

In Situ Measurement Activities at the NASA Orbital Debris Program Office

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Outline



- Objective
- Past projects
 - PINDROP
 - LAD-C
 - LADEE-DEBIE
 - MMSE
 - FOMIS
 - IMMUSE
- Present project
 - SDS
- Future project
 - High LEO altitude DRAGONS

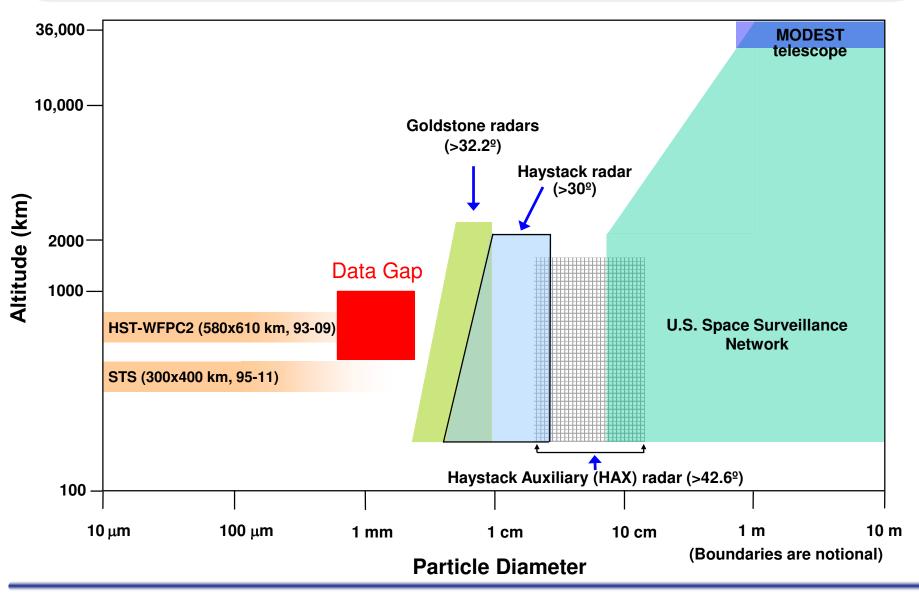
Objective



- Conduct in situ measurements to better characterize the small (millimeter or smaller) orbital debris (OD) populations in low Earth orbit (LEO)
 - Orbital debris <3 mm pose the highest penetration risk to most spacecraft.
 - For the flux for particles <3 mm, orbital debris model validation for altitudes above 600 km is most effective using in situ data.

(NASA/TM-2015-218780)

Orbital Debris Measurement Coverage



Past Projects (1/6)

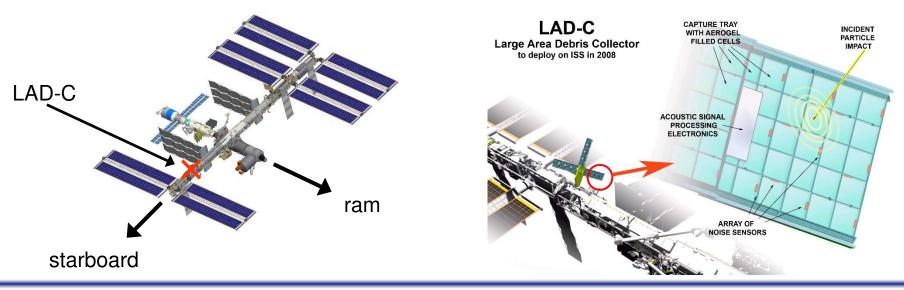


- <u>Particle Impact Noise Detection and Ranging on</u> Autonomous <u>Platform (PINDROP)</u>
 - Objective: To develop acoustic sensors for MMOD impact detections
 - Funding source: NASA's Planetary Instrument Definition and Development Program (PIDDP), 2003-2005
 - Team: Naval Research Lab (NRL), NASA Orbital Debris Program Office (ODPO)
 - Best sensor material: Polyvinylidene fluoride (PVDF)
 - Hypervelocity impact targets (with PVDF sensors attached) tests:
 - Al, aerogel, mylar, circuit board, Kevlar, spectra shield, multi-layer insulation

Past Projects (2/6)



- Large Area Debris Collector (LAD-C) on the ISS
 - Objective: To characterize the large (0.1 1 mm) MMOD environment at ISS altitude with 10 m² aerogel and PVDF acoustic sensors
 - U.S. funding sources: DoD STP, NASA ODPO
 - Team: NRL/USNA, NASA ODPO, ESA, JAXA, Kent (UK)
 - DoD SERB Nov 2004, PDR May 2006, terminated Jan 2007



Past Projects (3/6)



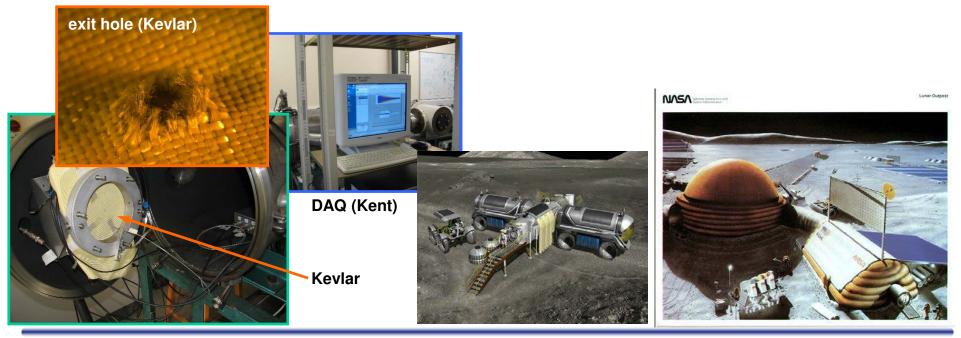
- <u>Lunar Atmosphere and Dust Environment Explorer</u> (LADEE) mission opportunity (2011)
 - Objective: A build-to-print approach to deploy a <u>Deb</u>ris <u>In-orbit</u> <u>E</u>valuator (DEBIE) to characterize the sub-µm lunar horizon glow particles at ~50 km above the lunar surface
 - Components: Impact ionization sensor, acoustic momentum sensor
 - Team: ODPO, ESA, MPIK, MSFC, NRL, UM, USNA, VT
 - Status: Proposal was not selected (December 2008)



Past Projects (4/6)



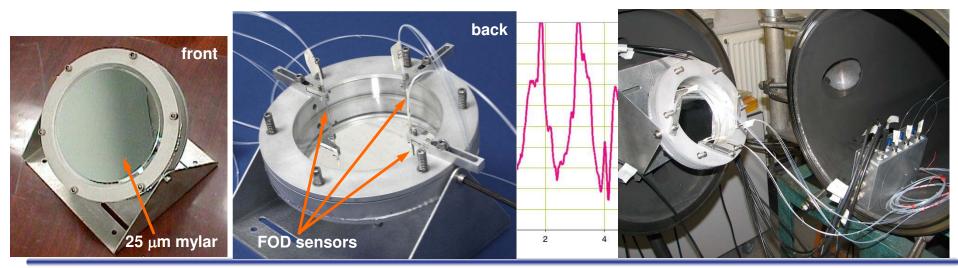
- <u>Micrometeoroid and Lunar Secondary Ejecta (MMSE)</u>
 - Objective: To explore the feasibility of using acoustic sensors to monitor potentially damaging impacts on large-scale instruments and lunar habitats
 - U.S. funding sources: NASA HQ OCE, NASA ODPO
 - Team: ODPO, Kent (UK), NRL, USNA, VT



Past Projects (5/6)



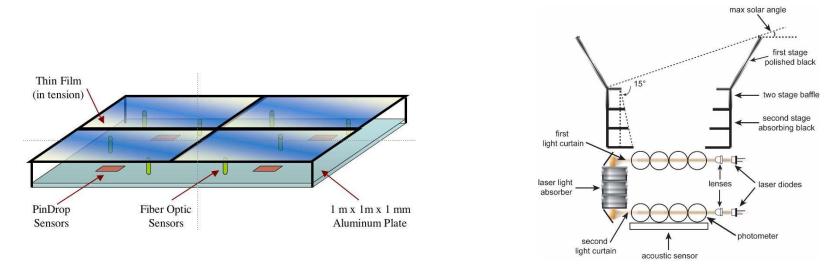
- <u>Fiber Optic Micrometeoroid Impact Sensor (FOMIS)</u>
 - Objective: To develop prototype FOMIS units that can be assembled into a large-area (≥tens of m²), low-cost, low-mass, and low-power MMSE impact detector on the lunar surface
 - Components: Thin film under tension, fiber optic displacement (FOD) sensors
 - U.S. funding sources: JSC (2008-2012)
 - Team: ODPO, Kent (UK), NRL, VT



Past Projects (6/6)



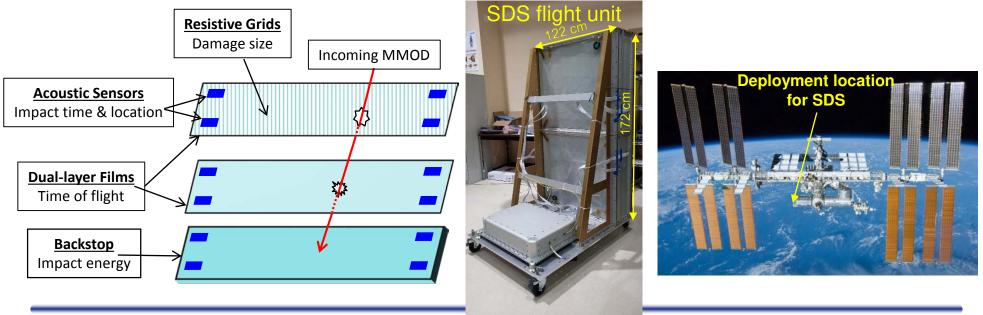
- <u>Impact Sensor for Micrometeoroid and Lunar Secondary</u>
 <u>Ejecta (IMMUSE)</u>
 - Objective: "Pre-Phase A" development of two integrated MMSE impact sensor systems
 - Components: FOMIS, PINDROP, dual-layer laser curtain sensors
 - U.S. funding sources: HQ SMD and ESMD (2009-2012)
 - Team: ODPO, Kent (UK), NRL, USNA, VT, MSFC



Present Project



- Space Debris Sensor (SDS)
 - The SDS, *i.e.*, the ISS Debris <u>Resistive/A</u>coustic <u>G</u>rid <u>O</u>rbital <u>Navy-NASA</u> <u>Sensor</u> (DRAGONS) has been funded by the ISS Technology Demonstration Office for an ISS deployment in early 2018
 - The objectives are to mature the DRAGONS technologies and demonstrate its capability to measure impact time, impact flux, particle size, impact speed, impact direction, and impact energy/particle density



Future Project



- The ODPO is actively pursuing flight opportunities to deploy a large area (≥1 m²) DRAGONS to 700-1000 km altitude
 - Rideshare
 - Secondary payload
- The ODPO will continue to improve DRAGONS and seek opportunities for collaboration to address the key millimeter-sized orbital debris data gap in LEO