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# A Method for the Measurement of Shielding Effectiveness of Planar Samples Requiring no Sample Edge Preparation or Contact

Andrew C Marvin, *Senior Member IEEE*, Linda Dawson, Ian Flintoft, *Member IEEE*  
and John Dawson, *Member IEEE*

**Abstract**—A method is presented for the measurement of shielding effectiveness of planar materials with non-conducting surfaces such as carbon fibre composites. The method overcomes edge termination problems with such materials by absorbing edge-diffracted energy. A dynamic range of up to 100 dB has been demonstrated over a frequency range of 1 to 8.5 GHz depending on the size of the sample under test. Comparison with ASTM D4935 and nested reverberation measurements of shielding effectiveness shows good agreement.

**Index Terms**—Shielding Effectiveness, Shielding Measurements

## I. INTRODUCTION

The measurement of the shielding effectiveness (SE) of a planar material sample is required in order to predict the suitability of the material to form an enclosed electromagnetic shield. A typical shielded enclosure has an ultimate shielding performance that is limited by the shielding performance of its structural features such as apertures penetrations and joints. For a high quality shielded enclosure the SE expressed logarithmically may exceed 100 dB when the enclosure is first commissioned. Typical equipment enclosures, or enclosures that have a secondary shielding requirement, such as vehicle bodies, may have SE values in the range 10 dB to 80 dB. This lower performance is a consequence of the structural features and the consequent requirement for the shielding performance of the structural material is lower, typically no more than 90 dB. Any sheet metal used as a structural material will have a SE considerably in excess of 90 dB. Structural materials formed from metallised plastic or other non-conducting substrates with an internal conducting component, such as carbon fibre re-enforced composite (CFC), may have SE values in the range below 90 dB and thus require measurement.

Such materials are conventionally measured in test systems that require a planar sample of the material to be placed across an aperture or within the cross-section of a transmission line. Examples are the use of nested reverberation chambers (NRC) [1,2], nested anechoic chambers and coaxial TEM waveguides, typically the standard ASTM cell [3]. In each of these systems the measured reduction of transmitted electromagnetic energy (insertion loss) through the aperture or cell with the sample present is compared to that without the sample present and the data is processed to estimate the SE of the sample material. With the sample present the energy flow is through the sample and possibly around the edge of the sample if a good conducting contact between the outer edge of the sample and the inner edge of the aperture/cell is not maintained along the entire sample perimeter. The edge contact requirement has been overcome in some coaxial systems [4,5] but can be a major source of measurement error in other systems that require contact with a buried conducting material or samples with conductor on one side only. These problems are particularly acute as the frequency of the measurement increases and especially in the microwave frequency range where the structural dimensions of the sample material features, for example the weave in the fabric of a composite material, become comparable to the wavelength.

## II. PROPOSED SYSTEM

In this paper we describe a measurement system that overcomes the edge contact problems by eliminating its requirement. Such a technique has been proposed before, where the planar sample is illuminated from one side and the transmitted energy is received on the other side along with edge diffracted energy and energy reflected from surrounding objects [6]. These latter unwanted artefacts, which follow a longer propagation path, are time gated out to leave the energy transmitted through the sample. The measurement dynamic range of this technique is limited to around 50 dB to 60 dB by the imperfections in the time gating process that results in some residual unwanted energy. The system is shown in outline diagrammatic form in Fig. 1.

The approach taken here is to accept the edge diffracted energy but to absorb it after diffraction in an absorber assembly fabricated from proprietary carbon loaded polyurethane foam [7]. Fig. 2 shows the principle of the new system where the absorber reduces the effect of the diffracted and reflected energy. Absorber has previously been used to attenuate diffracted energy in specialized gasket shielding measurement systems [8], however the effectiveness and optimization of the absorber has not been addressed. In the work reported here the absorber is explicitly used for the termination of planar samples and the measurements system is optimized for composite materials.

The prototype system is shown in Fig. 3. The dimensions of the prototype system are 600 mm by 600 mm by 330 mm with the aperture in the absorber being 140 mm by 150 mm, chosen to fit the dimensions of the ridged waveguide horn used as the receiving antenna (embedded in the bottom of the cavity). In practice larger apertures could be used. The absorber is enclosed in a metal box and the receive antenna signal is fed to a port on one side of the box using semi-rigid cable embedded in the absorber.

### III. PREDICTED PERFORMANCE OF THE SYSTEM

The absorber structure was modelled in order to predict and optimise its performance. The dynamic range of the system depends on the attenuation of the incident energy through the absorber underneath the sample and is therefore determined by the sample size and the attenuation constant of the absorber. Absorbers are commercially available with a variety of different attenuation levels and the question of which type provides the optimal performance arises when practical limitations of the absorber system are taken into account. The model was also used to evaluate the minimum size of sample required for a given dynamic range and to verify that the system was actually measuring the SE of the samples.

The Transmission Line Matrix (TLM) algorithm with a node spacing of 5 mm was used to construct a three-dimensional model of the absorber system depicted in Fig. 2 [9]. The model, shown in Fig. 4, included the metal outer box, the absorber, the cavity and the sample. A symmetry plane through the centre of the system was utilised to improve the computational efficiency. The model did not include representations of the horn antennas; instead a plane wave was used for the excitation and the field at the centre of the cavity was observed to estimate the transmission through the sample. The frequency dependent conductivity and permittivity of the absorber were taken from the manufacturers data sheets [7]. The node spacing of 5 mm is adequate to give a resolution of one-tenth of a wavelength up to 6 GHz; note that the relative permittivity of the absorber, which is typically around 10 at 1 GHz, falls rapidly with frequency. The sample was modelled using TLM thin boundaries, described in more detail below.

Fig. 5 shows the results of the model's estimate of the dynamic range of the system obtained by comparing the electric field computed in the antenna cavity with no sample present to the electric field at the sample location with a 300 mm square perfect electric conductor (PEC) sheet used in place of the material sample. The PEC sheet has effectively infinite shielding effectiveness in the model so the predicted SE in Fig. 5 indicates the energy diffracted around the sample that is not blocked by the absorber. The points are the computed result and the line is the measured result. The noise on the measured result above 4 GHz is the equipment noise floor. The dynamic range of the system is greater for larger samples at lower frequencies due to the increasing attenuation of energy coupled around the edge of the sample.

Verification that the system was indeed measuring SE was obtained by modelling samples with different characteristics using a sub-scale partially reflecting/transmitting TLM boundary with defined transmission coefficients [10]. This is the TLM equivalent of impedance boundary models in FDTD, which are often used to simulate composite materials [11]. The TLM boundary model is parameterised by the scattering matrix of the transverse field components at the boundary. To show more explicitly the effect of the sample size and absorber characteristics on the dynamic range of the system a range of "fictitious" samples with frequency independent transmission coefficients ranging from 20 dB to 80 dB were modelled (the reflection coefficient is chosen to ensure energy conservation and provide a little absorption in the boundary). The transmission coefficient defined for the boundary is implicitly the plane-wave infinite planar SE of the "fictitious" sample.

Fig. 6 shows the results of this simulation for samples with SE values in the range 20 dB to 80 dB. In these simulations a frequency independent model of the absorber was used for computational efficiency, taking the manufacturers material parameters at 3 GHz. The ratio of the predicted field value in the cavity below the sample to the predicted value in the cavity with the sample removed was evaluated.

It can be seen that the predicted SE values are very close to the infinite plane-wave SE values of the computational samples. The low frequency fall-off of the predicted SE value is due to energy leakage around the edge of the sample. The ultimate predicted dynamic range for a sample of this size is that obtained for the PEC sheet.

The energy leakage around the sample edge is reduced by the presence of the absorber, however the SE obtained in the prototype system was lower than predicted. This was found to be a function of the flatness of the sample and the absorber. Any air gap between the sample and absorber allowed energy to be guided between them. In the simulations described above the interface between the sample and the absorber was idealised, with the absorber in direct contact with the sample over the entire area of overlap between the sample and absorber. In the practical system direct contact cannot be achieved over the whole overlap region so the dynamic range is further limited. Also, many composite samples are likely to be layered in structure with

the buried conducting layer having a dielectric layer on either side. Thus a non-conducting gap is inevitable between the absorber and the conducting layer of the sample. The effect of this gap was modelled using a quasi-two-dimensional slice model of the absorber system as shown in Fig. 7.

The model is one cell deep with perfect magnetic conductor (PMC) boundaries on the front and back and PEC boundaries at the top and bottom. The node spacing is 0.5 mm. A plane wave is launched towards the gap, which is taken to be uniform in size, and the power flow at the far side of the gap is evaluated. The wave is terminated by absorbing boundary conditions (ABC) on either side of the problem space. A range of absorber types with increasing carbon content is available, listed as LS14 to LS30 [7]. The model evaluates the total power flow through the plane shown in the figure with and without the absorber present. Since this plane is the only entry point in the model for external energy into the inner cavity the ratio of these quantities, called the isolation, is a figure of merit indicating the effectiveness of the absorber in attenuating the diffractive wave in the full absorber system.

The predicted isolation values offered by the absorber with different gap sizes at a frequency of 3 GHz are listed in Table 1. The bold figures show the optimum absorber type for the gap size listed. The LS22 type absorber was used in the final system; it is optimal for gap sizes greater than 2 mm and near optimal for smaller gaps.

The gap between the conducting layer of the sample and the absorber can be either air due to imperfections in absorber or sample flatness or dielectric material forming the outer layer of the sample. Fig. 8 shows the effect of differing relative permittivity of the material in a 1mm gap compared to a system with no gap. A number of effects are present in this simulation.

At higher relative permittivity values the electrical depth of the gap is larger indicating greater power flow through the gap. However the higher permittivity values also mean that the lossy waveguide formed by the gap with absorber on one side is electrically longer and has greater attenuation. The presence of a permittivity change at each end of the gap also introduces an impedance mismatch. The overall effect of an increase in permittivity in the gap is an increase in isolation at frequencies below 4 GHz and a reduction in isolation at higher frequencies.

The presence of the absorber around the sample results in the electromagnetic field structure being different from a plane electromagnetic wave. The absorber forms a rectangular waveguide with “soft walls” which has a cut-off frequency of about 1 GHz for the lowest mode. Fig 9. shows the variation of the x and y polarised electric field across the plane of the sample at 3 GHz predicted by the full TLM model. The figures show only half the area of the cavity in the absorber since the model has a symmetry plane along the y-axis. The system was illuminated by a y-polarised electric field of magnitude 1 V/m (0 dB V/m). The field uniformity of the dominant y polarisation is about 5 dB at this frequency. The co-polar (x-polarised) electric field is about 20 dB lower than the dominant polarisation. The wave-impedance of the field in the plane of the sample, i.e. the local ratio of the y-polarised electric field to the x-polarised magnetic field varies, from 340  $\Omega$  to 460  $\Omega$  across the sample at 3 GHz. The wave impedance varies with frequency, being somewhat larger (typically 800  $\Omega$ ) at 1 GHz near the cut-off of the lowest mode, and approaching an average of the free-space impedance of 377  $\Omega$  at higher frequencies. The electromagnetic field that illuminates the sample is therefore not a pure plane-wave and the correlation of the SE measured by the system with other techniques is investigated by measurements below.

#### IV. FINAL SYSTEM CONFIGURATION

The final system is shown in cross-section in Fig. 10 and photographs of the system are shown in Figs. 11 and 12. The receiving antenna embedded in the absorber and the antenna above the sample are commercial 1 GHz to 18 GHz ridged waveguide horns. An expanded polystyrene block is used to hold the antenna above the sample. Enhanced reduction of the edge-diffracted energy is obtained by placing a block of layered absorber around the polystyrene block as shown. This also helps to keep thin samples flat and reduces uncontrolled air gaps. The sample shown in Fig. 11 is circular to allow measurements to be made at arbitrary angles of polarization relative to the sample structure.

#### V. SYSTEM CALIBRATION

The measurement of SE in this system relies on the suppression of reflected and edge diffracted energy relative to that penetrating directly through the sample. The frequency range of operation of the prototype system is from 1 GHz (limited by the efficiency of the absorber and sample size), to 8.5 GHz (limited by the instrumentation available). A vector network analyser was used for the measurements; simplistically the SE may be defined as the insertion of loss through the system when the sample is put in place:

$$SE_{\text{measured}} = \frac{S_{21}^{\text{nosample}}}{S_{21}^{\text{withsample}}} . \quad (1)$$

Samples exhibiting significant shielding due to the presence of one or more conducting layers within the sample have an

electric field reflection coefficient close to -1. A standing wave is therefore set up in the cavity formed in the absorber below the sample. Thus a simple measurement of SE performed by taking the ratio of the received field with and without the sample present has a superimposed ripple.

In order to overcome the ripple effect a reference sample technique was implemented using a sample of known SE, in this case a brass sheet of thickness 300  $\mu\text{m}$  with a square array of circular holes of diameter 3 mm and spacing 10 mm etched into it. The SE of this sample was measured using an ASTM D4935 cell [3] and a NRC using the NIST corrected calibration technique [1,2]. In either measurement case, the metal reference sample presented no contact problems with the sample holder and has no anisotropy.

The SE of the reference sample was also computed using TLM assuming that the sample is a thin PEC sheet and then also accounting for the thickness of the sample material. In this case a single cell of the array was modelled in a problem space bounded with appropriate PEC and PMC walls to simulate an infinite array by reflection. The TLM mesh size was 50  $\mu\text{m}$  in order to enable the model to account for the finite thickness of the sheet.

The results of these measurements and computation are shown in Fig. 13. Note that the ASTM cell only works in the frequency range below 1.5 GHz due to over-moding in the cell and that the NRC measurement is limited to frequencies above 1 GHz due to minimum mode density requirements in the smaller of the nested chambers. The measured results are in good agreement in the overlap frequency range and the computed result accounting for the sample thickness provides the best estimate of the SE. The actual SE of the reference sample ( $SE_{\text{ref}}$ ) was assumed to be that of the ASTM cell measurement in its frequency range and a straight-line projection (in the dB scale of Fig. 13) of the ASTM result in the higher frequency range. This straight-line projection overlays the noisier reverberation chamber measurement in the higher frequency range.

The results of measuring the reference sample using the absorber system are shown in Fig. 14. It can be seen that the agreement between the absorber system and the ASTM cell is good and that between the measurements and the simulation is also good, even though the superimposed ripples due to the reflections in the absorber cavity are present in this uncorrected measurement. The agreement between the absorber system and the reverberation chamber shown in Fig. 13 is also good.

Using the reference sample calibration technique the SE of the sample-under-test,  $SE_{\text{sample}}$ , is calculated from equation (2):

$$SE_{\text{sample}} = SE_{\text{ref}} \frac{S_{21}^{\text{ref}}}{S_{21}^{\text{sample}}} . \quad (2)$$

$SE_{\text{ref}}$  is the assumed known SE of the reference sample,  $SE_{21}^{\text{ref}}$  is the measured transmission through the reference sample and  $SE_{21}^{\text{sample}}$  is the measured transmission through the sample-under-test.

Fig. 15 shows this technique applied to a sample comprising a second 300  $\mu\text{m}$  thick brass sheet with a square array of circular holes of diameter 1.5 mm and spacing 5mm etched into it; the array spacing and hole size of this plate are thus half those of the reference sample. For comparison, the ASTM cell results for this sample and the uncorrected measurement defined by equation (1) are also plotted. In the overlap frequency range the two techniques agree within 1 dB and the measured SE at frequencies above this range follows the same gradient as the computed result.

Fig. 16 shows the results of corrected measurements on two conducting fabric samples and a copper coated plastic film. In the case of the samples with the lowest SE the limitation of the reference sample calibration technique can be observed at the highest frequencies where the SE is lowest. In these cases the standing wave pattern is not established in the same way as for the high performance samples and the ripple re-appears.

## VI. MEASUREMENT OF ANISOTROPIC SAMPLES

In principle the absorber system has some capability to measure samples with limited anisotropy allowing where necessary both amplitude and phase measurements. Two orthogonal SE values can be obtained by rotating the sample. Fig. 17 shows the measured and computed results for an anisotropic sample comprising a 300  $\mu\text{m}$  thick brass sheet with an array of parallel rows of slots each 20 mm long by 1 mm wide. The computed results were obtained from a model similar to that used for the reference sample. The end-to-end slot gap and the gap between adjacent rows of slots is 10 mm. The agreement between computed and measured SE is good for the x-polarisation where the incident electric field is normal to the slot length and the SE is relatively low. There is no agreement between computed and measured results in the orthogonal polarisation where the computed result shows high SE. This is due to the polarisation characteristics of the fields in the absorber system, which has a cross-polar field 20 dB lower than the dominant polarisation and the limited cross-polar performance (again about 20 dB) of the two ridged waveguide horn antennas used in the system. Both these factors limit the level of anisotropy that can be measured to about 15 dB. As the anisotropy of a sample approaches 15 dB the quality of the measurement will be compromised. Note that neither the ASTM cell nor the NRC is capable of making pairs of orthogonal measurements on anisotropic samples.

Fig. 18 shows the measurement of the SE of an isotropic single-layer CFC sample. The sample was an aerospace 5-harness satin weave fabric moulded into a 0.50 mm thick laminate. Two orthogonal polarisations are shown for the absorber system measurement and the single measurement obtained from an ASTM4935 and NRC measurements. The sample preparation for the NRC measurement included ablation of the outer resin coating and application of silver paint around the perimeter of the sample in order to ensure good contact with the sample holder. The ASTM4935 sample relies on capacitive coupling to the conducting fibers and is subject to reliability issues.

Note that the two absorber box results are almost identical and that the NRC result exhibits a variability associated with the statistical processes of the measurement. Within these limitations the absorber system and NRC measurements are in good agreement. The ASTM measurement is 6 dB lower than the NRC and absorber system.

Fig. 19 shows the measurements of an anisotropic two-layer CFC sample. This sample consisted of two layers of the same 5-harness satin weave fabric moulded into a 0.65 mm thick balanced laminate. Here the difference between the two orthogonal polarisations can be observed. The NRC result tracks that of the polarisation with the lowest SE, i.e. the highest energy penetration through the sample. In these results the anisotropy displayed is up to 10 dB so in view of the discussion above the measurement is expected to be reliable. The ATSM measurement is also consistent in level with the absorber system and NRC measurement, however, there is an indication of a different trend in the ASTM result.

## VII. CONCLUSIONS

This paper has described a method that may be used to assess the shielding effectiveness of planar materials with non-conducting surface finishes (such as CFC). These materials are difficult to measure as the imperfect edge terminations which are found when using such materials tend to dominate the measurements. The method assumes that fields will propagate around the edges of the material sample and aims to attenuate these fields sufficiently to minimise their impact on the measured results. The proposed method utilises a relatively simple and low cost test setup, which is easy to use but has still been able to measure a shielding effectiveness of up to 100 dB over the frequency range 1 GHz to 8.5 GHz. The frequency range demonstrated was limited by the test equipment available. In principle the dynamic range and frequency range of the measurement could be increased by enlarging the test jig and the samples which may be tested in it.

It has been shown that the results obtained are similar to those obtained using ASTM D4935 and nested reverberation chamber test methods at frequencies where the methods overlap. The proposed method also has a limited capability of showing the shielding effectiveness for different incident wave polarisations where the material under test is anisotropic, with an anisotropy less than 15 dB. Where the anisotropy is greater, the co-polar field in the absorber system and non-ideal performance of the antennas used limits the accuracy of the measurement of the polarisation with the higher SE.

The measurement method is easy and rapid to carry out compared to alternative techniques and the test jig is simple to construct and more robust than the alternatives, making the technique very cost effective.

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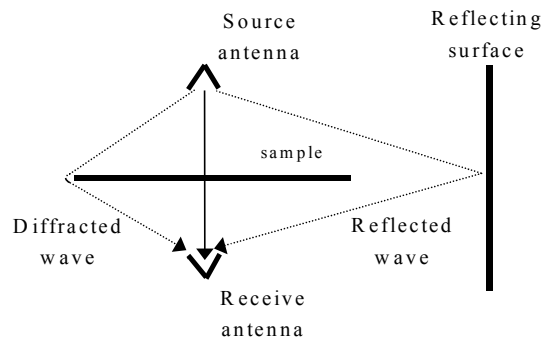


Fig.1. Time gating shielding effectiveness measurement.

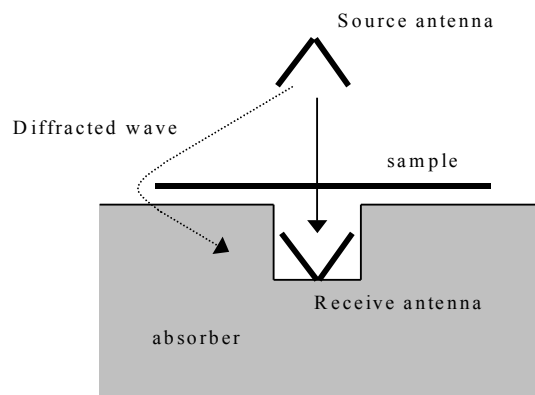


Fig. 2. Proposed method showing absorption of diffracted energy.

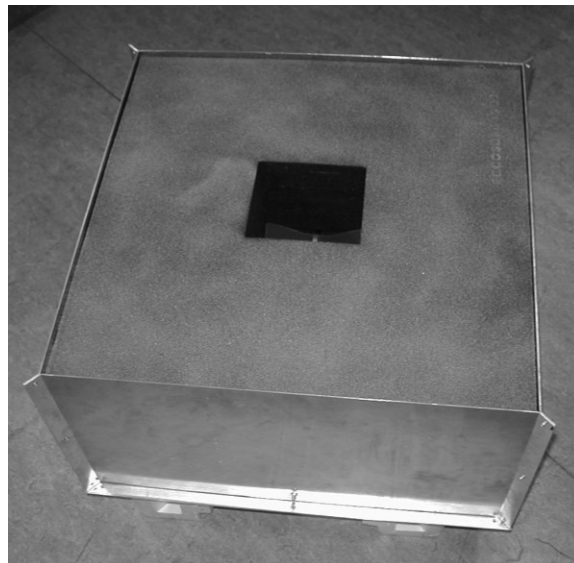


Fig.3. Prototype absorber system showing cavity with part of horn antenna just visible at the bottom.

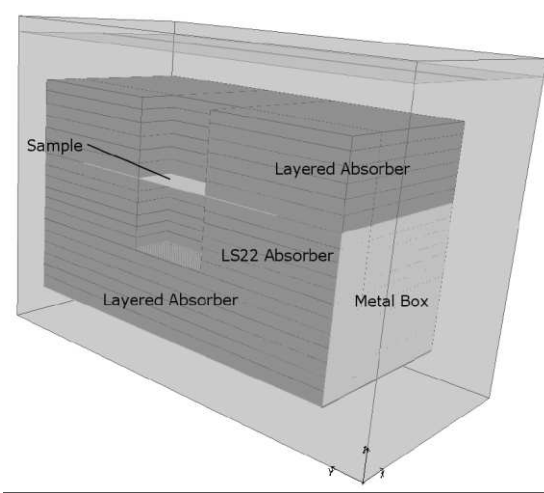


Fig. 4. Representation of the TLM absorber system model. Only half the structure is included in the model with a symmetry plane enforced in the front plane.

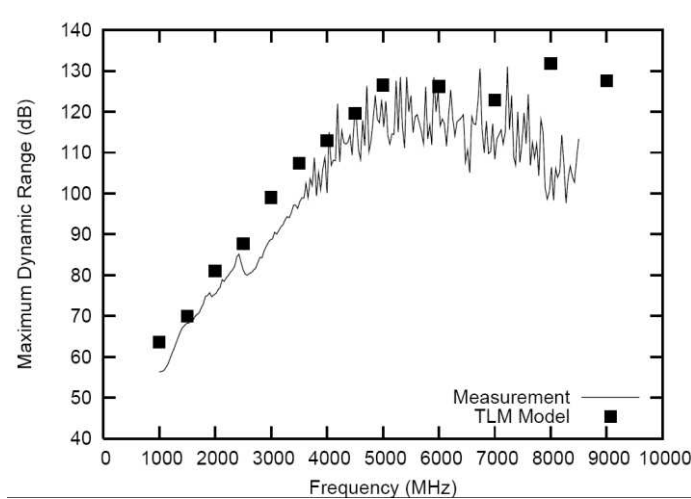


Fig. 5. Modelled and measured dynamic range of the absorber system with a 300 mm square sample.

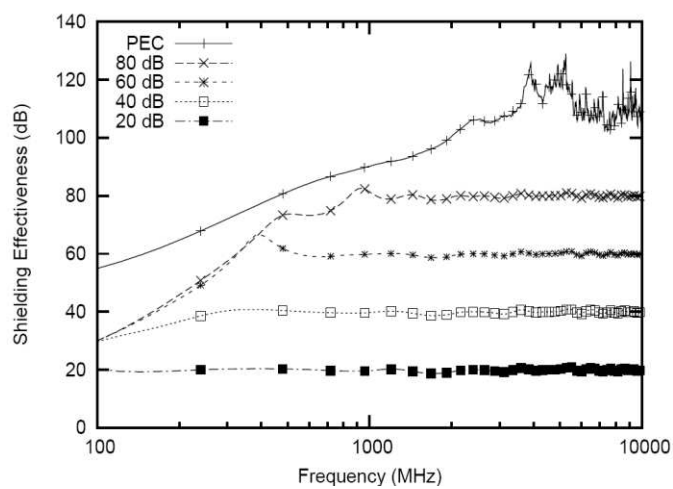


Fig. 6. Modelled performance in absorber box of fictitious materials with frequency independent SE from 20 dB to 80 dB. The PEC line indicates the maximum dynamic range of the system for the same size of sample..

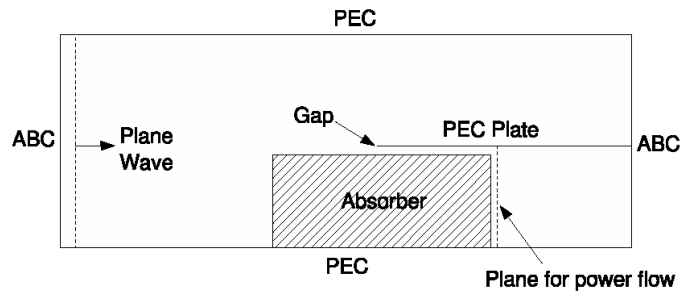


Fig. 7. Modelling the effect of a gap between the sample and absorber using a quasi-two-dimensional TLM model.

TABLE 1: ISOLATION PREDICTIONS (IN dB) BETWEEN THE EXTERNAL FIELD AND THE INTERNAL CAVITY FROM THE SLICE MODEL AT 3 GHz FOR VARIOUS ABSORBER TYPES AND GAP SIZES. OPTIMUM ABSORBER TYPE FOR EACH GAP SIZE IS SHOWN IN BOLD.

| LS Type | Gap Size     |             |             |             |             |             |
|---------|--------------|-------------|-------------|-------------|-------------|-------------|
|         | 0 mm         | 1 mm        | 2 mm        | 3 mm        | 4 mm        | 5 mm        |
| LS14    | 7.5          | 7.2         | 7           | 6.8         | 6.6         | 6.5         |
| LS16    | 18.3         | 17.7        | 17.3        | 16.9        | 16.5        | 16.2        |
| LS18    | 33.4         | 32.4        | 31.5        | 30.7        | 29.9        | 29.2        |
| LS20    | 35.2         | 35.2        | 35.6        | 35.2        | 34.0        | 32.6        |
| LS22    | 100.6        | 76.8        | <b>61.6</b> | <b>51.5</b> | <b>44.5</b> | <b>39.3</b> |
| LS24    | 116.5        | <b>79.3</b> | 59.9        | 48.9        | 41.7        | 36.5        |
| LS26    | 147.1        | 76.1        | 54.0        | 42.9        | 36.0        | 31.2        |
| LS28    | <b>158.9</b> | 59.3        | 40.8        | 32.1        | 26.9        | 23.4        |
| LS30    | 149.4        | 54.4        | 37.2        | 29.2        | 24.5        | 21.3        |

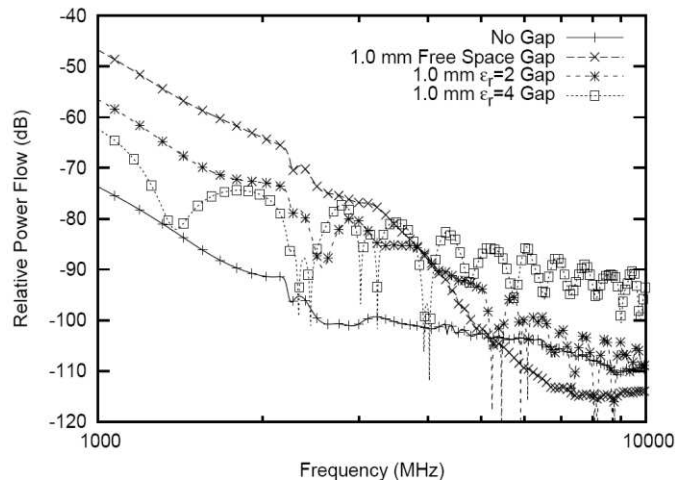


Fig. 8. Effect of permittivity in the gap between a conducting sample and the absorber.

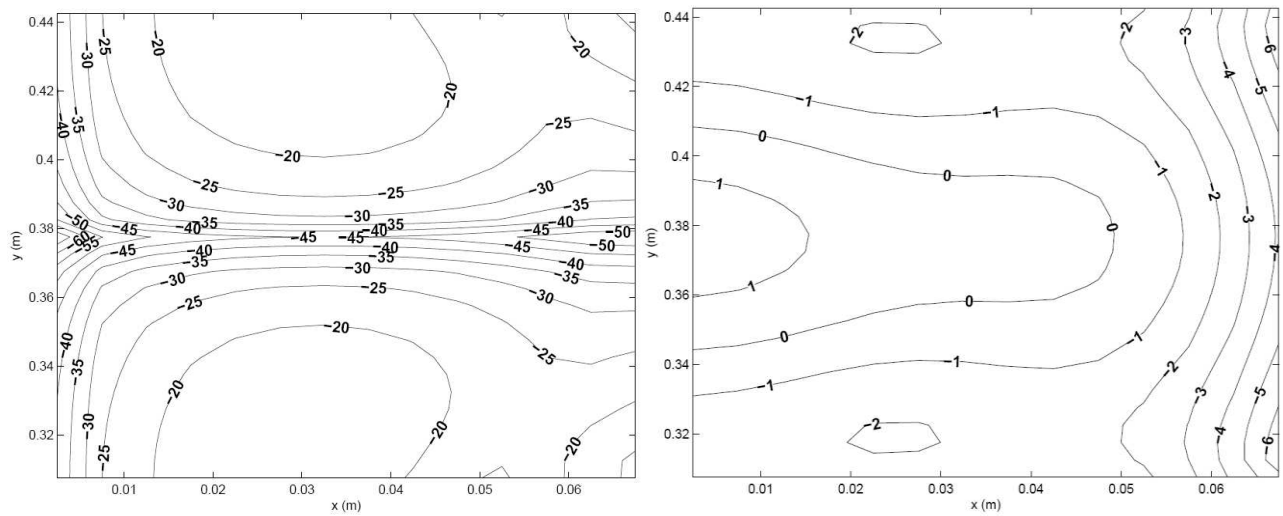


Fig.9 Electric field contours in the plane of the sample in dBV/m at 3 GHz with a y polarised horn incident field. Left is the x polarised field and right is the y polarised field. Note only half the cross-section of the cavity is shown, there is a symmetry plane along the axis.

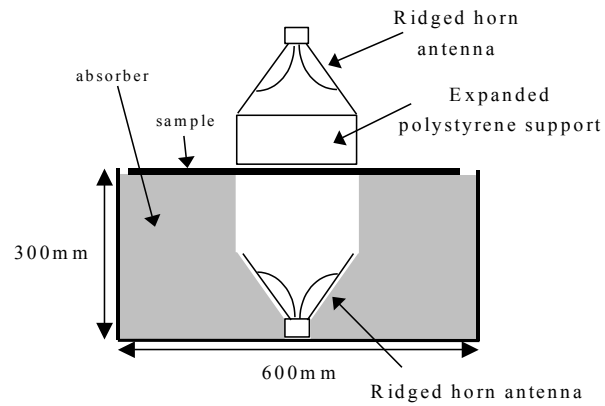


Fig. 10. Cross sectional view of the final absorber system.



Fig. 11. Final system showing sample and antenna support on top of absorber.



Fig. 12. Final system with the source antenna in place. The receive antenna port is visible at the bottom of the left hand wall of the box.

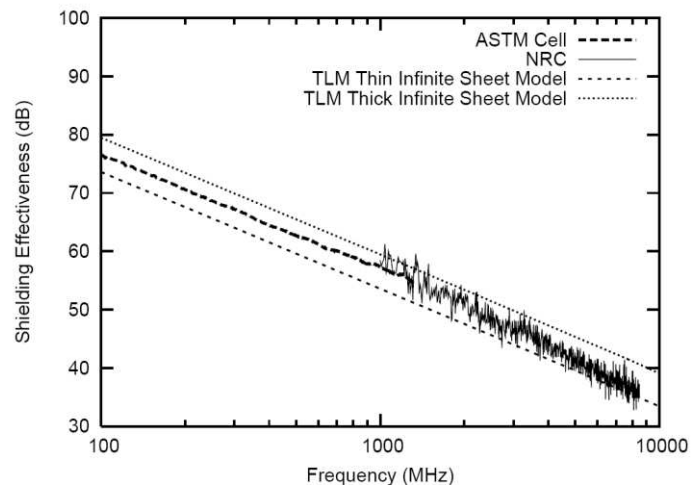


Fig. 13. Computed SE of thin and thick perforated brass sample compared with measured results using ASTM cell and NRC.

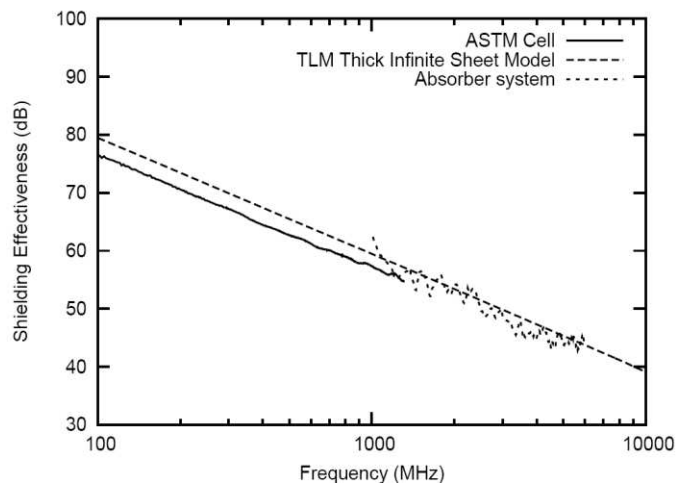


Fig. 14. Comparison of computed SE of perforated brass plate with measured SE using ASTM cell and uncorrected measurement in absorber system.

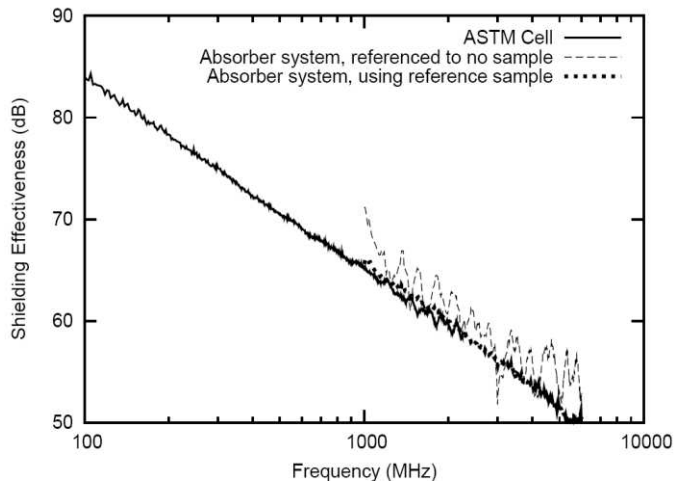


Fig. 15. Corrected and uncorrected measurements of second perforated brass plate (in absorber system) and ASTM cell measurement.

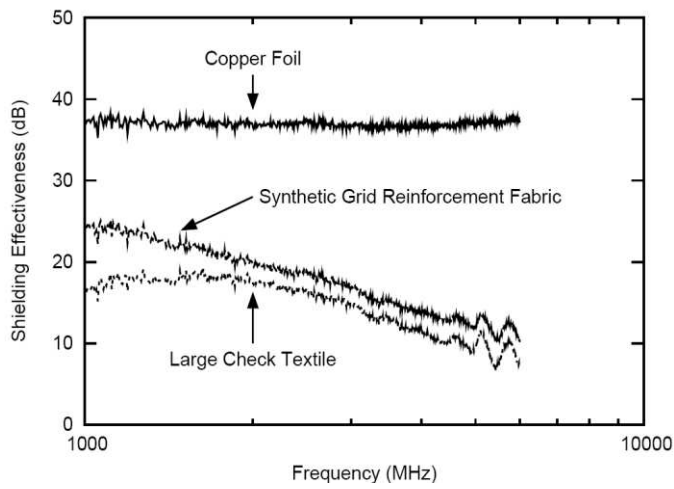


Fig. 16. Measured results on two conducting fabrics and copper coated plastic film.

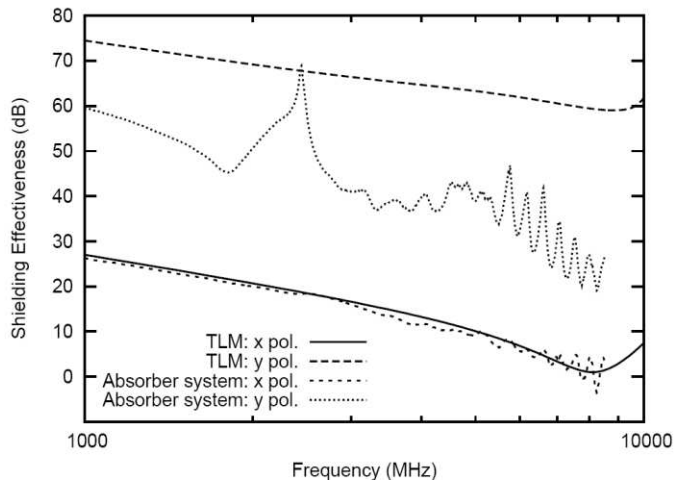


Fig. 17. Computed and measured results for an anisotropic perforated plate with the field polarisation across (x pol.) and along (y pol.) the array of slots.

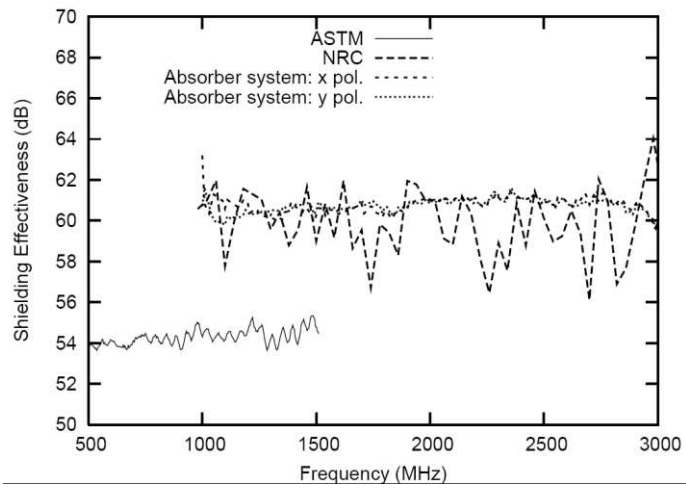


Fig. 18. Measured results on a 1-layer isotropic CFC sample.

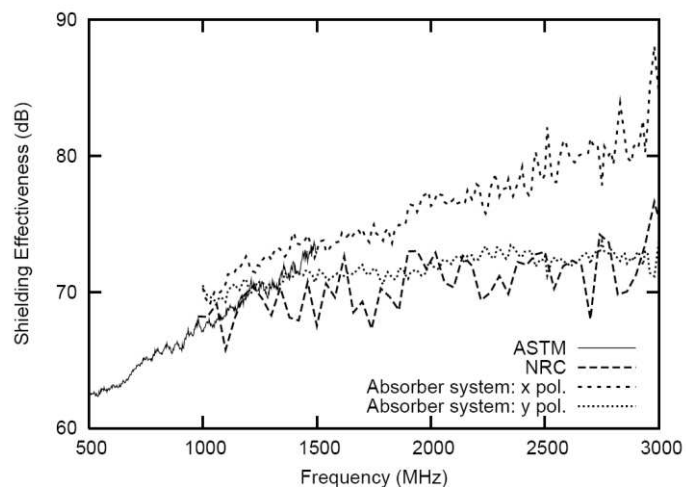


Fig. 19. Measured results on a 2-layer anisotropic CFC sample.