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Dynamic pull-in instability of geometrically nonlinear actuated micro-beams based on the modified couple stress theory

Abstract

This paper investigates the dynamic pull-in instability of vibrating micro-beams undergoing large deflection under electrosatically actuation. The governing equation of motion is derived based on the modified couple stress theory. Homotopy Perturbation Method is employed to produce the high accuracy approximate solution as well as the second-order frequency- amplitude relationship. The nonlinear governing equation of micro beam vibrations predeformed by an electric field includes both even and odd nonlinearities. The influences of basic non-dimensional parameters on the pull-in instability as well as the natural frequency are studied. It is demonstrated that two terms in series expansions are sufficient to produce high accuracy solution of the micro-structure. The accuracy of proposed asymptotic approach is validated via numerical results. The phase portrait of the system exhibits periodic and homoclinic orbits.

Keywords

Dynamic pull-in instability, Modified couple stress theory, Homotopy Perturbation Method, Frequency – amplitude relationship, Homoclinic orbit.

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1 INTRODUCTION

With development of modern technology, micro and nano-electro mechanical systems have shown enormous popularity on engineering and industry. The application of micro-electro-mechanical systems (MEMS) devices especially the electrically actuated MEMS devices which require low actuation voltage levels are continuously growing. Actuated MEMS devices are extensively used in inkjet printers, switches, gyroscopes, chemo-sensors and so on. Recently several numerical and experimental studies have been conducted on the pull-in instability and dynamic behavior of MEMS devices (Ansari et al., 2013; Zhang and Fu, 2012; Rajabi and Ramezani, 2013; Nayfeh et al., 2005; He et al., 2009; Sedighi and Shirazi, 2013; Jia et al., 2011; Younis and Nayfeh, 2003; Mobki et al., 2013; Rahaeifard et al., 2013; Caruntu et al., 2013; Batra et al., 2008; Moghimi Zand and Ahmadian,

2009; Nabian et al., 2013). However, the amplitude dependence of nonlinear frequency and pull-in instability has not been developed, till present.

A distributed size-dependent model based on the modified strain gradient elasticity theory (MSGT) was developed by Ansari et al. (2013)in order to investigate the pull-in instability of circular microplates subjected to the uniform hydrostatic and non-uniform electrostatic actuations. They employed step by step linearization of the equation of motion and utilized the generalized differential quadrature (GDQ) method to solve the problem numerically. A new model for a viscoelastic beam based on the simplified couple stress theory was developed by Zhang and Fu(2012). They investigated the effect of the beam size on the instantaneous pull-in voltage, durable pull-in voltage and pull-in delay time of the system. Rajabi and Ramezani (2013)studied the dynamic behaviour of the micro scale nonlinear beam model based on strain gradient elasticity. They demonstrated that by increasing the beam thickness, the strain gradient effect on increasing the natural frequency decreases and geometric nonlinearity plays the main role on its trend. Nayfeh et al. (2005) reviewed the development of reduced-order models (node and domain methods) for MEMS devices and discussed the advantages and disadvantages of each implementation.

The improved macromodel of the fixed-fixed microbeam-based of MEMS capacitive switch was developed by He et al. (2009) to predict the electromechanical behaviors of electrically actuated MEMS capacitive switch. Their model accounted for moderately large deflections, dynamic loads, axial stress induced by the midplane stretching and the residual stress. Sedighi and Shirazi (2013) presented a new asymptotic procedure to predict the nonlinear vibrational behaviour of classical micro-beams pre-deformed by an electric field using Parameter expansion method. Jia et al. (2011) investigated the pull-in instability of micro-switches under the combined electrostatic and intermolecular forces. They accounted for the effect of axial residual stress, the force nonlinearity and geometric nonlinearity in their research and solved the governing equation using the differential quadrature method.

An investigation into the response of a resonant microbeam to an electric actuation was presented by Younis and Nayfeh (2003). They employed the nonlinear model to account for the mid-plane stretching, a DC electrostatic force, and an AC harmonic force and discussed the effect of the design parameters on the dynamic response of the micro-structure. Mobki et al. (2013) studied the mechanical and bifurcation behavior of a capacitive micro-beam suspended between two conductive stationary plates. They used a modified non-linear mass-spring model in order to study the global stability of the fixed points and showed the homoclinic and heteroclinic orbits by plotting phase plane diagrams. Rahaeifard et al. (2013) investigated the dynamic behavior of micro-cantilevers under suddenly applied DC voltage based on the modified couple stress theory. They utilized multiple scales method for analytical analysis and their numerical approach was based on a hybrid finite element/finite difference method. Caruntu et al. (2013)employed the reduced order model (ROM) method to investigate the nonlinear parametric dynamics of electrostatically actuated microelectromechanical systems (MEMS) cantilever resonators under soft alternating current (AC) voltage of frequency near half natural frequency. They found that the fringe effect affects significantly the behavior of the MEMS resonator especially for the narrower micro-cantilevers. Batra et al. (2008) studied the vibrations of narrow microbeams pre-deformed by an electric field with the electromechanical model that estimated the electrostatic fringing field due to both the finite width and the finite thickness of the microbeam. Moghimi Zand et al (2009) studied the dynamic pull-in instability of microbeams subjected to step voltages using homotopy analysis method under the consideration of electrostatic force and midplane stretching. Nabian et al. (2013) examined the stability of a functionally graded clamped-clamped micro-plate subjected to hydrostatic and electrostatic pressures and illustrated that the micro-system undergoes a saddle node and homoclinic bifurcations.

The current work intends to compute the second-order frequency-amplitude relation in a micro clamped-clamped beam due to electrostatic voltage. Recently, considerable progresses had been made in asymptotic approximate solutions of nonlinear differential equations (Sedighi et al., 2013; Sedighi and Shirazi, 2012; Sedighi et al., 2012). There have been several approaches employed to solve the governing nonlinear differential equations to study the nonlinear vibrations such as Parametrized Perturbation Method (PPM) (Barari et al., 2011), Energy Balance Method (Ghadimi et al., 2012), Variational Iteration Method and Hamiltonian Approach (HA)(Sedighi et al., 2012), Laplace Transform Method (Rafieipour et al., 2012), Max-Min Approach (He, 2008), Homotopy Analysis Method (HAM)(Sedighi et al., 2012), Parameter Expansion Method (Sedighi et al., 2012), Iteration Perturbation Method (IPM) (He, 2001) and Homotopy Perturbation Method (HPM) (He, 1999). It is well known that while the perturbation methods provide the most versatile tools for the nonlinear analysis of engineering problems, they have also some limitations. In order to overcome these drawbacks, combining the standard homotopy and the perturbation method, known as the Homotopy Perturbation Method (HPM), improves the drawbacks of both approaches. He (1999) developed the homotopy perturbation method for solving a variety of problems including the linear and the nonlinear as well as the initial and the boundary value problems by merging two aforementioned techniques. Benefiting from easily computable components and rapid convergence, it has been applied to a wide class of functional equations.

The aim of the present article is to investigate the dynamic pull-in instability of geometrically nonlinear actuated micro-beams by introducing the second-order frequency amplitude relation. The nonlinear equation of motion is derived based on the modified couple stress theory using Hamilton's principle. The effect of vibrational amplitude and system parameters on the pull-in instability and natural frequency is studied via HPM. In this direction, analytical expressions for vibrational response of actuated micro-beams are presented. The proposed analytical method demonstrates that two terms in series expansions is sufficient to obtain a highly accurate solution of micro-beam vibration.

2 MATHEMATICAL MODELING

Consider a double-clamped micro-beam suspended above a rigid plate and under electro-statically actuation voltage as shown in Fig. 1.The actuated micro-beam has length L, cross section area A_0 , height h, width b, density r, moment of inertia I and modulus of elasticity E. The air initial gap is d_{gap} and an attractive electrostatic force which originates from voltage V causes the micro-beam to deflect. Assume that the micro-beam considered here, be the Euler–Bernoulli beam. The strain of a material point located at a distance z from the middle plane for the micro-beam caused by the large rotation and large displacement of the cross-section is represented as (Sedighi et al., 2012):

$$\varepsilon = -z \frac{d\theta}{dx} = -z \left[w_{xx} \left(1 + w_x^2 \right)^{-3/2} \right] = -z \left[w_{xx} - \frac{3}{2} w_{xx} w_x^2 \right]$$
 (1)

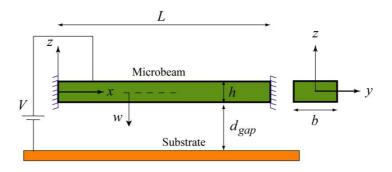


Figure 1 Configuration of a clamped-clamped actuated micro-beam

Taking into account the linear relation between the stress and strain, we have $\sigma = E\varepsilon$, The strain energy of the micro-beam can be calculated from:

$$U = \frac{1}{2E} \int_{V} \sigma^{2} dV = \frac{E}{2} \int_{V} \left(-z \left(w_{xx} - \frac{3}{2} w_{xx} w_{x}^{2} \right) \right)^{2} dV$$
 (2)

Furthermore, the strain energy of an elastic continuum medium using the modified couple stress theory (MCST) can be mentioned as (Rahaeifard et al., 2013):

$$U_{MCST} = \frac{1}{2} \int_{V} \mu l^2 w_{xx}^2 \, dV = \frac{1}{2} \int_{0}^{L} \mu A_0 l^2 w_{xx}^2 \, dx \tag{3}$$

where μ represents the shear modulus and l denotes the material length scale parameter, using the descriptions:

$$A_0 = \int_A dA, \ I = \int_A z^2 \, dA \tag{4}$$

The total strain energy of the micro-structure incorporating the MCST micro-beam model can be expressed as:

$$U_{tot} = \frac{EI}{2} \int_{0}^{L} \left[w_{xx} - \frac{3}{2} w_{xx} w_{x}^{2} \right]^{2} dx + \frac{\mu A_{0} l^{2}}{2} \int_{0}^{L} w_{xx}^{2} dx$$
 (5)

The virtual work W performed by the axial and electrical forces incorporating the von Karman type nonlinear strain can be written as (Batra et al., 2008):

$$W = -\frac{1}{2} \int_{0}^{L} \left[N_{i} + \frac{EA}{2L} \int_{0}^{L} w_{x}^{2} dx \right] w_{x}^{2} dx - \int_{0}^{L} F_{es} w dx$$
 (6)

where N_i is the axial force and F_{es} is the electrostatic force per unit length of the micro beam which is described by (Moghimi Zand et al, 2009):

$$F_{es} = \frac{b\varepsilon V^2}{2(d_{gap} - w)^2} \left(1 + \beta \frac{d_{gap} - w}{b}\right) \tag{7}$$

where b = 0.65 for double-clamped micro-beam. The kinetic energy is obtained as follows

$$T = \frac{\rho A_0}{2} \int_0^L w_t^2 dx \tag{8}$$

Applying the Hamilton's principle leads to:

$$\delta \left(\int_{t_1}^{t_2} \left(T - U_{tot} + W \right) dt \right) = 0 \tag{9}$$

After some mathematical computations, the nonlinear governing equation of motion for actuated MCST micro-beam model is expressed as follows:

$$\rho A_0 w_{tt} + EI \left(w_{xxxx} - 3w_{xxx}^2 w_x^2 - 3w_{xx}^3 - 12w_{xx} w_x w_{xxx} \right) + \mu A_0 l^2 w_{xxxx} - \left(N_i + \frac{EA_0}{2L} \int\limits_0^L w_x^2 dx \right) w_{xx} - F_{es} = 0 \tag{10}$$

By introducing the dimensionless parameters as:

$$\tau = \sqrt{\frac{EI}{\rho b h L^4}} t, \quad W = \frac{w}{d_{gap}}, \quad \xi = \frac{x}{L}, \quad \alpha = 6 \left(\frac{d_{gap}}{h}\right)^2, \quad \overline{V}^2 = \frac{24\varepsilon L^4 V^2}{Eh^3 d_{gap}^3}, \quad f_i = \frac{N_i L^2}{EI},$$

$$\gamma = \frac{d_{gap}}{b}, \quad \mu_s = \frac{12\mu}{E(h/l)^2}, \quad \kappa = \left(\frac{d_{gap}}{L}\right)^2$$

$$(11)$$

the non-dimensional nonlinear equation of the micro-beam based on the MCST model can be written as follows:

$$\begin{split} &\frac{\partial^{2}W}{\partial\tau^{2}} + \left[1 + \mu_{s}\right] \frac{\partial^{4}W}{\partial\xi^{4}} - 3\kappa \left(\frac{\partial^{4}W}{\partial\xi^{4}} \left(\frac{\partial W}{\partial\xi}\right)^{2} + \left(\frac{\partial^{2}W}{\partial\xi^{2}}\right)^{3} + 4\frac{\partial^{2}W}{\partial\xi^{2}} \frac{\partial^{3}W}{\partial\xi^{3}} \frac{\partial W}{\partial\xi}\right) \\ &- \left[f_{i} + \alpha \int_{0}^{1} \left(\frac{\partial W}{\partial\xi}\right)^{2} d\xi\right] \frac{\partial^{2}W}{\partial\xi^{2}} - \frac{\overline{V}^{2}}{4\left(1 - W\right)^{2}} \left(1 + \gamma\beta\left(1 - W\right)\right) = 0 \end{split} \tag{12}$$

Using Taylor expansion for electrostatic force in equation (12) results in

$$\begin{split} &\frac{\partial^{2}W}{\partial\tau^{2}} + \left[1 + \mu_{s}\right] \frac{\partial^{4}W}{\partial\xi^{4}} - 3\kappa \left(\frac{\partial^{4}W}{\partial\xi^{4}} \left(\frac{\partial W}{\partial\xi}\right)^{2} + \left(\frac{\partial^{2}W}{\partial\xi^{2}}\right)^{3} + 4\frac{\partial^{2}W}{\partial\xi^{2}} \frac{\partial^{3}W}{\partial\xi^{3}} \frac{\partial W}{\partial\xi}\right) \\ &- \left[f_{i} + \alpha \int_{0}^{1} \left(\frac{\partial W}{\partial\xi}\right)^{2} d\xi\right] \frac{\partial^{2}W}{\partial\xi^{2}} - \frac{\overline{V}^{2}}{4} \sum_{n=0}^{\infty} \left(n + 1 + \gamma\beta\right) W^{n} = 0 \end{split} \tag{13}$$

Assuming $W(\xi,\tau) = q(\tau)\phi(\xi)$, where $\phi(\xi)$ is the first eigen mode of the clamped-clamped beam and can be expressed as:

$$f(x) = \cosh(l_{cc}x) - \cos(l_{cc}x) - \frac{\cosh(l_{cc}) - \cos(l_{cc})}{\sinh(l_{cc}) - \sin(l_{cc})} \left(\sinh(l_{cc}x) - \sin(l_{cc}x)\right)$$

$$(14)$$

where $\lambda_{cc} = 4.73$ is the root of characteristic equation for first eigen mode. Applying the Bubnov-Galerkin decomposition method, the non-dimensional nonlinear governing equation of motion can be written as follows:

$$\frac{d^2q}{d\tau^2} + \beta_1 q(\tau) + \left[\beta_2 \left(q(\tau)\right)^2 + \beta_3 \left(q(\tau)\right)^3 + \beta_4 \left(q(\tau)\right)^4 + \beta_0\right] = 0 \tag{15}$$

where the parameters $\beta_0,...,\beta_4$ have been described in the Appendix.

3 BASIC IDEA OF HOMOTOPY PERTURBATION METHOD

Consider the following nonlinear differential equation (He, 1999):

$$A(u) - f(r) = 0, \ r \in \Omega \tag{16}$$

subjected to the following boundary condition:

$$B\left(u, \frac{\partial u}{\partial t}\right) = 0 \quad r \in \Gamma \tag{17}$$

where A is a general differential operator, B a boundary operator, f(r) is a known analytical function, Γ is the boundary of the solution domain (Ω) , and $\partial u/\partial t$ denotes differentiation along the outwards normal to Γ . Generally, the operator A may be divided into two parts: a linear part L and a nonlinear one N. Therefore, Eq. (16) may be rewritten as follows:

$$L(x) + N(x) - f(r) = 0 (18)$$

In cases where the nonlinear Eq. (16) includes no small parameter, one may construct the following homotopy equation

$$H(\nu, p) = (1 - p)[L(\nu) - L(u_0)] + p[A(\nu) - f(r)] = 0$$

$$\tag{19}$$

In Eq. (19), $p \in [0,1]$ is an embedding parameter and u_0 is the first approximation that satisfies the boundary condition. One may assume that solution of Eq. (19) may be written as a power series in p, as the following:

$$\nu = \nu_0 + p\nu_1 + p^2\nu_2 + \dots \tag{20}$$

The homotopy parameter p is also used to expand the square of the unknown angular frequency as follows:

$$\omega_0 = \omega^2 - p\omega_1 - p^2\omega_2 - \dots \tag{21}$$

where ω_0 is the coefficient of u(r) in Eq. (16) and should be substituted for the right hand side of Eq. (19). Besides, $\omega_i (i=1,2,3,...)$ are unknown parameters. The best approximations for the solution and the angular frequency ω are

$$u = \lim_{p \to 1} \nu = \nu_0 + \nu_1 + \nu_2 + \dots \tag{22}$$

$$\omega^2 = \omega_0 + \omega_1 + \omega_2 + \dots \tag{23}$$

Now we apply the homotopy perturbation method on Eq. (15). We construct a homotopy in the following form:

$$H(q,p) = (1-p)[\ddot{q} + \beta_1 q] + p[\ddot{q} + \beta_1 q + \beta_2 q^2 + \beta_3 q^3 + \beta_4 q^4 + \beta_0] = 0$$
(24)

According to the HPM, we assume that the solution of Eq. (24) can be expressed in a series of p;

$$q\left(\tau\right) = q_{\scriptscriptstyle 0}\left(\tau\right) + pq_{\scriptscriptstyle 1}\left(\tau\right) + p^2q_{\scriptscriptstyle 2}\left(\tau\right) + \dots \tag{25}$$

the coefficient of q is expanded into a series in p in a similar way (He, 1999):

$$\beta_1 = \omega^2 - p\omega_1 - p^2\omega_2 + \dots {26}$$

After substituting Eq. (26) and Eq. (25) into Eq. (24), and rearranging based on powers of p-terms, we have:

$$p^{0}: \ \ddot{q}_{_{0}}\left(\tau\right)+\omega^{2}q_{_{0}}\left(\tau\right)=0, \quad \ q_{_{0}}\left(0\right)=A, \quad \dot{q}_{_{0}}\left(0\right)=0 \tag{27-a}$$

$$p^{1}: \ \ddot{q}_{1}\left(\tau\right) + \omega^{2}q_{1}\left(\tau\right) = \omega_{1}q_{0}\left(\tau\right) - \left[\beta_{2}\left(q_{0}\left(\tau\right)\right)^{2} + \beta_{3}\left(q_{0}\left(\tau\right)\right)^{3} + \beta_{4}\left(q_{0}\left(\tau\right)\right)^{4} + \beta_{0}\right], \ \ q_{1}\left(0\right) = 0, \ \ \dot{q}_{1}\left(0\right) = 0 \tag{27-b}$$

$$p^{2}: \ddot{q}_{2}(\tau) + \omega^{2}q_{2}(\tau) = \omega_{1}q_{1}(\tau) + \omega_{2}q_{0}(\tau) - \left[2\beta_{2}q_{0}(\tau)q_{1}(\tau) + 3\beta_{3}(q_{0}(\tau))^{2}q_{1}(\tau) + 4\beta_{4}(q_{0}(\tau))^{3}q_{1}(\tau)\right] - \left[\beta_{2}(q_{0}(\tau))^{2} + \beta_{3}(q_{0}(\tau))^{3} + \beta_{4}(q_{0}(\tau))^{4} + \beta_{0}\right], \quad q_{2}(0) = 0, \quad \dot{q}_{2}(0) = 0$$

$$(27-c)$$

Since the solution of Eq. (27-a) is $q_0 = A\cos(\omega\tau)$, the solution of Eq. (27-b) should not contain the so-called secular term $\cos(\omega\tau)$. Substitution of this result into the right-hand side of equation (27-b) yields:

$$\begin{split} \ddot{q}_{_{1}}\left(\tau\right) + \omega^{2}q_{_{1}}\left(\tau\right) &= \left(\omega_{_{1}}A - \frac{3}{4}\beta_{_{3}}A^{_{3}}\right)\cos\left(\omega\tau\right) + \left(-\frac{1}{2}\beta_{_{4}}A^{_{4}} - \frac{1}{2}\beta_{_{2}}A^{_{2}}\right)\cos\left(2\omega\tau\right) \\ &- \frac{1}{2}\beta_{_{2}}A^{_{2}} - \frac{3}{8}\beta_{_{4}}A^{_{4}} - \frac{1}{4}\beta_{_{3}}A^{_{3}}\cos\left(3\omega\tau\right) - \beta_{_{0}} - \frac{1}{8}\beta_{_{4}}A^{_{4}}\cos\left(4\omega\tau\right), \end{split} \tag{28}$$

No secular terms in $q_1(\tau)$ require eliminating contributions proportional to $\cos(wt)$ on the right-hand side of equation (28), we have:

$$\omega_{\scriptscriptstyle 1} = \frac{3}{4}\beta_{\scriptscriptstyle 3}A^2 \tag{29}$$

Solving equation (28) for $q_1(t)$ gives the following second order approximation for $q(\tau)$ as:

$$\begin{split} q_{1}\left(\tau\right) &= \frac{\cos\left(\omega\tau\right)\!\left(48\beta_{4}A^{4} + 160\beta_{2}A^{2} - 15\beta_{3}A^{3} + 480\beta_{0}\right)}{480\omega^{2}} + \frac{\cos\left(2\omega\tau\right)\!\left(80\beta_{4}A^{4} + 80\beta_{2}A^{2}\right)}{480\omega^{2}} \\ &+ \frac{\beta_{3}A^{3}\cos\left(3\omega\tau\right)}{32\omega^{2}} + \frac{\beta_{4}A^{4}\cos\left(4\omega\tau\right)}{120\omega^{2}} + \frac{-480\beta_{0} - 180\beta_{4}A^{4} - 240\beta_{2}A^{2}}{480\omega^{2}} \end{split} \tag{30}$$

equation (26) for two terms approximation of series respect to p and for p=1 yields:

$$\omega_2 = \omega^2 - \beta_1 - \omega_1 \tag{31}$$

substitution of this result into the right-hand side of equation (27-c) for $q_2(\tau)$ and eliminating the secular terms proportional to $\cos(\omega \tau)$ results in:

$$S(\omega) = -\frac{5}{6}\beta_2^2 A^3 - \frac{7}{4}\beta_4 \beta_2 A^5 + \frac{1}{2}\beta_3 \beta_2 A^4 + \frac{3}{2}\beta_3 \beta_0 A^2 - \frac{63}{80}\beta_4^2 A^7 + \frac{3}{10}\beta_3 \beta_4 A^6$$

$$+ \beta_1 A\omega^2 - 2\beta_2 \beta_0 A + \frac{3}{4}\beta_3 A^3 \omega^2 - \frac{3}{128}\beta_3^2 A^5 - 3\beta_4 \beta_0 A^3 - A\omega^4 = 0$$

$$(32)$$

solving equation (32) for the fundamental frequency gives the following second-order frequency-amplitude relationship for actuated micro-beam vibrations as:

$$\omega(A) = \left[\frac{\beta_1}{2} + \frac{3}{8}\beta_3 A^2 + \left(\frac{\beta_1^2}{4} + \frac{3\beta_1 \beta_3 A^2}{8} + \frac{15\beta_3^2 A^4}{128} - \frac{7\beta_2 \beta_4 A^4}{4} + \frac{\beta_2 \beta_3 A^3}{2} + \frac{3\beta_0 \beta_3 A}{2} - \frac{5\beta_2^2 A^2}{6} + \frac{3\beta_3 \beta_4 A^5}{10} - 3\beta_4 \beta_0 A^2 - \frac{63\beta_4^2 A^6}{80} - 2\beta_2 \beta_0 \right]^{1/2} \right]^{1/2}$$

$$(33)$$

4 RESULTS AND DISCUSSION

In order to verify the effectiveness of the present modeling and the approximate approach, the obtained results in this work are compared with the results of Rahaeifard et al. (2013) for microcantilever beam vibrations. As can be seen in Fig. 2, the second order approximation for q(t) based on the modified couple stress theory exhibits an excellent agreement with the results obtained by Rahaeifard et al. (2013)using a hybrid finite difference method. It should be noted that in order to achieve the best second-order approximation with zero initial conditions, $q_0(\tau) = 1 - \cos(\omega \tau)$ is substituted for the first trial solution.

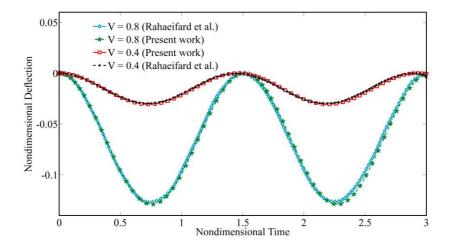


Figure 2 Comparison of the approximate periodic solutions with the reported results by Rahaeifard et al. (2013)

The effect of normalized parameters on the natural frequency and pull-in instability of actuated clamped-clamped micro-beams have been illustrated in Figs. 3 to 10. Fig.3 shows the characteristic curves of natural frequency for a micro-beam as a function of \overline{V} under some assigned values of normalized amplitude A. It is demonstrated that this non-dimensional parameter has a significant effect on predicting the pull-in phenomenon. It is obvious from Fig. 3 that by increasing the initial amplitude, the pull-in instability occurs at the lower values of actuation parameter \overline{V} . In addition, it is found that, the fundamental frequency of actuated micro-beam decreases by increasing the applied voltage until the natural frequency vanishes and the micro-beam drops to the rigid plate.

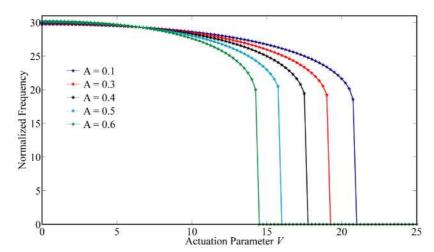


Figure 3 The effect of actuation parameter $ar{V}$ on the fundamenal frequency of micro-beam

Fig.4 examines how the nonlinear fundamental frequency of actuated micro-beam is affected by the axial force parameter f_i . As can be observed, the fundamental frequency increases by decreasing initial condition A. Also, it appears from this Fig. that when the normalized amplitude A decreases,

pull-in instability occurs at the lower values of parameter f_i . Furthermore, pull-in phenomenon vanishes by increasing the non-dimensional parameter f_i . The influence of nonlinearity parameter κ on the fundamental frequency is investigated in Fig. 5. According to the illustrated results, it is obvious that the fundamental frequency decreases as the parameter κ increases. Moreover, the pull-in voltage shifts downward by increasing this parameter.

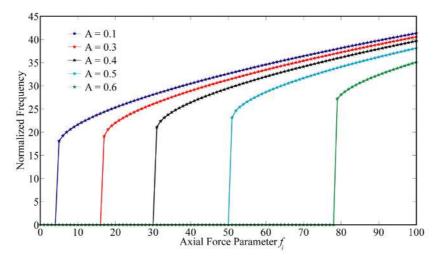


Figure 4 The effect of axial force parameter f_i on the natural frequency of micro-beam

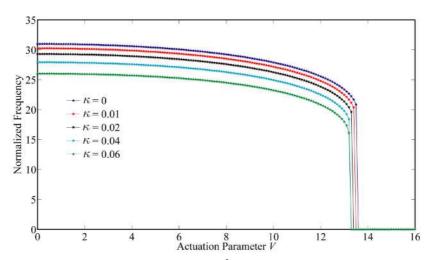


Figure 5 The effect of nonlinearity parameter $\,k\,$ on the natural frequency of micro-beam

Fig. 6 represents the impact of length scale parameter h/l on the pull-in instability of microsystems. It is clear from the Fig. that, when the beam thickness is in order of the material length scale, the normalized amplitude has no significant effect on the pull-in behavior of the structure. The fundamental frequency decreases by increasing the length scale parameter. In addition, as the initial condition increases, pull-in phenomenon occurs at lower values of non-dimensional parameter h/l. As mentioned earlier, the normalized amplitude and actuation voltage play substantial roles on

the pull-in behavior of the system. To this end, the effects of these parameters on the dynamic behavior of micro-beams are studied by plotting the time history and phase portrait diagrams. Figs. 7 and 8 investigate the nonlinear behavior of the system as a function of initial condition A.It is concluded that the time period of oscillation increases by increasing the normalized amplitude. In the vicinity of pull-in point (here A=0.65), a small increase in the amplitude, changes the dynamic behavior of micro-system. In this situation, when the initial amplitude increases, the system loses its stability and drops to the substrate beyond the pull-in point. According to Figs. 9 and 10, at less values of actuation voltage, the system exhibits periodic motion around the stable center point in the phase plane. Before pull-in point, when the actuation parameter increases, time period of vibration increases. As the actuation parameter approaches to the pull-in voltage (here $\bar{V}=19.35$), the motion trajectories in the phase plane approach to the unstable saddle node. There exists homoclinicorbit which starts from the unstable branch and goes back to the saddle node at the stable one. By increasing the applied voltage and above the pull-in voltage, the micro-beam becomes dynamically unstable and collapse onto the rigid plate.

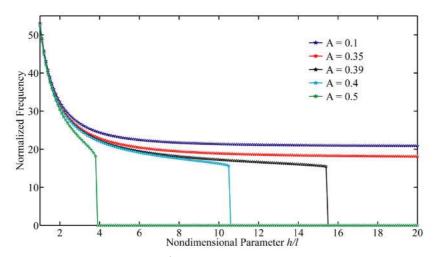


Figure 6 The effect of ratio h/l on the nonlinear frequency of actuated micro-beam

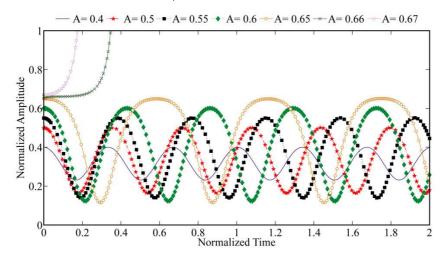


Figure 7 Time history of micro-beam deflection for different values of initial amplitude $\,A\,$

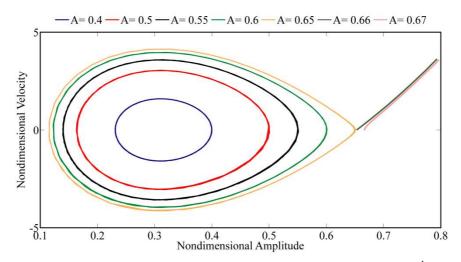


Figure 8 Phase portrait of micro-beam vibration for different values of initial amplitude $\,A\,$

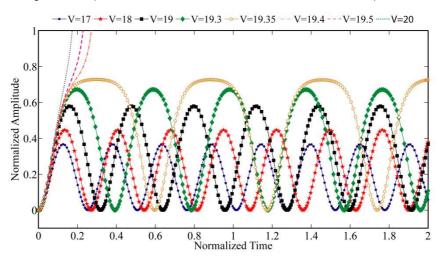


Figure 9 Time history of micro-beam deflection for different values of applied voltage $\, \overline{V} \,$

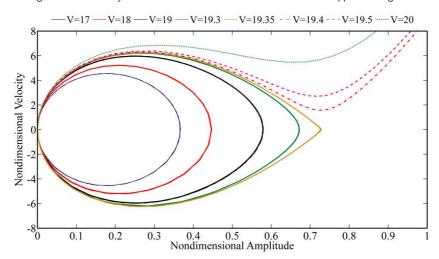


Figure 10 Phase portrait of micro-beam vibration for different values of applied voltage $\,\,\overline{\!V}$

4 CONCLUDING REMARKS

In this research, the Homotopy Perturbation Method was employed to solve governing equation of geometrically nonlinear actuated micro-beams based on the modified couple stress theory. An excellent analytical solution using asymptotic approach was obtained. The integrity of the obtained analytical solutions is verified in comparison with the results in the literature. The presented results showed thatas the normalized amplitude increases, the pull-in phenomenon occurs at lower values of actuation voltage. In addition, the pull-in stability disappears by increasing the axial force parameter. The phase plane portrait of the system illustrated the periodic orbits around the stable center point and homoclinic orbit near the unstable saddle node.

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Appendix

$$\begin{split} \beta_1 &= \left[1 + \mu_s\right] \lambda_{cc}^4 - f_i \int_0^1 \phi \phi'' \, d\xi - \frac{\lambda^2}{4} \Big(2 + \gamma \beta\Big) \\ \beta_2 &= -\frac{\lambda^2}{4} \Big(3 + \gamma \beta\Big) \int_0^1 \phi^3 \, d\xi \\ \beta_3 &= -\frac{\lambda^2}{4} \Big(4 + \gamma \beta\Big) \int_0^1 \phi^4 \, d\xi - \alpha \Bigg[\int_0^1 \phi \phi'' \Bigg[\int_0^1 \phi'^2 \, d\xi \Bigg] d\xi \Bigg] - 3\kappa \int_0^1 \phi \Big(\phi^{(4)} \phi'^2 + \phi''^3 + 4\phi' \phi'' \phi'''\Big) d\xi \\ \beta_4 &= -\frac{\lambda^2}{4} \Big(5 + \gamma \beta\Big) \int_0^1 \phi^5 \, d\xi \\ \beta_0 &= -\frac{\lambda^2}{4} \Big(1 + \gamma \beta\Big) \int_0^1 \phi \, d\xi \end{split}$$