

IRROTATIONAL FLOW ABOUT SHIP FORMS

by

L. Landweber and M. Macagno

Sponsored by
Office of Naval Research
Contract Nonr - 1611(07)

IIHR Report No. 123

Iowa Institute of Hydraulic Research
The University of Iowa
Iowa City, Iowa

December 1969

This document has been approved for public
release and sale; its distribution is unlimited.

IRROTATIONAL FLOW ABOUT SHIP FORMS

Introduction

A method of computing the potential flow about ship forms would, in spite of the neglect of viscous effects, be valuable in the preliminary design of a ship, or in investigating means of improving the performance of an existing ship. If an efficient procedure for performing such calculations were available, one could determine, without recourse to model tests, whether the streamlines along the forebody are such that bilge vortices would be generated and what would be the effect on these streamlines of various modifications of the bow. If separation at the stern is not severe and the bow not too blunt near the free surface, useful results could be obtained for the wavemaking of a ship form on the assumptions of irrotational flow and the linearized free-surface condition, but with the exact boundary condition on the hull. The last condition is significant since it would enable the effects of local modifications of form, especially at the bow, to be studied.

A well-known method of calculating the potential flow about a three-dimensional form is that of Hess and Smith [1]*. In common with the Hess-Smith approach, the method to be described herein determines a distribution of sources on the surface of a given body by solving the basic integral equation of potential theory for such a Neumann problem [2]. The methods differ in the treatment of the singularity of the kernel of the integral equation, the selection of an iteration formula for solving the integral equation, the quadrature formula used to reduce the integral equation to a set of linear equations, and the procedure employed to calculate the velocity distribution along the hull once the source distribution has been determined.

Although the procedure to be described has been available as a computer program for several years, and early results with a body of revolution (a spheroid) and a three-dimensional form (an ellipsoid) showed very good agreement with the exact solutions, publication has been delayed because anomalous results were obtained when the method was applied to a

*Numbers in brackets indicate references.

mathematical form with parabolic lines and sections, having sharp edges at the bow, stern and keel. Presently an attempt is being made to apply the method to a Series-60 model [3]; but since this form is serving as a vehicle for development of a procedure for fitting a mathematical equation to an arbitrary ship form, the hull coordinates and direction cosines of the normals to the hull, required as input in the potential flow program, are not yet available. Consequently it has been decided to present the method and the computer program without further delay, since others may be more efficient in obtaining the required input for their ship forms.

Statement of Problem

Our problem is to develop means of computing the flow about a ship form, including free-surface effects. We shall assume the fluid to be incompressible and inviscid, and the flow irrotational. We shall suppose that a ship form has been prescribed, and that its draft and trim are known. We shall assume that the surface disturbance is sufficiently small that the boundary condition on the free surface may be linearized. The boundary condition on the hull will be satisfied exactly.

An obvious criticism of these assumptions is that, in neglecting the effects of viscosity and sinkage and trim, and employing the approximate linearized form of the free-surface boundary condition, in comparison with the elegant, classical, thin-ship theory, only one of several equally important corrections will have been made. Our view is that the solution of the present problem can be made the basis for incorporating additional corrections by iterative techniques. For example, the resulting pressure distribution can be used to calculate the equilibrium trim and sinkage of the ship, which can then be applied to obtain a second approximation for the flow about it.

Let $f(x, y, z) = 0$ be the equation of the hull surface S , with x in the direction of the stream U , and z positive upwards, with the origin in the undisturbed level of the free surface. Denote the velocity potential by

$$\Phi = Ux + \phi \quad (1)$$

where ϕ is the disturbance potential which, as well as Φ , satisfies the

Laplace equation

$$\nabla^2 \phi = 0 \quad (2)$$

The boundary condition on S is then

$$\frac{\partial \phi}{\partial N} = -U \frac{\partial x}{\partial N} \quad (3)$$

where N denotes distance in the direction of the outward normal to S. The free-surface boundary condition will be taken in the linearized form

$$\frac{\partial^2 \phi}{\partial x^2} + k_0 \frac{\partial \phi}{\partial z} = 0; \quad k_0 = \frac{g}{U^2}, \quad z = 0 \quad (4)$$

For a source of unit strength at the point P(ξ, η, ζ) in the same uniform stream, the velocity potential ϕ_S which satisfies (2) and boundary condition (4) may be written in the form

$$\phi_S = Ux - \frac{1}{R} - \frac{1}{R'} + H(P, Q) \quad (5)$$

where Q is a point Q(x, y, z) below the free surface, R is the distance from P to Q, R' is the distance from P'($\xi, \eta, -\zeta$), the mirror image of P in the free surface, to Q,

$$R = [(x - \xi)^2 + (y - \eta)^2 + (z - \zeta)^2]^{\frac{1}{2}}, \quad R' = [(x - \xi)^2 + (y - \eta)^2 + (z + \zeta)^2]^{\frac{1}{2}} \quad (6)$$

and H(P, Q) is regular harmonic in the lower half space $z < 0$ and given by

$$H(P, Q) = \frac{1}{\pi} \int_0^\infty \int_0^{2\pi} \frac{e^{k(z + \zeta) + ik[(x - \xi) \cos \theta + (y - \eta) \sin \theta]}}{k - k_0 \sec^2 \theta} k dk d\theta$$

$$- 4k_0 \int_0^{\pi/2} e^{k_0(z + \zeta) \sec^2 \theta} \sin[k_0(x - \xi) \sec \theta] \cos[k_0(y - \eta) \tan \theta \sec \theta] \sec^2 \theta \cdot d\theta \quad (7)$$

where \int denotes the "Cauchy principal part". The velocity potential of (5) and (6) also satisfies the "radiation condition", that waves are propagated downstream from the source [4].

Now let M(P) denote the strength of a source distribution on S. The velocity potential of this source distribution which satisfies the free-surface condition (4) and the radiation condition, by (5), is given by

$$\phi(Q) = Ux - \int_{S+S'} \frac{M(P)}{R} dS_P + \int_S M(P) H(P, Q) dS_P \quad (7)$$

where S' denotes the mirror image of S in the plane $z = 0$. Taking into account the discontinuity in the normal derivative of the potential at a surface distribution, the boundary condition on the hull surface S then yields

$$M(Q) = -\frac{U}{2\pi} \frac{\partial x}{\partial N_Q} + \frac{1}{2\pi} \int_{S+S'} M(P) \frac{\partial}{\partial N_Q} \left(\frac{1}{R} \right) dS_P - \frac{1}{2\pi} \int_S M(P) \frac{\partial H(P, Q)}{\partial N_Q} dS_P \quad (8)$$

a Fredholm integral equation of the second kind. The development of a procedure for solving this integral equation numerically for a given ship form is our principal objective.

An essential difficulty in the numerical solution of (8) is that both of the integrands are singular. A means of removing these singularities will be described and justified in the following two sections.

Treatment of Double-Model Integral

At points of S where the normal is continuous, we have by Gauss's flux theorem

$$- \int_{S+S'} \frac{\partial}{\partial N_P} \left(\frac{1}{R} \right) dS_P = 2\pi \quad (9)$$

This enables us to write

$$\frac{1}{2\pi} \int_{S+S'} M(P) \frac{\partial}{\partial N_Q} \left(\frac{1}{R} \right) dS_P = \frac{1}{2\pi} \int_{S+S'} \left[M(P) \frac{\partial}{\partial N_Q} \left(\frac{1}{R} \right) - M(Q) \frac{\partial}{\partial N_P} \left(\frac{1}{R} \right) \right] dS_P - M(Q) \quad (10)$$

We shall now show that the singularity of the left member of (10) when P coincides with Q is not present in the right member.

The direction cosines at the point Q of S are given by

$$l_Q = \left(\frac{\partial n}{\partial N} \right)_Q = \frac{f_x}{D_Q}, \quad m_Q = \frac{f_y}{D_Q}, \quad n_Q = \frac{f_z}{D_Q} \quad (11)$$

where f_x, f_y, f_z denote partial derivatives of $f(x, y, z)$ with respect to x, y, z , and

$$D_Q = [f_x^2 + f_y^2 + f_z^2]^{\frac{1}{2}}$$

Then we have

$$\left. \begin{aligned} \frac{\partial}{\partial N_Q} \left(\frac{1}{R} \right) &= -\frac{1}{R^3} [(x - \xi)l_Q + (y - \eta)m_Q + (z - \zeta)n_Q] \\ &= -\frac{1}{R^3 D_Q} [(x - \xi)f_x + (y - \eta)f_y + (z - \zeta)f_z] \end{aligned} \right\} \quad (12)$$

and

$$\begin{aligned} M(P) \frac{\partial}{\partial N_Q} \left(\frac{1}{R} \right) - M(Q) \frac{\partial}{\partial N_P} \left(\frac{1}{R} \right) &= -\frac{1}{R^3} \left\{ \frac{M(P)}{D_Q} [(x - \xi)f_x + (y - \eta)f_y + (z - \zeta)f_z] \right. \\ &\quad \left. + \frac{M(Q)}{D_P} [(x - \xi)f_\xi + (y - \eta)f_\eta + (z - \zeta)f_\zeta] \right\} \end{aligned} \quad (13)$$

When P is near Q, we can write the Taylor expansion

$$\begin{aligned} f(\xi, \eta, \zeta) &= f(x, y, z) - (x - \xi)f_x - (y - \eta)f_y - (z - \zeta)f_z + \frac{1}{2}[(x - \xi)^2 f_{xx} \\ &\quad + (y - \eta)^2 f_{yy} + (z - \zeta)^2 f_{zz} + 2(y - \eta)(z - \zeta)f_{yz} + \dots] + \dots \end{aligned}$$

Since

$$f(x, y, z) = f(\xi, \eta, \zeta) = 0$$

we obtain

$$(x - \xi)f_x + (y - \eta)f_y + (z - \zeta)f_z = \frac{1}{2}[(x - \xi)^2 f_{xx} + \dots] + \text{3rd order terms} \quad (14)$$

Similarly we have

$$(\xi - x)f_\xi + (\eta - y)f_\eta + (\zeta - z)f_\zeta = \frac{1}{2}[(x - \xi) f_{\xi\xi} + \dots] + \text{3rd order terms}$$

or

$$(x - \xi)f_\xi + (y - \eta)f_\eta + (z - \zeta)f_\zeta = -\frac{1}{2}[(x - \xi)^2 f_{xx} + \dots] + \text{3rd order terms} \quad (15)$$

Also we have

$$M(P) = M(Q) + (\xi - x)M_x + (\eta - y)M_y + (\zeta - z)M_z + \dots \quad (16)$$

and

$$\frac{1}{D_P} = \frac{1}{D_Q} + (\xi - x) \frac{\partial}{\partial x} \left(\frac{1}{D_Q} \right) + \dots \quad (17)$$

Substituting the results in (14), (15), (16) and (17) into (13), we observe that terms of the first and second order in $(x - \xi)$, $(y - \eta)$, $(z - \zeta)$ cancel, leaving antisymmetric terms of the third order in the numerator. Since the denominator R^3 is also of the third order, the ratio is indeterminate as R approaches zero. The integral of this ratio over a small area symmetric about Q , however, is zero. For this reason we propose to set the integrand of the right member of (10) equal to zero when P coincides with Q .

Treatment of Wave Integral

As in the prior case, let us consider the integral of the transposed kernel, $\partial H(Q, P)/\partial N_P$. For points Q below the free surface ($z < 0$), $H(Q, P)$ is a regular harmonic function of $P(\xi, \eta, \zeta)$ for $\zeta \leq 0$. Hence, by Gauss's flux theorem, the flux of $H(Q, P)$ through the closed surface, consisting of S and the surface S_0 of the plane $z = 0$ capping S , is zero; i.e.

$$\int_S \frac{\partial H(Q, P)}{\partial N_P} dS_P = - \int_{S_0} \frac{\partial H(Q, P)}{\partial \zeta} \Big|_{\zeta=0} dS_P \quad (18)$$

Then we can write the wave integral in (8) in the form

$$\int_S M(P) \frac{\partial H(P, Q)}{\partial N_Q} dS_P = \int_S \left[M(P) \frac{\partial H(P, Q)}{\partial N_Q} - M(Q) \frac{\partial H(Q, P)}{\partial N_P} \right] dS_P - M(Q) \int_{S_0} \frac{\partial H(Q, P)}{\partial \zeta} \Big|_{\zeta=0} dS_P \quad (19)$$

From (6) we have

$$\begin{aligned} \frac{\partial H(P, Q)}{\partial N_Q} &= \frac{1}{\pi} \int_0^\infty \int_0^{2\pi} \frac{e^{k(z + \zeta) + ik[(x - \xi) \cos \theta + (y - \eta) \sin \theta]}}{k - k_0 \sec^2 \theta} k^2 \cdot \\ &\quad [n_Q + i(\ell_Q \cos \theta + m_Q \sin \theta)] d\theta dk \\ &+ 4k_0^2 \int_0^{\pi/2} \sec^3 \theta \cdot e^{k_0(z + \zeta) \sec^2 \theta} \left\{ \ell_Q \cos [k_0(x - \xi) \sec \theta] \cdot \right. \\ &\quad \left. \cos [k_0(y - \eta) \tan \theta \sec \theta] \right\} \end{aligned}$$

$$\left. \begin{aligned} & - m_Q \tan \theta \sin [k_0(x - \xi) \sec \theta] \sin [k_0(y - \eta) \tan \theta \sec \theta] \\ & + n_Q \sec \theta \sin [k_0(x - \xi) \sec \theta] \cos [k_0(y - \eta) \tan \theta \sec \theta] \end{aligned} \right\} d\theta \quad (20)$$

By comparison of the double integral in (20) with that in the relation

$$\begin{aligned} 2 \frac{\partial}{\partial N_Q} [(z + \zeta)^2 + (x - \xi)^2 + (y - \eta)^2]^{-\frac{1}{2}} = \\ \frac{1}{\pi} \int_0^{\infty} \int_0^{2\pi} e^{k(z + \zeta) + ik[(x - \xi) \cos \theta + (y - \eta) \sin \theta]} \cdot \\ k[n_Q + i(\ell_Q \cos \theta + n_Q \sin \theta)] dk d\theta \end{aligned}$$

the integrands of which are asymptotically equal for very large values of k , we see that the former integral is singular only at the free surface $z = \zeta = 0$ when P coincides with Q . As in the previous section, this singularity is removed from the first integral on the right of (19) by subtracting the transpose from the original integrand, although the problem of treating the last integral of (19), which is also singular, remains.

Next let us consider the second integral of (20). With the substitution $\lambda = \tan \theta$, the integral becomes

$$\begin{aligned} \int_0^{\infty} \sqrt{1 + \lambda^2} e^{k_0(z + \zeta)(1 + \lambda^2)} \left\{ \ell_Q \cos [k_0(x - \xi)\sqrt{1 + \lambda^2}] \cos [k_0(y - \eta)\lambda\sqrt{1 + \lambda^2}] \right. \\ \left. - m_Q \lambda \sin [k_0(x - \xi)\sqrt{1 + \lambda^2}] \sin [k_0(y - \eta)\lambda\sqrt{1 + \lambda^2}] \right. \\ \left. + n_Q \sqrt{1 + \lambda^2} \sin [k_0(x - \xi)\sqrt{1 + \lambda^2}] \cos [k_0(y - \eta)\lambda\sqrt{1 + \lambda^2}] \right\} d\lambda \quad (21) \end{aligned}$$

Convergence problems arise only when $z = \zeta = 0$. For this case let us expand the integrand of (21) in powers of $1/\lambda$. We have, with

$$\begin{aligned} \alpha & = k_0(x - \xi), \quad \beta = k_0(y - \eta) \\ \cos(\alpha\sqrt{1 + \lambda^2}) & = \cos \left[\alpha \left(\lambda + \frac{1}{2\lambda} - \frac{1}{8\lambda^3} + \dots \right) \right] \\ & = \cos \alpha \lambda \cos \left[\alpha \left(\frac{1}{2\lambda} - \frac{1}{8\lambda^3} + \dots \right) \right] \\ & \quad - \sin \alpha \lambda \sin \left[\alpha \left(\frac{1}{2\lambda} - \dots \right) \right] \end{aligned}$$

Hence, for very large values of λ , we have

$$\cos (\alpha \sqrt{1+\lambda^2}) \approx \left(1 - \frac{\alpha^2}{4\lambda^2}\right) \cos \alpha \lambda - \frac{\alpha}{2\lambda} \sin \alpha \lambda \quad (22a)$$

Similarly

$$\sin (\alpha \sqrt{1+\lambda^2}) \approx \left(1 - \frac{\alpha^2}{4\lambda^2}\right) \sin \alpha \lambda + \frac{\alpha}{2\lambda} \cos \alpha \lambda \quad (22b)$$

$$\cos (\beta \lambda \sqrt{1+\lambda^2}) \approx \cos [\beta(\lambda^2 + \frac{1}{2})] + \frac{\beta}{8\lambda^2} \sin [\beta(\lambda^2 + \frac{1}{2})] \quad (22c)$$

$$\sin (\beta \lambda \sqrt{1+\lambda^2}) \approx \sin [\beta(\lambda^2 + \frac{1}{2})] + \frac{\beta}{8\lambda^2} \cos [\beta(\lambda^2 + \frac{1}{2})] \quad (22d)$$

The the terms of the integrand of (21) become asymptotically

$$\begin{aligned} \sqrt{1+\lambda^2} \cos [\alpha \sqrt{1+\lambda^2}] \cos [\beta \lambda \sqrt{1+\lambda^2}] &\approx \lambda \cos \alpha \lambda \cos [\beta(\lambda^2 + \frac{1}{2})] \\ &\quad - \frac{\alpha}{2} \sin \alpha \lambda \cos [\beta(\lambda^2 + \frac{1}{2})] \end{aligned} \quad (23a)$$

$$\begin{aligned} \lambda \sqrt{1+\lambda^2} \sin [\alpha \sqrt{1+\lambda^2}] \sin [\beta \lambda \sqrt{1+\lambda^2}] &\approx \lambda^2 - \frac{\alpha^2}{4} + \frac{1}{2} \sin \alpha \lambda \sin [\beta(\lambda^2 + \frac{1}{2})] \\ &\quad + \frac{\alpha \lambda}{2} \cos \alpha \lambda \sin [\beta(\lambda^2 + \frac{1}{2})] - \frac{\beta}{8} \sin \alpha \lambda \cos [\beta(\lambda^2 + \frac{1}{2})] \end{aligned} \quad (23b)$$

$$\begin{aligned} (1+\lambda^2) \sin [\alpha \sqrt{1+\lambda^2}] \cos [\beta \lambda \sqrt{1+\lambda^2}] &\approx (\lambda^2 - \frac{\alpha^2}{4} + 1) \cos \alpha \lambda \sin [\beta(\lambda^2 + \frac{1}{2})] \\ &\quad + \frac{\alpha \lambda}{2} \cos \alpha \lambda \cos [\beta(\lambda^2 + \frac{1}{2})] + \frac{\beta}{8} \sin \alpha \lambda \sin [\beta(\lambda^2 + \frac{1}{2})] \end{aligned} \quad (23c)$$

Considering $\sin [\beta(\lambda^2 + \frac{1}{2})]$ and $\cos [\beta(\lambda^2 + \frac{1}{2})]$ in the forms

$$\begin{aligned} \sin [\beta(\lambda^2 + \frac{1}{2})] &= \sin (\beta \lambda^2) \cos \frac{\beta}{2} + \cos (\beta \lambda^2) \sin \frac{\beta}{2} \\ \cos [\beta(\lambda^2 + \frac{1}{2})] &= \cos (\beta \lambda^2) \cos \frac{\beta}{2} - \sin (\beta \lambda^2) \sin \frac{\beta}{2} \end{aligned}$$

it is seen that, with $z = \zeta = 0$, the asymptotic form of the integrand of (21) is a linear combination of the terms

$$\sin \alpha \lambda \frac{\sin}{\cos} (\beta \lambda^2), \quad \lambda \cos \alpha \lambda \frac{\sin}{\cos} (\beta \lambda^2), \quad \lambda^2 \sin \alpha \lambda \frac{\sin}{\cos} (\beta \lambda^2) \quad (24)$$

But from the table of definite integrals by Gröbner and Hofreiter [5], we have

$$\int_0^{\infty} \sin \alpha \lambda \frac{\sin}{\cos} (\beta \lambda^2) d\lambda = 0, \quad \beta \neq 0 \quad (25)$$

Since the derivatives of the first member of (24) with respect to α and β yield the second and third members, one is tempted to conclude from these

derivations of (25) that the infinite integrals of the second and third members are also zero. However, since the integrals resulting from these differentiations are not uniformly convergent, this conclusion may not be valid, as is also shown by the integration by parts

$$\int_0^L \sin \alpha \lambda \sin (\beta \lambda^2) d\lambda = -\frac{1}{\alpha} \cos \alpha L \sin \beta L^2 \Big|_0^L + \frac{2\beta}{\alpha} \int_0^L \lambda \cos \alpha \lambda \cos (\beta \lambda^2) d\lambda \quad (26)$$

Although the limit of the left member as $L \rightarrow \infty$ is zero, the oscillation of the first term on the right between $\pm 1/\alpha$ indicates that the last integral is indeterminate. The mean value of the last integral would, however, be zero in the limit.

If z and ζ were not zero, the asymptotic forms occurring in (24) would have been multiplied by $e^{k_0(z + \zeta)\lambda^2}$. With this factor the integral of the first member of (24) and its derivatives would be uniformly convergent, and consequently the derivative of the integral would be equal to the integral of the derivative. For example we would have

$$\frac{\partial}{\partial \alpha} \int_0^\infty e^{k_0(z + \zeta)\lambda^2} \sin \alpha \lambda \sin (\beta \lambda^2) d\lambda = \int_0^\infty e^{k_0(z + \zeta)\lambda^2} \lambda \cos \alpha \lambda \sin (\beta \lambda^2) d\lambda$$

with a determinate value for the right member, no matter how close $z + \zeta$ is to zero. Again integrating by parts, we have

$$\int_0^L e^{k_0(z + \zeta)\lambda^2} \sin \alpha \lambda \sin (\beta \lambda^2) d\lambda = -\frac{1}{\alpha} e^{k_0(z + \zeta)L^2} \cos \alpha L \sin \beta L^2 \Big|_0^L + \frac{2}{\alpha} \int_0^L [k_0(z + \zeta) \sin (\beta \lambda^2) + \beta \cos (\beta \lambda^2)] \lambda e^{k_0(z + \zeta)\lambda^2} \cos \alpha \lambda d\lambda \quad (27)$$

Now the oscillation of the first member on the right with increasing values of L is damped by the exponential factor and yields in the limit

$$\int_0^\infty e^{k_0(z + \zeta)\lambda^2} \sin \alpha \lambda \sin (\beta \lambda^2) d\lambda = \frac{2}{\alpha} \int_0^\infty [k_0(z + \zeta) \sin (\beta \lambda^2) + \beta \cos (\beta \lambda^2)] \lambda e^{k_0(z + \zeta)\lambda^2} \cos \alpha \lambda d\lambda \quad (28)$$

which, when $z + \zeta$ is very small, becomes

$$\int_0^{\infty} e^{k_0(z + \zeta)\lambda^2} \sin \alpha \lambda \sin (\beta \lambda^2) d\lambda \approx \frac{2\beta}{\alpha} \int_0^{\infty} e^{k_0(z + \zeta)\lambda^2} \lambda \cos \alpha \lambda \cos (\beta \lambda^2) d\lambda \quad (29)$$

Hence, since by (25) the limit of the integral on the left of (29), as $z + \zeta \rightarrow 0$, is zero, we see that

$$\begin{aligned} \lim_{z + \zeta \rightarrow 0} \int_0^{\infty} e^{k_0(z + \zeta)\lambda^2} \lambda \cos \alpha \lambda \frac{\sin}{\cos} (\beta \lambda^2) d\lambda \\ = \text{M.V.O.} \int_0^L \lambda \cos \alpha \lambda \frac{\sin}{\cos} (\beta \lambda^2) d\lambda = 0 \end{aligned} \quad (30)$$

where M.V.O. denotes the "mean value of the oscillation" for large values of L . Clearly the foregoing result applies to either $\sin (\beta \lambda^2)$ or $\cos (\beta \lambda^2)$, as is indicated in (30). Similarly, by integrating the last integrals in (26) and (27) again by parts, we can show that

$$\begin{aligned} \lim_{z + \zeta \rightarrow 0} \int_0^{\infty} e^{k_0(z + \zeta)\lambda^2} \lambda^2 \sin \alpha \lambda \frac{\sin}{\cos} (\beta \lambda^2) d\lambda \\ = \text{M.V.O.} \int_0^L \lambda^2 \sin \alpha \lambda \frac{\sin}{\cos} (\beta \lambda^2) d\lambda = 0 \end{aligned} \quad (31)$$

In the above analysis it was assumed in (25) that $\beta = k_0(y - \eta) \neq 0$. If $\beta = 0$, $\alpha \neq 0$, the terms of (24) become

$$\sin \alpha \lambda, \quad \lambda \cos \alpha \lambda, \quad \lambda^2 \sin \alpha \lambda$$

and we can show by integration by parts that

$$\begin{aligned} \lim_{z + \zeta \rightarrow 0} \int_0^{\infty} e^{k_0(z + \zeta)\lambda^2} \sin \alpha \lambda d\lambda &= \lim_{z + \zeta \rightarrow 0} \int_0^{\infty} \lambda e^{k_0(z + \zeta)\lambda^2} \cos \alpha \lambda d\lambda \\ &= \lim_{z + \zeta \rightarrow 0} \int_0^{\infty} \lambda^2 e^{k_0(z + \zeta)\lambda^2} \sin \alpha \lambda d\lambda = 0 \end{aligned} \quad (32)$$

If $\alpha = 0$ also, then we have

$$\int_0^{\infty} e^{k_0(z + \zeta)\lambda^2} \lambda d\lambda = \frac{1}{2k_0|z + \zeta|}$$

which indicates that the second of the limits in (32) does not exist.

Our conclusion is that the integral in (21) is determinate except when P and Q coincide and are at the free surface. When P and Q are at the free surface, but not coincident, the integral must be determined as the limiting value as $z + \zeta \rightarrow 0$ through negative values. Finally, the singularity when P and Q coincide is removed from the first integral on the right of (19) by subtracting the transpose from the original integrand, although, as for the double integral of (20), the last integral of (19) remains to be treated.

Let us now consider the last integral in (19),

$$\int_{S_0} \frac{\partial H(Q, P)}{\partial \zeta} \Big|_{\zeta=0} dS_P = \frac{1}{\pi} \int_{S_0} \int_0^\infty \int_0^{2\pi} \frac{e^{kz - ik[(x - \xi) \cos \theta + (y - \eta) \sin \theta]}}{k - k_0 \sec^2 \theta} \cdot k^2 d\theta dk dS_P$$

$$-4k_0^2 \int_{S_0} \int_0^{\pi/2} \sec^4 \theta \cdot e^{k_0 z \sec^2 \theta} \sin [k_0(x - \xi) \sec \theta] \cos [k_0(y - \eta) \tan \theta \sec \theta] d\theta dS_P \quad (33)$$

Let $\eta = \pm \eta(x)$ be the equation of the hull waterplane at $\zeta = 0$. Take the origin at the midship section and let l denote the half length of the ship. Then, interchanging the order of integration in (33) with the integration over S_0 taken first, we are led to consider the integral,

$$F(k, \theta) = \int_{-l}^l \int_{-\eta(x)}^{\eta(x)} e^{ik(\xi \cos \theta + \eta \sin \theta)} d\eta d\xi \quad (34)$$

This becomes

$$F(k, \theta) = \frac{2 \csc \theta}{k} \int_{-l}^l e^{ik \xi \cos \theta} \sin [k \eta(\xi) \sin \theta] d\xi \quad (35)$$

or, integrating by parts and noting that $\eta(l) = \eta(-l) = 0$, we obtain

$$F(k, \theta) = \frac{2i \sec \theta}{k} \int_{-l}^l e^{ik \xi \cos \theta} \eta'(\xi) \cos [k \eta(\xi) \sin \theta] d\xi \quad (36)$$

where $\eta'(\xi)$ denotes the derivative of η with respect to ξ .

Along the parallel middle body of a ship form, $\eta'(\xi) = 0$, and near the bow and stern, $\eta(\xi)$ is very small. This suggests the approximation

$$F(k, \theta) = \frac{2i \sec \theta}{k} \int_{-l}^l e^{ik \xi \cos \theta} \eta'(\xi) [1 - \frac{1}{2}k^2 \eta(\xi)^2 \sin^2 \theta] d\xi \quad (37)$$

Additional terms may be taken in the expansion of $\cos [k \eta(\xi) \sin \theta]$ if required for greater accuracy. If the parallel middle body extends over the range $a \leq \xi \leq b$, (37) may be written

$$F(k, \theta) = \frac{2i \sec \theta}{k} \left\{ \int_{-l}^a + \int_b^l \right\} e^{ik \xi \cos \theta} \cdot$$

$$[\eta'(\xi) - \frac{1}{2}k^2 \eta'(\xi) \eta(\xi)^2 \sin^2 \theta] d\xi$$

or, introducing $\mu = \frac{\xi - a}{l + a}$ in the first integral, and $\mu = \frac{\xi - b}{l - b}$ in the second,

$$F(k, \theta) = \frac{2i \sec \theta}{k} \left\{ e^{ika \cos \theta} \int_{-1}^0 e^{ik(l+a)\mu \cos \theta} \cdot \right.$$

$$[\eta'(\xi) - \frac{1}{2}k^2 \eta'(\xi) \eta(\xi)^2 \sin^2 \theta] d\mu$$

$$+ e^{ikb \cos \theta} \int_0^1 e^{ik(l-b)\mu \cos \theta} [\eta'(\xi) - \frac{1}{2}k^2 \eta'(\xi) \eta(\xi)^2 \sin^2 \theta] d\mu \quad (38)$$

One can now fit either Fourier series or polynomials in μ to the functions $\eta'(\xi)$ and $\eta'(\xi) \eta(\xi)^2$, the choice depending upon the particular form. If $\eta'(\xi)$ becomes infinite at the bow and stern, as it will if the radii of curvature are not zero at the extremities, a suitable fit which can satisfy this condition can be obtained from a polynomial for $\eta(\xi)^2$ of the form

$$\eta(\xi)^2 = (1 - \mu^2) p(\mu) \quad (39)$$

where $p(\mu)$ is a polynomial such that $p(\pm 1) \neq 0$. For then

$$\eta'(\xi) = [\frac{1}{2} (1 - \mu^2) p'(\mu) - \mu p(\mu)] \frac{d\mu/d\xi}{\eta(\xi)}$$

is seen to become infinite at $\mu = \pm 1$. In this way $F(k, \theta)$ can be expressed as a series of functions of k and θ , each of which is regular even as z approaches zero.

Although not evident from the form of the last integral in (33), by returning to the complex exponential form from which its integrand was derived one finds that the function $F(k, \theta)$ of (34), with $k = k_0 \sec^2 \theta$, applies to this integral as well. This will not be developed here in detail, nor will the analysis of the wave kernel be carried any further, since the application to a particular case, on which, it has been seen, the nature of the subsequent analysis would depend, has not yet been performed.

Convergence of Iteration Formulas

Hereafter we shall consider only the case where the boundary condition on the plane $z = 0$ is

$$\frac{\partial \Phi}{\partial z} = 0, \quad z = 0 \tag{40}$$

i.e., the case of "zero" Froude number. The integral equation (8) then reduces to

$$M(Q) = F(Q) + \int_{S+S'} M(P) K(P, Q) dS_P \tag{41}$$

where

$$K(P, Q) = \frac{1}{2\pi} \frac{\partial}{\partial N_Q} \left(\frac{1}{R} \right), \quad F(Q) = -\frac{U}{2\pi} \ell_Q \tag{42}$$

Equation (9) shows that the homogeneous integral equation

$$f(Q) = \lambda \int_{S+S'} K(Q, P) f(P) dS_P$$

has the eigenfunction $f(P) = 1$ when $\lambda = -1$. Thus $\lambda = -1$ is an eigenvalue of the kernel $K(Q, P)$, and hence also of its transpose $K(P, Q)$.

Consider the inhomogeneous integral equation

$$M(Q) = F(Q) + \lambda \int_{S+S'} M(P) K(P, Q) dS_P \tag{43}$$

which reduces to (41) when $\lambda = 1$. The theory of this integral equation states that $M(Q)$, considered as a function of the complex variable λ , is regular in the unit circle about $\lambda = 0$, and has a simple pole at $\lambda = -1$. Writing

$$M(Q) = F(Q) + \lambda F_1(Q) + \lambda^2 F_2(Q) + \dots, \quad |\lambda| < 1 \quad (44)$$

and substituting (44) into (43), yields the relation

$$F_{n+1}(Q) = \int_{S+S'} F_n(P) K(P, Q) dS_P \quad (45)$$

Put

$$M_n(Q) = F(Q) + \lambda F_1(Q) + \dots + \lambda^n F_n(Q) \quad (46)$$

Then, by (45), we obtain the iteration formula

$$M_{n+1}(Q) = F(Q) + \lambda \int_{S+S'} M_n(P) K(P, Q) dS_P \quad (47)$$

According to (44), however, the sequence of functions $M_{n+1}(Q)$ defined by (47) may not converge when $\lambda = 1$.

We can eliminate the pole at $\lambda = -1$ by considering

$$(1 + \lambda) M(Q) = F(Q) + \lambda(F + F_1) + \lambda^2(F_1 + F_2) + \dots, \quad |\lambda| < |\lambda_2| \quad (48)$$

where λ_2 denotes the next eigenvalue of $K(P, Q)$, arranged in the order of increasing absolute magnitude. Defining

$$M'_n = \frac{1}{1 + \lambda} [F(Q) + \lambda(F + F_1) + \dots + \lambda^n(F_{n-1} + F_n)] \quad (49)$$

then, by (45),

$$M'_n(Q) = F(Q) + \lambda \int_{S+S'} M'_n(P) K(P, Q) dS_P \quad (50)$$

Comparison with (47) shows that the sequences M_n and M'_n are obtained from the identical iteration formula, but these sequences differ because of the change in the initial approximations,

$$M_0(Q) = F(Q), \quad M'_0(Q) = \frac{F(Q)}{1 + \lambda} \quad (51)$$

Thus, when $\lambda = 1$, $M'_0(Q) = \frac{1}{2} F(Q)$. Alternatively, if we observe from (44) and (49) that

$$M'_n = \frac{1}{1 + \lambda} (M_n + \lambda M_{n-1}) \quad (52)$$

we obtain when $\lambda = 1$

$$M'_n = \frac{1}{2} (M_n + M_{n-1})$$

i.e., the arithmetic means of successive pairs of members of the sequence $M_n(Q)$ form a convergent sequence.

Let us consider the modification of the iteration formula (50),

$$2M_{n+1}''(Q) = M_n''(Q) + F(Q) + \lambda \int_{S+S'} M_n''(Q) K(P, Q) dS_P \quad (53)$$

with $M_0'' = M_0'$ given by (51). We have, by (50),

$$M_1'' = \frac{1}{2} [M_0' + F + \lambda \int M_0' K dS] = \frac{1}{2} (M_0' + M_1')$$

$$M_2'' = \frac{1}{2} [\frac{1}{2} (M_0' + M_1') + F + \lambda \int \frac{1}{2} (M_0' + M_1') K dS] = \frac{1}{4} (M_0' + 2M_1' + M_2')$$

We can now show by mathematical induction that

$$M_n'' = \frac{1}{2^n} [M_0' + \binom{n}{1} M_1' + \binom{n}{2} M_2' + \dots + \binom{n}{n} M_n'] \quad (54)$$

For if (54) is valid, then by (53)

$$2M_{n+1}'' = \frac{1}{2^n} \left\{ [M_0' + \binom{n}{1} M_1' + \dots + M_n'] + 2^n F + \lambda \int [M_0' + \binom{n}{1} M_1' + \dots + M_n'] K dS \right\}$$

But

$$2^n F \equiv (1 + 1)^n F = [1 + \binom{n}{1} + \binom{n}{2} + \dots + \binom{n}{n}] F$$

Then, by (50),

$$2^n F + \lambda \int [M_0' + \binom{n}{1} M_1' + \dots + M_n'] K dS = M_1' + \binom{n}{1} M_2' + \binom{n}{2} M_3' + \dots + M_{n+1}'$$

Thus we have

$$2M_{n+1}'' = \frac{1}{2^n} \left\{ M_0' + [\binom{n}{0} + \binom{n}{1}] M_1' + [\binom{n}{1} + \binom{n}{2}] M_2' + \dots + [\binom{n}{n-1} + \binom{n}{n}] M_n' + M_{n+1}' \right\}$$

or since

$$\binom{n}{n} + \binom{n}{n+1} = \binom{n+1}{n+1}$$

then

$$M_{n+1}'' = \frac{1}{2^{n+1}} [M_0' + \binom{n+1}{1} M_1' + \binom{n+1}{2} M_2' + \dots + M_{n+1}']$$

as we wished to show to complete the proof by induction.

In order to investigate the convergence of the sequence $\{M''_n\}$, let us take N sufficiently large so that, for $r > N$, $|M'_r - M| < \epsilon/2$, where M is the limit of the sequence $\{M'_n\}$. From (54) we have

$$M''_n - M = \frac{1}{2^n} [(M'_0 - M) + \binom{n}{1}(M'_1 - M) + \cdots + \binom{n}{N}(M'_N - M) + \cdots + \binom{n}{n}(M'_n - M)]$$

Then

$$|M''_n - M| < \frac{1}{2^n} [|M'_0 - M| + \binom{n}{1} |M'_1 - M| + \cdots + |M'_N - M| \binom{n}{N}] \\ + \frac{\epsilon}{2^{n+1}} [\binom{n}{N+1} + \binom{n}{N+2} + \cdots + \binom{n}{n}]$$

But

$$\frac{1}{2^n} [\binom{n}{N+1} + \binom{n}{N+2} + \cdots + \binom{n}{n}] < \frac{1}{2^n} [1 + \binom{n}{1} + \binom{n}{2} + \cdots + \binom{n}{n}] = 1$$

Then, if ρ is an upper bound of $|M'_i - M|$, $i = 0, 1, 2, \dots, N$, and $n \ll 2N$, we have

$$|M''_n - M| < \frac{\rho}{2^n} [1 + \binom{n}{1} + \binom{n}{2} + \cdots + \binom{n}{N}] + \frac{\epsilon}{2} < \frac{\rho}{2^n} (N+1) \binom{n}{N} + \frac{\epsilon}{2} \\ < \frac{\rho(N+1)}{N!} \frac{n^N}{2^n} + \frac{\epsilon}{2} < \epsilon$$

by taking n sufficiently large. Hence the sequence $\{M''_n\}$ also converges to M .

The alternative iteration formulas for M' or M'' arise when relation (10) is applied to remove the singularity of the kernel in the integral equation (41). We obtain

$$M(Q) = F(Q) - M(Q) + \int_{S+S'} [M(P) K(P, Q) - M(Q) K(Q, P)] dS_P \quad (55)$$

and the iteration formulas

$$M_{n+1} = F - M_n + \int [M_n(P) K(P, Q) - M_n(Q) K(Q, P)] dS_P \quad (56)$$

or

$$2M_{n+1} = F + \int [M_n(P) K(P, Q) - M_n(Q) K(Q, P)] dS_P \quad (57)$$

By (9), the first of these is seen to be of the form (50), the second of the form (53). Both begin with the same initial approximation which, by (51), is given by

$$M_0(Q) = \frac{1}{2} F(Q) = -\frac{U}{4\pi} \rho_Q \quad (58)$$

Although there is no a-priori basis for preferring one iteration formula over the other, comparison of the numerical results with the known exact solution for the case of an ellipsoid has shown that the sequence given by (57) converged much more rapidly than that obtained from (56).

At points where the normal to S is not continuous, the integral equation (8), and of course the above iteration formulae, are not valid. At such points we can either set $M(Q) = 0$, as can be justified, or round sharp edges with small, nonzero curvature and continue to use the iteration formula (57).

Distribution of Velocity Potential on S

Once the source distribution M has been found, the velocity potential ϕ can be computed from (7). For points Q on S we again encounter a singularity when P coincides with Q. This singularity may be removed as follows:

Let $N(P)$ be a source distribution on $S + S'$ which makes the surface an equipotential of potential ϕ_0 . This distribution satisfies the homogeneous integral equation

$$\int_{S+S'} N(P) K(P, Q) dS_P = -N(Q) \quad (59)$$

with the same kernel as in (42). In fact, $N(P)$ is the eigenfunction of $K(P, Q)$ associated with the eigenvalue $\lambda = -1$. This equation can be solved by means of the iteration formula

$$N_{n+1}(Q) = - \int_{S+S'} N_n(P) K(P, Q) dS_P$$

which, by applying (9), may be written in the singularity-free form

$$N_{n+1}(Q) = N_n(Q) - \int_{S+S'} [N_n(P) K(P, Q) - N_n(Q) K(Q, P)] dS_P \quad (60)$$

Since the matrices occurring in (60) have already been obtained for the numerical evaluation of $M(Q)$ from (57), the corresponding values of $N(Q)$ can be obtained from (60) with little additional computer time. Since the potential is constant in the interior of an equipotential surface, its value may conveniently be computed at the origin as

$$\phi_0 = - \int_{S+S'} \frac{N(P)}{[x^2 + y^2 + z^2]^{\frac{3}{2}}} dS_P \quad (61)$$

We can now apply the solution of this Dirichlet problem to eliminate the singularity from the expression for the velocity potential (7), by writing

$$\begin{aligned} \phi(Q) = Ux - \int_{S+S'} \frac{1}{R} [M(P) - N(P) \frac{M(Q)}{N(Q)}] dS_P + \phi_0 \frac{M(Q)}{N(Q)} \\ + \int_S M(P) H(P, Q) dS_P \end{aligned} \quad (62)$$

Here also we can justify setting the first integrand of (62) equal to zero when P coincides with Q , by the same argument as was used in equation (10).

Application to a Double Ship Model — Zero Froude Number

Since the x - y and x - z planes are planes of symmetry, it is necessary to determine the source distribution over only one-fourth of the hull surface of the double model. Let us consider only points Q for $y, z, \leq 0$. Denote by S_1, S_2, S_3, S_4 the parts of $S + S'$ for which $y, z > 0; y < 0, z > 0; y < 0, z < 0; y > 0, z < 0$, respectively. Put

$$\begin{aligned} R_1 &= [(x - \xi)^2 + (y - \eta)^2 + (z - \zeta)^2]^{\frac{1}{2}} \\ R_2 &= [(x - \xi)^2 + (y + \eta)^2 + (z - \zeta)^2]^{\frac{1}{2}} \\ R_3 &= [(x - \xi)^2 + (y + \eta)^2 + (z + \zeta)^2]^{\frac{1}{2}} \\ R_4 &= [(x - \xi)^2 + (y - \eta)^2 + (z + \zeta)^2]^{\frac{1}{2}} \end{aligned}$$

the distances from $Q(x, y, z)$ to the congruent points $P \equiv P_1(\xi, \eta, \zeta)$, $P_2(\xi, -\eta, \zeta)$, $P_3(\xi, -\eta, -\zeta)$, $P_4(\xi, \eta, -\zeta)$. At congruent points we have

$$M(P) = M(P_2) = M(P_3) = M(P_4)$$

and, denoting the direction cosines at P_i by $l_i, m_i, n_i, i = 1, 2, 3, 4$, we obtain the following relations:

$$l_p \equiv l_1 = l_2 = l_3 = l_4$$

$$m_p \equiv m_1 = -m_2 = -m_3 = m_4$$

$$n_p \equiv n_1 = n_2 = -n_3 = -n_4$$

If the values of the integrand of (57) at congruent points P are collected, the resulting integral would extend only over S_1 and is found to be of the form

$$M_{n+1}(Q) = F(Q) + \frac{1}{4\pi} \int_{S_1} [M_n(P) J(P, Q) - M_n(Q) J(Q, P)] dS_P \quad (63)$$

where

$$J(P, Q) = [(\xi - x)l_Q + (\eta - y)m_Q + (\zeta - z)n_Q] \left(\frac{1}{R_1^3} + \frac{1}{R_2^3} + \frac{1}{R_3^3} + \frac{1}{R_4^3} \right) - 2 \left(\frac{m_Q n}{R_2^3} + \frac{m_Q n + n_Q \zeta}{R_3^3} + \frac{n_Q \zeta}{R_4^3} \right); \quad P \neq Q \quad (64)$$

and, since the integrand of (63) vanishes when P coincides with Q , we may set

$$J(Q, Q) = 0 \quad (65)$$

Similarly the integrations over $S + S'$ in connection with the Dirichlet problem in (60), (61) and (62) can be expressed in terms of integrals over S_1 . Thus (60) becomes

$$N_{n+1}(Q) = N_n(Q) - \int_{S_1} [N_n(P) J(P, Q) - N_n(Q) J(Q, P)] dS_P \quad (66)$$

in which $J(P, Q)$ and $J(Q, Q)$ are given in (64) and (65), (61) becomes

$$\phi_0 = -4 \int_{S_1} \frac{N(P)}{[x^2 + y^2 + z^2]^{\frac{3}{2}}} dS_P \quad (67)$$

and (62), without the wave integral, assumes the form

$$\phi(Q) = Ux - \int_{S_1} L(P, Q) [M(P) - N(P) \frac{M(Q)}{N(Q)}] dS_P + \phi_0 \frac{M(Q)}{N(Q)} \quad (68)$$

where

$$\left. \begin{aligned} L(P, Q) &= \frac{1}{R_1} + \frac{1}{R_2} + \frac{1}{R_3} + \frac{1}{R_4} ; \quad P \neq Q \\ \text{and} \quad L(Q, Q) &= 0 \end{aligned} \right\} \quad (69)$$

Application to an Ellipsoid

With the equation of the ellipsoid in the alternative forms

$$\left. \begin{aligned} \frac{x^2}{a^2} + \frac{y^2}{b^2} + \frac{z^2}{c^2} &= 1 \\ \text{or} \quad y &= \frac{b}{a} \sqrt{a^2 - x^2} \cos \theta, \quad z = \frac{c}{a} \sqrt{a^2 - x^2} \sin \theta \end{aligned} \right\} \quad (70)$$

the direction cosines are

$$l_Q = \frac{A(Q)x}{a^2}, \quad m_Q = \frac{A(Q)y}{b^2}, \quad n_Q = \frac{A(Q)z}{c^2}$$

where

$$A(Q) = \left[\frac{x^2}{a^4} + \frac{y^2}{b^4} + \frac{z^2}{c^4} \right]^{-\frac{1}{2}}$$

Also we have

$$(\xi - x)l_Q + (\eta - y)m_Q + (\zeta - z)n_Q = -A(Q) \left[\frac{(\xi - x)^2}{a^2} + \frac{(\eta - y)^2}{b^2} + \frac{(\zeta - z)^2}{c^2} \right] \quad (71)$$

The right member of (71) is preferable to the first for numerical computations, especially when P is near Q, since all terms on the right are then of second order of smallness, and a loss of numerical accuracy would be expected if the same result were obtained from the sum of the first-order terms on the left. In terms of x and θ as the independent variables, the element of area dS_P becomes

$$dS_P = \left[\frac{b^2 c^2 x^2}{a^4} + \left(1 - \frac{x^2}{a^2}\right) (b^2 \sin^2 \theta + c^2 \cos^2 \theta) \right]^{\frac{1}{2}} d\theta dx$$

with θ varying from 0 to 2π and x from $-a$ to a .

Exact solutions of the foregoing integral equations for the ellipsoid are expressible in terms of the Lamé ellipsoidal coordinates (ρ, μ, ν) , which are related to the rectangular coordinates (x, y, z) by [6]

$$x = \frac{\rho\mu\nu}{hk}, \quad y^2 = \frac{(\rho^2 - k^2)(k^2 - \mu^2)(k^2 - \nu^2)}{k^2(k^2 - h^2)}, \quad z^2 = \frac{(\rho^2 - h^2)(\mu^2 - h^2)(h^2 - \nu^2)}{h(k^2 - h^2)} \quad (72)$$

When $\rho = a$, $k = \sqrt{a^2 - b^2}$, $h = \sqrt{a^2 - c^2}$, (72) is equivalent to the equation of the ellipsoid (70). Here

$$k^2 \leq \rho^2 < \infty, \quad h^2 \leq \mu^2 \leq k^2, \quad 0 \leq \nu^2 \leq h^2$$

Solutions of the present problem can be expressed in terms of the Lamé functions of the first and second kind, E_n^m and F_m^n . For the source distribution we find

$$M(Q) = \frac{3U\mu\nu}{4\pi ahk \dot{F}_1^1(a) \sqrt{(a^2 - \mu^2)(a^2 - \nu^2)}} = -\frac{U}{4\pi} \left[1 - \frac{F_1^1(a)}{\dot{F}_1^1(a)} \right] \ell_Q \quad (73)$$

and for the velocity potential,

$$\phi = \left[1 - \frac{F_1^1(a)}{\dot{F}_1^1(a)} \right] Ux \quad (74)$$

Here \dot{F}_1^1 denotes the derivative of F_1^1 with respect to its argument, and

$$F_1^1(a) = 3a \int_a^\infty \frac{d\rho}{\rho^2 \sqrt{(\rho^2 - h^2)(\rho^2 - k^2)}} = \frac{3a}{kh^2} [F(\phi, \lambda) - E(\phi, \lambda)] \quad (75)$$

$$\phi = \arccos \frac{b}{a}, \quad \lambda = \frac{h}{k}$$

where $F(\phi, \lambda)$, $E(\phi, \lambda)$ are the Legendre incomplete elliptic integrals of modulus λ and amplitude ϕ of first and second kinds.

Exact solutions of the equipotential problem are also of interest. If the total strength of the distribution over the ellipsoid is G , the source strength $N(Q)$ is given by

$$N(Q) = \frac{GA(Q)}{4\pi abc} \quad (76)$$

On and within the ellipsoid the potential has the constant value

$$\phi_0 = \frac{G}{2} \int_0^\infty \frac{ds}{[(a^2 + s)(b^2 + s)(c^2 + s)]^{3/2}} = \frac{G}{k} F(\phi, \lambda) \quad (77)$$

Calculations have been performed for the case $a = 1$, $b = 0.25$, $c = 0.50$, for which

$$1 - \frac{F_1^1(a)}{F_1^1(a)} = 1.12659$$

Then

$$M(Q) = -1.12659 \frac{U}{4\pi} xA(Q)$$

$$\phi = 1.12659 Ux$$

$$N(Q) = \frac{2}{\pi} GA(Q)$$

$$\phi_0 = 1.76984 G$$

FORTTRAN Program

In the FORTRAN program given in the Appendix, the Gauss 16-point quadrature formula was selected for purposes of illustration. Because of symmetry about the x-z and x-y planes, with this quadrature formula there are 16 i's but only 8 j's, so that the matrices $J_{ij, kl}$ and $L_{ij, kl}$ contain $128 \times 128 = 16384$ elements. An advantage of the Gauss formula is that it gives finer intervals at the bow and stern than amidships, as is required by the rapid changes in form of the transverse sections at a ship's extremities. Nevertheless, for a particular form, it may be desired to use a quadrature formula of the Simpson-rule type, with a fine mesh near the bow and stern and a coarser one over the remainder of the hull.

The input data of the FORTRAN program are the coordinates (x_i, y_{ij}, z_{ij}) , the direction cosines l_{ij}, m_{ij}, n_{ij} , and the weighting factors of the selected quadrature formula, A_k and A_l . The various do-loops in the program perform the following operations:

Loop 2 computes the first approximations for the iteration formulas (63) and (66), $(M_0)_{ij} = SDI(I, J)$ and $(N_0)_{ij} = SDO(I, J)$. Here is also computed the product of the area element $E_{ij} = E(I, J)$ by the weighting factors, $A_i A_j E_{ij} = F(I, J)$.

Loop 9 computes the kernel matrix of the integral equations (63) and (66), $J_{ij, kl}$ which, multiplied by the weighting factors, is denoted by $C(IJ, KL)$. The program yields simultaneously the transpose of the matrix with its corresponding weighting factors, and then computes directly the diagonal terms of the matrix. Also computed in this loop is the matrix $L_{ij, kl} = P1(IJ, KL)$, including the weighting factors and the element of area, used in computing ϕ_i from (68).

Loops 22-23 compute successive approximations to the source distribution $M_n = SDI$ from the iteration formula (63), for 10 iterations.

Loops 25-26 compute successive approximations to the source distribution for the equipotential problem, $N_n = SDO$, from the iteration formula (66), for 20 iterations.

Loop 28 computes $\phi_0 = PHO$ from (67).

Loop 35 computes $\phi_{ij} = PHI(I, J)$ from the expression for the velocity potential in (68).

Results for an Ellipsoid

The program was tested by applying it to an ellipsoid with axes a, b, c in the proportions

$$a : b : c = 4 : 1 : 2$$

Input data are given in Table 1, in which θ is the parametric angle of the equation of the ellipsoid, (70). The values of x and θ in the table correspond to the abscissas of the Gauss 16-point quadrature formula.

Values of the source distribution, $M(Q)$, computed from the exact expression as well as from the discretized iteration formula with 10 iterations, are given in Table 2. The results are seen to agree to within four significant figures for θ near zero, but to only three significant figures for θ near $\pi/2$. Repetition of the calculations employing double-precision arithmetic showed essentially the same results, indicating that the errors are due principally to the discretization of the equations, rather than round-off errors by the computer.

Table 3 gives the results for the source distribution $N(Q)$ when the hull is treated as an equipotential surface. Since the total source

strength on the hull was not normalized in the iterations, an unknown constant factor is present in the values computed from the iteration formula. The ratios of the exact values to those computed with twenty iterations show consistency to five significant figures. The exact value of the constant potential on the hull, $\phi_{OE} = -1.76977$, and the value of the potential at the origin computed from the source distribution of the twentieth iteration, $\phi_{OC} = 4.07197$, are in essentially the same ratio as that given in the Table, $\phi_{OE}/\phi_{OC} = 0.434621$.

Finally, the distribution of the velocity potential over the hull is given in Table 4. The exact values are seen to agree with those computed from the previously calculated source distributions, $M(Q)$ and $N(Q)$, to within at least five significant figures. This is remarkable in view of the aforementioned result that the values of the source distribution $M(Q)$ from the tenth iteration have considerably larger errors. A reexamination of the values of $M(Q)$ in Table 2 shows that the computed values tend to be larger than the exact ones at small values of θ , and smaller than the exact ones at the large values of θ , a possible explanation for the unexpectedly good agreement in the values of the velocity potential.

References

- [1] J. L. Hess and A. M. O. Smith, "Calculation of Potential Flow about Arbitrary Bodies", *Progress in Aeronautical Sciences*, Vol. 8, Pergamon Press, New York, 1966.
- [2] O. D. Kellogg, *Foundations of Potential Theory*, Frederick Ungar Publishing Company, New York, 1929.
- [3] F. H. Todd, "Some Further Experiments on Single-Screw Merchant Ship Forms — Series 60", *Transactions of the Society of Naval Architecture and Marine Engineering*, Vol. 61, 1953.
- [4] J. V. Wehausen, "Surface Waves", *Encyclopedia of Physics*, edited by S. Flügge, Vol. IX, Fluid Dynamics III, Springer Verlag, Berlin, 1960.
- [5] W. Gröbner and N. Hofreiter, *Integraltafel, Zweiter Teil, Bestimmte Integrale*, Springer-Verlag, Wien, New York, 1966.
- [6] E. W. Hobson, *The Theory of Spherical and Ellipsoidal Harmonics*, Chelsea Publishing Company, New York, 1955.
- [7] C. von Kerczek and E. O. Tuck, "The Representation of Ship Hulls by Conformal Mapping Functions", *Journal of Ship Research*, vol. 13, no. 4, December 1969.

TABLE I

INPUT DATA - COORDINATES AND DIRECTION COSINES
OF A 4 : 1 : 2 ELLIPSOID

x = 0.755404

| θ | y | z | l | m | n |
|----------|----------|----------|----------|----------|----------|
| 0.149245 | 0.161994 | 0.048716 | 0.279081 | 0.957566 | 0.071991 |
| 0.442341 | 0.148048 | 0.140244 | 0.296377 | 0.929365 | 0.220094 |
| 0.719450 | 0.123216 | 0.215898 | 0.331168 | 0.864286 | 0.378599 |
| 0.970557 | 0.092529 | 0.270360 | 0.380958 | 0.746614 | 0.545380 |
| 1.186584 | 0.061402 | 0.303743 | 0.435264 | 0.566082 | 0.700069 |
| 1.359729 | 0.034320 | 0.320359 | 0.476405 | 0.346307 | 0.808152 |
| 1.483732 | 0.014244 | 0.326388 | 0.495196 | 0.149402 | 0.855839 |
| 1.554146 | 0.002727 | 0.327584 | 0.499237 | 0.028841 | 0.865985 |

x = 0.865631

| θ | y | z | l | m | n |
|----------|----------|----------|----------|----------|----------|
| 0.149245 | 0.123773 | 0.037224 | 0.399553 | 0.914131 | 0.068726 |
| 0.442341 | 0.113123 | 0.107160 | 0.421931 | 0.882226 | 0.208931 |
| 0.719450 | 0.094149 | 0.164968 | 0.465774 | 0.810548 | 0.355059 |
| 0.970557 | 0.070701 | 0.206581 | 0.525660 | 0.686941 | 0.501791 |
| 1.186584 | 0.046917 | 0.232090 | 0.586995 | 0.509045 | 0.629532 |
| 1.359729 | 0.026224 | 0.244785 | 0.630642 | 0.305678 | 0.713338 |
| 1.483732 | 0.010884 | 0.249393 | 0.649769 | 0.130718 | 0.748808 |
| 1.554146 | 0.002084 | 0.250306 | 0.653818 | 0.025186 | 0.756233 |

x = 0.944575

| θ | y | z | l | m | n |
|----------|----------|----------|----------|----------|----------|
| 0.149245 | 0.081162 | 0.024407 | 0.587148 | 0.807201 | 0.060687 |
| 0.442341 | 0.074174 | 0.070265 | 0.612317 | 0.769333 | 0.182195 |
| 0.719450 | 0.061733 | 0.108169 | 0.658909 | 0.689016 | 0.301822 |
| 0.970557 | 0.046359 | 0.135455 | 0.716912 | 0.562963 | 0.411229 |
| 1.186584 | 0.030764 | 0.152180 | 0.769949 | 0.401221 | 0.496186 |
| 1.359729 | 0.017195 | 0.160505 | 0.804046 | 0.234186 | 0.546504 |
| 1.483732 | 0.007137 | 0.163526 | 0.818094 | 0.098896 | 0.566518 |
| 1.554146 | 0.001367 | 0.164125 | 0.821000 | 0.019004 | 0.570613 |

x = 0.989401

| θ | y | z | l | m | n |
|----------|----------|----------|----------|----------|----------|
| 0.149245 | 0.035899 | 0.010796 | 0.864215 | 0.501707 | 0.037719 |
| 0.442341 | 0.032808 | 0.031079 | 0.877978 | 0.465817 | 0.110316 |
| 0.719450 | 0.027306 | 0.047845 | 0.900792 | 0.397762 | 0.174239 |
| 0.970557 | 0.020505 | 0.059914 | 0.925044 | 0.306740 | 0.224065 |
| 1.186584 | 0.013607 | 0.067311 | 0.943869 | 0.207696 | 0.256855 |
| 1.359729 | 0.007605 | 0.070993 | 0.954546 | 0.117401 | 0.273970 |
| 1.483732 | 0.003157 | 0.072330 | 0.958656 | 0.048936 | 0.280329 |
| 1.554146 | 0.000604 | 0.072595 | 0.959486 | 0.009378 | 0.281599 |

TABLE I, Continued

x = 0.095012

| θ | y | z | l | m | n |
|----------|----------|----------|----------|----------|----------|
| 0.149245 | 0.246102 | 0.074009 | 0.024054 | 0.996897 | 0.074948 |
| 0.442341 | 0.224916 | 0.213060 | 0.025683 | 0.972764 | 0.230372 |
| 0.719450 | 0.187191 | 0.327995 | 0.029045 | 0.915587 | 0.401071 |
| 0.970557 | 0.140571 | 0.410734 | 0.034092 | 0.807036 | 0.589518 |
| 1.186584 | 0.093283 | 0.461450 | 0.039995 | 0.628266 | 0.776970 |
| 1.359729 | 0.052139 | 0.486692 | 0.044815 | 0.393482 | 0.918240 |
| 1.483732 | 0.021640 | 0.495853 | 0.047138 | 0.171776 | 0.984008 |
| 1.554146 | 0.004144 | 0.497669 | 0.047648 | 0.033248 | 0.998311 |

x = 0.281604

| θ | y | z | l | m | n |
|----------|----------|----------|----------|----------|----------|
| 0.149245 | 0.237216 | 0.071337 | 0.073784 | 0.994468 | 0.074766 |
| 0.442341 | 0.216794 | 0.205367 | 0.078753 | 0.970062 | 0.229732 |
| 0.719450 | 0.180432 | 0.316152 | 0.088994 | 0.912339 | 0.399648 |
| 0.970557 | 0.135495 | 0.395903 | 0.104319 | 0.803100 | 0.586642 |
| 1.186584 | 0.089915 | 0.444788 | 0.122155 | 0.624060 | 0.771769 |
| 1.359729 | 0.050256 | 0.469118 | 0.136646 | 0.390183 | 0.910542 |
| 1.483732 | 0.020859 | 0.477948 | 0.143600 | 0.170185 | 0.974893 |
| 1.554146 | 0.003994 | 0.479699 | 0.145126 | 0.032933 | 0.988865 |

x = 0.458017

| θ | y | z | l | m | n |
|----------|----------|----------|----------|----------|----------|
| 0.149245 | 0.219765 | 0.066089 | 0.128809 | 0.988879 | 0.074345 |
| 0.442341 | 0.200846 | 0.190259 | 0.137376 | 0.963859 | 0.228263 |
| 0.719450 | 0.167159 | 0.292894 | 0.154966 | 0.904908 | 0.396393 |
| 0.970557 | 0.125528 | 0.366778 | 0.181103 | 0.794152 | 0.580107 |
| 1.186584 | 0.083300 | 0.412067 | 0.211201 | 0.614585 | 0.760052 |
| 1.359729 | 0.046559 | 0.434608 | 0.235364 | 0.382812 | 0.893341 |
| 1.483732 | 0.019324 | 0.442788 | 0.246862 | 0.166645 | 0.954615 |
| 1.554146 | 0.003700 | 0.444410 | 0.249376 | 0.032234 | 0.967870 |

x = 0.617876

| θ | y | z | l | m | n |
|----------|----------|----------|----------|----------|----------|
| 0.149245 | 0.194384 | 0.058456 | 0.194329 | 0.978176 | 0.073541 |
| 0.442341 | 0.177649 | 0.168285 | 0.206949 | 0.952019 | 0.225459 |
| 0.719450 | 0.147853 | 0.259066 | 0.232674 | 0.890834 | 0.390228 |
| 0.970557 | 0.111030 | 0.324417 | 0.270396 | 0.777425 | 0.567888 |
| 1.186584 | 0.073680 | 0.364476 | 0.312994 | 0.597176 | 0.738522 |
| 1.359729 | 0.041182 | 0.384413 | 0.346471 | 0.369481 | 0.862231 |
| 1.483732 | 0.017092 | 0.391649 | 0.362157 | 0.160293 | 0.918231 |
| 1.554146 | 0.003273 | 0.393083 | 0.365566 | 0.030982 | 0.930270 |

TABLE 2
SOURCE DISTRIBUTION ON ELLIPSOID, M(P)

| Exact | | Computed - 10th Interaction | | | | | |
|-----------|-----------|-----------------------------|-----------|-----------|-----------|-----------|-----------|
| -0.002157 | -0.006615 | -0.011548 | -0.017422 | -0.025020 | -0.035820 | -0.052639 | -0.077478 |
| -0.002303 | -0.007060 | -0.012316 | -0.018553 | -0.026571 | -0.037827 | -0.054895 | -0.078712 |
| -0.002604 | -0.007978 | -0.013893 | -0.020860 | -0.029690 | -0.041757 | -0.059072 | -0.080757 |
| -0.003056 | -0.009352 | -0.016236 | -0.024241 | -0.034153 | -0.047126 | -0.064272 | -0.082931 |
| -0.003586 | -0.010951 | -0.018935 | -0.028060 | -0.039022 | -0.052625 | -0.069027 | -0.084619 |
| -0.004018 | -0.012250 | -0.021101 | -0.031062 | -0.042710 | -0.056538 | -0.072084 | -0.085576 |
| -0.004226 | -0.012874 | -0.022131 | -0.032468 | -0.044395 | -0.058253 | -0.073343 | -0.085945 |
| -0.004272 | -0.013011 | -0.022357 | -0.032774 | -0.044757 | -0.058616 | -0.073604 | -0.086019 |
| -0.002157 | -0.006615 | -0.011548 | -0.017422 | -0.025021 | -0.035822 | -0.052640 | -0.077497 |
| -0.002303 | -0.007061 | -0.012317 | -0.018555 | -0.026573 | -0.037830 | -0.054902 | -0.078720 |
| -0.002605 | -0.007981 | -0.013897 | -0.020866 | -0.029699 | -0.041771 | -0.059094 | -0.080748 |
| -0.003058 | -0.009356 | -0.016243 | -0.024251 | -0.034166 | -0.047141 | -0.064278 | -0.082904 |
| -0.003585 | -0.010950 | -0.018932 | -0.028056 | -0.039014 | -0.052608 | -0.068994 | -0.084577 |
| -0.004014 | -0.012238 | -0.021079 | -0.031030 | -0.042666 | -0.056480 | -0.072016 | -0.085525 |
| -0.004219 | -0.012853 | -0.022096 | -0.032417 | -0.044328 | -0.058172 | -0.073258 | -0.085889 |
| -0.004264 | -0.012988 | -0.022318 | -0.032718 | -0.044685 | -0.058529 | -0.073515 | -0.085963 |

TABLE 4
VELOCITY POTENTIAL ON SURFACE OF ELLIPSOID

| | Exact | | | | | | |
|----------|----------|----------|----------|----------|----------|----------|----------|
| 0.107040 | 0.317252 | 0.515997 | 0.696093 | 0.851031 | 0.975211 | 1.064149 | 1.114649 |
| | Computed | | | | | | |
| 0.107038 | 0.317247 | 0.515989 | 0.696082 | 0.851017 | 0.975195 | 1.064130 | 1.114623 |
| 0.107038 | 0.317247 | 0.515989 | 0.696082 | 0.851017 | 0.975195 | 1.064129 | 1.114628 |
| 0.107038 | 0.317246 | 0.515988 | 0.696081 | 0.851016 | 0.975193 | 1.064127 | 1.114636 |
| 0.107038 | 0.317246 | 0.515988 | 0.696081 | 0.851016 | 0.975195 | 1.064134 | 1.114643 |
| 0.107039 | 0.317248 | 0.515991 | 0.696086 | 0.851023 | 0.975204 | 1.064146 | 1.114650 |
| 0.107040 | 0.317251 | 0.515997 | 0.696093 | 0.851033 | 0.975215 | 1.064156 | 1.114654 |
| 0.107041 | 0.317254 | 0.516001 | 0.696098 | 0.851038 | 0.975221 | 1.064161 | 1.114655 |
| 0.107041 | 0.317254 | 0.516002 | 0.696100 | 0.851040 | 0.975223 | 1.064162 | 1.114656 |

APPENDIX

FORTTRAN IV(G) Program for IBM 360/
Program for Computing Source Distribution
and Velocity Potential on a Ship Surface

```
DIMENSION X(16),Y(16,8),Z(16,8),EL(16,8),EM(16,8),EN(16,8),A(16),  
1CAS(128),SD(128),SDA(128),SD3(128),SDE(16,8),SDO(16,8),SD1(16,8)  
2PHI(16,8),C(128,128),P1(128,128),F(16,8),E(16,8)  
P = 12.56637061  
READ(5,1) (X(I),I=1,16)  
1 FORMAT(8F10.8)  
READ(5,1) (A(I),I=1,16)  
READ(5,1) ((Y(I,J),I=1,8),J=1,8)  
READ(5,1) ((Y(I,J),I=9,16),J=1,8)  
READ(5,1) ((Z(I,J),I=1,8),J=1,8)  
READ(5,1) ((Z(I,J),I=9,16),J=1,8)  
READ(5,1) ((EL(I,J),I=1,8),J=1,8)  
READ(5,1) ((EL(I,J),I=9,16),J=1,8)  
READ(5,1) ((EM(I,J),I=1,8),J=1,8)  
READ(5,1) ((EM(I,J),I=9,16),J=1,8)  
READ(5,1) ((EN(I,J),I=1,8),J=1,8)  
READ(5,1) ((EN(I,J),I=9,16),J=1,8)  
DO 2 I = 1,16  
X2 = X(I) * X(I)  
DO 2 J = 1,8  
J1 = J + 8  
Y2 = Y(I,J) * Y(I,J)  
Z2 = Z(I,J) * Z(I,J)  
E(I,J) = .....  
F(I,J) = E(I,J) * A(I) * A(J1)  
SD1(I,J) = - EL(I,J) / P  
2 SDO(I,J) = SQRT(X2 + Y2 + Z2)  
KL = 0  
DO 9 K = 1,16  
DO 9 L = 1,8  
KL = KL + 1  
SD3(KL) = SDO(K,L)  
CAS(KL) = SD1(K,L)  
SD(KL) = SD1(K,L)  
IJ = 0  
SUM = 0.0  
DO 8 I = 1,16  
DO 8 J = 1,8  
IJ = IJ + 1  
X0 = X(K) - X(I)  
Y0 = Y(K,L) - Y(I,J)  
Y1 = Y(K,L) + Y(I,J)  
Z0 = Z(K,L) - Z(I,J)  
Z1 = Z(K,L) + Z(I,J)  
X02 = X0 * X0  
Y02 = Y0 * Y0  
Y12 = Y1 * Y1  
Z02 = Z0 * Z0  
Z12 = Z1 * Z1
```


APPENDIX, Continued

```
R11 = SQRT(X02 + Y02 + Z02)
R22 = SQRT(X02 + Y12 + Z02)
R33 = SQRT(X02 + Y12 + Z12)
R44 = SQRT(X02 + Y02 + Z12)
R1 = R11 * R11 * R11
R2 = R22 * R22 * R22
R3 = R33 * R33 * R33
R4 = R44 * R44 * R44
V2 = (1.0/R2) + (1.0/R3)
V3 = (1.0/R4) + (1.0/R3)
W2 = EM(I,J) * Y(K,L) * V2
W3 = EN(I,J) * Z(K,L) * V3
PIJ = X0 * EL(I,J) + Y0 * EM(I,J) + Z0 * EN(I,J)
IF(K.EQ.1.AND.L.EQ.J) GO TO 10
V11 = (1.0/R11) + (1.0/R22) + (1.0/R33) + (1.0/R44)
V1 = (1.0/R1) + (1.0/R2) + (1.0/R3) + (1.0 / R4)
P1(IJ,KL) = V11
BA = PIJ * V1 - 2.0 * (W2 + W3)
BE = BA * F(I,J)
GO TO 11
10 BA = 0.00000000
BE = 0.00000000
P1(IJ,KL) = BE
11 C(IJ,KL) = BA * F(K,L)
8 SUM = SUM + BE
9 C(KL,KL) = C(KL,KL) - SUM
LA = 1
13 DO 22 IJ = 1,128
SUM = 0.0
DO 21 KL = 1,128
21 SUM = SUM + SD(KL) * C(IJ,KL)
22 SDA(IJ) = 0.125 * SUM + CAS(IJ)
IJ = 0
DO 23 I = 1,16
DO 23 J = 1,8
IJ = IJ + 1
SD(IJ) = SDA(IJ)
23 SDE(I,J) = SDA(IJ)
LA = LA + 1
IF(LA.LE.10) GO TO 13
WRITE(6,102)((SDE(I,J),I=1,8),J=1,8)
102 FORMAT(40H SOURCE DISTRIBUTION AFTER 10 ITERATIONS/(8F10.6))
WRITE(6,110)((SDE(I,J),I=9,16),J=1,8)
110 FORMAT(1H//((8F10.6))
LA = 1
27 DO 25 IJ = 1,128
SUM = 0.0
DO 24 KL = 1,128
24 SUM = SUM + SD3(KL) * C(IJ,KL)
25 SD(IJ) = SD3(IJ) - 0.25 * SUM
IJ = 0
DO 26 I = 1,16
DO 26 J = 1,8
IJ = IJ + 1
SD3(IJ) = SD(IJ)
```

APPENDIX, Continued

```
26 SDO(I,J) = SD(IJ)
   LA = LA + 1
   IF(LA.LE.20) GO TO 27
   WRITE(6,104)((SDO(I,J),I=1,8),J=1,8)
104 FORMAT(40H DIRICHLET PROBLEM AFTER 20 ITERATIONS /(8F10.6))
   PHO = 0.0
   DO 28 I = 1,16
   DO 28 J = 1,8
   VV= SQRT(X(I)*X(I) + Y(I,J)*Y(I,J) + Z(I,J)*Z(I,J))
28  PHO = PHO - 4.0*SDO(I,J) * F(I,J) * P / VV
   IJ = 0
   DO 35 I = 1,16
   DO 35 J = 1,8
   IJ = IJ + 1
   COEF = SDE(I,J) / SDO(I,J)
   KL = 0

   SUM = 0.0
   DO 33 K = 1,16
   DO 33 L = 1,8
   KL = KL + 1
33  SUM = SUM + P1(IJ,KL)*(SDA(KL)-SD3(KL)*COEF) * F(K,L) * P
35  PHI(I,J) = X(I) - SUM + PHO * COEF
   WRITE(6,107) (X(I),I=1,8)
107 FORMAT(1H ///(8F10.6))
   WRITE(6,110)((PHI(I,J),I=1,8),J=1,8)
   WRITE(6,107) (X(I),I=9,16)
   WRITE(6,110)((PHI(I,J),I=9,16),J=1,8)
   CALL EXIT
   END
```

DISTRIBUTION LIST FOR
TECHNICAL REPORTS ISSUED UNDER
CONTRACT Nonr-1611(07) TASK 062-183
OFFICE OF NAVAL RESEARCH

Technical Library
Building 131
Aberdeen Proving Ground, Maryland 21005

Defense Documentation Center (20)
Cameron Station
Alexandria, Virginia 22314

Technical Library
Naval Ship Research and
Development Center
Annapolis Division
Annapolis, Maryland 21402

Professor Bruce Johnson
Engineering Department
Naval Academy
Annapolis, Maryland 21402

Library
Naval Academy
Annapolis, Maryland 21402

Professor W. P. Graebel
Department of Engineering
Mechanics
The University of Michigan
College of Engineering
Ann Arbor, Michigan 48104

Professor W. R. Debler
Department of Engineering Mechanics
University of Mechanics
Ann Arbor, Michigan 48108

Dr. Francis Ogilvie
Department of Naval Architecture
and Marine Engineering
University of Michigan
Ann Arbor, Michigan 48108

Professor S. D. Sharma
Department of Naval Architecture
and Marine Engineering
University of Michigan
Ann Arbor, Michigan 48108

Professor W. W. Willmarth
Department of Aerospace Engineering
University of Michigan
Ann Arbor, Michigan 48108

Professor Finn C. Michelsen
Naval Architecture and Marine
Engineering
445 West Engineering Bldg.
University of Michigan
Ann Arbor, Michigan 48104

AFOSR (REM)
1400 Wilson Boulevard
Arlington, Virginia 22204

Dr. J. Menkes
Institute for Defense Analyses
400 Army-Navy Drive
Arlington, Virginia 22204

Professor S. Corrsin
Mechanics Department
The Johns Hopkins University
Baltimore, Maryland 20910

Professor O. M. Phillips
The Johns Hopkins University
Baltimore, Maryland 20910

Professor L. S. G. Kovaszny
The Johns Hopkins University
Baltimore, Maryland 20910

Librarian
Department of Naval Architecture
University of California
Berkeley, California 94720

Professor P. Lieber
Department of Mechanical Engineering
University of California
Institute of Engineering Research
Berkeley, California 94720

Professor M. Holt
Division of Aeronautical Sciences
University of California
Berkeley, California 94720

Professor J. V. Wehausen
Department of Naval Architecture
University of California
Berkeley, California 94720

Professor J. R. Paulling
Department of Naval Architecture
University of California
Berkeley, California 94720

Professor E. V. Laitone
Department of Mechanical Engineering
University of California
Berkeley, California 94720

School of Applied Mathematics
Indiana University
Bloomington, Indiana 47401

Commander
Boston Naval Shipyard
Boston, Massachusetts 02129

Director
Office of Naval Research
Branch Office
495 Summer Street
Boston, Massachusetts 02210

Professor M. S. Uberoi
Department of Aeronautical Engineering
University of Colorado
Boulder, Colorado 80303

Naval Applied Science Laboratory
Technical Library
Bldg. 1 Code 222
Flushing and Washington Avenues
Brooklyn, New York 11251

Professor J. J. Foody
Chairman, Engineering Department
State University of New York
Maritime College
Bronx, New York 10465

Dr. Irving C. Statler, Head
Applied Mechanics Department
Cornell Aeronautical Laboratory, Inc.
P. O. Box 235
Buffalo, New York 14221

Dr. Alfred Ritter
Assistant Head, Applied Mechanics Dept.
Cornell Aeronautical Laboratory, Inc.
Buffalo, New York 14221

Professor G. Birkhoff
Department of Mathematics
Harvard University
Cambridge, Massachusetts 02138

Commanding Officer
NROTC Naval Administrative Unit
Massachusetts Institute of Technology
Cambridge, Massachusetts 02139

Professor N. Newman
Department of Naval Architecture and
Marine Engineering
Massachusetts Institute of Technology
Cambridge, Massachusetts 02139

Professor A. H. Shapiro
Department of Mechanical Engineering
Massachusetts Institute of Technology
Cambridge, Massachusetts 02139

Professor C. C. Lin
Department of Mathematics
Massachusetts Institute of Technology
Cambridge, Massachusetts 02139

Professor E. W. Merrill
Department of Mathematics
Massachusetts Institute of Technology
Cambridge, Massachusetts 02139

Professor M. A. Abkowitz
Department of Naval Architecture
and Marine Engineering
Massachusetts Institute of Technology
Cambridge, Massachusetts 02139

Professor G. H. Carrier
Department of Engineering and
Applied Physics
Harvard University
Cambridge, Massachusetts 02139

Professor E. Mollo-Christensen
Room 54-1722
Massachusetts Institute of Technology
Cambridge, Massachusetts 02139

Professor A. T. Ippen
Department of Civil Engineering
Massachusetts Institute of Technology
Cambridge, Massachusetts 02139

Commander
Charleston Naval Shipyard
U. S. Naval Base
Charleston, South Carolina 29408

A. R. Kuhlthau, Director
Research Laboratories for the
Engineering Sciences
Thorton Hall, University of Virginia
Charlottesville, Virginia 22903

Director
Office of Naval Research
Branch Office
219 Dearborn Street
Chicago, Illinois 60604

Library
Naval Weapons Center
China Lake, California 93557

Library MS 60-3
NASA Lewis Research Center
21000 Brookpark Road
Cleveland, Ohio 44135

Professor J. M. Burgers
Institute of Fluid Dynamics and
Applied Mathematics
University of Maryland
College Park, Maryland 20742

Acquisition Director
NASA Scientific & Technical
Information
P. O. Box 33
College Park, Maryland 20740

Professor Pai
Institute for Fluid Dynamics
and Applied Mathematics
University of Maryland
College Park, Maryland 20740

Technical Library
Naval Weapons Laboratory
Dahlgren, Virginia 22448

Computation & Analyses Laboratory
Naval Weapons Laboratory
Dahlgren, Virginia 22448

Professor C. S. Wells
LTV Research Center
Dallas, Texas 75222

Dr. R. H. Kraichnan
Dublin, New Hampshire 03444

Commanding Officer
Army Research Office
Box CM, Duke Station
Durham, North Carolina 27706

Professor A. Charnes
The Technological Institute
Northwestern University
Evanston, Illinois 60201

Dr. Martin H. Bloom
Polytechnic Institute of Brooklyn
Graduate Center, Dept. of Aerospace
Engineering and Applied Mechanics
Farmingdale, New York 11735

Technical Documents Center
Building 315
U. S. Army Mobility Equipment
Research and Development Center
Fort Belvoir, Virginia 22060

Professor J. E. Cermak
College of Engineering
Colorado State University
Ft. Collins, Colorado 80521

Technical Library
Webb Institute of Naval Architecture
Glen Cove, Long Island, New York 11542

Professor E. V. Lewis
Webb Institute of Naval Architecture
Glen Cove, Long Island, New York 11542

Library MS 185
NASA, Langley Research Center
Langley Station
Hampton, Virginia 23365

Dr. B. N. Pridmore Brown
Northrop Corporation
NORAIR-Div.
Hawthorne, California 90250

Dr. J. P. Breslin
Stevens Institute of Technology
Davidson Laboratory
Hoboken, New Jersey 07030

Mr. D. Savitsky
Stevens Institute of Technology
Davidson Laboratory
Hoboken, New Jersey 07030

Mr. C. H. Henry
Stevens Institute of Technology
Davidson Laboratory
Hoboken, New Jersey 07030

Professor J. F. Kennedy, Director
Iowa Institute of Hydraulic Research
State University of Iowa
Iowa City, Iowa 52240

Professor L. Landweber
Iowa Institute of Hydraulic Research
State University of Iowa
Iowa City, Iowa 52240

Professor E. L. Resler
Graduate School of
Aeronautical Engineering
Cornell University
Ithaca, New York 14851

Professor John Miles
I.G.P.P.
University of California, San Diego
La Jolla, California 92038

Director
Scripps Institution of Oceanography
University of California
La Jolla, California 92037

Dr. B. Sternlicht
Mechanical Technology Incorporated
968 Albany-Shaker Road
Latham, New York 12110

Mr. P. Eisenberg, President
Hydronautics
Pindell School Road
Howard County
Laurel, Maryland 20810

Professor A. Ellis
University of California, San Diego
Department of Aerospace & Mech. Engrg. Sci.
La Jolla, California 92037

Mr. Alfonso Alcedan L. Director
Laboratorio Nacional De Hydraulics
Antigui Cameno A. Ancon
Casilla Jostal 682
Lima, Peru

Commander
Long Beach Naval Shipyard
Long Beach, California 90802

Professor John Laufer
Department of Aerospace Engineering
University Park
Los Angeles, California 90007

Professor J. Ripkin
St. Anthony Falls Hydraulic Lab.
University of Minnesota
Minneapolis, Minnesota 55414

Professor J. M. Killen
St. Anthony Falls Hydraulic Lab.
University of Minnesota
Minneapolis, Minnesota 55414

Lorenz G. Straub Library
St. Anthony Falls Hydraulic Lab.
Mississippi River at 3rd Avenue SE.
Minneapolis, Minnesota 55414

Dr. E. Silberman
St. Anthony Falls Hydraulic Lab.
University of Minnesota
Minneapolis, Minnesota 55414

Superintendent
Naval Postgraduate School
Library Code 0212
Monterey, California 93940

Professor A. B. Metzner
University of Delaware
Newark, New Jersey 19711

Technical Library
USN Underwater Weapons &
Research & Engineering Station
Newport, Rhode Island 02840

Technical Library
Underwater Sound Laboratory
Fort Trumbull
New London, Connecticut 06321

(2)

Professor J. J. Stoker
Institute of Mathematical Sciences
New York University
251 Mercer Street
New York, New York 10003

Engineering Societies Library
345 East 47th Street
New York, New York 10017

Office of Naval Research
New York Area Office
207 W. 24th Street
New York, New York 10011

Commanding Officer
Office of Naval Research
Branch Office
Box 39
FPO New York, New York 09510 (25)

Professor H. Elrod
Department of Mechanical Engineering
Columbia University
New York, New York 10027

Society of Naval Architects and
Marine Engineering
74 Trinity Place
New York, New York 10006

Professor S. A. Piascek
Department of Engineering Mechanics
University of Notre Dame
Notre Dame, Indiana 46556

United States Atomic Energy Commission
Division of Technical Information
Extension
P. O. Box 62
Oak Ridge, Tennessee 37830

Miss O. M. Leach, Librarian
National Research Council
Aeronautical Library
Montreal Road
Ottawa 7, Canada

Technical Library
Naval Ship Research and
Development Center
Panaman City, Florida 32401

Library
Jet Propulsion Laboratory
California Institute of Technology
4800 Oak Grove Avenue
Pasadena, California 91109

Professor M. S. Plesset
Engineering Division
California Institute of Technology
Pasadena, California 91109

Professor H. Liepmann
Department of Aeronautics
California Institute of Technology
Pasadena, California 91109

Technical Library
Naval Undersea Warfare Center
3202 E. Foothill Boulevard
Pasadena, California 91107

Dr. J. W. Hoyt
Naval Undersea Warfare Center
3202 E. Foothill Boulevard
Pasadena, California 91107

Professor T. Y. Wu
Department of Engineering
California Institute of Technology
Pasadena, California 91109

Director
Office of Naval Research
Branch Office
1030 E. Green Street
Pasadena, California 91101

Professor A. Acosta
Department of Mechanical Engineering
California Institute of Technology
Pasadena, California 91109

Naval Ship Engineering Center
Philadelphia Division
Technical Library
Philadelphia, Pennsylvania 19112

Technical Library (Code 249B)
Philadelphia Naval Shipyard
Philadelphia, Pennsylvania 19112

Professor R. C. Mac Camy
Department of Mathematics
Carnegie Institute of Technology
Pittsburgh, Pennsylvania 15213

Dr. Paul Kaplan
Oceanics, Inc.
Plainview, Long Island, New York 11803

Technical Library
Naval Missile Center
Point Mugu, California 93441

Technical Library
Naval Civil Engineering Lab.
Port Hueneme, California 93041

Commander
Portsmouth Naval Shipyard
Portsmouth, New Hampshire 03801

Commander
Norfolk Naval Shipyard
Portsmouth, Virginia 23709

Professor F. E. Bisshopp
Division of Engineering
Brown University
Providence, Rhode Island 02912

Dr. L. L. Higgins
TRW Space Technology Labs, Inc.
One Space Park
Redondo Beach, California 90278

Redstone Scientific Information Center
Attn: Chief, Document Section
Army Missile Command
Redstone Arsenal, Alabama 35809

Dr. H. N. Abramson
Southwest Research Institute
8500 Culebra Road
San Antonio, Texas 78228

Editor
Applied Mechanics Review
Southwest Research Institute
8500 Culebra Road
San Antonio, Texas 78206

Librarian
Naval Command Control Communications
Laboratory Center
San Diego, California 92152

Library & Information Services
General Dynamics-Convair
P. O. Box 1128
San Diego, California 92112

Commander (Code 246P)
Pearl Harbor Naval Shipyard
Box 400
FPO San Francisco, California 96610

Technical Library (Code H245C-3)
Hunters Point Division
San Francisco Bay Naval Shipyard
San Francisco, California 94135

Office of Naval Research
San Francisco Area Office
1076 Mission Street
San Francisco, California 94103

Dr. A. May
Naval Ordnance Laboratory
White Oak
Silver Spring, Maryland 20910

Fenton Kennedy Document Library
The Johns Hopkins University
Applied Physics Laboratory
8621 Georgia Avenue
Silver Spring, Maryland 20910

Librarian
Naval Ordnance Laboratory
White Oak
Silver Spring, Maryland 20910

Dr. Bryne Perry
Department of Civil Engineering
Stanford University
Stanford, California 94305

Professor Milton Van Dyke
Department of Aeronautical Engineering
Stanford University
Stanford, California 94305

Professor E. Y. Hsu
Department of Civil Engineering
Stanford University
Stanford, California 94305

Dr. R. L. Street
Department of Civil Engineering
Stanford University
Stanford, California 94305

Professor S. Eskinazi
Department of Mechanical Engineering
Syracuse University
Syracuse, New York 13210

Professor R. Pfeffer
Florida State University
Geophysical Fluid Dynamics Institute
Tallahassee, Florida 32306

Professor J. Foa
Department of Aeronautical Engineering
Rensselaer Polytechnic Institute
Troy, New York 12180

Professor R. C. Di Prima
Department of Mathematics
Rensselaer Polytechnic Institute
Troy, New York 12180

Dr. M. Sevik
Ordnance Research Laboratory
Pennsylvania State University
University Park, Pennsylvania 16801

Professor J. Lumley
Ordnance Research Laboratory
Pennsylvania State University
University Park, Pennsylvania 16801

Dr. J. M. Robertson
Department of Theoretical and
Applied Mechanics
University of Illinois
Urbana, Illinois 61803

Shipyard Technical Library
Code 130L7 Building 746
San Francisco Bay Naval Shipyard
Vallejo, California 94592

Code L42
Naval Ship Research and
Development Center
Washington, D.C. 20007

Code 800
Naval Ship Research and
Development Center
Washington, D.C. 20007

Code 2027
U. S. Naval Research Laboratory
Washington, D.C. 20390

(6)

Code 438
Chief of Naval Research
Department of the Navy
Washington, D.C. 20360

(3)

Code 513
Naval Ship Research and
Development Center
Washington, D.C. 20007

Science & Technology Division
Library of Congress
Washington, D.C. 20540

ORD 913 (Library)
Naval Ordnance Systems Command
Washington, D.C. 20360

Code 6420
Naval Ship Engineering Center
Concept Design Division
Washington, D.C. 20360

Code 500
Naval Ship Research and
Development Center
Washington, D.C. 20007

Code 901
Naval Ship Research and
Development Center
Washington, D.C. 20007

Code 520
Naval Ship Research and
Development Center
Washington, D.C. 20007

Code 0341
Naval Ship Systems Command
Department of the Navy
Washington, D.C. 20360

Code 2052 (Technical Library)
Naval Ship Systems Command
Department of the Navy
Washington, D.C. 20360

Mr. J. L. Schuler (Code 03412)
Naval Ship Systems Command
Department of the Navy
Washington, D.C. 20360

Dr. J. H. Huth (Code 031)
Naval Ship Systems Command
Department of the Navy
Washington, D.C. 20360

Code 461
Chief of Naval Research
Department of the Navy
Washington, DC. 20360

Code 530
Naval Ship Research and
Development Center
Washington, D.C. 20360

Code 466
Chief of Naval Research
Department of the Navy
Washington, D.C. 20360

Office of Research and Development
Maritime Administration
441 G. Street, NW.
Washington, D.C. 20235

Code 463
Chief of Naval Research
Department of the Navy
Washington, D.C. 20360

National Science Foundation
Engineering Division
1800 G. Street, NW.
Washington, D.C. 20550

Dr. G. Kulin
National Bureau of Standards
Washington, D.C. 20234

Department of the Army
Coastal Engineering Research Center
5201 Little Falls Road, NW.
Washington, D.C. 20011

Code 521
Naval Ship Research and
Development Center
Washington, D.C. 20007

Code 481
Chief of Naval Research
Department of the Navy
Washington, D.C. 20390

Code 421
Chief of Naval Research
Department of the Navy
Washington, D.C. 20360

Commander
Naval Ordnance Systems Command
Code ORD 035
Washington, D.C. 20360

Librarian Station 5-2
Coast Guard Headquarters
1300 E. Street, NW.
Washington, D.C. 20226

Division of Ship Design
Maritime Administration
441 G. Street, NW.
Washington, D.C. 20235

HQ USAF (AFRSTD)
Room 1D 377
The Pentagon
Washington, D.C. 20330

Commander
Naval Ship Systems Command
Code 6644C
Washington, D.C. 20360

Code 525
Naval Ship Research and
Development Center
Washington, D.C. 20007

Dr. A. Powell (Code 01)
Naval Ship Research and
Development Center
Washington, D.C. 20007

Director of Research Code RR
National Aeronautics & Space Admin.
600 Independence Avenue, SW.
Washington, D.C. 20546

Commander
Naval Ordnance Systems Command
Code 03
Washington, D.C. 20360

Code ORD 05411
Naval Ordnance Systems Command
Washington, D.C. 20360

AIR 5301
Naval Air Systems Command
Department of the Navy
Washington, D.C. 20360

Mr. Ralph Lacey (Code 6114)
Naval Ship Engineering Center
Department of the Navy
Washington, D.C. 20360

AIR 604
Naval Air Systems Command
Department of the Navy
Washington, D.C. 20360

Dr. A. S. Iberall, President
General Technical Services, Inc.
451 Penn Street
Yeadon, Pennsylvania 19050

Dr. John Craven (PM 1100)
Deep Submergence Systems
Project
Department of the Navy
Washington, D.C. 20360

Dr. H. Cohen
IBM Research Center
P. O. Box 218
Yorktown Heights, New York 10598

Code 522
Naval Ship Research and
Development Center
Washington, D.C. 20007

Commander
Naval Oceanographic Office
Washington, D.C. 20390

Chief of Research & Development
Office of Chief of Staff
Department of the Army
The Pentagon, Washington, D.C. 20310

Code 6342A
Naval Ship Systems Command
Department of the Navy
Washington, D.C. 20360

Code 468
Chief of Naval Research
Department of the Navy
Washington, D.C. 20360

Director
U. S. Naval Research Laboratory
Code 6170
Washington, D.C. 20390

Code 473
Chief of Naval Research
Department of the Navy
Washington, D.C. 20360

Code 6100
Naval Ship Engineering Center
Department of the Navy
Washington, D.C. 20360

DOCUMENT CONTROL DATA - R & D

Security classification of title, body of abstract and indexing annotation must be entered when the overall report is classified

| | |
|--|--|
| 1. ORIGINATING ACTIVITY (Corporate author) IOWA INSTITUTE OF HYDRAULIC RESEARCH | 2a. REPORT SECURITY CLASSIFICATION Unclassified |
| | 2b. GROUP |

3. REPORT TITLE

Irrotational Flow About Ship Forms

4. DESCRIPTIVE NOTES (Type of report and, inclusive dates)
Interim Report

5. AUTHOR(S) (First name, middle initial, last name)

Louis Landweber and Matilde Macagno

| | | |
|---------------------------------|------------------------------|----------------------|
| 6. REPORT DATE December 1969 | 7a. TOTAL NO. OF PAGES 33 | 7b. NO. OF REFS 7 |
|---------------------------------|------------------------------|----------------------|

| | |
|--|---|
| 8a. CONTRACT OR GRANT NO. Nonr-1611(07) | 9a. ORIGINATOR'S REPORT NUMBER(S) IIHR Report No. 123 |
| b. PROJECT NO. Task 062-183 | |
| c. d. | 9b. OTHER REPORT NO(S) (Any other numbers that may be assigned this report) |

10. DISTRIBUTION STATEMENT

This document has been approved for public release and sale; its distribution is unlimited.

| | |
|-------------------------|--|
| 11. SUPPLEMENTARY NOTES | 12. SPONSORING MILITARY ACTIVITY Office of Naval Research |
|-------------------------|--|

13. ABSTRACT

A method of computing the potential flow about a ship model in terms of a source distribution on the hull surface is described. After a formulation and discussion of the flow with wavemaking, the zero-Froude number, double-ship-model case is treated in detail. The Fredholm singular integral equation of the second kind for the source distribution is solved numerically by removing the singularity, replacing the integral by a quadrature formula, and solving the resulting high-order set of linear equations by an iteration formula for which convergence is proved. The corresponding velocity potential on the hull surface, the evaluation of which from the source distribution also requires the calculation of a singular integral, is obtained by first solving the equipotential problem for the hull form (which employs the same kernel as the original integral equation). The solution of this Dirichlet problem is then used to remove the singularity from the velocity-potential integral, and then the latter is computed by means of a quadrature formula. The method is applied to an ellipsoid, results from which are compared with the exact solution. A FORTRAN program is included as an Appendix.

| 14. KEY WORDS | LINK A | | LINK B | | LINK C | |
|---|--------|----|--------|----|--------|----|
| | ROLE | WT | ROLE | WT | ROLE | WT |
| Ship Form Irrotational Flow Three Dimensional Flow Source Distributions Integral Equations Wave Making | | | | | | |