FLUID FREE SURFACE PROXIMITY EFFECT ON A SPHERE VERTICALLY ACCELERATED FROM REST

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## CALIFORNIA INSTITUTE OF TECHNOLOGY

Hydrodynamics Laboratory

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## ABSTRACT

Theory is developed to estimate the effect of free surface proximity on the initial added mass of a sphere accelerated vertically upward from rest in an ideal fluid. It is assumed that the acceleration regime is sufficiently brief that inertial forces predominate and gravitational effects may be neglected. Results of tests in water indicate that while there are slight viscous and gravitational effects over the acceleration regime, the agreement between theory and experiment is good. It is concluded that over briefer acceleration regimes these effects would decrease and the agreement would improve.

# FLUID FREE SURFACE PROXIMITY EFFECT ON A SPHERE VERTICALLY ACCELERATED FROM REST 

J. G. Waugh and A. T. Ellis

## 1. Introduction

This report comprises the final technical report on work supported by the Department of the Navy, Bureau of Naval Weapons (and subsequently the Naval Ordnance Systems Command) on Contract N600(19)59368 to the California Institute of Technology, and Task Assignment No. RRRE-04001/216-1/R009-01-01 to the U. S. Naval Ordnance Test Station, Pasadena, California. One previous technical report was issued (Ref. 1) and three papers were published in the open literature (Refs. 2-4).

A major problem in the design of bodies moving through a fluid is the evaluation of the inertial resistance of the fluid when the body is accelerated. This inertial resistance affects the motion of the body and is usually described in terms of an apparent increase in the mass (added mass) of the body (Ref. 5). Although in theory added masses can be computed for any flow situation and used to determine a complete expression for the ideal pressure forces, the actual solution of the problem often proves mathematically intractable. The greatest difficulties are found in the determination of motion of bodies not deeply submerged; that is, bodies close to a free surface. In this category are weapons of primary importance to national defense, such as aircraft torpedoes, other air-to-water missiles, and water-to-air missiles such as POLARIS and SUBROC. The importance of added mass in service missile technology is evidenced by the fact that these missiles experience perturbations in passing through the water-air interface which pose problems in their subsequent operation, and which can in part be ascribed to added mass effects.

Despite the importance of water-exit perturbations in missile behavior and the large effect that added mass must have in these perturbations, there appears to be little information of a basic nature. It is known that potential frictionless flow exists in a real fluid during the first instant after a body is accelerated from rest (Ref. 6). Generally speaking, resistance to acceleration from rest in a real fluid should agree with ideal fluid theory and quite good agreement has been obtained for a sphere when all solid boundaries are remote (Ref. 5), but there do not appear to be any tests of this hypothesis for a body in the presence of a free surface. This suggests the possibility of developing theory for a body of mathematically tractable shape in the presence of a free surface and correlating this theory with experimental results obtained during the first instant after this body has been accelerated from rest. It was decided to conduct theoretical and experimental studies on the initial added mass of a sphere accelerated vertically upward from rest at varied depths below a free fluid surface. These studies are described in this report.

## 2. Theory

In the following theoretical discussion we assume an ideal fluid with orthogonal coordinate axes $x, y, z$ such that the $x, z$-plane lies in the undisturbed free fluid surface and the y-axis is directed positively vertically upward (Fig. 1). Let the velocity potential which satisfies all boundary conditions be $\phi(x, y, z, t)$, the fluid velocity (assumed to be small everywhere) be $v=-\nabla \phi$, and all disturbances arising from $\phi$ be negligible at a great distance. Then the pressure equation for time-dependent irrotational flow is

$$
\begin{equation*}
\frac{p}{\rho}+\frac{1}{2}\left(\phi_{x}^{2}+\phi_{y}^{2}+\phi_{z}^{2}\right)+g y-\frac{\partial \phi}{\partial t}=C(t) \tag{1}
\end{equation*}
$$

where $C(t)$ is an arbitrary function of time. If $\eta=\eta(x, z, t)$ is the height of the free surface (assumed to be small) measured from the plane $y=0$, the linearized formula for surface deformation may be obtained from the following considerations. The velocity components
$\phi_{x}, \phi_{y}, \phi_{z}$ must be small at the surface, and therefore the quadratic terms $\phi_{x}^{2}, \phi_{y}^{2}, \phi_{z}^{2}$ may be dropped from Eq. (1). Furthermore $p$ is constant at the free surface. Then Eq. (1) becomes

$$
\begin{equation*}
\frac{p}{\rho}+g \eta-\frac{\partial \phi}{\partial t}=C(t) \tag{2}
\end{equation*}
$$

If $C(t)$ and the additive constant, $p / \rho$, are supposed merged in the value of $\partial \phi / \partial t$, Eq. (2) becomes

$$
\begin{equation*}
g_{\eta}=\left(\frac{\partial \phi}{\partial t}\right)_{y=\eta} \approx\left(\frac{\partial \phi}{\partial t}\right)_{y=0} \tag{3}
\end{equation*}
$$

At the free surface, the velocity of each fluid particle in the surface must be tangential to the surface. Expressing the equation of the free surface as an implicit function

$$
\begin{equation*}
F(x, y, z, t) \equiv y-\eta(x, z, t) \equiv 0 \tag{4}
\end{equation*}
$$

this kinematic condition is given by

$$
\begin{equation*}
\frac{\mathrm{dF}}{\mathrm{dt}}=0 \tag{5}
\end{equation*}
$$

From Eqs. (4) and (5)

$$
\begin{equation*}
-\frac{\partial \eta}{\partial t}+\frac{\partial \phi}{\partial x} \frac{\partial \eta}{\partial x}+\frac{\partial \phi}{\partial z} \frac{\partial \eta}{\partial z}-\frac{\partial \phi}{\partial y}=0 \tag{6}
\end{equation*}
$$

Neglecting the product terms in Eq. (6)

$$
\begin{equation*}
\frac{\partial \eta}{\partial t}+\frac{\partial \phi}{\partial y}=0 \quad(y=0) \tag{7}
\end{equation*}
$$

which states that the surface deformation velocity, $\partial_{\eta} / \partial t$, must be equal to the surface perturbation velocity, $-\partial \phi / \partial y$, evaluated in the plane $y=0$. The pressure condition on $\phi$ from Eq. (3), and the kinematic condition from Eq. (7), can be combined as follows

$$
\begin{equation*}
\frac{\partial}{\partial t}\left(\frac{1}{g} \frac{\partial \phi}{\partial t}\right)=\frac{\partial \eta}{\partial t}=-\frac{\partial \phi}{\partial y} \tag{8}
\end{equation*}
$$

or

$$
\begin{equation*}
\frac{\partial^{2} \phi}{\partial t^{2}}+g \frac{\partial \phi}{\partial y}=0 \quad(y=0) \tag{9}
\end{equation*}
$$

Now consider a disturbance generated in the fluid involving high fluid accelerations, but considered over a sufficiently brief interval of time that fluid displacements and velocities are small. Then the gravitational effect will be negligible and inertial forces will predominate. Under these assumptions, the term $g(\partial \phi / \partial y)$ may be dropped in Eq. (9) to yield

$$
\begin{equation*}
\frac{\partial^{2} \phi}{\partial t^{2}}=0 \quad(y=0) \tag{10}
\end{equation*}
$$

and neglecting $g$ (i.e., setting $g=0$ ) in Eq. (3) implies

$$
\begin{equation*}
\frac{\partial \phi}{\partial t}=0 \quad(y=0) \tag{11}
\end{equation*}
$$

We may satisfy the conditions of Eq. (10) and (11) by constructing $\phi(x, y, z, t)$ to be independent of $t$ in $y=0$. Thus if we assume $\phi_{0}$ is the velocity potential in the absence of the surface, we can construct $\phi$ by writing

$$
\begin{equation*}
\phi=\phi_{o}(x, y, z, t)+\phi_{1}(x, y, z, t) \tag{12}
\end{equation*}
$$

where $\phi_{1}$ is a potential function which corrects $\phi_{0}$ for the presence of the fluid surface. In view of the above conditions, it suffices to take

$$
\begin{equation*}
\phi_{1}(x, o, z, t)=-\phi_{o}(x, o, z, t) \tag{13}
\end{equation*}
$$

If the disturbance is caused by a moving body in the fluid, we may assume a function $\phi_{2}$ which will correct Eq. (12) to satisfy the boundary condition on the body. That is,

$$
\begin{equation*}
\phi=\phi_{0}(x, y, z, t)+\phi_{1}(x, y, z, t)+\phi_{2}(x, y, z, t) \tag{14}
\end{equation*}
$$

satisfies the body boundary condition. Assuming a function $\phi_{3}$ Eq. (14) may be corrected for the presence of the fluid surface, etc. In this way we can construct a sequence of potential functions $\phi_{i}$ such that

$$
\begin{equation*}
\phi(x, y, z, t)=\phi_{0}(x, y, z, t)+\sum_{i=1}^{\infty} \phi_{i}(x, y, z, t) \tag{15}
\end{equation*}
$$

satisfies the fluid surface and boundary conditions. If, in addition, the body has a mathematically tractable shape such as a sphere, and experimental conditions approximate those assumed in the theory, a correlation between theory and experimental data may be obtained.

We consider, then, a sphere accelerated vertically upward from rest at different depths below the fluid surface. Referring to Fig. 1, the potential function for a sphere of radius, a, in fluid of infinite extent, moving upward along the $y$-axis is

$$
\begin{equation*}
\phi_{0}=\frac{U(t) a^{3}}{2} \frac{\cos \theta}{r_{o}^{2}}=\frac{U(t) a^{3}}{2} \frac{y+h_{o}(t)}{r_{o}^{3}} \tag{16}
\end{equation*}
$$

where $U(t)$ is the sphere velocity and $h_{0}(t)$ is the instantaneous position of the center of the sphere below the origin. Equation (16) is the velocity potential of a doublet with its axis directed along positive $y$ and strength $U(t) a^{3} / 2$. The boundary condition on the sphere is given by

$$
\begin{equation*}
-\frac{\partial \phi_{\mathrm{o}}}{\partial \mathrm{r}_{\mathrm{o}}}=\mathrm{U}(\mathrm{t}) \cos \theta, \quad\left(\mathrm{r}_{\mathrm{o}}=\mathrm{a}\right) \tag{17}
\end{equation*}
$$

From the preceding theory, through the use of successive "image" doublets in the sphere and free surface the desired velocity potential may be obtained.

If we denote the doublet strengths in the primary sphere by $\mu_{\text {s } v}$ and those in the free surface by $\mu_{i v}$, we obtain for the doublet strengths and locations

$$
\begin{gather*}
\mu_{\text {so }}=\mu_{\text {io }}=\frac{U(t) a^{3}}{2}  \tag{18}\\
\mu_{\text {sv }}=\mu_{i v}=\frac{U(t) a^{3}}{2} \prod_{p=1}^{v}(-1)^{v}\left(\frac{a}{h_{0}+h_{p-1}}\right)^{3}, \\
(v=1,2,3, \ldots) \tag{19}
\end{gather*}
$$

where

$$
\begin{equation*}
h_{p}=h_{o}-\frac{a^{2}}{h_{o}+h_{p-1}}, \quad(p=1,2,3, \ldots) \tag{20}
\end{equation*}
$$

The primary sphere image doublet vertical positions, $\mathrm{y}_{\mathrm{S} v}$, are given by $-\mathrm{y}_{\mathrm{s} v}=\mathrm{h}_{v}$ and the surface image vertical positions, $\mathrm{y}_{\mathrm{i} \nu}$, by $\mathrm{y}_{\mathrm{i} \nu}=\mathrm{h}_{\nu}$. The limit points for the sphere and surface image doublet positions are derived in the Appendix. Note that $-y_{\text {So }}=h_{o}=y_{\text {io }}$. Then the desired velocity potential is given by

$$
\begin{equation*}
\phi=\frac{\mathrm{U}(\mathrm{t}) \mathrm{a}^{3}}{2} \mathrm{G} \tag{21}
\end{equation*}
$$

where

$$
\begin{align*}
\mathrm{G}= & \frac{\mathrm{y}+\mathrm{h}_{\mathrm{o}}}{\mathrm{r}_{\mathrm{o}}{ }^{3}}+\sum_{v=1}^{\infty}(-1)^{v} \prod_{\mathrm{p}=1}^{v}\left(\frac{\mathrm{a}}{\mathrm{~h}_{\mathrm{o}}+\mathrm{h}_{\mathrm{p}-1}}\right)^{3}\left(\frac{\mathrm{y}+\mathrm{h}_{v}}{\mathrm{r}_{v}{ }^{3}}\right) \\
& +\frac{\mathrm{y}-\mathrm{h}_{\mathrm{o}}}{\mathrm{~s}_{0}^{3}}+\sum_{v=1}^{\infty}(-1)^{v} \prod_{\mathrm{p}=1}^{v}\left(\frac{\mathrm{a}}{\mathrm{~h}_{\mathrm{o}}+\mathrm{h}_{\mathrm{p}-1}}\right)^{3}\left(\frac{\mathrm{y}-\mathrm{h}_{v}}{\mathrm{~s}_{v}{ }^{3}}\right) \tag{22}
\end{align*}
$$

and

$$
\begin{array}{ll}
\mathrm{r}_{v}{ }^{2}=\mathrm{R}^{2}+\left(\mathrm{y}+\mathrm{h}_{v}\right)^{2} & (v=0,1,2, \ldots) \\
\mathrm{s}_{\nu}{ }^{2}=\mathrm{R}^{2}+\left(\mathrm{y}-\mathrm{h}_{\nu}\right)^{2} & (v=0,1,2, \ldots) \tag{24}
\end{array}
$$

It should be noted that the $h_{p}$ as well as $h_{o}$ are functions of $t$. This is not shown explicitly in Eq. (19), (20), and (22) to avoid complication.

Differentiating Eq. (21) with respect to time,

$$
\begin{equation*}
\frac{\partial \phi}{\partial t}=\frac{\dot{\mathrm{U}}(\mathrm{t}) \mathrm{a}^{3}}{2} \mathrm{G}+\frac{\mathrm{U}(\mathrm{t}) \mathrm{a}^{3}}{2} \frac{\partial \mathrm{G}}{\partial \mathrm{t}} \tag{25}
\end{equation*}
$$

The terms of $\partial G / \partial t$ contain factors of the order of $U(t)$ which decrease and vanish as $U(t) \longrightarrow 0$. It can be shown that $\partial G / \partial t$ converges for finite $U(t)$ and vanishes for $U(t)=0$. Under the assumption that $U(t)$ is small over the time interval considered, Eq. (25) becomes

$$
\begin{equation*}
\frac{\partial \phi}{\partial t}=\frac{\dot{U}(t) a^{3}}{2} G \tag{26}
\end{equation*}
$$

The pressure on the sphere is given by

$$
\begin{equation*}
p=\rho\left(\frac{\partial \phi}{\partial t}-g y-\frac{\left|v^{2}\right|}{2}\right) \tag{27}
\end{equation*}
$$

The term gy in Eq. (27) on integration over the sphere gives the hydrostatic or buoyancy force. This does not contribute to the fluid resistance and will be neglected. Experimentally, this force can be eliminated by making the sphere neutrally buoyant. The velocity $|\vec{v}|$ is of the order of $U(t)$ and hence the quadratic term $\left|v^{2}\right|$ will be neglected. From Eq. (26) and (27), and integrating the pressure, $p$, over the sphere surface, we obtain for the force, $F$, arising from fluid resistance on the sphere,

$$
\begin{align*}
F & =-2 \pi a^{2} \int_{0}^{\pi} p \sin \theta \cos \theta d \theta \\
& =-\dot{U}(t)\left[\pi a^{5} \rho \int_{0}^{\pi} G \sin \theta \cos \theta d \theta\right] \tag{28}
\end{align*}
$$

This may be written

$$
\begin{equation*}
K_{0}=\frac{F}{-\frac{4}{3} \pi a^{3} \rho \dot{U}(t)}=\frac{3}{4} a^{2} \int_{0}^{\pi} G \sin \theta \cos \theta d \theta \tag{29}
\end{equation*}
$$

where $K_{o}$, the added mass coefficient, is the initial apparent increase in the mass of the sphere on acceleration from rest neglecting gravitation effects ( $g=0$ ), expressed in terms of the mass of fluid displaced by the sphere. In order to compute the added mass coefficients it is convenient to introduce the terms

$$
\begin{aligned}
\mathrm{b}_{\mathrm{s} v}= & \text { distance of the } v \text { th doublet within the sphere } \\
& \text { from the primary sphere doublet } \\
\mathrm{b}_{\mathrm{i} v}= & \text { distance of the } v \text { th free surface image } \\
& \text { doublet from the primary sphere doublet }
\end{aligned}
$$

Then

$$
\begin{align*}
& b_{s v}=\frac{a^{2}}{h_{0}+h_{v-1}}  \tag{30}\\
& b_{i v}=2 h_{o}-\frac{a^{2}}{h_{o}+h_{v-1}}=2 h_{o}-b_{s v} \tag{31}
\end{align*}
$$

The following expansions of the image doublets in terms of the primary sphere coordinates are obtained from surface zonal harmonic theory (Ref. 7)

$$
\begin{align*}
\frac{y+h_{v}}{r_{v}^{3}} & =\frac{\cos \theta_{s v}}{r_{v}^{2}}=\frac{1}{r_{o}^{2}} P_{1}(\cos \theta)+\frac{2 b_{s v}}{r_{o}^{3}} P_{2}(\cos \theta) \\
& +\frac{3 b_{s v}^{2}}{r_{o}^{4}} P_{3}(\cos \theta)+\ldots,\left(r_{o}>b_{s v}\right)  \tag{32}\\
\frac{y-h_{v}}{s_{v}^{3}} & =\frac{\cos \theta_{i v}}{s_{v}^{2}}=-\left[\frac{1}{b_{i v}^{2}}+\frac{2 r_{o} P_{1}(\cos \theta)}{b_{i v}^{3}}\right. \\
+ & \left.\frac{3 r_{o}^{2} P_{2}(\cos \theta)}{b_{i v}^{4}}+\ldots .\right], \tag{33}
\end{align*}
$$

We now define the non-dimensionalized doublet positions

$$
\begin{equation*}
h_{v}^{\prime}=\frac{h_{v}}{a}, \quad(v=0,1,2, \ldots) \tag{34}
\end{equation*}
$$

From Eqs. (20), (30), (31), and (34)

$$
\begin{align*}
& h_{v}^{\prime}=h_{o}^{\prime}-\frac{1}{h_{o}^{\prime}+h_{v-1}^{\prime}} \\
& b_{s v}^{\prime}=\frac{1}{h_{o}^{\prime}+h_{v-1}^{\prime}}  \tag{35}\\
& b_{i v}^{\prime}=2 h_{o}^{\prime}-\frac{1}{h_{o}^{\prime}+h_{v-1}^{\prime}}=2 h_{o}^{\prime}-b_{s v}^{\prime}
\end{align*}
$$

Note that $b_{\text {so }}^{\prime}=0$ and $b_{i o}^{\prime}=2 h_{o}^{\prime}$. To evaluate Eq. (29), we substitute Eq. (22) for $G$ in the integrand. Making use of Eqs. (30) through (35) and the integral properties of surface zonal harmonics, integrals of the form

$$
\left.\begin{array}{l}
a^{2} \int_{0}^{\pi} \frac{y+h_{v}}{r_{v}{ }^{3}} \cos \theta \sin \theta d \theta=\frac{2}{3}, \quad\left(r_{o}=a, b_{s v}^{\prime}<1\right) \\
a^{2} \int_{0}^{\pi} \frac{y-h_{v}}{s_{v}{ }^{3}} \cos \theta \sin \theta d \theta=-\frac{4}{3 b_{i v}^{\prime 3}}, \quad\left(r_{o}=a, b_{i v}^{\prime}>1\right)
\end{array}\right\}
$$

are obtained which apply to the sphere and surface image doublets, respectively. Substituting the values of the integrals, we obtain for
the added mass coefficient

$$
\begin{align*}
\mathrm{K}_{\mathrm{o}}=\frac{1}{2}+\frac{1}{2} \sum_{v=1}^{\infty}(-1)^{v} & \prod_{\mathrm{p}=1}^{v} \mathrm{~b}_{\mathrm{sp}}^{\prime 3}-\frac{1}{\mathrm{~b}_{\mathrm{io}}^{\prime 3}} \\
& -\sum_{v=1}^{\infty}(-1)^{v} \prod_{\mathrm{p}=1}^{v} \frac{\mathrm{~b}_{\mathrm{sp}}^{\prime 3}}{\mathrm{~b}_{\mathrm{i} v}^{\prime 3}} \tag{37}
\end{align*}
$$

It can be shown that Eq. (37) converges for $1 \leq h_{o}^{\prime} \leq \infty$, that is, for all depths of complete sphere submergence. At great depths, $\mathrm{h}_{\mathrm{o}}^{\prime}=\infty$ and $\mathrm{K}_{\mathrm{o}}=0.5$, the added mass coefficient of a sphere in fluid of infinite extent. At the least depth of complete sphere submergence where the sphere is tangent to the fluid surface, $h_{o}^{\prime}=1$ and the added mass coefficient becomes

$$
\begin{equation*}
\mathrm{K}_{\mathrm{o}}=\frac{3}{2} \sum_{v=1}^{\infty} \frac{(-1)^{v-1}}{v^{3}}-1=0.35214 \ldots \tag{38}
\end{equation*}
$$

Values of $K_{o}$ corresponding to various depths are given in Table 1 and $K_{o}$ is shown plotted as a function of sphere depth in Fig. 2.

The preceding theory has been developed on the assumption that $g=0$, i.e., only inertial forces are considered. If we set $g=\infty$, no surface deformation can take place and we have the case of a sphere moving vertically toward a rigid wall (Ref. 7) for which the added mass coefficient is given by

$$
\begin{equation*}
K_{\infty}=\frac{1}{2}+\frac{3}{2}\left(\frac{a}{2 h_{o}}\right)^{3}=\frac{1}{2}+\frac{3}{2}\left(\frac{1}{2 h_{o}^{1}}\right)^{3} \tag{39}
\end{equation*}
$$

where $h_{o}$ is now the distance of the sphere center from the wall
(fluid surface). The added mass coefficient is shown plotted as a function of sphere depth in Fig. 2. Note that $K_{\infty}>K_{o}$ for all depths, $K_{\infty}$ decreasing asymptotically and $K_{o}$ increasing asymptotically to 0.5 as the depth becomes infinite. In experimental studies where sphere displacement and velocity are necessarily involved and $0<g<\infty$, both viscous and gravitational effects would arise and experimentally obtained added mass coefficients, $K$, should be greater than the corresponding theoretical values $K_{o}$, and even greater than $K_{\infty}$ for sufficiently viscous liquids. Where a valid correction can be made for viscous effects, it seems reasonable to assume that $\mathrm{K}_{\mathrm{o}}<\mathrm{K}<\mathrm{K}_{\mathrm{\infty}}$.

## 3. Experiment

A highly polished hollow steel sphere was used in these studies. It was $1.004 \pm 0.0001 \mathrm{in}$. in diameter and weighted to 9.3055 gm . so that it was 0.6411 gm . negatively buoyant with respect to water at an ambient temperature of $20^{\circ} \mathrm{C}$. The sphere was positioned below the center and on the axis of a horizontal electromagnetic coil and accelerated vertically upward by discharging a bank of heavy-duty capacitors through the coil. An Ignitron mercury switch in the circuit prevented current oscillation and limited the electromagnetic effect to a half-wave sinusoidal pulse. The duration of the pulse was about 4 ms . and sphere displacement over the pulse interval less than $3 / 16$ in. By accelerating the sphere in water and in air under otherwise similar conditions, it was possible to determine the added mass of the sphere.

A diagram of the electromagnetic acceleration apparatus is shown in Fig. 3. For launchings in water, the coil was positioned above the water surface to avoid hydrodynamic interference effects and the sphere was positioned at a suitable operating distance below the coil by means of a suspending thread. Varying sphere depths
below the water surface were obtained by adjusting the water level. Since the negative buoyancy of the sphere was negligible, it was not necessary to consider any stored energy in the thread in the impulse to the sphere. For calibration launchings in air, where this would not be the case and where estimation of the energy in the thread would be uncertain, the sphere was supported by a Lucite rod. Because of hydrodynamic interference, a supporting rod could not be used in the tests with water. Tests were conducted in the test section of a non-magnetic Lucite water tank (Ref. 2) whose walls were at least 9 inches distant from the sphere; therefore wall effects could be ignored. An optical technique (Ref. 3) was used to obtain sphere displacement-time data. A more complete description of the electromagnetic accelerating technique, together with the theory involved, is given in Ref. 1.

Experimental data for sphere depths of $1.0,1.5$ and 2.5 diameters below the water surface are given in Table 2. For sphere accelerations in water, if we equate the total impulse acting on the sphere to the momentum change, we obtain

$$
\begin{align*}
\int_{0}^{T / 2}\left(F_{w}-D-G_{w}\right) d t & =\int_{0}^{T / 2} d\left[(M+m) U_{\infty}\right] \\
& =\left[(M+m) U_{\infty}\right]_{t=0}^{t=T / 2}=(M+m) U_{\infty}(T / 2) \tag{40}
\end{align*}
$$

where

$$
\begin{aligned}
& \mathrm{t}=0 \quad=\text { time at start of acceleration regime } \\
& \mathrm{t}=\mathrm{T} / 2=\text { time at end of acceleration regime } \\
& \mathrm{F}_{\mathrm{W}} \quad=\text { magnetic propulsive force in water } \\
& \mathrm{D} \\
& \mathrm{C}_{\mathrm{w}} \quad=\text { viscous drag force due to wall shear stress } \\
& \mathrm{C}_{\mathrm{W}} \quad \text { negative-buoyancy force in water }
\end{aligned}
$$

$$
\begin{aligned}
\mathrm{M} & =\text { mass of sphere } \\
\mathrm{m} & =\text { added mass of sphere } \\
\mathrm{U}_{\mathrm{OO}}(\mathrm{t}) & =\text { velocity of sphere through water }
\end{aligned}
$$

For the sphere launched in air the impulse momentum relationship (the added mass is negligible) is

$$
\begin{equation*}
\int_{0}^{T / 2}\left(F_{a}-G_{a}\right) d t=M U_{a}(T / 2) \tag{41}
\end{equation*}
$$

where

$$
\begin{aligned}
& \mathrm{U}_{\mathrm{a}}(\mathrm{t})=\text { velocity of sphere in air } \\
& \mathrm{F}_{\mathrm{a}}=\text { magnetic propulsive force in air } \\
& \mathrm{G}_{\mathrm{a}}=\text { gravitational force on sphere }
\end{aligned}
$$

Defining the impulses by

$$
\begin{equation*}
\mathrm{I}_{a}=\int_{0}^{\mathrm{T} / 2} \mathrm{~F}_{a} \mathrm{dt} \tag{42}
\end{equation*}
$$

and using Eqs. (40) and (41), the following relationship can be deduced:

$$
\begin{equation*}
\frac{\mathrm{m}}{\mathrm{M}}=\frac{1}{\mathrm{U}_{\infty}(\mathrm{T} / 2)}\left[\frac{\mathrm{U}_{\mathrm{a}}(\mathrm{~T} / 2)+\frac{{ }_{\mathrm{I}_{\mathrm{a}}}}{M}}{\frac{\mathrm{I}_{\mathrm{F}}}{\mathrm{I}_{\mathrm{F}_{\mathrm{w}}}}}-\frac{\mathrm{I}_{\mathrm{D}}}{\mathrm{M}}-\frac{{ }_{\mathrm{I}_{\mathrm{G}}}}{\mathrm{M}}\right]-1 \tag{43}
\end{equation*}
$$

Now the magnetic propulsive force on the sphere is not only a function of time but also a function of sphere position. Since the acceleration of the sphere will be greater in air than in water, its
displacement toward the coil for corresponding times during the pulse period will be greater and its impulse greater. This difference in impulse was estimated by measuring the sphere displacement-time records and numerically integrating the data, making use of electromagnetic theory given in Ref. 1. For the data presented in Table 2, the impulse ratio $\mathrm{I}_{\mathrm{F}_{\mathrm{a}}} / \mathrm{I}_{\mathrm{F}_{\mathrm{W}}}=1.002$. The pulse duration, $\mathrm{T} / 2$, was about 4.02 ms . and $\mathrm{g}=386.1 \mathrm{in} . \mathrm{sec} .^{-2}$. For launchings in air, the sphere left the supporting rod 0.08 ms . after pulse initiation (Ref. 1), but the correction to the magnetic force impulse due to this was found to be negligible. Therefore $I_{G_{a}} / M=1.52 \mathrm{in} . \mathrm{sec}^{-1}$. Now $M=9.3055$ gm. and $G_{w}=0.6411 \mathrm{gm}$. from which $\mathrm{I}_{\mathrm{G}_{\mathrm{w}}}=0.107 \mathrm{in} . \mathrm{sec} .^{-1}$.

In order to compute the drag impulse, $I_{D}$, we will assume that boundary-layer separation did not occur over the acceleration regime, and flow outside the boundary layer was potential or very nearly so. Later we will show that these assumptions are justified. Referring to Eq. (22), the principal contribution to the velocity potential is given by the primary sphere and first surface and sphere image doublets; the contributions of succeeding doublets diminish rapidly, especially at greater depths. We then assume the approximate velocity potential,

$$
\begin{equation*}
\phi=\frac{U(t) a^{3}}{2}\left[\frac{y+h_{o}}{r_{o}^{3}}+\frac{y-h_{o}}{s_{o}^{3}}-\left(\frac{a}{2 h_{o}}\right)^{3} \frac{y+h_{1}}{r_{1}^{3}}\right] \tag{44}
\end{equation*}
$$

From Eqs. (32,), (33) and (44), we obtain the following approximate velocity potential in the neighborhood of the primary sphere

$$
\phi=\frac{U(t) a^{3}}{2}\left[\frac{\cos \theta}{r_{o}^{2}}-\frac{1}{\left(2 h_{o}\right)^{2}}-\frac{2 r_{o} \cos \theta}{\left(2 h_{o}\right)^{3}}-\left(\frac{a}{2 h_{o}}\right)^{3} \frac{\cos \theta}{r_{o}^{2}}\right]
$$

By adding to Eq. (45) the velocity potential for a uniform stream, $U(t)$, in the negative $y$-direction, $\quad \phi=U(t) r_{o} \cos \theta$, we obtain the approximate velocity potential for fluid flow outside the boundary layer

$$
\begin{gather*}
\phi=\frac{U(t) a^{3}}{2}\left[\frac{\cos \theta}{r_{0}^{2}}-\frac{1}{\left(2 h_{0}\right)^{2}}-\frac{2 r_{0} \cos \theta}{\left(2 h_{o}\right)^{3}}-\left(\frac{a}{2 h_{0}}\right)^{3} \frac{\cos \theta}{r_{0}^{2}}\right] \\
 \tag{46}\\
+U(t) r_{0} \cos \theta
\end{gather*}
$$

Then the fluid velocity outside the boundary layer over the sphere is given by

$$
\begin{equation*}
-\left(\frac{\partial \phi}{r_{0} \partial^{\partial}}\right)_{r_{0}=a}=\frac{3 U(t)}{2}\left[1-\frac{1}{\left(2 h_{0}^{1}\right)^{3}}\right] \sin \theta \tag{47}
\end{equation*}
$$

Equation (47) indicates that depth has only a slight effect on the initial flow over the sphere on starting from rest. At great depth, Eq. (46) assumes the velocity potential of a sphere in uniform fluid flow of infinite extent. For unit velocity from Eq. (47),

$$
\begin{equation*}
U(\theta)=\frac{3}{2}\left[1-\frac{1}{\left(2 h_{0}^{\prime}\right)^{3}}\right] \sin \theta \tag{48}
\end{equation*}
$$

From theory (Ref. 1 and 4), the drag impulse is given by

$$
\begin{align*}
& 2 \rho \sqrt{\frac{\nu}{\pi}} \int_{0}^{\pi} 2 \pi \mathrm{a}^{2} \sin ^{2} \theta \mathrm{U}(\theta) \mathrm{d} \theta \\
& \int_{0}^{\mathrm{T} / 2} \mathrm{U}_{0}(\mathrm{~T} / 2) \int_{0}^{\mathrm{T} / 2} \sqrt{\frac{\mathrm{~T}}{2}-\tau} \mathrm{e}^{-2 a \tau_{\sin }{ }^{2} \omega_{\tau} \mathrm{d} \mathrm{~d}_{\tau}} \tag{49}
\end{align*}
$$

where

$$
\begin{aligned}
\rho \quad= & \text { density of water, } 0.998 \mathrm{gm} \cdot \mathrm{~cm} .^{-3} \text { at } 20^{\circ} \mathrm{C} \\
v= & \text { kinematic viscosity of water, } 1.007 \times 10^{-2} \mathrm{~cm} .{ }^{2} \mathrm{sec} .^{-1} \\
& \text { at } 20^{\circ} \mathrm{C} \\
a= & \text { damping constant of accelerating circuit, } \\
& 1.955 \times 10^{2} \text { rad.sec. }{ }^{-1} \\
\omega= & \text { natural frequency of accelerating circuit, } \\
& 7.789 \times 10^{2} \text { rad.sec. }{ }^{-1} \\
\mathrm{a}= & \text { radius of sphere, } 0.502 \text { in., } 1.275 \mathrm{~cm} . \\
\mathrm{T} / 2= & \text { pulse duration, } 4.02 \times 10^{-3} \mathrm{sec} . \\
\mathrm{U}(\mathrm{~T} / 2)= & \text { velocity of sphere in water at end of pulse regime, } \\
& \text { in. sec. }
\end{aligned}
$$

The units of measurement have been mixed for convenience. In the laboratory the scale used for weighing was calibrated in grams and measurements of distances were made in inches.

Drag impulses for various sphere depths were calculated from Eq. (48) and Eq. (49) and the values of $U_{\infty}(T / 2)$ given in Table 2. The right-hand integral in Eq. (49) was evaluated by Simpson's Rule and found to be $4.214 \times 10^{-5} \mathrm{sec} .^{3 / 2}$. The other two integrals are easily evaluated, the left-hand integral in the numerator (considering only the integrand $\sin ^{3} \theta$ ) having the value $4 / 3$ and the integral in the denominator the value $9.670 \times 10^{-4} \mathrm{sec}$.

The results show that $I_{D}=5.28,5.23$ and 5.25 gm. in. sec. ${ }^{-1}$ for sphere depths of $2.5,1.5$ and 1.0 diameters respectively. Substituting these values and the values of $\mathrm{U}_{\mathrm{a}}(\mathrm{T} / 2)$ and $\mathrm{U}_{\infty}(\mathrm{T} / 2)$ from Table 2 into Eq。 (43) we obtain $\mathrm{m} / \mathrm{M}$, the added-mass-to-sphere-mass ratio. The mass of fluid displaced by the sphere is $M-G_{w}=9.3055-0.6411$ $=8.6644 \mathrm{gm}$. , and therefore $\mathrm{M} /\left(\mathrm{M}-\mathrm{G}_{\mathrm{w}}\right)=1.0740$. Multiplying $\mathrm{m} / \mathrm{M}$ by this factor we obtain $\mathrm{m} /\left(\mathrm{M}-\mathrm{G}_{\mathrm{w}}\right)=\mathrm{K}$, the added mass coefficient.

For a comparison of theory and experiment to be valid, it is necessary to show that boundary-layer separation did not occur and fluid flow outside the boundary layer was essentially potential over the acceleration regime. The maximum (final) Reynolds number is $3.41 \times 10^{4}$, well below the critical $\left(3 \times 10^{5}\right)$, and laminar boundarylayer flow should obtain. E. Boltze showed (Ref. 6) that the distance a sphere launched impulsively from rest travels before separation starts is $S=0.392$ a. H. Blasius (Ref. 6) showed in the case of twodimensional flow that separation occurs at longer distances from the starting point for constant acceleration than for motion started impulsively. In brief, motion started impulsively appears to be the worst case. In these tests $\mathrm{a}=0.502 \mathrm{in}$. and if the motion were started impulsively $S$ would be 0.197 in. From data measurements, the displacement of the sphere during the force pulse was at most $0.188 \mathrm{in} .$, and since its acceleration was a damped half-sine wave with respect to time, the sphere must have travelled some distance after pulse termination before separation began.

Added mass coefficients have been calculated from the data of Table 2 making use of the above theory. They are also presented in Table 2 together with fiducial limits corresponding to the 0.9546 and 0.9973 probability levels. These were calculated from statistical theory given in Ref. 1. Added mass coefficients obtained under similar conditions (Ref. l) have been subsequently corrected for the drag impulse and their fiducial limits calculated. These data,
given in Table 3, are slightly more accurate than those of Table 2, for they were obtained at a later time when more precise techniques for positioning the sphere with respect to the coil had been developed and electrical instrumentation for more accurately controlling launching conditions had been obtained. This probably accounts for the data of Table 2 being consistently below that of Table 3 and the low experimental value for a sphere depth of 2.5 diameters.

## 4. Discussion and Conclusions

The data of Tables 1 and 2 which are illustrated in Fig. 2 show good agreement between theory ( $\mathrm{K}_{\mathrm{o}}$ ) and experiment. In no case did the experimental results and theory deviate significantly (0.9546 probability level), although the deviation for a depth of 2.5 sphere diameters is close to this level.

The data of Tables 1 and 3 show good agreement for sphere depths of $1 / 2$ and $5 / 8$ diameter, but deviate significantly ( 0.9546 probability level) for a depth of 1-1/8 diameters and highly significantly (0.9973 probability level) for depths of $3 / 4,7 / 8,1,1-1 / 4$, and $1-3 / 8$ diameters.

From Fig. 2 it is seen that these significant deviations all lie above the theoretical curve for $K_{o}$ and below the theoretical curve for $K_{\infty}$, indicating that there may have been some gravitational effect during the acceleration regime. However, the good agreement between theory ( $\mathrm{K}_{\mathrm{O}}$ ) and experiment from the standpoint of overall added mass coefficient values indicates that the theory provides a good approximation to the experimental initial added mass.

It is concluded that for briefer acceleration regimes, viscous and gravitational effects would decrease and the agreement between theory and experiment would improve.

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## References

1. Mellsen, S. B., Ellis, A. T. and Waugh, J. G., "On the Added Mass of a Sphere in a Circular Cylinder Considering Real Fluid Effects," California Institute of Technology, Hydrodynamics Laboratory Report No. E-124.1, March 1966.
2. Waugh, J. G. and Ellis, A. T., "The Variable-Atmosphere Wave Tank, " Cavitation Research Facilities and Techniques, A.S.M.E., 1964.
3. Waugh, J. G., Ellis, A. T., and Mellsen, S. B., "Techniques for Metric Photography, " J. of the Society of Motion Picture and Television Engineers, Vol. 75, No. 1, January 1966.
4. Mellsen, S. B., Waugh, J. G., and Ellis, A. T., "Real Fluid Effects on an Accelerated Sphere Before Boundary-Layer Separation, "A.S.M.E. Paper 66-WA/UNT-6, 1966.
5. Birkhoff, G., "Hydrodynamics," Princeton University Press, 1960, Chapter 6.
6. Schlichting, H., "Boundary Layer Theory," 4th Edition, McGrawHill, 1960, pp. 120, 185, 218-221.
7. Milne-Thomson, L. M., "Theoretical Hydrodynamics," Macmillan and Co., 1949, Chapter 16, pp. 437-447.
8. Khintchine, A. Y., "Continued Fractions," P. Noordhoff, Ltd., Groningen, The Netherlands, 1963, p. 15.

## Nomenclature

a - radius of sphere
$b_{i v}$
$b_{s v} \quad-$ distance of the $v$ th image doublet within the sphere from the primary sphere doublet
$C(t)$ - arbitrary function of time
D - viscous drag force on sphere due to wall shear stress
F - fluid resistance on sphere due to inertial effects
$\mathrm{F}_{\mathrm{a}}$ - magnetic propulsive force on sphere in air
$\mathrm{F}_{\mathrm{w}}$ - magnetic propulsive force on sphere in water
g - gravitational acceleration
$\mathrm{G}_{\mathrm{a}}$

G
$h_{v}$

K

M - mass of sphere
p - fluid pressure


Table 1

Theoretical Initial Added Mass Coefficient of a Sphere Accelerated Vertically From Rest Below an Ideal Fluid Surface ( $\mathrm{g}=0$ )

## Initial Depth of Sphere Center (radii)

Added Mass
Coefficient
K。

| 1 | 0.3521 |
| :--- | :--- |
| 1.1 | 0.3806 |
| 1.2 | 0.4038 |
| 1.3 | 0.4218 |
| 1.4 | 0.4360 |
| 1.5 | 0.4473 |
| 1.75 | 0.4661 |
| 2 | 0.4770 |
| 2.25 | 0.4837 |
| 2.5 | 0.4881 |
| 2.75 | 0.4910 |
| 3 | 0.4931 |
| 4 | 0.4970 |
| 5 | 0.4985 |
| $\infty$ | 0.5000 |

1.1
0.3806
1.2
0.4038
1.3
0.4218

1. 4
0.4473
2. 75
0.4661

2
0.4837
2.5
0.4881
2. 75
0.4931

4
$\infty$
0.5000

## Table 2

Effect of Depth of Submergence on the Added Mass of a 1.004-Inch Diameter Sphere Accelerated Vertically Upward From Rest in Water

|  | Launchings in water |  |  | Launchings |
| :---: | :---: | :---: | :---: | :---: |
| Depth of sphere center below water surface (diameters) | 2.5 | 1.5 | $\underline{1.0}$ | in air |
| Sphere velocities just after acceleration regime (inches/second) | 52.73 | 53.03 | 52.95 | 76.03 |
|  | 52.53 | 51.80 | 53.29 | 75.75 |
|  | 52.35 | 52.35 | 53.02 | 75.83 |
|  | 51.99 | 52.29 | 52.85 | 75.16 |
|  | 53.04 | 51.95 | 52.61 | 75.14 |
|  |  | 52.18 | 53.23 | 75.49 |
| Average sphere velocity just after acceleration regime (in/sec) | 52.528 | 52.267 | 52.991 | 75.567 |
| Average added mass coefficient (K) | 0.487 | 0.495 | 0.474 |  |
| Average added mass coefficient (95.46\% limits) | 0.475 | 0.483 | 0.465 |  |
|  | 0.499 | 0.507 | 0.482 |  |
| Average added mass coefficient (99.73\% limits) | 0.469 | 0.477 | 0.461 |  |
|  | 0.506 | 0.513 | 0.486 |  |

Table 3

Effect of Depth of Submergence on the Added Mass of a 1.004-Inch Diameter Sphere Accelerated Vertically Upward From Rest in Water**

| Initial Depth <br> of Sphere Center <br> (diameters) | Average <br> Added Mass <br> Coefficient <br> (K) | Fiducial Limits for Average <br> Added Mass Coefficient <br> $95.46 \%$ | $99.73 \%$ |
| :---: | :---: | :---: | :---: |
| $1 / 2$ | 0.348 | $\pm 0.009$ | $\pm 0.013$ |
| $5 / 8$ | 0.415 | $\pm 0.006$ | $\pm 0.009$ |
| $3 / 4$ | 0.471 | $\pm 0.015$ | $\pm 0.023$ |
| $7 / 8$ | 0.477 | $\pm 0.006$ | $\pm 0.009$ |
| 1 | 0.488 | $\pm 0.006$ | $\pm 0.009$ |
| $1-1 / 8$ | 0.492 | $\pm 0.006$ | $\pm 0.009$ |
| $1-1 / 4$ | 0.502 | $\pm 0.007$ | $\pm 0.010$ |
| $1-3 / 8$ | 0.504 | $\pm 0.005$ | $\pm 0.007$ |

* The average added mass coefficients were obtained from six tests at each depth and were originally presented in Table 7 of Reference 1. They have subsequently been corrected for the drag impulse and their fiducial limits computed.


Figure 1 - Coordinate Systems for Fluid and Sphere


Figure 2 - Effect of Water Surface Proximity on the Added Mass of a 1-Inch
Diameter Sphere Accelerated Vertically Upward from Rest


Figure 3 - Techniques Used to Accelerate Sphere Under Water and In Air

## Appendix

## $\underline{\text { Limit Points of Image Doublet Positions }}$

It is of interest to determine the limit point of the primary sphere image doublet positions and hence the surface image doublet positions. For convenience, we assume positive values for the sphere doublet positions. It is obvious from their definition that the primary sphere doublet positions, $h_{v}^{\prime}$, must lie on the upper vertical radius of the primary sphere. From the recurrence relations, Eq. (35)

$$
\begin{equation*}
h_{0}^{\prime}>h_{1}^{\prime}>h_{2}^{\prime}>\ldots>h_{v}^{\prime}>h_{v+1}^{\prime}>\ldots \tag{50}
\end{equation*}
$$

and the $h_{v}^{\prime}$ form a monotonically decreasing sequence. On the other hand, the point $h_{o}^{\prime}-1$ which lies on the upper vertical radius and circumference of the primary sphere is the greatest lower bound of points on the upper vertical radius and hence a lower bound of doublet positions. Since the doublet positions are bounded it follows from well-known limit theory that they must approach a limit. That is,

$$
\begin{equation*}
\lim _{v \rightarrow \infty}\left(h_{v-1}^{\prime}-h_{v}^{\prime}\right)=0, \quad \lim _{v \rightarrow \infty} h_{v-1}^{\prime}=h_{v}^{\prime} \tag{51}
\end{equation*}
$$

From the recurrence relations, Eq. (35) we obtain

$$
\begin{equation*}
h_{v}^{\prime} h_{v-1}^{\prime}=h_{o}^{\prime 2}-1+h_{0}^{\prime}\left(h_{v-1}^{\prime}-h_{v}^{\prime}\right) \tag{52}
\end{equation*}
$$

Then in the limit as $\nu \longrightarrow \infty$

$$
\begin{equation*}
\lim _{v \rightarrow \infty} h_{v}^{\prime 2}=h_{o}^{\prime 2}-1 \tag{53}
\end{equation*}
$$

or

$$
\begin{equation*}
\lim _{v \rightarrow \infty} h_{v}^{\prime}=\left(h_{o}^{\prime 2}-1\right)^{\frac{1}{2}} \tag{54}
\end{equation*}
$$

and $\sqrt{h_{0}^{\prime 2}-1}$ is the greatest lower bound or limit point. To show that the limit point (and hence sphere image doublets) lies within the primary sphere doublet, it is sufficient to show that

$$
\begin{equation*}
\left(h_{0}^{\prime 2}-1\right)^{\frac{1}{2}} \geqslant h_{o}^{\prime}-1 \tag{55}
\end{equation*}
$$

Since $\quad h_{0}^{\prime} \geqslant 1$

$$
\begin{equation*}
h_{0}^{\prime 2}-1 \geqslant h_{0}^{\prime 2}-2 h_{0}^{\prime}+1=\left(h_{0}^{\prime}-1\right)^{2} \tag{56}
\end{equation*}
$$

and taking positive square roots, Eq. (55) follows.
The convergence of the doublet positions may also be established directly from the recurrence relations Eq. (35). From these the $h_{\nu}^{\prime}$ can be shown as continuing fractions

$$
\begin{equation*}
h_{v}^{\prime}=h_{0}^{\prime}-\frac{1}{2 h_{0}^{\prime}-} \frac{1}{2 h_{0}^{\prime}-} \frac{1}{2 h_{0}^{\prime}-} \cdots \frac{1}{2 h_{0}^{\prime}} \tag{57}
\end{equation*}
$$

to $v$ terms. From the theory of continued fractions (Ref. 8), the necessary and sufficient condition that $h_{v}{ }^{\prime}$ converges as $v \rightarrow \infty$ is that

$$
\sum_{v=1}^{\infty}\left(2 h_{0}^{\prime}\right)_{v}
$$

diverges, which it obviously does since $h_{o}^{\prime} \geqslant 1$. Hence $h_{v}^{\prime}$ approaches a limit as $v \longrightarrow \infty$, and since $h_{v}$ decreases monotonically, it follows from limit theory that the limit is the greatest lower bound. The determination of the limit point follows, as shown above.

The limit point for surface image doublets is simply the reflection of the primary sphere image limit point in the surface. When the primary sphere is tangent to the fluid surface, both limit points coincide and lie on the surface point of tangency of the primary sphere doublet.
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Theory is developed to estimate the effect of free surface proximity on the initial added mass of a sphere accelerated vertically upward from rest in an ideal fluid. It is assumed that the acceleration regime is sufficiently brief that inertial forces predominate and gravitational effects may be neglected. Results of tests in water indicate that while there are slight viscous and gravitational effects over the acceleration regime, the agreement between theory and experiment is good. It is concluded that over briefer acceleration regimes these effects would decrease and the agreement would improve.

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