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Analysis of Vortex-Induced Counter Torque and Fin Pressure on a Finned Body of Revolution

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Analysis of Vortex-Induced Counter Torque and Fin Pressure on a Finned Body of Revolution

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Abstract

Finned bodies of revolution firing lateral jets in flight may experience lower spin rates than predicted. This reduction in spin rate is a result of vortices generated by the interaction between the lateral jets and freestream air flowing past the body. The vortices change the pressure distribution on the fins, inducing a counter torque that opposes the desired spin. Wind tunnel data measuring roll torque and fin pressures were collected for a full-scale model at varying angle of attack, roll angle, airspeed, and jet strength. The current analysis builds upon previously written code that computes torque by integrating pressure over the fin surfaces at 0° angle of attack. The code was modified to investigate the behavior of counter torque at different angles of attack and roll angles as a function of J , the ratio of jet dynamic pressure to freestream dynamic pressure. Numerical error analysis was applied to all data to assist with interpretation of results. Results show that agreement between balance and fin pressure counter torque at 0° angle of attack was not as close as previously believed. Counter torque at 4° angle of attack was higher than at 0° , and agreement between balance and fin pressure counter torque was closer. Plots of differential fin pressure coefficient revealed a region of high pressure at the leading edge and an area of low pressure over the center and aft regions of the tapped surface. Large differences in the counter-torque coefficient were found between various freestream dynamic pressures, especially at Mach 0.95 and 1.1. Roll angle had significant effect only for cases at angle of attack, where it caused counter torque to change unpredictably.

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Nomenclature

A_{fin}	fin area
A_{tr}	tapped region area
J	nozzle dynamic pressure ratio
C_{CT}	counter torque coefficient
C_P	coefficient of pressure
CT	counter torque due to vortex-fin interaction
\mathbf{F}	force vector
FBR	finned body of revolution
P	pressure measured by transducer
P_∞	freestream pressure
Q_{nozzle}	dynamic pressure on nozzle centerline at nozzle exit
Q_∞	freestream dynamic pressure
\mathbf{r}	radius vector
T	total torque on finned body of revolution
T_F	torque from fins with jet off
T_J	torque from jets with tunnel off
\mathbf{T}_{fin}	torque on fin
\mathbf{T}_{tr}	torque on tapped region
U_{CT}	uncertainty in counter torque
U_{CP}	uncertainty in coefficient of pressure
U_P	uncertainty in pressure transducer
U_{Q_∞}	uncertainty in freestream dynamic pressure
U_r	uncertainty in parameter r
VFI	vortex-fin interaction
α	angle of attack
θ_i	sensitivity coefficient
ΔP	difference between odd and even side pressure

Analysis of Vortex-Induced Counter Torque and Fin Pressure on a Finned Body of Revolution

1 Introduction

Finned bodies of revolution (FBR) firing lateral jets in flight may experience lower spin rates than expected due to an interaction of the jets with freestream air. This interaction creates vortices at the nozzle exits that are swept downstream and affect the pressure distribution over the fin surfaces. Known as Vortex-Fin Interaction (VFI), this phenomenon creates a counter torque opposing the spin of the body. The mechanism of VFI is illustrated in Figure 1, reproduced from Reference 1. A counter-rotating vortex pair is formed at each nozzle and is believed to be the primary cause of interaction with the downstream fins.² The objective of the current work is to characterize the behavior of counter torque at various flight conditions by deriving pressure-induced counter torque and comparing these results to balance torque. A thorough investigation of pressure-induced torque may provide a better understanding of the physics of VFI that cannot be obtained from the balance data. Previous MATLAB³ code applied integration to pressure data for a 0° angle of attack trial to obtain counter torque as a function of J , the ratio of jet-to-freestream dynamic pressure.⁴ The parameter J is the dominant variable for correlating counter torque with various run conditions.¹ It is defined as:

$$J \equiv Q_{nozzle} / Q_{\infty} \quad (1)$$

The variables Q_{nozzle} and Q_{∞} denote dynamic pressures on the nozzle centerline intersecting the vehicle surface and the freestream air, respectively.

The original code is expanded to accommodate various angles of attack, roll angles, fin positions, and jet nozzle configurations. Behavior of counter-torque coefficient is also investigated in order to identify trends associated with test conditions. In all cases, comparisons are made between results obtained from fin pressure and from the aerodynamic balance. An error analysis is conducted to complement this comparison and assist in validation of results. In addition, visualization of the pressure distribution across the fin is

accomplished with contour plots of pressure coefficient variations. This approach facilitates a better understanding of the physical effects of VFI on fin pressure distribution, and may assist in the interpretation of the findings of the counter-torque plots. These tasks are implemented using numerical integration and interpolation techniques in MATLAB.

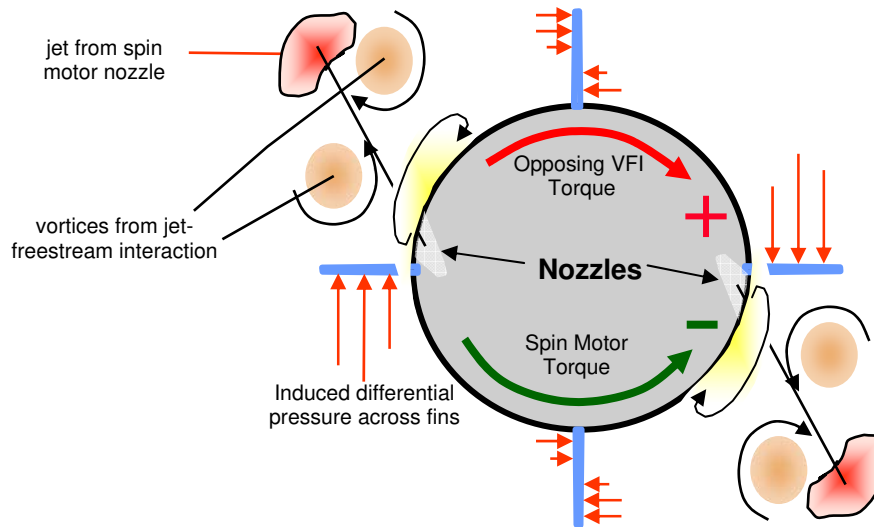


Figure 1. Vortex-Fin Interaction on FBR (View from Fore to Aft)

2 Experimentation

Wind tunnel testing of a full-scale FBR model was conducted in the Arnold Engineering Development Center (AEDC) Tunnel 16-T and the NASA Ames 11-ft Unitary Tunnel. Tests were conducted at Mach 0.8, 0.95, and 1.1 and freestream dynamic pressures of 220, 440, and 880 psf. Other variables include fin cant angle, angle of attack, roll angle, carriage lugs, varying nozzle size, repositioned fins, and jet strength.¹ Schematics of the FBR are shown in Figures 2-4, reproduced from Reference 2. An aerodynamic balance is placed within the model in order to measure forces and moments upon the body.

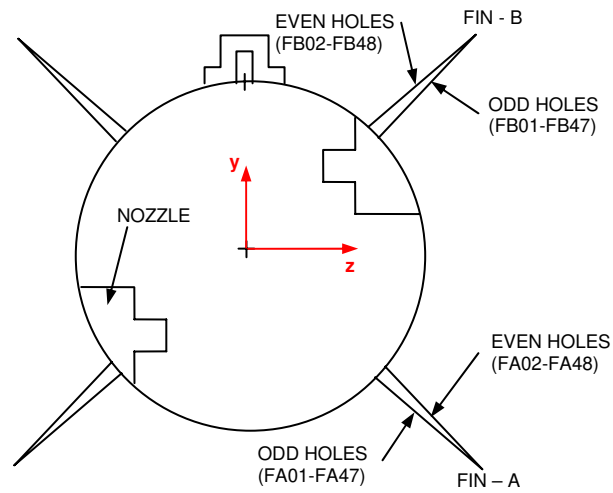


Figure 2. Fined Body of Revolution (View from Fore to Aft)

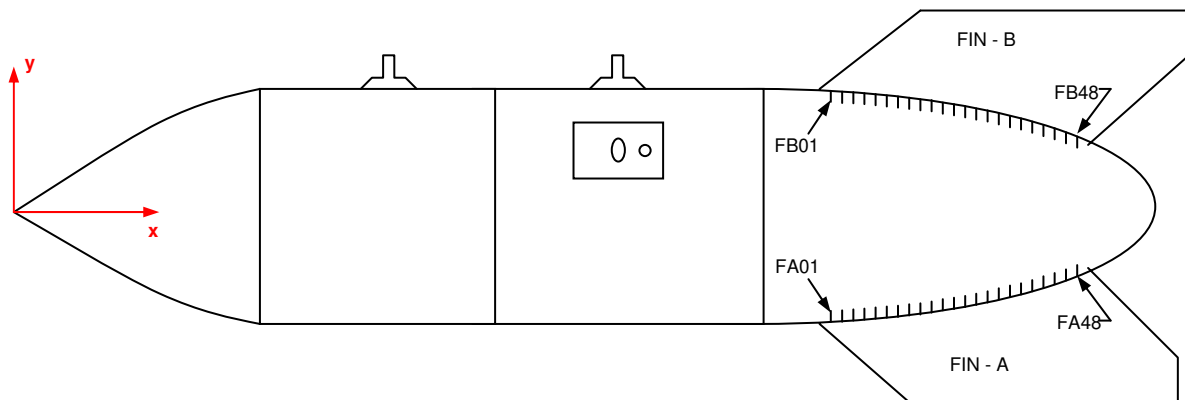


Figure 3. Side View of FBR

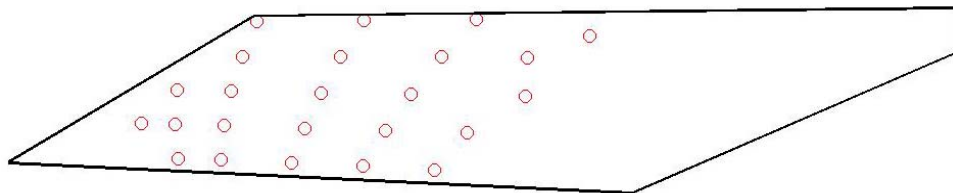


Figure 4. Fin Shape and Pressure Taps (red)

The FBR has four fins spaced 90° apart, and fins A and B were instrumented with pressure taps. The taps were placed in the pattern seen in Figure 1, but slightly staggered on

opposing sides due to the fin's small thickness. Taps were numbered 1 to 48 with 24 on each side; odd numbers on one side, even numbers on the other. During the run, J was gradually increased while continuously recording pressure and balance data. These results are directly input into the MATLAB code for analysis after headers and titles are stripped. Tare runs were also conducted with tunnel on, jets off to determine torque resulting solely from vehicle geometry and jets on, tunnel off to determine torque due to jet thrust.

3 Counter Torque Computation

The FBR experiences torque from the jets, the fins, and the VFI counter torque. The counter torque (CT) is defined as follows:¹

$$CT = -(T - T_J - T_F) \quad (2)$$

T is the total torque acting on the body, T_J is the torque from the jets with no freestream flow, and T_F is the torque from the fins in the absence of the jets. The torque from the jets without a freestream flow does not affect the fin pressures and the torque from the fins without the jets can be calculated using the tare runs. In order to determine counter torque acting upon the body, the 24 pressure readings on each of the 4 instrumented surfaces must be integrated and the torque from the tare runs subtracted. Using Delaunay Triangulation³, a mesh is formed between the 24 taps, generating roughly 35 triangles as seen in Figure 5. A mean pressure is then determined for each triangle using the average of the pressures at its 3 vertices.

By computing triangle areas and taking the cross products of 2 legs of each triangle, an array of area vectors is generated. An array of force vectors (\mathbf{F}_i) is created by multiplying the area vectors by the corresponding mean pressures. Since the forces are assumed to act at the centroid of each triangle, the moment arm is defined as the distance from the coordinate system origin to the centroid. By computing the moment arm corresponding to each triangle (\mathbf{r}_i), the total torque on a single surface (\mathbf{T}_{surf}) is given by:⁴

$$\mathbf{T}_{surf} = \sum_{i=1}^N \mathbf{r}_i \times \mathbf{F}_i \quad (3)$$

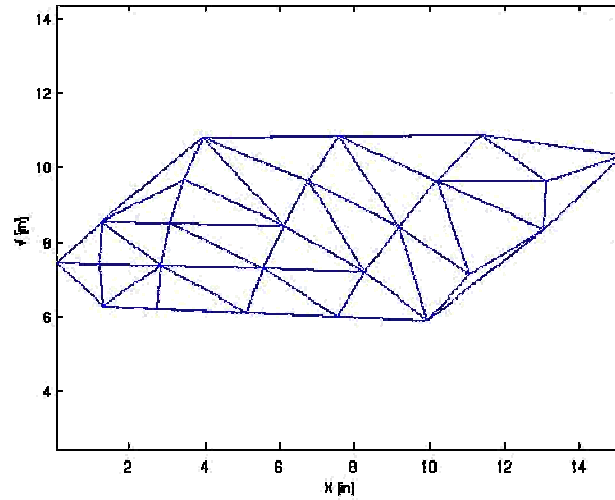


Figure 5. Meshed Pressure Tap Region Generated with Delaunay Triangulation

The ratio of total torque acting upon the fin surface to torque acting upon the tapped region was then assumed to be equal to the ratio of the respective areas, *i.e.*,

$$\mathbf{T}_{\text{fin}} = \mathbf{T}_{\text{tr}} \frac{A_{\text{fin}}}{A_{\text{tr}}} \quad (4)$$

\mathbf{T}_{fin} , \mathbf{T}_{tr} , A_{fin} , and A_{tr} denote the torques and areas of the fin and tapped region respectively. Implicit in the use of Eq. (4) is the assumption that the tapped region adequately captures the nature of the pressure in the untapped region. The validity of the assumption is discussed in more detail in the following sections.

When the torque upon a single fin surface is obtained, the torque from the tare run can be subtracted out to obtain the counter torque. Repeating the process for all 4 instrumented surfaces and summing the torques produces the total torque acting upon two of the FBR's four fins. Assuming symmetry in the model at 0° angle of attack, the total torque upon the FBR is computed by multiplying the torque on the two instrumented fins by 2. This counter torque can be plotted and compared with the torque measured by the aerodynamic balance. The same basic procedure can be applied to tests under various conditions with some modifications to the MATLAB code.

4 MATLAB Implementation

The original MATLAB code, *VFI_torque.m*,⁴ read data for a run at 0° angle of attack and plotted the resulting counter torque together with torque from the aerodynamic balance as a function of J . Plots of torque from the individual tapped surfaces and fins were also

generated. The code was modified to include the effect of torque from the fins with jets off, which had previously been neglected. The modified code was then used to produce plots for all trials at 0° angle of attack, including those at a 45° roll angle. Major modifications were made to the code to accept angles of attack other than 0° , though the basic integration scheme remained the same. Error analysis was conducted numerically using the original code (see below), while new code was written to create pressure contour and counter-torque coefficient plots. A brief description of each code is given, and the complete codes are included in Appendix A.

4.1 VFI_torque.m

As it existed at the start of the current effort, the code initially asks for user input of the directory and filename of the trial to be plotted. The complete data set is loaded, and all fin pressures and nozzle dynamic pressure ratios (J) extracted. Subroutines were then called for all 4 tapped surfaces: *FAodd.m*, *FAeven.m*, *FBodd.m*, *FBeven.m*. *FAodd.m* corresponds to “Fin A, odd side,” and the appropriate data are fed into the subroutine. The subroutine *vert_and_area.m* is then called to generate a triangular mesh from the x , y , and z coordinates using the MATLAB Delaunay function. The vertex coordinates of each triangle are read into an array of x , y , and z vertex coordinates and sent to the subroutine *tri3Darea.m*, which returns the area magnitude. *vert_and_area.m* then computes the tapped region area, an array of triangle centroid coordinates, and an array of area vectors (calculated from the cross product of two legs of each triangle). The subfunction *triangle_pressures.m* computes the mean pressure for each triangle from the pressures at its 3 vertices and outputs the results in an array. The subfunction *moment.m* then accepts the triangle mean pressures, area vectors, and centroid coordinates and computes torque over the tapped region. Using the known areas of the tapped region and fin, this torque is scaled to the entire fin surface as shown in Eq. (4). The process is repeated for the entire range of J , resulting in a vector of torques. Each of the 4 main subroutines follows this process, resulting in torques for all 4 fin surfaces. After plotting these torques, they are summed and multiplied by 2, resulting in the total torque on the FBR. This result is then plotted against the balance measured torque as a function of J for comparison. A more detailed explanation of the code can be found in Reference 4.

4.2 Fin Torque Removal

The fin pressures resulting from the tare runs are in the same format as the data files, and are read into *VFI_torques_tare.m* (for 4° angle of attack cases) and *VFI_torques_0tare.m* (for 0° angle of attack cases). The output from the code is a set of 4 tare torques (one for each tapped surface) for each combination of conditions: Mach 0.8, 0.95, and 1.1; Q 220, 440, and 880 psf; 0° and 4° angle of attack; and 0° and 45° roll angle. The sets of torques at all Mach numbers and freestream dynamic pressures for a given roll angle and angle of attack were compiled together in a text file. These text files are used in *VFI_torques_0_AOA.m* and *VFI_torques_AOA.m* to remove the tare torque from the torque on each tapped surface. The subroutines *FAodd.m*, *FAeven.m*, *FBodd.m*, *FBeven.m* were modified to identify the tare torque corresponding to the current set of conditions, subtract the tare torque from the fin surface torque, and output the results as before to the main

program. This procedure eliminates the torque from the fins that was neglected in the original code. The results from the modified program show significant differences from those generated with the original code.

4.3 Angle of Attack

The program *VFI_torques_AOA.m* is a modification of *VFI_torque.m* that facilitates an investigation of counter torque behavior at an angle of attack of $\pm 4^\circ$. Because of the symmetric vehicle geometry and the use of only two instrumented fins, a data set at both $+4^\circ$ and -4° angle of attack must be used to completely describe the torque upon the FBR. *VFI_torques_AOA* accepts two data files, one at positive and one at negative angles of attack. The program now works with 8 distinct torque surfaces (2 for each of the 4 fins), calling subroutines *FA_odd_pos.m* and *FA_odd_neg.m* instead of *FA_odd.m* and so forth for the other 7 surfaces. These subfunctions return 8 torque vectors to the main program, where the torques resulting from positive and negative angles of attack are summed separately. This gives a torque vector as a function of J for each half of the FBR at angle of attack. Since the two runs correspond to slightly different J values, their torques cannot be summed directly. A uniform array of J values is generated based upon the range of the two J sets, and both torque vectors are interpolated onto this array. Summing the interpolated torques produces the total counter torque upon the FBR at angle of attack as a function of J . The two balance-measured torques are also extracted from the dataset and interpolated onto the uniform J vector.

4.4 Error Analysis

Error analysis of the experimental data and MATLAB results was conducted to aid in the comparison of pressure-derived torques to experimentally measured torques. The program *VFI_torquesuncertainty.m* was used to solve for errors numerically. The uncertainty equation that was used is given in its basic form as follows:⁵

$$U_r^2 = \sum_{i=1}^J \theta_i^2 U_{X_i}^2 \quad (5)$$

U_r is the uncertainty in the relevant parameter (in this case the counter torque CT). U_{X_i} denotes the uncertainty in each independently measured variable.⁵ θ_i is the sensitivity coefficient, defined as

$$\theta_i = \frac{\partial r}{\partial X_i} \quad (6)$$

The only source of uncertainty is the fin pressure — thus θ_i is given by:⁵

$$\theta_i = \frac{\partial CT}{\partial P_i} \quad (7)$$

U_{x_i} represents the uncertainty within each pressure transducer. This is assumed to be constant at 0.25% of full scale, or 0.0375 psi. The only other variable is the sensitivity coefficient — a measure of the sensitivity of counter torque with respect to pressure change. A coefficient must be computed for each of the 48 taps on a fin. Symmetry can be used to obtain the uncertainty for the other fins. While these sensitivity coefficients can be determined analytically, the algebra is quite complex and requires keeping track of each triangle's area, unit normal, and moment arm vector. This procedure is thus very error-prone. To reduce the possibility of introducing errors in the process, it was decided to compute the coefficients numerically using existing MATLAB scripts.

The MATLAB code directly computes the sensitivity coefficients for fin A and uses symmetry to compute the total uncertainty. Initially, the value of tap 1 was set to one and the rest to zero before calling the functions *FA_odd.m* and *FA_even.m*. The torque output from the function is the sensitivity coefficient of the tap. The value of tap 2 was then set to one and all others to zero. This process was repeated for all 48 taps and the resulting torques were recorded in a vector. With the sensitivity coefficients on one fin, the uncertainty of the counter torque for all fins, U_{CT} , can be calculated from the following:

$$U_{CT} = \sqrt{a \sum_{i=1}^{48} (\theta * U_p)_i^2} \quad (8)$$

The uncertainty in the pressure transducers, U_p , is constant at 0.0375 psi as stated previously. The coefficient a is a correction factor used to scale the uncertainty of a fin to the entire FBR. For tests at angle of attack, data are obtained from 4 instrumented fins and hence a takes the value of 4. At 0° angle of attack, data from only 2 instrumented fins are used. In this case, a equals 8 — a factor of 4 corresponding to the 4 fins, while a factor of 2 appears since the error is doubled by obtaining torque from only 2 instrumented fins. Thus, uncertainty is less for trials at angle of attack by a factor of $\sqrt{2}$.

The error in the torque measured by the aerodynamic balance was computed from the following equation:⁵

$$U_r^2 = B_r^2 + P_r^2 \quad (9)$$

Both the bias, B_r , and precision, P_r , of the aerodynamic balance are known quantities provided by the wind tunnel facility. Using these values, the uncertainty in the balance, U_r , was determined.

4.5 Pressure Coefficient Distribution

Plots of the pressure coefficient variation across the tapped region were created using *finpress_plot.m*. Pressure coefficient C_p , is defined as:

$$C_p = (P - P_\infty) / Q_\infty \quad (10)$$

P is the pressure from the pressure transducer and the freestream pressure P_∞ is determined from the jet-off tare. The program takes a directory and run number as inputs and generates a shaded pressure coefficient distribution across the tapped region. Five slices are also extracted from the plot along the leading edge and chord-wise across the tapped region and the variation of pressure coefficient is plotted against distance along the slice. Plots were only generated for Fin B since no significant variations were seen in Fin A. As seen in Figure 2, Fin B is almost directly downstream of the nozzle and experiences a much larger effect from the induced vortices than Fin A.

A uniform grid was first generated over the tapped region using the MATLAB “meshgrid” command. Because a differential pressure coefficient was desired (i.e. the difference between pressure coefficients on the odd and even sides), the pressures and coordinates of both the odd and even taps were extracted from the data. The odd and even pressures were then interpolated onto the mesh using the “griddata” command. A separate error analysis was also conducted upon the pressure coefficient data using equations 5 and 7. The resulting equation for uncertainty in the coefficient of pressure, U_{CP} , is as follows:

$$U_{CP} = \sqrt{\left(\frac{1}{Q_\infty} U_P\right)^2 + \left(-\Delta P / Q_\infty^2 U_{Q_\infty}\right)^2} \quad (11)$$

The uncertainty in pressure, U_P , again corresponds to uncertainty in the pressure transducers (0.0375 psi). The term U_{Q_∞} refers to uncertainty in the tunnel dynamic pressure and was obtained directly from AEDC’s uncertainty analysis. Since differential pressure coefficient is computed, ΔP corresponds to the difference in pressure coefficient between the even and odd side of the fin. Uncertainty is computed for all points on the grid, although only the maximum is reported since variation between points is minimal.

4.6 Counter Torque Coefficient

The counter torque coefficient is a dimensionless parameter that enables the comparison of the magnitude of counter torque under different conditions. It is defined as:¹

$$C_{CT} = \frac{CT}{Q_\infty AD} \quad (12)$$

The counter torque is represented by CT , while A is the model maximum cross-sectional area ($\pi D^2 / 4$). The reference length D is the maximum body diameter (13.3 inches). Plots with error bars were generated for all run conditions using *VFI_CTcoeff.m* and *VFI_CT_AOA.m*. Again using equations 5 and 7, uncertainty in the counter torque coefficient is given by:

$$U_{CCT} = \sqrt{\left(\frac{1}{Q_\infty AD} U_{CT}\right)^2 + \left(\frac{CT}{Q_\infty^2 AD} U_{Q_\infty}\right)^2} \quad (13)$$

5 Results

Previous results for 0° angle of attack runs were obtained while neglecting the torque from the tare runs. As mentioned previously, the code was corrected and plots were generated for models of configurations 2, 3, and 4 at all run conditions. Error bands were also included on the plots, represented by dashed lines above and below the data. The uncertainties in the counter torques determined from the error analysis are displayed in Table 1.

Table 1. Uncertainty in Pressure-Derived and Balance-Measured Counter Torques

Counter Torque	Uncertainty (\pm in-lbf)
Pressure-derived at 0° angle of attack	39.33
Pressure-derived at 4° angle of attack	27.81
Balance-measured	13.28

Total counter torques for configuration 2, 3, and 4 at 0° angle of attack can be seen in Figures 6-8. Over a similar range of nozzle dynamic pressure ratio, J , the counter torques for the three configurations are essentially identical within the experimental uncertainty. This trend is seen at nearly all run conditions. Configurations 3 and 4 have different lateral jet nozzles from configuration 2 -- the baseline model upon which the rest of this investigation will focus. The "RMS difference" reported on the plots is the root mean square of the difference between the balance and fin pressure counter torques. Before $J = 10$ there is close agreement between the balance and fin pressure counter torques. After this point, the agreement between the two torques is not as close, although they are near or within the error limits.

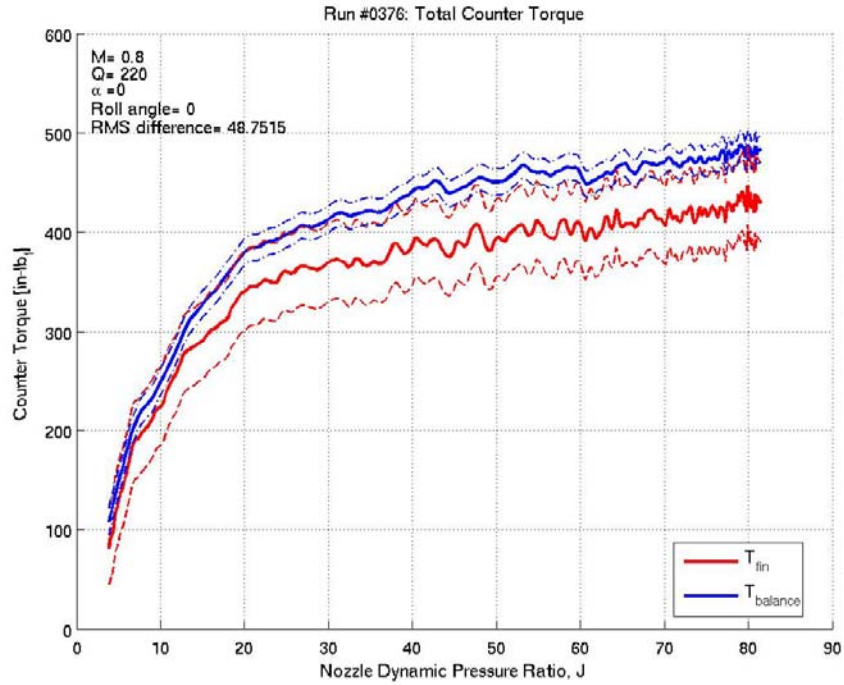


Figure 6. Fin and Balance Counter Torque vs. Nozzle Dynamic Pressure Ratio - Configuration 2

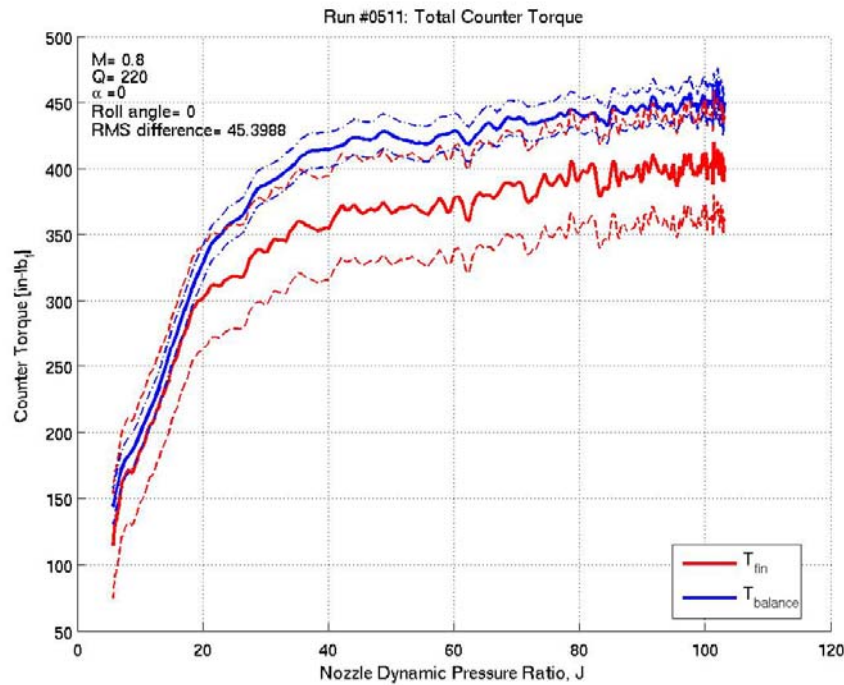


Figure 7. Fin and Balance Counter Torque vs. Nozzle Dynamic Pressure Ratio - Configuration 3

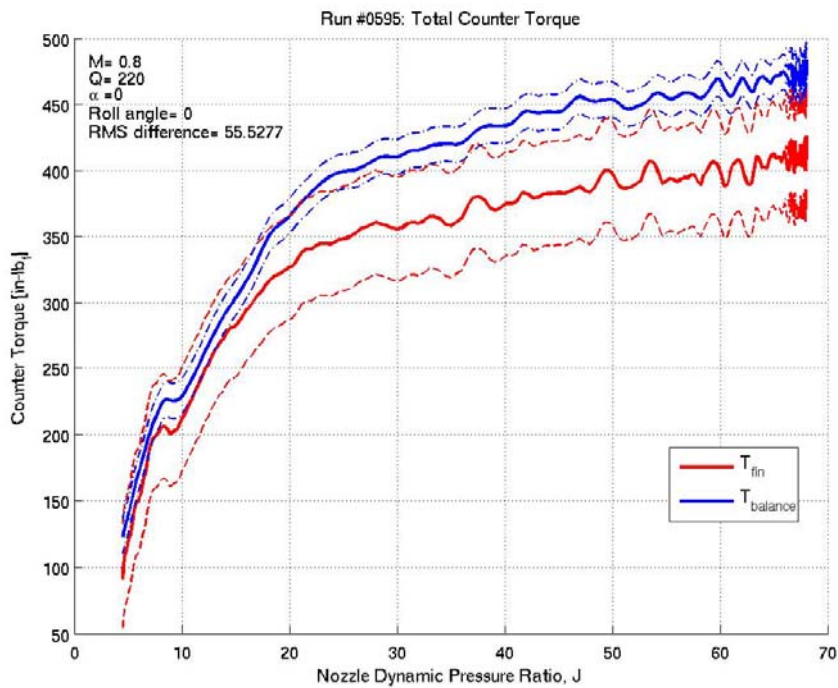


Figure 8. Fin and Balance Counter Torque vs. Nozzle Dynamic Pressure Ratio - Configuration 4

The remaining figures in this section show data for configuration 2. Results were first produced for the 0° angle of attack runs. Figures 9 and 10 show counter torques at $Q = 220$ psf for Mach 0.95 and 1.1.

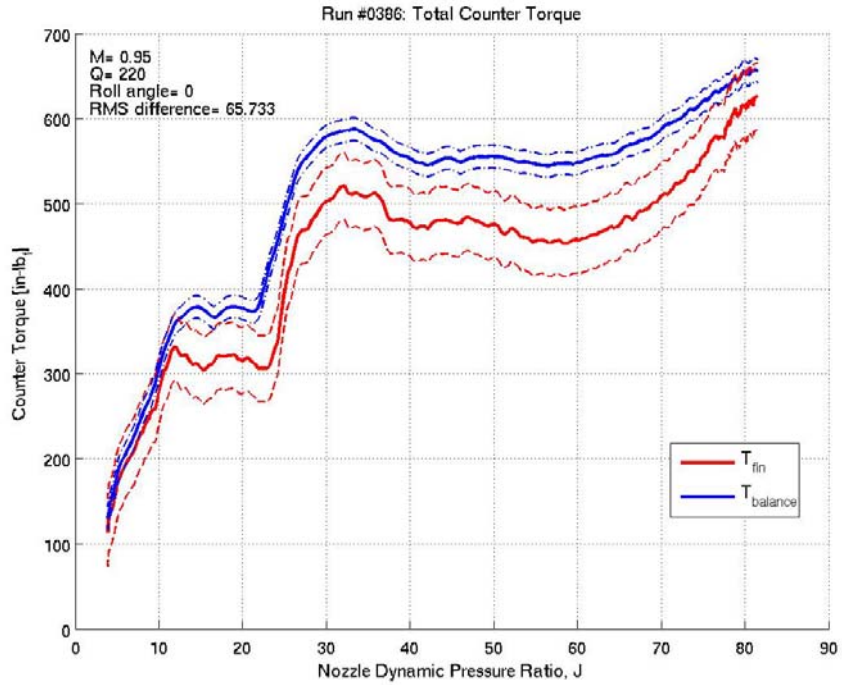


Figure 9. Fin and Balance Counter Torque vs. Nozzle Dynamic Pressure Ratio for Run #0386.

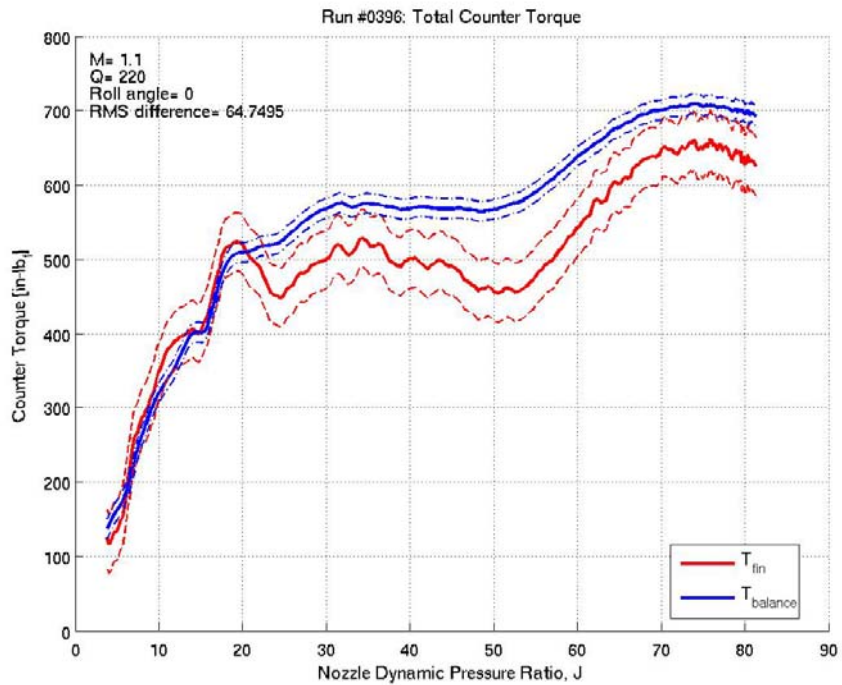


Figure 10. Fin and Balance Counter Torque vs. Nozzle Dynamic Pressure Ratio for Run #0396

The fin pressure counter torque determined by the modified *VFI_torque.m* does not match plots created using the original code.⁴ After subtracting the tare torques, the fin pressure counter torque is consistently less than the balance measured torque. Agreement between the two plotted torques decreases as Mach number increases, which is reflected by an increase in RMS difference. Increase in Mach number also corresponds to an increase in the magnitude of the counter torque.

As the freestream dynamic pressure is increased from 220 to 880 psf, the magnitude of the counter torque increases for all cases. However, the agreement between balance and fin pressure torques remains close at Mach 0.8, but generally slightly in excess of the measurement uncertainty. The agreement between the two torques becomes poor at Mach 0.95 and 1.1 as seen in Figures 11 and 12.

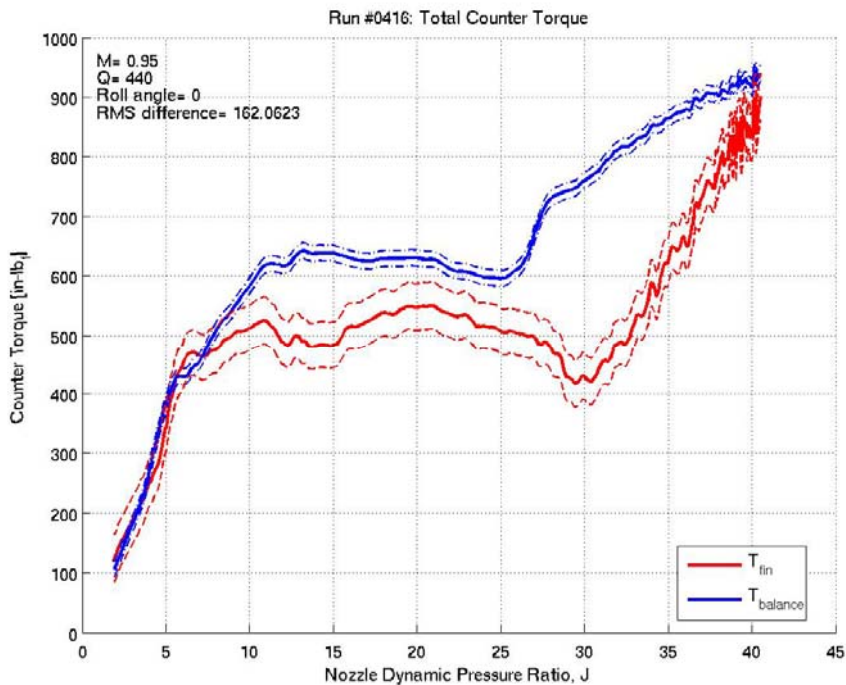


Figure 11. Fin and Balance Counter Torque vs. Nozzle Dynamic Pressure Ratio for Run #0416

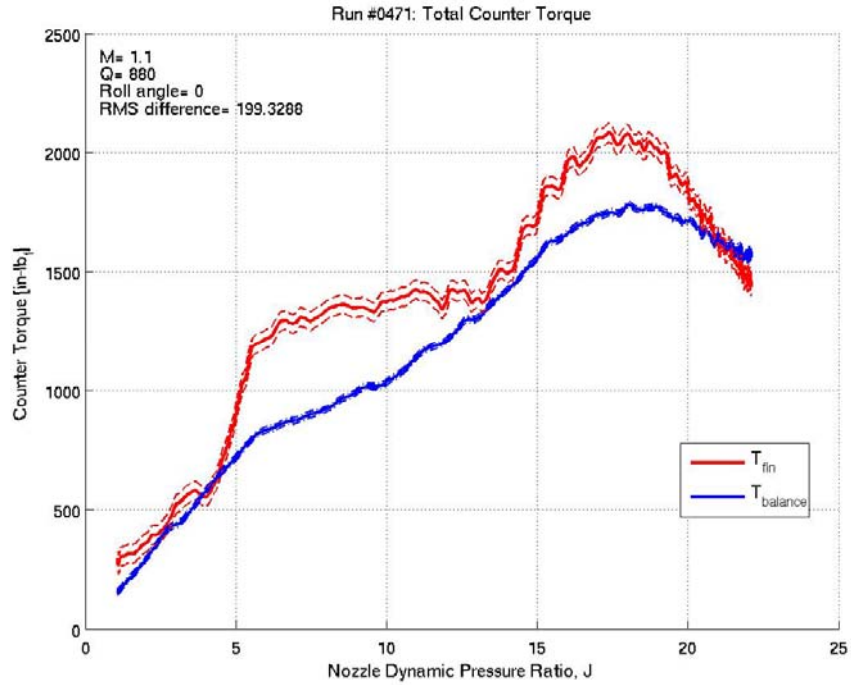


Figure 12. Fin and Balance Counter Torque vs. Nozzle Dynamic Pressure Ratio for Run #0451.

Plots of counter torque on the FBR at a roll angle of 45° were also produced using *VFI_torque.m*. The magnitude and shape of the counter torque plots at a 45° roll angle are similar to those at 0° roll as seen in Figures 13-15. This trend is seen at all Mach numbers and freestream dynamic pressures. It is thus verified that roll angle has little effect upon the behavior of counter torque at 0° angle of attack.

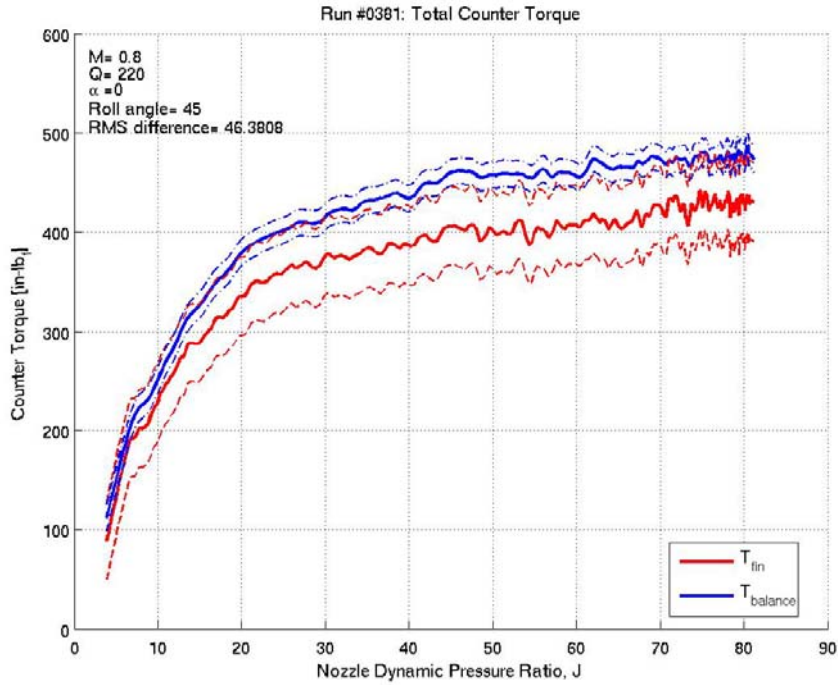


Figure 13. Fin and Balance Counter Torque vs. Nozzle Dynamic Pressure Ratio for Run #0381.

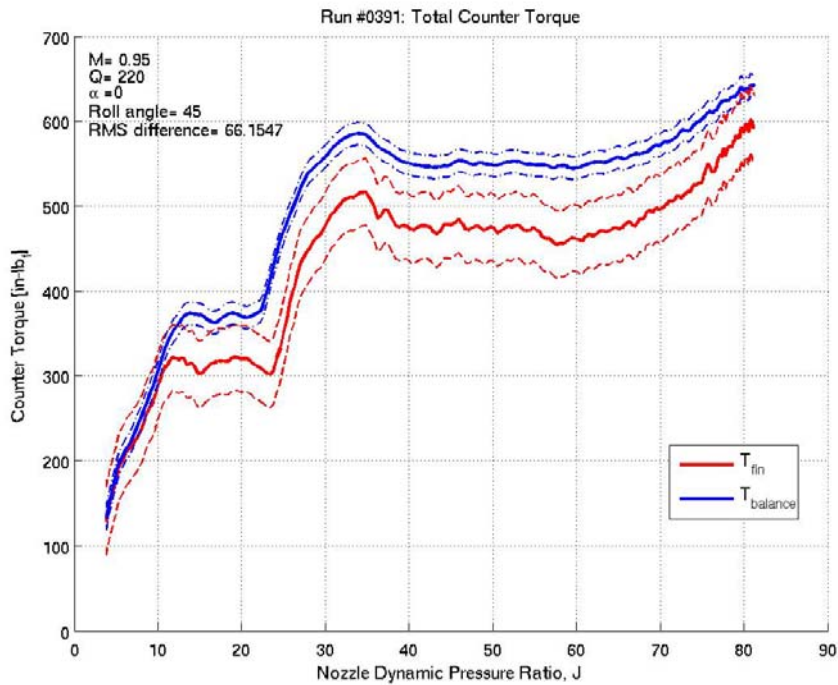


Figure 14. Fin and Balance Counter Torque vs. Nozzle Dynamic Pressure Ratio for Run #0391.

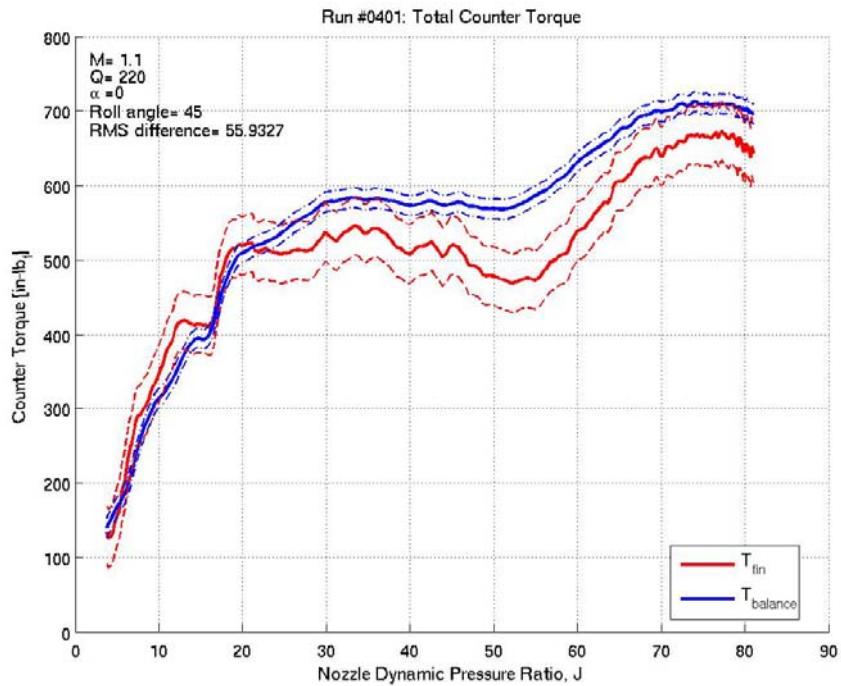


Figure 15. Fin and Balance Counter Torque vs. Nozzle Dynamic Pressure Ratio for Run #0401.

The next investigation involved trials at an angle of attack of 4° . As mentioned previously, data were combined from 2 individual trials to determine counter torque on the FBR at angle of attack. Figures 16-18 display the counter torque for trials at $Q = 220$ psf and are representative of the data at other freestream dynamic pressures. The agreement between fin pressure and balance counter torque is clearly better at angle of attack though differences still often exceed the uncertainty. The magnitude and behavior of the counter torque at Mach 0.8 is similar for 0° and 4° angle of attack. However, the magnitude of counter torque increases significantly at angle of attack for Mach 0.95 and 1.1.

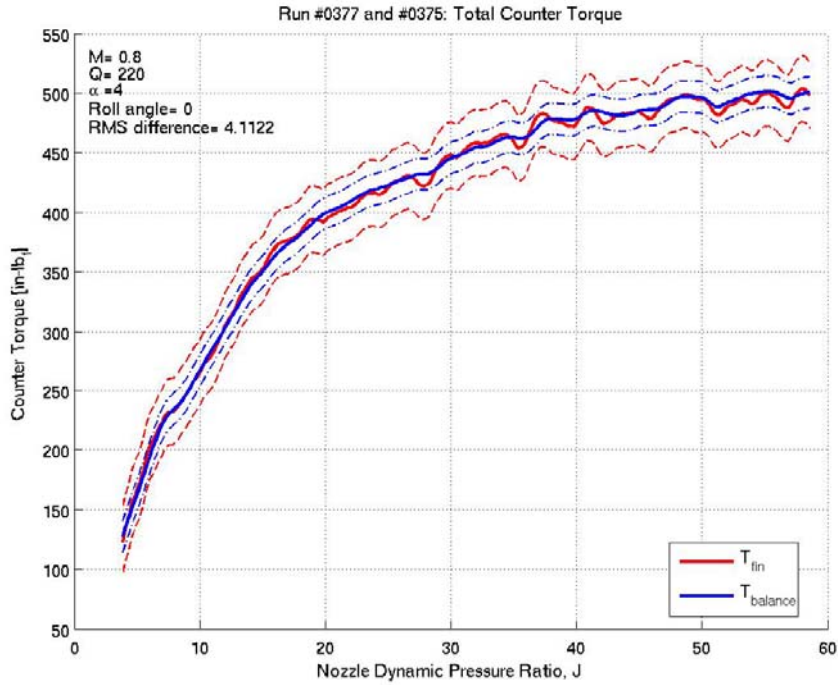


Figure 16. Fin and Balance Counter Torque vs. Nozzle Dynamic Pressure Ratio for Runs #0377 and 0375.

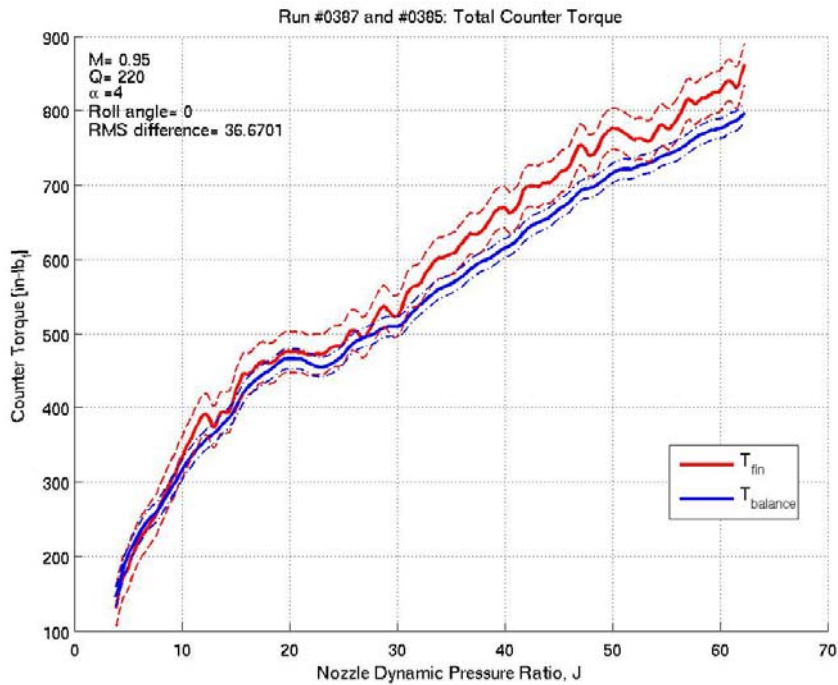


Figure 17. Fin and Balance Counter Torque vs. Nozzle Dynamic Pressure Ratio for Runs #0387 and 0385.

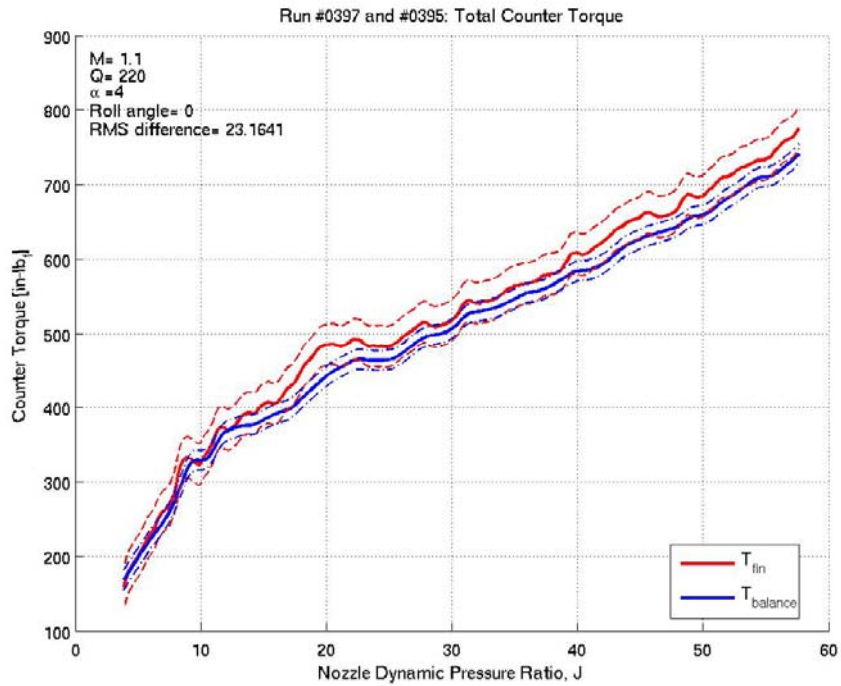


Figure 18. Fin and Balance Counter Torque vs. Nozzle Dynamic Pressure Ratio for Runs #0397 and 0395.

At a given Mach number for cases at angle of attack, the counter torque curve maintains the same general shape but increases in magnitude as the freestream dynamic pressure increases. The agreement between the fin pressure and balance counter torque decreases slightly as Mach number and freestream dynamic pressure are increased, but is nowhere near the marked decrease in agreement seen in zero angle of attack cases. These trends can be seen by comparing Figures 18-20. For a given Mach number and free stream dynamic pressure, the agreement between the balance and fin pressure counter torques is always better at angle of attack.

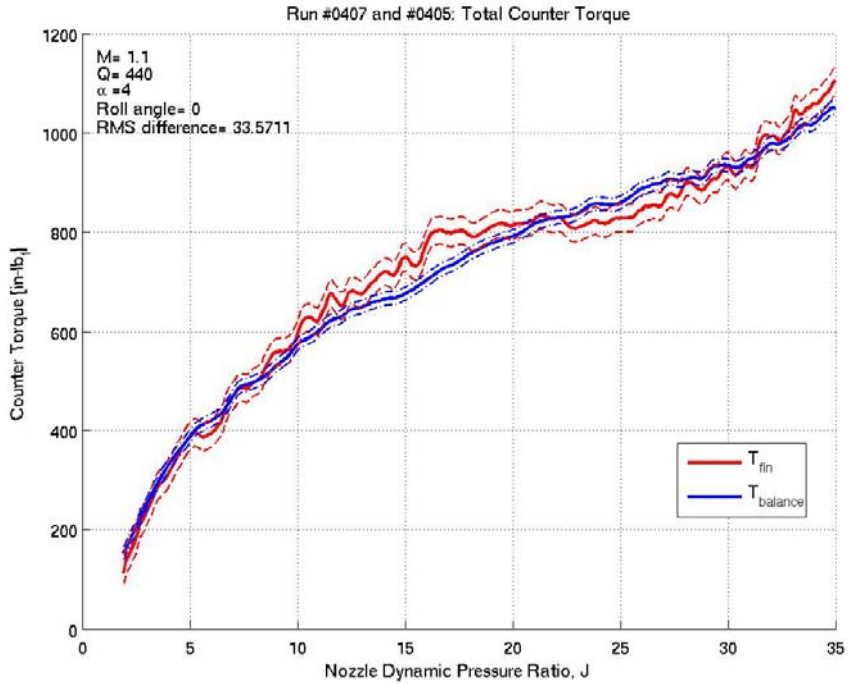


Figure 19. Fin and Balance Counter Torque vs. Nozzle Dynamic Pressure Ratio for Runs #0407 and 0405.

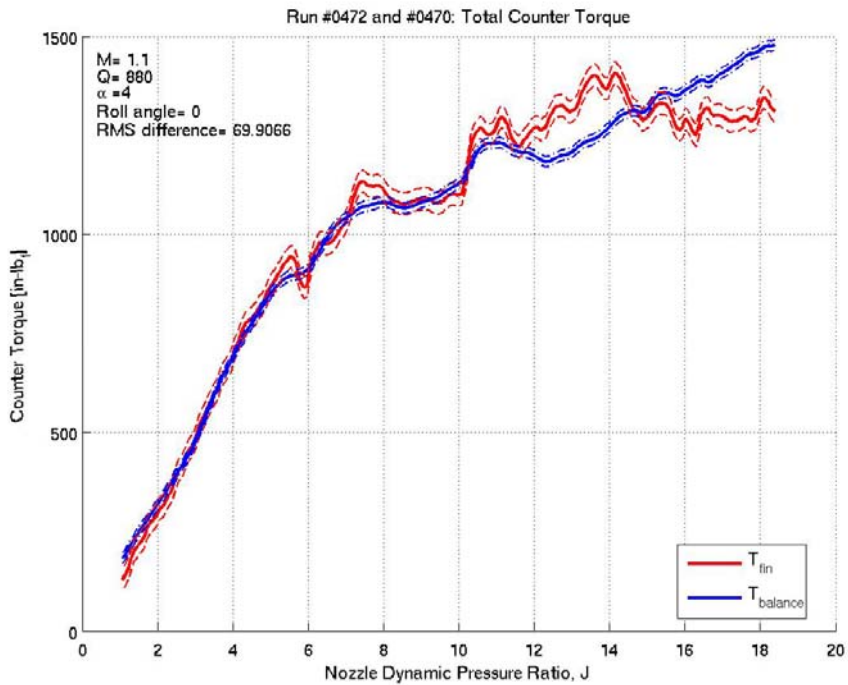


Figure 20. Fin and Balance Counter Torque vs. Nozzle Dynamic Pressure Ratio for Runs #0472 and 0470.

Fin differential pressure coefficient distributions for runs at zero angle of attack are seen in Figures 21-23. Since most of the counter torque acts upon Fin B, only the pressure distribution across this fin is plotted. For a given freestream dynamic pressure and Mach number, plots were generated at $J = 8, 16,$ and 36 . The plots show the differential pressure coefficient variation (odd side C_p minus even side C_p) over the tapped surface. All the plots are characterized by a high pressure region at the leading edge and a large low pressure region over the middle of the tapped surface. The change in pressure coefficient with Mach number can be seen in Figures 22, 24, and 25. An expansion of the large pressure differential is seen at the leading edge as well as an increase in the overall magnitude of the pressure coefficient as Mach number increases. No discernible trend can be seen as freestream dynamic pressure increases in Figures 22, 26, and 27. Similar behavior is seen at all three values of J . Uncertainties in the pressure coefficient are dependent only on freestream dynamic pressure and are displayed in Table 2.

Table 2. Uncertainty in Pressure Coefficient

Q (psf)	C_p Uncertainty
220	0.0245
440	0.0123
880	0.0061

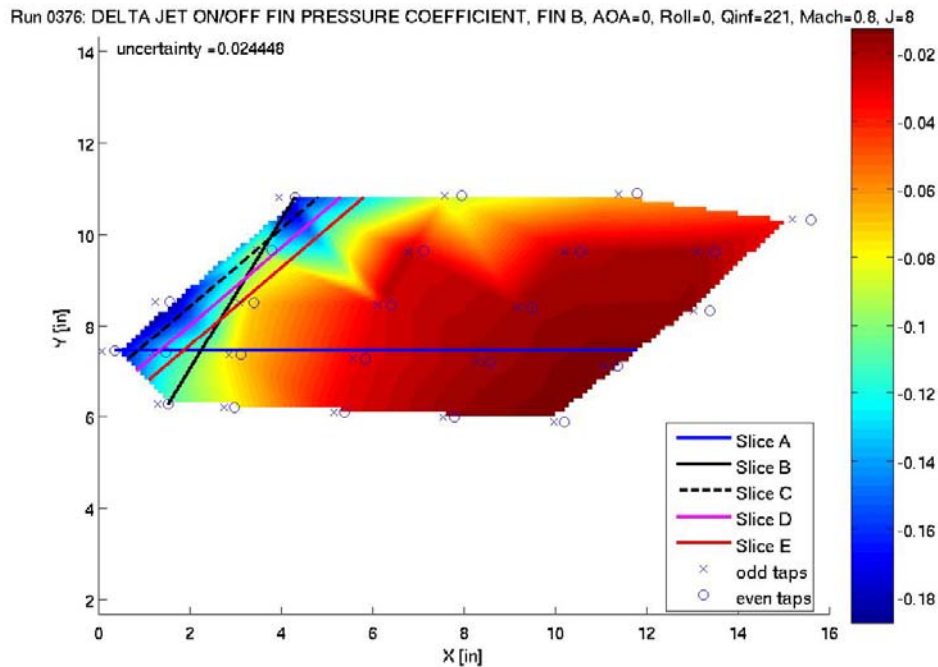


Figure 21. Differential Pressure Coefficient for Run #0376 at J=8.

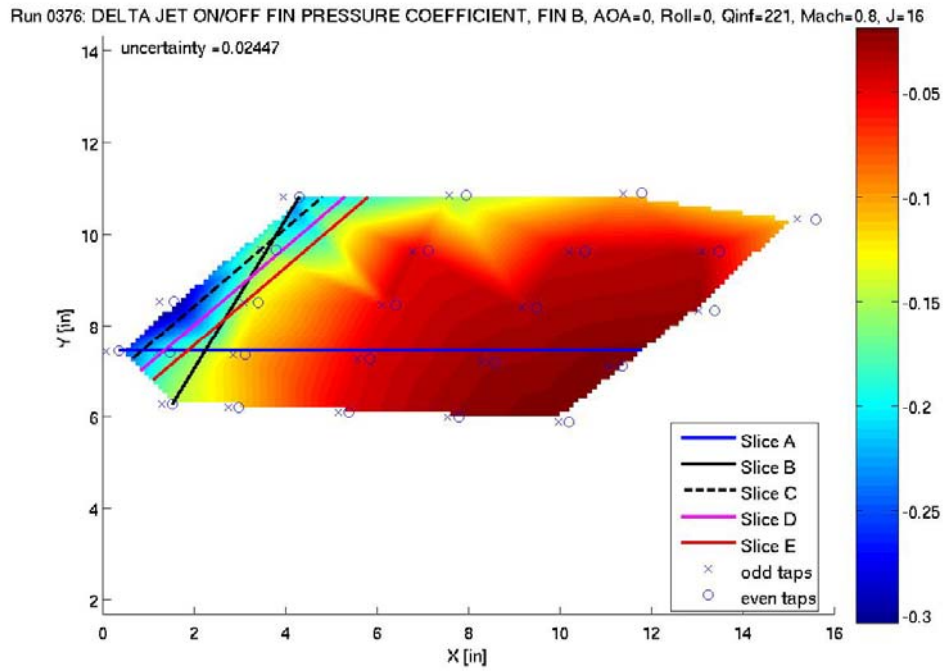


Figure 22. Differential Pressure Coefficient for Run #0376 at J=16.

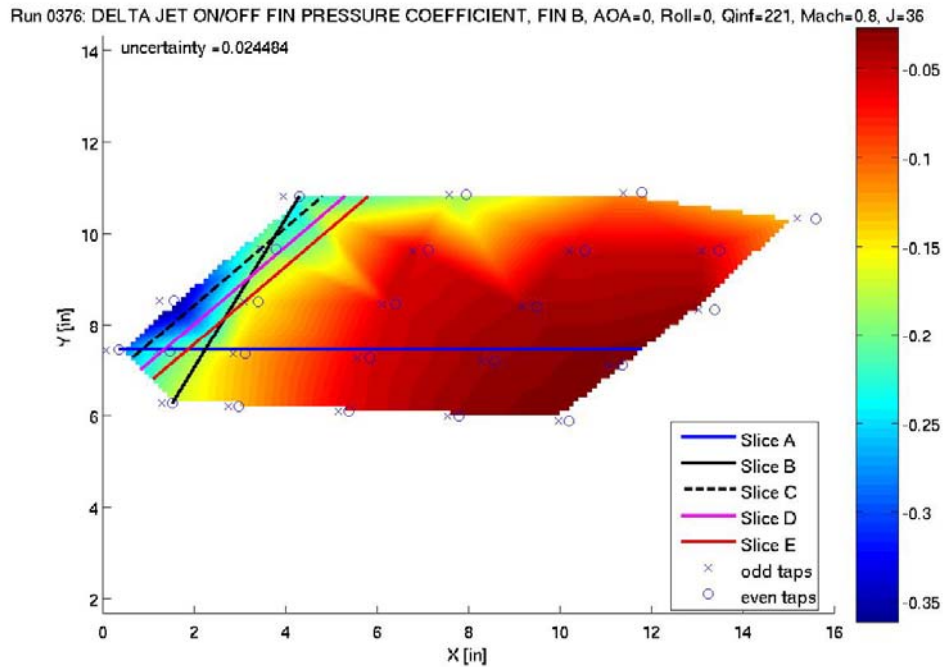


Figure 23. Differential Pressure Coefficient for Run #0376 at J=36.

Run 0386: DELTA JET ON/OFF FIN PRESSURE COEFFICIENT, FIN B, AOA=0, Roll=0, Qinf=221, Mach=0.95, J=16

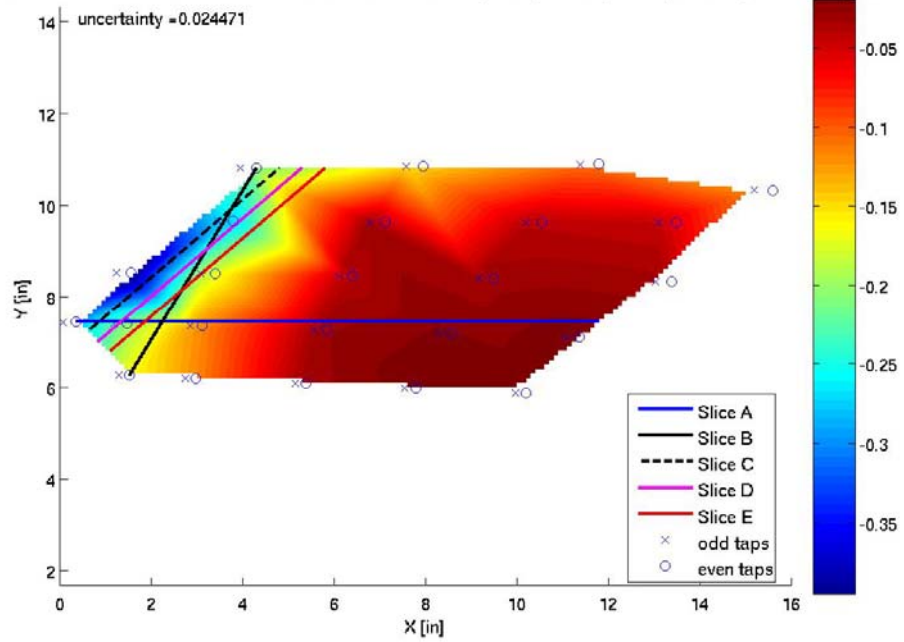


Figure 24. Differential Pressure Coefficient for Run #0386 at J=16.

Run 0396: DELTA JET ON/OFF FIN PRESSURE COEFFICIENT, FIN B, AOA=0, Roll=0, Qinf=221, Mach=1.1, J=16

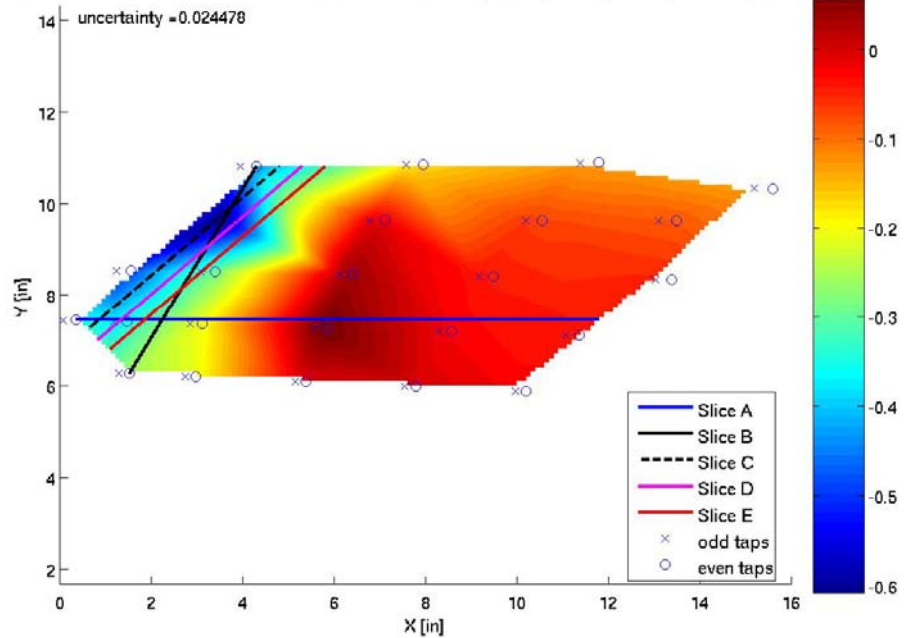


Figure 25. Differential Pressure Coefficient for Run #0396 at J=16.

Run 0406: DELTA JET ON/OFF FIN PRESSURE COEFFICIENT, FIN B, AOA=0, Roll=0, Qinf=441, Mach=1.1, J=16

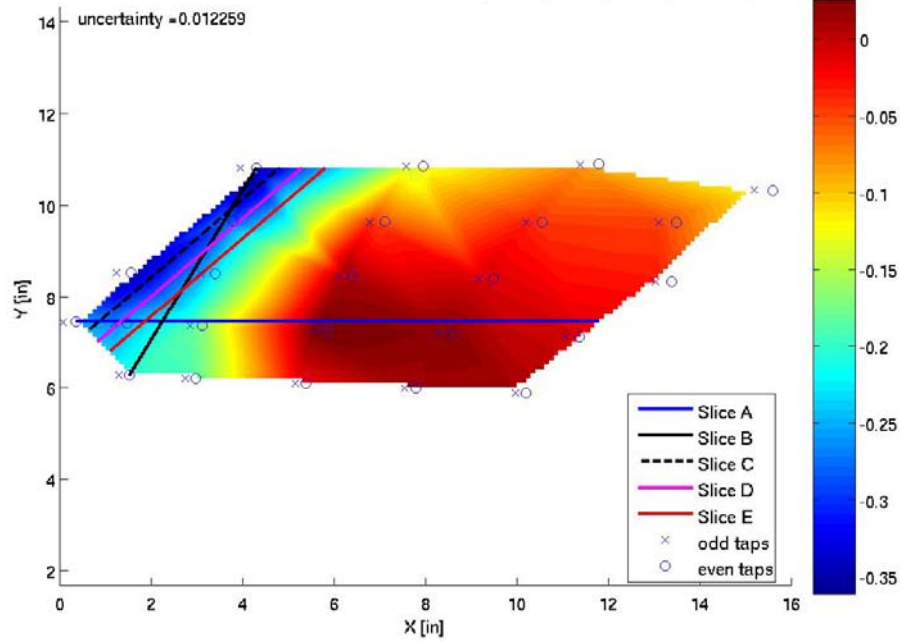


Figure 26. Differential Pressure Coefficient for Run #0406 at J=16.

Run 0471: DELTA JET ON/OFF FIN PRESSURE COEFFICIENT, FIN B, AOA=0, Roll=0, Qinf=882, Mach=1.1, J=16

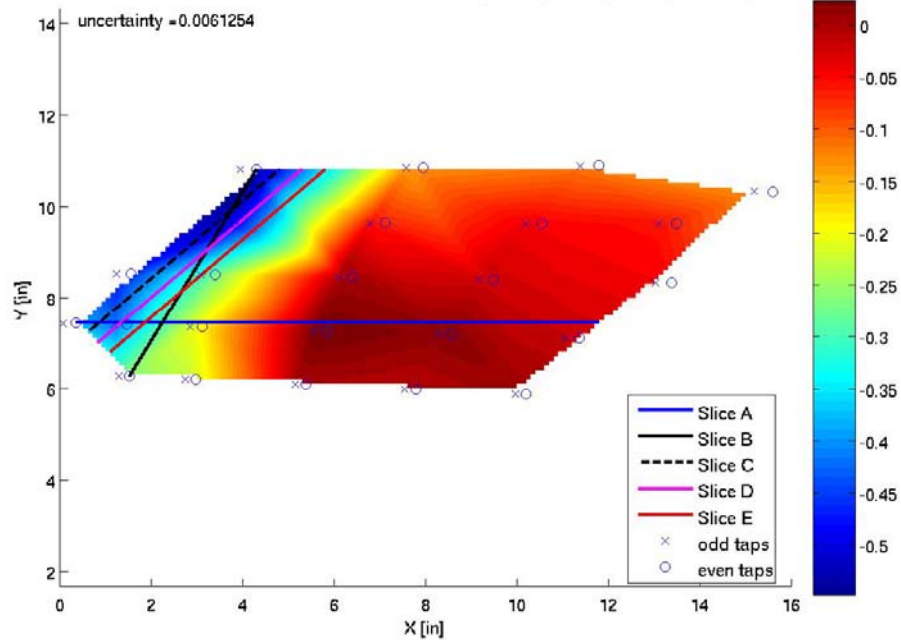


Figure 27. Differential Pressure Coefficient for Run #0471 at J=16.

In order to quantify the variation of pressure coefficient, slices were taken longitudinally across the tapped region (Slice A) and along the leading edge (Slices B-E). According to Figures 28-30, the coefficient of pressure behaves similarly along the longitudinal slice at Mach 0.8 and 0.95 although the magnitude is slightly greater in the latter. At Mach 1.1, a small positive pressure differential arises toward the center of the tapped region. The greatest pressure coefficient magnitude occurs at Mach 1.1, particularly in the peak at the leading edge. There is little change in the pressure coefficient distribution as freestream dynamic pressure varies from 220 to 880 psf. The most significant change in the pressure coefficient distribution occurs between Mach 0.95 and Mach 1.1.

The high pressure area at the leading edge clearly extends beyond the boundaries of the tapped region. This calls into question the validity of the assumption that the ratio of the pressure in the tapped region to the pressure over the fin surface is proportional to the ratio of their respective areas. The existence of complexities beyond the tapped area decreases the accuracy of this assumption. The poor agreement of the balance and pressure-derived torques at 0° angle of attack as previously mentioned may be a result of this problem.

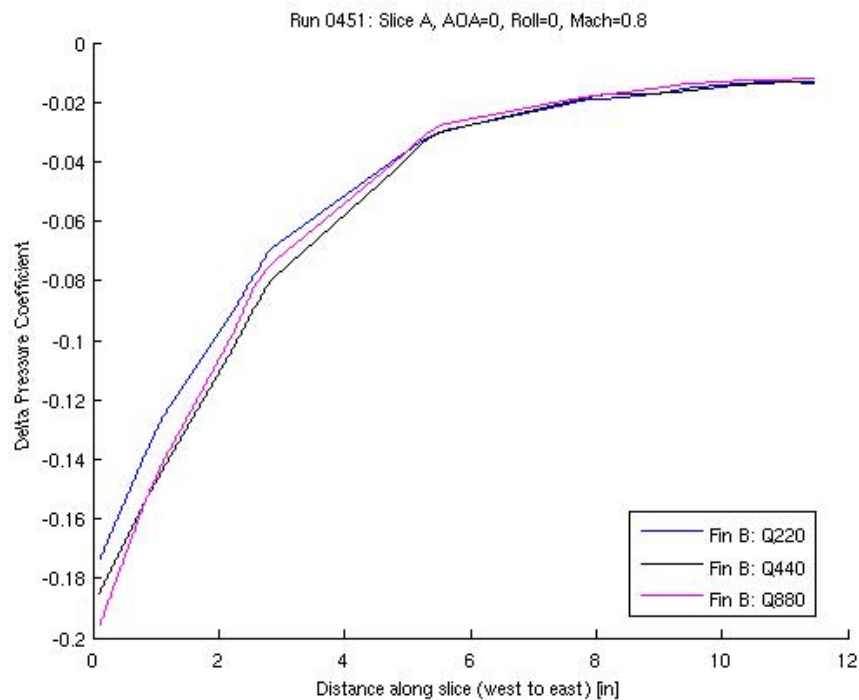


Figure 28. Pressure Coefficient Variation along Slice A at Mach 0.8, J=8.

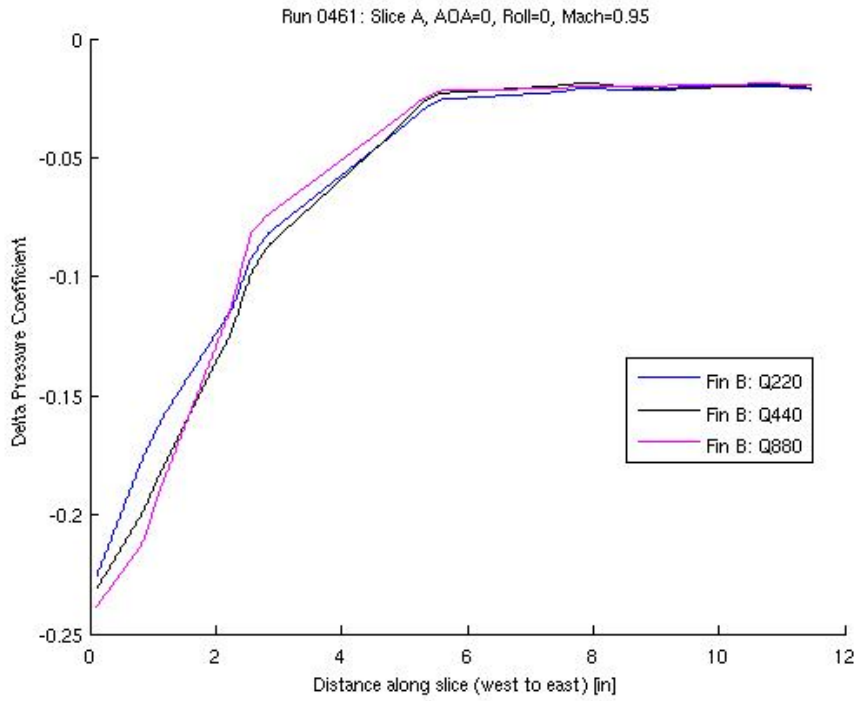


Figure 29. Pressure Coefficient Variation along Slice A at Mach 0.95, J=8.

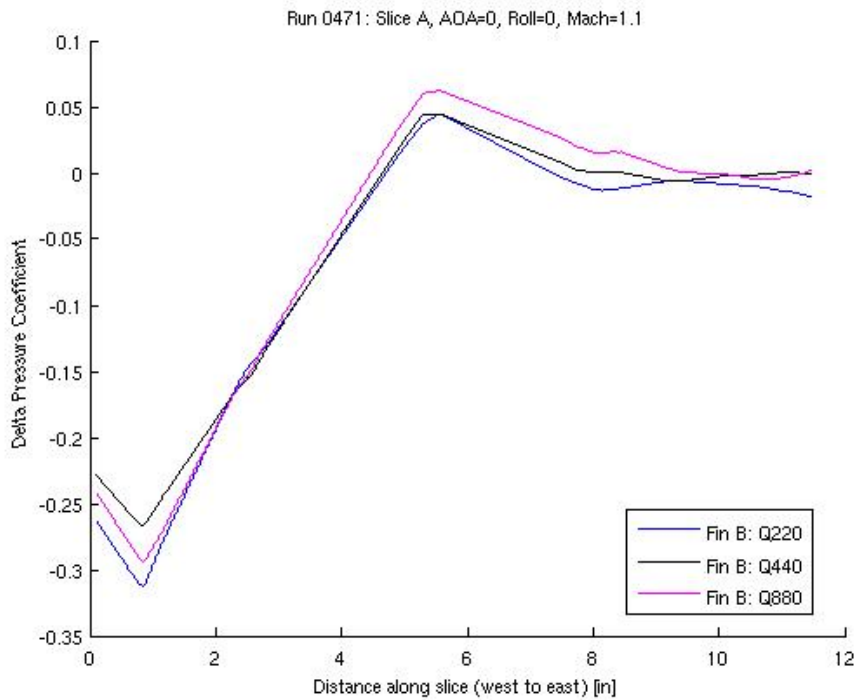


Figure 30. Pressure Coefficient Variation along Slice A at Mach 1.1, J=8.

Plots of counter torque coefficient as a function of J were generated for all run conditions. The counter torque curves at $Q = 220, 440,$ and 880 psf were plotted together for comparison for each Mach number in Figures 31-33. In all cases at zero angle of attack, discrepancies between the balance and fin pressure counter torque coefficients became significant at approximately $J > 10$. However at Mach 0.95 and 1.1 larger differences between the coefficients appear, particularly at $Q = 440$ psf. This behavior reflects similar results obtained in Reference 1. According to Figures 32 and 33, the fin pressure torque follows the general pattern of the balance torque, but is offset downward by a nearly constant factor.

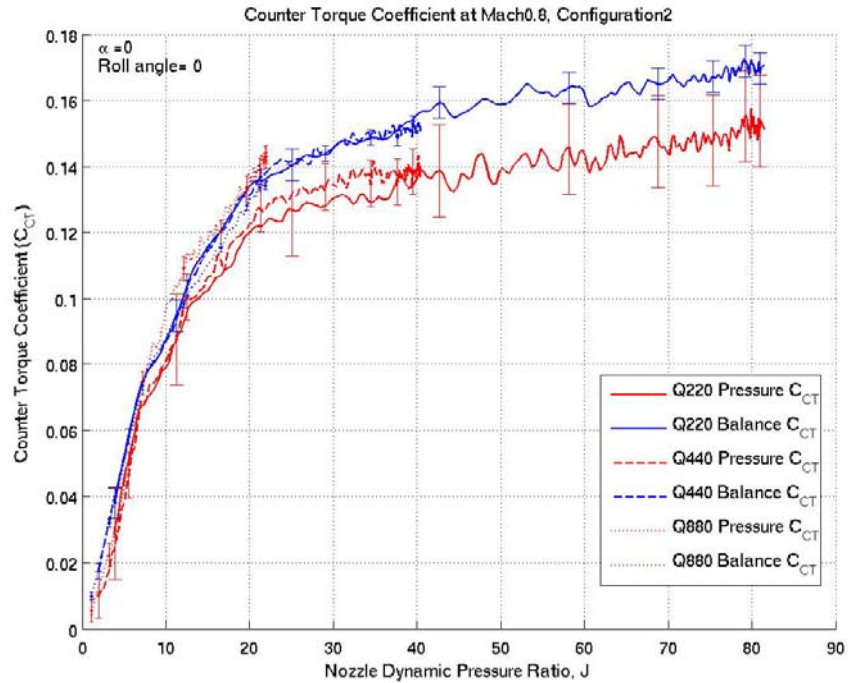


Figure 31. Counter Torque Coefficients at Mach 0.8, 0° Angle of Attack, 0° Roll.

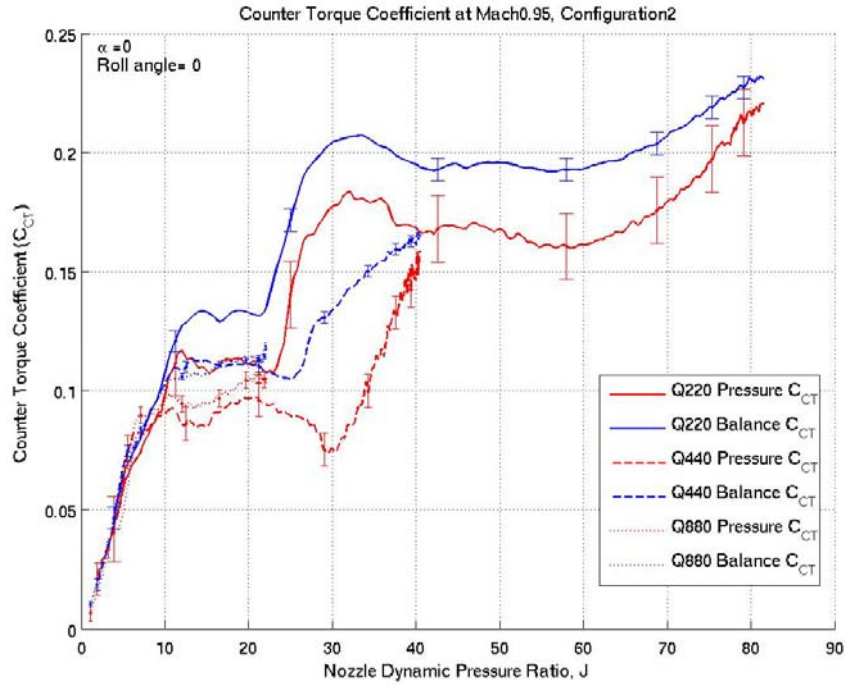


Figure 32. Counter Torque Coefficients at Mach 0.95, 0° Angle of Attack, 0° Roll.

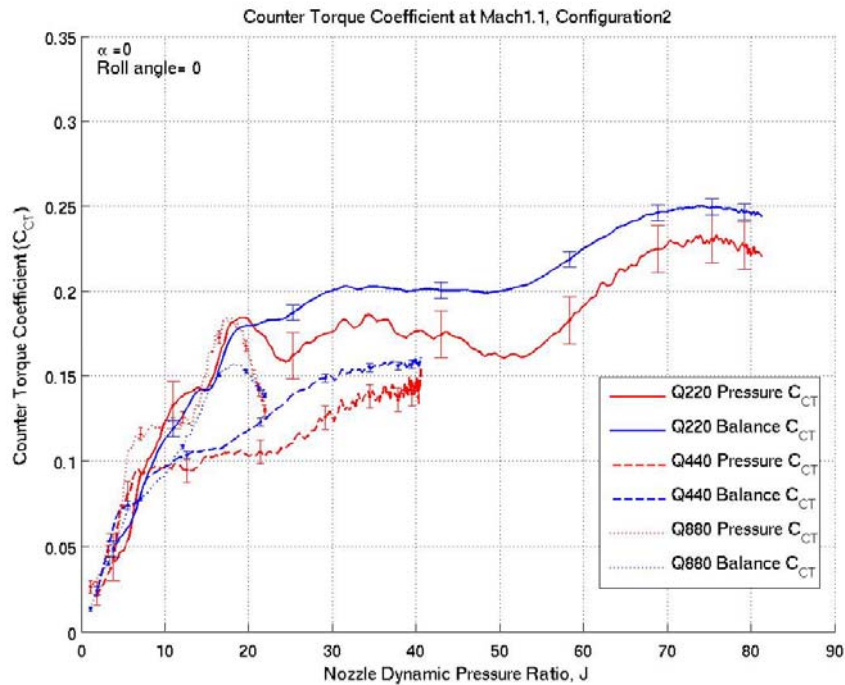


Figure 33. Counter Torque Coefficients at Mach 1.1, 0° Angle of Attack, 0° Roll.

More consistent results were obtained for cases at an angle of attack of 4° and 0° roll angle. The disagreement seen in 0° angle of attack cases is no longer as prominent in Figures 34-36. The behavior of the counter torque coefficient at $Q = 440$ psf is again the most anomalous, though differences are not as large as those at 0° angle of attack.

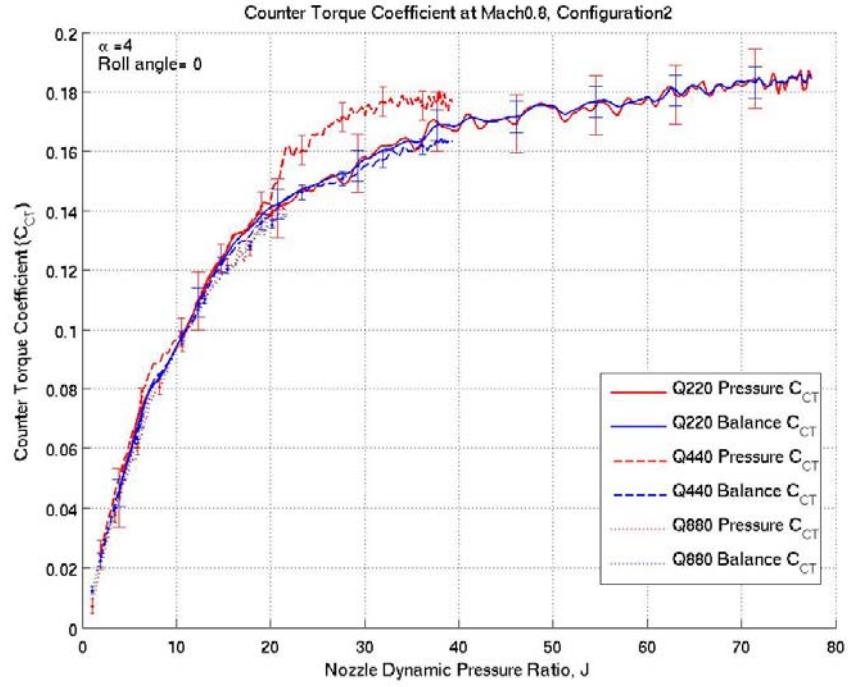


Figure 34. Counter Torque Coefficients at Mach 0.8, 4° Angle of Attack, 0° Roll.

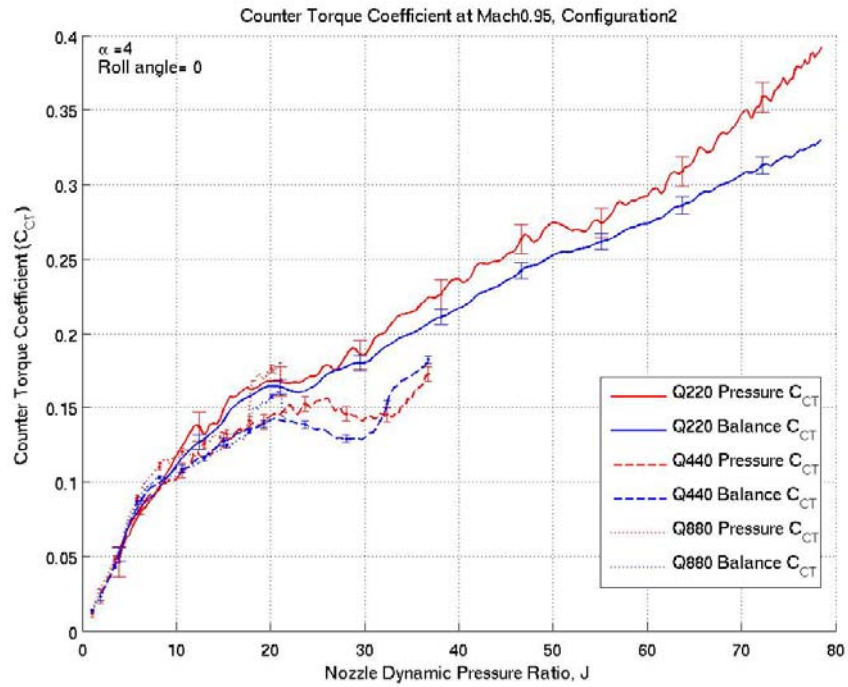


Figure 35. Counter Torque Coefficients at Mach 0.95, 4° Angle of Attack, 0° Roll.

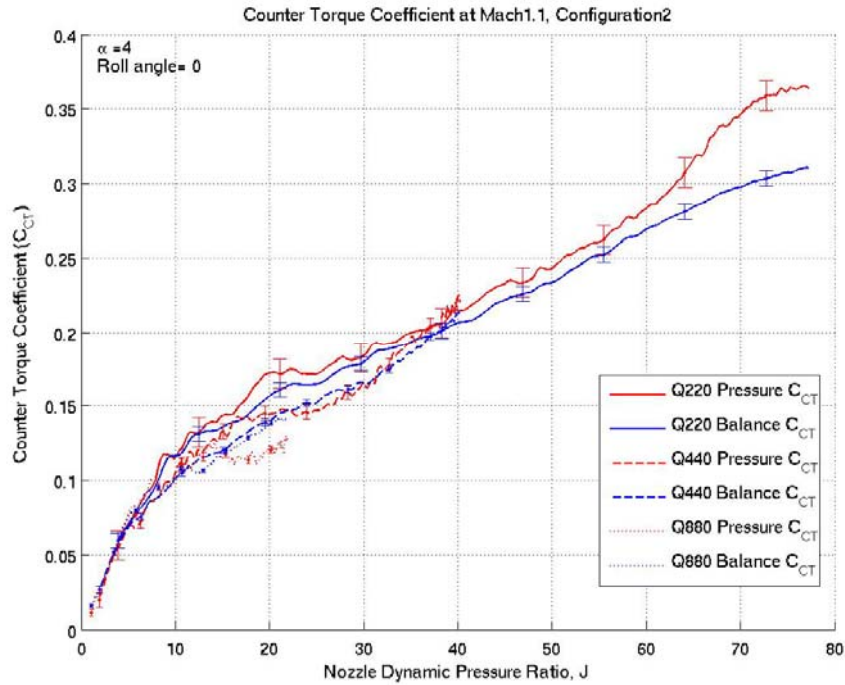


Figure 36. Counter Torque Coefficients at Mach 1.1, 4° Angle of Attack, 0° Roll.

At 0° angle of attack, the counter torque coefficients at 0° and 45° roll angle are nearly identical. Counter torque coefficient behaves similarly at 45° roll and 4° angle of attack as seen in Figures 37-39. The close agreement between the counter-torque coefficients in Figures 34-36 is lost when the FBR is at a roll angle. Again, the most conspicuous discrepancies in the data occur at $Q = 440$ psf.

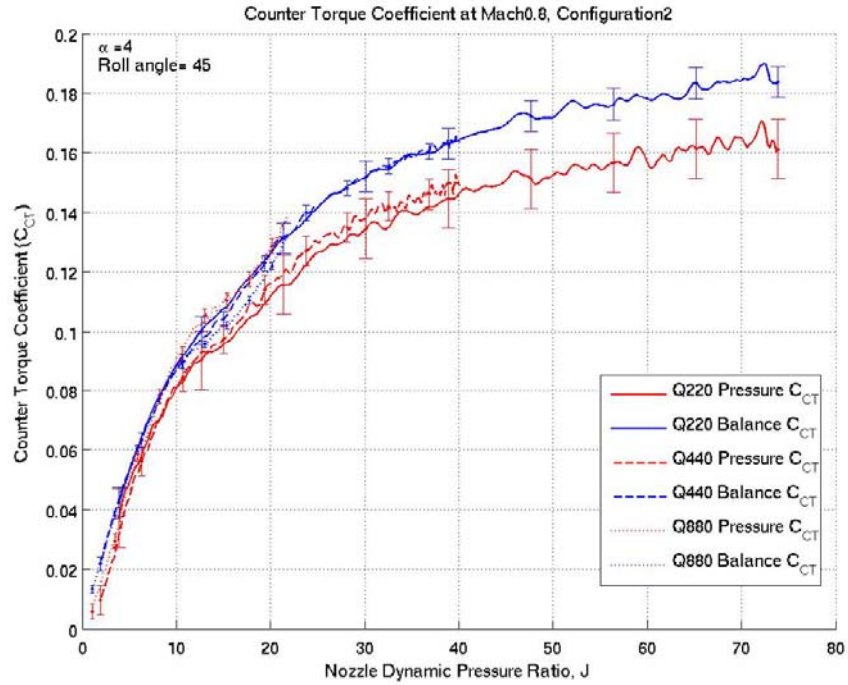


Figure 37. Counter Torque Coefficients at Mach 0.8, 4° Angle of Attack, 45° Roll.

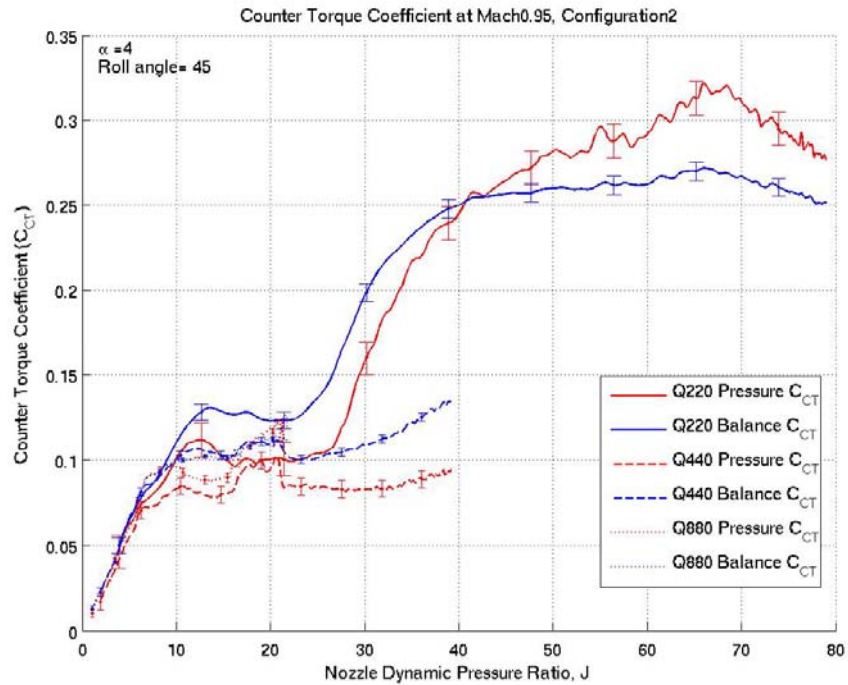


Figure 38. Counter Torque Coefficients at Mach 0.95, 4° Angle of Attack, 45° Roll.

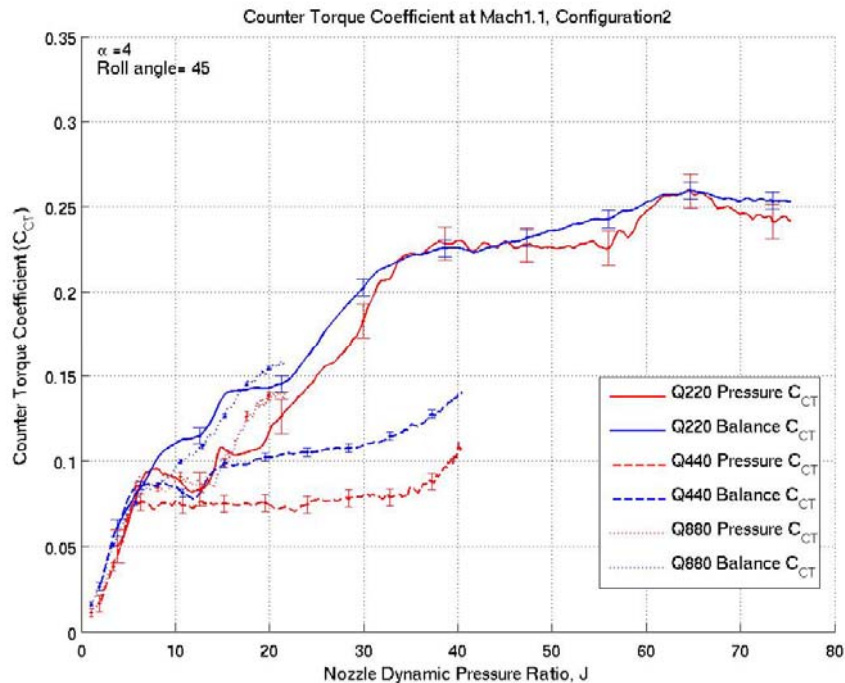


Figure 39. Counter Torque Coefficients at Mach 1.1, 4° Angle of Attack, 45° Roll.

6 Summary and Conclusions

The main objective of the project was to define the behavior of counter torque at various flowfield conditions as derived from fin pressure data. Modification of the original code from Reference 2 revealed that agreement between the fin pressure and balance counter torques was poorer than previously believed because of the jet-off tares. The investigation of pressure coefficient distributions over the tapped region at zero angle of attack implied that the complexity of the pressure distribution beyond the instrumented surface cannot be neglected. MATLAB code was successfully produced to generate counter torque plots for runs at angle of attack. In this case, there was good agreement between the fin pressure and balance results. Roll angle was found to be a significant factor only at angle of attack. In addition, different nozzle configurations had little effect on counter torque. An investigation of the counter torque coefficient verified the aberrant behavior of counter torque at Mach 0.95 and 1.1 as previously observed.

The MATLAB code successfully produced counter torque based on fin pressure. However, there is a large inconsistency between the accuracy of results at 0° and 4° angle of attack. Closer agreement between the balance and fin pressure counter torques was consistently obtained at 4° angle of attack. An investigation of the pressure coefficient distribution across the fins at 4° angle of attack may help to identify the problem. A better

method of obtaining the fin counter torque from the torque acting upon the tapped region may need to be developed at zero angle of attack. Also, correlation of the fin pressure distributions to the Particle Image Velocimetry results in Reference 4 may provide useful information.

References

1. Peterson, C. W., Wolfe, W. P., and Payne, J. L., "Experiments and Computations of Roll Torque Induced by Vortex-fin Interaction," Paper No. AIAA-2004-1069, 42nd AIAA Aerospace Sciences Meeting, Reno, NV, January 2004.
2. Beresh, S. J. "Stereoscopic PIV for Jet/Fin Interaction Measurements on a Full-Scale Flight Vehicle Configuration," Paper No. AIAA2005-0442, 43rd AIAA Aerospace Sciences Meeting, Reno, NV, January 2005.
3. "MATLAB: The Language of Technical Computing," Version 6, The MathWorks, Inc. Natick, MA 2000.
4. Vijlee, S. Z., "An Automated Procedure for Analyzing the Effects of Vortex-Induced Fin Pressure on Roll Torque for a Finned Body of Revolution," SAND2004-4378, Sandia National Laboratories, Albuquerque, NM, September 2004.
5. Coleman, H. W. and Steele, Jr., W. G. *Experimentation and Uncertainty Analysis for Engineers*, 2nd Edition, John Wiley & Sons; New York, NY: 1999.

Appendix A: MATLAB M-Files

VFI_torques_0_AOA.m

```
%plots counter torque at 0 angle of attack
clear all
clc

%global variable declaration
global dir run J FA_p FB_p tap_coords fin_area
global M data tare_pressures

%conversions
psf2psi = 0.0069444;%convert from psf to psi

%user input
dir = input('Enter directory name (ex. C2_Q440_M8): ','s');%the
directory with DAT files
run = input('Enter the run number (ex. 0426): ','s');%the first DAT
file

%file input
data = load([dir '/RUN0' run '.dat']);%loads entire DAT file
tap_coords = load(['fin_taps.dat']);%loads entire taps file that
contains coordinates for the taps

%determining minimum and maximum J values allowed for data
a=find (data(:,427)>4.6, 1);
b=find (data(:,427)>2.3, 1);
c=find (data(:,427)>1.1, 1);

i=find (data(:,427)>73.7, 1);
j=find (data(:,427)>36.8, 1);
k=find (data(:,427)>18.4, 1);

%finds point at which J begins to decrease and discards subsequent
data
if isempty(i)==1 i(1) =
min(find(diff(data(100:size(data,1),427))<0)); end
if isempty(j)==1 j(1) =
min(find(diff(data(100:size(data,1),427))<0)); end
if isempty(k)==1 k(1) =
min(find(diff(data(100:size(data,1),427))<0)); end

if 215<=(round (data(1,7)/10))*10<=225%sets minimum J value for Q220
data
```

```

    minsize = a(1);
elseif 435<=(round (data(1,7)/10))*10<=445%sets minimum J value for
Q440 data
    minsize = b(1);
else minsize = c(1);%sets minimum J value for Q880 data
end

if 215<=(round (data(1,7)/10))*10<=225%sets maximum J value for Q220
data
    maxsize = i(1);
elseif 435<=(round (data(1,7)/10))*10<=445%sets maximum J value for
Q440 data
    maxsize = j(1);
else maxsize = k(1);%sets maximum J value for Q880 data
end

%calculating area of whole fin
fin_dim = load('fin_shape.dat');%loads coordinates of fin edge
points
fin_x = fin_dim(:,1);%maintains edge x-coordinates
fin_y = fin_dim(:,2);%maintains edge y-coordinates
fin_area = polyarea(fin_x,fin_y);%calculates area of whole fin

%data filter
J = data(:,427);%maintains Nozzle Dynamic Pressure Ratio from DAT
file, in column 427
FA_p = data(:, [87:134]) * psf2psi;%maintains 48 columns of pressure
data for Fin A
FB_p = data(:, [135:182]) * psf2psi;%maintains 48 columns of pressure
data for Fin B
T_roll = abs(data(:,29));%maintains Roll Torques from Aerodynamic
Balance

%run identification
AOA = int2str (round (data(1,13)));%reads angle of attack from DAT
file
M= num2str((round (data(1,4)*100))/100);%reads Mach number from DAT
file
Q= dir(5:7);%reads Q from DAT file
rollangle= num2str(round(data(1,14)));%reads roll angle from DAT
file

%loading tare pressures
tare_pressures = load ([dir(1:2) '_0AOA_' rollangle
'rolltare.txt']);

%function calls
T_FAodd = FAodd;%calls function FAodd stores fin pressure, roll
torques on Fin A, Odd
T_FBodd = FBodd;%calls function FBodd stores fin pressure, roll
torques on Fin B, Odd

```

```

T_FAeven = FAeven;%calls function FAeven stores fin pressure, roll
torques on Fin A, Even
T_FBEven = FBeven;%calls function FBeven stores fin pressure, roll
torques on Fin B, Even

%calculations
T_pressure = 2*(T_FAodd + T_FAeven + T_FBodd + T_FBeven);%determines
total torque from fin pressures,
...factor of 2 is for the other 2 symmetric fins
T_FA = T_FAodd + T_FAeven;%determines total torque on Fin A
T_FB = T_FBodd + T_FBeven;%determines total torque on Fin B

%error analysis
T_diff= T_roll-T_pressure;%determines difference between balance and
fin pressure torques
RMSdiff= num2str((mean(T_diff.^2))^0.5);%determines RMS difference
between balance and fin pressure torques

%plots both balance-measured torque and fin pressure torque on same
axis
%and saves it in the same directory with DAT file
hold on
plot(J,T_pressure, 'r', 'LineWidth', 2);
plot(J,T_roll, 'b', 'LineWidth', 2);
plot(J,(T_pressure+39.3264), 'r--', 'LineWidth', 1);%plots error in
pressure torques
plot(J,(T_pressure-39.3264), 'r--', 'LineWidth', 1);%plots error in
pressure torques
plot(J,(T_roll+13.2827), 'b-.', 'LineWidth', 1);%plots error in
balance torques
plot(J,(T_roll-13.2827), 'b-.', 'LineWidth', 1);%plots error in
balance torques
text(0.025, 0.975,['M= ' M ''], 'sc');
text(0.025, 0.945,['Q= ' Q ''], 'sc');
text(0.025, 0.915,['\alpha = ' AOA ''], 'sc');
text(0.025, 0.885,['Roll angle= ' rollangle''], 'sc');
text(0.025, 0.855,['RMS difference= ' RMSdiff''], 'sc');
grid on
legend('T_f_i_n', 'T_b_a_l_a_n_c_e', 'Location', 'Best');
title(['Run #' run ': Total Counter Torque']);
xlabel('Nozzle Dynamic Pressure Ratio, J');
ylabel('Counter Torque [in-lb_f]');
saveas(gcf, [dir '/RUN0' run 'total'], 'jpg');
hold off
close

%plots total roll torque on Fin A versus J and saves it in the
directory with
%the DAT files
plot(J, T_FA, 'LineWidth', 1);
grid on
title(['Run #' run ': Total Counter Torque on Fin A']);

```

```

xlabel('Nozzle Dynamic Pressure Ratio, J');
ylabel('Counter Torque [in-lb_f]');
saveas(gcf,[dir '/RUN0' run 'finAtotal'],'jpg');
close

%plots total roll torque on Fin B versus J and saves it in the
directory with
%the DAT files
plot(J, T_FB, 'LineWidth', 1);
grid on
title(['Run #' run ': Total Counter Torque on Fin B']);
xlabel('Nozzle Dynamic Pressure Ratio, J');
ylabel('Counter Torque [in-lb_f]');
saveas(gcf,[dir '/RUN0' run 'finBtotal'],'jpg');
close

```

FAodd.m

```

function T_FAodd = FAodd

%global variable declaration
global dir run J FA_p tap_coords fin_area tare_pressures

%data filter
p = FA_p(:,[1:2:47]);%maintains pressure data for Fin B, odd
pressure taps
x = tap_coords(:,1);%maintains x coordinates for Fin A, odd pressure
taps
y = tap_coords(:,2);%maintains y coordinates for Fin A, odd pressure
taps
z = tap_coords(:,3);%maintains z coordinates for Fin A, odd pressure
taps

%reading appropriate tare pressure from file
if str2num(dir(10:length(dir)))==8 && str2num(dir(5:7))==220
    tare_pressure=tare_pressures(1);
elseif str2num(dir(10:length(dir)))==95 && str2num(dir(5:7))==220
    tare_pressure=tare_pressures(5);
elseif str2num(dir(10:length(dir)))==11 && str2num(dir(5:7))==220
    tare_pressure=tare_pressures(9);
elseif str2num(dir(10:length(dir)))==8 && str2num(dir(5:7))==440
    tare_pressure=tare_pressures(13);
elseif str2num(dir(10:length(dir)))==95 && str2num(dir(5:7))==440
    tare_pressure=tare_pressures(17);
elseif str2num(dir(10:length(dir)))==11 && str2num(dir(5:7))==440
    tare_pressure=tare_pressures(21);
elseif str2num(dir(10:length(dir)))==8 && str2num(dir(5:7))==880
    tare_pressure=tare_pressures(25);
elseif str2num(dir(10:length(dir)))==95 && str2num(dir(5:7))==880
    tare_pressure=tare_pressures(29);

```

```

elseif str2num(dir(10:length(dir)))==11 && str2num(dir(5:7))==880
    tare_pressure=tare_pressures(34);
else disp('Error identifying tare pressure')
end

%function calls
[vertices, area_vectors, r, a_surf] = vert_and_area(x, y, z);%calls
function to determine triangle vertices,
...area vectors, triangle centroids, and area on Fin A, Odd within
tapped region
p_triangle = triangle_pressures(p, vertices);%calls function to
determine mean pressures for each triangle
T_tot = moment(p_triangle, area_vectors, r);%calls function to
determine torque for the entire surface, Fin A:Odd

%calculations
T_tot_x = -1.*T_tot(:,1);

%plots roll torque for Fin A, Odd versus J and saves it in directory
with
%DAT file
plot(J, T_tot_x, 'LineWidth', 1)
grid on
ylabel('Counter Torque [in-lb_f]')
xlabel('Nozzle Dynamic Pressure Ratio, J');
title(['Run #' run ': Fin A, Odd Taps']);
saveas(gcf,[dir '/RUN0' run 'finAodd'],'jpg');
close

%scales torque to approximate the torque over the entire Fin A, Odd
T_FAodd = T_tot_x*(fin_area/a_surf)-tare_pressure;

```

VFI_torques_AOA.m

```

clear all
clc
format long

%global variable declaration
global dir run_pos run_neg J_pos J_neg FA_p_pos FA_p_neg FB_p_pos
FB_p_neg tap_coords fin_area
global AOA_pos AOA_neg M data_pos data_neg tare_pressures

%conversions
psf2psi = 0.0069444;%convert from psf to psi

%user input
dir = input('Enter directory name (ex. C2_Q440_M8): ','s');%the
directory with DAT files

```

```

run_pos = input('Enter the run number for test with positive angle
of attack (ex. 0426): ','s');%the first DAT file
run_neg = input('Enter the run number for test with negative angle
of attack (ex. 0426): ','s');%the second DAT file

%file input
data_pos = load([dir '/RUN0' run_pos '.dat']);%loads positive angle
of attack DAT file
data_neg = load([dir '/RUN0' run_neg '.dat']);%loads negative angle
of attack DAT file
tap_coords = load(['fin_taps.dat']);%loads entire taps file that
contains coordinates for the taps

%constructing range of J values allowed for data
a=find (4.6<data_pos(:,427)<73.7);%J limits on Q220 data, positive
AOA
b=find (2.3<data_pos(:,427)<36.8);%J limits on Q440 data, positive
AOA
c=find (1.1<data_pos(:,427)<18.4);%J limits on Q880 data, positive
AOA
d=find (4.6<data_neg(:,427)<73.7);%J limits on Q220 data, negative
AOA
e=find (2.3<data_neg(:,427)<36.8);%J limits on Q440 data, negative
AOA
f=find (1.1<data_neg(:,427)<18.4);%J limits on Q880 data, negative
AOA

%sets range of J values for Q220 data, positive AOA
if str2num(dir(5:7))==220 && round (data_pos(1,13))==4
    if isempty(find(diff(data_pos(:,427))<0))==0%filtering out bad
data (decreasing J values)
        if
min(find(diff(data_pos(1:min(size(data_neg,1),size(data_pos,1)),427)
)<0))<20%filters bad data at beginning

s=find(diff(data_pos(1:min(size(data_neg,1),size(data_pos,1)),427))<
0);
            s=1+max(s(s<20));
            a_start=s;
        else a_start=1;
        end

        if
isempty(find(diff(data_pos(100:min(size(data_neg,1),size(data_pos,1)
),427))<0))==0%filters bad data at end
            a_end=
min(find(diff(data_pos(100:min(size(data_neg,1),size(data_pos,1)),42
7))<0));
            else a_end=size(data_pos,1);
            end

        datasize_pos=a(a_start:a_end);

```

```

else
    datasize_pos = a;
end

%sets range of J values for Q440 data, positive AOA
elseif str2num(dir(5:7))==440 && round (data_pos(1,13))==4
    if isempty(find(diff(data_pos(:,427))<0))==0%filtering out bad
data (decreasing J values)
        if
min(find(diff(data_pos(1:min(size(data_neg,1),size(data_pos,1)),427)
)<0))<20%filters bad data at beginning

s=find(diff(data_pos(1:min(size(data_neg,1),size(data_pos,1)),427))<
0);
            s=1+max(s(s<20));
            b_start=s;
        else b_start=1;
        end

        if
isempty(find(diff(data_pos(100:min(size(data_neg,1),size(data_pos,1)
),427))<0))==0%filters bad data at end
            b_end=
min(find(diff(data_pos(100:min(size(data_neg,1),size(data_pos,1)),42
7))<0));
        else b_end=size(data_pos,1);
        end

        datasize_pos=b(b_start:b_end);
    else
        datasize_pos = b;
    end

%sets range of J values for Q880 data, positive AOA
elseif str2num(dir(5:7))==880 && round (data_pos(1,13))==4
    if isempty(find(diff(data_pos(:,427))<0))==0%filtering out bad
data (decreasing J values)
        if
min(find(diff(data_pos(1:min(size(data_neg,1),size(data_pos,1)),427)
)<0))<20%filters bad data at beginning

s=find(diff(data_pos(1:min(size(data_neg,1),size(data_pos,1)),427))<
0);
            s=1+max(s(s<20));
            c_start=s;
        else c_start=1;
        end

        if
isempty(find(diff(data_pos(100:min(size(data_neg,1),size(data_pos,1)
),427))<0))==0%filters bad data at end

```



```

        c_end=
min(find(diff(data_pos(100:min(size(data_neg,1),size(data_pos,1)),42
7))<0));
        else c_end=size(data_pos,1);
        end

        datasize_pos=c(c_start:c_end);
    else
        datasize_pos = c;
    end

else disp('Error in data limits')
end

%sets range of J values for Q220 data, negative AOA
if str2num(dir(5:7))==220 && round(data_neg(1,13))==-4
    if isempty(find(diff(data_neg(:,427))<0))==0%filtering out bad
data (decreasing J values)
        if
min(find(diff(data_neg(1:min(size(data_neg,1),size(data_pos,1)),427)
)<0))<20%filters bad data at beginning

t=find(diff(data_neg(1:min(size(data_neg,1),size(data_pos,1)),427))<
0);
            t=1+max(t(t<20));
            d_start=t;
        else d_start=1;
        end

        if
isempty(find(diff(data_neg(100:min(size(data_neg,1),size(data_pos,1)
),427))<0))==0%filters bad data at end
            d_end=
min(find(diff(data_neg(100:min(size(data_neg,1),size(data_pos,1)),42
7))<0));
            else d_end=size(data_neg,1);
            end

            datasize_neg=d(d_start:d_end);
        else
            datasize_neg = d;
        end

%sets range of J values for Q440 data, negative AOA
elseif str2num(dir(5:7))==440 && round(data_neg(1,13))==-4
    if isempty(find(diff(data_neg(:,427))<0))==0%filtering out bad
data (decreasing J values)
        if
min(find(diff(data_neg(1:min(size(data_neg,1),size(data_pos,1)),427)
)<0))<20%filters bad data at beginning

```

```

t=find(diff(data_neg(1:min(size(data_neg,1),size(data_pos,1)),427))<
0);
    t=1+max(t(t<20));
    e_start=t;
    else e_start=1;
    end

    if
isempty(find(diff(data_neg(100:min(size(data_neg,1),size(data_pos,1)
),427))<0))==0%filters bad data at end
        e_end=
min(find(diff(data_neg(100:min(size(data_neg,1),size(data_pos,1)),42
7))<0));
        else e_end=size(data_neg,1);
        end

        datasize_neg=e(e_start:e_end);
    else
        datasize_neg = e;
    end

%sets range of J values for Q880 data, negative AOA
elseif str2num(dir(5:7))==880 && round (data_neg(1,13))==-4
    if isempty(find(diff(data_neg(:,427))<0))==0%filtering out bad
data (decreasing J values)
        if
min(find(diff(data_neg(1:min(size(data_neg,1),size(data_pos,1)),427)
)<0))<20%filters bad data at beginning

t=find(diff(data_neg(1:min(size(data_neg,1),size(data_pos,1)),427))<
0);
    t=1+max(t(t<20));
    f_start=t;
    else f_start=1;
    end

    if
isempty(find(diff(data_neg(100:min(size(data_neg,1),size(data_pos,1)
),427))<0))==0%filters bad data at end
        f_end=
min(find(diff(data_neg(100:min(size(data_neg,1),size(data_pos,1)),42
7))<0));
        else f_end=size(data_neg,1);
        end

        datasize_neg=f(f_start:f_end);
    else
        datasize_neg = f;
    end

else disp ('Error in data limits')

```

```

end

%calculating area of whole fin
fin_dim = load('fin_shape.dat');%loads coordinates of fin edge
points
fin_x = fin_dim(:,1);%maintains edge x-coordinates
fin_y = fin_dim(:,2);%maintains edge y-coordinates
fin_area = polyarea(fin_x,fin_y);%calculates area of whole fin

%data filter
J_pos = data_pos(datasize_pos,427);%maintains Nozzle Dynamic
Pressure Ratio from positive AOA DAT file, in column 427
J_neg = data_neg(datasize_neg,427);%maintains Nozzle Dynamic
Pressure Ratio from negative AOA DAT file, in column 427
FA_p_pos = data_pos(datasize_pos,[87:134]) * psf2psi;%maintains 48
columns of pressure data for positive AOA Fin A
FA_p_neg = data_neg(datasize_neg,[87:134]) * psf2psi;%maintains 48
columns of pressure data for negative AOA Fin A
FB_p_pos = data_pos(datasize_pos,[135:182]) * psf2psi;%maintains 48
columns of pressure data for positive AOA Fin B
FB_p_neg = data_neg(datasize_neg,[135:182]) * psf2psi;%maintains 48
columns of pressure data for negative AOA Fin B
T_roll_pos = abs(data_pos(datasize_pos,29));%maintains Roll Torques
from positive AOA Aerodynamic Balance
T_roll_neg = abs(data_neg(datasize_neg,29));%maintains Roll Torques
from negative AOA Aerodynamic Balance

%run identification
AOA_pos = int2str (round (data_pos(1,13)));%reads angle of attack
from DAT file
AOA_neg = int2str (round (data_neg(1,13)));%reads angle of attack
from DAT file
M= num2str((round (data_pos(1,4)*100))/100);%reads Mach number from
DAT file
Q= dir(5:7);%reads Q from DAT file
rollangle= num2str(round(data_pos(1,14)));%reads roll angle from DAT
file

%loading tare pressures
tare_pressures = load ([dir(1:2) 'AOA' rollangle 'rolltare.txt']);

%function calls
T_FAodd_pos = FAodd_pos;%calls function FAodd_pos stores fin
pressure, roll torques on Fin A, Odd, positive AOA
T_FAodd_neg = FAodd_neg;%calls function FAodd_neg stores fin
pressure, roll torques on Fin A, Odd, negative AOA
T_FBodd_pos = FBodd_pos;%calls function FBodd_pos stores fin
pressure, roll torques on Fin B, Odd, positive AOA
T_FBodd_neg = FBodd_neg;%calls function FBodd_neg stores fin
pressure, roll torques on Fin B, Odd, negative AOA
T_FAEven_pos = FAEven_pos;%calls function FAEven_pos stores fin
pressure, roll torques on Fin A, Even, positive AOA

```

```

T_FAeven_neg = FAeven_neg;%calls function FAeven_neg stores fin
pressure, roll torques on Fin A, Even, negative AOA
T_FBEven_pos = FBeven_pos;%calls function FBeven_pos stores fin
pressure, roll torques on Fin B, Even, positive AOA
T_FBEven_neg = FBeven_neg;%calls function FBeven_neg stores fin
pressure, roll torques on Fin B, Even, negative AOA

%calculations
T_FA_pos= T_FAodd_pos + T_FAeven_pos;%determines total torque on Fin
A, positive AOA
T_FA_neg= T_FAodd_neg + T_FAeven_neg;%determines total torque on Fin
A, negative AOA
T_FB_pos= T_FBodd_pos + T_FBEven_pos;%determines total torque on Fin
B, positive AOA
T_FB_neg= T_FBodd_neg + T_FBEven_neg;%determines total torque on Fin
B, negative AOA

%interpolates J values for total fin pressure from the two sets of J
data
U=[(max (J_pos(1), J_neg(1))):(range (J_pos))/350:(min
(J_pos(min(length(J_pos),length(J_neg))),
J_neg(min(length(J_pos),length(J_neg)))))]';
%creates uniform matrix with approximately 350 J values from the
limits of the two sets of data
T_pos= T_FA_pos + T_FB_pos;%determines total torque on the positive
AOA side
T_neg= T_FA_neg + T_FB_neg;%determines total torque on the negative
AOA side

TU_pos= interp1(J_pos,T_pos,U);%interpolates torque from positive
AOA side onto uniform matrix
TU_neg= interp1(J_neg,T_neg,U);%interpolates torque from negative
AOA side onto uniform matrix
T_pressure= TU_pos + TU_neg;%determines total roll torque

%interpolates torque from aerodynamic balance from the two sets of
data
T_rollUpos= interp1(J_pos,T_roll_pos,U);%interpolates balance torque
data from positive AOA onto uniform J matrix
T_rollUneg= interp1(J_neg,T_roll_neg,U);%interpolates balance torque
data from negative AOA onto uniform J matrix
T_roll=(T_rollUpos + T_rollUneg)/2;%averages the 2 balance torques
at each J value

%error analysis
T_diff= T_roll - T_pressure;%determines difference between balance
and fin pressure torques
RMSdiff= num2str((mean(T_diff.^2))^0.5)%determines RMS difference
between balance and fin pressure torques
format short
mean_Taero_pos=mean(T_roll_pos)%determines mean of balance torque
data, positive AOA

```

```

mean_Taero_neg=mean(T_roll_neg)%determines mean of balance torque
data, negative AOA
mean_Tpressure=mean(T_pressure)%determines mean of pressure torque
data
stdev_Taero_pos=std(T_roll_pos)%determines standard deviation of
balance torque data, positive AOA
stdev_Taero_neg=std(T_roll_neg)%determines standard deviation of
balance torque data, negative AOA
stdev_Tpressure=std(T_pressure)%determines standard deviation of
pressure torque data

%plots positive and negative AOA balance-measured torque vs. J and
saves it
%in the directory with the DAT files
hold on
plot(J_pos,T_roll_pos, 'r', 'LineWidth', 1);
plot(J_neg,T_roll_neg, 'b', 'LineWidth', 1);
grid on
legend('T_b_a_l_a_n_c_e_p_o_s', 'T_b_a_l_a_n_c_e_n_e_g','Location',
'Best');
title(['Run #' run_pos ' and #' run_neg ': Balance Measured
Torques']);
xlabel('Nozzle Dynamic Pressure Ratio, J');
ylabel('Counter Torque [in-lb_f]');
saveas(gcf,[dir '/RUN0' run_pos run_neg 'aero'],'jpg');
hold off
close

%plots both balance-measured torque and fin pressure torque on same
axis
%and saves it in the same directory with DAT file
hold on
plot(U,T_pressure, 'r', 'LineWidth', 2);
plot(U,T_roll, 'b', 'LineWidth', 2);
plot(U,(T_pressure+27.808), 'r--', 'LineWidth', 1);%plots error in
pressure torques
plot(U,(T_pressure-27.808), 'r--', 'LineWidth', 1);%plots error in
pressure torques
plot(U,(T_roll+13.2827), 'b-.', 'LineWidth', 1);%plots error in
balance torques
plot(U,(T_roll-13.2827), 'b-.', 'LineWidth', 1);%plots error in
balance torques
text(0.025, 0.975,['M= ' M '''],'sc');
text(0.025, 0.945,['Q= ' Q '''],'sc');
text(0.025, 0.915,['\alpha = ' AOA_pos '''],'sc');
text(0.025, 0.885,['Roll angle= ' rollangle'],'sc');
text(0.025, 0.855,['RMS difference= ' RMSdiff'],'sc');
grid on
legend('T_f_i_n', 'T_b_a_l_a_n_c_e','Location', 'Best');
title(['Run #' run_pos ' and #' run_neg ': Total Counter Torque']);
xlabel('Nozzle Dynamic Pressure Ratio, J');
ylabel('Counter Torque [in-lb_f]');

```

```

saveas(gcf,[dir '/RUN0' run_pos run_neg 'total'],'jpg');
hold off
close

%plots total roll torque on Fin A, positive AOA versus J and saves
it in the directory with
%the DAT files
plot(J_pos, T_FA_pos, 'LineWidth', 1);
grid on
title(['Run #' run_pos ' M ' M ', AOA=' AOA_pos ' degrees: Total
Counter Torque on Fin A']);
xlabel('Nozzle Dynamic Pressure Ratio, J');
ylabel('Counter Torque [in-lb_f]');
saveas(gcf,[dir '/RUN0' run_pos 'finAtotal'],'jpg');
close

%plots total roll torque on Fin A versus J, negative AOA and saves
it in the directory with
%the DAT files
plot(J_neg, T_FA_neg, 'LineWidth', 1);
grid on
title(['Run #' run_neg ' M ' M ', AOA=' AOA_neg ' degrees: Total
Counter Torque on Fin A']);
xlabel('Nozzle Dynamic Pressure Ratio, J');
ylabel('Counter Torque [in-lb_f]');
saveas(gcf,[dir '/RUN0' run_neg 'finAtotal'],'jpg');
close

%plots total roll torque on Fin B, positive AOA versus J and saves
it in the directory with
%the DAT files
plot(J_pos, T_FB_pos, 'LineWidth', 1);
grid on
title(['Run #' run_pos ' M ' M ', AOA=' AOA_pos ' degrees: Total
Counter Torque on Fin B']);
xlabel('Nozzle Dynamic Pressure Ratio, J');
ylabel('Counter Torque [in-lb_f]');
saveas(gcf,[dir '/RUN0' run_pos 'finBtotal'],'jpg');
close

%plots total roll torque on Fin B, negative AOA versus J and saves
it in the directory with
%the DAT files
plot(J_neg, T_FB_neg, 'LineWidth', 1);
grid on
title(['Run #' run_neg ' M ' M ', AOA=' AOA_neg ' degrees: Total
Counter Torque on Fin B']);
xlabel('Nozzle Dynamic Pressure Ratio, J');
ylabel('Counter Torque [in-lb_f]');
saveas(gcf,[dir '/RUN0' run_neg 'finBtotal'],'jpg');
close

```

```

%plots total roll torque on negative AOA side versus J and saves it
in the directory with
%the DAT files
plot(J_neg, T_neg, 'LineWidth', 1);
grid on
title(['Run #' run_neg ' M ' M ', AOA=' AOA_neg ' degrees: Total
Counter Torque']);
xlabel('Nozzle Dynamic Pressure Ratio, J');
ylabel('Counter Torque [in-lb_f]');
saveas(gcf,[dir '/RUN0' run_neg 'finABtotal'],'jpg');
close

%plots total roll torque on positive AOA side versus J and saves it
in the directory with
%the DAT files
plot(J_pos, T_pos, 'LineWidth', 1);
grid on
title(['Run #' run_pos ' M ' M ', AOA=' AOA_pos ' degrees: Total
Counter Torque']);
xlabel('Nozzle Dynamic Pressure Ratio, J');
ylabel('Counter Torque [in-lb_f]');
saveas(gcf,[dir '/RUN0' run_pos 'finABtotal'],'jpg');
close

```

FAodd_pos.m

```

function T_FAodd_pos = FAodd_pos

%global variable declaration
global dir run_pos J_pos FA_p_pos tap_coords fin_area AOA_pos
AOA_neg M data_pos tare_pressures

%data filter
p = FA_p_pos(:, [1:2:47]);%maintains pressure data for Fin A, odd
pressure taps, positive AOA
x = tap_coords(:,1);%maintains x coordinates for Fin A, odd pressure
taps
y = tap_coords(:,2);%maintains y coordinates for Fin A, odd pressure
taps
z = tap_coords(:,3);%maintains z coordinates for Fin A, odd pressure
taps

%reading appropriate tare pressure from file
if str2num(dir(10:length(dir)))==8 && str2num(dir(5:7))==220
    tare_pressure=tare_pressures(5);
elseif str2num(dir(10:length(dir)))==95 && str2num(dir(5:7))==220
    tare_pressure=tare_pressures(13);
elseif str2num(dir(10:length(dir)))==11 && str2num(dir(5:7))==220
    tare_pressure=tare_pressures(21);

```

```

elseif str2num(dir(10:length(dir)))==8 && str2num(dir(5:7))==440
    tare_pressure=tare_pressures(29);
elseif str2num(dir(10:length(dir)))==95 && str2num(dir(5:7))==440
    tare_pressure=tare_pressures(37);
elseif str2num(dir(10:length(dir)))==11 && str2num(dir(5:7))==440
    tare_pressure=tare_pressures(45);
elseif str2num(dir(10:length(dir)))==8 && str2num(dir(5:7))==880
    tare_pressure=tare_pressures(53);
elseif str2num(dir(10:length(dir)))==95 && str2num(dir(5:7))==880
    tare_pressure=tare_pressures(61);
elseif str2num(dir(10:length(dir)))==11 && str2num(dir(5:7))==880
    tare_pressure=tare_pressures(69);
else disp('Error identifying tare pressure')
end

%function calls
[vertices, area_vectors, r, a_surf] = vert_and_area(x, y, z);%calls
function to determine triangle vertices,
...area vectors, triangle centroids, and area on Fin A, Odd within
tapped region
p_triangle = triangle_pressures(p, vertices);%calls function to
determine mean pressures for each triangle
T_tot = moment(p_triangle, area_vectors, r);%calls function to
determine torque for the entire surface, Fin A:Odd

%calculations
T_tot_x = -1.*T_tot(:,1);

%plots roll torque for Fin A, Odd, positive AOA versus J and saves
it in directory with
%DAT file
plot (J_pos, T_tot_x, 'LineWidth', 1)
grid on
ylabel('Counter Torque [in-lb_f]')
xlabel('Nozzle Dynamic Pressure Ratio, J');
title(['Run #' run_pos ' M ' M ', AOA=' AOA_pos ' degrees: Fin A,
Odd Taps']);
saveas(gcf, [dir '/RUN0' run_pos 'finAodd'],'jpg');
close

%scales torque to approximate the torque over the entire Fin A, Odd
T_FAodd_pos = T_tot_x*(fin_area/a_surf) - tare_pressure;

```

VFI_torques_AOAtare.m

```

clear all
clc

%global variable declaration

```



```

global dir run FA_p_pos FB_p_pos tap_coords fin_area FA_p_neg
FB_p_neg

%conversions
psf2psi = 0.0069444;%convert from psf to psi

%user input
dir = input('Enter directory name (ex. C2_Q440_M8): ','s');%the
directory with DAT files
run = input('Enter the run number (ex. 0426): ','s');%the first DAT
file

%file input
data = load([dir '/RUN0' run '.dat']);%loads entire DAT file
tap_coords = load(['fin_taps.dat']);%loads entire taps file that
contains coordinates for the taps

%calculating area of whole fin
fin_dim = load('fin_shape.dat');%loads coordinates of fin edge
points
fin_x = fin_dim(:,1);%maintains edge x-coordinates
fin_y = fin_dim(:,2);%maintains edge y-coordinates
fin_area = polyarea(fin_x,fin_y);%calculates area of whole fin

%data filter
FA_p_neg = data(8,[87:134]) * psf2psi;%maintains 48 columns of
pressure data for Fin A, negative AOA
FB_p_neg = data(8,[135:182]) * psf2psi;%maintains 48 columns of
pressure data for Fin B, negative AOA
FA_p_pos = data(6,[87:134]) * psf2psi;%maintains 48 columns of
pressure data for Fin A, positive AOA
FB_p_pos = data(6,[135:182]) * psf2psi;%maintains 48 columns of
pressure data for Fin B, positive AOA

%function calls
T_FAodd_neg = FAodd_tare_neg%calls function, stores fin pressure,
roll torques on Fin A, Odd, negative AOA
T_FBodd_neg = FBodd_tare_neg%calls function, stores fin pressure,
roll torques on Fin B, Odd, negative AOA
T_FAeven_neg = FAeven_tare_neg%calls function, stores fin pressure,
roll torques on Fin A, Even, negative AOA
T_FBeven_neg = FBeven_tare_neg%calls function, stores fin pressure,
roll torques on Fin B, Even, negative AOA
T_FAodd_pos = FAodd_tare_pos%calls function, stores fin pressure,
roll torques on Fin A, Odd, positive AOA
T_FBodd_pos = FBodd_tare_pos%calls function, stores fin pressure,
roll torques on Fin B, Odd, positive AOA
T_FAeven_pos = FAeven_tare_pos%calls function, stores fin pressure,
roll torques on Fin A, Even, positive AOA
T_FBeven_pos = FBeven_tare_pos%calls function, stores fin pressure,
roll torques on Fin B, Even, positive AOA

```

```

%calculations
T_FA_pos = T_FAodd_pos + T_FAeven_pos;%determines total torque on Fin
A, positive AOA
T_FA_neg = T_FAodd_neg + T_FAeven_neg;%determines total torque on Fin
A, negative AOA
T_FB_pos = T_FBodd_pos + T_FBeven_pos;%determines total torque on Fin
B, positive AOA
T_FB_neg = T_FBodd_neg + T_FBeven_neg;%determines total torque on Fin
B, negative AOA
T_FAB_pos = T_FA_pos + T_FB_pos;%determines total torque on Fin A,
positive AOA
T_FAB_neg = T_FA_neg + T_FB_neg;%determines total torque on Fin B,
negative AOA

T_sum=T_FAB_pos + T_FAB_neg;%determines sum of positive and negative
AOA torques
T_ratio=T_sum/T_FAB_pos;%determines ratio of summed torque to a
torque on a single side

```

VFI_torques_Otare.m

```

clear all
clc

%global variable declaration
global dir run J FA_p FB_p tap_coords fin_area

%conversions
psf2psi = 0.0069444;%convert from psf to psi

%user input
dir = input('Enter directory name (ex. C2_Q440_M8): ','s');%the
directory with DAT files
run = input('Enter the run number (ex. 0426): ','s');%the first DAT
file

%file input
data = load([dir '/RUN0' run '.dat']);%loads entire DAT file
tap_coords = load(['fin_taps.dat']);%loads entire taps file that
contains coordinates for the taps

%calculating area of whole fin
fin_dim = load('fin_shape.dat');%loads coordinates of fin edge
points
fin_x = fin_dim(:,1);%maintains edge x-coordinates
fin_y = fin_dim(:,2);%maintains edge y-coordinates
fin_area = polyarea(fin_x,fin_y);%calculates area of whole fin

%data filter

```

```

J = data(:,427);%maintains Nozzle Dynamic Pressure Ratio from DAT
file, in column 427
FA_p = data(2,[87:134]) * psf2psi;%maintains 48 columns of pressure
data for Fin A
FB_p = data(2,[135:182]) * psf2psi;%maintains 48 columns of pressure
data for Fin B

%function calls
T_FAodd = FAodd0%calls function FAodd0 stores fin pressure, roll
torques on Fin A, Odd
T_FBodd = FBodd0%calls function FBodd0 stores fin pressure, roll
torques on Fin B, Odd
T_FAeven = FAeven0%calls function FAeven0 stores fin pressure, roll
torques on Fin A, Even
T_FBeven = FBeven0%calls function FBeven0 stores fin pressure, roll
torques on Fin B, Even

```

FAeven0.m

```

function T_FAeven = FAeven

%global variable declaration
global dir run J FA_p tap_coords fin_area

%data filter
p = FA_p(:,[2:2:48]);%maintains pressure data for Fin A, even
pressure taps
x = tap_coords(:,4);%maintains x coordinates for Fin A, even
pressure taps
y = tap_coords(:,5);%maintains y coordinates for Fin A, even
pressure taps
z = tap_coords(:,6);%maintains z coordinates for Fin A, even
pressure taps

%function calls
[vertices, area_vectors, r, a_surf] = vert_and_area(x, y, z);%calls
function to determine triangle vertices,
...area vectors, triangle centroids, and area on Fin A, Even within
tapped region
p_triangle = triangle_pressures(p, vertices);%calls function to
determine mean pressures for each triangle
T_tot = moment(p_triangle, area_vectors, r);%calls function to
determine torque for the entire surface, Fin A:Even
%calculations
T_tot_x = T_tot(:,1);

%scales torque to approximate the torque over the entire Fin A, Even
T_FAeven = T_tot_x*(fin_area/a_surf);

```

VFI_CTcoeff.m

```
%plots counter torque coefficient for a Mach number at various Q
clear all
clc
mkdir('CTplots')

%global variable declaration
global dir J FA_p FB_p tap_coords fin_area
global M data tare_pressures

%conversions
psf2psi = 0.00694444;%convert from psf to psi

%user input
config = input('Enter configuration (ex. 2, 3, 4): ','s');%the
configuration
M = input('Enter Mach number (0.8, 0.95, 1.1): ','s');%Mach number
rollangle = input('Enter roll angle (0, 45): ','s');%roll angle
if str2num (config)~=2 & str2num (config)~=3 & str2num (config)~=4
disp('Error in configuration input'); end
if str2num (M)~=0.8 & str2num (M)~=0.95 & str2num (M)~=1.1
disp('Error in Mach number input'); end
if str2num (rollangle)~=0 &str2num (rollangle)~=45 disp('Error in
roll angle input'); end

%identifying first input file for given configuration, Mach number,
and
...roll angle
if str2num(rollangle) == 0
    if str2num(config)==2 & str2num(M)==0.8
file1='C2_Q220_M8/RUN00376.dat'; end
    if str2num(config)==2 & str2num(M)==0.95
file1='C2_Q220_M95/RUN00386.dat'; end
    if str2num(config)==2 & str2num(M)==1.1
file1='C2_Q220_M11/RUN00396.dat'; end

    if str2num(config)==3 & str2num(M)==0.8
file1='C3_Q220_M8/RUN00511.dat'; end
    if str2num(config)==3 & str2num(M)==0.95
file1='C3_Q220_M95/RUN00521.dat'; end
    if str2num(config)==3 & str2num(M)==1.1
file1='C3_Q220_M11/RUN00531.dat'; end

    if str2num(config)==4 & str2num(M)==0.8
file1='C4_Q220_M8/RUN00595.dat'; end
    if str2num(config)==4 & str2num(M)==0.95
file1='C4_Q220_M95/RUN00605.dat'; end
    if str2num(config)==4 & str2num(M)==1.1
file1='C4_Q220_M11/RUN00615.dat'; end
end
```

```

if str2num(rollangle) ==45
    if str2num(config)==2 & str2num(M)==0.8
file1='C2_Q220_M8/RUN00381.dat'; end
    if str2num(config)==2 & str2num(M)==0.95
file1='C2_Q220_M95/RUN00391.dat'; end
    if str2num(config)==2 & str2num(M)==1.1
file1='C2_Q220_M11/RUN00401.dat'; end

    if str2num(config)==3 & str2num(M)==0.8
file1='C3_Q220_M8/RUN00516.dat'; end
    if str2num(config)==3 & str2num(M)==0.95
file1='C3_Q220_M95/RUN00526.dat'; end
    if str2num(config)==3 & str2num(M)==1.1
file1='C3_Q220_M11/RUN00536.dat'; end

    if str2num(config)==4 & str2num(M)==0.8
file1='C4_Q220_M8/RUN00600.dat'; end
    if str2num(config)==4 & str2num(M)==0.95
file1='C4_Q220_M95/RUN00610.dat'; end
    if str2num(config)==4 & str2num(M)==1.1
file1='C4_Q220_M11/RUN00620.dat'; end
end

if str2num(config)==2
    n=3;
else n=2;
end

%determining counter torque coefficient for varying Q
for o=1:n

    %file input
    data = load([file1]);%loads appropriate DAT file
    tap_coords = load(['fin_taps.dat']);%loads tap coordinates
    if str2num(M)==0.8 dir = file1(1:10); end%identifies directory
    if str2num(M)~=0.8 dir = file1(1:11); end%identifies directory

    %determining minimum and maximum J values allowed for data
    if abs(data(1,7)-220<5) a=find (data(:,427)>4.6, 1); end
    if abs(data(1,7)-440<5) b=find (data(:,427)>2.3, 1); end
    if abs(data(1,7)-880<5) c=find (data(:,427)>1.1, 1); end

    if abs(data(1,7)-220<5) i=find (data(:,427)>73.7, 1); end
    if abs(data(1,7)-440<5) j=find (data(:,427)>36.8, 1); end
    if abs(data(1,7)-880<5) k=find (data(:,427)>18.4, 1); end

    %finds point at which J begins to decrease and discards
subsequent data
    if isempty(i)==1 i(1) =
min(find(diff(data(100:size(data,1),427))<0)); end

```

```

    if isempty(j)==1 j(1) =
min(find(diff(data(100:size(data,1),427))<0)); end
    if isempty(k)==1 k(1) =
min(find(diff(data(100:size(data,1),427))<0)); end

    if (round ((data(1,7)/10)))*10==220%minimum J value for Q220
data
        minsize = a(1);
    elseif (round ((data(1,7)/10)))*10==440%minimum J value for Q440
data
        minsize = b(1);
    else minsize = c(1);%minimum J value for Q880 data
    end

    if (round ((data(1,7)/10)))*10==220%maximum J value for Q220
data
        maxsize = i(1);
    elseif (round ((data(1,7)/10)))*10==440%maximum J value for Q440
data
        maxsize = j(1);
    else maxsize = k(1);%maximum J value for Q880 data
    end

    %calculating area of whole fin
    fin_dim = load('fin_shape.dat');%loads coordinates of fin edge
points
    fin_x = fin_dim(:,1);%maintains edge x-coordinates
    fin_y = fin_dim(:,2);%maintains edge y-coordinates
    fin_area = polyarea(fin_x,fin_y);%calculates area of whole fin

    %data filter
    J(1:size(data,1),o) = data(:,427);%maintains Nozzle Dynamic
Pressure Ratio from DAT file, in column 427
    FA_p = data(:, [87:134]) * psf2psi;%maintains 48 columns of
pressure data for Fin A
    FB_p = data(:, [135:182]) * psf2psi;%maintains 48 columns of
pressure data for Fin B
    T_roll = abs(data(:,29));%maintains Roll Torques from
Aerodynamic Balance

    %run identification
    AOA = int2str (round (data(1,13)));%reads angle of attack from
DAT file
    rollangle= num2str(round(data(1,14)));%reads roll angle from DAT
file
    Qpsi=data(1,12);%reads free stream dynamic pressure from DAT
file

    %loading tare pressures
    tare_pressures = load (['C' config '_0AOA_' rollangle
'rolltare.txt']);

```

```

%function calls
T_FAodd = FAoddCT;%calls function FAodd stores fin pressure,
roll torques on Fin A, Odd
T_FBodd = FBoddCT;%calls function FBodd stores fin pressure,
roll torques on Fin B, Odd
T_FAeven = FAevenCT;%calls function FAeven stores fin pressure,
roll torques on Fin A, Even
T_FBeven = FBevenCT;%calls function FBeven stores fin pressure,
roll torques on Fin B, Even

%calculations
T_pressure = 2*(T_FAodd + T_FAeven + T_FBodd +
T_FBeven);%determines total torque from fin pressures,
...factor of 2 is for the other 2 symmetric fins
CCT_fin(1:size(data,1),o) =
(T_pressure)/(Qpsi*pi/4*(13.3)^3);%calculates coefficient from
pressure data
CCT_roll(1:size(data,1),o) =
(T_roll)/(Qpsi*pi/4*(13.3)^3);%calculates coefficient from balance
data

%error analysis
%Q uncertainties from AEDC
if 215<=(round (data(1,7)/10))*10<=225 && str2num(M)==0.8
Uq=1.2786; end
if 215<=(round (data(1,7)/10))*10<=225 && str2num(M)==0.95
Uq=1.0391; end
if 215<=(round (data(1,7)/10))*10<=225 && str2num(M)==1.1
Uq=0.7639; end
if 435<=(round (data(1,7)/10))*10<=445 && str2num(M)==0.8
Uq=1.3004; end
if 435<=(round (data(1,7)/10))*10<=445 && str2num(M)==0.95
Uq=1.0541; end
if 435<=(round (data(1,7)/10))*10<=445 && str2num(M)==1.1
Uq=0.7708; end
if 875<=(round (data(1,7)/10))*10<=885 && str2num(M)==0.8
Uq=1.4525; end
if 875<=(round (data(1,7)/10))*10<=885 && str2num(M)==0.95
Uq=1.1365; end
if 875<=(round (data(1,7)/10))*10<=885 && str2num(M)==1.1
Uq=0.8102; end

%computing uncertainty in pressure and balance derived counter
torque coefficient
for a=1:length(T_pressure)
U_CCT_fin(a,o)= sqrt((1/(Qpsi*pi/4*13.3^3)*39.3264)^2 +
(T_pressure(a)/(Qpsi^2*pi/4*13.3^3)*Uq/144)^2);
U_CCT_roll(a,o)=sqrt((1/(Qpsi*pi/4*13.3^3)*13.2827)^2 +
(T_pressure(a)/(Qpsi^2*pi/4*13.3^3)*Uq/144)^2);
end

%setting up next run

```

```

if str2num(M)==0.8
    if o==1
        p=num2str(str2num(file1(17:19))+50);
        q=num2str(str2num(file1(5:7))*2);
    elseif o==2
        p=num2str(str2num(file1(17:19))+25);
        q=num2str(str2num(file1(5:7))*2);
    end
    file1(17:19)=p;
    file1(5:7)=q;
end

if str2num(M)==0.95
    if o==1
        p=num2str(str2num(file1(18:20))+30);
        q=num2str(str2num(file1(5:7))*2);
    elseif o==2
        p=num2str(str2num(file1(18:20))+45);
        q=num2str(str2num(file1(5:7))*2);
    end
    file1(18:20)=p;
    file1(5:7)=q;
end

if str2num(M)==1.1
    if o==1
        p=num2str(str2num(file1(18:20))+10);
        q=num2str(str2num(file1(5:7))*2);
    elseif o==2
        p=num2str(str2num(file1(18:20))+65);
        q=num2str(str2num(file1(5:7))*2);
    end
    file1(18:20)=p;
    file1(5:7)=q;
end
end

%plots both coefficients from pressure and balance data and error
bars
hold on
%determining final data point for each set of data
r=find(CCT_roll(:,1)==0,1)-1;
if isempty(r) r=size(CCT_roll,1); end
s=find(CCT_roll(:,2)==0,1)-1;
if isempty(s) s=size(CCT_roll,1); end
if str2num(config)==2
    t=find(CCT_roll(:,3)==0,1)-1;
    if isempty(t) t=size(CCT_roll,1); end
end

if str2num(config)==2
    Jlimit=J(t,3);

```



```

else Jlimit=J(s,2);
end

%increment for error bar
b=1:40:r;
c=1:40:s;
if str2num(config)==2
    d=1:40:t;
end

%plotting coefficients and error bars
plot(J(1:r,1),CCT_fin(1:r,1), 'r', 'LineWidth', 1);%Q220 pressure
plot(J(1:r,1),CCT_roll(1:r,1), 'b', 'LineWidth', 1);%Q220 balance
plot(J(1:s,2),CCT_fin(1:s,2), 'r--', 'LineWidth', 1);%Q440 pressure
plot(J(1:s,2),CCT_roll(1:s,2), 'b--', 'LineWidth', 1);%Q440 balance

if str2num(config)==2%Q880
    plot(J(1:t,3),CCT_fin(1:t,3), 'r:', 'LineWidth', 1);
    plot(J(1:t,3),CCT_roll(1:t,3), 'b:', 'LineWidth', 1);

errorbar(J(d,3),CCT_fin(d,3),U_CCT_fin(d,3),U_CCT_fin(d,3),'r.');
```

```

errorbar(J(d,3),CCT_roll(d,3),U_CCT_roll(d,3),U_CCT_roll(d,3),'b.');
```

```

end

Q220dev=mean(abs(CCT_fin(1:find(abs(J(1:r,1)-Jlimit)<1,1),1)-
CCT_roll(1:find(abs(J(1:r,1)-Jlimit)<1,1),1)));
Q440dev=mean(abs(CCT_fin(1:find(abs(J(1:s,2)-Jlimit)<1,1),2)-
CCT_roll(1:find(abs(J(1:s,2)-Jlimit)<1,1),2)));
if str2num(config)==2 Q880dev=mean(abs(CCT_fin(1:t,3)-
CCT_roll(1:t,3))); end

%Q220 & Q440 error bars
errorbar(J(b,1),CCT_fin(b,1),U_CCT_fin(b,1),U_CCT_fin(b,1),'r.');
```

```

errorbar(J(b,1),CCT_roll(b,1),U_CCT_roll(b,1),U_CCT_roll(b,1),'b.');
```

```

errorbar(J(c,2),CCT_fin(c,2),U_CCT_fin(c,2),U_CCT_fin(c,2),'r.');
```

```

errorbar(J(c,2),CCT_roll(c,2),U_CCT_roll(c,2),U_CCT_roll(c,2),'b.');
```

```

if str2num(config)~=2
    legend(['Q220 Pressure C_C_T'], ['Q220 Balance C_C_T'], ['Q440
Pressure C_C_T'], ['Q440 Balance C_C_T'], 'Location', 'Best')
else legend(['Q220 Pressure C_C_T'], ['Q220 Balance C_C_T'], ['Q440
Pressure C_C_T'], ['Q440 Balance C_C_T'], ['Q880 Pressure C_C_T'],
['Q880 Balance C_C_T'],'Location', 'Best')
end

text(0.025, 0.985,['\alpha = ' AOA ''],'sc');%displays angle of
attack
text(0.025, 0.955,['Roll angle= ' rollangle ''],'sc');%displays roll
angle
grid on

```

```

title(['Counter Torque Coefficient at Mach' M ', Configuration'
config '']);
xlabel('Nozzle Dynamic Pressure Ratio, J');
ylabel('Counter Torque Coefficient (C_C_T)');
saveas(gcf,['CTplots/Mach' M ', Config' config ' ' rollangle
'roll.jpg'],'jpg');
hold off
close

%plotting deviation between pressure and balance measured
coefficients
hold on
plot(J(1:r,1),abs(CCT_fin(1:r,1)-CCT_roll(1:r,1)), 'r', 'LineWidth',
1);
plot(J(1:s,2),abs(CCT_fin(1:s,2)-CCT_roll(1:s,2)), 'b', 'LineWidth',
1);
if str2num(config)==2 plot(J(1:t,3),abs(CCT_fin(1:t,3)-
CCT_roll(1:t,3)), 'k', 'LineWidth', 1); end

if str2num(config)~=2
    legend(['Q220 deviation'], ['Q440 deviation'], 'Location',
'Best');
elseif str2num(config)==2
    legend(['Q220 deviation'], ['Q440 deviation'], ['Q880
deviation'], 'Location', 'Best');
end

text(0.025, 0.985,['\alpha = ' AOA ''],'sc');%displays angle of
attack
text(0.025, 0.955,['Roll angle= ' rollangle ''],'sc');%displays roll
angle
text(0.025, 0.925,['Q220 deviation= ' num2str(Q220dev)
'],'sc');%displays Q220 deviation
text(0.025, 0.895,['Q440 deviation= ' num2str(Q440dev)
'],'sc');%displays Q440 deviation
if str2num(config)==2 text(0.025, 0.865,['Q880 deviation= '
num2str(Q880dev) ''],'sc'); end%displays Q880 deviation

grid on
title(['Deviation Between Pressure and Balance C_C_T at Mach' M ',
Configuration' config '']);
xlabel('Nozzle Dynamic Pressure Ratio, J');
ylabel('Counter Torque Coefficient (C_C_T) Deviation');
saveas(gcf,['CTplots/Mach' M ', Config' config '_' rollangle
'roll_deviation.jpg'],'jpg');
hold off
close

```

FAoddCT.m

```
function T_FAodd = FAoddCT

%global variable declaration
global dir run J FA_p tap_coords fin_area tare_pressures

%data filter
p = FA_p(:, [1:2:47]);%maintains pressure data for Fin B, odd
pressure taps
x = tap_coords(:,1);%maintains x coordinates for Fin A, odd pressure
taps
y = tap_coords(:,2);%maintains y coordinates for Fin A, odd pressure
taps
z = tap_coords(:,3);%maintains z coordinates for Fin A, odd pressure
taps

%reading appropriate tare pressure from file
if str2num(dir(10:length(dir)))==8 && str2num(dir(5:7))==220
    tare_pressure=tare_pressures(1);
elseif str2num(dir(10:length(dir)))==95 && str2num(dir(5:7))==220
    tare_pressure=tare_pressures(5);
elseif str2num(dir(10:length(dir)))==11 && str2num(dir(5:7))==220
    tare_pressure=tare_pressures(9);
elseif str2num(dir(10:length(dir)))==8 && str2num(dir(5:7))==440
    tare_pressure=tare_pressures(13);
elseif str2num(dir(10:length(dir)))==95 && str2num(dir(5:7))==440
    tare_pressure=tare_pressures(17);
elseif str2num(dir(10:length(dir)))==11 && str2num(dir(5:7))==440
    tare_pressure=tare_pressures(21);
elseif str2num(dir(10:length(dir)))==8 && str2num(dir(5:7))==880
    tare_pressure=tare_pressures(25);
elseif str2num(dir(10:length(dir)))==95 && str2num(dir(5:7))==880
    tare_pressure=tare_pressures(29);
elseif str2num(dir(10:length(dir)))==11 && str2num(dir(5:7))==880
    tare_pressure=tare_pressures(34);
else disp('Error identifying tare pressure')
end

%function calls
[vertices, area_vectors, r, a_surf] = vert_and_area(x, y, z);%calls
function to determine triangle vertices,
...area vectors, triangle centroids, and area on Fin A, Odd within
tapped region
p_triangle = triangle_pressures(p, vertices);%calls function to
determine mean pressures for each triangle
T_tot = moment(p_triangle, area_vectors, r);%calls function to
determine torque for the entire surface, Fin A:Odd

%calculations
```

```
T_tot_x = -1.*T_tot(:,1);
```

```
%scales torque to approximate the torque over the entire Fin A, Odd  
T_FAodd = T_tot_x*(fin_area/a_surf)-tare_pressure;
```

VFI_CTcoeff_AOA.m

```
%plots counter-torque coefficient at angle of attack  
clear all  
clc  
format long  
  
%global variable declaration  
global dir run_pos run_neg J_pos J_neg FA_p_pos FA_p_neg FB_p_pos  
FB_p_neg tap_coords fin_area  
global AOA_pos AOA_neg M data_pos data_neg tare_pressures  
  
%conversions  
psf2psi = 0.0069444;%convert from psf to psi  
  
%user input  
config = input('Enter configuration (ex. 2, 3, 4): ','s');%the  
configuration  
M = input('Enter Mach number (0.8, 0.95, 1.1): ','s');%Mach number  
rollangle = input('Enter roll angle (0, 45): ','s');%roll angle  
if str2num(config)~=2 & str2num(config)~=3 & str2num(config)~=4  
disp('Error in configuration input'); end  
if str2num(M)~=0.8 & str2num(M)~=0.95 & str2num(M)~=1.1  
disp('Error in Mach number input'); end  
  
%identifying first input file for given configuration, Mach number,  
and  
...roll angle  
if str2num(rollangle) == 0  
    if str2num(config)==2 & str2num(M)==0.8  
file1='C2_Q220_M8/RUN00375.dat'; end  
    if str2num(config)==2 & str2num(M)==0.95  
file1='C2_Q220_M95/RUN00385.dat'; end  
    if str2num(config)==2 & str2num(M)==1.1  
file1='C2_Q220_M11/RUN00395.dat'; end  
  
    if str2num(config)==3 & str2num(M)==0.8  
file1='C3_Q220_M8/RUN00510.dat'; end  
    if str2num(config)==3 & str2num(M)==0.95  
file1='C3_Q220_M95/RUN00520.dat'; end  
    if str2num(config)==3 & str2num(M)==1.1  
file1='C3_Q220_M11/RUN00530.dat'; end  
  
    if str2num(config)==4 & str2num(M)==0.8  
file1='C4_Q220_M8/RUN00594.dat'; end
```

```

        if str2num(config)==4 & str2num(M)==0.95
file1='C4_Q220_M95/RUN00604.dat'; end
        if str2num(config)==4 & str2num(M)==1.1
file1='C4_Q220_M11/RUN00614.dat'; end
end

if str2num(rollangle) ==45
    if str2num(config)==2 & str2num(M)==0.8
file1='C2_Q220_M8/RUN00382.dat'; end
    if str2num(config)==2 & str2num(M)==0.95
file1='C2_Q220_M95/RUN00392.dat'; end
    if str2num(config)==2 & str2num(M)==1.1
file1='C2_Q220_M11/RUN00402.dat'; end

    if str2num(config)==3 & str2num(M)==0.8
file1='C3_Q220_M8/RUN00517.dat'; end
    if str2num(config)==3 & str2num(M)==0.95
file1='C3_Q220_M95/RUN00527.dat'; end
    if str2num(config)==3 & str2num(M)==1.1
file1='C3_Q220_M11/RUN00537.dat'; end

    if str2num(config)==4 & str2num(M)==0.8
file1='C4_Q220_M8/RUN00601.dat'; end
    if str2num(config)==4 & str2num(M)==0.95
file1='C4_Q220_M95/RUN00611.dat'; end
    if str2num(config)==4 & str2num(M)==1.1
file1='C4_Q220_M11/RUN00621.dat'; end
end

if str2num(config)==2
    n=3;
else n=2;
end

%determining counter torque coefficient for varying Q
for o=1:n
    file2 = file1;
    if str2num(rollangle)==0 file2(end-4)=num2str(str2num(file1(end-
4))+2); end
    if str2num(rollangle)==45 file2(end-
4)=num2str(str2num(file1(end-4))-2); end

    %file input
    data_neg = load([file1]);%loads negative AOA DAT file
    data_pos = load([file2]);%loads positive AOA DAT file
    tap_coords = load(['fin_taps.dat']);%loads tap coordinates
    if str2num(M)==0.8 dir = file1(1:10); end%identifies directory
    if str2num(M)~=0.8 dir = file1(1:11); end%identifies directory

    %constructing range of J values allowed for data
    a=find (4.6<data_pos(:,427)<73.7);%J limits on Q220 data,
positive AOA

```

```

    b=find (2.3<data_pos(:,427)<36.8);%J limits on Q440 data,
positive AOA
    c=find (1.1<data_pos(:,427)<18.4);%J limits on Q880 data,
positive AOA
    d=find (4.6<data_neg(:,427)<73.7);%J limits on Q220 data,
negative AOA
    e=find (2.3<data_neg(:,427)<36.8);%J limits on Q440 data,
negative AOA
    f=find (1.1<data_neg(:,427)<18.4);%J limits on Q880 data,
negative AOA

    %sets range of J values for Q220 data, positive AOA
    if str2num(dir(5:7))==220 && round (data_pos(1,13))==4
        if isempty(find(diff(data_pos(:,427))<0))==0%filtering out
bad data (decreasing J values)
            if
min(find(diff(data_pos(1:min(size(data_neg,1),size(data_pos,1)),427)
)<0))<20)%filters bad data at beginning

s=find(diff(data_pos(1:min(size(data_neg,1),size(data_pos,1)),427))<
0);
            s=1+max(s(s<20));
            a_start=s;
            else a_start=1;
            end

            if
isempty(find(diff(data_pos(100:min(size(data_neg,1),size(data_pos,1)
),427))<0))==0%filters bad data at end
                a_end=
min(find(diff(data_pos(100:min(size(data_neg,1),size(data_pos,1)),42
7))<0)+100);
            else a_end=size(data_pos,1);
            end

            datasize_pos=a(a_start:a_end);
        else
            datasize_pos = a;
        end

    %sets range of J values for Q440 data, positive AOA
    elseif str2num(dir(5:7))==440 && round (data_pos(1,13))==4
        if isempty(find(diff(data_pos(:,427))<0))==0%filtering out
bad data (decreasing J values)
            if
min(find(diff(data_pos(1:min(size(data_neg,1),size(data_pos,1)),427)
)<0))<20)%filters bad data at beginning

s=find(diff(data_pos(1:min(size(data_neg,1),size(data_pos,1)),427))<
0);
            s=1+max(s(s<20));
            b_start=s;

```

```

        else b_start=1;
        end

        if
isempty(find(diff(data_pos(100:min(size(data_neg,1),size(data_pos,1)
),427))<0))==0%filters bad data at end
            b_end=
min(find(diff(data_pos(100:min(size(data_neg,1),size(data_pos,1)),42
7))<0)+100);
            else b_end=size(data_pos,1);
            end

            datasize_pos=b(b_start:b_end);
        else
            datasize_pos = b;
        end

        %sets range of J values for Q880 data, positive AOA
        elseif str2num(dir(5:7))==880 && round (data_pos(1,13))==4
            if isempty(find(diff(data_pos(:,427))<0))==0%filtering out
bad data (decreasing J values)
                if
min(find(diff(data_pos(1:min(size(data_neg,1),size(data_pos,1)),427)
)<0))<20%filters bad data at beginning

s=find(diff(data_pos(1:min(size(data_neg,1),size(data_pos,1)),427))<
0);
                    s=1+max(s(s<20));
                    c_start=s;
                else c_start=1;
                end

                if
isempty(find(diff(data_pos(100:min(size(data_neg,1),size(data_pos,1)
),427))<0))==0%filters bad data at end
                    c_end=
min(find(diff(data_pos(100:min(size(data_neg,1),size(data_pos,1)),42
7))<0)+100);
                    else c_end=size(data_pos,1);
                    end

                    datasize_pos=c(c_start:c_end);
                else
                    datasize_pos = c;
                end

                else disp ('Error in data limits')
                end

                %sets range of J values for Q220 data, negative AOA
                if str2num(dir(5:7))==220 && round (data_neg(1,13))==-4

```

```

        if isempty(find(diff(data_neg(:,427))<0))==0%filtering out
bad data (decreasing J values)
            if
min(find(diff(data_neg(1:min(size(data_neg,1),size(data_pos,1)),427)
)<0))<20%filters bad data at beginning

t=find(diff(data_neg(1:min(size(data_neg,1),size(data_pos,1)),427))<
0);
                t=1+max(t(t<20));
                d_start=t;
            else d_start=1;
            end

            if
isempty(find(diff(data_neg(100:min(size(data_neg,1),size(data_pos,1)
),427))<0))==0%filters bad data at end
                d_end=
min(find(diff(data_neg(100:min(size(data_neg,1),size(data_pos,1)),42
7))<0)+100);
            else d_end=size(data_neg,1);
            end

            datasize_neg=d(d_start:d_end);
        else
            datasize_neg = d;
        end

        %sets range of J values for Q440 data, negative AOA
        elseif str2num(dir(5:7))==440 && round (data_neg(1,13))==-4
            if isempty(find(diff(data_neg(:,427))<0))==0%filtering out
bad data (decreasing J values)
                if
min(find(diff(data_neg(1:min(size(data_neg,1),size(data_pos,1)),427)
)<0))<20%filters bad data at beginning

t=find(diff(data_neg(1:min(size(data_neg,1),size(data_pos,1)),427))<
0);
                    t=1+max(t(t<20));
                    e_start=t;
                else e_start=1;
                end

                if
isempty(find(diff(data_neg(100:min(size(data_neg,1),size(data_pos,1)
),427))<0))==0%filters bad data at end
                    e_end=
min(find(diff(data_neg(100:min(size(data_neg,1),size(data_pos,1)),42
7))<0)+100);
                else e_end=size(data_neg,1);
                end

                datasize_neg=e(e_start:e_end);

```



```

else
    datasize_neg = e;
end

%sets range of J values for Q880 data, negative AOA
elseif str2num(dir(5:7))==880 && round (data_neg(1,13))==-4
    if isempty(find(diff(data_neg(:,427))<0))==0%filtering out
bad data (decreasing J values)
        if
min(find(diff(data_neg(1:min(size(data_neg,1),size(data_pos,1)),427)
)<0))<20%filters bad data at beginning
t=find(diff(data_neg(1:min(size(data_neg,1),size(data_pos,1)),427))<
0);
            t=1+max(t(t<20));
            f_start=t;
        else f_start=1;
        end

        if
isempty(find(diff(data_neg(100:min(size(data_neg,1),size(data_pos,1)
),427))<0))==0%filters bad data at end
            f_end=
min(find(diff(data_neg(100:min(size(data_neg,1),size(data_pos,1)),42
7))<0)+100);
        else f_end=size(data_neg,1);
        end

            datasize_neg=f(f_start:f_end);
        else
            datasize_neg = f;
        end

else disp ('Error in data limits')
end

%calculating area of whole fin
fin_dim = load('fin_shape.dat');%loads coordinates of fin edge
points
fin_x = fin_dim(:,1);%maintains edge x-coordinates
fin_y = fin_dim(:,2);%maintains edge y-coordinates
fin_area = polyarea(fin_x,fin_y);%calculates area of whole fin

%data filter
J_pos = data_pos(datasize_pos,427);%maintains Nozzle Dynamic
Pressure Ratio from positive AOA DAT file, in column 427
J_neg = data_neg(datasize_neg,427);%maintains Nozzle Dynamic
Pressure Ratio from negative AOA DAT file, in column 427
FA_p_pos = data_pos(datasize_pos,[87:134]) * psf2psi;%maintains
48 columns of pressure data for positive AOA Fin A
FA_p_neg = data_neg(datasize_neg,[87:134]) * psf2psi;%maintains
48 columns of pressure data for negative AOA Fin A

```

```

    FB_p_pos = data_pos(datasize_pos,[135:182]) * psf2psi;%maintains
48 columns of pressure data for positive AOA Fin B
    FB_p_neg = data_neg(datasize_neg,[135:182]) * psf2psi;%maintains
48 columns of pressure data for negative AOA Fin B
    T_roll_pos = abs(data_pos(datasize_pos,29));%maintains Roll
Torques from positive AOA Aerodynamic Balance
    T_roll_neg = abs(data_neg(datasize_neg,29));%maintains Roll
Torques from negative AOA Aerodynamic Balance
    Qpsi_pos=data_pos(datasize_pos,12);
    Qpsi_neg=data_neg(datasize_neg,12);

%run identification
    AOA = int2str (round (data_pos(1,13)));%reads angle of attack
from DAT file
    M= num2str((round (data_pos(1,4)*100))/100);%reads Mach number
from DAT file
    Q= dir(5:7);%reads Q from DAT file

%loading tare pressures
    tare_pressures = load ([dir(1:2) 'AOA' rollangle
'rolltare.txt']);

%function calls
    T_FAodd_pos = FAodd_posCT;%calls function FAodd_pos stores fin
pressure, roll torques on Fin A, Odd, positive AOA
    T_FAodd_neg = FAodd_negCT;%calls function FAodd_neg stores fin
pressure, roll torques on Fin A, Odd, negative AOA
    T_FBodd_pos = FBodd_posCT;%calls function FBodd_pos stores fin
pressure, roll torques on Fin B, Odd, positive AOA
    T_FBodd_neg = FBodd_negCT;%calls function FBodd_neg stores fin
pressure, roll torques on Fin B, Odd, negative AOA
    T_FAEven_pos = FAEven_posCT;%calls function FAEven_pos stores
fin pressure, roll torques on Fin A, Even, positive AOA
    T_FAEven_neg = FAEven_negCT;%calls function FAEven_neg stores
fin pressure, roll torques on Fin A, Even, negative AOA
    T_FBEven_pos = FBEven_posCT;%calls function FBEven_pos stores
fin pressure, roll torques on Fin B, Even, positive AOA
    T_FBEven_neg = FBEven_negCT;%calls function FBEven_neg stores
fin pressure, roll torques on Fin B, Even, negative AOA

%calculations
    T_FA_pos= T_FAodd_pos + T_FAEven_pos;%determines total torque on
Fin A, positive AOA
    T_FA_neg= T_FAodd_neg + T_FAEven_neg;%determines total torque on
Fin A, negative AOA
    T_FB_pos= T_FBodd_pos + T_FBEven_pos;%determines total torque on
Fin B, positive AOA
    T_FB_neg= T_FBodd_neg + T_FBEven_neg;%determines total torque on
Fin B, negative AOA

%interpolates J values for total fin pressure from the two sets
of J data

```

```

    U_1=[(max (J_pos(1), J_neg(1))):(range (J_pos))/350:(min
(J_pos(min(length(J_pos),length(J_neg))),
J_neg(min(length(J_pos),length(J_neg)))))]';
    %creates uniform vector with approximately 350 J values from the
limits of the two sets of data
    U(1:length(U_1),o) = U_1;%assigns vector a column in array U
    T_pos= T_FA_pos + T_FB_pos;%determines total torque on the
positive AOA side
    T_neg= T_FA_neg + T_FB_neg;%determines total torque on the
negative AOA side
    TU_pos= interp1(J_pos,T_pos,U(:,o));%interpolates torque from
positive AOA side onto uniform array
    TU_neg= interp1(J_neg,T_neg,U(:,o));%interpolates torque from
negative AOA side onto uniform array

    Qpsi_posU= interp1(J_pos,Qpsi_pos,U(:,o));%interpolates Q from
positive AOA side onto uniform array
    Qpsi_negU= interp1(J_neg,Qpsi_neg,U(:,o));%interpolates Q from
negative AOA side onto uniform array

    %interpolates torque from aerodynamic balance from the two sets
of data
    T_rollUpos= interp1(J_pos,T_roll_pos,U(:,o));%interpolates
balance torque data from positive AOA onto uniform J matrix
    T_rollUneg= interp1(J_neg,T_roll_neg,U(:,o));%interpolates
balance torque data from negative AOA onto uniform J matrix

    CCT_fin(1:size(U,1),o) =
(TU_pos)./(Qpsi_posU)/(pi/4*(13.3)^3)+(TU_neg)./(Qpsi_negU)/(pi/4*(1
3.3)^3);%calculates coefficient from pressure data
    CCT_roll(1:size(U,1),o) =
((T_rollUpos)./(Qpsi_posU)/(pi/4*(13.3)^3)+(T_rollUneg)./(Qpsi_negU)
/(pi/4*(13.3)^3))/2;%calculates coefficient from balance data

    %error analysis
    %Q uncertainties from AEDC
    if 215<=(round (data_pos(1,7)/10))*10<=225 && str2num(M)==0.8
Uq=1.2786; end
    if 215<=(round (data_pos(1,7)/10))*10<=225 && str2num(M)==0.95
Uq=1.0391; end
    if 215<=(round (data_pos(1,7)/10))*10<=225 && str2num(M)==1.1
Uq=0.7639; end
    if 435<=(round (data_pos(1,7)/10))*10<=445 && str2num(M)==0.8
Uq=1.3004; end
    if 435<=(round (data_pos(1,7)/10))*10<=445 && str2num(M)==0.95
Uq=1.0541; end
    if 435<=(round (data_pos(1,7)/10))*10<=445 && str2num(M)==1.1
Uq=0.7708; end
    if 875<=(round (data_pos(1,7)/10))*10<=885 && str2num(M)==0.8
Uq=1.4525; end
    if 875<=(round (data_pos(1,7)/10))*10<=885 && str2num(M)==0.95
Uq=1.1365; end

```

```

    if 875<=(round (data_pos(1,7)/10))*10<=885 && str2num(M)==1.1
    Uq=0.8102; end

    %computing uncertainty in pressure and balance derived counter
    torque coefficient
    for a=1:length(CCT_fin)
        U_CCT_fin(a,o)=
    sqrt((1/((Qpsi_posU(a)+Qpsi_negU(a))/2*pi/4*13.3^3)*27.80799)^2 +
    (TU_pos(a)/(Qpsi_posU(a)^2*pi/4*13.3^3)*Uq/144+TU_neg(a)/(Qpsi_negU(
    a)^2*pi/4*13.3^3)*Uq/144)^2);

    U_CCT_roll(a,o)=sqrt((1/((Qpsi_posU(a)+Qpsi_negU(a))/2*pi/4*13.3^3)*
    13.2827)^2 +
    (T_rollUpos(a)/(Qpsi_posU(a)^2*pi/4*13.3^3)*Uq/144+T_rollUneg(a)/(Qp
    si_negU(a)^2*pi/4*13.3^3)*Uq/144)^2);
    end

    %setting up next run
    if str2num(M)==0.8
        if o==1
            p=num2str(str2num(file1(17:19))+50);
            q=num2str(str2num(file1(5:7))*2);
        elseif o==2
            p=num2str(str2num(file1(17:19))+25);
            q=num2str(str2num(file1(5:7))*2);
        end
        file1(17:19)=p;
        file1(5:7)=q;
    end

    if str2num(M)==0.95
        if o==1
            p=num2str(str2num(file1(18:20))+30);
            q=num2str(str2num(file1(5:7))*2);
        elseif o==2
            p=num2str(str2num(file1(18:20))+45);
            q=num2str(str2num(file1(5:7))*2);
        end
        file1(18:20)=p;
        file1(5:7)=q;
    end

    if str2num(M)==1.1
        if o==1
            p=num2str(str2num(file1(18:20))+10);
            q=num2str(str2num(file1(5:7))*2);
        elseif o==2
            p=num2str(str2num(file1(18:20))+65);
            q=num2str(str2num(file1(5:7))*2);
        end
        file1(18:20)=p;
        file1(5:7)=q;
    end

```

```

    end
end

%plots balance and pressure derived counter torque coefficient vs. J
and saves it
...in the directory CTplots
hold on
%determining final data point for each set of data
r=find(CCT_roll(:,1)==0,1)-1;
if isempty(r) r=size(CCT_roll,1); end
s=find(CCT_roll(:,2)==0,1)-1;
if isempty(s) s=size(CCT_roll,1); end
if str2num(config)==2
    t=find(CCT_roll(:,3)==0,1)-1;
    if isempty(t) t=size(CCT_roll,1); end
end

%increment for error bar
b=1:40:r;
c=1:40:s;
if str2num(config)==2
    d=1:40:t;
end
%plotting coefficients and error bars
plot(U(1:r,1),CCT_fin(1:r,1), 'r', 'LineWidth', 1);%Q220 pressure
plot(U(1:r,1),CCT_roll(1:r,1), 'b', 'LineWidth', 1);%Q220 balance
plot(U(1:s,2),CCT_fin(1:s,2), 'r--', 'LineWidth', 1);%Q440 pressure
plot(U(1:s,2),CCT_roll(1:s,2), 'b--', 'LineWidth', 1);%Q440 balance

if str2num(config)==2%Q880
plot(U(1:t,3),CCT_fin(1:t,3), 'r:', 'LineWidth', 1);
plot(U(1:t,3),CCT_roll(1:t,3), 'b:', 'LineWidth', 1);
errorbar(U(d,3),CCT_fin(d,3),U_CCT_fin(d,3),U_CCT_fin(d,3),'r. ');
errorbar(U(d,3),CCT_roll(d,3),U_CCT_roll(d,3),U_CCT_roll(d,3),'b. ');
end

%Q220 & Q440 error bars
errorbar(U(b,1),CCT_fin(b,1),U_CCT_fin(b,1),U_CCT_fin(b,1),'r. ');
errorbar(U(b,1),CCT_roll(b,1),U_CCT_roll(b,1),U_CCT_roll(b,1),'b. ');
errorbar(U(c,2),CCT_fin(c,2),U_CCT_fin(c,2),U_CCT_fin(c,2),'r. ');
errorbar(U(c,2),CCT_roll(c,2),U_CCT_roll(c,2),U_CCT_roll(c,2),'b. ');

if str2num(config)~=2
    legend(['Q220 Pressure C_C_T'], ['Q220 Balance C_C_T'], ['Q440
Pressure C_C_T'], ['Q440 Balance C_C_T'], 'Location', 'Best')
else legend(['Q220 Pressure C_C_T'], ['Q220 Balance C_C_T'], ['Q440
Pressure C_C_T'], ['Q440 Balance C_C_T'], ['Q880 Pressure C_C_T'],
['Q880 Balance C_C_T'], 'Location', 'Best')
end

text(0.025, 0.985,['\alpha = ' AOA ''],'sc');%displays angle of
attack

```

```

text(0.025, 0.955,['Roll angle= ' rollangle ''],'sc');%displays roll
angle
grid on
title(['Counter Torque Coefficient at Mach' M ', Configuration'
config '']);
xlabel('Nozzle Dynamic Pressure Ratio, J');
ylabel('Counter Torque Coefficient (C_C_T)');
saveas(gcf,['CTplots/Mach' M ', Config' config ' AOA ' rollangle
'roll.jpg'],'jpg');
hold off
close

```

FAodd_posCT.m

```

function T_FAodd_pos = FAodd_pos

%global variable declaration
global dir run_pos J_pos FA_p_pos tap_coords fin_area AOA_pos
AOA_neg M data_pos tare_pressures

%data filter
p = FA_p_pos(:, [1:2:47]);%maintains pressure data for Fin A, odd
pressure taps, positive AOA
x = tap_coords(:,1);%maintains x coordinates for Fin A, odd pressure
taps
y = tap_coords(:,2);%maintains y coordinates for Fin A, odd pressure
taps
z = tap_coords(:,3);%maintains z coordinates for Fin A, odd pressure
taps

%reading appropriate tare pressure from file
if str2num(dir(10:length(dir)))==8 && str2num(dir(5:7))==220
    tare_pressure=tare_pressures(5);
elseif str2num(dir(10:length(dir)))==95 && str2num(dir(5:7))==220
    tare_pressure=tare_pressures(13);
elseif str2num(dir(10:length(dir)))==11 && str2num(dir(5:7))==220
    tare_pressure=tare_pressures(21);
elseif str2num(dir(10:length(dir)))==8 && str2num(dir(5:7))==440
    tare_pressure=tare_pressures(29);
elseif str2num(dir(10:length(dir)))==95 && str2num(dir(5:7))==440
    tare_pressure=tare_pressures(37);
elseif str2num(dir(10:length(dir)))==11 && str2num(dir(5:7))==440
    tare_pressure=tare_pressures(45);
elseif str2num(dir(10:length(dir)))==8 && str2num(dir(5:7))==880
    tare_pressure=tare_pressures(53);
elseif str2num(dir(10:length(dir)))==95 && str2num(dir(5:7))==880
    tare_pressure=tare_pressures(61);
elseif str2num(dir(10:length(dir)))==11 && str2num(dir(5:7))==880
    tare_pressure=tare_pressures(69);

```

```

else disp('Error identifying tare pressure')
end

%function calls
[vertices, area_vectors, r, a_surf] = vert_and_area(x, y, z);%calls
function to determine triangle vertices,
...area vectors, triangle centroids, and area on Fin A, Odd within
tapped region
p_triangle = triangle_pressures(p, vertices);%calls function to
determine mean pressures for each triangle
T_tot = moment(p_triangle, area_vectors, r);%calls function to
determine torque for the entire surface, Fin A:Odd

%calculations
T_tot_x = -1.*T_tot(:,1);

%plots roll torque for Fin A, Odd, positive AOA versus J and saves
it in directory with
%DAT file
plot (J_pos, T_tot_x, 'LineWidth', 1)
grid on
ylabel('Counter Torque [in-lb_f]')
xlabel('Nozzle Dynamic Pressure Ratio, J');
title(['Run #' run_pos ' M ' M ', AOA=' AOA_pos ' degrees: Fin A,
Odd Taps']);
saveas(gcf,[dir '/RUN0' run_pos 'finAodd'],'jpg');
close

%scales torque to approximate the torque over the entire Fin A, Odd
T_FAodd_pos = T_tot_x*(fin_area/a_surf) - tare_pressure;

```

finpress_plot.m

```

%finpress_plot.m
%plots fin pressure coefficient distribution

clear all
clc
%reading input file
dir = input('Enter directory name (ex. C2_Q440_M8): ','s');%the
directory with DAT file
run = input('Enter the run number (ex. 0426): ','s');%the DAT file
BIG=load([dir '/RUN0' run '.dat']);%loads entire DAT file
mkdir(dir, 'finplots')

%tap coordinates

```

```

Xodd=[0.06 1.28 1.18 1.24 2.73 2.84 3.08 3.45 3.94 5.14 5.56
6.11 6.78 7.58 7.54 8.28 9.17 10.2 11.39 9.96 11.07 13.11 13.04
15.19];
Yodd=[7.43 6.27 7.4 8.53 6.21 7.35 8.5 9.65 10.81 6.1 7.27
8.45 9.63 10.84 5.99 7.19 8.39 9.62 10.88 5.89 7.1 9.62
8.33 10.33];
Xeven=[0.34 1.52 1.46 1.55 2.97 3.11 3.39 3.78 4.35.38 5.83
6.41 7.12 7.95 7.78 8.56 9.48 10.55 11.79 10.2 11.36 13.49
13.39 15.61];
Yeven=[7.44 6.27 7.4 8.53 6.2 7.35 8.5 9.65 10.81 6.1 7.27 8.45 9.64
10.85 5.99 7.19 8.4 9.63 10.89 5.89 7.1 9.62 8.33 10.31];

%reading tare values
tares=load([dir '/RUN00' num2str(str2num(run)-2) '.dat']);
tare_odd=tares(2,135:2:181);
tare_even=tares(2,136:2:182);

%plotting fin B
Aodd=BIG(:,135:2:181);
Aeven=BIG(:,136:2:182);

%setting J values to be plotted
J_1=8;
J_2=16;
J_3=36;

if BIG(end,427) < J_2
    k=1;
elseif BIG(end,427) < J_3
    k=2;
else k=3;
end
for j=1:k
    if j==1 J=J_1; end
    if j==2 J=J_2; end
    if j==3 J=J_3; end

    %Q uncertainties from AEDC
    if 215<=(round (BIG(1,7)/10))*10<=225 && str2num(dir(10:end))==8
Uq=1.2786; end
    if 215<=(round (BIG(1,7)/10))*10<=225 &&
str2num(dir(10:end))==95 Uq=1.0391; end
    if 215<=(round (BIG(1,7)/10))*10<=225 &&
str2num(dir(10:end))==11 Uq=0.7639; end
    if 435<=(round (BIG(1,7)/10))*10<=445 && str2num(dir(10:end))==8
Uq=1.3004; end
    if 435<=(round (BIG(1,7)/10))*10<=445 &&
str2num(dir(10:end))==95 Uq=1.0541; end
    if 435<=(round (BIG(1,7)/10))*10<=445 &&
str2num(dir(10:end))==11 Uq=0.7708; end
    if 875<=(round (BIG(1,7)/10))*10<=885 && str2num(dir(10:end))==8
Uq=1.4525; end

```



```

        if 875<=(round (BIG(1,7)/10))*10<=885 &&
str2num(dir(10:end))==95 Uq=1.1365; end
        if 875<=(round (BIG(1,7)/10))*10<=885 &&
str2num(dir(10:end))==11 Uq=0.8102; end

        %plotting fin at prescribed J
        i=find (floor(BIG(:,427))==J);
        Zodd=(Aodd(i(1),:)-tare_odd)/BIG(i(1),7);%normalized odd
pressures
        Zeven=(Aeven(i(1),:)-tare_even)/BIG(i(1),7);%normalized even
pressures
        [xi,yi] = meshgrid(0:.1:16,5:.1:11);%generates uniform grid over
entire tapped surface
        zi_odd=griddata(Xodd,Yodd,Zodd,xi,yi,'linear');%interpolates X-Y
tap coordinates and pressures onto uniform grid
        zi_even=griddata(Xeven,Yeven,Zeven,xi,yi,'linear');%interpolates
X-Y tap coordinates and pressures onto uniform grid
        deltapressure=(zi_odd-zi_even);

        for p=1:size(deltapressure,1)%computes uncertainty for each
point in tapped region
            for q=1:size(deltapressure,2)

uncertainty(p,q)=sqrt((1/BIG(i(1),12)*0.0375)^2+(deltapressure(p,q)/
BIG(i(1),12)^2*Uq/144)^2);
                end
            end
            Umax=max(max(uncertainty,[],2));%maximum uncertainty in tapped
region

            peakmax=min(min(deltapressure,[],2));%determines maximum
pressure coefficient
            peakmin=(1-1/exp(1))*min(min(deltapressure,[],2));%identifies
region with > 67% of maximum pressure coefficient
            n=find(deltapressure<peakmin);
            xin=xi(n);%x coordinates of peak region
            yin=yi(n);%y coordinates of peak region
            tri=delaunay(xin,yin);%triangulates peak region
            tri_xcoord= xin(tri);%x coordinates of triangles
            tri_ycoord= yin(tri);%y coordinates of triangles
            area= tri2Darea(tri_xcoord, tri_ycoord);%determines area of
triangles
            ik=area/(sum(area)/length(area));
            b = find(ik> 1.35 | ik<0.35);%finds triangles with area > or <
35% of mean area
            area(b)=[];%deletes triangles
            area_peaks=area;

            h=find (isnan(deltapressure)==0);%determines total tapped area
            xik=xi(h);%tapped area x coordinates
            yik=yi(h);%tapped area y coordinates
            tri_tot=delaunay(xik,yik);%triangulates tapped area

```

```

tri_xcoord_tot= xik(tri_tot);%x coordinates of triangles
tri_ycoord_tot= yik(tri_tot);%y coordinates of triangles
area= tri2Darea(tri_xcoord_tot, tri_ycoord_tot);%determines area
of triangles
ik=area/(sum(area)/length(area));
b = find(ik> 1.35 | ik<0.35);%finds triangles > or < 35% of mean
area(b)=[];%deletes triangles

percent_peak=sum(area_peaks)/sum(area)*100;%percent peak region
peak_mean=mean(deltapressure(n));%mean pressure coefficient in
peak region

%generating slices A through E
x_1=[0.34:0.001:11.8];
y_1=7.44*ones(1,length(x_1));
z_1(1,:,j)=griddata(xi,yi,deltapressure,x_1,y_1,'linear');
x_2=[1.52:0.001:4.3];
y_2=(4.54/2.78)*x_2+3.7877;
z_2(1,:,j)=griddata(xi,yi,deltapressure,x_2,y_2,'linear');
x_3=(0.5987:0.001:4.8);
y_3=(3.334/3.914)*x_3+6.7112928;
z_3(:, :, j)=griddata(xi,yi,deltapressure,x_3,y_3,'linear');
x_4=(0.8281:0.001:5.3);
y_4=(3.334/3.914)*x_4+6.2953858;
z_4(:, :, j)=griddata(xi,yi,deltapressure,x_4,y_4,'linear');
x_5=(1.1061:0.001:5.8);
y_5=(3.334/3.914)*x_5+5.8594788;
z_5(:, :, j)=griddata(xi,yi,deltapressure,x_5,y_5,'linear');

%shaded view of tapped region with slices and taps visible
hold on
plot(x_1,y_1,'linewidth',2)
plot(x_2,y_2,'k','linewidth',2)
plot(x_3,y_3,'k--','linewidth',2)
plot(x_4,y_4,'m','linewidth',2)
plot(x_5,y_5,'r','linewidth',2)
plot(Xodd,Yodd,'xr')
plot(Xeven,Yeven,'o')
text(0.025, 0.985,['uncertainty =' num2str(Umax) ''],'sc');
legend('Slice A','Slice B','Slice C','Slice D','Slice E','odd
taps','even taps','location','SouthEast')

ylabel('Y [in]')
pcolor(xi,yi,deltapressure)%creates shaded pressure plot
axis equal
axis off

colorbar
shading interp
title(['Run ' run ': DELTA JET ON/OFF FIN PRESSURE COEFFICIENT,
FIN B, AOA=' num2str(round (BIG(1,13))) ', Roll=' num2str(round

```

```

(BIG(1,14)) ', Qinf=' num2str(round (BIG(1,7))) ', Mach='
num2str(round((BIG(1,4))*100)/100) ', J=' num2str(J) ');
    saveas(gcf,[dir '/finplots' '/RUN0' run ' M '
num2str(round((BIG(1,4))*100)/100) ', J ' num2str(J)
'FinBplot.jpg'],'jpg');
    close
end

%plotting pressure vs. distance along slice A
hold on
plot(sqrt((x_1-0.34).^2+(y_1-7.44).^2),z_1(:, :,1),'b')
if k~=1 plot(sqrt((x_1-0.34).^2+(y_1-7.44).^2),z_1(:, :,2),'k');
end
if k==3 plot(sqrt((x_1-0.34).^2+(y_1-7.44).^2),z_1(:, :,3),'m');
end

if k==3
    legend(['Fin B: J=' num2str(J_1) ''], ['Fin B: J=' num2str(J_2)
''], ['Fin B: J=' num2str(J_3) ''], 'Location', 'Best')
elseif k==2 legend(['Fin B, J=' num2str(J_1) ''], ['Fin B, J='
num2str(J_2) ''], 'Location', 'Best')
else legend(['Fin B: J=' num2str(J_1) ''],'Location','Best')
end
xlabel('Distance along slice (west to east) [in]')
ylabel('Delta Pressure Coefficient')
hold off
title(['Run ' run ': Slice B, AOA=' num2str(round (BIG(1,13))) ',
Roll=' num2str(round (BIG(1,14))) ', Mach='
num2str(round((BIG(1,4))*100)/100) '']);
    saveas(gcf,[dir '/finplots' '/RUN0' run ' M '
num2str(round((BIG(1,4))*100)/100) ', sliceAplot'],'jpg');
    close

%plotting pressure vs. distance along slice B
hold on
plot(sqrt((x_2-1.52).^2+(y_2-6.27).^2),z_2(:, :,1),'b')
if k~=1 plot(sqrt((x_2-1.52).^2+(y_2-6.27).^2),z_2(:, :,2),'k');
end
if k==3 plot(sqrt((x_2-1.52).^2+(y_2-6.27).^2),z_2(:, :,3),'m');
end

if k==3
    legend(['Fin B: J=' num2str(J_1) ''], ['Fin B: J=' num2str(J_2)
''], ['Fin B: J=' num2str(J_3) ''], 'Location', 'Best')
elseif k==2 legend(['Fin B, J=' num2str(J_1) ''], ['Fin B, J='
num2str(J_2) ''], 'Location', 'Best')
else legend(['Fin B: J=' num2str(J_1) ''],'Location','Best')
end
xlabel('Distance along slice (west to east) [in]')
ylabel('Delta Pressure Coefficient')
hold off

```

```

    title (['Run ' run ': Slice B, AOA=' num2str(round (BIG(1,13))) ',
Roll=' num2str(round (BIG(1,14))) ', Mach='
num2str(round((BIG(1,4))*100)/100) ']);
    saveas(gcf,[dir '/finplots' '/RUN0' run ' M '
num2str(round((BIG(1,4))*100)/100) ', sliceBplot'],'jpg');
    close

%plotting pressure vs. distance along slice C
hold on
plot(sqrt((x_3-0.5987).^2+(y_3-7.221274).^2),z_3(:, :,1),'b')
if k~=1 plot(sqrt((x_3-0.5987).^2+(y_3-
7.221274).^2),z_3(:, :,2),'k'); end
if k==3 plot(sqrt((x_3-0.5987).^2+(y_3-
7.221274).^2),z_3(:, :,3),'m'); end

if k==3
    legend(['Fin B: J=' num2str(J_1) ''], ['Fin B: J=' num2str(J_2)
''], ['Fin B: J=' num2str(J_3) ''], 'Location', 'Best')
elseif k==2 legend (['Fin B, J=' num2str(J_1) ''], ['Fin B, J='
num2str(J_2) ''], 'Location', 'Best')
else legend(['Fin B: J=' num2str(J_1) ''],'Location','Best')
end
xlabel('Distance along slice (west to east) [in]')
ylabel('Delta Pressure Coefficient')
hold off
    title (['Run ' run ': Slice C, AOA=' num2str(round (BIG(1,13))) ',
Roll=' num2str(round (BIG(1,14))) ', Mach='
num2str(round((BIG(1,4))*100)/100) ']);
    saveas(gcf,[dir '/finplots' '/RUN0' run ' M '
num2str(round((BIG(1,4))*100)/100) ', sliceCplot'],'jpg');
    close

%plotting pressure vs. distance along slice D
hold on
plot(sqrt((x_4-0.8281).^2+(y_4-7.000773).^2),z_4(:, :,1),'b')
if k~=1 plot(sqrt((x_4-0.8281).^2+(y_4-
7.000773).^2),z_4(:, :,2),'k'); end
if k==3 plot(sqrt((x_4-0.8281).^2+(y_4-
7.000773).^2),z_4(:, :,3),'m'); end

if k==3
    legend(['Fin B: J=' num2str(J_1) ''], ['Fin B: J=' num2str(J_2)
''], ['Fin B: J=' num2str(J_3) ''], 'Location', 'Best')
elseif k==2 legend (['Fin B, J=' num2str(J_1) ''], ['Fin B, J='
num2str(J_2) ''], 'Location', 'Best')
else legend(['Fin B: J=' num2str(J_1) ''],'Location','Best')
end
xlabel('Distance along slice (west to east) [in]')
ylabel('Delta Pressure Coefficient')
hold off

```

```

    title (['Run ' run ': Slice D, AOA=' num2str(round (BIG(1,13))) ',
Roll=' num2str(round (BIG(1,14))) ', Mach='
num2str(round((BIG(1,4))*100)/100) ']);
    saveas(gcf,[dir '/finplots' '/RUN0' run ' M '
num2str(round((BIG(1,4))*100)/100) ', sliceDplot'],'jpg');
    close

    %plotting pressure vs. distance along slice E
    hold on
    plot(sqrt((x_5-1.1061).^2+(y_5-6.80167).^2),z_5(:, :,1),'b')
    if k~=1 plot(sqrt((x_5-1.1061).^2+(y_5-
6.80167).^2),z_5(:, :,2),'k'); end
    if k==3 plot(sqrt((x_5-1.1061).^2+(y_5-
6.80167).^2),z_5(:, :,3),'m'); end

    if k==3
        legend(['Fin B: J=' num2str(J_1) ''], ['Fin B: J=' num2str(J_2)
''], ['Fin B: J=' num2str(J_3) ''], 'Location', 'Best')
    elseif k==2 legend (['Fin B, J=' num2str(J_1) ''], ['Fin B, J='
num2str(J_2) ''], 'Location', 'Best')
    else legend(['Fin B: J=' num2str(J_1) ''],'Location','Best')
    end
    xlabel('Distance along slice (west to east) [in]')
    ylabel('Delta Pressure Coefficient')
    hold off
    title (['Run ' run ': Slice E, AOA=' num2str(round (BIG(1,13))) ',
Roll=' num2str(round (BIG(1,14))) ', Mach='
num2str(round((BIG(1,4))*100)/100) ']);
    saveas(gcf,[dir '/finplots' '/RUN0' run ' M '
num2str(round((BIG(1,4))*100)/100) ', sliceEplot'],'jpg');
    close

```

VFI_torquesuncertainty.m

```

clear all
clc

%global variable declaration
global dir run J FA_p tap_coords fin_area pressure_precision

tap_coords = load(['fin_taps.dat']);%loads entire taps file that
contains coordinates for the taps

%calculating area of whole fin
fin_dim = load('fin_shape.dat');%loads coordinates of fin edge
points
fin_x = fin_dim(:,1);%maintains edge x-coordinates
fin_y = fin_dim(:,2);%maintains edge y-coordinates
fin_area = polyarea(fin_x,fin_y);%calculates area of whole fin

```

```

%data filter
for i=1:48
    FA_p = zeros(1,48);
    FA_p(1,i) = 1;

    %function calls
    T_FAodd(i) = FAoddUC;%calls function FAodd stores fin pressure,
roll torques on Fin A, Odd
    T_FAeven(i) = FAevenUC;%calls function FAeven stores fin
pressure, roll torques on Fin A, Even

    T_FA(i) = T_FAodd(i) + T_FAeven(i);
end

U=sqrt(2*sum((T_FA.*0.0375).^2));
U_OAOA=2*U
U_AOA=sqrt(2)*U

```

FAoddUC.m

```

function T_FAodd = FAoddUC

%global variable declaration
global dir run J FA_p tap_coords fin_area pressure_precision

%data filter
p = FA_p(:,[1:2:47]);%maintains pressure data for Fin B, odd
pressure taps
x = tap_coords(:,1);%maintains x coordinates for Fin A, odd pressure
taps
y = tap_coords(:,2);%maintains y coordinates for Fin A, odd pressure
taps
z = tap_coords(:,3);%maintains z coordinates for Fin A, odd pressure
taps

%function calls
[vertices, area_vectors, r, a_surf] = vert_and_area(x, y, z);%calls
function to determine triangle vertices,
...area vectors, triangle centroids, and area on Fin A, Odd within
tapped region
p_triangle = triangle_pressures(p, vertices);%calls function to
determine mean pressures for each triangle
T_tot = moment(p_triangle, area_vectors, r);%calls function to
determine torque for the entire surface, Fin A:Odd

%calculations
T_tot_x = -1.*T_tot(:,1);

%scales torque to approximate the torque over the entire Fin A, Odd
T_FAodd = T_tot_x*(fin_area/a_surf);

```

Distribution

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