

# Boundary Layer Flow past a Stretching Cylinder and Heat Transfer with Variable Thermal Conductivity

Rekha R. Rangi, Naseem Ahmad

Department of Mathematics, Jamia Millia Islamia, New Delhi, India

Email: rdrekhadahiya@gmail.com

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## ABSTRACT

The boundary layer flow of viscous incompressible fluid over a stretching cylinder has been considered to study flow field and temperature field. Due to non-linearity, a numerical approach called Keller-box technique has been used to compute the values of velocity function  $f$  and temperature field at different points of dynamic region. The expressions for skin friction and Nusselt number have also been obtained. The dependence of velocity profile and temperature profile on the dimensionless parameter of practical interest has been analyzed in detail by graphs. The dependence of Skin friction and Nusselt number has been seen through tables.

**Keywords:** Boundary Layer Equation; Keller-Box Method; Energy Equation; Variable Thermal Conductivity; Prandtl Number

## 1. Introduction

The boundary layer flow and heat transfer due to stretching flat plates or cylinders are of practical importance in fiber technology and extrusion processes, and of theoretical interest as well. The production of polymer sheets and plastic films is based on this technology. There are number of examples which include the cooling of an infinite metallic plate in a cooling bath, the boundary layer along material handling conveyers, the aerodynamic extrusion of plastic sheets, the boundary layer along a liquid film in condensation processes, paper production, glass blowing, metal spinning and drawing plastic films, and polymer extrusion. The quality of the final product depends on the rate of heat transfer at the stretching surface. Sakiadis [1] was the first to consider the boundary layer flow on a moving continuous solid surface. Crane [2] extended this concept to a stretching sheet with linearly varying surface speed and presented an exact solution for the steady two-dimensional flow over stretching surface in a quiescent fluid. Since then, many authors have considered various aspects of this problem and obtained similarity solutions. A similarity solution is one in which the number of independent variables is reduced by at least one, usually by a coordinate transformation. The idea is analogous to dimensional analysis, but instead of parameters, like the Reynolds number, the coordinates themselves are collapsed into dimensionless groups that scale the velocities [3].

The boundary layer flow due to a stretching vertical

surface in a quiescent viscous and incompressible fluid when the buoyancy forces are taken into consideration have been considered in the papers [4-11]. Lin and Shih [12,13] considered the laminar boundary layer and heat transfer along horizontally and vertically moving cylinders with constant velocity and found that the similarity solutions could not be obtained due to the curvature effect of the cylinder. In the present paper, we find velocity and temperature field for boundary layer flow past a stretching cylinder where cylinder is continuously moving with  $U(x) = U_0(x/l)$  velocity. Due to nonlinearity of the problem, we use the numerical approach called Keller-box technique [14]. Since the lateral surface of cylinder changes according to curvature parameter  $\gamma = 0.0, 0.25, 0.5, 0.75, 1$ , so we have studied the influence of  $\gamma$  on velocity and temperature fields. The expressions of skin friction  $C_f$  and Nusselt number  $Nu_x$  have been obtained. They have been discussed in respect to pertinent parameters like prandtl number  $Pr$ , curvature parameter  $\gamma$  and thermal variable parameter  $\varepsilon$ .

## 2. Mathematical Formulation of the Problem

Consider a steady, axisymmetric boundary layer flow of a viscous incompressible fluid along a continuously stretching cylinder as shown in **Figure 1**. It is assumed that the stretching velocity  $U(x)$  is of the form  $U(x) = U_0(x/l)$  where  $U_0$  is positive constant and  $l$  is characteristic length. Under these assumptions along with the boundary layer approximations, the equations which

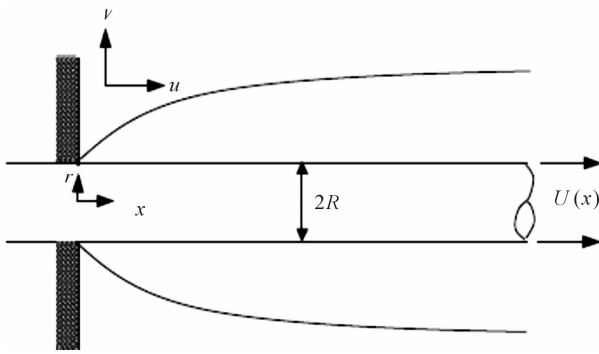


Figure 1. Physical model and coordinate system.

model the problem under consideration are

$$\frac{\partial}{\partial x}(ru) + \frac{\partial}{\partial r}(rv) = 0 \tag{2.1}$$

$$u \frac{\partial u}{\partial x} + v \frac{\partial u}{\partial r} = \frac{v}{r} \frac{\partial}{\partial r} \left( r \frac{\partial u}{\partial r} \right) \tag{2.2}$$

together with boundary conditions

$$\begin{aligned} u = U(x), v = 0 \text{ at } r = R, \\ u \rightarrow 0 \text{ as } r \rightarrow \infty \end{aligned} \tag{2.3}$$

where  $u$  and  $v$  are the velocity components in the  $x$ - and  $r$ -directions, respectively. The continuity Equation (2.1) can be satisfied by introducing a stream function  $\psi$ , such that  $u = \frac{1}{r} \frac{\partial \psi}{\partial r}$  and  $v = -\frac{1}{r} \frac{\partial \psi}{\partial x}$ . Following [15,16], we define  $\eta$  and  $\psi$  as

$$\eta = \frac{r^2 - R^2}{2R} \sqrt{\frac{U}{\nu x}}, \quad \psi = \sqrt{U \nu x} R f(\eta) \tag{2.4}$$

so that the momentum equation becomes

$$(1 + 2\gamma\eta) f''' + 2\gamma f'' + ff'' - f'^2 = 0 \tag{2.5}$$

together with boundary conditions

$$f(0) = 0, f'(0) = 1, f'(\infty) \rightarrow 0 \tag{2.6}$$

The Equations (2.5) and (2.6) constitute a non-linear boundary-value problem in infinite domain  $[0, \infty)$ . Since no prescribed methods are available to deal with non-linear boundary equations that to be in infinite domain, so we solve it by numerical method called Keller-box method for different curvature parameter  $\gamma$ .

### 3. Skin Friction Coefficient

Surface shear stress is calculated by

$$\tau_w = \mu \left( \frac{\partial u}{\partial r} \right)_{r=R} = \mu U \sqrt{\frac{U}{\nu x}} f''(0) \tag{3.1}$$

Therefore, the skin friction coefficient is given by

$$C_f = \frac{\tau_w}{\rho U^2 / 2} \Rightarrow \frac{1}{2} C_f \text{Re}_x^{1/2} = f''(0) \tag{3.2}$$

### 4. Heat Transfer Problem

The heat flow problem for the stretching cylinder to the fluid in the absence of energy dissipation, is governed by

$$u \frac{\partial T}{\partial x} + v \frac{\partial T}{\partial r} = \frac{1}{r} \frac{\partial}{\partial r} \left( \alpha r \frac{\partial T}{\partial r} \right) \tag{4.1}$$

with boundary conditions

$$\begin{aligned} T = T_p, r = R \\ T \rightarrow T_\infty, r \rightarrow \infty \end{aligned} \tag{4.2}$$

Defining the dimensionless temperature by

$$\theta = \frac{T - T_\infty}{T_p - T_\infty}$$

It has been observed that for liquid metals the thermal conductivity  $\alpha$  varies with temperature in an approximately linear manner in the range from  $0^\circ\text{F}$  to  $400^\circ\text{F}$ . Therefore, we assume  $\alpha$  as  $\alpha = \alpha_\infty (1 + \varepsilon\theta)$

Now, the Equation (4.1) reduces to

$$\begin{aligned} (1 + 2\gamma\eta) \theta'' + 2\gamma\theta' + Pr f \theta' \\ + \varepsilon \left[ (1 + 2\gamma\eta) (\theta'^2 + \theta\theta'') + 2\gamma\theta\theta' \right] = 0 \end{aligned} \tag{4.3}$$

Equation (4.3) again constitutes a non-linear boundary value problem involving velocity function  $f$ . We again use Keller-box method for different values of flow parameter  $\varepsilon$ ,  $\gamma$  and  $Pr$ .

### 5. Local Nusselt Number

The energy is transit by virtue of temperature gradient is given by

$$q_w = -k \left( \frac{\partial T}{\partial r} \right)_{r=R} = -k(T_w - T_\infty) \sqrt{\frac{U}{\nu x}} \theta'(0) \tag{5.1}$$

Hence, nusselt number is calculated as

$$Nu_x = \frac{xq_w}{k(T_w - T_\infty)} \Rightarrow Nu_x / \text{Re}_x^{1/2} = -\theta'(0) \tag{5.2}$$

where  $\text{Re}_x = Ux/\nu$  is the local Reynolds number.

### 6. Discussions and Results

In the course of analysis of the problem of boundary layer flow past a stretching cylinder and heat transfer with variable thermal conductivity, we come across the following results:

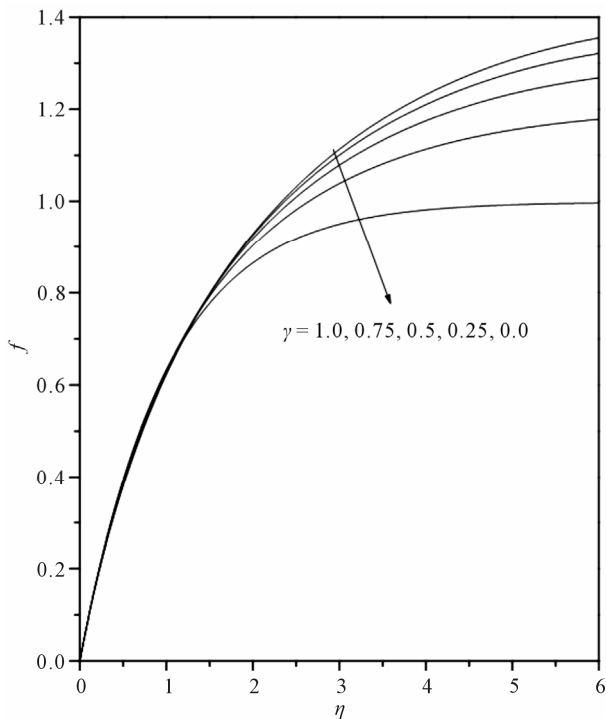
The Equation (2.5) together with boundary conditions (2.6) constitutes a non-linear boundary value problem of order three. We use Keller-box [16] technique of implicit

finite difference to find  $f(\eta)$  and  $f'(\eta)$  so that the velocity components of viscous incompressible fluid over stretching cylinder may be obtained. Referring **Figure 2**, we see that the transverse velocity component has not been influenced by curvature parameter  $\gamma$  within the dynamic region  $[0, 1.25]$ , thereafter, as curvature of stretching cylinder decreases, this velocity component decreases. Physically, as  $\gamma \rightarrow 0$ , the outer surface of cylinder behaves like a flat surface. It means as  $\gamma \rightarrow 1.0$ , the viscosity effect reduces due to contact area of surface with fluid tends to the tangential position.

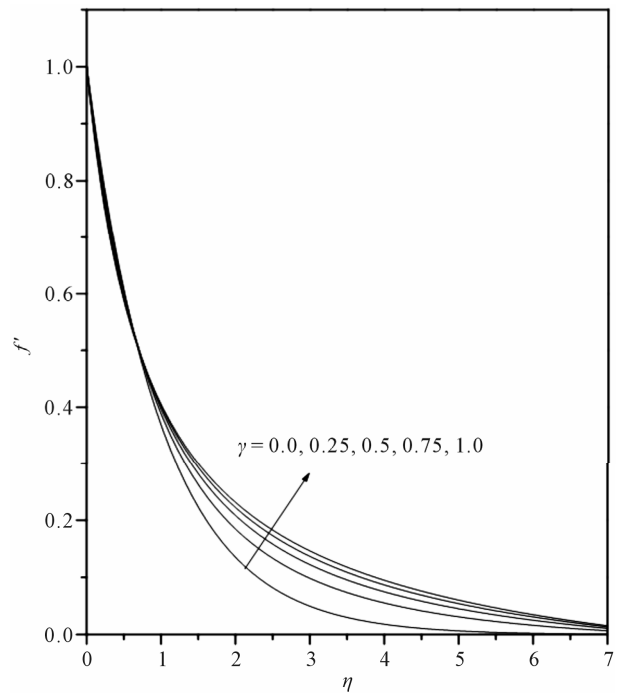
Looking at the **Figure 3**, we see that the effect of curvature parameter on horizontal component of velocity field is almost nil within the dynamic region  $[0, 0.75]$ . This velocity component approaches to zero asymptotically within the region  $[0.75, \infty)$ . In this case, the velocity within  $[0.75, \infty)$  is free stream velocity and in this region, as  $\gamma$  increases, the velocity component increases.

Reading the **Table 1**, we see that  $C_f \propto Re_x^{-1/2}$ . Thus, as Reynolds number increases  $|C_f|$  decreases. It is well known fact that as  $Re_x$  increases, the viscous forces decreases, in turn  $|C_f|$  decreases. In the present case,  $C_f$  depends on curvature parameter  $\gamma$ . We see that as  $\gamma$  increases  $\frac{1}{2}|C_f Re_x|$  increases which means as  $\gamma$  increases, the viscous forces start reducing.

Reading the **Figures 4 and 5**, we see the trend of the behavior of temperature profile. According to **Figure 4**, where  $Pr = 1.0$ , we note the following:



**Figure 2.** Transverse velocity profiles for different values of  $\gamma$ .



**Figure 3.** Horizontal velocity profiles for different values of  $\gamma$ .

**Table 1.** The variation of  $\frac{1}{2}C_f\sqrt{Re_x}$  with respect to curvature parameter  $\gamma$ .

	0.0	0.25	0.5	0.75	1.0
$f''(0)$	-1.0000	-1.094378	-1.188715	-1.281833	-1.459308

1) In case constant thermal conductivity  $\varepsilon = 0$ , curvature parameter  $\gamma$  has no effects on temperature field within dynamic region  $[0, 0.7]$ . As the flow boundary layer is thinner than the thermal boundary layer, therefore, this dynamic region is sub interval of flow dynamic region. In free stream, as curvature increases, the heat transfer from surface to fluid enhances.

2) When thermal conductivity is variable, that is  $\varepsilon = 0.2$ , we see that there is no region which has not been affected by curvature parameter  $\gamma$ . In this case, as  $\gamma$  increases, the heat transfers becomes fast from surface to fluid too.

3) If further  $\varepsilon = 0.4$ , transfer of heat is in better position than the case where  $\varepsilon = 0.2$ .

The **Figure 5** shows the pattern of temperature field variation when the thermal conductivity parameter  $\varepsilon = 0.1$ . We see that in the case  $Pr = 1.0$ , we see that curvature parameter  $\gamma$  does not having any effect on heat transfer within dynamic region  $[0, 0.75]$  while for  $Pr = 2.0$  and  $Pr = 3.0$ , the curvature parameter influences the heat transfer everywhere. The enhancement in  $\gamma$  accelerates the heat transfer in both the cases  $Pr = 0.2$  and  $Pr = 0.4$ .

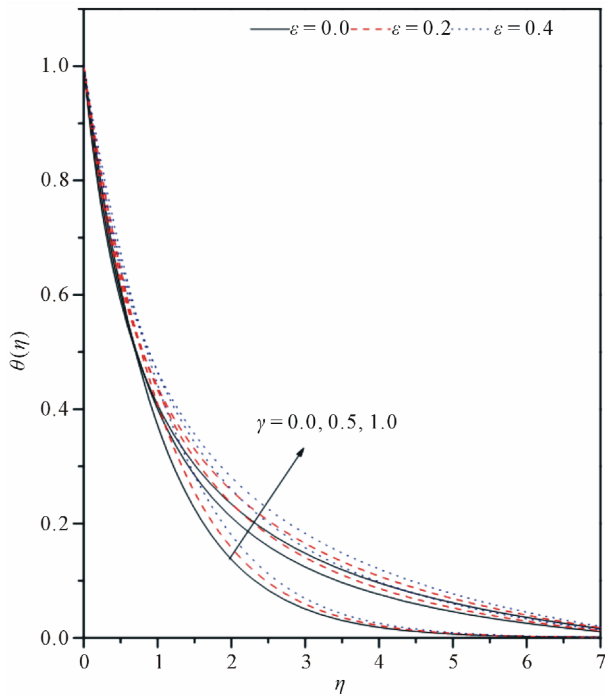


Figure 4. Temperature profiles for different values of  $\gamma$  and  $\epsilon$  with  $Pr = 1.0$ .

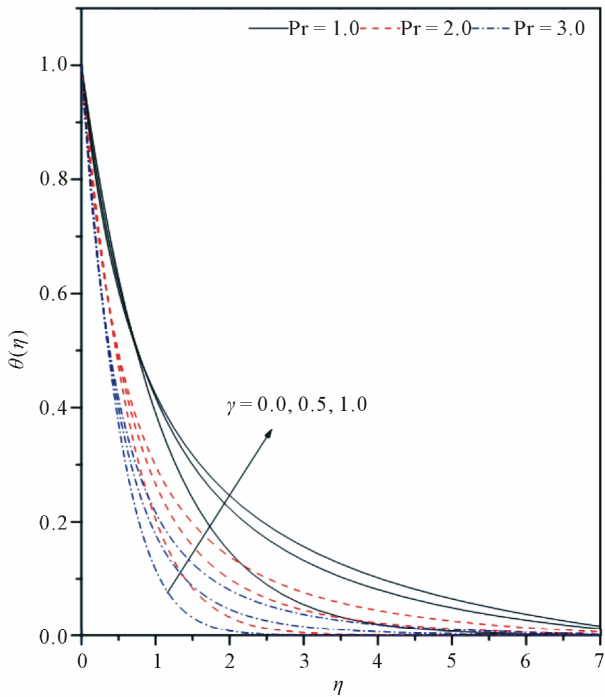


Figure 5. Temperature profiles for different values of  $\gamma$  and  $Pr$  with  $\epsilon = 0.1$ .

The Equation (5.2) establishes the relation between the coefficient of convective heat transfer, *i.e.* Nusselt number  $Nu_x$  and Reynolds number. We see that  $Nu_x^2 \propto Re_x$ . Further, we conclude that as Reynolds number increases,

Table 2. Temperature gradient  $-\theta'(0)$  at the outer surface of cylinder for different values of pertinent parameters.

Pr	$\gamma$	$\epsilon = 0.0$	$\epsilon = 0.2$	$\epsilon = 0.4$
1.0	0.0	-0.985286	-0.862122	-0.770792
	0.25	-1.079447	-0.949659	-0.853754
	0.5	-1.173899	-1.037605	-0.937245
	0.75	-1.267214	-1.124397	-1.09536
0.1	1.0	-1.359308	-1.209949	-1.100545
	0.0	-0.918691	-1.403426	-1.774590
	0.25	-1.000921	-1.490347	-1.862539
	0.5	-1.100088	-1.574003	-1.946728
0.1	0.75	-1.189821	-1.656050	-2.028463
	1.0	-1.278325	-1.736990	-2.108424

*i.e.*  $Re_x = \frac{Ux}{\nu}$  increases when viscosity decreases, so

the decrease in viscosity enhances the magnitude of rate of convective heat transfer in case of stretching cylinder. Also, the coefficient of convective heat depends on Prandtl number  $Pr$ , curvature parameter  $\gamma$  and thermal variable parameter  $\epsilon$ . The behavior of the coefficient of convective heat transfer may be studied from Table 2 with respect to pertinent parameters.

### 7. Conclusion

The curvature of stretching cylinder is very vital parameter affecting flow and temperature field both. In case of thermal conductivity, curvature helps to enhance the heat transfer. The reduction in fluid viscosity accelerates the rate of convective heat transfer.

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