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Increased Efficiency in SI Engine with Air Replaced by Oxygen in Argon Mixture

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Abstract

Basic engine thermodynamics predicts that spark ignited engine efficiency is a function of both the compression ratio of the engine and the specific heat ratio of the working fluid. In practice the compression ratio of the engine is often limited due to knock. Both higher specific heat ratio and higher compression ratio lead to higher end gas temperatures and increase the likelihood of knock. In actual engine cycles, heat transfer losses increase at higher compression ratios and limit efficiency even when the knock limit is not reached. In this paper we investigate the role of both the compression ratio and the specific heat ratio on engine efficiency by conducting experiments comparing operation of a single-cylinder variable-compression-ratio engine with both hydrogen-air and hydrogen-oxygen-argon mixtures. For low load operation it is found that the hydrogen-oxygen-argon mixtures result in higher indicated thermal efficiencies. Peak efficiency for the hydrogen-oxygen-argon mixtures is found at compression ratio 5.5 whereas for the hydrogen-air mixture with an equivalence ratio of 0.24 the peak efficiency is found at compression ratio 13. We apply a three-zone model to help explain the effects of specific heat ratio and compression ratio on efficiency. Operation with hydrogen-oxygen-argon mixtures at low loads is more efficient because the lower compression ratio results in a substantially larger portion of the gas to reside in the adiabatic core rather than in the boundary layer and in the crevices, leading to less heat transfer and more complete combustion.

Keywords: Hydrogen, Internal Combustion Engine, Noble Gas, Argon

1. Introduction

Reducing emissions of carbon dioxide (CO_2) has become increasingly urgent. Several possible paths exist to achieve CO_2 emissions reductions in the transportation sector: increasing efficiency of petroleum fueled engines, increasing use of alternative fuels, and increasing powertrain electrification with batteries or hydrogen fuel cells. While all of these paths hold some degree of promise, the use of renewable hydrogen (H_2) as a transportation fuel will reduce not only CO_2 emissions, but all carbon based emissions, such as, carbon monoxide (CO) and unburned hydrocarbons. Hydrogen fuel cells are efficient and produce no pollution; however, their high cost is a major barrier limiting their introduction to the mass market. H_2 fueled internal combustion (IC) engines are a more viable option since they rely on mature IC engine technology. The only harmful engine-out emissions are oxides of nitrogen (NO_x), since CO_2 and hydrocarbons from oil consumption are negligible in well-tuned engines. Depending on the operating regime, three-way catalysts, NO_x traps, or ultra lean combustion can efficiently control NO_x emissions.

A more radical approach to NO_x control that may also contribute to increased efficiency consists of replacing nitrogen from the air by an optimized diluent. Noble gases such as helium and argon are promising diluents because of their monatomic structure. These gases are nonreactive and have a high specific heat ratio ($\gamma = 1.67$ compared to $\gamma < 1.4$ for air) because noble gases have only one mode of energy storage: translational motion. Basic engine cycle thermodynamics predicts that the use of a gas with high specific heat ratio leads to high engine efficiency. The indicated engine efficiency for an ideal Otto cycle can be written as a function of the engine's compression ratio (CR) and the specific heat ratio of in-cylinder gas [1],

$$\eta_i = 1 - \frac{1}{CR^{\gamma-1}} \quad (1)$$

This relationship between engine efficiency and the specific heat ratio inspired a patent by Laumann et al. [2] in which argon is utilized in an internal combustion (IC) engine. It was proposed that the water from the exhaust would be condensed out and the remaining argon could then be recycled in a closed loop system. This concept was later tested experimentally in a single cylinder spark ignited (SI) Cooperative Fuel Research (CFR) engine by De Boer and Hulet [3]. High efficiency operation was achieved for high concentrations of argon, but knock was a problem at the compression ratios tested, which ranged from 5.5 to 12. Direct injection compression ignition engines do not suffer from knock and might be a better solution to take advantage of the thermodynamic properties of noble gases. Ikegami et al. [4] experimentally tested this approach and achieved indicated thermal efficiencies close to 50% using compression ratios between 10 and 16. The amount of H₂ they could inject at higher compression ratios was limited due to the injection pressure of their system. High pressure H₂ gas injection is difficult due to the high diffusivity and low lubricity of H₂ gas [5]. Therefore, reliable high pressure injection of H₂ gas remains an obstacle for the realization of such an engine.

In the near term a low pressure port injection as opposed to a high pressure direct injection approach is more practical; the following arguments are therefore targeted at port fuel injection spark ignited engines. From Eq. (1) it appears we can arbitrarily increase the compression ratio of the engine and γ to achieve high efficiency. However, Eq. (1) assumes isentropic compression with constant specific heat ratio from bottom dead center (BDC) at the beginning of the compression process to top dead center (TDC) just before the combustion event [1]. Given these assumptions the following relation is used,

$$\frac{T_{TDC}}{T_{BDC}} = \left(\frac{V_{BDC}}{V_{TDC}} \right)^{\gamma-1} = CR^{\gamma-1}, \quad (2)$$

where T_{TDC} is the temperature of the in-cylinder gas at TDC corresponding to the motored (without chemical reactions) temperature and T_{BDC} is the temperature of the gas at BDC and V_{TDC} and V_{BDC} are the cylinder volumes at TDC and BDC, respectively. Substituting Eq. (2) into Eq. (1) we get a relationship for the thermal efficiency for an ideal Otto cycle as a function of temperature,

$$\eta_i = 1 - \frac{T_{BDC}}{T_{TDC}}. \quad (3)$$

Note that we can increase the efficiency by either decreasing T_{BDC} or increasing T_{TDC} . Because the end gas temperature is directly related to the motored temperature of the gas at TDC, T_{TDC} can be considered representative of the knock propensity of the engine. In practice T_{TDC} is typically limited by knock, which effectively limits how much the compression ratio or specific heat ratio can be increased (see Eq. (2)). Therefore, knock often limits the engine's efficiency. This description simply illustrates the tradeoffs between knock, compression ratio, specific heat ratio, and efficiency; in practice knock behavior is far more complicated and depends on additional factors such as engine's geometry (i.e. spark plug location, cylinder shape, bore, location of valves), fuel properties, wall temperatures, and engine speed. The key point is that increasing either compression ratio or specific heat ratio tends to increase indicated efficiency, but also increase the propensity for knock.

Because knock effectively limits the temperature of the in-cylinder gas at TDC it is not obvious that increasing the specific heat ratio of the gas is more beneficial than simply increasing the compression ratio of the engine. In this paper the differences between obtaining high

efficiency in a H₂ fueled SI engine by increasing the compression ratio and increasing the specific heat ratio of the working fluid are investigated. Specifically, H₂-O₂-Ar mixtures are compared with H₂-air mixtures using a variable compression ratio CFR engine.

First we describe the experimental apparatus and tests conducted. Then we present and discuss the experimental results. We finally present a three-zone model of SI combustion to further interpret the experimental results.

2. Experimental Apparatus and Data Acquisition

Experiments were conducted using a single cylinder, port fuel injected, variable compression ratio, ASTM-CFR engine. Engine specifications can be found in Table 1 and a schematic of the experimental configuration is shown in Fig. 1.

In-cylinder pressure was measured using a 6052B Kistler piezoelectric pressure transducer in conjunction with a 5044A Kistler charge amplifier. Intake pressure was measured using a 4045A5 Kistler piezoresistive pressure transducer in conjunction with a 4643 Kistler amplifier module. Sampling of in-cylinder and intake pressures was hardware timed using an optical encoder, and recorded every 0.1 crank angle (CA) degree. An electric motor directly coupled to the engine crankshaft was controlled by an ABB variable speed frequency drive that in turn controlled the engine speed. Mass of H₂ injected was measured using a mass flow sensor (Alicat M Series Mass Flow Meter) and also verified using a Horiba six-gas emissions analyzer. When operating at stoichiometric conditions, $\phi = 1$, a wide band lambda sensor was primarily used.

Three sets of high-pressure gas cylinders were used: one containing pure Ar, the second with a mixture of O₂ and Ar, and the third with pure H₂ supplied the engine at specifiable ratios. Pure O₂ could be blended with Ar, for safety reasons, we use O₂ premixed with Ar, which was then blended with Ar. Flow rates of compressed Ar and Ar-O₂ were determined using calibrated sonic orifices. Software controlled electronic pressure regulators set the flow rates by adjusting the gas pressure upstream of the sonic orifices.

A Motec M4 ECU (Engine Control Unit) controlled spark timing, injection timing, and injection pulse width. Initially, when running the engine on H₂-air mixtures, significant problems with backflash (ignition of the fuel before the intake valve closes) occurred. It was found that the backflash occurred because of residual charge in the spark plug wire created from the previous cycle's ignition [6]. Backflash problems ceased after the spark plug wires in our system were modified as described by Kondo et al. [6].

All tests were conducted without throttling the engine at constant intake pressure of 0.98 bar. The engine was pre-heated by burning H₂-air, bringing the coolant temperature to 75°C. This temperature was maintained for the recorded tests using a closed-loop controller. The crankcase was purged with nitrogen (N₂), preventing accidental explosion of any hydrogen that might have blown by the rings.

3. Experimental Results

We first present results from running the CFR engine on H₂-O₂-Ar mixtures at various Ar concentrations and compression ratios. We then compare these results to operation with H₂-air at various compression ratios.

3.1. Operation with H₂-O₂-Ar

We investigate the H₂-O₂-Ar engine concept with a stoichiometric equivalence ratio ($\phi=1$). The concentration of Ar is used to alter the power produced by the engine (intake pressure is held constant; therefore, increasing the Ar fraction decreases the O₂ fraction, and thus, amount of the H₂-O₂ mixture, reducing power). Previous experiments showed that high Ar concentrations resulted in the highest efficiency [3], leading us to look at Ar concentrations of 84%, 86% and 88% by volume (with the balance being a stoichiometric mixture of H₂ and O₂). The composition of the three blends is listed in Table 2. For each blend the compression ratio is varied from 4.5 to 7, in increments of one half. Compression ratios higher than 7 sporadically result in autoignition of the mixture before the spark fired, and they are therefore not reported. The spark timing is swept at each combination of Ar concentration and compression ratio, and we report the spark timing that produces the maximum brake torque. Knock occurs at advanced spark timing with almost all combinations of Ar concentration and compression ratio. Typically we found that the torque increases with advancing spark timing, but advance is limited by the onset of knock.

The net indicated thermal efficiency is plotted versus compression ratio for the three Ar concentrations in Fig. 2. For most operating points Ar concentration of 86% by volume results in the highest efficiency, then 88% Ar has the next highest efficiency, and 84% Ar typically results in the lowest efficiency. For all three blends of Ar with O₂, the peak efficiency occurs at compression ratio 5.5. The overall highest net indicated thermal efficiency achieved is 44.8% for an Ar concentration of 86% at compression ratio 5.5.

Tables 3 to 5 present performance metrics calculated using the in-cylinder pressure trace for the three H₂-O₂-Ar blends. The indicated mean effective pressure (IMEP) is defined as the work per engine cycle divided by the displacement and allows engine performance comparison over a range of sizes and speeds. The tables present the IMEP averaged over 300 engine cycles (IMEP_{ave}). Table 3 shows that 84% Ar concentration has the highest IMEP_{ave} because this blend has the most fuel of the three blends (Table 2).

The cyclic variability of the operating point is quantified using the coefficient of variation in IMEP (COV_{IMEP}) and is defined as the standard deviation in IMEP from cycle-to-cycle over 300 cycles divided by the mean IMEP and multiplied by 100 to yield a percent. From Table 3 we can see that COV_{IMEP} increases with compression ratio, because spark timing is simultaneously being delayed to avoid knock. The crank angle at which 50% of the fuel energy has been released, CA50 (fourth column in Tables 3-6), similarly illustrates the trend of delaying the combustion process to avoid knock with increasing compression ratio. If knock were not occurring, more spark timing advance and lower COV_{IMEP} could be achieved for some compression ratios.

The last three columns of the table describe the burning process. The crank angle interval from initiation of the spark to the crank angle at which 10% of the fuel energy has been released, CA10-CA0, describes the early flame development period. CA90-CA10 is the crank angle interval required to burn the bulk of the fuel and is the difference between the crank angle at which 90% and 10% of the fuel energy has been released, CA90 and CA10 respectively. Finally, CA90-CA0 defines the overall burning process and is the crank angle interval between spark plug firing and 90% of the fuel energy being released. Figures 3 to 5 graphically show the

burning process for the three blends of H_2 , O_2 and Ar. These plots display the crank angle location of spark ignition, and the crank angles at which 10%, 50%, and 90% of the fuel energy has been released. The trend of delaying spark timing with increasing compression ratio can clearly be seen in these plots. For an Ar concentration of 84% the duration of the early flame development CA10-CA0 decreases with increasing compression ratio; however, the duration of the bulk combustion CA90-CA10 is fairly constant with changes to the compression ratio and it is only lengthened for the highest compression ratio due to the very late spark timing. These trends also hold for Ar concentrations of 86% and 88%. The compression temperature increases with compression ratio and the increased temperature leads to higher flame speeds during the initial flame development period, as will be discussed in more detail in the modeling.

Considering the advances in engine technology that have occurred since the CFR engine was designed, the low engine speed tested here, and that the CFR engine was designed specifically to knock it is expected that the efficiency of this concept could be further improved using a modern engine that is less prone to knock. Furthermore, increasing the combustion rate with advanced ignition systems [7,8] could be used to extend the knock limit. Direct injection compression ignition is also a promising route to achieve higher efficiencies if the injectors can be made more reliable.

3.2. Operation with H_2 -air

For comparison, the engine is also operated with H_2 -air mixtures. We chose an equivalence ratio of 0.24 because this mixture requires the same mass of H_2 as used for H_2 - O_2 -Ar mixtures with 86% Ar. The compression ratio is varied from 6 to 17 and the spark timing is swept at each compression ratio. We report the spark timing that produced the maximum brake torque. We can see by comparing Tables 6 and 4 that the overall burn duration CA90-CA0 is substantially longer

for the H₂-air mixture than for all the H₂-O₂-Ar mixtures. Both the early flame development interval CA10-CA0 and the bulk burn interval CA90-CA10 are between one and a half to two times longer than the 86%Ar mixture. H₂-air flame speed is very sensitive to temperature and this is likely the reason for the difference. The burn behavior of H₂-air does not change much with compression ratio as can also be seen in Fig. 6.

Figure 7 shows the net indicated thermal efficiency plotted versus compression ratio. From this plot we can see that the efficiency is much lower than for the H₂-O₂-Ar mixtures. The efficiency initially increases as the compression ratio is increased from 6, but then quickly levels out and peaks at a compression ratio of 13 to 35.9% and then decreases. Heat transfer becomes increasingly important as the compression ratio is increased and has the effect of reducing engine efficiency.

4. Three-Zone SI Combustion Model

A three-zone model, consisting of a burned zone, unburned zone, and crevices, is used to model combustion in the CFR engine. Because the CFR engine is port fuel injected, the air and fuel are assumed to be perfectly mixed. During the compression stroke, the combustible mixture flows into the crevices. The gas pressure is assumed uniform in all zones such that the mass exchange rate between the zones can be solved simultaneously with energy and species equations [9,10]. The combustion process is divided into two stages. The first stage is the ignition and initiation of the three zones in combustion space. The second stage is the propagation of the flame front. Once a small nucleus of the fuel-air mixture has combusted, the combustion chamber is subdivided into two zones. Following spark ignition, there is a delay period that is modeled empirically [11]. Once combustion has commenced, the mean burned gas front is approximated by a sphere centered on the spark plug location. The burning rate is

modeled as the product of the spherical burning area and the turbulent burning rate [1]. Heat transfer between the gases in all three zones and the cylinder walls at a rate proportional to the chamber surface area wetted by the gases is modeled using the Woschni correlation [12]. During the expansion stroke, unburned gases stored in the crevices flow back into the combustion chamber accounting for unburned fuel produced in the experiment. Intake and exhaust flows are modeled using measured valve lifts. The mass flows through the intake and exhaust valves are modeled with a discharge coefficient $C_d=0.6$. The effect of residual burned gas from previous cycles is included by running the model through eight complete cycles.

4.1. Details of model

The turbulent burning rate is computed as [1]

$$\frac{dm_b}{dt} = \rho_u \cdot A_f \cdot S_T, \quad (4)$$

where m_b denotes the mass of burned mixture, S_T is the turbulent flame speed, ρ_u and A_f are the unburned density and flame surface area, respectively.

An empirical model is used for the turbulent flame speed,

$$S_T = S_L \cdot f \cdot \frac{\rho_u / \rho_b}{(\rho_u / \rho_b - 1) \cdot Y_{mb} + 1}, \quad (5)$$

where S_L is the laminar flame speed, $f = 1 + 0.0018 \cdot \text{RPM}$ is the turbulent enhancement factor, Y_{mb} is the mass fraction of burned mixture, ρ_b is the density of burned mixture. The laminar flame speed is computed as

$$S_L = S_{L0} \left(\frac{T}{T_{ref}} \right)^\alpha \left(\frac{p}{p_{ref}} \right)^\beta \quad (\text{cm/s}). \quad (6)$$

where T and p are the temperature and pressure, T_{ref} and p_{ref} are the reference temperature and pressure, and the parameters S_{L0} , α , and β can be found in Table 8.

4.2. Model Results and Discussion

Given equation (3) it may be expected that similar efficiencies could be achieved by operating the engine with H₂-O₂-Ar and H₂-air mixtures when the same mass of fuel and compression ratios that resulted in similar motored temperatures at TDC were used; however, as shown in Fig. 7 the efficiencies are not the same. This difference is explored using the three zone model; a comparison is made between a H₂-O₂-Ar mixture and a H₂-air mixture with the same mass of fuel and compression ratios that resulted in similar motored temperatures at TDC. We compare blend 2 with 86% Ar concentration by volume to an H₂-air mixture with $\phi = 0.24$. The motored temperature at TDC for the H₂-O₂-Ar mixture with a compression ratio of 6 and the H₂-air mixture with a compression ratio of 16 are both about 900 K, Eq. (3) therefore predicts equal thermal efficiencies.

Figure 8 shows the modeling results. Pressure traces are displayed in (a) for the two conditions, both from the model and experiments. The model results compare well with the experimental pressure, giving us reason to trust the other outputs from the model. The magnitude of the pressure for the two mixtures is quite different; peak pressure for the H₂-air mixture is almost 45 bar, whereas peak pressure for the H₂-O₂-Ar mixture is around 20 bar. The peak pressure has implications on how robust the engine must be since the pressure exerts force on the engine components.

Figure 8 (b) and (c) present the temperatures of the three zones in the model: burned, unburned, and crevices versus crank angle degree; as well as the net heat release for the H₂-O₂-Ar mixture and H₂-air mixture, respectively. Comparing plots (b) and (c) we can see that the burned gas temperature is much higher for the H₂-O₂-Ar mixture; in large part because the heat capacity is much lower for this mixture and the fuel's chemical energy raises the mixtures temperature more than for the H₂-air mixture. From Eqs. (5) and (6) we can see that the flame speed is very sensitive to temperature, which helps explain the differences in burn duration between the two mixtures as shown in Tables 4 and 6 and Figs. 4 and 6. The model also predicts that about 6% of the hydrogen (by mass) remains unburned for the H₂-air mixture due to the large mass in the cold crevices. The amount of gas in the crevices for the H₂-air mixture is substantially larger because of the high compression ratio. We can see from plots (b) and (c) that the net heat release is lower for the H₂-air mixture because of the difference in the amount of unburned fuel. Tang et al. [13] swept the equivalence ratio in a H₂-air SI engine and found that the indicated efficiency dropped off at $\phi=0.22$ due to increases in unburned hydrogen and increased burn duration. Another effect likely important for the experiments is that the highest rates of heat transfer occur in the boundary layer close to the wall, and due to the high compression ratio used for the H₂-air mixture a larger portion of the mixture resides within the boundary layer, therefore higher heat transfer rates are expected [14].

5. Conclusions

- A Compression ratio of 5.5 is the most efficient for each of the three H₂-O₂-Ar blends tested. Peak net indicated thermal efficiency (44.8%) occurred with a dilution level of 86% Ar and compression ratio 5.5.

- For comparison, experiments are conducted with H₂-air at an equivalence ratio of 0.24. Peak net indicated thermal efficiency of 35.9% is reached at compression ratio 13, which is substantially lower than that of the H₂-O₂-Ar mixtures tested.
- A three-zone model is used to compare an H₂-O₂-Ar mixture with an Ar concentration of 86% and compression ratio 6 to an H₂-air with an equivalence ratio of 0.24 and compression ratio 16, these two mixtures have the same mass of fuel and almost identical peak motoring temperatures. The model predicts that the H₂-air mixture has a substantial amount of unburned hydrogen arising from the increased mass residing in the crevices.

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References

- [1] J.B. Heywood, Internal combustion engine fundamentals, New York, McGraw-Hill, 1988.
- [2] E.A. Laumann, R.K. Reynolds, Hydrogen-fueled engine, 1978.
- [3] P. deBoer, J. Hulet, Performance of a hydrogen-oxygen-noble gas engine, Int. J. Hydrogen Energy. 5 (1980) 439-452.
- [4] M. Ikegami, K. Miwa, M. Shioji, A study on hydrogen-fueled compression-ignition engines, Int. J. Hydrogen Energy. (1982) 341-353.
- [5] A.B. Welch, D. Mumford, S. Munshi, J. Holbery, B. Boyer, M. Younkins, et al., Challenges in Developing Hydrogen Direct Injection Technology for Internal Combustion Engines, SAE Paper, No. 2008-01-2379. (2008).
- [6] T. Kondo, S. Iio, M. Hiruma, A Study on the Mechanism of Backfire in External Mixture Formation Hydrogen Engines, SAE Paper, No. 971704. (1997).
- [7] Y. Ikeda, A. Nishiyama, M. Kaneko, Microwave Enhanced Ignition Process for Fuel Mixture at Elevated Pressure of 1MPa, in: 47th AIAA Aerospace Sciences Meeting Including the New Horizons Forum and Aerospace Exposition, 2009.
- [8] A.A. Boretti, H.C. Watson, Enhanced combustion by jet ignition in a turbocharged cryogenic port fuel injected hydrogen engine, International Journal of Hydrogen Energy. 34 (2009) 2511-2516.
- [9] T. Jensen, J. Schramm, A Three-Zone Heat Release Model for Combustion Analysis in a Natural Gas SI Engine – Effects of Crevices and Cyclic Variation on UHC Emissions, SAE

- Paper, No. 2000-01-2802. (2000).
- [10] G. Chin, J. Chen, Development of a Consistent Crevices Model For Multi-Zone Modeling of Piston Engines, in: University of California at Irvine, 2009.
 - [11] M. Al-Baghdadi, A Simulation Model for a Single Cylinder Four-Stroke Spark Ignition Engine Fueled with Alternative Fuels, Turkish J. Eng. Env. Sci. 30 (2006) 331-350.
 - [12] G. Woschni, A universally applicable equation for instantaneous heat transfer in the internal combustion engine, SAE Paper, No. 670931. (1967).
 - [13] X. Tang, D.M. Kabat, R.J. Natkin, W.F. Stockhausen, J.W. Heffel, Ford P2000 Hydrogen Engine Dynamometer Development, SAE Paper No. 2002-01-0242. (2002).
 - [14] J.R. Smith, Optimized hydrogen piston engines, in: 1994 Society of Automotive Engineers' Convergence, 1994.

Tables

Table 1
Engine Specifications

Displacement	Bore	Stroke	Compression Ratio	rpm
0.616 L	82.5 mm	114 mm	4.5-17	900

Table 2
Fuel-Oxidizer-Ar Mixture Concentrations by Volume

	H2	O2	Ar
Blend 1	10.67%	5.33%	84.0%
Blend 2	9.33%	4.67%	86.0%
Blend 3	8.00%	4.00%	88.0%

Table 3
Pressure trace analysis 84% Ar

Compression Ratio	IMEP _{ave} [bar]	COV _{IMEP} [%]	CA50	CA10-CA0	CA90-CA10	CA90-CA0
4.5	4.1	1.1	14.3	20.4	19.5	40.0
5.0	4.3	1.5	17.9	17.4	18.9	36.3
5.5	4.3	1.5	22.5	15.0	19.1	34.0
6.0	4.2	1.6	27.9	14.5	20.3	34.8
6.5	4.2	1.9	28.1	12.4	20.0	32.4
7.0	3.8	2.9	40.2	13.6	25.4	39.0

Table 4
Pressure trace analysis 86% Ar

Compression Ratio	IMEP _{ave} [bar]	COV _{IMEP} [%]	CA50	CA10-CA0	CA90-CA10	CA90-CA0
4.5	3.8	1.0	11.8	27.8	23.5	51.4
5.0	3.7	1.5	16.9	22.1	23.5	45.6
5.5	3.8	1.9	20.7	19.6	24.1	43.8
6.0	3.5	2.3	23.8	17.0	25.9	42.8
6.5	3.6	3.3	25.5	15.2	24.5	39.7
7.0	3.5	3.7	33.1	14.5	30.6	45.2

Table 5
Pressure trace analysis 88% Ar

Compression Ratio	IMEP _{ave} [bar]	COV _{IMEP} [%]	CA50	CA10-CA0	CA90-CA10	CA90-CA0
4.5	3.0	2.0	8.5	46.9	31.2	78.1
5.0	3.2	2.1	13.6	33.2	29.2	62.4
5.5	3.2	2.4	18.7	26.1	30.1	56.2
6.0	3.0	4.4	23.4	23.0	33.0	56.0
6.5	2.6	7.9	30.2	15.6	43.9	59.4
4.5	3.0	2.0	8.5	46.9	31.2	78.1

Table 6
Pressure trace analysis H₂ air, $\phi = 0.24$

Compression Ratio	IMEP _{ave} [bar]	COV _{IMEP} [%]	CA50	CA10-CA0	CA90-CA10	CA90-CA0
6.0	3.2	1.8	5.7	33.8	46.5	80.2
8.0	3.5	2.0	7.0	31.4	45.8	77.1
10.0	3.5	2.0	8.5	30.4	47.5	77.9
12.0	3.5	2.1	10.2	31.2	47.6	78.8
13.0	3.6	2.5	10.3	27.7	48.7	76.4
14.0	3.5	2.6	15.3	27.0	49.6	76.6
15.0	3.5	2.8	17.0	28.5	50.3	78.9
16.0	3.5	2.8	18.3	27.8	50.6	78.4
17.0	3.4	3.3	20.3	30.0	50.2	80.2

Table 7
Fuel-Oxidizer-Ar Mixture Concentrations by Volume

Mixture	S _{L0}	α	β
H ₂ -Ar 84%	131	1.52	-0.17
H ₂ -Ar 86%	92	1.52	-0.17
H ₂ -Ar 88%	56	1.52	-0.17
H ₂ -Air $\phi=0.24$	25.8	2.8	-0.29

Figures

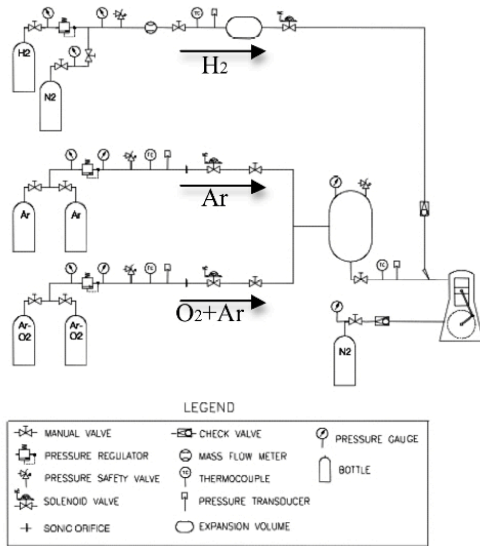


Fig. 1 Schematic diagram of the experimental configuration.

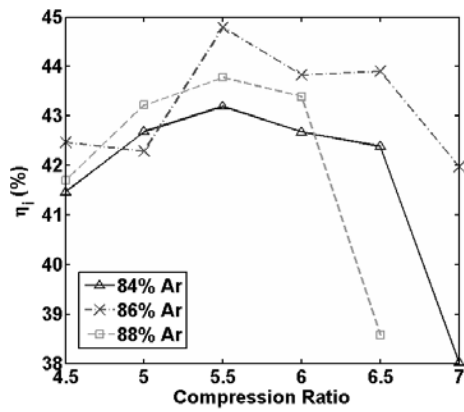


Fig. 2. Indicated thermal efficiency versus compression ratio for the three Ar concentrations. The peak thermal efficiency obtained occurs at a CR of 5.5 for each concentration.

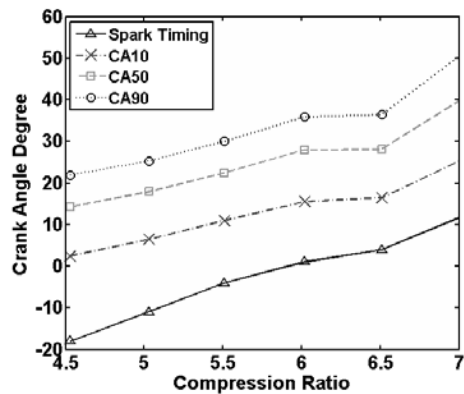


Fig. 3. Evolution of the heat release at each compression ratio for the H₂-O₂-Ar mixture with 84% Ar. This plots shows the MBT spark timings, and the crank angle degrees at which 10%, 50%, and 90% of the fuels energy has been released at each compression ratio.

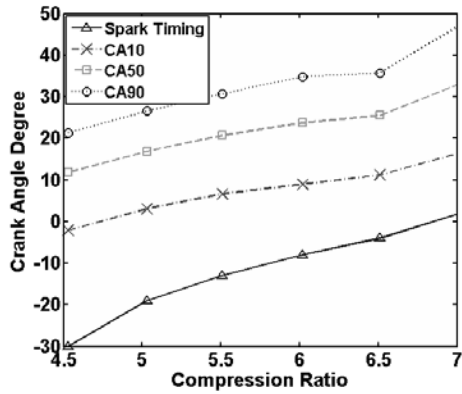


Fig. 4. Evolution of the heat release at each compression ratio for the H₂-O₂-Ar mixture with 86% Ar. This plots shows the MBT spark timings, and the crank angle degrees at which 10%, 50%, and 90% of the fuels energy has been released at each compression ratio.

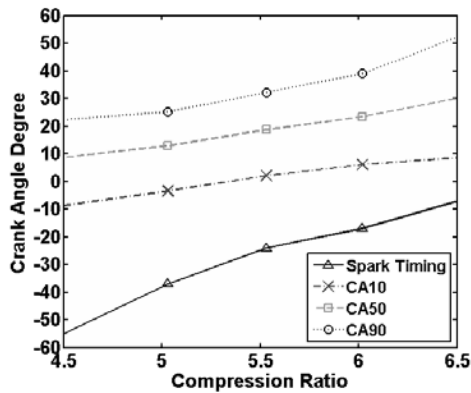


Fig. 5. Evolution of the heat release at each compression ratio for the H₂-O₂-Ar mixture with 88% Ar. This plots shows the MBT spark timings, and the crank angle degrees at which 10%, 50%, and 90% of the fuels energy has been released at each compression ratio.

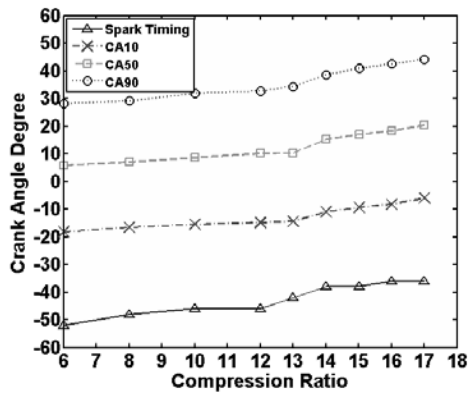


Fig. 6. Evolution of the heat release at each compression ratio for the H₂-air mixture with $\phi=0.24$. This plots shows the MBT spark timings, and the crank angle degrees at which 10%, 50%, and 90% of the fuels energy has been released at each compression ratio.

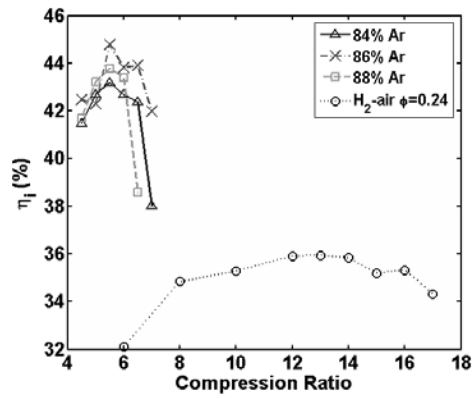


Fig. 7. Indicated thermal efficiency versus compression ratio for H₂-O₂-Ar mixtures of 84%, 86%, and 88% Ar concentration and H₂-air mixture with $\phi=0.24$. The peak thermal efficiency obtained for the H₂-air mixture is much lower than for the H₂-O₂-Ar mixtures.

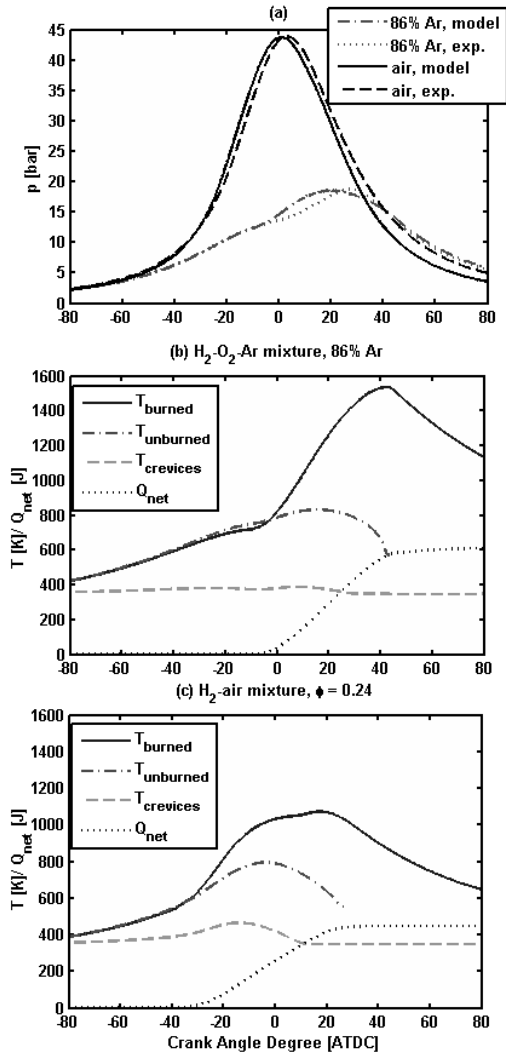


Fig. 8. Modeling comparison between H_2-O_2-Ar mixture of 86% Ar, and H_2 -air mixture with $\phi=0.24$. (a) In-cylinder pressure, (b) temperature in burned zone, temperature in unburned zone, temperature in crevices, and cumulative heat release of 86% Ar mixture with a compression ratio of 6, (c) temperature in burned zone, temperature in unburned zone, temperature in crevices, and cumulative heat release of H_2 -air mixture with a compression ratio of 16.

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