

1 Flow over hills: A Large-Eddy Simulation of the Bolund
2 case

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21 **Abstract.** Simulation of local atmospheric flows around complex topography is
22 important for several applications in wind energy (short-term wind forecasting and
23 turbine siting and control), local weather prediction in mountainous regions and
24 avalanche risk assessment. However, atmospheric simulation around steep mountain
25 topography remains challenging, and a number of different approaches are used to
26 represent such topography in numerical models. The immersed boundary method
27 (IBM) is particularly well-suited for efficient and numerically stable simulation of
28 flow around steep terrain. It uses a homogenous grid and permits a fast meshing of
29 the topography. Here, we use the IBM in conjunction with a large-eddy simulation
30 (LES) and test it against two unique datasets. In the first comparison, the LES
31 is used to reproduce experimental results from a wind-tunnel study of a smooth
32 three-dimensional hill. In the second comparison, we simulate the wind field around
33 the Bolund Hill, Denmark, and make direct comparisons with field measurements.
34 Both cases show good agreement between the simulation results and the experimen-
35 tal data, with the largest disagreement observed near the surface. The source of
36 error is investigated by performing additional simulations with a variety of spatial
37 resolutions and surface roughness properties.

38 **Keywords:** Bolund, Complex terrain, Computational fluid dynamics, Immersed
39 boundary method, Large-eddy simulation, Navier-Stokes equation, Topography, Val-
40 idation of computational fluid dynamics models

Introduction

42 Accurate wind field modelling in complex topography has many rele-
43 vant applications. For example, wind fields are essential in evaluating
44 wind power potential, for formulating real time responses to accidental
45 releases of hazardous materials in urban areas, and for assessing the
46 redistribution of the alpine snowpack and resulting avalanche risk. In
47 all of these cases, the wind field must be evaluated around topographic
48 elements such as buildings or mountains. This creates unique numerical
49 challenges, i.e. prescribing the proper boundary conditions while also
50 solving a high Reynolds number turbulent flow. Steep slopes challenge
51 numerical methods (Tseng and Ferziger, 2003). The immersed bound-
52 ary method (IBM) (Peskin, 1972; Peskin, 2002; Mittal and Iaccarino,
53 2005) is a numerical technique that can be used to incorporate to-
54 pography into conventional simulation approaches. It is well suited for
55 environmental applications in which the terrain is steep. In regard to
56 turbulence, direct numerical simulation (DNS), which resolves the full
57 turbulence spectra of atmospheric flows, is not practical with currently
58 available computational power (Voller and Porte-Agel, 2002). Large-
59 eddy simulation (LES) (Deardorff, 1970; Bou-Zeid et al., 2004) is a
60 numerical technique in which large-scale turbulent motion is resolved
61 and small-scale motion is numerically parametrized. Here, we couple
62 the IBM with LES to simulate the turbulent flow around topographic
63 elements.

64 The following section describes LES and IBM techniques along with
65 details concerning the numerical model. We then compare simulation
66 results with measurements. In Sect. 3, a wind-tunnel experiment of flow
67 around a steep three-dimensional hill is described, and the experimental
68 results are compared directly to LES. Sect. 4 describes a 2009 field
69 campaign on the Bolund Hill in Denmark; these field measurements are
70 likewise compared with LES results. In both cases we find favourable
71 agreement between measurements and simulation results. Finally, the
72 boundary conditions and the resolution of the simulation are modified
73 to investigate the sources of error and identify potential improvements.

74

1. Model description

1.1. LARGE-EDDY SIMULATION

76 Currently available computational power is not sufficient to resolve all
 77 of the relevant scales of turbulent motion ($O(1 \text{ mm})$ - $O(10 \text{ km})$) in the
 78 atmosphere. Nonetheless, a substantial portion of the relevant scales of
 79 motion can be resolved. In the atmosphere, the largest scales of motion
 80 are the most energetic and are responsible for the majority of turbulent
 81 transport. The smallest scales of motion are more isotropic and thus
 82 amenable to parametrization. This dichotomy provides the conceptual
 83 foundation of LES in which turbulent motions are separated into re-
 84 solved scales that are computed on a grid of a given resolution, and
 85 smaller, so-called subgrid scales (SGS) that can be parametrized (Lilly,
 86 1967). Applying a filtering operation to the Navier-Stokes equations
 87 results in the LES equations:

$$\frac{\partial \tilde{u}_i}{\partial t} + \tilde{u}_j \frac{\partial \tilde{u}_i}{\partial x_j} = -\frac{1}{\rho} \frac{\partial \tilde{p}}{\partial x_i} - \frac{1}{\rho} \frac{\partial \tau_{ij}}{\partial x_j}, \quad (1)$$

$$\frac{\partial \tilde{u}_i}{\partial x_i} = 0, \quad (2)$$

88 where $\tau_{ij} = \widetilde{u_i u_j} - \tilde{u}_i \tilde{u}_j$ is the SGS stress tensor that must be parametrized.
 89 In these equations, the tilde represents the filtering operation, x_i repre-
 90 sent the three spatial axes, u_i represent the three velocity components,
 91 ρ is the air density and p is pressure. Many models have been de-
 92 veloped to deal with the SGS terms. The first, and most popular, the
 93 so-called Smagorinsky model (Smagorinsky, 1963), uses a mixing length
 94 approach:

$$\tau_{ij}^{Smag} = -2C_s^2 \Delta^2 |\tilde{S}| \tilde{S}_{ij}, \quad (3)$$

95 in which C_s is the Smagorinsky constant, Δ is the filter length and S_{ij}
 96 the strain rate tensor defined by $S_{ij} = \frac{1}{2}(\frac{du_i}{dx_j} + \frac{du_j}{dx_i})$. For this model for-
 97 mulation, the central issue is the proper choice of the Smagorinsky con-
 98 stant C_s , which can be computed dynamically (Germano et al., 1991;
 99 Porte-Agel et al., 2000; Meneveau and Katz, 2000) using a Lagrangian
 100 scale-dependent implementation (Bou-Zeid et al., 2005).

101 If the simulation includes topography, a more appropriate SGS model
 102 might be the Lagrangian model, which allows for a more natural av-
 103 eraging defined by the fluid flow, where the Eulerian model simply
 104 averages over a plane. These different approaches have been compared
 105 in the literature (Nieuwstadt et al., 1991; Andren et al., 1994) and
 106 simulation results have been verified against field measurements over

107 topographic elements such as the Askervein Hill (Taylor and Teunissen,
108 1987) and the Bolund Hill (Bechmann et al., 2009; Berg et al., 2011).

109 The LES model from Ecole Polytechnique Fédérale de Lausanne
110 (EPFL-LES) was developed from the original model described by Al-
111 bertson and Parlange (1999). In this work, the simulations were per-
112 formed using the scale-dependent Lagrangian dynamic subgrid model
113 developed by Bou-Zeid et al. (2005). To compute the Smagorinsky
114 coefficient, C_s , the scale-dependent model uses two filters at sizes twice
115 and four times the LES grid size Δ . Therefore, $C_{s,2\Delta}$ and $C_{s,4\Delta}$ are
116 computed for the two grid filters and a scaling relationship between the
117 two scales is assumed by $\beta = C_{s,4\Delta}^2 / C_{s,2\Delta}^2$. This result is then used to
118 compute $C_{s,\Delta}$. The local values of C_s are then averaged in space. If the
119 flow is homogeneous and encounters no obstacles or surface roughness
120 changes, a planar averaging can be performed. This is not the case in
121 the current study where topography is included. Thus, the averaging
122 is done following a Lagrangian scheme (following the path lines of the
123 fluid). The EPFL-LES uses the second-order Adams-Bashforth scheme,
124 in neutral conditions and without buoyancy or Coriolis effects. Spectral
125 methods are used in the horizontal directions coupled with fast Fourier
126 transform to solve the equations. Other methods, such as the iterative
127 Kyrlov solver (Saad, 1981) exist but are not considered here. The flow
128 is forced with a static pressure gradient and it is assumed that the
129 Reynolds number of the flow is sufficiently high that the molecular
130 viscosity may be neglected.

131 1.2. IMMERSED BOUNDARY METHOD

132 The most common numerical technique used to implement topography
133 in atmospheric simulations is the terrain-following coordinate system
134 (Gal-Chen and Somerville, 1975). However, other methods have been
135 developed to implement an obstacle in a simulation, including the
136 bluff-body technique (Tseng et al., 2006) and the immersed boundary
137 method (IBM), which was first proposed by Peskin (1972) and is still
138 widely used in biological fluid mechanics (Peskin, 2002; Mittal and
139 Iaccarino, 2005). Using IBM with LES is discussed by Balaras (2004);
140 in this case, SGS elements were modelled dynamically (Germano et al.,
141 1991). The efficiency and the low computational cost of IBM com-
142 pared to terrain-following techniques (body-fit meshes) is discussed in
143 Cristallo and Verzicco (2006). A full description of the implementation
144 of IBM in the EPFL-LES model is given in Chester et al. (2007). Flow
145 is computed on a homogenous grid, and the topography is described
146 separately with a level set function. To define the level set, every point
147 of the computational grid is associated with the shortest distance to

148 the topographic surface by a signed distance function, $\phi(x)$. The value
 149 is positive if the point is in the free atmosphere and negative if it is
 150 within a topographic element. During the simulation, body forces are
 151 applied to all points within the topography (i.e. $\phi(x) \leq 0$) to prevent
 152 flow. A layer above the topographic surface with thickness $\delta = 1.1\Delta$,
 153 where Δ is the grid spacing, is defined. For each point in this layer,
 154 a new coordinate system is then created, using the flow direction, the
 155 wall normal and their vectorial product. The velocity at a distance δ
 156 from the wall is defined by \mathbf{v} . In this new coordinate system, the wall
 157 stress is found using a logarithmic wind law:

$$\tau_w = -\rho \left[\frac{\kappa |\mathbf{v}_t|}{\ln(1 + \delta/z_0)} \right]^2, \quad (4)$$

158 where z_0 is the aerodynamic roughness length, $\kappa = 0.4$ is the von
 159 Karman constant, δ is the distance to the surface and \mathbf{v}_t is the tangen-
 160 tial component of the velocity \mathbf{v} . A second layer is defined under the
 161 surface, in the area $-\delta \leq \phi(x) < 0$, in which the stress is extrapolated
 162 to ensure a smooth boundary transition. For the remaining points with
 163 $\phi(x) < -\delta$, the stress profile is smoothed using five successive overrelax-
 164 ation iterations (Ferziger and Péric, 1999) that blend with the stresses
 165 at the points close to the immersed boundary and thus yield to a stress
 166 field smooth enough to be compatible with the spectral differentiation.
 167 Chester et al. (2007) stress that this results in a smearing of the vari-
 168 ables over length scales comparable to the grid scale, leading to a loss
 169 of accuracy near the surface.

170 Because the EPFL-LES model is designed with periodic boundary
 171 conditions and we require flow fields for a single obstacle, an alternative
 172 strategy to enforce the inflow boundary condition is needed. Therefore,
 173 the simulation is performed in two steps. First, a precursor simulation is
 174 run to acquire a well-defined, time-varying, turbulent inflow and these
 175 velocity components are then stored. During this simulation, no topog-
 176 raphy is present and the flow is forced through the domain along the
 177 x -axis using a constant mean pressure gradient: $f_p = -d\bar{p}/dx > 0$ per
 178 unit volume, where \bar{p} is the mean pressure. This precursor simulation
 179 provides the inflow boundary conditions as well as the initial condi-
 180 tions for the main simulation with topography. The last eighth of the
 181 domain is used as a buffer region; here, we force the velocity back to the
 182 stored velocity components from the precursor simulation: $\tilde{u}(0, y, z) =$
 183 $\tilde{u}_p(y, z)$, where $x = 0$ is the inlet coordinate of the domain and $\tilde{u}_p(y, z)$
 184 denotes the velocity from the precursor simulation. In the buffer area
 185 ($7L/8 \leq x \leq L$, L being the size of the domain), the velocity is imposed
 186 according to $\tilde{u}(x, y, z) = \tilde{u}(7L/8, y, z) + w(x)(\tilde{u}_p(y, z) - \tilde{u}(7L/8, y, z))$,
 187 where $w(x) = \frac{1}{2}(1 - \cos(\pi(x - 7L/8)/(L/8)))$. This allows us to simulate

188 non-periodic flow in the x -direction using pseudospectral numerics.
 189 A typical simulation with 256 nodes in each horizontal direction and
 190 128 nodes in the vertical takes about 12 days using 64 processors. The
 191 domain is divided into horizontal slices and each processor is assigned
 192 one or several of these slices. The bulk of the computing time is devoted
 193 to creating a converged inflow dataset. The simulation with topography
 194 is completed in 24 hours, computing 40,000 timesteps.

195 2. Wind-tunnel comparison

196 2.1. EXPERIMENTAL SETUP

197 A first test of the EPFL-LES was performed by comparing simulation
 198 results with data from an idealized wind-tunnel experiment conducted
 199 by Ishihara et al. (1999). A smooth three-dimensional (3D) hill was
 200 placed in a wind-tunnel, and wind profiles were measured at seven
 201 locations along the centre hill axis. Horizontal profiling was also per-
 202 formed at several locations above the hill and in its wake. The wind
 203 speed outside the boundary layer was maintained at 5.8 m s^{-1} and
 204 monitored throughout the experiment. This dataset is one of the few
 205 that investigates flow around 3D hills in a wind-tunnel. Several other
 206 wind-tunnel experiments focus on two-dimensional hills (Loureiro et al.,
 207 2009). In addition, many previous comparisons between LES and wind-
 208 tunnel data focus exclusively on vertical profiles (Brown et al., 2001;
 209 Allen and Brown, 2002; Tamura et al., 2007). The main advantage of
 210 the Ishihara dataset is that comparisons can also be made along the
 211 horizontal plane. In this study, the hill is described by the equation:

$$z(x, y) = H \cos^2 \left(\frac{\pi \sqrt{x^2 + y^2}}{2l} \right), \quad (5)$$

212 with the hill height $H = 40 \text{ mm}$ and $l = 100 \text{ mm}$. This topography
 213 and the 5.8 m s^{-1} wind-speed result in a Reynolds number ($\text{Re} =$
 214 UL/ν , where ν is the kinematic viscosity of the fluid, here air) around
 215 14,500, which is in the lower part of the turbulent range. The wind field
 216 was then measured in seven different locations along the flow axis: five
 217 points on the hill and two points downstream from the hill. A sampling
 218 time of 60 s was used for mean velocities and turbulence statistics.
 219 Horizontal transects of the flow in the wake region are also presented.

220 2.2. SIMULATION SETUP

221 The LES is performed in a 0.64 m x 0.64 m horizontal and 0.32 m ver-
 222 tical domain, in which x is the streamwise direction and z the vertical
 223 one. The domain was discretized into 256 x 256 x 128 points, resulting
 224 in a resolution of $dx = dy = dz = 2.5$ mm. The roughness length was set
 225 to 0.025 mm to match the conditions in the wind-tunnel experiment.
 226 The simulation had a stress-free upper boundary condition, which is
 227 not representative of the conditions found in a wind-tunnel, but the
 228 distance from the top of the hill to the top of the domain is much larger
 229 than the size of the hill, so the upper boundary condition does not have
 230 a significant impact on the flow in the vicinity of the topography. The
 231 simulation's domain height is eight hill heights, and we observe the
 232 same phenomenon as Yue et al. (2007). About three hill heights above
 233 the topography, the flow returns to an undisturbed logarithmic profile.
 234 The LES results were averaged in time over 2 sec and compared with
 235 the experimental values, representing 80,000 iterations.

236 2.3. COMPARISON

237 The results of the comparison between the measured wind-tunnel flow
 238 velocities and the computed LES flow velocities are presented in Fig. 1a.
 239 Upwind of the peak, measurements and simulation results agree well,
 240 but on the lee side of the hill, differences between the LES and the
 241 measurements are apparent. The comparison becomes more favourable
 242 at greater distances from the wall. This is further confirmed when
 243 we compare streamwise velocity measurements and simulation results
 244 taken along the y -axis in the wake of the hill at two separate elevations:
 245 at $H/8$ (Fig. 2a) and at H (Fig. 2b). Due to the periodic boundary
 246 conditions, the flow develops spanwise heterogeneities, as observed by Calaf
 247 et al. (2010) and Cal et al. (2010). To eliminate these, we perform a
 248 symmetric averaging along the central line: $u_i(x, y, z) = u_i(x, -y, z) =$
 249 $0.5(u_i(x, y, z) + u_i(x, -y, z))$, where the flow is along the x -axis and
 250 $y = 0$ is the centre of the domain. The assumption of very high Reynolds
 251 number in our model can explain some mismatches seen in the model-
 252 data comparison as the Reynolds number for the wind-tunnel is around
 253 14,500. The mismatch near the ground could be either due to the wall
 254 model, or to an incorrect prediction of the separation point location.
 255 The viscous sublayer that ultimately gives rise to this separation point
 256 is not resolved in the LES, thus the separation point is difficult to
 257 predict on very smooth surfaces such as the present example. Improve-
 258 ments in agreement at the ground-air interface for both stress and
 259 velocity fields may be achieved by applying the smoothing technique
 260 proposed by Fang et al. (2011). The reduced accuracy on the lee side

261 of the hill is also seen in Fig. 1b in which we compare the streamwise
 262 velocity variance in the $x - z$ plane defined by $\sigma_u = \sqrt{u'^2}$. All the data
 263 are normalized by the mean velocity u_H obtained at the hill height, H ,
 264 in the undisturbed boundary layer. Here, the largest disagreement be-
 265 tween the wind-tunnel measurements and LES simulation occurs in the
 266 profiles immediately downstream of the peak. This effect is expected to
 267 diminish in real topographical situations, due to the presence of sharp
 268 corners and discontinuities that enable the position of separation points
 269 to be established. This conjecture is addressed in the next section.

270 The impact of elevation on the accuracy of the results can be seen
 271 in Fig. 3, in which the ratios of simulated variances to the measured
 272 variances are displayed. Near the ground and up to an elevation equal
 273 to the height of the hill ($z/H \leq 1$), and at locations upwind of the
 274 topography, the LES underpredicts the variance. Conversely the LES
 275 overpredicts the variance on the lee side of the hill ($x/H = 3.75$ and
 276 $x/H = 5$), particularly for elevations between $z/H = 0.5$ and $z/H = 1$.
 277 All data converge at elevations above the height of the hill ($z/H \geq 1$)
 278 where the simulated variances underestimate the measurements.

279 3. The Bolund comparison

280 3.1. THE BOLUND FIELD CAMPAIGN

281 The Askervein Hill field campaign (Taylor and Teunissen, 1987) was,
 282 until recently, the standard case used to test numerical simulations of
 283 turbulent flows over topography (Lopes et al., 2007; Chow and Street,
 284 2009). The Bolund field campaign, which took place from December
 285 2008 to February 2009 on the Bolund Hill in Denmark, near the Risø
 286 Campus of the Technical University of Denmark, has more challenging
 287 topography, including steep slopes and cliffs, than that of the Askervein
 288 Hill campaign. The details of the experiment are described in Bech-
 289 mann et al. (2009). The hill is 130 m long (east - west axis) and 75
 290 m wide (north - south axis), with a maximum height of 11.7 m. The
 291 dominant winds are from the west and south-west. Thus the wind has
 292 an extensive upwind fetch over the sea before encountering land, leading
 293 to a "steady" flow on the windward side of the hill. The wind field first
 294 encounters a 10-m vertical cliff, after which air flows back down to
 295 sea level on the east side of the hill. A recirculation zone and a flow
 296 separation are expected due to this abrupt change of slope. During the
 297 campaign, 35 anemometers were deployed over the hill. The location
 298 of the measurement devices can be seen on Fig. 4. Instrumentation
 299 includes 23 sonic anemometers, 12 cup anemometers and two lidars. At

300 each measurement location, the three components of the wind velocity
 301 vector and their variances were recorded for four different dominant
 302 wind directions, three westerly winds originating from the sea (268° ,
 303 254° and 242°) and one easterly wind originating from the land (95°).
 304 The mean wind speed during the measurements was around 10 m s^{-1} ,
 305 leading to a Reynolds number of $\text{Re} = Uh/\nu \approx 10^7$ (Berg et al., 2011).
 306 The measured values are 10-min averages of measurements sampled at
 307 20 Hz for sonic anemometers. We followed Bechmann et al. (2009) and
 308 considered the atmosphere as neutral.

309 3.2. SIMULATION SET-UP

310 The simulations used to compute the wind fields on the Bolund Hill
 311 were performed using a domain of $512 \times 256 \times 128$ elements with
 312 dimensions of $512 \text{ m} \times 256 \text{ m} \times 128 \text{ m}$ (wind direction, cross-wind
 313 direction and vertical direction) giving a resolution of $dx = dy = dz =$
 314 1 m . A roughness length of $z_0 = 0.3 \text{ mm}$ was prescribed for the entire
 315 domain. This value is consistent with the measured surface roughness
 316 of water, but smaller than the measured roughness length over the
 317 land, viz. $z_0 = 15 \text{ mm}$. It is assumed that the topography has a much
 318 larger impact on the near-surface flow dynamics than the small change
 319 in surface roughness length. Simulations were performed for the four
 320 different wind directions observed during the field campaign (i.e. wind
 321 direction from 268° , 254° , 242° and 95°). Each simulation was run for
 322 40,000 5 ms timesteps, i.e. 200 sec, using a precursor simulation coupled
 323 with the buffer technique as described in Sect. 1. These settings yielded
 324 a mean running time of four advection times along the whole domain
 325 or 15 advection times along the hill.

326 3.3. COMPARISON

327 Bechmann et al. (2011) compared results from several different mod-
 328 elling techniques with the field measurements of the wind around the
 329 hill, but few LES results were included in these.

330 We first compare our results with the experiment using the same
 331 methodology used by Bechmann et al. (2011) using the data for flow
 332 along the 242° axis (Fig. 4). First, the inflow wind-speed profile given
 333 by the simulations is compared with the measurements at mast M0,
 334 which is about 150 m upstream of the hill. The comparison can be
 335 seen in Fig. 5. The red line is the theoretical line obtained using a
 336 logarithmic profile:

$$u(z_{agl}) = \frac{u_{*0}}{\kappa} \ln\left(\frac{z_{agl}}{z_0}\right), \quad (6)$$

337 where $z_0 = 0.3$ mm is the roughness length, $\kappa = 0.4$ is the von Karman
 338 constant, $u_{*0} = 0.45$ m s⁻¹ is the friction velocity measured during the
 339 field campaign and z_{agl} is the elevation above ground level. The points
 340 represent the field data that were used to calibrate the theoretical
 341 profile and the blue line shows the LES profile. Note that these data are
 342 normalized by u_* . The matching is not perfect but would be considered
 343 accurate by Bechmann et al. (2011) (less than 10% error).

344 In order to investigate discrepancies between simulations and mea-
 345 surements occurring after the first mast (M0), we follow Bechmann et al.
 346 (2011) and quantify changes in the wind field as either changes in speed
 347 (speed-up) or in direction (turning). Speed-up is defined by:

$$\Delta S_m = \frac{\langle \bar{s}/u_{*0} \rangle_{z_{agl}} - \langle \bar{s}_0/u_{*0} \rangle_{z_{agl}}}{\langle \bar{s}_0/u_{*0} \rangle_{z_{agl}}}, \quad (7)$$

348 where \bar{s} is the mean wind speed at the sensor location and \bar{s}_0 is the
 349 mean wind speed at the inflow mast M0. Turning is defined as the
 350 difference between the wind direction at the measurement point and
 351 that at M0. The comparison is made for two different elevations, 2 and
 352 5 m above the ground level. The results for the speed-up (Fig. 6) and
 353 for the turning of the wind (Fig. 7) both show excellent agreement with
 354 the experimental data.

355 3.4. COMPARISON, PART II

356 Excellent agreement between the EPFL-LES and the field measure-
 357 ments was found when using the comparison protocol of Bechmann
 358 et al. (2011), but these tests do not include all data for all wind di-
 359 rections. Further tests were conducted on the entire measurement set.
 360 We present here the results for the inflow velocity along the 242 ° axis,
 361 but the other directions show similar results and are summarized in
 362 Table I. Fig. 8 shows the comparison between the LES results and all
 363 experimental data for the case of the westerly wind (242°). In this plot,
 364 the wind speeds are normalized by u_{*0} , which is the friction velocity at
 365 a reference point, M0, located on the station far upstream of the hill.
 366 There is good agreement between the field data and the LES results.
 367 In Fig. 9 the ratio of LES wind speeds to measured wind speeds is
 368 displayed as a function of elevation (Fig. 9a) and measurement location
 369 (Fig. 9b). It is clear that the largest mismatch between the LES results
 370 and the experimental data occurs at the lowest points. Many factors
 371 contribute to this lower accuracy near the surface.

- 372 1. Interpolation of the velocity field near the land surface, this is highly
 373 dependent on the wall model used;

- 374 2. The smoothing of the stress field induced by the IBM;
- 375 3. The SGS scales are more important near the surface, leading to a
376 higher ratio of the parametrized to the resolved terms in that area.

377 Two of the least accurate results originate from mast M8, which
378 is on the lee side of the hill for the 242° wind direction and located
379 on a slope parallel to the main wind direction. Mast M5 is also on a
380 slope parallel to the 242° wind direction, and mast M6, where we also
381 see a single point of disparity, is at the top of a cliff perpendicular
382 to the main flow. Fig. 10 shows the simulated wind directions versus
383 the measured wind directions. The point sizes are proportional to the
384 magnitude of the measured wind speed at that location. The wind
385 speed at the mispredicted point from M8 is very low, and is once again
386 associated with the lowest-altitude wind measurement from mast M8.
387 The field measurements show that the wind direction at this point is
388 parallel to the prevailing wind, but the simulation predicts it to have a
389 wind direction rotated with respect the main flow. The results obtained
390 for the two other sensor locations on mast M8 fit the field data with
391 respect to the wind direction. The simulated wind direction for the
392 middle sensor at M8 also agrees with field data (the highest sensor is
393 a cup anemometer that only gives wind-speed). Further evidence that
394 the wind components are more difficult to capture in certain locations
395 can be seen in Fig. 11, which shows the simulated versus the measured
396 speeds for the u and v components of the wind field sorted by station
397 location. The larger discrepancy at mast M8 is once again apparent.
398 The ratio of the simulated u -component to the measured u -component
399 is shown in Fig. 12. In this plot, the v -component is not displayed
400 because some of the measured values are very close to zero. Greater
401 disparity between simulations and measurements can again be seen for
402 masts M6 and M8. Fig. 13 compares the variances, $\overline{u'u'}$ and $\overline{v'v'}$, and
403 shows the ratio of simulated to experimental data. Here, the locations
404 with the least accurate results are situated on masts M6 and M8 for
405 the u -variances and on masts M2 and M6 for the v -variances. At their
406 location, on the top of the cliff, there is significant turbulence caused by
407 orographic lifting. Note that the LES overestimates the variance for the
408 lee side of the hill (M8) and underestimates it for the other masts at the
409 top of the cliff (M2 and M6). This trend is similar to the one observed
410 for the wind-tunnel comparison (Fig. 3). The LES overestimates the
411 variances on the lee side and underestimates them on the windward
412 side and at elevations higher than the hill height.

413 The same analysis was conducted for the other wind directions.
414 For winds from the sea (254° and 268°), the results are very similar.
415 Comparisons of the results with other wind directions verify the finding

416 that LES is most challenged when the wind is oriented parallel to the
 417 slope. For example, for the 268° wind direction, the towers M4 and
 418 M5 have lower agreement, while the mast M8 gains accuracy near the
 419 ground. Table I shows a numerical comparison of the results using the
 420 speed-up error, defined by:

$$R_S = 100(\Delta S_s - \Delta S_m) \quad (8)$$

421 where subscript s and m denote the simulated and measured speed-
 422 up (defined by Eq. 7), respectively. With a mean error of 7.1%, the
 423 242° case provides an accurate prediction of the speed-up error. The
 424 254° simulation gives the best result (mean error of 5.9%) and the 268°
 425 direction has a mean speed-up error of 12.1%.

426 The last case, with the wind originating from the land (95°), has
 427 the largest mean speed-up error (24%). For this case, a separate inflow
 428 file was built to match the high reference shear stress ($u_{*0} = 0.51 \text{ m}$
 429 s^{-1}) and the higher surface roughness ($z_0 = 15 \text{ mm}$ compared to $z_0 =$
 430 0.3 mm for the cases with flow from the sea). As a result, the mean
 431 wind speed is about 1.6 times lower at heights ranging from 2 to 20 m
 432 above the ground than in the three first cases. For the 95° case, we see
 433 the same trend as in the previous results: masts M1 and M7 located
 434 on the lee side of the hill, near the edge of the cliff, show the only
 435 inaccurate wind direction results. This mismatch is also observed when
 436 comparing wind speeds: all towers show good agreement except for the
 437 lower sensors of masts M1 and M7.

438
 439 One possible source of error is the prescribed surface roughness
 440 used for the inflow file and in simulation. For example when the wind
 441 direction was 95° (from land), the whole simulation was run with a
 442 single surface roughness of $z_0 = 0.3 \text{ mm}$. During the blind test with
 443 flow from the land (inflow file with $z_0 = 15 \text{ mm}$), the airflow also
 444 crosses water (simulation with $z_0 = 0.3 \text{ mm}$) before interacting with
 445 the hill. Notice that the amount of water between the coast and the
 446 hill was not specified in the blind test and that a band of land emerges
 447 from the sea at low tide. The speed-up error at a height of 2 m has
 448 a value of 32%, whereas at 5 m it decreases to 16%. This does not
 449 follow the trend observed for flow from the sea and suggests that the
 450 boundary conditions on the ground may need to be modified. To further
 451 investigate the impact of surface roughness on the accuracy of the LES
 452 results, further simulations were performed.

453 3.5. SENSITIVITY AND PERFORMANCE

454 The assumption that surface roughness has a much lower importance
 455 than topography was tested by running a new simulation for the wind
 456 direction from 242° using two different surface roughnesses: $z_0 = 15$
 457 mm for land and $z_0 = 0.3$ mm for sea and inflow file. The sensitivity
 458 of the model to the grid resolution was also tested by running another
 459 simulation involving two different surface roughnesses but with a grid of
 460 $256 \times 128 \times 64$ cubes, i.e with half the elements in each direction, leading
 461 to a grid size of $2 \text{ m} \times 2 \text{ m} \times 2 \text{ m}$. Table II presents these results. The use
 462 of two different surface roughnesses in the higher-resolution simulations
 463 lowers the mean speed-up error from 7.1% to 5.1%. As expected, the
 464 use of a low-resolution grid leads to less accurate results, and a mean
 465 speed-up error of 11.9%. Notice that, in this case, the error near the
 466 ground is larger than it is at 5 m, which is the opposite of the case with
 467 a higher resolution.

468
 469 The results obtained with the EPFL-LES are compared to the mod-
 470 eling efforts performed by Bechmann et al. (2011) in Table III. When
 471 all the cases are included, the EPFL-LES is among the most accurate
 472 models, and if the 95° case is excluded, it shows better results than any
 473 other model.

474 4. Conclusion

475 To simulate the wind fields around topography, we have implemented
 476 an immersed boundary method within a large-eddy simulation model
 477 (LES).

478 Results were first tested against wind-tunnel data. For this case,
 479 we see good agreement between simulation and experimental data,
 480 except for some inaccuracies near the ground and on the lee side of
 481 the hill; these are likely due to the fact that the separation point is
 482 hard to resolve on such a smooth surface. The EPFL-LES is further
 483 tested against experimental data from the Bolund Hill field campaign.
 484 A comparison is made for four main measured wind directions, with the
 485 focus on the wind speed, direction and variance. In general, the LES
 486 shows good agreement with field measurements, however the points
 487 closest to steep slopes aligned parallel to the main wind direction or
 488 near the ground appear to be the most difficult to predict accurately.
 489 Nevertheless, the results are still favourable. The observed disparities
 490 indicate that there is room for improvement in the wall model used as
 491 boundary condition for ground, where LES predictions are difficult. A

492 sensitivity study confirms the importance of using appropriate surface
 493 roughness for the sea and the land surfaces and shows that a coarse
 494 grid significantly lowers accuracy near the ground. The comparison of
 495 our results with other approaches presented in Bechmann et al. (2011)
 496 is very favourable.

497

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Table I. Dependence of the speed-up error (defined by Eq. 8) to the wind direction. Values in percents (%).

| Wind direction | Error at 2 m | Error at 5 m | Average error |
|----------------|--------------|--------------|---------------|
| 95 | 32.0 | 16.0 | 24.0 |
| 242 | 5.7 | 8.5 | 7.1 |
| 254 | 6.2 | 5.7 | 5.9 |
| 268 | 9.0 | 15.2 | 12.1 |

Table II. Improvements of speed-up error (defined by Eq. 8) obtained by increasing the resolution of the grid and/or using different surfaces roughnesses for land and sea. Values in percent (%).

| Size of the grid | Different z_0 ? | Error at 2 m | Error at 5 m | Average error |
|------------------|-------------------|-----------------|-----------------|------------------|
| 256 x 128 x 64 | yes | 14.5 | 9.2 | 11.9 |
| 512 x 256 x 128 | no | 5.7 | 8.5 | 7.1 |
| 512 x 256 x 128 | yes | 4.7 | 5.6 | 5.1 |

Table III. Comparison between the speed-up errors (defined by Eq. 8) from models taking part to the blind test, measured data and results from the EPFL-LES model. Cases 1 and 3 represent the flow from the sea from 242° and 268° . Values in percent (%).

| Model | All cases (best) | Case 1+3 (best) |
|--------------------|------------------|-----------------|
| Two-equations RANS | 13.6 (10.2) | 15.1 (11.4) |
| Experiment | | 14.7 (13.3) |
| One-equation RANS | 16.3 (12.2) | 17.2 (13.8) |
| LES | 16.0 (13.5) | 17.3 (14.1) |
| Linearized | 21.0 (18.5) | 23.7 (20.6) |
| All models | 15.8 | 17.3 |
| EPFL-LES | 12.3 | 9.6 |

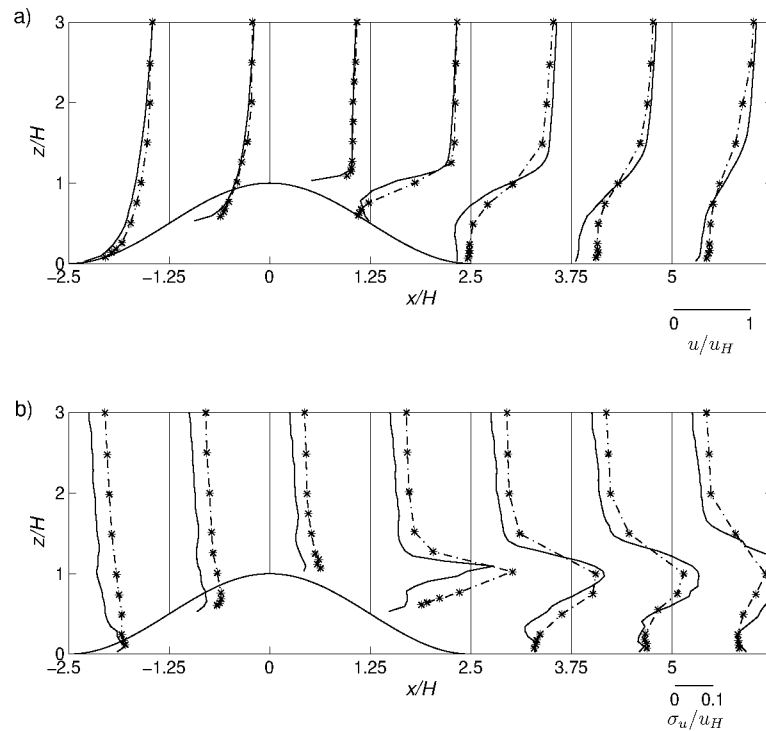


Figure 1. Stars are the wind-tunnel measurements, the solid line represents the LES results. Data are given for seven different locations represented by the vertical lines. a) Wind-tunnel comparisons in streamwise direction along the centre of the domain (through the middle of the hill). b) Corresponding streamwise velocity variances. Velocity and variance are normalized by u_H , which is the velocity at "hill height" for a flow field without a hill (4.9 m s^{-1} for our LES).

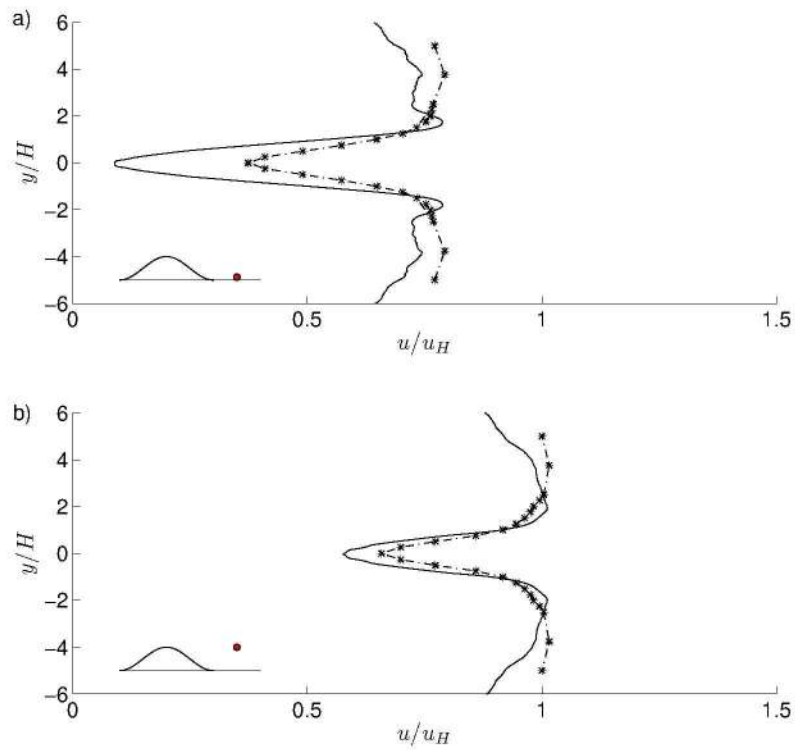


Figure 2. Stars are the wind-tunnel measurements, the solid line represents the LES results a) Wind-tunnel comparisons of the u -component of velocity at $H/8$ in the wake region at a distance $x/H = 3.75$ behind the centre of the hill. b) Wind-tunnel comparison of the u -component of velocity at hill height in the wake region at a distance $x/H = 3.75$ behind the top of the hill. Velocity is normalized by u_H , which is the velocity at "hill height" for a flow field without a hill (4.9 m s^{-1} for our LES).

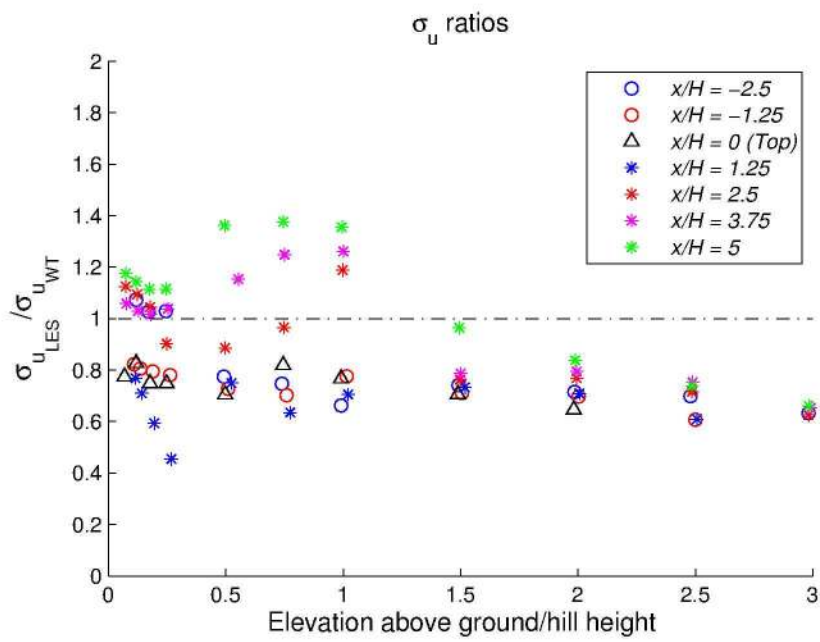


Figure 3. Ratio of LES results to wind-tunnel data for variances of the u -component of velocity, results sorted by locations.

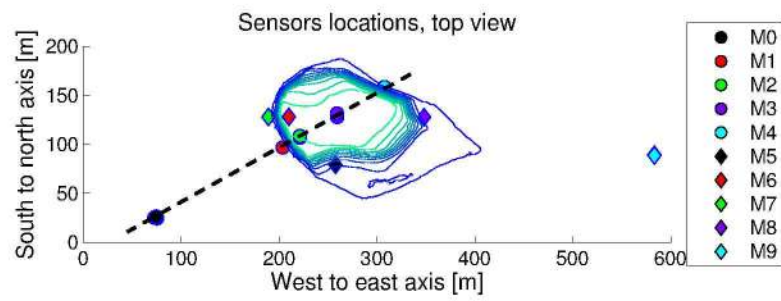


Figure 4. Station locations on the Bolund Hill. North is upwards, the dashed line is along the 242° wind direction (west to east).

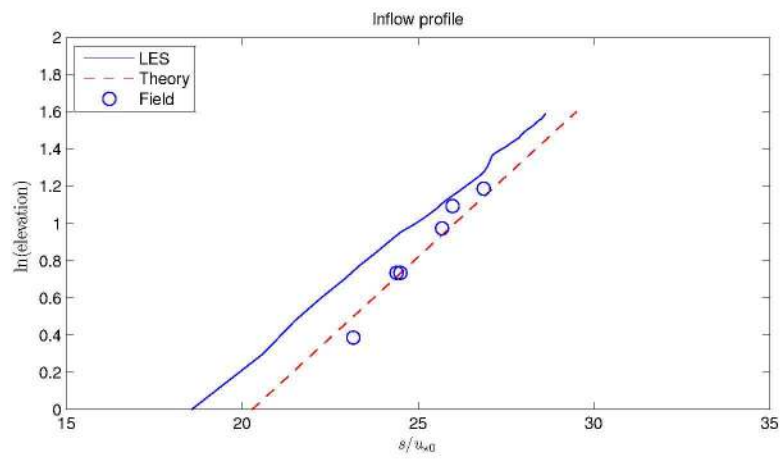


Figure 5. Comparison of inflow wind profiles taken at mast M0.

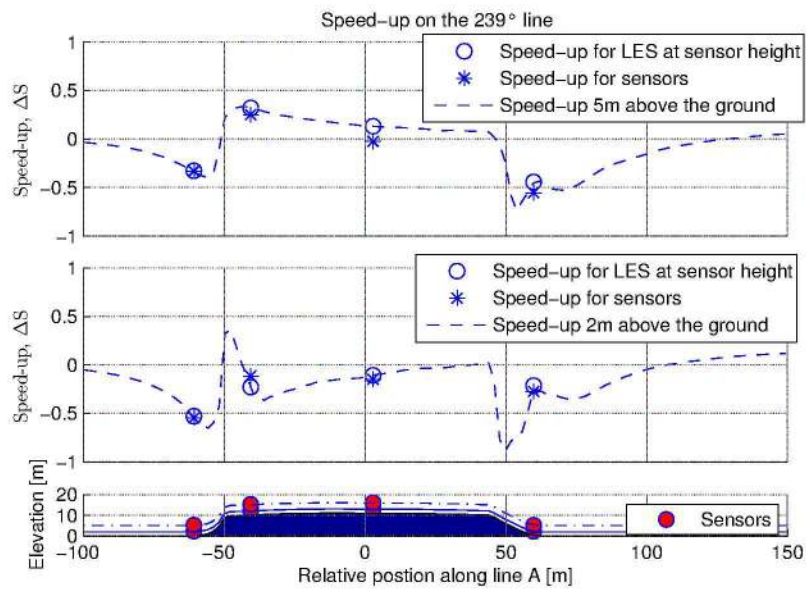


Figure 6. Speed-up of the wind along the Bolund Hill. Wind direction is from 242° .

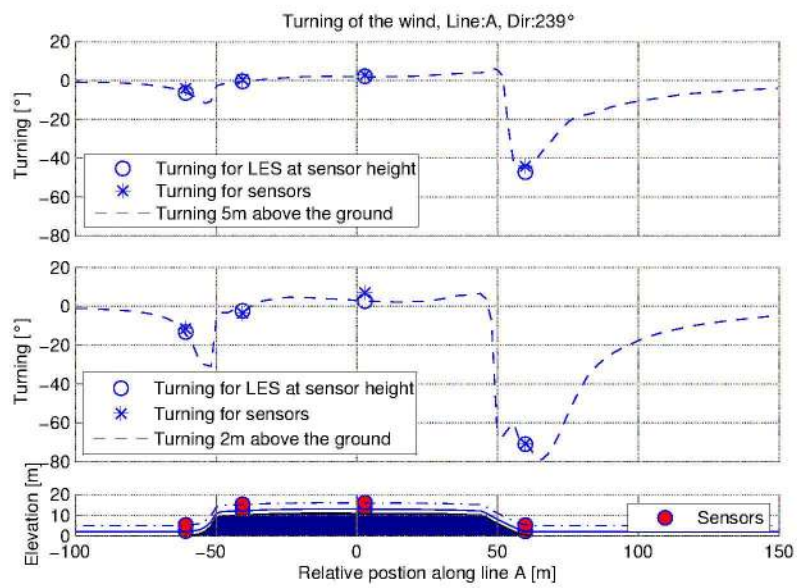


Figure 7. Turning of the wind direction along the Bolund Hill. Wind direction is from 242°.

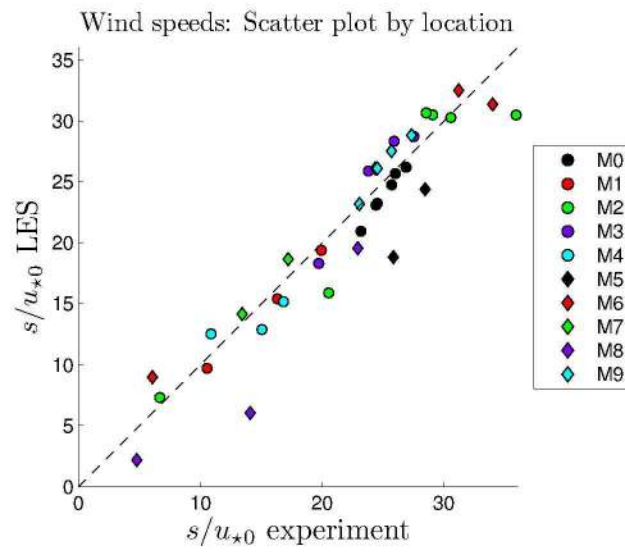


Figure 8. Scatter plot of total speeds, field data against LES results. Wind direction is from 242° .

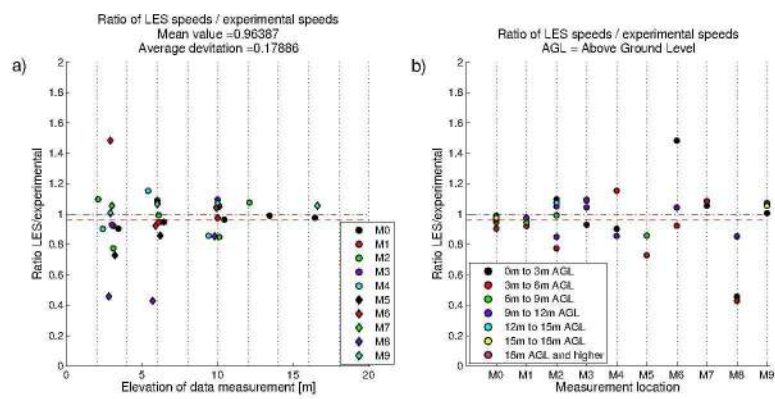


Figure 9. Ratio of LES results to field data for total speeds, the dot colours are sorted by location (a) and elevation (b). Red line is the average value, black line is at ratio = 1. Wind direction is from 242° .

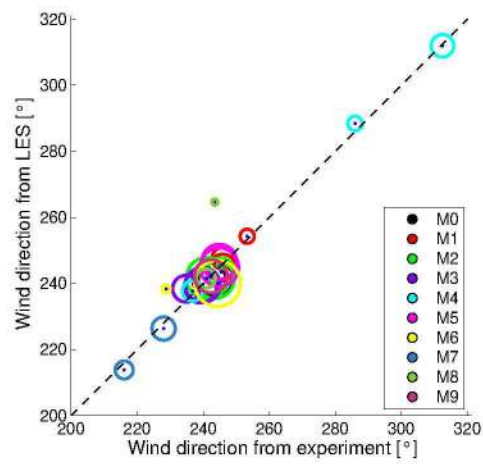


Figure 10. Scatter plot for wind directions, field data against LES results. The size of the dot is proportional to the total wind speed at the sensor's location. Wind direction is from 242°.

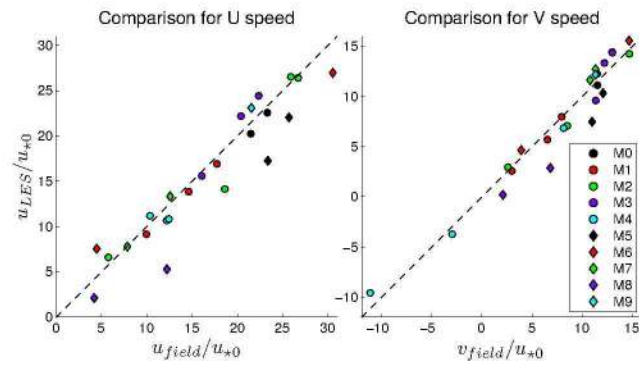


Figure 11. Scatter plots of u - and v -components of speed, field data against LES results. Wind direction is from 242° .

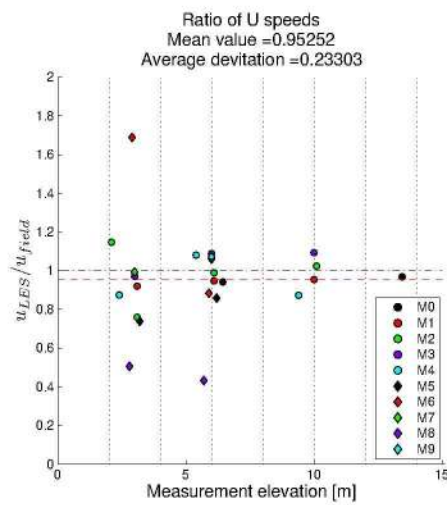


Figure 12. Ratio of LES results to field data for u -component of velocity, sorted by locations. Red line is the average value, black line is at ratio = 1. Wind direction is from 242° .

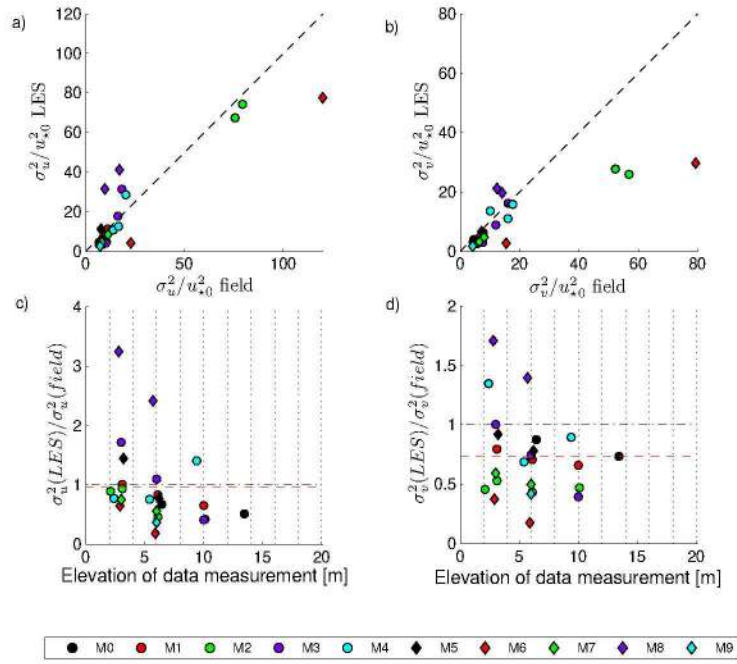


Figure 13. Top: Scatter plots of variances of u - and v -components of velocity, field data against LES results
 Bottom: Ratio of variances of u - and v -components of speed, LES results over field data. Red line is the average value, black line is at ratio = 1. Wind direction is from 242° .