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Recommended Citation

Dabiri, J.O., S. Bose, Gemmell, S.P.Colin, J.H. Costello.2013. "An algorithm to estimate unsteady and quasi-steady pressure fields from velocity field measurements," *Journal of Experimental Biology* 217 (3): 331-336.

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1 **An algorithm to estimate unsteady and quasi-steady pressure fields from velocity field**
2 **measurements**

3

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15

16 Short title: Pressure estimation from velocity fields

17

18 Keywords: swimming, flying, wakes, feeding, particle image velocimetry

19

20 *Submitted to Methods and Techniques section of Journal of Experimental Biology*

21

22 **Summary**

23 We describe and characterize a method for estimating the pressure field corresponding to
24 velocity field measurements, such as those obtained by using particle image velocimetry. The
25 pressure gradient is estimated from a time series of velocity fields for unsteady calculations or
26 from a single velocity field for quasi-steady calculations. The corresponding pressure field is
27 determined based on median polling of several integration paths through the pressure gradient
28 field in order to reduce the effect of measurement errors that accumulate along individual
29 integration paths. Integration paths are restricted to the nodes of the measured velocity field,
30 thereby eliminating the need for measurement interpolation during this step and significantly
31 reducing the computational cost of the algorithm relative to previous approaches. The method is
32 validated by using numerically-simulated flow past a stationary, two-dimensional bluff body and
33 a computational model of a three-dimensional, self-propelled anguilliform swimmer to study the
34 effects of spatial and temporal resolution, domain size, signal-to-noise ratio, and out of plane
35 effects. Particle image velocimetry measurements of a freely-swimming jellyfish medusa and a
36 freely-swimming lamprey are analyzed using the method to demonstrate the efficacy of the
37 approach when applied to empirical data.

40 **Introduction**

41 A longstanding challenge for empirical observations of fluid flow is the inability to
42 directly access the instantaneous pressure field using techniques analogous to those established
43 to measure the velocity field. Recent approaches have made significant progress, especially in
44 the measurement of pressure associated with unsteady fluid-structure interactions (e.g. Hong and
45 Altman, 2008; Jardin et al. 2009a, 2009b; David et al., 2009; Rival et al. 2010a, 2010b; David et
46 al., 2012; Tronchin et al., 2012; van Oudheusden, 2013; Liu and Katz, 2013). However, prior
47 efforts have not achieved explicit pressure estimation for moving bodies with time-dependent
48 shape, such as those characteristic of animal locomotion and feeding. The pressure field of
49 swimming animals is complicated by the interaction between pressure associated with vortices in
50 the flow and the irrotational pressure field due to acceleration of the body, often referred to as the
51 acceleration reaction or added mass (Daniel, 1984).

52

53 Existing methods for empirical pressure estimation often require relatively complex
 54 measurement techniques such as multi-camera or time-staggered, multi-exposure particle image
 55 velocimetry (Jensen and Pedersen, 2004; Liu and Katz, 2006). In addition, significant
 56 computational costs can be associated with the post-processing required to derive the pressure
 57 field from measurements of the velocity or acceleration fields. These post-processing approaches
 58 generally fall into one of two categories. In the first case, the pressure field is computed as a
 59 solution to a Poisson equation, e.g. in an inviscid flow:

$$61 \quad \nabla^2 p = -\rho \left(\nabla \cdot \frac{D\mathbf{u}}{Dt} \right), \quad (1)$$

62
 63 where p is the pressure, \mathbf{u} is the velocity vector, ρ is the fluid density, and D/Dt is the material
 64 derivative, i.e., the time rate of change of an idealized infinitesimal fluid particle in the flow.
 65 Solution of equation (1) poses challenges in practice because measurement errors accumulate
 66 due to the required temporal and spatial derivatives of \mathbf{u} , the condition number (i.e. sensitivity)
 67 of the Laplacian operator (Golub and Van Loan, 1996), and measurement uncertainty in the
 68 boundary conditions, especially at fluid-solid interfaces (Gurka et al., 1999). For attached flows
 69 at high Reynolds number, the Neumann boundary condition specifying the pressure gradient at
 70 fluid-solid interfaces is given by the boundary layer approximation as $\partial p / \partial n \approx 0$, where n is the
 71 direction of the local normal surface vector (Rosenhead, 1963). However, for separated flows at
 72 moderate or low Reynolds numbers, such as those commonly found in animal locomotion, *a*
 73 *priori* determination of the appropriate fluid-solid boundary conditions for solution of (1) can be
 74 intractable.

75 A second category of approaches for pressure field estimation is those based on direct
 76 integration of the pressure gradient term in the Navier-Stokes equation, e.g. for incompressible
 77 flow:

$$79 \quad \nabla p = -\rho \left(\frac{D\mathbf{u}}{Dt} - \nu \nabla^2 \mathbf{u} \right) \quad (2)$$

80

81 where ν is the kinematic viscosity of the fluid. The pressure difference between two points in the
82 domain is determined by integration of equation (2) between the two points. For example, the
83 difference in pressure between two points \mathbf{x}_1 and \mathbf{x}_2 is given by

84

$$85 \quad p_2 - p_1 = \int_{\mathbf{x}_1}^{\mathbf{x}_2} \nabla p d\mathbf{x} \quad (3)$$

86

87 Because measurement errors accumulate along the path of integration from \mathbf{x}_1 to \mathbf{x}_2 in equation
88 (3), various techniques have been employed to make this approach less sensitive to measurement
89 uncertainty. A common strategy is to take advantage of the scalar property of the pressure field,
90 such that its local value is independent of integration path. Therefore, each independent
91 integration path that arrives at a point in the flow is in principle an independent estimate of the
92 pressure at that point, provided that measurement errors are uncorrelated. By polling a large
93 number of integration paths, an estimate of the local pressure can be achieved. For example, one
94 successful method (Liu and Katz, 2006) uses an iterative scheme that averages $2m(n+m) +$
95 $2n(2m+n)$ integration paths on an $m \times n$ grid in order to estimate the instantaneous pressure field.

96 The aforementioned iterative scheme, while effective in limiting the influence of
97 measurement errors, still incurs a relatively high computational cost. For example, for a $128 \times$
98 128 grid of velocity vectors that is commonly acquired using PIV, the method requires 1.6×10^5
99 integration paths per iteration of velocity field integration; and several iterations can be required
100 for convergence of the method (Liu and Katz, 2006). Furthermore, if each integration path is
101 taken as a straight line through the domain, then the method requires interpolation of the
102 estimated pressure gradient field in order to evaluate integration path points that do not coincide
103 with the original data grid. While these requirements are not necessarily prohibitive for two-
104 dimensional calculations, they are time-consuming and are indeed a showstopper for extension
105 of the method to three dimensions.

106 We present a simple yet demonstrably effective approach for pressure estimation that is
107 in the spirit of the second category of pressure estimation methods. The method is validated by
108 using two numerically-simulated flows: flow past a two-dimensional, stationary bluff body and
109 the flow created by a three-dimensional, self-propelled anguilliform swimmer. The first flow is
110 used to characterize a quasi-steady implementation of the algorithm, in which the pressure field

111 is estimated from a single velocity field measurement. The second flow demonstrates the ability
 112 of the method to accurately estimate the pressure on unsteady, deformable bodies such as those
 113 of relevance in animal locomotion. Both flows are used to characterize the method, including its
 114 numerical convergence properties and sensitivity to domain size, signal-to-noise ratio, and out of
 115 plane effects. Furthermore, we apply the method to PIV measurements of a freely-swimming
 116 jellyfish medusa and a freely-swimming lamprey, showing that this tool can be applied to the
 117 type of measurement data commonly acquired in research.

118

119 **Materials and Methods**

120 *Material acceleration estimation*

121 The instantaneous fluid particle acceleration $D\mathbf{u}/Dt$ required for calculation of the
 122 pressure gradient in equation (2) is estimated by advecting idealized infinitesimal fluid particles
 123 in the measured velocity fields. For quasi-steady estimation, the material acceleration is derived
 124 from a single velocity field as

125

$$126 \quad \frac{D\mathbf{u}}{Dt}(\mathbf{x}_i) \approx \frac{\mathbf{u}_{PIV}(\mathbf{x}_i^a) - \mathbf{u}_{PIV}(\mathbf{x}_i)}{\Delta t} \quad (4)$$

127

128 where $i = 1, 2, \dots, m \times n$ (i.e. the dimensions of the velocity grid), \mathbf{x}_i are the positions of fluid
 129 particles coincident with the grid points in the PIV velocity field, and \mathbf{x}_i^a are the positions of
 130 those fluid particles after being advected by the instantaneous velocity field for a period Δt :

131

$$132 \quad \mathbf{x}_i^a \approx \mathbf{x}_i + \mathbf{u}_{PIV}(\mathbf{x}_i)\Delta t \quad (5)$$

133

134 In order for equations (4) and (5) to remain valid, Δt is limited to values much smaller
 135 than the characteristic time scale of the flow, yet sufficiently large that there is a measurable
 136 change in the fluid particle velocity.

137

138 For many flows, especially those involving accelerating or deforming bodies, the
 139 aforementioned constraint on Δt cannot be satisfied. For these inherently unsteady fluid-structure
 139 interactions, we derive the material acceleration from two sequential velocity fields as

140

141
$$\frac{D\mathbf{u}}{Dt}(\mathbf{x}_i, t_1) \approx \frac{\mathbf{u}_{PIV}(\mathbf{x}_i^a, t_2) - \mathbf{u}_{PIV}(\mathbf{x}_i, t_1)}{t_2 - t_1} \quad (6)$$

142 where $\mathbf{x}_i^a \approx \mathbf{x}_i + \left(\frac{\mathbf{u}_{PIV}(\mathbf{x}_i, t_1) + \mathbf{u}_{PIV}(\mathbf{x}_i, t_2)}{2} \right) (t_2 - t_1)$ (7)

143

144 Equation (7) is akin to a Crank-Nicolson (i.e. trapezoidal) scheme for the particle positions, in
 145 contrast to the forward Euler scheme in equation (5). Hence, the convergence of the method with
 146 time step is second order (Crank and Nicolson, 1947).

147 The primary source of measurement error in this type of unsteady estimate of the material
 148 acceleration $D\mathbf{u}/Dt$ arises from temporal noise in the measured velocity components at each node
 149 in the velocity field. We address this by applying a temporal filter to the time series of velocity
 150 fields, which results in a smoothing spline approximation \mathbf{u}^* to the velocity \mathbf{u} at each node in the
 151 velocity field. The spline approximations are defined such that they minimize, for each
 152 component of \mathbf{u} , the parameter

153

154
$$S_{\mathbf{u}} = \phi \sum_{\tau=1}^N (\mathbf{u}_{\tau} - \mathbf{u}_{\tau}^*)^2 + (1 - \phi) \int_{t_{\min}}^{t_{\max}} \left(\frac{d^2 \mathbf{u}^*}{dt^2} \right)^2 dt \quad (8)$$

155

156 where $\tau = 1 \dots N$ is the temporal sequence of velocity fields to be analyzed, \mathbf{u}_{τ} is a velocity vector
 157 corresponding to velocity field τ in the sequence, \mathbf{u}_{τ}^* is the spline-approximated value of the
 158 same velocity vector for the same velocity field in the sequence, t_{\min} and t_{\max} are the temporal
 159 bounds on the sequence of velocity fields, and ϕ is a weight between the first and second terms
 160 and has a value between 0 and 1. In effect, the parameter $S_{\mathbf{u}}$ quantifies both the deviation of the
 161 spline approximation from the original data (i.e. the first term) and the total curvature magnitude
 162 of the spline approximation (i.e. the second term). For $\phi = 0$, only the second term is minimized,
 163 resulting in a least-squares fit with zero curvature, i.e. a linear fit to the data. For $\phi = 1$, only the
 164 first term is minimized, giving a cubic spline fit that passes through each original data point. In
 165 all that follows, we set $\phi = 0.05$, a value we have identified as enabling effective temporal noise
 166 filtering without discarding true temporal trends in the measurement data.

167 Further characterization of the temporal filter is provided in Appendix 2. In particular, it
168 is shown that the use of the temporal filter increases the order of temporal convergence above
169 second order, as anticipated by theory (Atkinson, 1968).

170 It is worth noting that the distinction between the quasi-steady and unsteady approaches
171 can be made explicit by decomposing the material acceleration into its Eulerian components:

172

$$173 \quad \frac{D\mathbf{u}}{Dt} \equiv \frac{\partial\mathbf{u}}{\partial t} + (\mathbf{u} \cdot \nabla)\mathbf{u} \quad (9)$$

174

175 The quasi-steady approximation in equations (4) and (5) implicitly neglects the first term on the
176 right-hand side of equation (9), whereas the unsteady calculation retains it.

177 The viscous term on the right-hand side of equation (2) is computed using centered finite
178 differences between adjacent nodes in the velocity field. The effect of the viscous term is
179 evaluated in the context of a numerical simulation described in the validation section.

180

181 *Pressure gradient integration*

182 Whereas previous methods that integrate the pressure gradient via many integration paths
183 assign to each grid point the arithmetic mean of the many integrations, in the present approach
184 the paths are polled by taking the median. The median is less sensitive to grossly erroneous
185 values that may arise on a few of the integration paths due to localized measurement errors or
186 due to localized errors created by the aforementioned material acceleration approximations in
187 equations (4) through (7). Hence, this approach enables a significant reduction in the total
188 number of integration paths per frame that are required to achieve accurate pressure estimates.

189 Fig. 1 illustrates the paths used presently. Eight families of integration paths are used, with each
190 family originating at the domain boundary and propagating toward each grid point from the left
191 (L), upper left (UL), top (T), upper right (UR), right (R), lower right (LR), bottom (B), and lower
192 left (LL), respectively.

193 Only 8 integration paths (one per family) per grid point are used, for a total of $8m \times n$
194 paths per velocity field. For the aforementioned example grid of 128×128 velocity vectors, 1.3
195 $\times 10^5$ integration paths are required, a 20 percent reduction from existing optimal methods (Liu
196 and Katz, 2006). More importantly, the integration paths are constrained to include only grid

197 points coincident with the original velocity field. For example, the UL integration path is
198 comprised of alternating integration steps in the -y and +x directions, originating at the domain
199 boundary and terminating at each grid point. Hence, no interpolation is required in order to
200 integrate the pressure gradient field. Furthermore, portions of many of the paths are redundant,
201 facilitating fast calculation using simple matrix manipulations. A forward Euler spatial
202 integration scheme is used throughout, resulting in first-order spatial convergence of the method
203 (see Appendix 1).

204 An important limitation of the present algorithm that arises from the trade-off between
205 speed and accuracy is that it assumes the pressure is zero at the point on the outer domain
206 boundary where each integration path is initiated. This does not imply, however, that the final
207 pressure estimate is constrained to be zero at the boundaries. Integration paths that originate from
208 the other domain boundaries and terminate at a given boundary may estimate a non-zero value of
209 pressure at the termination point. If the median of all paths terminating at that point on the
210 domain boundary is non-zero, then the final pressure estimate at that point will also be non-zero.
211 Note that for all points in the domain, the final pressure estimate is relative to a zero reference
212 pressure, as that is the pressure at the origin of each integration path. The impact of these
213 assumptions on the robustness of the technique is quantified below, and it is shown to be modest
214 for the external flows tested. At the same time, the net result of this tradeoff in the algorithm
215 design is a more than order-of-magnitude reduction in computational time compared to previous
216 methods (see Appendix 2).

217 A common source of localized error that can affect pressure estimates is the presence of
218 solid objects in the flow. Typical PIV measurements are often unreliable in the region close to
219 solid objects, which compromises pressure integration paths that cross the fluid-solid interface,
220 especially in previous methods that average the erroneous data instead of discarding it via
221 median polling (or in Poisson solvers that rely on the pressure gradient at the fluid-solid interface
222 as a boundary condition). In the present algorithm, integration paths that cross a fluid-solid
223 interface in the flow can be nullified by assigning the nodes nearest to the interface an undefined
224 pressure gradient. Hence, when that value is integrated along any integration path, the pressure
225 value for that path also becomes undefined and therefore does not contribute to the median
226 calculation.

227

228 *Validation data sets*

229 To validate the accuracy of the quasi-steady pressure estimates achieved using this
230 algorithm, a numerical simulation of flow past a two-dimensional square cross-section cylinder
231 at a Reynolds number of 100 was used. This numerical data set enabled quantification of the
232 effects of spatial resolution, domain size, and signal-to-noise ratio, while providing a known
233 pressure field standard for comparison (see Appendix 1). The numerical simulation was
234 executed using a solver that computes on arbitrary polyhedra (Ham and Iaccarino, 2004). In the
235 present case, a regular Cartesian mesh was utilized and subsequently interpolated onto coarser
236 grids of varying sizes typical of PIV data. The viscous term in equation (2) was retained in all of
237 the calculations to demonstrate the robustness of the median polling approach to errors normally
238 associated with application of the Laplacian operator. For all calculations of equations (4) and
239 (5) in this validation, we set $\Delta t = 0.01h/U_{max}$, where h is the mean grid spacing and U_{max} is the
240 maximum flow speed in the measurement domain. The results described below were insensitive
241 to order of magnitude larger and smaller values of Δt . Where noted, nearest-neighbor Gaussian
242 smoothing was applied both to the pressure gradient before integration and to the resulting
243 pressure field.

244 The accuracy of the fully unsteady pressure estimates was validated by using a published
245 numerical simulation of a three-dimensional, self-propelled anguilliform swimmer (Kern and
246 Koumoutsakos, 2006). The Reynolds number based on swimmer length and speed was
247 approximately 2400. Time steps between sequential velocity fields from $0.02T$ to $0.08T$ (where T
248 is the swimming stroke duration) were studied to quantify the temporal convergence of the
249 method. The validation results described below are based on calculations of equations (6) and (7)
250 using velocity fields separated by $0.02T$.

251

252 *Empirical data sets*

253 The present method was also applied to particle image velocimetry measurements of a
254 freely-swimming *Aurelia aurita* (Linnaeus, 1758) jellyfish medusa and a freely-swimming
255 *Anguilla rostrata* (LeSueur, 1817) lamprey to demonstrate the performance of the algorithm with
256 empirical data inputs and, in the case of the jellyfish, without treatment of fluid-solid interfaces.
257 The swimming Reynolds numbers of the jellyfish and lamprey were approximately 1000 and
258 10,000, respectively, and the time between sequential velocity fields was 5 ms ($t/T \approx 0.013$) and

259 4 ms ($t/T \approx 0.015$), respectively. Details of the PIV implementation can be found in published
260 literature (Colin et al., 2012).

261

262 **Results and discussion**

263 *Quasi-steady pressure estimation*

264 Fig. 2 compares an instantaneous pressure field from the numerical simulation of flow
265 past a stationary bluff body with the pressure field estimated from the corresponding velocity
266 field using the quasi-steady form of the present algorithm. A vector field spatial resolution of
267 $D/16$ (where D is the side length of the bluff body) is used in the horizontal and vertical
268 directions to mimic typical PIV measurements. The salient features of the flow, especially the
269 high pressure on the upstream face of the bluff body and the low pressure in the shear layers and
270 near-wake vortices, are well captured by the algorithm (see Appendix 1 for discussion of
271 discrepancies in the far wake). Furthermore, the maximum and minimum pressures in the field
272 are in quantitative agreement (Fig. A1). To be sure, nearest-neighbor Gaussian smoothing creates
273 a spurious thin layer of undefined pressure at the fluid-solid interface and moves the pressure
274 peak on the upstream face of the body away from the interface. However, the correct near-body
275 pressure can be recovered by increasing the grid resolution so that the nearest-neighbor filter
276 artifact on the body surface is limited to a smaller region very close to the body. Additional
277 surface pressure calculations for the quasi-steady case (Appendix 1) are based on a velocity
278 vector spacing of $D/64$. Note that a similar increase in resolution using a PIV camera would
279 require a concomitant reduction in the measurement window size due to limits on camera pixel
280 resolution.

281 Additional characterization of the quasi-steady algorithm is detailed in Appendix 1,
282 including analysis of spatial convergence; the relative contribution of each integration path to the
283 median pressure field; robustness to measurement noise; and the effects of domain size, fluid
284 viscosity, and fluid-solid interfaces.

285

286 *Unsteady pressure estimation*

287 Fig. 3 compares an instantaneous pressure field from the numerical simulation of a self-
288 propelled anguilliform swimmer with the pressure field estimated from the corresponding
289 velocity field using the unsteady form of the present algorithm. A vector field spatial resolution

290 of $L/42$ (where L is the length of the swimmer) is used in the horizontal and vertical directions.
291 No smoothing is applied to this data set in order to contrast the results with those in the previous
292 section and to limit the spatial extent of the region of undefined pressure near the fluid-solid
293 interface. The algorithm is effective in capturing the high-low pressure couples formed on the
294 sides of the swimmer head and tail as they accelerate in the positive- y direction; the low-high
295 pressure couple formed at the mid-body as it accelerates in the negative- y direction; and the
296 pressure in the wake vortices.

297 The importance of the unsteady term in equation (9) is illustrated by comparison with the
298 pressure field estimated using the quasi-steady approximation, shown in Fig. 3C. Low pressure
299 in the wake vortices is captured, but the high-low pressure couples on the body surface due to the
300 body added mass are missing entirely, as is the high pressure in the wake due to vortex added
301 mass (Dabiri, 2006). The comparison is further quantified in Fig. 4, which plots the pressure on a
302 contour surrounding the swimmer and immediately adjacent to the region of undefined pressure.
303 At each of the four phases of the swimming cycle shown, good agreement is achieved between
304 the pressure computed in the numerical simulation and the pressure estimated from the velocity
305 field using the unsteady algorithm. By contrast, the pressure estimated by the quasi-steady
306 algorithm is erroneous everywhere except near the forming wake vortex at the tail.

307 Additional characterization of the unsteady algorithm is provided in Appendix 2,
308 including analysis of temporal convergence and out-of-plane effects for three-dimensional flows.

309 To demonstrate the efficacy of the present method for analyzing empirically measured
310 velocity fields, Fig. 5 shows measured velocity and vorticity fields for the freely-swimming
311 jellyfish and lamprey (panels A and C) along with the corresponding pressure fields estimated
312 using the unsteady algorithm (panels B and D). The full measurement domain is shown in both
313 cases; the velocity vector field is plotted at half of the full resolution. Only the left half of the
314 jellyfish body is visible in the measurement domain; its exumbrellar surface is indicated by a
315 black curve in panels A and B. The full lamprey body is visible in panels C and D.

316 In both cases, the pressure field derived from the velocity field measurements captures
317 key features near the body surface and in the wake. In particular, the jellyfish data set indicates
318 low pressure in the forming starting vortex and high pressure where the bell margin is
319 accelerating inward and pushing the adjacent fluid. The results are consistent with the measured
320 vorticity field (panel C), with the region of low pressure corresponding to the core of the starting

321 vortex. The presence of low and high pressure regions near the bell margin is also in agreement
322 with previous numerical simulations of a swimming jellyfish with similar body shape and
323 kinematics (Sahin et al., 2009).

324 The lamprey data set shares similarities with the three-dimensional numerical model
325 shown previously. The vorticity and pressure fields are less smooth and show finer structure in
326 the empirical measurements, which is attributable in part to the Reynolds number being
327 approximately four times higher than that of the numerical simulation.

328 The ease of implementation of this algorithm, both in terms of data acquisition and
329 velocity field post-processing, and its relatively low computation cost (see Appendix 2) gives it
330 the potential to find use in a broad range of problems of interest in biological fluid mechanics.
331 Because the temporal filter implemented in the unsteady algorithm does add considerable time to
332 the pressure calculation (cf. Fig. A10), in practice one should first evaluate the results of both the
333 quasi-steady and the fully unsteady implementations of the algorithm on a sample of the data of
334 interest to determine whether unsteady effects are important. If they are not, then the quasi-
335 steady calculation provides the most efficient tool for determination of the pressure field.

336 Although the present evaluation focused on two-dimensional velocity fields, it is
337 straightforward to extend the algorithm to three dimensions by the addition of a limited number
338 of new integration paths consistent with the geometry in Fig. 1. In that case, even greater
339 reductions in computation expense can be achieved relative to existing methods due to the
340 relatively small total number of required integration paths and the elimination of associated
341 velocity field interpolation during integration of the pressure gradients.

342 A free MATLAB implementation of this algorithm is available at
343 <http://dabiri.caltech.edu/software.html>.

344

345 **Appendix 1: Additional Characterization of Quasi-steady Algorithm**

346 *Effect of median polling*

347 To illustrate the contribution of each integration path to the final pressure estimate, Fig.
348 A1 plots the pressure on the body surface (at $0.1D$ away from the fluid-solid interface, to avoid
349 the spatial filter artifact) and on two additional concentric square contours in the domain (e.g. the
350 dashed contour in Fig. A2A), as computed using each of the 8 families of integration paths. The
351 results illustrate the benefit of median polling versus an average of the integration path results.

352 For example, only 5 of the integration path families intersect the upstream face of the bluff body
353 without passing through the body itself. The median of these curves is in good quantitative
354 agreement with the correct surface pressure (Fig. A1A). The pressure profiles for the two
355 concentric square contours in the domain (i.e. Figs. A1B and A1C) indicate that the contribution
356 of each family to the final pressure estimate is spatially non-uniform. This is illustrated
357 qualitatively in Fig. A1D, which is a contour plot that colors each point in the domain according
358 to the path family that corresponds to the median pressure at that point. Because there are 8 path
359 families, the median is always the average of the two intermediate values (where none of the
360 paths is undefined due to intersection with the solid body). To reveal the individual integration
361 path family contributions, a ninth pressure value equal to the mean of the 8 path families is
362 included in Fig. A1D, so that the median pressure is from either a single integration path family
363 or from the mean. The contour plot indicates that each integration path family contributes to the
364 final pressure field estimate, but the contributions are often spatially localized. The pressure
365 estimates for the R family of integration paths are noticeably less accurate than the other families
366 (e.g. Fig. A1A) and yet, as illustrated in Fig. A1D, these paths determine the pressure estimate in
367 the far wake. This leads to the observed poorer pressure estimate in that region of the flow (e.g.
368 Fig. 2B). The underlying source of this effect is discussed below in the section examining the
369 effect of boundary conditions.

370

371 *Effect of global measurement error*

372 Perhaps the most important test of the algorithm is its robustness to global measurement
373 errors, such as those associated with empirical measurements. Fig. A2 illustrates the streamwise
374 velocity contours for data sets with increasing levels of Gaussian white noise superimposed on
375 the u and v velocity components. The highest levels of noise, corresponding to the lowest signal-
376 to-noise ratios, are higher than typical PIV data but possibly representative of instantaneous two-
377 dimensional data collected in a highly-turbulent flow field, where out-of-plane motion can
378 reduce data quality. Comparison of the pressure profiles on a square contour centered on the
379 bluff body and with side length $3D$ so that it passes through the salient flow features (i.e. Fig.
380 A2A) indicates that, with the exception of the highest noise level tested, the quantitative pressure
381 estimates remain consistent with the noise-free result despite relatively high noise (Fig. A3A).

382 Error in the pressure estimate is not additive with the increasing noise level because

383 errors do not accumulate uniformly on the 8 paths that arrive at each point in the domain. Hence,
384 median polling remains an effective filter irrespective of the noise magnitude, up to the second-
385 highest noise level tested. At higher noise levels, contour plots of the pressure estimate begin to
386 exhibit spatial discontinuities reminiscent of the median contributions in Fig. A1D. Because the
387 pressure estimates from each integration path family begin to diverge in the presence of high
388 noise levels, median polling in this case leads to spatially discontinuous changes in pressure.
389 Result of this sort are an indication that measurement noise in the input velocity data has become
390 unacceptably large.

391

392 *Effect of boundary conditions*

393 As mentioned previously, a major assumption implicit in the present algorithm is that the
394 pressure on each integration path is zero at its originating point on the boundary, to avoid the
395 need for a computationally expensive iteration scheme to solve for the boundary pressure as part
396 of the field solution (Liu and Katz, 2006). Although this assumption can be reasonable for large
397 domains, it is prudent to investigate the dependence of the pressure estimate on the domain size.
398 Fig. A3B plots the pressure on a square contour centered on the bluff body (see Fig. A2A) for
399 domains ranging in size from $H/D = 2$ to 30, where H is the half-width of the domain. The results
400 indicate that the accuracy of the algorithm (and hence, the assumption regarding the boundary
401 pressure) is not significantly compromised until the domain shrinks to $H/D = 2$. This limitation is
402 important to keep in mind when designing experiments that will make use of the present
403 algorithm.

404 Notwithstanding the demonstrated efficacy of the aforementioned assumption regarding
405 the boundary pressure, examination of the individual pressure estimates on each family of
406 integration paths reveals that some individual estimates are severely compromised by this
407 assumption. Most notably, the R family of integration paths originate at the downstream
408 boundary of the domain, where vortices shed by the bluff body exit the measurement window
409 and create a non-zero pressure on that boundary. Hence this family of pressure estimates is
410 significantly less accurate than the others, as seen in Fig. A1A for example. The benefit of the
411 median polling approach is that this estimate is usually discarded in determining the final
412 pressure estimate. In contrast, previous methods would include pressure estimates affected by the
413 downstream boundary in the final averaged pressure estimate, and therefore require additional

414 computational effort to resolve the correct pressure on that boundary via iterative processes.

415 However, the present method does suffer in that the pressure in regions close to the
416 downstream boundary is based either on integration paths that originate at the downstream
417 boundary where the pressure is nonzero (i.e. R, UR, and LR families) or on long integration
418 paths from the other boundaries. The relatively large error accumulated on the long integration
419 paths can make them an even poorer estimate of the local pressure near the R boundary (cf. Fig.
420 A1D); hence the median pressure in this region is less accurate than in the rest of the domain.
421 This limitation is inherent in the present method and should be kept in mind when using the
422 technique for flows with large velocity gradients at any of the boundaries.

423

424 *Effect of fluid viscosity*

425 It is useful to examine the role of the viscous term in equation (2), as many previous
426 pressure estimation methods neglect this term. Fig. A4A plots the pressure estimates on the body
427 surface for each integration path family as in Fig. A1A, but for a pressure estimate that neglects
428 the viscous term in the Navier-Stokes equation. The effect is most noticeable in integration paths
429 orthogonal to the mean flow (i.e. T and B), especially near the upstream face of the bluff body.
430 This result can be understood by considering the contributions to the pressure gradient from the
431 streamwise and transverse material acceleration components, Du/Dt and Dv/Dt , relative to the
432 contributions from the Laplacian of the streamwise and transverse velocity components in the
433 viscous term. As the flow approaches the upstream face of the bluff body, the material
434 acceleration is dominated by streamwise fluid particle deceleration Du/Dt . However, the pressure
435 computed on integration paths that are orthogonal to the mean flow (i.e. T and B) is independent
436 of this term. Instead, on these paths the pressure depends to a significant degree on the local
437 velocity curvature (i.e. second spatial derivative) as the flow is turned around the bluff body.
438 This effect is captured in part by the Laplacian of the transverse velocity. Hence, its neglect leads
439 to an underestimate of the pressure on those integration paths. The net effect of the neglected
440 viscous term is minimal due to the median polling approach implemented presently, i.e., the T
441 and B paths do not represent the median pressure estimate on the upstream face of the bluff body
442 and are therefore not a factor in the final pressure estimate in that region of the flow.

443

444 *Effect of fluid-solid interfaces*

445 An aspect of the present algorithm that can be cumbersome is the treatment of the fluid-
446 solid interface to eliminate the effect of integration paths that pass through solid objects in the
447 flow. For example, for moving objects, this approach requires the identification of the fluid-solid
448 interface in each data frame. To illustrate the effect of the interface treatment in the algorithm,
449 Fig. A4B plots pressure estimates on the body surface, where the algorithm has been
450 implemented without treating the fluid-solid interface. The accuracy of the algorithm is
451 noticeably affected due to additional spurious pressure estimates on paths that cross the body.
452 However, it is noteworthy that the final pressure estimate is still qualitatively consistent with the
453 correct pressure field. It may therefore be acceptable to bypass the fluid-solid interface treatment
454 where only a quantitative approximation of the pressure field is sought. The results of the
455 analysis in Fig. 5B, which did not identify the boundary of the medusa as was done for the
456 numerical data, suggest that suitable pressure estimates can still be achieved where treatment of
457 the fluid-solid interfaces is not practical.

458

459 *Spatial convergence*

460 The spatial convergence of the quasi-steady algorithm was evaluated by computing the
461 pressure on a square contour immediately adjacent to the region of undefined pressure on the
462 bluff body, and by integrating the pressure to compute the net force in the streamwise and lateral
463 directions. Fig. A5 plots the fractional error in these calculations (using the pressure from the
464 numerical simulation (CFD) as the true value, i.e. $|F_{CFD} - F_{estimate}|/F_{CFD}$) versus the grid
465 resolution. At relatively large grid spacing, the log-log curve has a slope of 1, indicating the
466 expected first-order spatial convergence of the method. As the grid spacing is further reduced,
467 the error decreases more slowly. This effect is due to a combination of inherent model error and
468 numerical round-off error. A convergence plot for calculations of the time-averaged streamwise
469 force is included. Its departure from first order convergence at small grid spacing confirms that
470 the quasi-steady approximation is not solely responsible for errors at small grid spacing. For grid
471 spacing less than $D/16$, the error falls below 5 percent for the instantaneous pressure and
472 approaches 10 percent for the time-averaged pressure.

473

474 **Appendix 2: Additional Characterization of Unsteady Algorithm**

475 *Effect of temporal filter*

476 Fig. A6 plots the time series of v component data at two selected points in the jellyfish
477 PIV data set. Despite the relatively smooth spatial distribution of velocity, as illustrated in Fig.
478 5A and Fig. A7A, there is non-trivial scatter in the temporal data at both spatial locations. Flow
479 accelerations computed by using finite differences of adjacent velocity fields would be subject to
480 large errors because the local slope varies considerably between adjacent pairs of velocity fields.
481 A temporal filter of the data is therefore essential in this case. Fig. A6 indicates the
482 corresponding smoothing splines that were fit to the data and subsequently used to compute the
483 material acceleration. The splines capture the true transient behavior of the flow while
484 eliminating the high-frequency noise. Comparison of Fig. A7A and A7B illustrates that the
485 spatial distribution of velocity is relatively unaffected by the temporal filter. It is prudent to note
486 that if a flow exhibits real, high-frequency oscillations in the velocity, e.g. in turbulence, it will
487 be essential that the PIV measurements are of sufficiently high temporal resolution such that the
488 smoothing spline does not discard those temporal trends. In those cases, it is important that the
489 frequency of PIV measurements satisfies the Nyquist sampling criterion with respect to the time
490 scale of turbulence fluctuations, while concurrently avoiding sampling at frequencies high
491 enough to incur numerical round-off errors in the calculation (Beckwith et al., 2007).

493 *Effect of out-of-plane flow*

494 Given that two-dimensional PIV data represents a projection of three-dimensional flow, it
495 is useful to characterize the impact of that limitation on the accuracy of the present methods. As
496 in prior work (Stamhuis and Videler, 1995), Fig. A8 characterizes the out-of-plane motion by
497 computing the divergence of a two-dimensional velocity field extracted from the three-
498 dimensional numerical simulation of the self-propelled swimmer and of the PIV data sets
499 examined in Fig. 5. The divergence is made dimensionless by multiplying it by the time step
500 between adjacent velocity fields, as this time scale is most relevant for calculation of the material
501 acceleration. The plots effectively quantify the fractional change in the volume of an idealized
502 infinitesimal fluid particle between adjacent velocity fields. Because the flows are
503 incompressible, the fractional volume change would be identically zero if the flows were two-

504 dimensional. Deviation from zero values can therefore be attributed to velocity gradients
505 perpendicular to the plane of the velocity field, i.e., out-of-plane flow.

506 The results in Fig. A8 indicate that the three-dimensional numerical simulation exhibits
507 greater out-of-plane flow than the PIV measurements. Given the demonstrated accuracy of the
508 algorithm in the case of the three-dimensional numerical data, we can conclude that the
509 algorithm is robust to out-of-plane effects at the magnitudes found in typical PIV measurements.
510 To be sure, the divergence metric does not capture out-of-plane flow where there is no flow
511 gradient in the perpendicular direction. However, in such cases, the PIV would itself be difficult
512 to acquire, as the seed particles would not remain in the plane of the laser sheet sufficiently long
513 to enable temporal cross-correlation of their positions.

514

515 *Temporal convergence*

516 The temporal convergence of the unsteady algorithm was evaluated by plotting the
517 fractional error in the pressure at the head of the self-propelled swimmer at an instant of high
518 acceleration (using the pressure from the numerical simulation (CFD) as the true value, i.e.
519 $|p_{CFD} - p_{estimate}| / p_{CFD}$) versus the time step between velocity fields (Fig. A9). Although the
520 available data was limited to time steps from $0.02T$ to $0.08T$, the results are consistent with
521 temporal convergence that is higher than second order, except as the smallest step size is
522 approached, where further reduction in error is limited by inherent model error and numerical
523 round-off error. At a temporal spacing of $0.02T$, the error in the pressure at the head is
524 approximately 8 percent.

525 When the unsteady algorithm is applied to a sequence of velocity fields that are spaced
526 too closely in time, leading to increased numerical error, the results appear similar to those
527 described in the context of global measurement error (Appendix 1) in which the pressure
528 contours exhibit spatial discontinuity reminiscent of Fig. A1D.

529

530 *Computational cost*

531 Fig. A10 plots the time required for a single 3-GHz processor to apply the temporal filter
532 and to compute the pressure field for velocity fields from 32×32 to 256×256 nodes, which is a
533 practical upper limit for typical PIV measurements due to camera pixel resolution. The time
534 required for the temporal filter scales linearly with the number of nodes in the velocity field. The

535 cost is independent of the number of velocity fields in the sequence of up to the tested several
536 hundred frames of data. The computational cost of the subsequent pressure calculation scales
537 sublinearly in the range tested, and it is significantly lower than the cost of the temporal filter in
538 absolute terms. For example, for a 128x128 velocity field, each pressure field is computed in
539 approximately 3 seconds, as compared to 46 seconds using an existing iterative algorithm (Liu
540 and Katz, 2006).

541

542

543 **Acknowledgments**

544 The authors gratefully acknowledge S. Kern and P. Koumoutsakos for providing the three-
545 dimensional numerical simulation of the self-propelled swimmer, and X. Liu for providing
546 access to the pressure gradient integration code used in Liu and Katz (2006). This research was
547 supported by Office of Naval Research awards N000140810918 and N000141010137 to J.O.D.

548

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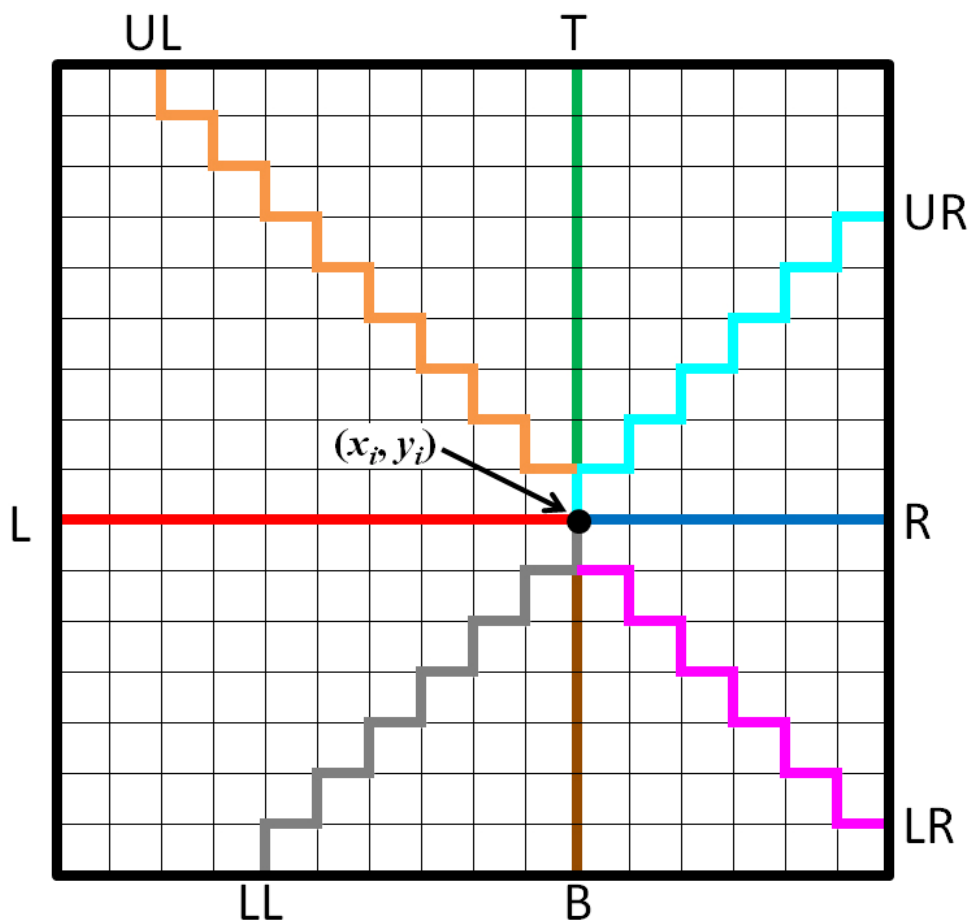
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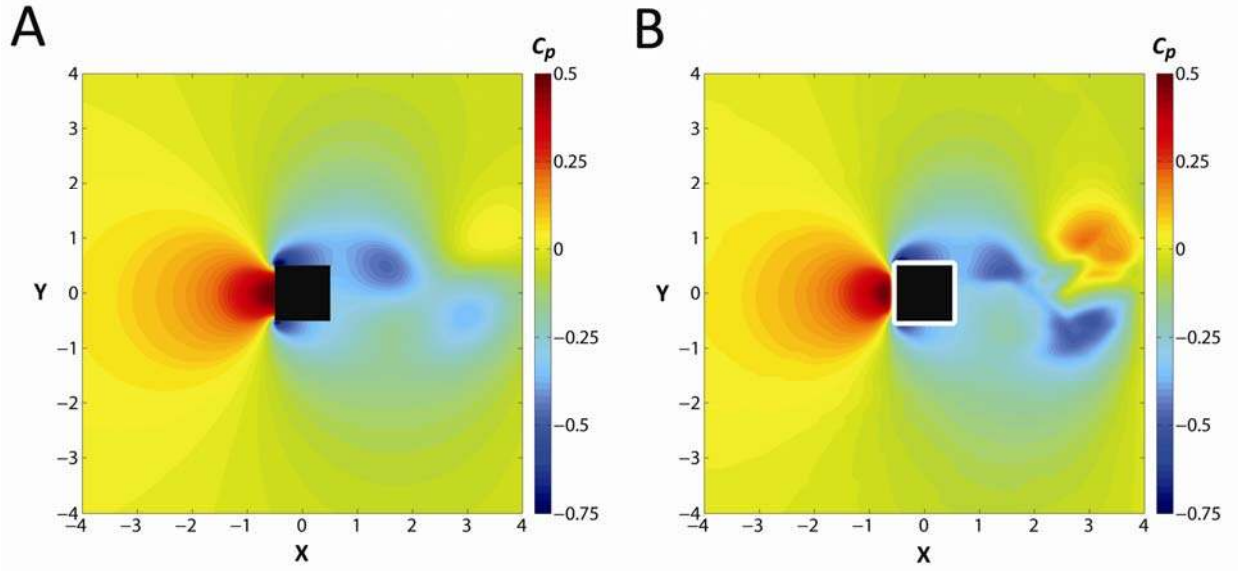
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606

607 **Figure 1.** Geometry of integration paths for pressure field estimation. Eight paths originate from
 608 the domain boundary and propagate toward each point (x_i, y_i) in the domain from the left (L),
 609 upper left (UL), top (T), upper right (UR), right (R), lower right (LR), bottom (B), and lower left
 610 (LL). The points on each path coincide with the measurement grid.

611



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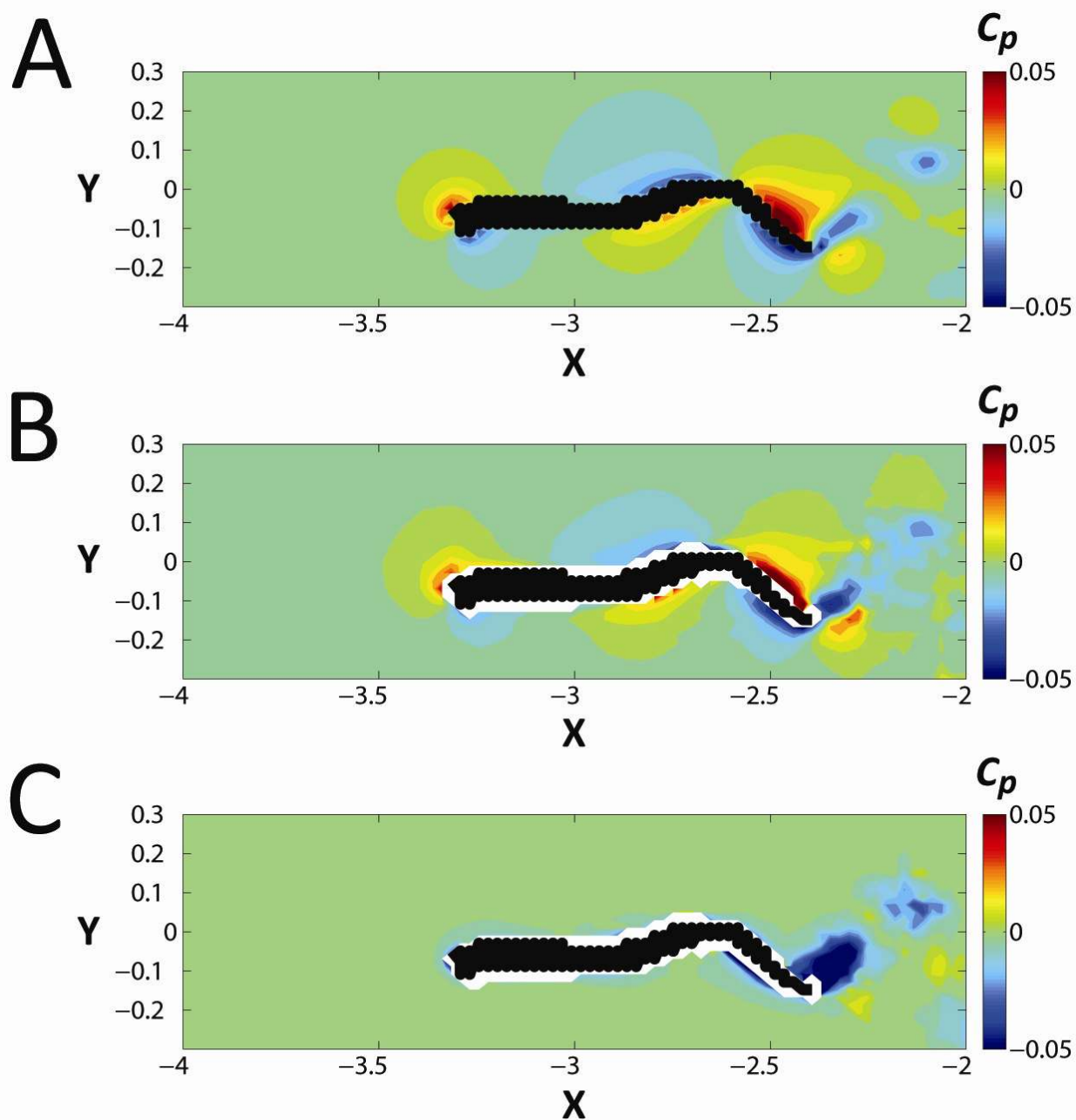
Figure 2. (A) Pressure field computed from numerical simulation of flow past two-dimensional square cylinder at a Reynolds number $Re = UD/\nu = 100$. The pressure coefficient $c_p = p/(\rho U^2)$.

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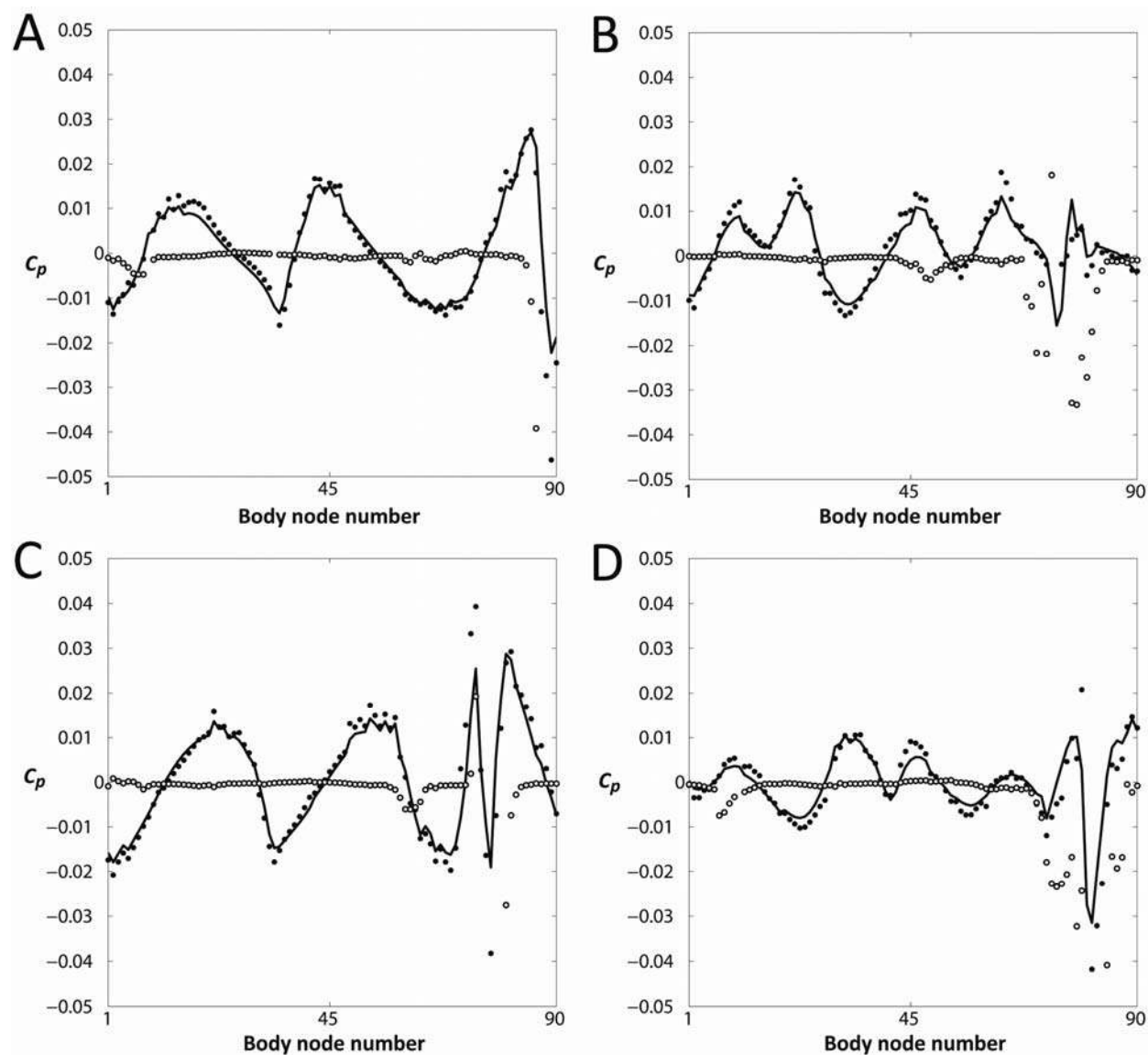
(B) Pressure field estimated using quasi-steady algorithm.

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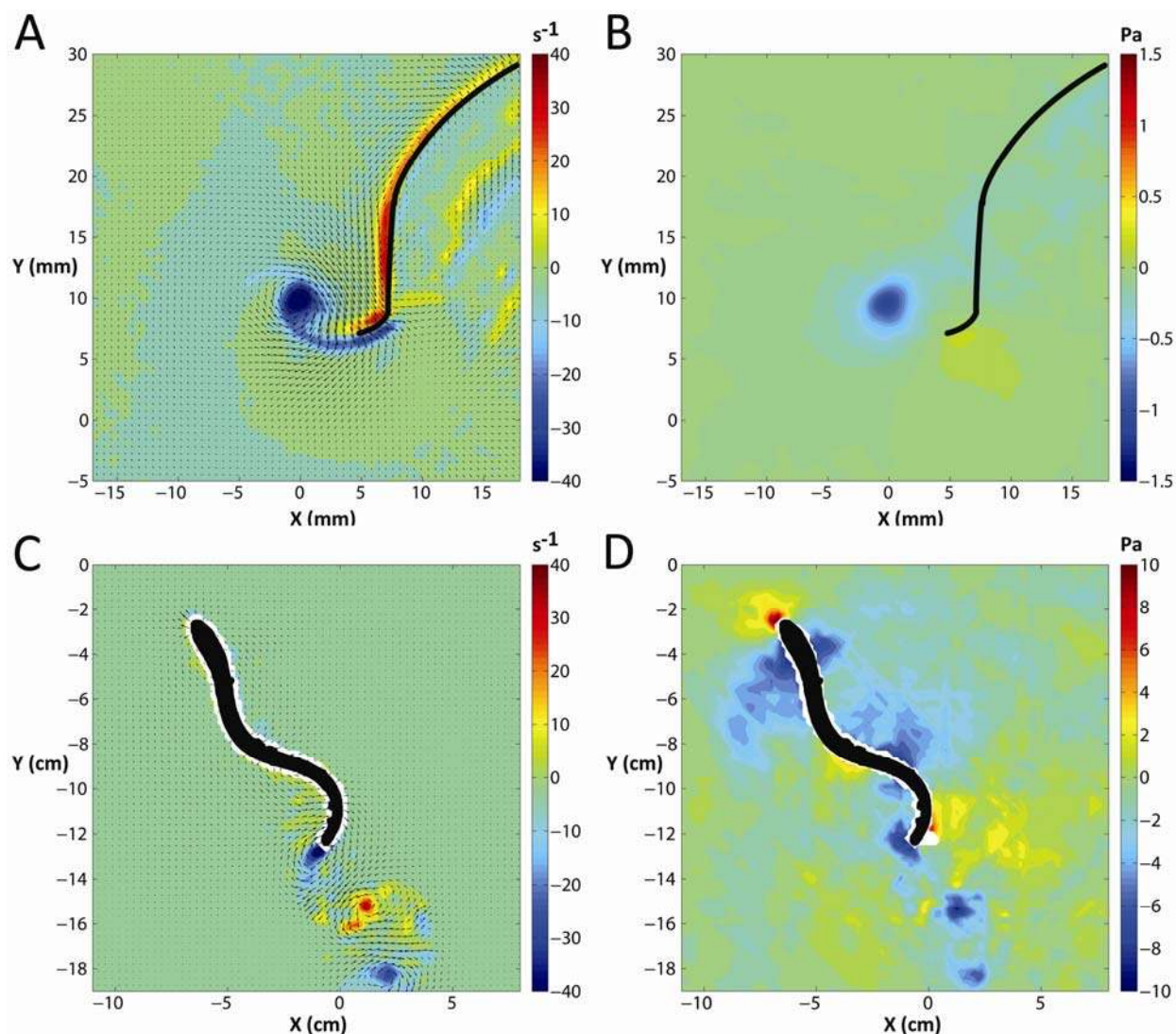
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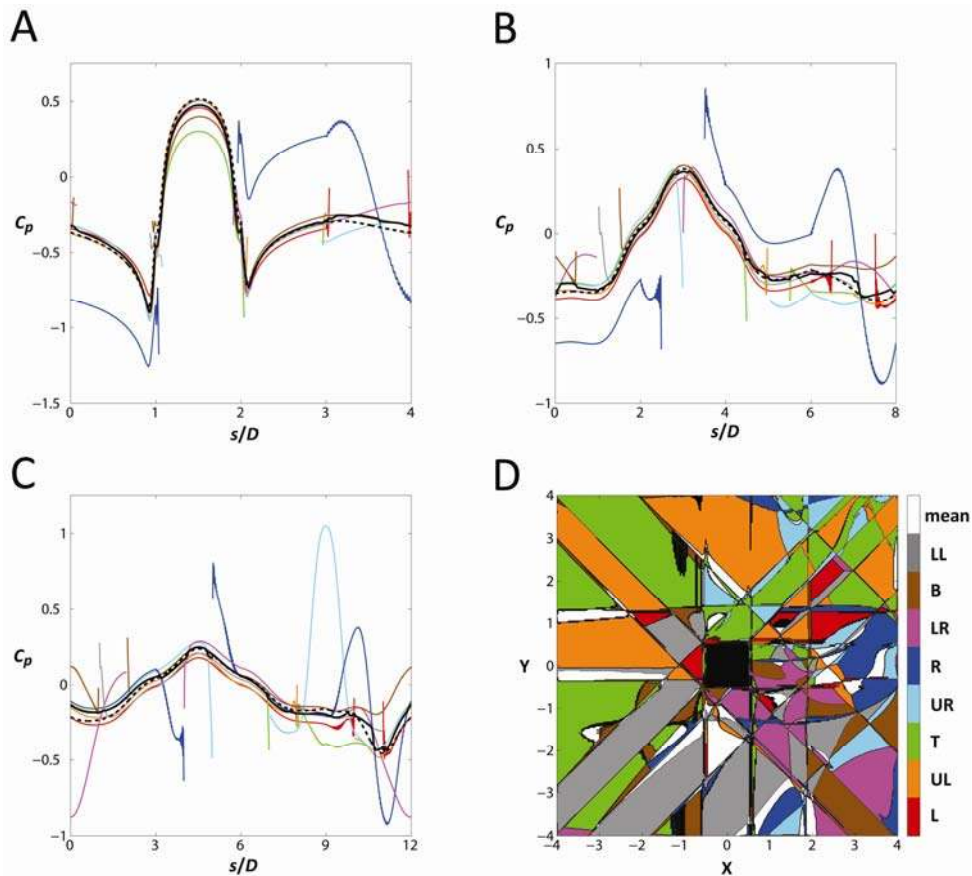
617
 618 **Figure 3.** (A) Pressure field computed from numerical simulation of three-dimensional self-
 619 propelled swimmer. The pressure coefficient $c_p = p/(\rho U^2)$. Velocity nodes completely inside
 620 swimmer body are indicated in black (body surface is smooth in numerical simulation). Spatial
 621 coordinates are normalized by swimmer length. (B) Pressure field estimated using unsteady
 622 algorithm. (C) Pressure field estimated using quasi-steady algorithm.
 623



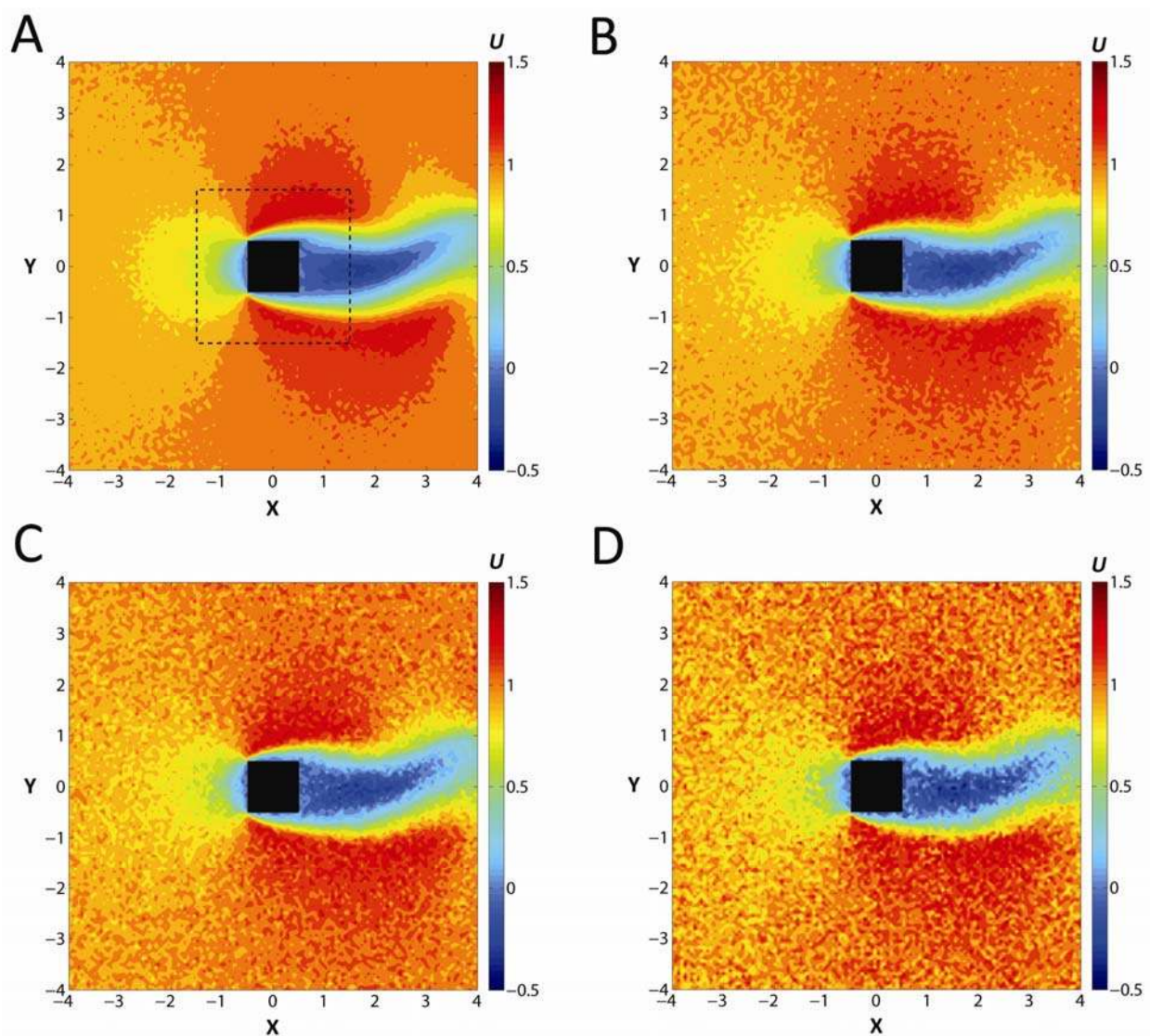
624
 625 **Figure 4.** Pressure on contour surrounding the self-propelled swimmer and immediately adjacent
 626 to the region of undefined pressure, at four instants during the swimming cycle duration T . Head
 627 is at body node number 45; tail is at body nodes 1 and 90. Solid curve, pressure computed from
 628 numerical simulation. Closed circles, pressure estimated from unsteady algorithm. Open circles,
 629 pressure estimated from quasi-steady algorithm. (A) $t/T = 1/4$. (B) $t/T = 1/2$. (C) $t/T = 3/4$. (D)
 630 $t/T = 1$.



631
 632 **Figure 5.** (A) Particle image velocimetry (PIV) measurement of a freely-swimming jellyfish
 633 medusa. Velocity field is plotted on vorticity contours. Maximum velocity vector is
 634 approximately 3 cm s^{-1} . Velocity field is plotted at half of full resolution. Left half of
 635 exumbrellar surface is indicated by black curve. (B) Pressure field estimated using unsteady
 636 algorithm. (C) Particle image velocimetry (PIV) measurement of a freely-swimming lamprey.
 637 Velocity field is plotted on vorticity contours. Maximum velocity vector is approximately 11 cm
 638 s^{-1} . Velocity field is plotted at half of full resolution. Animal body is approximately indicated in
 639 black. (D) Pressure field estimated using unsteady algorithm.
 640



641
 642 **Figure A1.** (A) Pressure on surface of bluff body estimated using quasi-steady algorithm.
 643 Measurement contour is offset by $0.1D$ from the fluid-solid interface. The pressure coefficient c_p
 644 $= p/(\rho U^2)$. s is the local surface coordinate and increases in the counter-clockwise direction from
 645 the upper right corner of the bluff body. Dashed black line, pressure from numerical simulation;
 646 solid black line, pressure estimated using quasi-steady algorithm; solid colored lines, pressure
 647 estimates based on each family of integration paths. Colors correspond to paths in Fig. 1 and to
 648 the legend in panel D. (B, C) Pressure on square contours centered on the bluff body and with
 649 side length $2D$ and $3D$, respectively (e.g. Fig. A2A). s is the local surface coordinate and
 650 increases in the counter-clockwise direction from the upper right corner of each square contour.
 651 The difference in abscissa in panels A-C reflects the different contour perimeters. (D) Contour
 652 plot that colors each point in the domain according to the path family that corresponds to the
 653 median pressure at that point. To reveal the individual integration path family contributions, a
 654 ninth pressure value equal to the mean of the 8 path families is included, so that the median
 655 pressure is from either a single integration path family or from the mean.

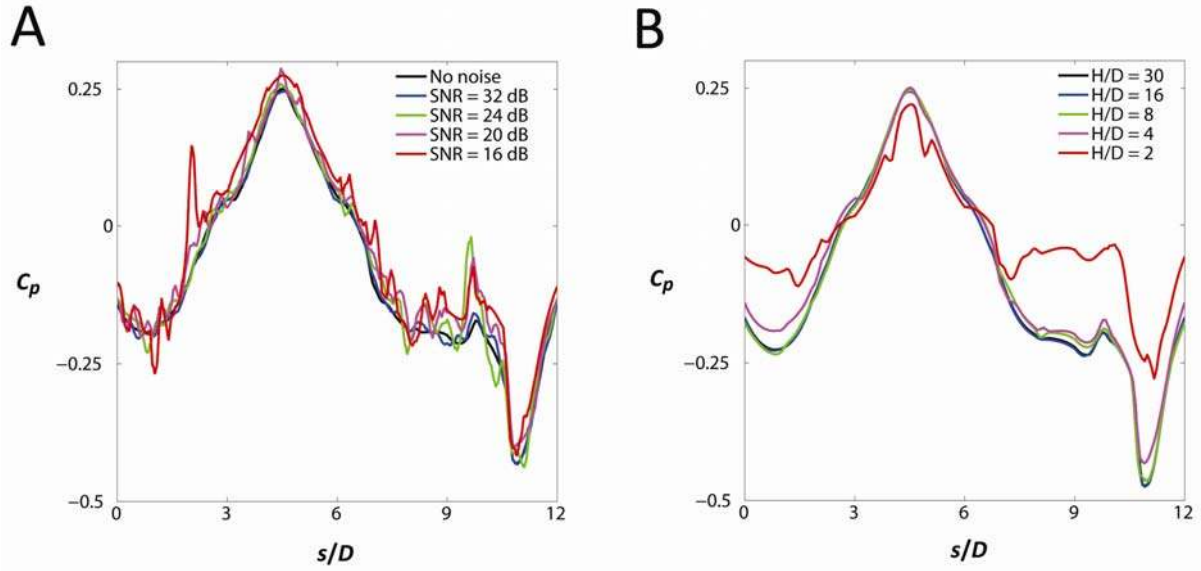


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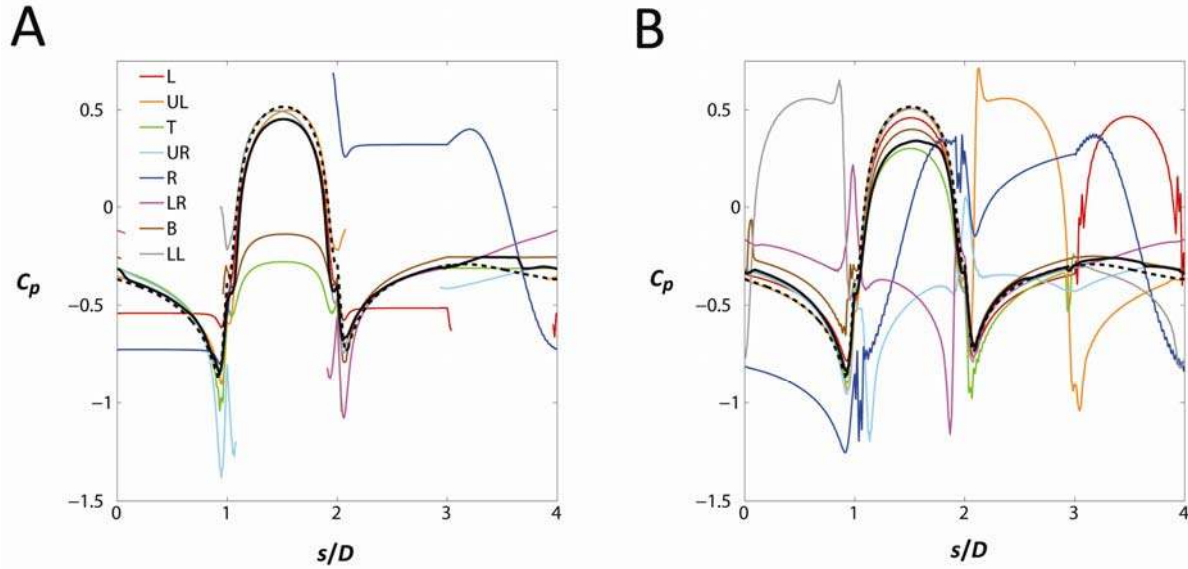
657 **Figure A2.** Streamwise velocity contours for flow field with Gaussian white noise added to
658 reduce the signal-to-noise ratio (SNR). (A) SNR = 32 dB. (B) SNR = 24 dB. (C) SNR = 20 dB.

659 (D) SNR = 16 dB. Dashed square in panel A indicates contour on which quasi-steady pressure
660 estimates are compared in Fig. A3A.

661

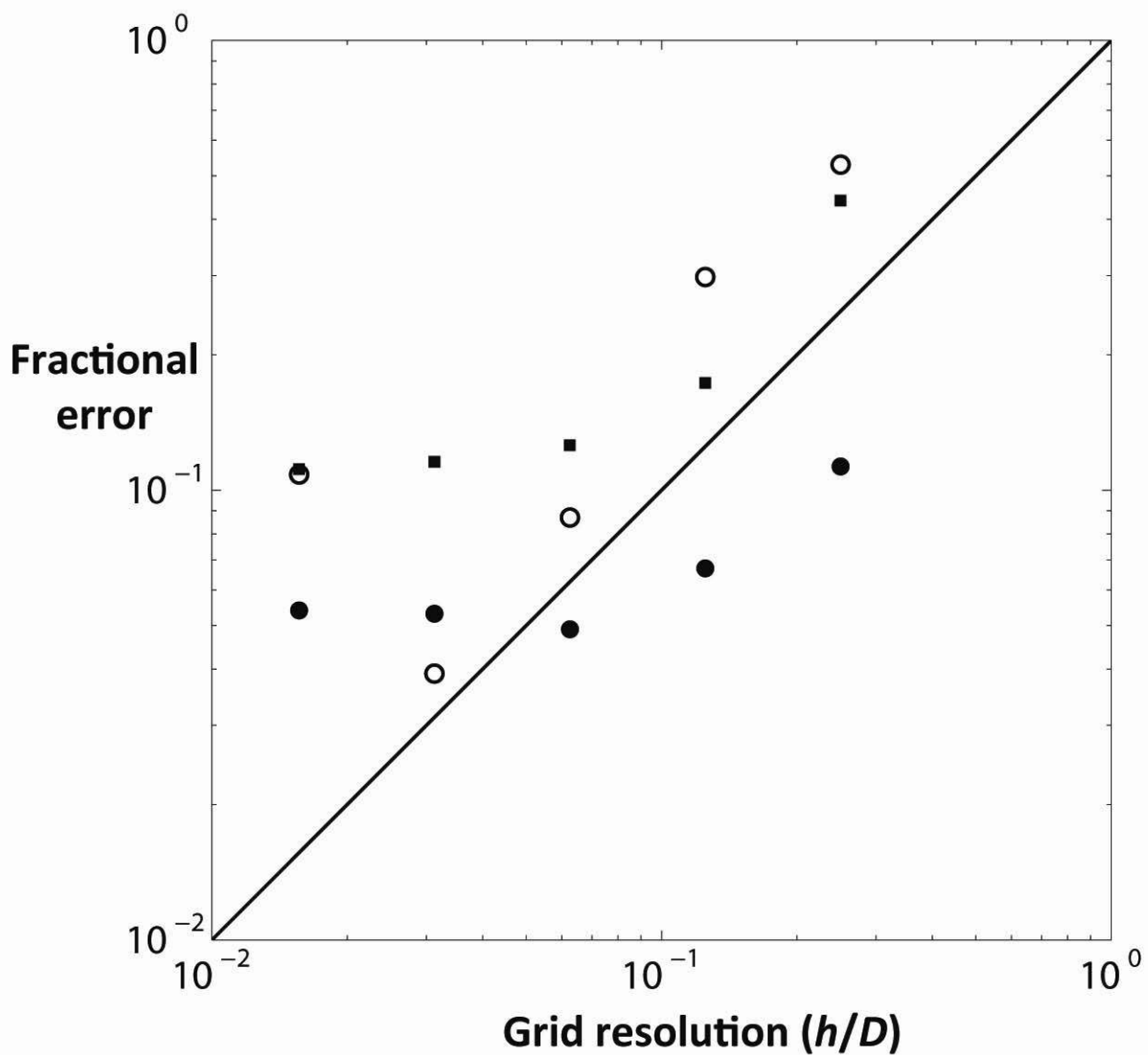


662
 663 **Figure A3.** (A) Quasi-steady pressure estimate on contour shown in Fig. A2A for varying signal-
 664 to-noise ratio. s is the local surface coordinate and increases in the counter-clockwise direction
 665 from the upper right corner of the square contour. (B) Quasi-steady pressure estimate on contour
 666 shown in Fig. A2A for varying measurement domain size. H is the half-width of the
 667 measurement domain.
 668



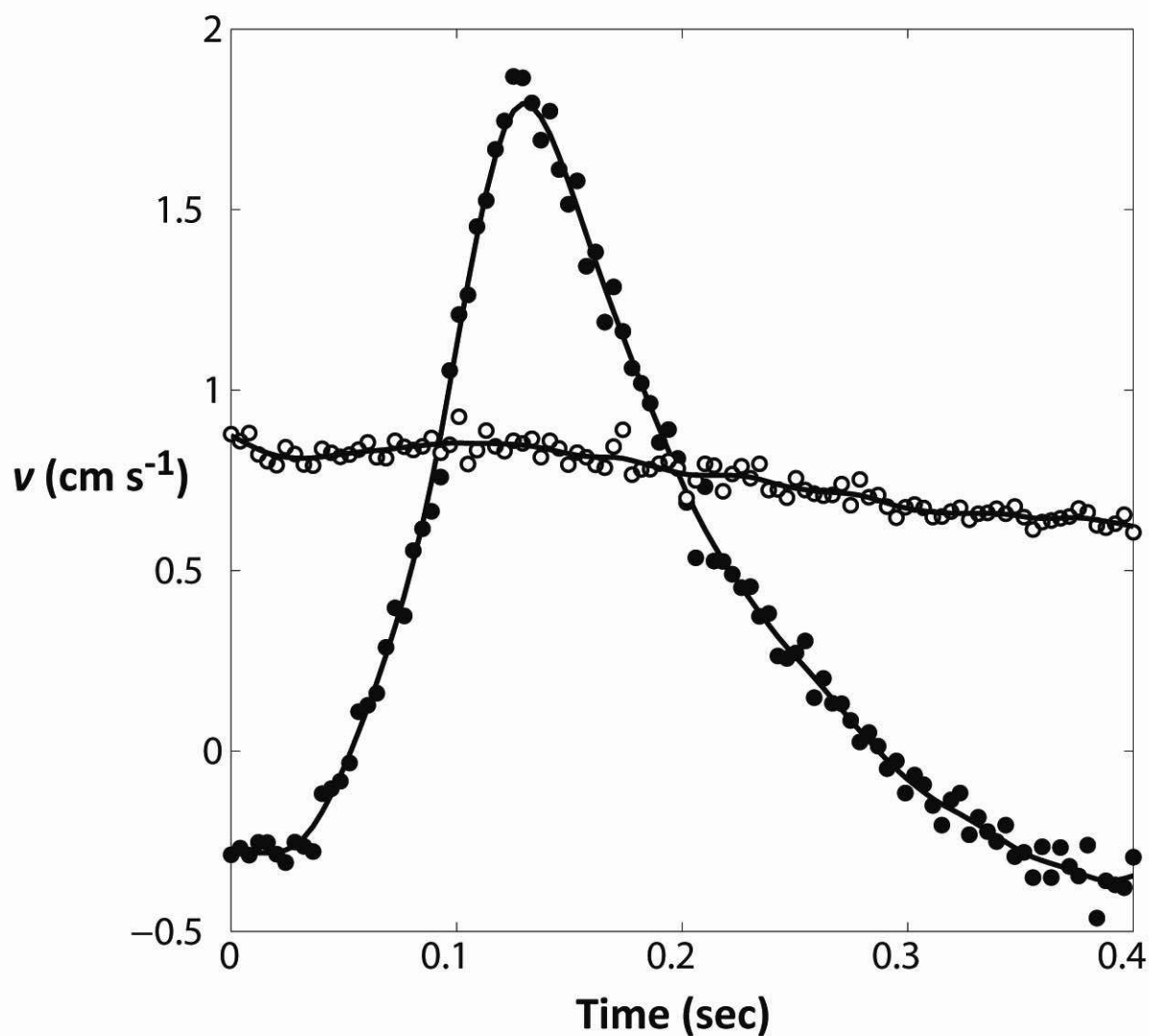
669
 670 **Figure A4.** (A) Pressure on surface of bluff body estimated using quasi-steady algorithm without
 671 viscous term. Measurement contour is offset by $0.1D$ from the fluid-solid interface. s is the local
 672 surface coordinate and increases in the counter-clockwise direction from the upper right corner
 673 of the bluff body. Dashed black line, pressure from numerical simulation; solid black line,
 674 pressure estimated using quasi-steady algorithm; solid colored lines, pressure estimates based on
 675 each family of integration paths. (B) Pressure on surface of bluff body estimated using quasi-
 676 steady algorithm without treatment of fluid-solid interfaces to remove integration paths that pass
 677 through the solid body.

678



679

680 **Figure A5.** Spatial convergence of the algorithm. Log-log plot of the fractional error in
 681 instantaneous streamwise (closed circles), instantaneous lateral (open circles), and time-averaged
 682 streamwise (closed squares) force coefficients versus grid resolution for numerical simulation of
 683 two-dimensional flow past the bluff body. Solid line indicates a slope of 1 corresponding to first-
 684 order convergence. Deviation from first-order convergence at small grid resolution is due to a
 685 combination of model error and numerical round-off error.



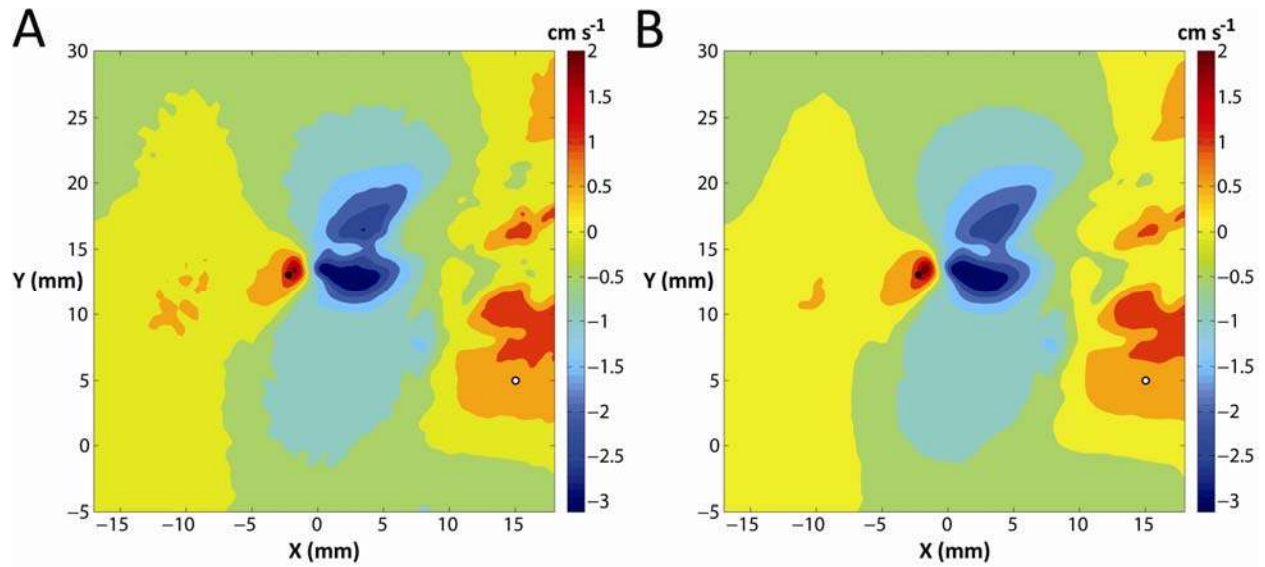
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687 **Figure A6.** Time series of v component data at two selected points in the jellyfish PIV data set.

688 Symbols indicate original PIV data at corresponding locations identified in Fig. A7. Solid curves

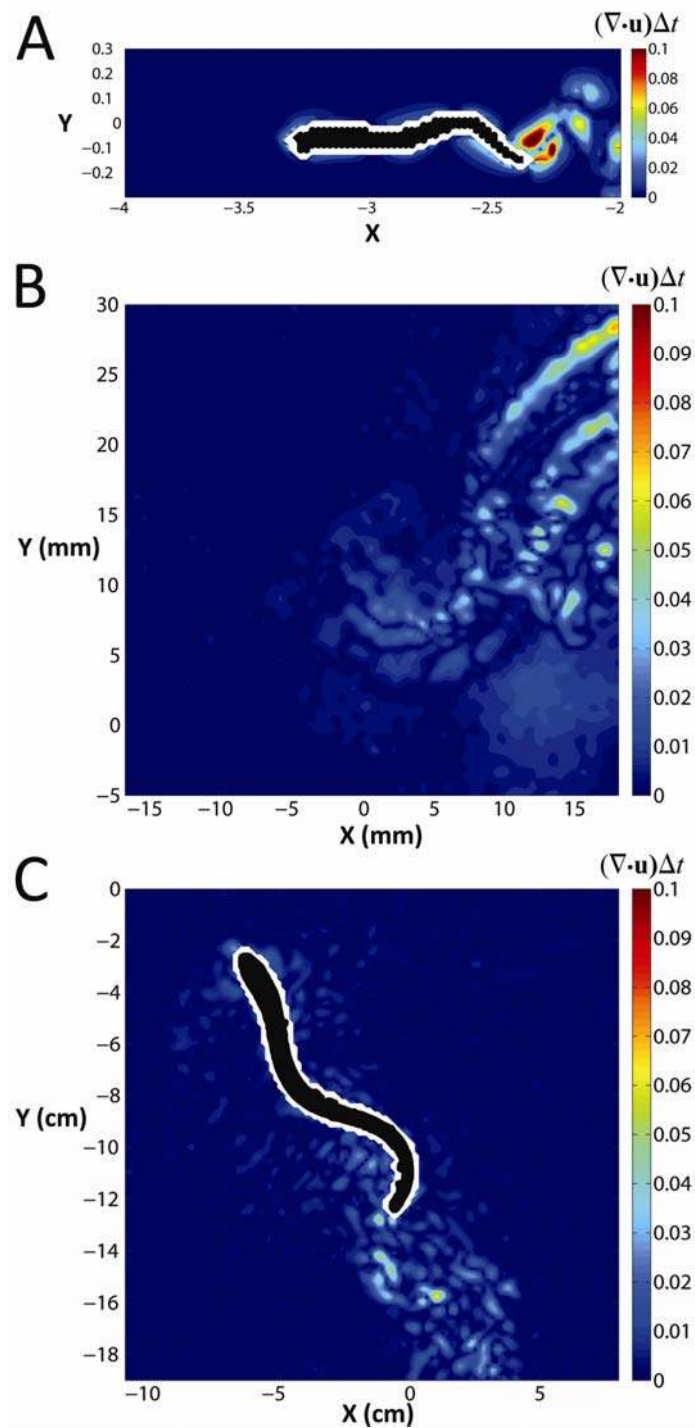
689 indicate respective smoothing splines.

690

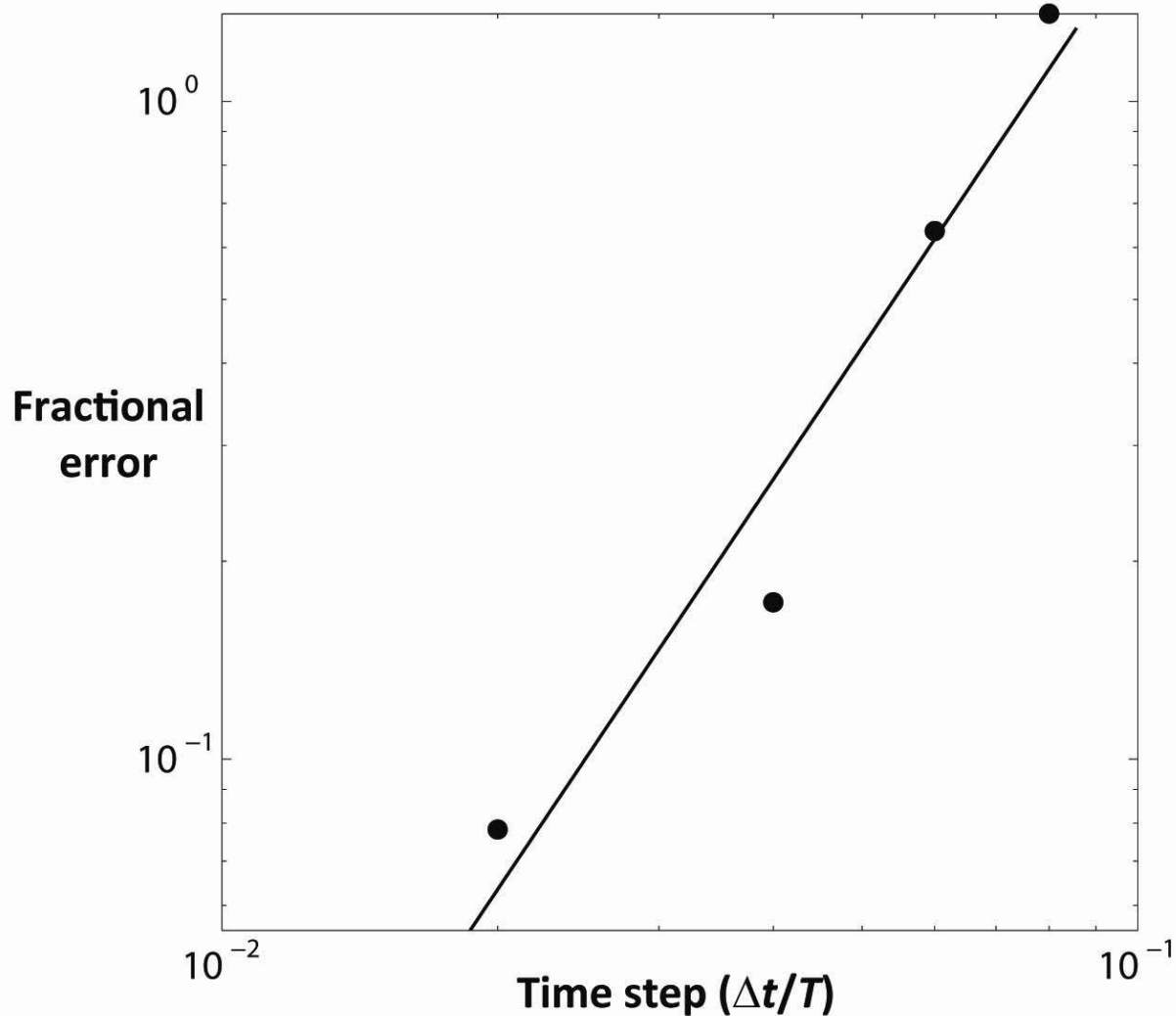


691
 692 **Figure A7.** (A) Contour plot of v component of original velocity measurement during middle of
 693 jellyfish bell contraction. (B) Contour plot of v component temporal spline-filtered velocity
 694 measurement during middle of jellyfish bell contraction. Location of animal is similar to that
 695 indicated in Fig. 5A, although earlier in the bell contraction phase. Closed circle near bell margin
 696 and open circle in wake indicate locations of temporal profiles in Fig. A6.

697



698
 699 **Figure A8.** Contour plots of normalized two-dimensional divergence for (A) three-dimensional
 700 numerical simulation of self-propelled swimmer, (B) PIV measurement of freely-swimming
 701 jellyfish (cf. Fig. 5A), (C) PIV measurement of freely-swimming lamprey (cf. Fig. 5C).
 702 Dimensional divergence is normalized by multiplying by the time step between sequential
 703 velocity fields in each case.



704

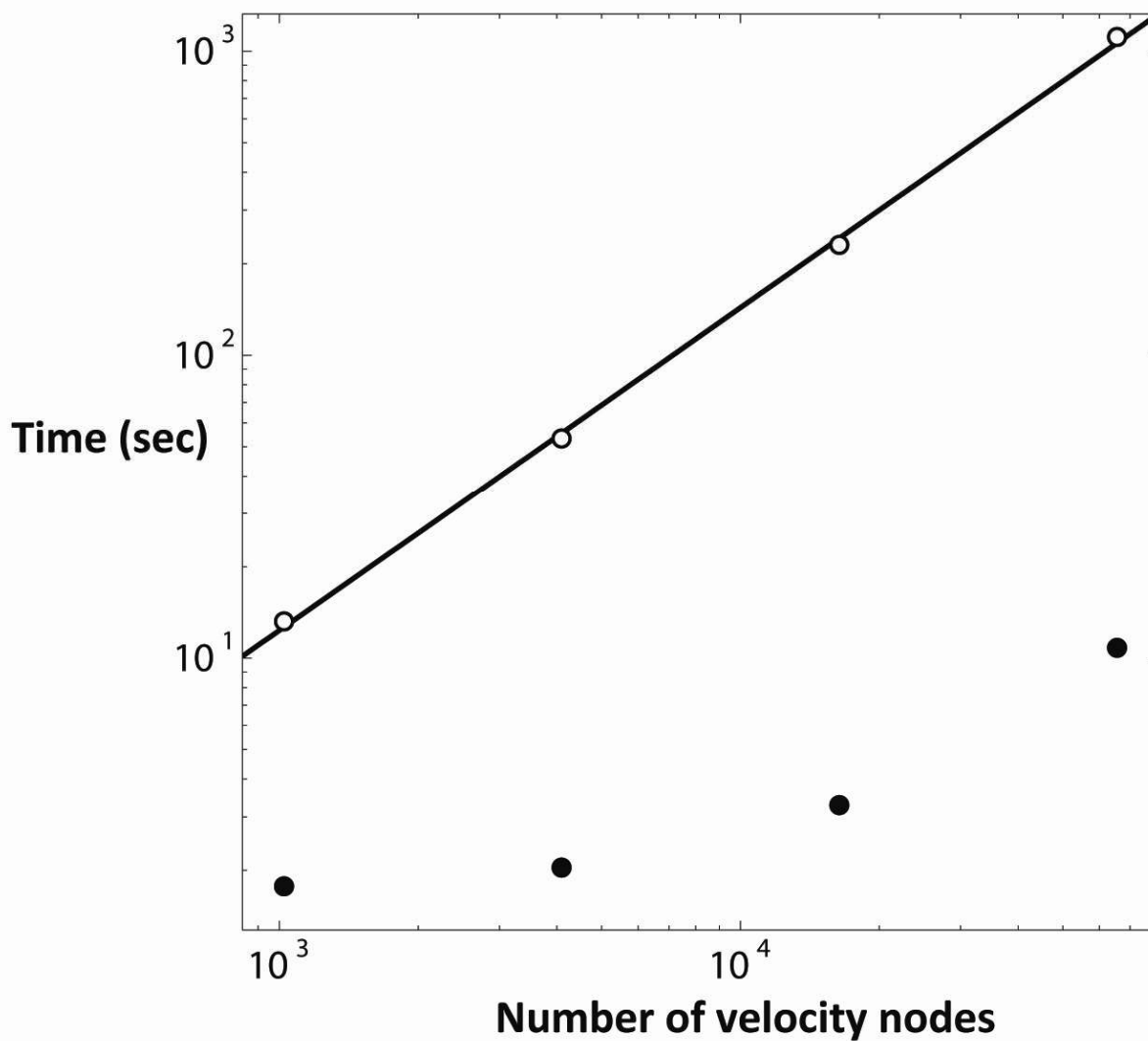
705 **Figure A9.** Temporal convergence of the algorithm. Log-log plot of the fractional error in

706 pressure at the head of the simulated three-dimensional self-propelled swimmer versus time step

707 between velocity fields (closed circles). Solid line indicates a slope of 2 corresponding to

708 second-order convergence.

709



710

711 **Figure A10.** Computational cost of the algorithm, as quantified by the time required for a single

712 3-GHz processor to apply the temporal filter (open circles) and to compute the pressure field

713 (closed circles) for velocity fields from 32x32 to 256x256 nodes. Solid line indicates slope of 1.