



Atmospheric Mining in the Outer Solar System

*Bryan Palaszewski
Glenn Research Center, Cleveland, Ohio*

NASA STI Program . . . in Profile

Since its founding, NASA has been dedicated to the advancement of aeronautics and space science. The NASA Scientific and Technical Information (STI) program plays a key part in helping NASA maintain this important role.

The NASA STI Program operates under the auspices of the Agency Chief Information Officer. It collects, organizes, provides for archiving, and disseminates NASA's STI. The NASA STI program provides access to the NASA Aeronautics and Space Database and its public interface, the NASA Technical Reports Server, thus providing one of the largest collections of aeronautical and space science STI in the world. Results are published in both non-NASA channels and by NASA in the NASA STI Report Series, which includes the following report types:

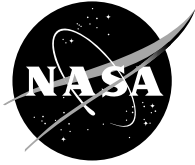
- **TECHNICAL PUBLICATION.** Reports of completed research or a major significant phase of research that present the results of NASA programs and include extensive data or theoretical analysis. Includes compilations of significant scientific and technical data and information deemed to be of continuing reference value. NASA counterpart of peer-reviewed formal professional papers but has less stringent limitations on manuscript length and extent of graphic presentations.
- **TECHNICAL MEMORANDUM.** Scientific and technical findings that are preliminary or of specialized interest, e.g., quick release reports, working papers, and bibliographies that contain minimal annotation. Does not contain extensive analysis.
- **CONTRACTOR REPORT.** Scientific and technical findings by NASA-sponsored contractors and grantees.

- **CONFERENCE PUBLICATION.** Collected papers from scientific and technical conferences, symposia, seminars, or other meetings sponsored or cosponsored by NASA.
- **SPECIAL PUBLICATION.** Scientific, technical, or historical information from NASA programs, projects, and missions, often concerned with subjects having substantial public interest.
- **TECHNICAL TRANSLATION.** English-language translations of foreign scientific and technical material pertinent to NASA's mission.

Specialized services also include creating custom thesauri, building customized databases, organizing and publishing research results.

For more information about the NASA STI program, see the following:

- Access the NASA STI program home page at <http://www.sti.nasa.gov>
- E-mail your question via the Internet to help@sti.nasa.gov
- Fax your question to the NASA STI Help Desk at 301-621-0134
- Telephone the NASA STI Help Desk at 301-621-0390
- Write to:
NASA STI Help Desk
NASA Center for AeroSpace Information
7121 Standard Drive
Hanover, MD 21076-1320



Atmospheric Mining in the Outer Solar System

Bryan Palaszewski
Glenn Research Center, Cleveland, Ohio

Prepared for the
41st Joint Propulsion Conference and Exhibit
cosponsored by AIAA, ASME, SAE, and ASEE
Tucson, Arizona, July 10–13, 2005

National Aeronautics and
Space Administration

Glenn Research Center
Cleveland, Ohio 44135

This work was sponsored by the Low Emissions Alternative
Power Project of the Vehicle Systems Program
at the NASA Glenn Research Center.

Level of Review: This material has been technically reviewed by technical management.

Available from

NASA Center for Aerospace Information
7121 Standard Drive
Hanover, MD 21076-1320

National Technical Information Service
5285 Port Royal Road
Springfield, VA 22161

Available electronically at <http://gltrs.grc.nasa.gov>

Atmospheric Mining in the Outer Solar System

Bryan Palaszewski
National Aeronautics and Space Administration
Glenn Research Center
Cleveland, Ohio 44135

Abstract

Overviews, analyses, and assessments of various mining options for outer planet atmospheres are presented. Options for mining include aerostat borne factories that process the atmosphere and cruisers or scoopers that dip into or reside in the atmosphere. Assessments of the mission complexity for several mining scenarios are presented. Propulsion options for these scenarios are also assessed. Options for mining and transporting atmospheric constituents from the outer planet of our solar system were reviewed. Comparisons of the mining and transportation techniques were also assessed. Comparisons of the complexity of lunar helium 3 (^3He) mining and mining in the outer solar system planetary atmospheres were conducted. Mission options that reduce the complexity of the transportation systems are suggested.

Nomenclature

^3He	Helium 3
^4He	Helium (or Helium 4)
delta-V	Change in velocity (km/s)
DS-1	Deep Space One
H_2	Hydrogen
He	Helium
ISRU	In Situ Resource Utilization
Isp	Specific Impulse (s)
KBO	Kuiper Belt Object
kWe	Kilowatts of electric power
MT	Metric tonnes
NEP	Nuclear Electric Propulsion
NTP	Nuclear thermal propulsion
O_2	Oxygen
PPB	Parts per billion
SR	Sample Return

Introduction

Exploration of the solar system is a monumental task. The distances to the outer planets are quite large, and the radiation produced by the planets can be deadly to humans (refs. 1 and 2). Also, the sunlight is quite weak at the outer planets, requiring new power systems for sustainable power for short and long duration missions (refs. 3, 4, and 5) to be developed. In the far future, enormous amounts of energy might be derived from extraterrestrial locations, even one day powering our entire planet (ref. 5).

Mining in the outer solar system is an important option for exploration and exploitation (refs. 5 to 10). Launching and transporting all of the materials for exploration from Earth is expensive and may make the idea of exploration untenable. The large reserves of atmospheric gases in the outer planets are an excellent resource for fuels and other life sustaining or colony building gases (ref. 10). The moons of the planets can be a great resource for oxygen, ceramic precursors, and metals. Outer planet moons, such

as Europa, Ganymede, and Callisto, may have reserves of liquid and frozen water. Specialized factories created in the outer solar system can leverage all of these resources and allow for extended stays in that cold, dark environment.

In-Situ Resources Utilization

Many decades of research have been focused on using the natural resources of the solar system to allow sustainable human exploration and exploitation of the environments of the planets and the Sun (refs. 4 to 23). Everything from preliminary experiments in propellant production to creation of human colonies in space has been proposed (refs. 11 to 23). Studies such as those of reference 10 have shown that the outer planets can provide the rich resources for interstellar exploration and the eventual human colonization of the galaxy.

As human exploration is initiated in the outer solar system, the travel time and other natural hazards (planetary radiation belts, solar coronal mass ejections, etc.) will create new challenges for the explorers. In-situ resources will likely be a great asset in this exploration. Shielding from radiation can be created with rock from the moons or with hydrogen and other liquefied gases from the planetary atmospheres. High speed travel will be augmented by nuclear fission (ref. 24) and advanced future fusion propulsion (ref. 25), fueled by the atmospheric gases.

Atmospheres of the Outer Planets

Highly energetic materials are available in the outer planet atmospheres. The composition of the outer planet atmospheres are shown in tables I and II (refs. 1, 26, 27, and 28). The elements of greatest interest are gases and ices. Figures 1 and 2 depict the atmospheric structures of the outer planets.

TABLE I.—OUTER PLANET ATMOSPHERE COMPOSITION
(by vol%, ref. 26)

Jupiter:		
Hydrogen	89.8	
Helium	10.2	
Other trace elements		
Saturn:		
Hydrogen	96.3	
Helium	3.3	
Other trace elements	0.4	
Uranus:		
Hydrogen	82.5	
Helium	15.2	
Methane	2.3	
Other trace elements	1.0	
Neptune:		
Hydrogen	80.0	
Helium	19.0	
Methane	1.0	
Other trace elements		

TABLE II.—SUMMARY OF ATTRACTIVE RESOURCES
(ref. 26)

Main Gases:	
	Hydrogen
	Helium
	Methane
Trace gaseous elements (present at ppm levels):	
	Helium 3
	Hydrogen deuteride
	Ethane
Ices deep within the atmosphere:	
	Hydrogen
	Hydrogen deuteride
	Methane
	Ammonia
	Water

The gases found in the atmospheres are excellent for fuels for chemical and nuclear propulsion systems. Hydrogen and methane are excellent fuels for chemical rockets that can be used for the ascent from and descent to the moons' surfaces. Also, hydrogen can be the fuel of choice for nuclear fission and fusion rockets. Helium 3 (${}^3\text{He}$) is another future fusion reactor fuel and can be found in these atmospheres (refs. 5 to 10). The hydrogen, helium, and / or other ices found deep in Uranus and Neptune may be crucial to the exploration beyond the solar system (refs. 5 to 10).

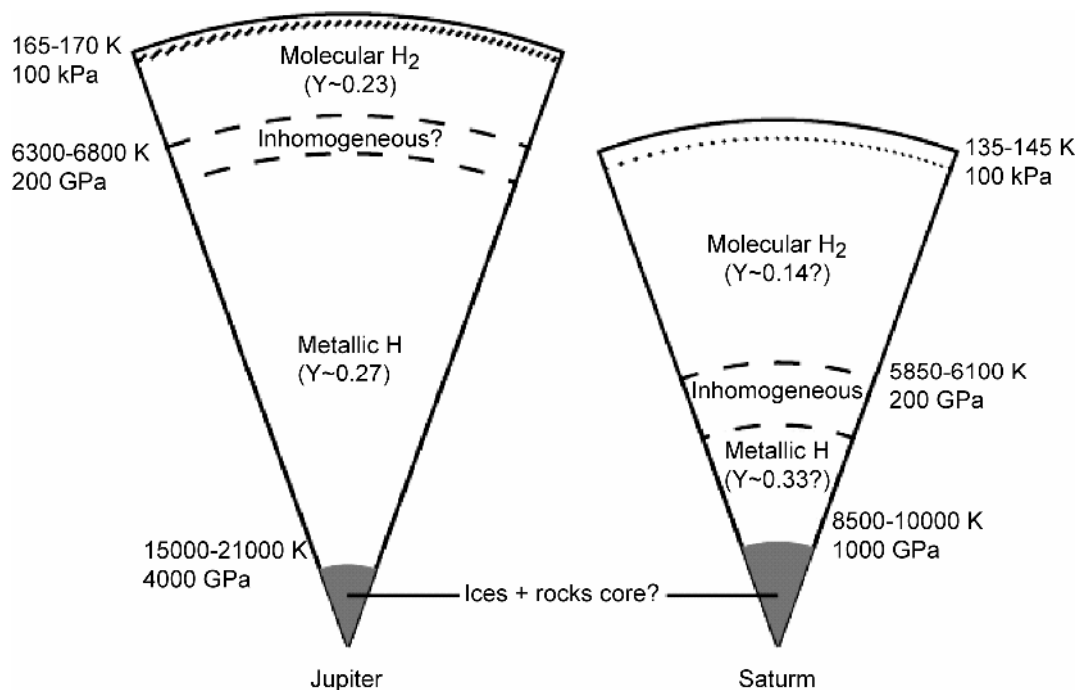


Figure 1.—Atmospheres of Jupiter and Saturn (Refs. 27 and 28).

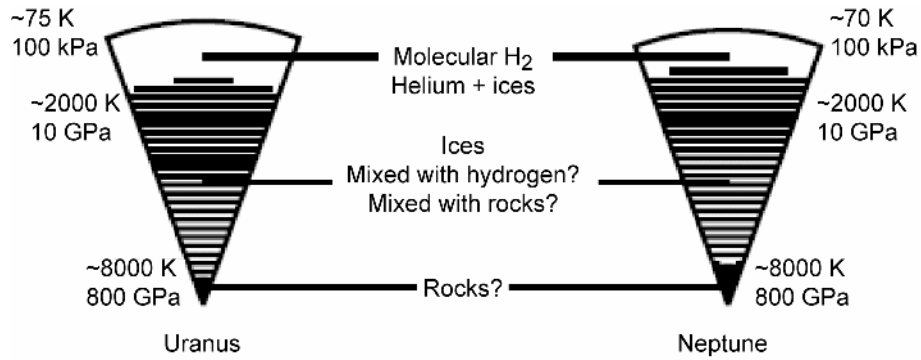


Figure 2.—Atmospheres of Uranus and Neptune (Refs. 27 and 28).

Why Perform Mining in Atmospheres?

It is possible that atmospheric mining may be easier than mining from outer planet moon surfaces. Also, the sheer abundances of gases and the locations of the planets give us a good place for way stations for exploration beyond the edges of the solar system. On the other hand, the delta-V required to closely approach Jupiter or any of the outer planets for repetitive atmospheric access and then returning to orbit can be quite high. Also, as the planets typically have powerful magnetic fields, the vehicle's resistance to radiation will be important for reliable operation.

One of the attractive materials that can be extracted from the solar system is ^3He . It is deposited in lunar regolith and exists in the outer planet atmospheres (ref. 5 to 19). This material can be an excellent nuclear fusion fuel and can reduce the neutron radiation created during the fusion process, extending the life of future power reactors (refs. 19 to 22).

Using ^3He from the lunar regolith has been extensively studied (refs. 11 to 22). Deposited over billions of years by the solar wind, ^3He is resting in the first hundred centimeters of lunar regolith, waiting for extraction. The fraction of ^3He in the lunar regolith is quite small: perhaps less than 5 to 100 parts per billion (ppb) in some regions. Thus, large robotic machines were envisioned to conduct the mining of the surface and slowly and carefully wrest the precious fuel from the Moon.

Predictions of the ^3He concentration in the helium of the atmosphere of Uranus are about 1×10^{-4} (by mass) (refs. 3 and 4). Though the levels seem quite small, the argument has been made that it is easier to extract the ^3He from the gaseous helium in the outer planet atmosphere. Factories stationed in the atmosphere could robotically mine the gas, and await orbital vehicles to gather the ^3He for delivery to other parts of the solar system.

If atmospheric mining can be made highly efficient, there may be great benefits for gathering ^3He from an atmosphere rather than the Moon's surface. Once a pipeline of deliveries is established, the flow of ^3He from the outer solar system might be more attractive than lunar regolith mining.

Wind Speeds in Atmosphere

Though atmospheric mining shows advantages, there are very high winds speeds and wind shears in these atmospheres (refs. 29 to 32). Figures 3 and 4 present the wind speeds and wind profiles for the 4 gas giants (ref. 29). Given the typical lightweight nature of aerospace vehicles, the predictability of wind speed in the atmosphere will have a powerful effect on the factory design and the related aeronautical vehicles.

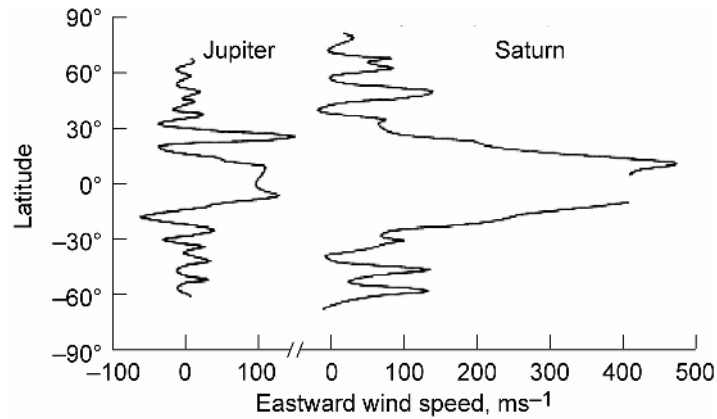


Figure 3.—Winds at Jupiter and Saturn (Ref. 29).

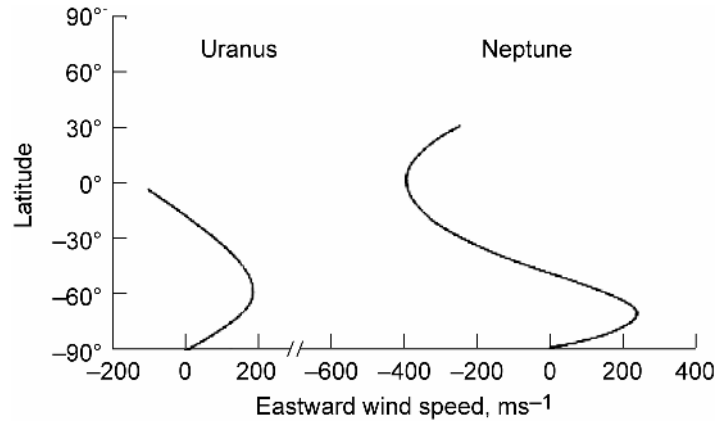


Figure 4.—Winds at Uranus and Neptune (Ref. 29).

While Jupiter and Saturn have more dramatic changes in wind shear, Uranus and Neptune appear to have more calm changes in wind profiles. The stresses imposed on any vehicle entering Jupiter's or Saturn's atmosphere may be too much for a conventional aerostat.

Due to more predictable atmospheric conditions and wind speeds, mining with aerostat borne stations seems more applicable for use at Uranus and Neptune, while more aggressive aerodynamic options (high speed cruisers) may be needed at Jupiter and Saturn.

Mining Methods

The nature of the mining stations may be a series of buoyant stations in the planetary atmosphere, aerodynamic scoopers that dive into the atmosphere or cruisers that ply the atmosphere, gathering the needed gases and liquefying them. The vehicle types and overall scenarios are described in figures 5, 6, and 7. In all cases, there are many highly interrelated systems and elements. The various scenarios are discussed below, with emphasis on the atmospheric mining elements.

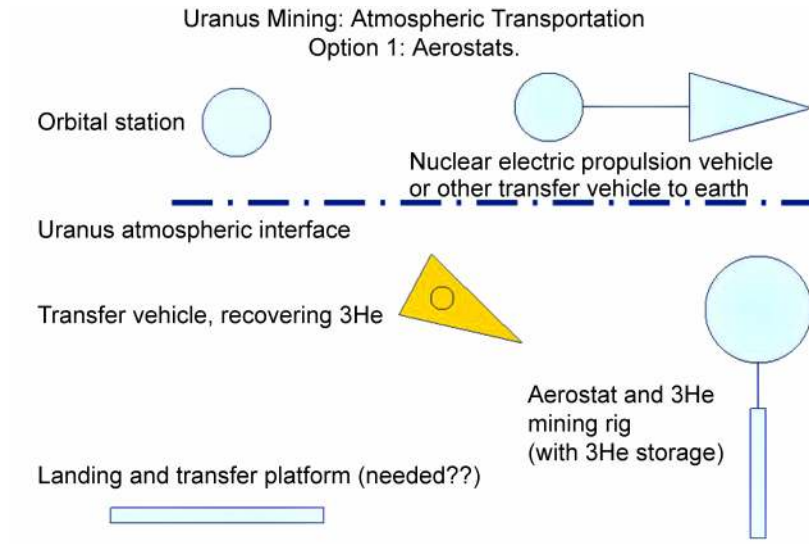


Figure 5.—Atmospheric mining with aerostats.

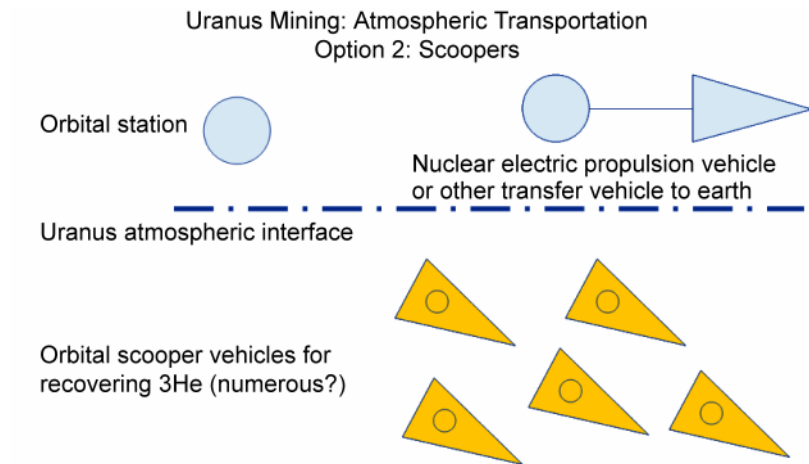


Figure 6.—Atmospheric mining with scoopers.

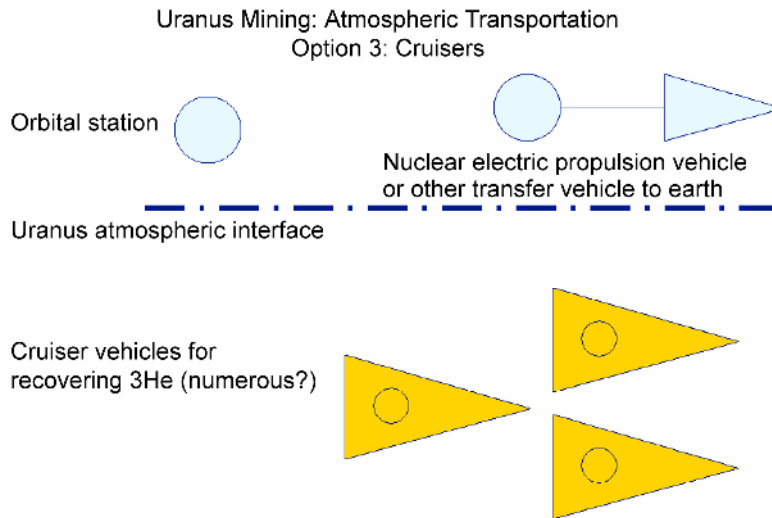


Figure 7.—Atmospheric mining with cruisers.

Aerostat stations

Figure 5 shows the aerostat scenario. Aerostat vehicles are buoyant stations that would persist in the atmosphere, and continually process atmospheric gases for the needed final products— ^3He , ^4He , and H_2 , etc. Orbital vehicles would perform round trip visits to the aerostat to deliver the final product to orbit or any other destination (a moon of the planet, transfer vehicle back to Earth, etc.). Aerostats have been considered for mining Jupiter’s atmosphere (refs. 33 to 38). The size of the overall plant was estimated with historical information, and the sizing of balloons (ref. 33). Figures 8 and 9 illustrate the overall conceptual design of an aerostat mining vehicles (ref. 33). The aerostat station was to be visited on a regular basis by vehicles in orbit, and the propellants and other gases derived from the atmosphere were delivered to the vicinity of Jupiter’s moon, Callisto. Callisto is an attractive way station as it can provide materials for vehicle construction from its surface, and it is outside Jupiter’s major radiation belts, thus making it safer for human explorers. Such a scenario may be the most attractive of the three atmospheric mining options.

Scoopers

Scoopers (ref. 33, 39 to 45) are atmospheric entry vehicles that scoop a small amount of the planet’s atmosphere and gather it aboard the vehicle for processing and/or consuming it as fuel for the vehicle itself. The scooper vehicles can provide a short term option for fuel capturing, and the processing may take place in orbit. The complexity, mass, and volume of the scooper may make it impractical to place all of the subsystems for processing on the entry vehicle.

To gather large amounts of gas, the scoopers would perform many hundreds of missions in to the atmosphere. Overall, the lifetime of the scoopers may be limited by the heating and other stresses due to multiple entries and exits, as well as the on-board propulsion component life. Also, the atmospheric composition may impose new stresses on materials. Flying through icy particles and potentially reactive materials (hydrogen, ethane, etc.) will no doubt create new challenges for long lived vehicles.

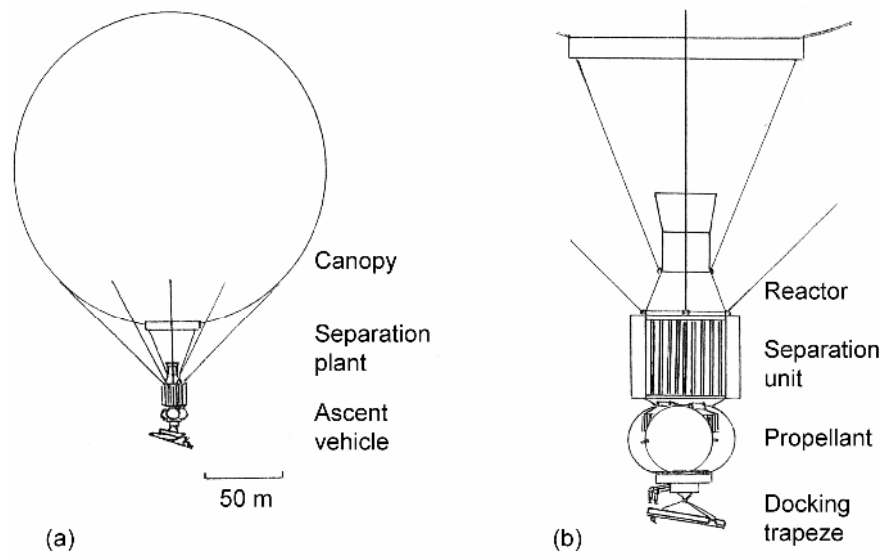


Figure 8.—Aerostat conceptual design (Ref. 33). (a) Overall scheme with the ascent vehicle docked. (b) Detail of the factory complex.

TABLE 1. Assessment of gross energy requirements*.

Liquefaction of 47.5 gm/sec. ^3He @ 3.2°K (Refrigeration factor 95.6)	= 4.7 MW
Liquefaction of 31.7 gm/sec. D_2 @ 23°K (Refrigeration factor 13.3)	= 0.5 MW
Liquefaction of 151.8 gm/sec. H_2 @ 20°K as ascent propellant ($c = 40$ km/s)	= 4.3 MW
Circulation pumps for 28 tonnes/sec. atmosphere	= 4.2 MW
Separation processes	= 3.0 MW
Heat leaks - 1°K temperature differential	
on Helium flow	= 24.7 MW
on Hydrogen flow	= 372.2 MW
Separation Plant Total	= 413.6 MW
Ascent energy for helium-3 (transport energy efficiency = 0.3)	= 68.4 MW
Ascent energy for deuterium	= 45.6 MW
Average power requirement for ascent ferry	= 114.0 MW

* All values averaged over all collection units and 20 year collection period.

Figure 9.—Estimated power requirements for aerostat conceptual design (Ref. 33).

Typical mission life for orbiting or flyby spacecraft has been from 8 years for Mars Global Surveyor (refs. 46 and 47) to 40 years for Voyager (ref. 48). The lifetime for vehicles which repetitively enter atmospheres is not known but can be estimated from historical data and engineering estimates. Hypersonic vehicles used in the past have usually been expendable vehicles, such as reentry bodies for national defense or military missiles. Their heating rates are quite high and they used ablative materials to withstand the heat fluxes for those missions. These vehicles were by definition designed for short lifetimes. The Space Shuttle has experienced many hypersonic reentries but only after much ground maintenance after every entry. Long life hypersonic vehicles may be a truly Olympian challenge.

Cooling of the vehicle surfaces to allow the entry and the scooping of the atmosphere has been studied (refs. 33 and 39 to 45). The aerothermodynamics of the inlet, liquefier and compressors will be a significant challenge. As the technology for this type of vehicle is very complex from many points of view, the overall vehicle complexity and potentially short vehicle lifetime of the scooper scenario may make it the least attractive of the three mining options.

Cruisers

A cruiser vehicle would be more akin to a traditional aircraft, remaining in the atmosphere, and gathering atmospheric gases as it flies. Smaller vehicles would visit the cruiser and return the liquefied gases to orbit. Many launch vehicle and transatmospheric vehicle designs have been investigated using a liquid air cycle engine (LACE) technology (refs. 49 to 52). A similar design for liquefying hydrogen, helium and ^3He would be used in the outer planet atmospheres. Due to the density differences between air and hydrogen, the vehicle size and mass would be much larger, or many more cyclical visits to the atmosphere and the cruisers would be needed to deliver the same amount of liquefied gases for the same mass of vehicle. Again, delivering this complex vehicle or fleets of vehicles from Earth may be a prohibitive task. In-situ construction and maintenance of such vehicles may be more attractive.

Interplanetary Transportation Options

After suggesting the possible mining options in the atmosphere, the transportation to and from the planet must be addressed. To support the scenarios discussed earlier, there will be both interplanetary spacecraft as well as transportation from orbit to the atmosphere and back to orbit. Earlier studies suggested that the potential complexity of lunar ^3He mining should lead to consideration of outer planet atmospheric mining. Theoretically, the extraction of ^3He from a gaseous atmosphere will be easier than extracting it from lunar regolith (refs. 5 to 19). The transportation complexity and time of flight may however create a barrier to using the outer planets for powering Earth related activities.

Figure 10 shows the changes in velocity (Δv) required for Neptune and other outer planet missions (ref. 3). The Δv s required are quite high and necessitate some advanced propulsion concept beyond chemical propulsion. A likely first choice is nuclear electric propulsion for the interplanetary missions. Solar energy is quite difficult to use at distances beyond Mars and the main asteroid belt. Nuclear electric transfer vehicles would be a critical part of any scenario. Additional vehicles for transportation in the atmosphere and factories for extracting the hydrogen, helium, or ^3He would also be required. Nuclear thermal propulsion (NTP) has also been suggested (ref. 7) however, due to the lower specific impulse (Isp) for NTP, the resulting higher total mass of propellant needed for NTP is higher than NEP. Therefore, especially for cargo missions where trip time is not a critical issue, the selection process for the vehicle usually favors NEP (refs. 47 and 48).

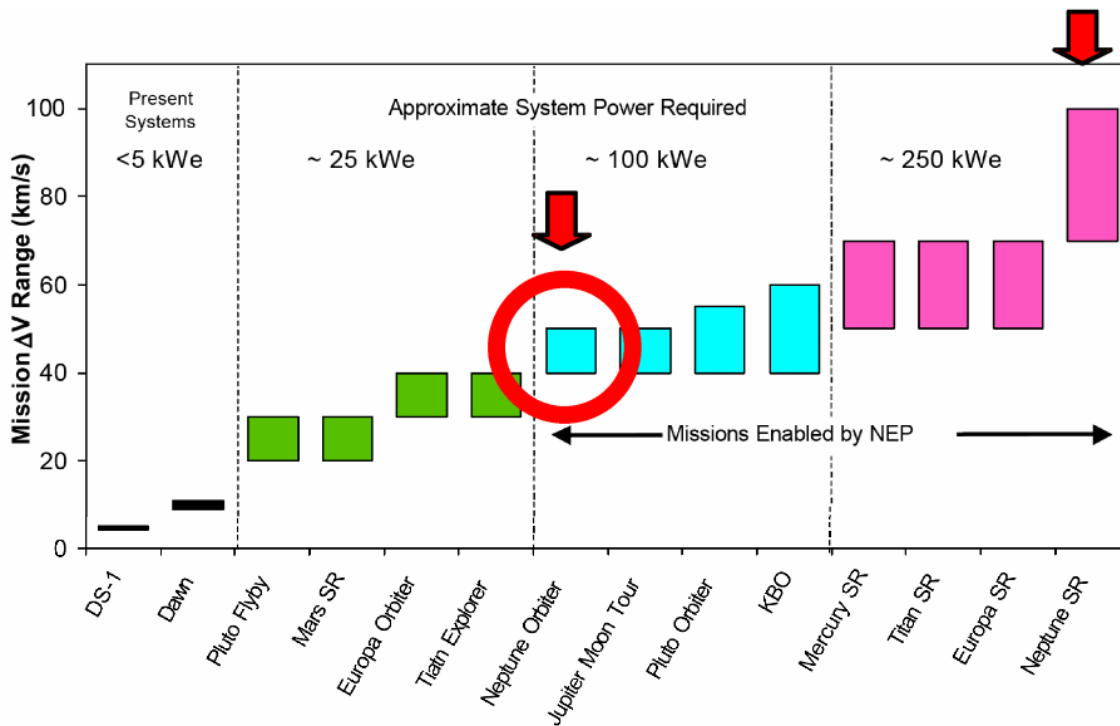


Figure 10.—Mission velocity changes for outer planet missions (Ref. 3).

Planning Scenarios for Development

The three scenarios were assessed and comparison of the maneuvers and their potential complexity were made. Figures 5, 6 and 7 illustrate the three options for atmospheric mining. Table III provides a list of possible vehicle maneuvers for a Uranus mission set. The major variations in complexity lie in the maneuvers in the atmosphere. In some cases, the scenarios have a large number of platforms and maneuvers required. The number of vehicles needed and their associate maneuvers will have to be managed to create a manageable set of vehicles for the selected scenario. Repetitive maneuvers requiring multiple entries and exits of the atmosphere will stress many aspects of materials, design, crews, robots, and scheduling.

Mission complexities

Mission complexity is a serious consideration for the success of any space endeavor. The limits of machines and their crews are often stressed during any space mission. There seem to be some fundamental limits to some of the atmospheric mining options. The number of maneuvers and the complexity of those maneuvers may make some options unacceptable for long term robotic or human missions.

The table III list of planned maneuvers for the atmospheric mining missions gives some implication of the number of and complexity of the maneuvers. With scoopers vehicles, the entry velocities are usually very high, and the time in the atmosphere is quite short—the gas scooping time may be only minutes to less than 1 hour. Cruising in the atmosphere is another possibility, but the size of such a cruiser may be prohibitive. The many visits of smaller ^3He recovery vehicles (going from orbit, to the atmosphere, and back to orbit) to off load the cruisers may take a toll during the many atmospheric visits. Previous studies of Liquid Air Cycle Engines (LACE) for air separation have shown that the vehicle may have to remain in the air for long periods to make a relatively small amount of propellants. Using aerostat borne factories may be the only realistic option for long term missions. They can remain for long periods in the atmosphere, and have ships visit to recover the processed gases and liquids from the factory. Deployment of the very large aerostats or balloons may be another issue.

TABLE III.—MISSION MANEUVER SUMMARY
FOR ATMOSPHERIC MINING

Maneuvers for Uranus mining flights

- Mission maneuvers
 - Depart from Earth
 - Launch from Earth, orbit established around Earth
 - Depart for Uranus
 - Trans Uranus injection
 - Coast to Uranus
 - Low thrust to Uranus
 - High thrust to Uranus
 - Arrival at Uranus
 - Uranus orbit insertion
 - Arrival at Uranus
 - Uranus orbit insertion
 - Aerocapture?
 - Propulsive orbit insertion
 - Descend to atmosphere
 - Descent orbit initiation
 - Aerobraking to lower orbit
 - Prior to descent into the atmosphere
 - Enter atmosphere
 - Maneuver to destination
 - Arrival at atmospheric station
 - Arrival at atmospheric scooping altitude
 - On-station maneuvers
 - Station keeping at atmospheric station
 - Station keeping at atmospheric scooping altitude
 - Exit the atmosphere
 - Depart from atmospheric station
 - Depart from atmospheric scooping altitude
 - Depart Uranus
 - Trans Uranus injection
 - Coast
 - Low thrust to Earth
 - High thrust to Earth
 - Return to Earth
 - Aerocapture
 - Propulsive capture, insertion
 - Rendezvous
 - Offload cargo
 - Refurbish or dispose of vehicle

Mass Delivery Implications

In previous studies (ref. 9), the mass of ^3He delivered to Earth was about 450 to 500 MT per year. In order to support Earth based power systems, the number of vehicles required may be extremely large. Overall, the processing of ^3He from the atmosphere will likely take hours to days to months, depending on the available power level. Only preliminary estimates were developed by the Daedalus Project (ref. 33). Reference 33 showed ^3He production of 1500 MT per year for 20 years for the Daedalus vehicle. It is likely that small amounts will be delivered to orbit at any one time, requiring many vehicle flights to traverse the atmosphere to recover the ^3He fuel. Until very high velocity vehicles are developed, delivering large masses to the Earth may have to wait. Until such fast transportation is available, performing initial experiments and perhaps constructing large vehicles in the outer solar system and using this precious fuel for exploration may be much more attractive in the near term.

Comparisons with Lunar Mining

Lunar Helium 3 (^3He) may have a large amount of time and effort associated with it, but the travel time to Earth is very short (refs. 9 to 19). The large mining equipment can be sent to the lunar surface and repaired in a low gravity environment, with spare parts available from Earth factories. Currently, servicing an outer planet factory, which is many years and millions of kilometers away may be a great hindrance.

Outer planet ^3He extraction has many issues associated with the additional atmospheric constituents. The number and complexity of maneuvers required for ^3He mining may make it prohibitive, as noted in table III. Also, returning a sample or large load of ^3He to Earth may take a prohibitive amount of time. A round trip time of flight for a Uranus mission may be several years. However, once a pipeline of many continuing flights is established, the long flight time may not be a critical issue.

Figure 11 shows the percentages of Helium in the lunar regolith, up to 70 parts per million (refs. 55 and 56). Figure 12 shows the fraction of ^3He in the lunar regolith or dust (refs. 55 and 56). Only a fraction of the total helium in the dust is ^3He (with overall only 5 to 100 PPB of ^3He in the dust). In the outer planet atmospheres, it is estimated that the ^3He mass fraction in the helium mass fraction in the atmosphere is 10^{-4} (refs. 5 to 9). Even if the total fraction of ^3He in the outer planet helium (^4He) is so much higher in the Uranus atmosphere, the other operational factors discussed in this paper may make the idea of ^3He mining in the outer planet regions difficult to implement.

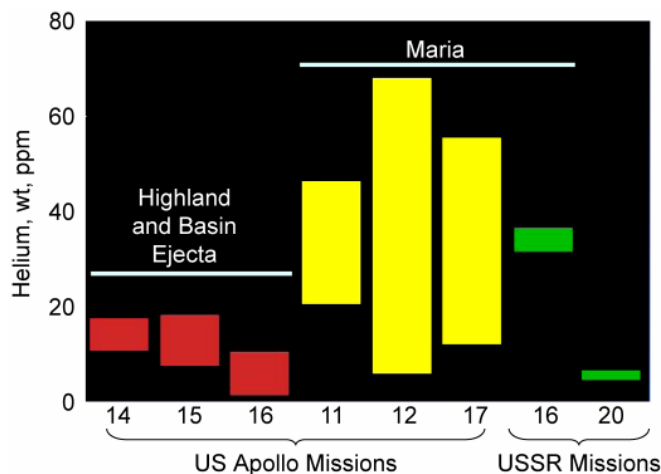


Figure 11.—Helium 4 in the lunar dust (ref. 55). <http://fti.neep.wisc.edu/neep602/FALL97/LEC19/lecture19.html>

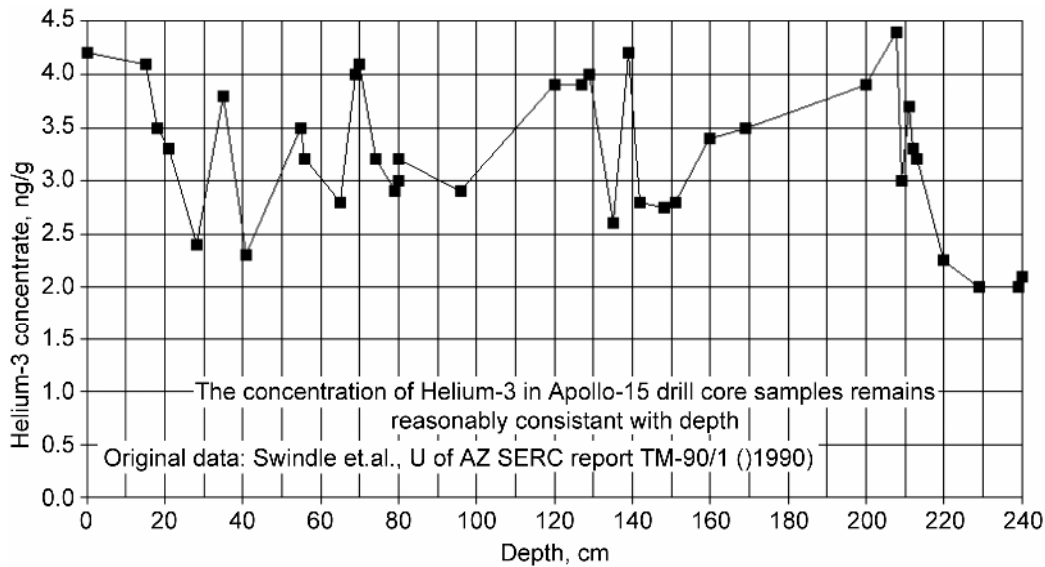


Figure 12.—Helium 3 in the lunar dust (ref. 55). <http://fti.neep.wisc.edu/neep602/LEC18/IMAGES/fig25.gif>

Concluding Remarks

It is likely that the delivery of 3He to Earth from the outer solar system will be a difficult and lengthy task. The number of maneuvers, the time scale for transportation, the complexity of the vehicles, and the lifetime of the mining factories or atmospheric vehicles are all important factors to investigate.

An alternative option is the use of the 3He in the outer solar system (not returning it to Earth). Interplanetary and interstellar missions have been investigated (refs. 25 and 33 to 38), and the fuel needed for these missions might be more effectively used in transportation in the solar system. Given the complexity, using the gathered fuel for transportation about or beyond the solar system is likely to be a better option than returning it to Earth.

Production rates for such in-situ missions will be important for mission effectiveness. Supplying the Earth with approximately 450 to 500 metric tonnes (MT) per year of 3He for fusion power has been suggested (ref. 9). Creating the infrastructure for sustainable Earth delivery will likely be a daunting task. Several hundred flights a year to Earth is an admirable goal, but right now it is beyond our reach. An even more audacious idea is the fueling of interstellar vehicles. An extreme example of very high production rates is the Daedalus Project. Based on the Daedalus Project work (refs. 33 to 38), the mass of an initial robotic interstellar vehicle may be 54,000 MT (ref. 35). Approximately 50,000 MT of this mass is the fuel needed for propulsion: 60 percent of the total propellant mass is 3He (ref. 35). At a production rate of 1 MT per visit to the atmosphere, the feasibility of such production systems is unattractive. So many flights would be needed; the time for the project would be prohibitive, extending many years/decades. The Daedalus Project suggested 1500 MT per year for a 20 year period and perhaps this rate may be attractive. Producing 50,000 MT per year is attractive, but may place an undue burden on the human crews in such a harsh environment. Robotic systems, tended by humans, will likely be needed for effective construction and fuel production.

Alternative methods of producing 3He on the Earth have been investigated (refs. 20 to 22). Production from past weapons caches (tritium decay) and production from primordial sources (natural gas and helium stores) seem to have a small upper limit on production mass of approximately 17 kg per year (refs. 20 to 22). For the nearer term, lunar production may prove to be effective in the future for many tons or ultimately hundreds of tons of 3He per year (refs. 5 and 11 to 19). With all such in situ resource

usage, small experiments will come first, to prove the technologies, and allow a more effective scale up to the enormous tasks of outer planet exploration.

Highly efficient ISRU plants will be needed to reduce the potential for damage to planetary environments that might be harbors for either past or future alien life. The human exploration of the solar system and others like it will come with time. In very advanced future times, creating new starships or titanic “world ships” (containing entire oceans, continents, and colonies of humans) will require the ability to manipulate or even dismantle entire moons of our outer planets or the moon about the planets of other solar systems (refs. 33 to 39).

Conclusions

Mining of the outer planets may take many directions, the two most important of which are mining the moons and the atmospheres of those planets. Using specialized factories, the energy of hydrogen, helium, and helium 3 gases can allow us to power nuclear fusion propulsion systems. Chemical rockets for landing and ascending from the moons can also derive their propellants from the moons’ ices and/or the planetary atmospheres.

Many issues may make the implementation of these ideas more complex than was originally thought. The dynamics of the atmosphere, radiation, and the energy for orbital transfer all call for very energetic and reliable propulsion systems to allow for rapid, reliable, and repetitive visits to the planets and their moons. For example, due to the more predictable atmospheric conditions and wind speeds at Uranus and Neptune, mining with aerostat borne stations seems more applicable for use there, while with more dynamic winds at Jupiter and Saturn, more aggressive aerodynamic options (high speed cruiser aircraft) may be needed.

The numerous maneuvers and extended vehicle lifetimes required for wresting the resources from the outer planets will likely lead us to more easily completed options. Mining of outer planet moons, especially if water is detected there, will be an excellent option. Initial experiments to prove the efficacy and lifetime of atmospheric scoopers need to be conducted before committing to such ambitious vehicles.

There are great possibilities for human exploration of the solar system. Effective gathering and conversion of the resources in outer planet atmospheres will control the timetables for this human exploration. Indeed the use of these resources of the outer planets can lead us to our first steps toward the stars.

References

1. Carl D. Murray and Stanley F. Dermott, Solar System Dynamics, Cambridge Press, May 2000.
2. John S. Lewis, Physics and Chemistry of the Solar System, Academic Press, San Diego, CA, 1995.
3. Ira Katz, John R. Brophy, John R. Anderson, James E. Polk, Dan M. Goebel, “Technologies to Improve Ion Propulsion System Performance, Life and Efficiency for NEP,” Advanced Space Propulsion Workshop, Huntsville, AL, April 17, 2003, available at:
http://www.spacetransportation.com/ast/2003_prop_workshop/pres-pdf/9a_katz.pdf
4. Dudenhoefer, James E. and George, Patrick J., “Space Solar Power Satellite Technology Development at the Glenn Research Center: An Overview,” NASA/TM—2000-210210, July 2000.
5. Kearney, John J., “Report of NASA Lunar Energy Enterprise Case Study Task Force,” NASA TM—101652, July 1, 1989.
6. Jeffrey Van Cleve, Carl Grillmair, Mark Hanna, “Helium-3 Mining Aerostats in the Atmosphere of Uranus,” Abstract from Space Resources Roundtable, October 24–26, 2001.
<http://www.mines.edu/research/srr/2001abstracts/vancleve.PDF>
7. Van Cleve, J.E.; Grillmair, C.J., “Small Nuclear-powered Hot Air Balloons for the Exploration of the Deep Atmosphere of Uranus and Neptune,” Forum on Innovative Approaches to Outer Planetary

- Exploration 2001–2020, p. 87, 01/2001.
<http://www.lpi.usra.edu/meetings/outerplanets2001/pdf/4107.pdf>
8. Eder, Dani, “Helium–3 Mining of Uranus,” Space manufacturing 8—Energy and materials from space; Proceedings of the 10th Princeton/AIAA/SSI Conference, Princeton, NJ, May 15–18, 1991. Washington, DC, American Institute of Aeronautics and Astronautics, 1991, p. 263–266.
 9. Paniagua, J., Powel, J., Maise, G., “A Cost Effective Space Infrastructure for Retrieval of Helium–3 from Uranus for Earth-based Fusion Power Systems Utilizing MITEE Nuclear Propulsion System,” Plus Ultra Technologies, Inc., Report number PUR–11, July 23, 1999, available at: <http://www.newworlds.com/reports/PUR–11.PDF>
 10. Bond, A.; Martin, A.R.; Buckland, R.A.; Lawton, A.T.; Mattinson, H.R.; Parfitt, J.A.; Parkinson, R.C.; Richards, G.R.; Strong, J.G.; Webb, G.M.; et al. , “Project Daedalus; Unmanned Flyby Of Barnard’s Star With Nuclear Pulse Rocket,” Spaceflight, vol. 19, Dec. 1977, p. 419–430.
 11. Sviatoslavsky, I.N. , “Processes And Energy Costs For Mining Lunar Helium–3,” NASA, Lewis Research Center Publication, Lunar Helium–3 and Fusion Power, pp. 129–146. Sept. 1, 1988.
 12. Crabb, J.C., White, S.W., Wainwright, L.P., Kratz, S.E., Kulcinski, G.L., “Fifty Years Research in Helium–3 Fusion and Helium–3 Resources,” WCSAR–TR–AR3–9312–1 December 1993 (revised May 1994) [also UWFD–935].
 13. Wittenberg, L.J., “In-Situ Extraction of Lunar Soil Volatiles,” WCSAR–TR–AR3–9311–3. Prepared for Space 94, The 4th International Conference and Exposition on Engineering, Construction and Operations in Space, and The Conference and Exposition/Demonstration on Robotics for Challenging Environments, Feb. 26–Mar. 3, 1994, Albuquerque NM, Nov. 1993.
 14. Sviatoslavsky, I.N., “The Challenge of Mining He–3 on the Lunar Surface: How All the Parts Fit Together,” WCSAR–TR–AR3–9311–2. Prepared for Space 94, The 4th International Conference and Exposition on Engineering, Construction and Operations in Space, and The Conference and Exposition/Demonstration on Robotics for Challenging Environments, Feb. 26–Mar. 3, 1994, Albuquerque, NM, Nov. 1993.
 15. K.R. Harris, H.Y. Khater, G.L. Kulcinski, “Remote Sensing of Astrofuel,” WCSAR–TR–AR3–9311–1 Prepared for Space 94, The 4th International Conference and Exposition on Engineering, Construction and Operations in Space, and The Conference and Exposition/Demonstration on Robotics for Challenging Environments, Feb. 26–Mar. 3, 1994, Albuquerque, NM, Nov. 1993.
 16. N. Duffie, G. Kulcinski, I. Sviatoslavsky, B. Bartos, S. Rutledge, L. Wittenberg, T. Ylikorpi, E. Mogahed, “Study of an Unmanned Lunar Mission for Volatile Gas Recovery (Phase 1—Final Report),” WCSAR–TR–AR3–9309–1 Sept. 1993.
 17. I.N. Sviatoslavsky, “Lunar He–3 Mining: Improvements on the Design of the UW Mark II Lunar Miner,” WCSAR–TR–AR3–9201–2 prepared for Space 92, The Third International Conference on Engineering, Construction, and Operations in Space, May 31–June 4 1992, Denver, CO, Jan. 1992.
 18. Everett K. Gibson, Jr., Roberta Bustin, David S. McKay, “Lunar Hydrogen: A Resource for Future Use at Lunar Bases and Space Activities,” Presented at the Symposium on Lunar Bases and Space Activities in the 21st Century, Houston, TX, Apr. 5–7, 1988.
 19. Sviatoslavsky, I.N. and Jacobs, M., “Mobile Helium–3 Mining and Extraction System and Its Benefits Toward Lunar Base Self-Sufficiency;” WCSAR–TR–AR3–8808–1, August 1988.
 20. Kulcinski, G.L., 1993, “History of Research on 3He Fusion,” University of Wisconsin WCSAR–TR–AR3–9307–3, p. 9, Presented at the Second Wisconsin Symposium on Helium–3 and Fusion Power; Proceedings of a Symposium held in Madison, WI, July 19–21, 1993.
 21. White, S.W., 1996, “A Current Bibliography of Helium–3 Research,” University of Wisconsin Report UWFD–1003, January 1996.
 22. Wittenberg, L.J. Santarius, J.F. and Kulcinski, G.L., 1986, “Lunar Source of He–3 for Commercial Fusion Power,” Fusion Technology, 10, p.167.
 23. “Humans in Space: Colonizing the Galaxy,” available at: <http://library.thinkquest.org/C003763/index.php?page=future05>

24. Melissa L. McGuire, Stanley K. Borowski, Lee M. Mason, and James Gilland “High Power MPD Nuclear Electric Propulsion (NEP) for Artificial Gravity HOPE Missions to Callisto,” NASA/TM—2003-212349, Dec. 2003.
25. Craig H. Williams, Leonard A. Dudzinski, Stanley K. Borowski, and Albert J. Juhasz, “Realizing “2001: A Space Odyssey””: Piloted Spherical Torus Nuclear Fusion Propulsion,” NASA/TM—2005-213559, AIAA—2001—3805, Mar. 2005.
26. National Space Science Data Center - Solar system web sites -
<http://nssdc.gsfc.nasa.gov/planetary/factsheet/jupiterfact.html>
<http://nssdc.gsfc.nasa.gov/planetary/factsheet/saturnfact.html>
<http://nssdc.gsfc.nasa.gov/planetary/factsheet/uranusfact.html>
<http://nssdc.gsfc.nasa.gov/planetary/factsheet/neptunefact.html>
27. Guillot, Tristan, “Interiors of Giant Planets Inside and Outside the Solar System,” *Science*, October 1999; vol. 286: pp. 72–77.
28. D.J. Stevenson, “Interiors of the Giant Planets,” *Annual Review of Earth and Planetary Sciences*, 10, 257, (1982).
29. Ingersoll, D. “Science Rationale for Giant Planet Probes,” Workshop on Extreme Environments Technologies for Space Exploration, May 14–16, 2003,
http://extenv.jpl.nasa.gov/presentations/Ingersoll-Science_Rationale_for_Giant_Planet_Probes.pdf.
30. Erich Karkoschka, “Clouds of High Contrast on Uranus,” *Science*, vol. 280, Apr. 24, 1998. pp. 570–572.
31. C.C. Porco, E. Baker, J. Barbara, K. Beurle, A. Brahic, J.A. Burns, S. Charnoz, N. Cooper, D.D. Dawson, A.D. Del Genio, T. Denk, L. Dones, U. Dyudina, M.W. Evans, B. Giese, K. Grazier, P. Helfenstein, A.P. Ingersoll, R.A. Jacobson, T.V. Johnson, A. McEwen, C.D. Murray, G. Neukum, W.M. Owen, J. Perry, T. Roatsch, J. Spitale, S. Squyres, P. Thomas, M. Tiscareno, E. Turtle, A.R. Vasavada, J. Veverka, R. Wagner, and R. West, “Cassini Imaging Science: Initial Results on Saturn’s Atmosphere,” *Science*, Feb. 25, 2005, vol. 307, pp. 1243–1247.
32. Jeffery L. Hall, Muriel A. Noca and Robert W. Bailey, “Cost—Benefit Analysis of the Aerocapture Mission Set,” AIAA—2003—4658, July 2003.
33. Parkinson, R.C., “Project Daedalus: Propellant Acquisition Techniques,” *Journal of the British Interplanetary Society (JBIS) Interstellar Studies, Supplement, Final Report of the BIS Starship Study*, 1978.
34. Bond, A., Martin, A.R., “Project Daedalus Reviewed,” IAF PAPER 85–490, Oct 1, 1985.
35. Bond, A.; Martin, A.R.; Buckland, R. A.; Lawton, A.T.; Mattinson, H.R.; Parfitt, J.A.; Parkinson, R.C.; Richards, G.R.; Strong, J.G.; Webb, G.M.; et al. “Project Daedalus; Unmanned Flyby Of Barnard’s Star With Nuclear Pulse Rocket,” *Spaceflight*, vol. 19, Dec. 1977, pp. 419–430.
36. Freitas, R.A., Jr., “A Self-Reproducing Interstellar Probe,” *Journal of the British Interplanetary Society, (Interstellar Studies)*, vol. 33, July 1980, p. 251–264.
37. Strong, J., “Further Thoughts On Interstellar Exploration,” *Spaceflight*, vol. 23, May 1981, p. 140–143.
38. Parkinson, B., “The Starship As An Exercise In Economics; Interstellar Travel As Function Of National Economies,” *Journal of the British Interplanetary Society*, vol. 27, Sept. 1974, pp. 692–696.
39. Minovitch, M.A.. “Solar powered, self-refueling, microwave propelled interorbital transportation system,” AIAA PAPER 83–1446, Jun 1, 1983.
40. Parkinson, R.C., “Earth-moon transport options in the Shuttle and advanced Shuttle era,” *Journal British Interplanetary Society (Space Technology)*, vol. 34, Feb. 1981, p. 51–57.
41. Reichel, R.H. “The air-scooping nuclear-electric propulsion concept for advanced orbital space transportation missions,” IAF Paper 76–161, Oct 1, 1976.
42. Fatkin, Iu.M.; Tokarev, V.V. Air scoops in problems of optimization of power-limited flight. *Inzhenernyi Zhurnal*, vol. 5, no. 3, 1965, P. 531–536. and *Soviet Engineering Journal*, vol. 5, May–Jun. 1965, pp. 438–442.

43. Hanford, D.R.; Reichel, R.H.; Smith, T.L., "Potentialities of air-scooping electrical space propulsion systems." American Rocket Society, Electric Propulsion Conference, Berkeley, CA, Mar. 14–16, 1963.
44. Hanford, D.R.; Reichel, R.H.; Smith, T.L., "Potentialities of air-scooping electrical space propulsion systems," Report Number: REPT.-2391-62, Presented At The American. Rocket Society, Electric Propulsion Conf., Berkeley, CA, 14–16 Mar. 1962.
45. Varvill, R. and Bond, A. "A Comparison of Propulsion Concepts for SSTO Reusable Launchers," *Journal of the British Interplanetary Society (JBIS)*, vol. 56, pp. 108–117, 2003.
46. Timothy N. Titus, Hugh H. Kieffer, and Phillip R. Christensen, "Exposed Water Ice Discovered near the South Pole of Mars," *Science*, 14 Feb. 2003, vol. 299, pp. 1048–1051.
47. Sean C. Solomon, Oded Aharonson, Jonathan M. Aurnou, W. Bruce Banerdt, Michael H. Carr, Andrew J. Dombard, Herbert V. Frey, Matthew P. Golombek, Steven A. Hauck, II, James W. Head, III, Bruce M. Jakosky, Catherine L. Johnson, Patrick J. McGovern, Gregory A. Neumann, Roger J. Phillips, David E. Smith, and Maria T. Zuber, "New Perspectives on Ancient Mars," *Science*, 25 Feb. 2005, vol. 307, pp. 1214–1220.
48. Richard A. Kerr, "Voyager 1 Crosses a New Frontier and May Save Itself From Termination," *Science*, May 27, 2005, vol. 308, pp. 1237–1238.
49. Dana G. Andrews and Jason E. Andrews "Air Collection and Enrichment System (ACES) for Advanced 2nd Generation RLVs," AIAA-2001-3702, July 2001.
50. Crocker, D. Andrews, S. White and J. Andrews, "Progress on ACES: Technology for Next Generation Space Transportation," AIAA-2003-4890, July 2003
51. G. Sadler and W. Taylor, A. Crocker, J. Roche, A. Wuerl, "Gryphon: A Feasible Horizontal Takeoff Next Generation Architecture Concept," AIAA-2004-3391, July 2004
52. Patrick Biltgen, Jarret Lafleur, Josh Loughman, Robert Martin, Kevin Flaherty, Min Cho, Keith Becker, Chester Ong, John R. Olds "StarRunner: A Single-Stage-to-Orbit, Airbreathing, Hypersonic Propulsion System," AIAA-2004-3729, July 2004.
53. Palaszewski, B., "Electric Propulsion For Lunar Exploration And Lunar Base Development," *The Second Conference on Lunar Bases and Space Activities of the 21st Century*, vol. 1 pp. 35–45, Sept. 1, 1992.
54. Palaszewski, B., Brophy, J., and King, David, "Nuclear-Electric Propulsion—Manned Mars Propulsion Options," *The Case For Mars III: Strategies For Exploration—Technical*, San Diego, CA, Univelt, Inc., 1989, p. 431–451, January 1, 1989.
55. Swindle, T.D., Glass, C.E., and Poulton, M.M., 1990, "Mining Lunar Soils for 3He," UA/NASA Space Engineering Research Center TM-90/1 (Tucson: UA/NASA SERC).
56. Johnson, J.R.; Swindle, T.D.; Lucey, P.G., "Solar-Wind-Implanted Volatiles in the Lunar Regolith," *Workshop on New Views of the Moon 2: Understanding the Moon Through the Integration of Diverse Datasets/29–31*; LPI-Contrib-980, 1999, <http://www.lpi.usra.edu/meetings/moon99/pdf/program.pdf>

REPORT DOCUMENTATION PAGE

Form Approved
OMB No. 0704-0188

Public reporting burden for this collection of information is estimated to average 1 hour per response, including the time for reviewing instructions, searching existing data sources, gathering and maintaining the data needed, and completing and reviewing the collection of information. Send comments regarding this burden estimate or any other aspect of this collection of information, including suggestions for reducing this burden, to Washington Headquarters Services, Directorate for Information Operations and Reports, 1215 Jefferson Davis Highway, Suite 1204, Arlington, VA 22202-4302, and to the Office of Management and Budget, Paperwork Reduction Project (0704-0188), Washington, DC 20503.

1. AGENCY USE ONLY (<i>Leave blank</i>)	2. REPORT DATE October 2006	3. REPORT TYPE AND DATES COVERED Technical Memorandum	
4. TITLE AND SUBTITLE Atmospheric Mining in the Outer Solar System		5. FUNDING NUMBERS WBS- 22-708-03-03	
6. AUTHOR(S) Bryan Palaszewski			
7. PERFORMING ORGANIZATION NAME(S) AND ADDRESS(ES) National Aeronautics and Space Administration John H. Glenn Research Center at Lewis Field Cleveland, Ohio 44135-3191		8. PERFORMING ORGANIZATION REPORT NUMBER E-15457	
9. SPONSORING/MONITORING AGENCY NAME(S) AND ADDRESS(ES) National Aeronautics and Space Administration Washington, DC 20546-0001		10. SPONSORING/MONITORING AGENCY REPORT NUMBER NASA TM-2006-214122 AIAA-2005-4319	
11. SUPPLEMENTARY NOTES Prepared for the 41st Joint Propulsion Conference and Exhibit cosponsored by AIAA, ASME, SAE, and ASEE, Tucson, Arizona, July 10-13, 2005. Responsible person, Bryan Palaszewski, organization code RTB, 216-977-7493, e-mail: bryan.a.palaszewski@nasa.gov.			
12a. DISTRIBUTION/AVAILABILITY STATEMENT Unclassified - Unlimited Subject Categories: 91, 07, 20, 28, and 15 Available electronically at http://gltrs.grc.nasa.gov This publication is available from the NASA Center for AeroSpace Information, 301-621-0390.		12b. DISTRIBUTION CODE	
13. ABSTRACT (<i>Maximum 200 words</i>) Overviews, analyses, and assessments of various mining options for outer planet atmospheres are presented. Options for mining include aerostat borne factories that process the atmosphere and cruisers or scoopers that dip into or reside in the atmosphere. Assessments of the mission complexity for several mining scenarios are presented. Propulsion options for these scenarios are also assessed. Options for mining and transporting atmospheric constituents from the outer planet of our solar system were reviewed. Comparisons of the mining and transportation techniques were also assessed. Comparisons of the complexity of lunar helium 3 (3He) mining and mining in the outer solar system planetary atmospheres were conducted. Mission options that reduce the complexity of the transportation systems are suggested.			
14. SUBJECT TERMS In-situ resources utilization; Propellants; Metallized gelled fuels; Cryogenics; Nuclear power; Planetary exploration		15. NUMBER OF PAGES 22	
		16. PRICE CODE	
17. SECURITY CLASSIFICATION OF REPORT Unclassified	18. SECURITY CLASSIFICATION OF THIS PAGE Unclassified	19. SECURITY CLASSIFICATION OF ABSTRACT Unclassified	20. LIMITATION OF ABSTRACT

