

Open access • Journal Article • DOI:10.1190/1.1451472

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Published on: 01 Jan 2002 - Geophysics (Society of Exploration Geophysicists)

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An Adaptive Finite-Difference Method for Traveltimes and Amplitudes

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ABSTRACT

The point source traveltime field has an upwind singularity at the source point. Consequently, all formally high-order finite-difference eikonal solvers exhibit first-order convergence and relatively large errors. Adaptive upwind finite-difference methods based on high-order Weighted Essentially NonOscillatory (WENO) Runge-Kutta difference schemes for the paraxial eikonal equation overcome this difficulty. The method controls error by automatic grid refinement and coarsening based on an a posteriori error estimation. It achieves prescribed accuracy at far lower cost than does the fixed-grid method. Moreover, the achieved high accuracy of traveltimes yields reliable estimates of auxiliary quantities such as takeoff angles and geometrical spreading factors.

INTRODUCTION

Many finite-difference methods have been introduced to compute the traveltime for isotropic media directly on regular grids (Reshef and Kosloff, 1986; Vidale, 1988; van Trier and Symes, 1991; Podvin and Lecomte, 1991; Schneider et al., 1992; Qin et al., 1992; Schneider, 1995; El-Mageed et al., 1997; Popovici and Sethian, 1997; Kim and Cook, 1999). The traveltime field is mostly smooth, suggesting that highorder finite-difference methods should be effective. The use of upwind differencing in all of the cited methods confines the errors due to singularities which develop away from the source point. The source point itself is, however, also an upwind singularity. The truncation error of a pth order method is dominated by the product of (p+1)st derivatives of the traveltime field and the (p+1)st power of the step(s). The (p+1)st derivatives of the traveltime field, however, behave like the (-p+1)th power of the distance to the source, since in the constant velocity case the traveltime is equal to the distance divided by the velocity. Therefore, near the source – when the distance is on the order of the step – the truncation error is quadratic in the step, i.e., first order. This inaccuracy spreads throughout the computation and renders all higher-order methods first-order convergent. Moreover, the resultant inaccuracy in traveltime prevents reliable computation of auxiliary quantities such as takeoff angles and amplitudes.

This inaccuracy afflicts all point source traveltime computations using gridded eikonal solvers. In the few published convergence tests, implementers have resorted to imposing a grid-independent region of constant velocity near the source, in which the traveltimes are initialized analytically. This is the approach taken for instance by Sethian (1999) in demonstrating second-order convergence for a version of his fast marching method. This approach has two obvious drawbacks: (1) the velocity may not be homogeneous near the source; and (2) the size of the region of analytic computation must be set by the user and bears no obvious relation to the grid parameters.

In principle, highly accurate ray-tracing methods could be used to alleviate the first difficulty, but the second remains: it introduces an arbitrary parameter into the use of eikonal solvers. Kim and Cook (1999) take a different approach, similar to the one we advocate: they refine the grid several times near the source so that the reduced grid spacing compensates for the increased truncation error. However, their grid refinement strategy appears to be ad-hoc, and it once again involves an arbitrary parameter, namely, the number of grid refinements near the source, without a clearcut selection criterion.

In this paper, we show how to use adaptive-gridding concepts commonplace in the numerical solution of ordinary differential equations (Gear, 1971) to resolve the difficulty caused by this inaccuracy. Adaptive gridding has already been used in numerical solutions of PDEs (Berger and Oliger, 1984; Berger and LeVeque, 1997). Generally, the grid refinement must be localized in several dimensions, leading to complex data structures. Fortunately, the nature of the traveltime field permits a relatively straightforward adaptive-gridding strategy (Belfi and Symes, 1998). The present work improves that of Belfi and Symes through the use of the more accurate Weighted Essentially NonOscillating (WENO) difference schemes and extends it to solutions of advection equations for various geometrical acoustics quantities. The efficiencies achieved by the adaptive gridding are considerable – usually more than an order of magnitude reduction in computation time for problems of typical exploration size, compared to fixed-grid methods giving the same level of accuracy. We also obtain dramatic improvements in the accuracy of computed geometrical acoustics quantities, such as takeoff angles and geometrical amplitudes.

The essential principle of the adaptive gridding is simple. It is based on a hierarchy of difference schemes of various orders. Presumably a higher-order step is more accurate than a lower-order step, so that the higher-order step can serve as a substitute for the exact solution in evaluating the local error in the lower-order step. Therefore, one can combine the step computations of two different orders to obtain a so-called a

posteriori estimate of the truncation error for the lower-order step. Since the lower-and higher- order truncation errors stand in a known asymptotic relation, this permits an estimate of the higher-order truncation error as well. The asymptotic form of the truncation error then permits prediction of a step that will result in a lower-order truncation error less than a user-specified tolerance. So long as the steps are selected to maintain this local error, standard theory predicts that the higher-order global error, i.e., the actual error in the solution computed using the higher-order scheme, will be proportional to the user-specified tolerance. This straightforward idea is embedded in most modern software packages for solutions of ordinary differential equations (Gear, 1971). Its use for partial differential equations is a little more complicated because it is usually necessary to adjust the grid of the non-evolution variables along with the evolution step. As first established by Belfi and Symes (1998), the solution of the (paraxial) eikonal equation changes in a sufficiently predictable way to make grid adjustment practical.

Then we formulate the advection equation for takeoff angles and present the amplitude formulae for a two-dimensional line source and point source, respectively. We briefly describe numerical schemes needed in the adaptive-gridding approach, presenting the details in Appendix A. With these ingredients in place, we introduce the adaptive-gridding principle for the eikonal equation with a point source and present a simple implementation. Numerical experiments demonstrate that the new approach gives us not only accurate traveltime fields, but accurate amplitude fields as well. We conclude with some discussion on adaptive gridding in the three-dimensional case.

PARAXIAL EIKONAL EQUATIONS

The traveltime field in an isotropic solid satisfies an eikonal equation. Denote by (x_s, z_s) the coordinates of a source point, and by (x, z) the coordinates of a general

point in the subsurface. The first-arrival traveltime field $\tau(x, z; x_s, z_s)$ is the viscosity solution of the eikonal equation (Lions, 1982),

$$\left(\frac{\partial \tau}{\partial x}\right)^2 + \left(\frac{\partial \tau}{\partial z}\right)^2 = s^2(x, z) \tag{1}$$

with the initial condition

$$\lim_{(x,z)\to(x_s,z_s)} \left(\frac{\tau(x,z;x_s,z_s)}{\sqrt{(x-x_s)^2 + (z-z_s)^2}} - \frac{1}{v(x,z)} \right) = 0$$

where v is the velocity, and $s = \frac{1}{v}$ is the slowness.

In some seismic applications, the traveltime field is needed only in regions where

$$\frac{\partial \tau}{\partial z} \ge s \cos \theta_{\max} > 0,$$

i.e., along downgoing, first-arriving rays making an angle $\leq \theta_{\text{max}} < \frac{\pi}{2}$ with the vertical. To enforce this condition, we modify the eikonal equation as an evolution equation in depth (Gray and May, 1994),

$$\frac{\partial \tau}{\partial z} = H\left(\frac{\partial \tau}{\partial x}\right) = \sqrt{\operatorname{smmax}\left(s^2 - \left(\frac{\partial \tau}{\partial x}\right)^2, s^2 \cos^2 \theta_{\max}\right)},\tag{2}$$

where smmax is a smoothed max function defined by

$$\operatorname{smmax}(x, a) = \begin{cases} \frac{1}{2}a & \text{if } x < 0, \\ \frac{1}{2}a + 2\frac{x^4}{a^3}(1 - \frac{4}{5}\frac{x}{a}) & \text{if } 0 \le x < \frac{a}{2}, \\ x + 2\frac{(x - a)^4}{a^3}(1 + \frac{4}{5}\frac{x - a}{a}) & \text{if } \frac{a}{2} \le x < a, \\ x & \text{if } x \ge a, \end{cases}$$

where a > 0 (Qian et al., 1999).

Equation (2) defines a stable nonlinear evolution in z, suitable for explicit finite-difference discretization. The smoothed max function makes the numerical Hamiltonian smooth enough to carry out standard truncation error analysis for schemes of up to third-order accuracy. The solution τ is identical to the solution of the eikonal equation provided that the ray makes an angle $\leq \theta_{\text{max}} < \frac{\pi}{2}$ with the vertical; if the ray makes an angle $\geq \theta_{\text{max}}$ with the vertical, the corresponding wavefront is replaced by an artificial plane wave.

ADVECTION EQUATIONS FOR TAKEOFF ANGLES

Based on the traveltime computed by solving the eikonal equation, we can approximate the amplitude field by solving a transport equation. The amplitude satisfies the following transport equation (Cerveny et al., 1977),

$$\nabla \tau \cdot \nabla A + \frac{1}{2} A \nabla^2 \tau = 0. \tag{3}$$

The equation (3) is a first-order advection equation for the amplitude A. The Laplacian of the traveltime field is involved in this advection equation, which implies that we need a third-order accurate traveltime field to get a first-order accurate amplitude field (Symes, 1995; El-Mageed et al., 1997).

For convenience in the following presentation, we introduce the ray coordinates. The ray coordinates are defined by $(\tau, \phi) = (\tau(x, z; x_s, z_s), \phi(x, z; x_s, z_s))$, where τ and ϕ are the traveltime and take-off angle of a ray from source point (x_s, z_s) to a general point (x, z) in the subsurface, respectively. In two-dimensional isotropic media with line sources, the amplitude also satisfies the formula (Cerveny et al., 1977; Friedlander, 1958),

$$A = \frac{v}{2\pi\sqrt{2}}\sqrt{|\nabla\tau\times\nabla\phi|},\tag{4}$$

where $\nabla \phi$ and $\nabla \tau$ are the gradients of the takeoff angle and the traveltime, respectively.

Since the take-off angle ϕ is constant along any ray,

$$\nabla \tau \cdot \nabla \phi = \frac{\partial \tau}{\partial x} \frac{\partial \phi}{\partial x} + \frac{\partial \tau}{\partial z} \frac{\partial \phi}{\partial z} = 0.$$
 (5)

That is, the wavefront normal $\nabla \tau$ is tangential to the ray; the gradient $\nabla \phi$ is tangential to the wavefront. Equation (5) is slightly easier to solve numerically than equation (3) because no second-order traveltime derivatives are explicitly involved in equation (5). Having solved equation (5) for ϕ , one produces the amplitude A through (4).

Since the typical seismic source is a point source, we need to compensate for the out-of-plane radiation in the two-dimensional (2-D) line-source amplitude formula. The 2-D amplitude with a point source (2.5-D amplitude) can be computed by

$$A = \frac{v}{4\pi} \sqrt{\tau_{yy} |\nabla \tau \times \nabla \phi|},\tag{6}$$

where the out-of-plane curvature τ_{yy} satisfies another advection equation (Symes et al., 1994),

$$\frac{\partial \tau}{\partial x} \frac{\partial \tau_{yy}}{\partial x} + \frac{\partial \tau}{\partial z} \frac{\partial \tau_{yy}}{\partial z} + \tau_{yy}^2 = 0. \tag{7}$$

Supposing that the amplitude is required to be first-order accurate, the two gradients $\nabla \tau$ and $\nabla \phi$ involved in the amplitude formulae should have at least first-order accuracy. However, because after discretization of equation (5) $\nabla \phi$ depends on second-order derivatives of traveltime τ , it implies that to get a first-order accurate $\nabla \phi$, the traveltime τ itself should have at least third-order accuracy. The final conclusion is that a third-order traveltime solver is required to get first-order accurate amplitudes, as noted before.

Zhang (1993) used equation (6) in polar coordinates to compute the geometrical spreading factor, but his computation of the takeoff angle was based on the first-order traveltime field. Consequently, the gradient of take-off angle computed by his scheme was inaccurate. Vidale etc. (1990) encountered a similar difficulty.

FINITE-DIFFERENCE SCHEMES

The literature suggests a large number of competing finite-difference and related schemes for the solution of the eikonal equation. We have chosen to use the essentially nonoscillatory (ENO) schemes (Osher and Sethian, 1988; Osher and Shu, 1991) and the related weighed ENO (WENO) schemes (Liu et al., 1994; Shu, 1998; Jiang and Peng, 2000) for the following reasons: (1) stable schemes of arbitrarily high-order

accuracy exist, permitting accurate solutions on coarse grids (a factor which is critical to the mesh refinement or coarsening); (2) versions exist in any dimension so that we can straightforwardly extend our methodology to the three-dimensional case (El-Mageed et al., 1997; Kim and Cook, 1999).

All of these schemes take the form of recursive depth stepping rules,

$$\tau \leftarrow \tau + \delta_n^n \tau, \tag{8}$$

$$z \leftarrow z + \Delta z. \tag{9}$$

Here δ_n^n is a nonlinear update operator expressing the WENO-Runge-Kutta rule of order n, defining a difference scheme of formal nth-order accuracy and depending on Δz , Δx , and the slowness field s. Since we want to emphasize the strategy of the adaptive-gridding approach, we put the detailed form of δ_n^n (n=2,3) in Appendix A.

Similarly, we solve the advection equation for the takeoff angle ϕ and the out-ofplane curvature τ_{yy} by using WENO schemes.

ADAPTIVE GRIDDING IMPLEMENTATION

To initialize our algorithm, the user supplies a local error tolerance ϵ ; σ_1 and σ_2 are two user-defined positive functions of ϵ which are used to control the coarsening and refinement. For example, we can take $\sigma_1 = 0.1\epsilon$ and $\sigma_2 = \epsilon$. We use the 2nd and 3rd order eikonal solvers (equations (A-1) and (A-2)) and estimate the truncation error of the 2nd-order scheme as $e_2 = \max |\delta_2^2 \tau - \delta_3^3 \tau|$ over the current depth. So long as $\sigma_1(\epsilon) \leq e_2 \leq \sigma_2(\epsilon)$ at every point of the current depth level, we simply proceed to the next step. It is well known (Gear, 1971) that for ordinary differential equations an efficient adaptive stepping requires rather loose control of the local error, hence the factor of 10 difference between σ_1 and σ_2 is reasonable and works pretty well in practice. When $e_2 < \sigma_1(\epsilon)$, we increase the step by a factor of two, i.e., $\Delta z \leftarrow 2\Delta z$,

and we recompute the τ update and e_2 . Similarly, when $e_2 > \sigma_2(\epsilon)$, we decrease the step by a factor of two. As soon as the local error is once again within the tolerance interval, we continue depth-stepping. A very important point is that we retain the 3rd-order (a more accurate one) computation of τ at the end of each depth step as the actual update, discarding the 2nd-order computation, which is used only in step control.

The usual step adjustment in ODE solvers would change Δz by a factor computed from the asymptotic form of the truncation error (Stoer and Bulirsch, 1992, 449). This is impractical for a PDE application because it would require an arbitrary adjustment of the spatial grid (i.e., the x-grid in the difference scheme) and, therefore, expensive interpolation. Scaling Δz by a factor of two, however, implies that the stability may be maintained by scaling Δx by the same factor. For coarsening, this means simply throwing out every other grid point, i.e., no interpolation at all, which dramatically reduces the floating point operations required. Since the typical behaviour of the traveltime field is to become smoother as one moves away from the source, the truncation errors tend in general to decrease. Therefore, most of the grid adjustments are coarsenings and very little or no interpolation is required. Since the slowness field comes to us in gridded form, an interpolation is always required to supply estimates of slowness at the points appearing in the WENO-Runge-Kutta formula. We use a local quadratic interpolation in x and z because the third-order accuracy of which is compatible with that of the difference scheme. For traveltimes, we will use a similar quadratic interpolation.

Since the traveltime field is nonsmooth at the source point, the truncation error analysis on which the adaptive step selection criterion is based is not valid there. Therefore, it is necessary to produce a smooth initial traveltime field. We do this by estimating the largest $z_{\rm init} > 0$ at which the constant velocity traveltime is in error by less than $\sigma_2(\epsilon)$. Details of the $z_{\rm init}$ calculation are given in Appendix B. Having initialized τ at $z_{\rm init}$, the algorithm invokes adaptive gridding. Since $z_{\rm init}$ is

quite small, τ changes rapidly, resulting in a large number of grid refinements at the outset. However, no interpolation is performed, as τ is given analytically on $z=z_{\rm init}$. This initially very fine grid is rapidly coarsened as the depth stepping proceeds.

In our current implementation, we maintain a data structure for the computational grid that is independent of the output grid; the desired quantities are calculated on the computational grid and interpolated back to the output grid. As a safeguard against pathological program behaviors, we specify a maximum number of permitted grid refinements, MAXREF.

A simplified algorithm framework is as follows:

- Input ϵ , x_s , z_s , θ_{max} , Δz , MAXREF.
- Initialize Δx , τ , $z=z_{\mathrm{init}}$, REF = 0.
- Do while z < target depth,
 - compute $e_2 = \max |\delta_2^2 \tau \delta_3^3 \tau|$ over the current depth level z;
 - if $e_2 \leq \sigma_1(\epsilon)$ and REF > 0,
 - * $\Delta z \leftarrow 2\Delta z$.
 - * $\Delta x \leftarrow 2\Delta x$,
 - * REF \leftarrow REF -1,
 - * upsample τ (throw out every other point).
 - else if $e_2 \geq \sigma_2(\epsilon)$ and REF \leq MAXREF,
 - * $\Delta z \leftarrow \Delta z/2$,
 - * $\Delta x \leftarrow \Delta x/2$,
 - * REF \leftarrow REF + 1,
 - * downsample τ (interpolate)
 - else

*
$$z \leftarrow z + \Delta z$$
,

*
$$\tau \leftarrow \tau + \delta_3^3 \tau$$
.

- end if

• end do

This description leaves out the output step of the algorithm: a full implementation monitors the depth level of the next set of output points and quadratically interpolates the traveltime field onto them as soon as z passes this depth, using the current and last two depth levels of τ . Local quadratic interpolation preserves the third-order accuracy of the computed τ .

To avoid unnecessary computations, we update τ only within the triangle

$$\{(x,z): |x-x_s| \le |z-z_s| \tan \theta_{\max} \}.$$

All rays with takeoff angles less than θ_{max} must lie inside this triangle, and it is only along such rays that the paraxial eikonal equation produces correct first-arrival times. Output points outside the triangle are assigned a very large number so that constructed ray paths will never reach those places. Because traveltimes at output points which are inside the triangle but not lying on rays with takeoff angles less than θ_{max} also receive erroneous time values, they must be washed out of any subsequent computations. For high frequency asymptotics computations, this masking is most easily accomplished by zeroing the geometrical amplitude at such points.

NUMERICAL EXPERIMENTS

To illustrate how the adaptive-gridding approach works, we test our method on a constant velocity model, $v=1 \mathrm{km/s}$, with two-dimensional geometry $\{(x,z):-0.5 \mathrm{km} \le x \le 0.5 \mathrm{km}, \ 0 \le z \le 1.0 \mathrm{\ km} \ \}$, in which case the behaviors of traveltime fields and amplitude fields are well understood.

In the constant velocity case, all the desired quantities have obvious analytical solutions to compare against the computed solutions. We compare the results obtained by fixed- and adaptive- grid algorithms. Both algorithms use a third-order WENO scheme (Appendix A) to compute τ ; the adaptive-grid scheme uses a second-order ENO scheme for local truncation error estimation. The ouput grid is 51×51 with $\Delta x = \Delta z = 0.02$ km. For adaptive-grid algorithms, MAXREF is set to be 5 with the coarsest grid 17×17 , $\sigma_1 = 0.1\epsilon$ and $\sigma_2 = \epsilon$.

Frist, we will compare the computation cost of the two methods. Tables 1 and 2 show the traveltime error and computation cost by the fixed-grid and adaptive-grid method, respectively, where Flops denote the number of floating point operations. The error is the maximum absolute error at the bottom row of the gridpoints (z = 1 km). The computed portion of this depth level ($-0.5 \text{km} \le x \le 0.5 \text{km}$) lies entirely within the computation aperture ($\theta_{\text{max}} = 78 \text{ degrees}$) and so consists of accurately computed τ values. From the two tables, we can see that to reach the same level of accuracy, the adaptive-gridding approach requires an order of magnitude lower computational cost than does the fixed-gridding approach.

Second, we will illustrate the difference of accuracy of the two algorithms. For the fixed-gridding algorithm, the computational grid is 200×200 with dx = 0.005 km. For the adaptive-gridding algorithm, the local error-tolerance ϵ is 0.00001. The traveltime contours (not shown here) produced by the two approaches have no obvious difference because the fixed-gridding algorithm still has first-order accuracy. Figure 1 shows contours of τ_x computed by two approaches. We can see that τ_x by the fixed grid is oscillating, but τ_x by the adaptive grid traveltime solver is convergent. Because the fixed-gridding approach gives us only first-order accurate traveltime field, the resultant traveltime derivatives have only zero-order accuracy and exhibit oscillations which do not decrease in magnitude as the grid is refined, as shown in Figure 1a. However, the adaptive-gridding approach yields far more accurate traveltime fields, thus the traveltime derivatives are still accurate, as shown in Figure 1b. Similar

phenomena are observed for τ_z .

Now we discuss the takeoff angle and its derivatives. Because the coefficients in the advection equation for takeoff angles depend on the traveltime gradient, the accuracy of ϕ will be decided by the traveltime solver we are using. Since the first-order traveltime field from the fixed-gridding approach results in inaccurate $\nabla \tau$, the resultant takeoff angle is not accurate to be differentiated. However, the takeoff angle based on the traveltime field from the adaptive-gridding approach is accurate enough to be differentiated. Figure 2 shows ϕ_x by the two approaches. Because the takeoff angle based on the traveltime field from the fixed-gridding approach is inaccurate, the resultant derivatives ϕ_x are divergent, as shown in Figure 2a. However, the adaptive-gridding approach produces accurate traveltime gradients, which leads to the convergent ϕ_x , as shown in Figure 2b. Similar observations hold for ϕ_z .

To further illustrate the differences of the accuracy between two approaches, Figure 3 shows the distribution of relative errors along the depth direction for ϕ_x . The error along the depth direction is defined as

$$e(z) = \frac{\max_{-0.5 \le x \le 0.5} |f^{\text{comp}}(x, z) - f^{\text{ana}}(x, z)|}{\max_{-0.5 < x < 0.5} |f^{\text{ana}}(x, z)|}$$
(10)

where f^{comp} is the computed solution, and f^{ana} the analytic solution. For instance, substituting f with ϕ_x in Equation (10), we get the error distribution for ϕ_x along the depth direction. From Figure 3, we can conclude that the adaptive-gridding approach produces much more accurate $\nabla \phi$ than does the fixed-gridding approach. The resultant amplitudes with a line source based on $\nabla \tau$ and $\nabla \phi$ by the two approaches are shown in Figure 4; one is divergent by the fixed-gridding approach, the other accurate by the adaptive-gridding approach. Note the episodic nature of the convergence for the adaptive-gridding algorithms. Because we have allowed the local error estimate to vary by an order of magnitude before adjusting the grid and then permitted only step changes by factors of 2, the error exhibits "sticky," discontinuous behaviour.

Finally, Figure 5 shows the computational results for the out-of-plane curvature

 τ_{yy} and the amplitude field with the point source by adaptive-gridding approach. The computed τ_{yy} is accurate and the resultant amplitude is convergent.

We have embedded the adaptive-grid traveltime and amplitude solver in twodimensional Kirchhoff prestack migration and inversion code (Symes et al., 1994). Figure 6 shows the impulse response of the inversion for a WENO third-order eikonal solver, where the Beylkin determinant required by the inversion is computed by using the information from traveltimes and takeoff angles. We will report the complete test result of the new adaptive traveltime and amplitude solver embeddedin migration and inversion in the near future.

CONCLUSIONS

In this paper first we stated a paraxial eikonal equation with depth as evolution direction and an advection equation for takeoff angles, then we presented high-order WENO difference schemes to solve the eikonal equation for the traveltime and the advection equation for the takeoff angle. To deal with the singularity of a point source, we proposed a new adaptive-grid eikonal solver and detailed the implementation. Numerical experiments showed that the new method yields an efficiency gain of more than an order of magnitude in computational time. Adaptive gridding does not altogether eliminate the arbitrary parameter feature, for which we criticized other approaches in the introductory section of this paper; however, our arbitrary parameter is the local-error tolerance ϵ . In principle, the local-error tolerance ϵ is proportional to the (global) error in the computed solution, but the relation is complex (as the numerical example shows) and not a by-product of the algorithm. Nonetheless we maintain that the simplicity and homogeneity of the algorithm, and the direct if not apparent relation between ϵ and the global solution error, make the adaptivegrid scheme easier to use than its alternatives. Also, the considerable success of the variable-step selection methods for ODEs, which have the same indirect error control

feature, supports this contention.

The extension to 3-D isotropic media is straightforward. Because all the difference schemes presented here can easily be extended to the three-dimensional case, there is no difficulty in implementing a 3-D version of the adaptive traveltime and amplitude solver. Moreover, we are expecting that the efficiency gain in computational cost will be even more dramatic in 3-D. Furthermore, we already extended the adaptive-gridding algorithm to computing traveltimes and amplitudes in anisotropic media (Qian, 2000). A fully adaptive eikonal solver based upon a posteriori error estimates for general numerical methods for Hamilton-Jacobi equations (Albert et al., 2000) will be the subject of a subsequent paper.

ACKNOWLEDGEMENTS

The authors performed part of the research reported here while guests of the Stanford Exploration Project; JLQ and WWS thank SEP members and director Prof. Jon Claerbout for their hospitality and many stimulating discussions. JLQ thanks Dr. S. Kim and Dr. J. Bee Bednar for their valuable comments. The authors also thank the associate editor, G. T. Schuster, and the reviewers for the constructive comments and suggestions which were helpful in finalizing the paper.

This work was partially supported by the National Science Foundation under grant number DMS 9627355, the Office of Naval Research under grant number N00014-96-1-0156, the U. S. Department of Energy under grant number DE-FG07-97 ER14827 and The Rice Inversion Project. TRIP Sponsors for 1999 are Amerada Hess, Amoco Research, Conoco Inc., Exxon Production Research Co., Geophysical Development Corporation, Landmark Graphics, Mobil, and Western Geophysical.

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APPENDIX A-WENO SCHEMES FOR EIKONAL EQUATIONS

Our adaptive scheme is based on the second- and third- order WENO difference schemes introduced by Jiang and Peng (Jiang and Peng, 2000). These are in turn extensions of second- and third-order ENO difference schemes, which we present first.

For a function f of the space variable (x, z) in the computational domain, we write

$$f_i^k = f(x_i, z_k),$$

$$(x_i, z_k) = (x_{\min} + (i-1)\Delta x, z_{\min} + (k-1)\Delta z).$$

Let

$$\tau_i^k = \tau(x_i, z_k; x_s, z_s)$$

and define the forward D^+ and backward D^- finite-difference operators

$$D_x^{\pm} \tau_i^k = \frac{\pm [\tau_{i\pm 1}^k - \tau_i^k]}{\Delta x}.$$

The second- and third-order ENO refinements of $D_x^{\pm} \tau$ are

$$\begin{split} D_x^{\pm,2}\tau_i &= D_x^{\pm}\tau_i \mp \frac{1}{2}\Delta x \ m(D_x^{\pm}D_x^{\pm}\tau_i, D_x^{-}D_x^{+}\tau_i), \\ D_x^{\pm,3}\tau_i &= D_x^{\pm,2}\tau_i - \frac{1}{6}(\Delta x)^2 \ m(D_x^{\pm}D_x^{\pm}D_x^{\pm}\tau_i, D_x^{+}D_x^{+}D_x^{-}\tau_i, D_x^{+}D_x^{-}T_i), \end{split}$$

where

$$m(x,y) = \min\left(\max\left(x,0\right), \max\left(y,0\right)\right) + \max\left(\min\left(x,0\right), \min\left(y,0\right)\right).$$

Similar refinements exist of any order.

The upwind ENO approximations for $\frac{\partial \tau}{\partial x}$ are

$$\widehat{D}_x^n\tau = \mathrm{modmax}(\mathrm{max}(D_x^{-,n}\tau,0),\mathrm{min}(D_x^{+,n}\tau,0)),$$

where the modmax function returns the larger value in modulus.

The second-order and third-order ENO Runge-Kutta steps are

$$\begin{split} \delta_2^1 \tau &= \Delta z H(\widehat{D}_x^2 \tau), \\ \delta_2^2 \tau &= \frac{1}{2} \left(\delta_2^1 \tau + \Delta z H(\widehat{D}_x^2 (\tau + \delta_2^1 \tau)) \right), \end{split} \tag{A-1}$$

and

$$\begin{split} \delta_3^1 \tau &= \Delta z H(\widehat{D}_x^3 \tau), \\ \delta_3^2 \tau &= \frac{1}{4} \left(\delta_3^1 \tau + \Delta z H(\widehat{D}_x^3 (\tau + \delta_3^1 \tau)) \right), \\ \delta_3^3 \tau &= \frac{1}{3} \left(2 \delta_3^2 \tau + 2 \Delta z H(\widehat{D}_x^3 (\tau + \delta_3^2 \tau)) \right). \end{split} \tag{A-2}$$

The depth step Δz must satisfy the stability condition

$$\Delta z \leq \Delta z_{
m cfl} = rac{\Delta x}{ an(heta_{
m max})}.$$

We have typically chosen $\Delta z = 0.9 \Delta z_{\rm cfl}$.

The nth-order scheme is then

$$\tau^{k+1} = \tau^k + \delta_n^n \tau^k \tag{A-3}$$

for $k = 0, 1, 2, \cdots$.

We have observed that the gradient of the take-off angle based on the third-order ENO traveltime is too noisy to lead to an accurate amplitude field. To alleviate this phenomenon, instead of ENO third-order refinements, we use WENO third-order refinement (Jiang and Peng, 2000) to compute $D_x^{\pm}\tau$ in the third-order Runge-Kutta step, which yields an accurate amplitude field.

The WENO third-order schemes for $D_x^{\pm} \tau_i$ are

$$D_x^{\pm W,3} \tau_i = \frac{1}{12} \left(-D_x^+ \tau_{i-2} + 7D_x^+ \tau_{i-1} + 7D_x^+ \tau_i - D_x^+ \tau_{i+1} \right)$$

$$\pm \Delta x \Phi^W \left(D_x^- D_x^+ \tau_{i\pm 2}, D_x^- D_x^+ \tau_{i\pm 1}, D_x^- D_x^+ \tau_i, D_x^- D_x^+ \tau_{i\mp 1} \right),$$

where

$$\Phi^{W}(a, b, c, d) = \frac{1}{3}w_0(a - 2b + c) + \frac{1}{6}(w_2 - \frac{1}{2})(b - 2c + d)$$

with weights defined as

$$w_0 = \frac{\alpha_0}{\alpha_0 + \alpha_1 + \alpha_2}, w_2 = \frac{\alpha_2}{\alpha_0 + \alpha_1 + \alpha_2},$$

$$\alpha_0 = \frac{1}{(\delta + \beta_0)^2}, \alpha_1 = \frac{1}{(\delta + \beta_1)^2}, \alpha_2 = \frac{1}{(\delta + \beta_2)^2},$$

$$\beta_0 = 13(a - b)^2 + 3(a - 3b)^2,$$

$$\beta_1 = 13(b - c)^2 + 3(b + c)^2,$$

$$\beta_2 = 13(c - d)^2 + 3(3c - d)^2.$$

In the denominators above, we added a small positive number δ to avoid dividing by zero. In the computation, δ is chosen to be 10^{-6} . In practice, the solution is not senstive to the choice of δ .

Next we have to compute the take-off angle ϕ and out-of-plane curvature τ_{yy} . To match with the evolution form of the eikonal equation in depth, we formulate the advection equation for take-off angles as an evolution equation in depth as well, i.e.,

$$\frac{\partial \phi}{\partial z} = -\left(\frac{\partial \tau}{\partial z}\right)^{-1} \frac{\partial \tau}{\partial x} \frac{\partial \phi}{\partial x}.$$
 (A-4)

To fully take advantage of the accuracy of traveltimes produced by the WENO Runge-Kutta third-order scheme for the eikonal equation and simplify the implementation, we embed the third-order scheme for equation (A-4) into the third-order scheme for the eikonal equation. Because the coefficient of the discretized advection equation has only second-order accuracy, which is computed from the eikonal equation by the third-order scheme, we use a second-order upwind WENO scheme to approximate the derivatives $\frac{\partial \phi}{\partial x}$. The advection equation for τ_{yy} is treated similarly. See Qian (2000) for details.

APPENDIX B-ESTIMATE THE INITIAL STEP

To initialize the traveltime for finite-difference schemes, we assumed that the velocity near the source is constant and equal to the velocity at the source. Now we desire to analyse the traveltime error due to this assumption and furthermore compute an *a priori* estimate of the initial step.

Assuming that the source is at the origin, we consider the two-dimensional raytracing equation. By the method of characteristics for the eikonal equation, we have

$$\dot{x} = v^2 p, \tag{B-1}$$

$$\dot{z} = v^2 q,\tag{B-2}$$

$$\dot{p} = -\frac{1}{v} \frac{\partial v}{\partial x},\tag{B-3}$$

$$\dot{q} = -\frac{1}{v} \frac{\partial v}{\partial z},\tag{B-4}$$

where the dot \cdot denotes the differentiation with respect to time t along the ray; $p = \frac{\partial \tau}{\partial x}$ and $q = \frac{\partial \tau}{\partial z}$.

Denoting the group angle as θ , we have

$$\dot{x} = v \sin \theta, \tag{B-5}$$

$$\dot{z} = v \cos \theta. \tag{B-6}$$

Furthermore, equations (B-1) and (B-2) yield

$$p = \frac{\sin \theta}{v(x, z)},\tag{B-7}$$

$$q = \frac{\cos \theta}{v(x, z)}. ag{B-8}$$

Differentiating equation (B-7) with respect to time t and simplifying the resultant equation, we have

$$\dot{\theta} = -\cos\theta \frac{\partial v}{\partial x} + \sin\theta \frac{\partial v}{\partial z}.$$
 (B-9)

Now we introduce polar coordinates, i.e.

$$x = r\sin\psi,\tag{B-10}$$

$$z = r\cos\psi. \tag{B-11}$$

Differentiating equations (B-10) and (B-11) with respect to time t and solving for \dot{r} and $\dot{\psi}$, we have

$$\dot{r} = v\cos(\theta - \psi),\tag{B-12}$$

$$\dot{\psi} = -\frac{v}{r}\sin(\theta - \psi). \tag{B-13}$$

Next we want to estimate $(\theta - \psi)$. First of all, we have $|\theta - \psi| < \pi$, since for the downward wave propagation both θ and ψ lie in the interval $(-\frac{\pi}{2}, \frac{\pi}{2})$. Defining

$$a(t) = \dot{\theta},\tag{B-14}$$

$$b(t) = \frac{vt}{r} \frac{\sin(\theta - \psi)}{(\theta - \psi)},\tag{B-15}$$

by (B-9) and (B-13) we have an ordinary differential equation for $(\theta - \psi)$,

$$\dot{\theta} - \dot{\psi} = a(t) - \frac{b(t)}{t} (\theta - \psi). \tag{B-16}$$

Its solution is

$$\theta - \psi = \int_0^t d\tau a(\tau) \exp\left(-\int_\tau^t d\sigma \frac{b(\sigma)}{\sigma}\right). \tag{B-17}$$

Because $b(t) \geq 0$ and the function a is bounded by a_{\max} , which is equal to the supremum of the length of gradient of the velocity, i.e., $|a| \leq a_{\max}$, equation (B-17) yields an estimate for $\theta - \psi$,

$$|\theta - \psi| \le a_{\text{max}}t. \tag{B-18}$$

Now we are ready to get an approximate relative error estimate for the traveltime. Denote t_0 as the approximation to the exact traveltime t when we are using the constant velocity v_0 at the source as the approximation to the exact velocity v. Since

$$\dot{t_0} = \frac{\dot{r}}{v_0} = \frac{v}{v_0} \cos{(\theta - \psi)},$$
 (B-19)

we have

$$\dot{t_0} - \dot{t} = (\frac{v}{v_0} - 1)\cos(\theta - \psi) + \cos(\theta - \psi) - 1;$$
 (B-20)

furthermore,

$$|\dot{t_0} - \dot{t}| \le \left| \frac{v}{v_0} - 1 \right| + |\cos(\theta - \psi) - 1|.$$
 (B-21)

Noticing that if $|\dot{t_0}-\dot{t}|\leq arepsilon,$ then $|t_0-t|\leq arepsilon t.$ So let's specify that

$$\left|\frac{v}{v_0} - 1\right| \le \frac{\varepsilon}{2} \tag{B-22}$$

and

$$|\cos(\theta - \psi) - 1| \le \frac{\varepsilon}{2}.$$
 (B-23)

Expanding v at the origin (the source) by Taylor theorem with remainder, we have

$$v(x,z) = v_0 + \frac{\partial v}{\partial x}(\zeta_1, \eta_1)x + \frac{\partial v}{\partial z}(\zeta_2, \eta_2)z,$$
 (B-24)

where (ζ_1, η_1) and (ζ_2, η_2) lie in

$$D = \{(\zeta, \eta) : \min(x, 0) \le \zeta \le \max(x, 0), 0 \le \eta \le z\}.$$
 (B-25)

As a consequence,

$$|v(x,z) - v_0| \le \sqrt{2}r \sup\{|\nabla v(\zeta,\eta)| : |\zeta| \le |x|, 0 \le \eta \le z\}$$
 (B-26)

by Cauchy inequality.

Because we are only bounding the error inside the aperture,

$$|x| \le z \tan \theta_{\max}, r \le \frac{z}{\cos \theta_{\max}};$$
 (B-27)

it follows that

$$\left| \frac{v - v_0}{v_0} \right| \leq \frac{\sqrt{2}r}{v_0} \sup\{ |\nabla v(\zeta, \eta)| : |\zeta| \leq z \tan \theta_{\max}, 0 \leq \eta \leq z \}
\leq \frac{\sqrt{2}zB}{v_0 \cos \theta_{\max}},$$
(B-28)

where z_{max} is the maximum depth and

$$B = \sup\{|\nabla v(\zeta, \eta)| : |\zeta| \le z_{\text{max}} \tan \theta_{\text{max}}, 0 \le \eta \le z_{\text{max}}\}.$$
 (B-29)

For (B-22) to hold, by (B-28) we should choose z such that

$$z \le z_1 = \frac{v_0 \varepsilon \cos \theta_{\text{max}}}{2\sqrt{2}B}.$$
 (B-30)

Finally we choose z so that (B-23) holds, and we need a lemma to do so.

Lemma 1 Along a ray segment $\{(x(\tau), z(\tau)) : 0 \le \tau \le t\}$, the following inequality holds:

$$t \le \frac{r}{v_{\min}},\tag{B-31}$$

where $r = \sqrt{x^2(t) + z^2(t)}$; v_{min} is the minimum velocity along the ray segment.

Proof Denote the true ray path as s and its length |s|, and the straight ray path as l and its length |l| which is equal to r. In addition, we are using l to approximate the true ray path s. Then by Fermat's principle, we have

$$t = \int_{s} d\sigma \frac{1}{v} \le \int_{l} d\sigma \frac{1}{v} \le \int_{l} d\sigma \frac{1}{v_{\min}} = \frac{r}{v_{\min}}.$$
 (B-32)

Using (B-18) and **Lemma 1**, we have

$$|\cos(\theta - \psi) - 1| = \left| -2\sin^2\frac{(\theta - \psi)}{2} \right| \le \frac{r^2 B^2}{v_{\min}^2},$$
 (B-33)

where we have used the relation $a_{\text{max}} \leq B$ inside the aperture. Hence to make (B-23) hold implies that

$$z \le z_2 = \sqrt{\frac{\varepsilon}{2}} \frac{v_{\min} \cos \theta_{\max}}{B}.$$
 (B-34)

So for error tolerance $\varepsilon,\,z_{\mathrm{init}}$ should be chosen such that

$$z_{\text{init}} = \min(z_1, z_2). \tag{B-35}$$

Although both z_1 and z_2 depend on B (the bound of gradient of velocity model), there are at least two ways to estimate B. One way is simply setting B to be a big number which is larger than the actual value; the other way is computing the gradient of velocity model from the given discretized model. Both ways will produce a reasonable initial step.

TABLES

TABLE 1. Fixed-grid eikonal solver: a constant velocity model

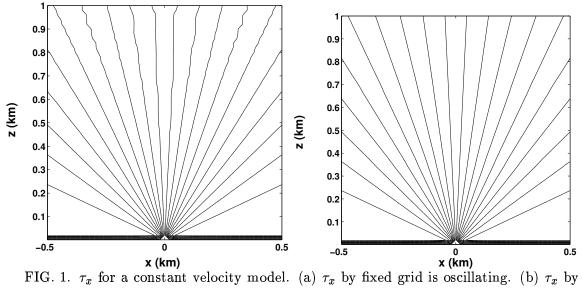
dx	$\mathrm{Abs.Err}(\tau,\mathrm{dx})(\mathrm{s})$	Flops
0.01	0.001232	261,590
0.00125	0.000219	16,632,765

TABLE 2. Adaptive-grid eikonal solver: a constant velocity model

ϵ	$\mathrm{Abs.Err}(\tau,\mathrm{dx})(\mathrm{s})$	Flops
0.000025	0.001041	39,815
0.00000169	0.000160	928,770

FIGURES

- FIG. 1. τ_x for a constant velocity model. (a) τ_x by fixed grid is oscillating. (b) τ_x by adaptive grid is convergent.
- FIG. 2. ϕ_x at z=1 for a constant velocity model. (a) Fixed grid; solid line (-): true solution; star (*): computed solution. (b) Adaptive grid; solid line (-): true solution; star (*): computed solution.
- FIG. 3. Relative errors in ϕ_x . (a) Fixed grid: maximum relative error is almost 45 percent. (b) Adaptive grid: maximum relative error is less than 1.5 percent.
- FIG. 4. 2-D amplitude with a line source for a constant velocity model. (a) The amplitude by fixed grid is divergent. (b) The amplitude by adaptive grid is convergent.
- FIG. 5. (a) τ_{yy} at z=1 for a constant velocity model by adaptive grid; solid line (-): true solution; star (*): computed solution. (b) 2-D amplitude with a point source for a constant velocity model by adaptive grid.
- FIG. 6. The impulse response by inversion with adaptive-gridding WENO traveltimeamplitude solver. The Beylkin determinant needed in the inversion is computed by using the information from traveltimes and takeoff angles, and the response is smooth as expected.



adaptive grid is convergent.

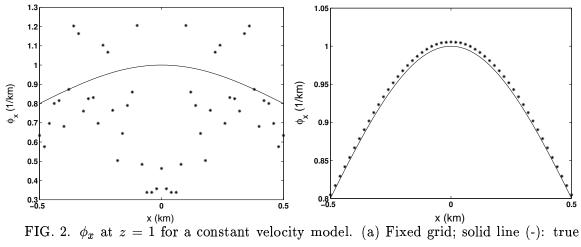


FIG. 2. ϕ_x at z=1 for a constant velocity model. (a) Fixed grid; solid line (-): true solution; star (*): computed solution. (b) Adaptive grid; solid line (-): true solution; star (*): computed solution.

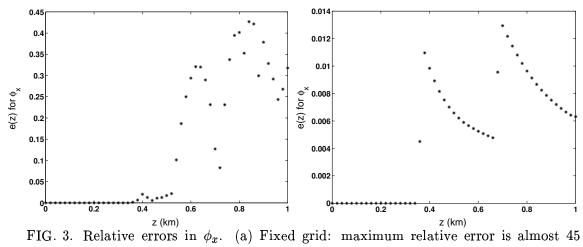


FIG. 3. Relative errors in ϕ_x . (a) Fixed grid: maximum relative error is almost 4. percent. (b) Adaptive grid: maximum relative error is less than 1.5 percent.

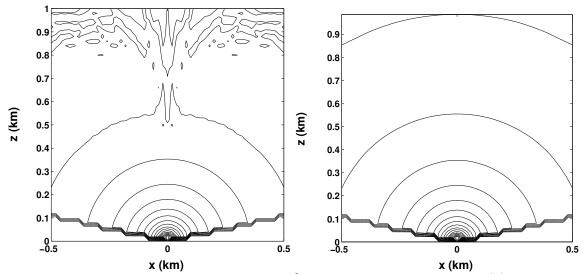


FIG. 4. 2-D amplitude with a line source for a constant velocity model. (a) The amplitude by fixed grid is divergent. (b) The amplitude by adaptive grid is convergent.

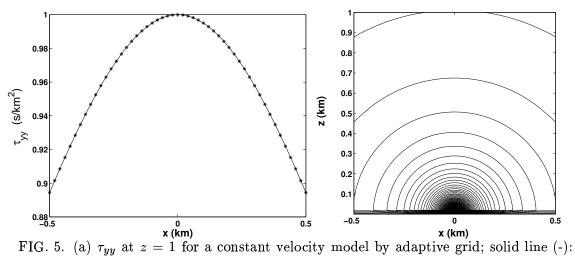


FIG. 5. (a) τ_{yy} at z=1 for a constant velocity model by adaptive grid; solid line (-): true solution; star (*): computed solution. (b) 2-D amplitude with a point source for a constant velocity model by adaptive grid.

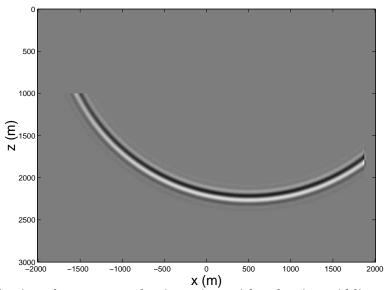


FIG. 6. The impulse response by inversion with adaptive-gridding WENO traveltime-amplitude solver. The Beylkin determinant needed in the inversion is computed by using the information from traveltimes and takeoff angles, and the response is smooth as expected.