t)+1}}_{\xi_3(t)}\ddot{y}_d(t). \quad (17)$$

Since there is no explicit solution to (17), the signals $v(t)$, $\dot{v}(t)$, and $\ddot{v}(t)$ for a given y_d are obtained through numerical integration. Utilizing (16) then gives the control input by

$$u_{ff}(t) = \underbrace{\frac{1}{a^2}}_{\mu_1}\ddot{v}(t) + \underbrace{\frac{2}{a}}_{\mu_2}\dot{v}(t) + v(t). \quad (18)$$

In state-space form, the feedforward controller can be written and implemented as

$$\dot{x}(t) = \underbrace{\begin{bmatrix} 0 & 1 \\ \xi_2(t) & \xi_1(t) \end{bmatrix}}_{A_{FF}(t)}x(t) + \underbrace{\begin{bmatrix} 0 \\ \xi_3(t) \end{bmatrix}}_{B_{FF}(t)}\ddot{y}_d(t),$$

$$u_{ff}(t) = \underbrace{[\mu_1\xi_2(t)+1 \quad \mu_1\xi_1(t)+\mu_2]}_{C_{FF}(t)}x(t) + \underbrace{[\mu_1\xi_3(t)]}_{D_{FF}(t)}\ddot{y}_d(t), \quad (19)$$

where $x(t) = [x_1(t) \ x_2(t)]^T = [v(t) \ \dot{v}(t)]^T$. Naturally, in the context of this control scheme, time-dependency of the compliance function and its derivatives is introduced by the POI function $r_p(t)$, thus yielding $C(r_p(t))$, $\dot{C}(r_p(t))$, and $\ddot{C}(r_p(t))$. A block diagram of this feedforward control scheme is depicted in Fig. 6. Note that the second time-derivative of y_d is often known a priori.

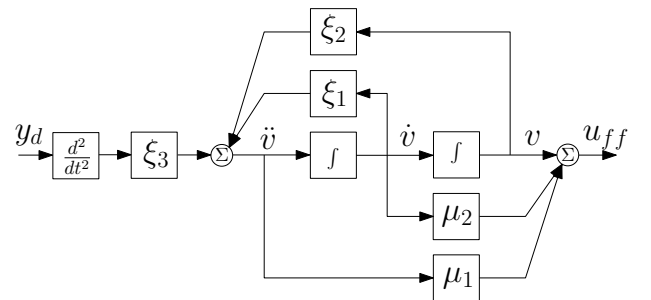


Fig. 6: Compliance compensating feedforward scheme.

B. Feasibility analysis

In terms of feasibility, as previously mentioned, division by zero mid-experiment should be excluded, otherwise (17) cannot have a solution. Moreover, for LTI cases, (17) needs to be asymptotically stable in order for the feedforward controller to be bounded-input, bounded-output (BIBO) stable.

Following this reasoning, a can be chosen as follows,

$$\begin{cases} a \in \mathbb{R}^+ & \text{if } C(t) > 0 \\ a < \frac{1}{\sqrt{-MC(t)}} & \text{if } C(t) < 0 \end{cases} \quad (20)$$

which also translates to keeping the eigenvalues of $A_{FF}(t)$ inside the left-half (complex) plane (LHP).

IV. NUMERICAL RESULTS

The simulation results given in this section illustrate the compliance compensating feedforward control scheme when applied to a beam system with a time-varying performance location. For the plant, an Euler-Bernoulli beam is considered [1][2], which is illustrated and specified in Fig. 7. These specifications induce a frequency response that has similarities to a typical wafer stage system. That is with respect to both the RB and the first NRB mode's frequency and magnitude.

In frequency domain, the PDE describing this beam is given by,

$$\frac{d^4}{dr_p^4} Y(s, r_p) + \frac{2325}{3s + 5 \cdot 10^8} s^2 Y(s, r_p) = 0. \quad (21)$$

The solution to (21) yields the position-dependent transfer function of the beam, $G_d(s, r_p)$ [13]. This infinite-dimensional transfer function can be expanded via modal approximation into its RB and infinite NRB modes, bringing it to the form that closely matches (11) but still being of infinite order. For simulation purposes, a finite-order model can be obtained through truncation. In this simulation model the RB and the first NRB mode of the beam will be included. It should however, be mentioned that the modeling strategy remains valid if more than one NRB modes are considered.

Given a pole λ_k , we define $a = \text{Re}(\lambda_k)$, $b = \text{Im}(\lambda_k)$, $c(r_p) = \text{Re}(\text{Res}(\lambda_k, r_p))$, $d(r_p) = \text{Im}(\text{Res}(\lambda_k, r_p))$, where $\text{Res}(\lambda_k, r_p)$ is the Cauchy residue of the pole λ_k (see [3]), which is dependent on the point of interest function r_p . The first NRB mode of the beam is given by

$$\begin{aligned} G_1(s, r_p) &= \frac{\text{Res}(\lambda_1, r_p)}{s - \lambda_1} + \frac{\text{Res}(\lambda_1^*, r_p)}{s - \lambda_1^*} \\ &= \frac{2(c(r_p)s - ac(r_p) - bd(r_p))}{s^2 - 2as + a^2 + b^2}. \end{aligned} \quad (22)$$

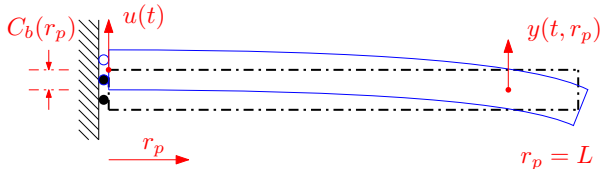


Fig. 7: Vertically-moving cantilever Euler-Bernoulli beam, where $u(t)$ the actuation force, $y(t, r_p)$ the displacement at the point-of-interest r_p , and $C_b(r_p)$ the position-dependent compliance; length $L = 0.6$ [m], cross-sectional area $A = h^2 = 10^{-4}$ [m²], mass density $\rho = 7.75 \cdot 10^3$ [kg/m³], Young's modulus $E = 2 \cdot 10^3$ [kg/(m · sec²)], second moment of area $I = h^4/12 = 10^{-4}/12$ [m⁴], Kelvin-Voigt damping $c_d = 10^{-3}$.

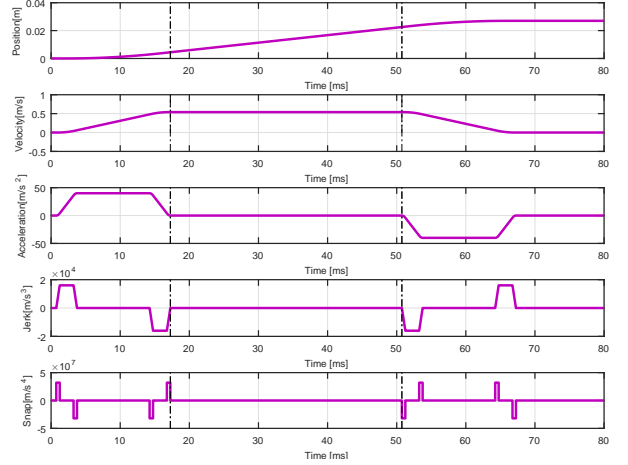


Fig. 9: Fourth-order reference setpoint y_d . The black dashed lines enclose the critical scanning interval (constant velocity).

Naturally the frequency response along the beam changes only through the residue, which affects only the zeros. The poles of the structure remain unchanged, regardless of the point of interest $r_p = r_p(t)$. From (22) it can be seen that the compliance of the single flexible mode is given by,

$$C_{\lambda_1}(r_p) = \lim_{s \rightarrow 0} G_1(s, r_p) = \frac{-2(ac(r_p) + bd(r_p))}{a^2 + b^2}. \quad (23)$$

Generally, the compliance function of the beam does not equal the compliance of its first NRB mode. Therefore an adjustment in the compliance function is required to correctly match the compliance to the infinite-dimensional beam system, similar to the approximation in (12), however now one NRB mode is preserved. The simulation model is given as follows,

$$G_s(s, r_p) = G_{\text{rb}}(s) + G_1(s, r_p) - C_{\lambda_1}(r_p) + C_b(r_p), \quad (24)$$

where the RB mode is given by

$$G_{\text{rb}}(s) = \frac{1}{\rho A L s^2}, \quad (25)$$

and

$$C_b(r_p) = \frac{6L^4 - 30L^2 r_p^2 + 20L r_p^3 - 5r_p^4}{120 E I L}, \quad (26)$$

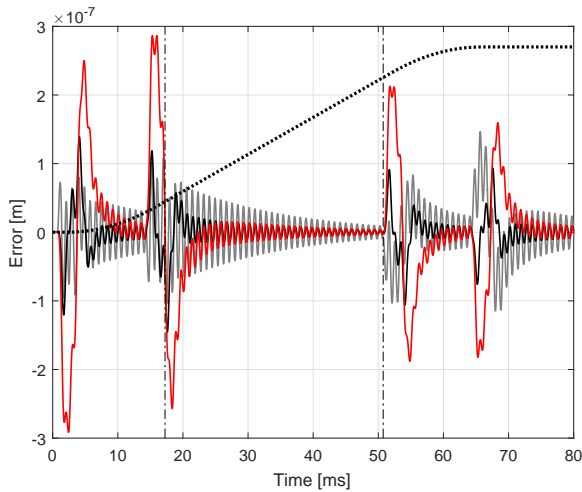
is the compliance function of the beam, see [13].

Simulations were performed in continuous time. The fourth-order trajectory y_d in Fig. 9 is used as the tracking setpoint, in order to facilitate a comparison with snap feedforward. Two example POI functions r_p are chosen, one staying in the area of the beam where $C(r_p) > 0$, and the other one also venturing toward the area where $C(r_p) < 0$:

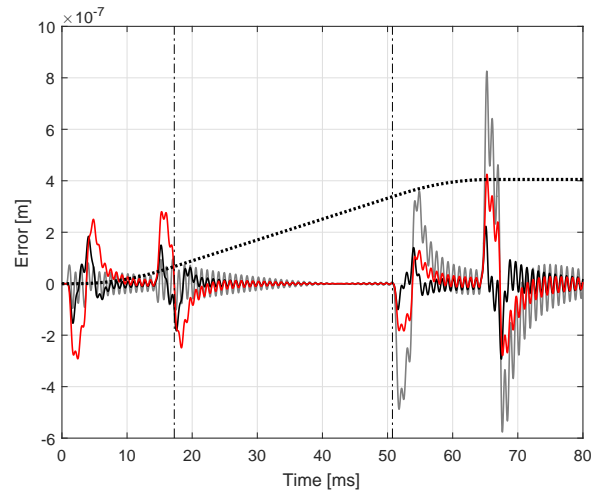
$$r_{p_1}(t) = 0.1(1 - \cos(12.5\pi t)), \quad (27)$$

and

$$r_{p_2}(t) = 0.3(1 - \cos(12.5\pi t)). \quad (28)$$



(a) POI function $r_{p1}(t)$



(b) POI function $r_{p2}(t)$

Fig. 8: Simulation with position-dependent (black), (fixed) snap (gray) and (fixed) acceleration feedforward (red) for the beam with a time-varying performance location, in closed loop; the scanning interval is enclosed by the vertical dashed lines, while the (scaled) setpoint is drawn in dotted curves.

The feedback controller, which robustly stabilizes the beam in the case that one or two NRB modes are included in the model, irrespective of the position r_p considered, is given by

$$C_{fb}(s) = C_{PID}(s)C_{1st}(s)N_1(s)N_2(s)N_s(s), \quad (29)$$

where

$$C_{PID}(s) = \frac{1.64 \cdot 10^8 s^2 + 6.48 \cdot 10^{10} s + 9.93 \cdot 10^{11}}{s}, \quad (30)$$

$$C_{1st}(s) = \frac{1}{s + 2.088 \cdot 10^5}, \quad (31)$$

$$N_1(s) = \frac{s^2 + 0.3184 s + 5.191 \cdot 10^7}{s^2 + 1.297 \cdot 10^4 s + 5.191 \cdot 10^7}, \quad (32)$$

$$N_2(s) = \frac{s^2 + 8.823 s + 1.516 \cdot 10^9}{s^2 + 7.008 \cdot 10^4 s + 1.516 \cdot 10^9}, \quad (33)$$

and

$$N_s(s) = \frac{6.25 \cdot 10^{-6} s^2 + 7.226 s + 2.088 \cdot 10^8}{s^2 + 2890 s + 2.088 \cdot 10^8}. \quad (34)$$

For the case r_{p1} in (27), a cut-off frequency $a_1 = 7400 [rad/sec]$ was chosen, and for r_{p2} in (28) the cut-off frequency is $a_2 = 4700 [rad/sec]$.

The simulation results are illustrated in Fig. 8, which show the tracking error in closed loop using the proposed feedforward controller (black), a fixed acceleration feedforward (red), and a position-independent (or fixed) snap feedforward controller (gray), which was tuned to account for the compliance at the base of the beam, i.e. $C_b(0)$ in (26). It can be seen that the proposed feedforward controller results in a smaller error when compared both to snap feedforward and acceleration feedforward control. Superiority over acceleration feedforward control can be understood in terms of the more complex (position-dependent) model used to calculate the proposed feedforward controller, which can

account for compliant part of the beam system and for which acceleration feedforward is unable to compensate. Snap feedforward control, while correctly tuned for the base of the beam, starts to deteriorate the error, observed for both POI functions. Moreover, the discontinuous, step-like control inputs produced by the snap feedforward controller lead to higher oscillations of the system, for which the smooth signal produced by the proposed LTV controller does not suffer from.

V. CONCLUSIONS AND REMARKS

Inspired by the snap feedforward [9] and the continuous compliance compensation scheme [13], on which an accurate compliance function was found, this paper further exploits this information. More precisely, a model-based compliance compensating feedforward controller is obtained for flexible structures with a time-varying performance location. A low-frequency approximation of the plant was used to produce the controller, modified sufficiently in order to be invertible, while its inverse is simultaneously stable. Simulations using an Euler-Bernoulli beam's PDE representation, a fourth-order setpoint trajectory and two arbitrary point-of-interest time functions indicate improved performance in terms of the closed loop tracking error, regarding the compliant part of the LTV system. In terms of wafer stage control, this may help in reducing internal chuck deformations otherwise resulting in the controlled stage dynamics.

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