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Modeling, Validation, and Performance of Low-Frequency Piezoelectric Energy Harvesters

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Abstract

Analytical and finite element electromechanical models that take into account the fact that the piezoelectric sheet does not cover the whole substrate beam are developed. A linear analysis of the analytical model is performed to determine the optimal load resistance. The analytical and finite element models are validated with experimental measurements. The results show that the analytical model that takes into account the fact that the piezoelectric patch does not cover the whole beam predicts accurately the experimental measurements. The finite element results yield a slight discrepancy in the global frequency and a slight overestimation in the value of the harvested power at resonance. On the other hand, using an approximate analytical model based on mode shapes of the full covered beam leads to erroneous results and overestimation of the global frequency as well as the

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8 level of harvested power. In order to design enhanced piezoelectric energy
9 harvesters that can generate energy at low frequency excitations, further
10 analysis is performed to investigate the effects of varying the length of the
11 piezoelectric material on the natural frequency and the performance of the
12 harvester. The results show that there is a compromise between the length
13 of the piezoelectric material, the electrical load resistance, and the available
14 excitation frequency. By quantifying this compromise, we optimize the per-
15 formance of beam-mass systems to efficiently harvest energy from a specified
16 low-frequency of the ambient vibrations.
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24 *Keywords:* Energy harvesting, Piezoelectric material, Low-frequency,
25 Distributed-parameter model, Finite element analysis.
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28 29 30 **1. Introduction**

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32 Different structural systems for energy harvesting from ambient or aeroe-
33 lastic vibrations have been proposed (Erturk et al., 2010; Sousa et al., 2010;
34 Abdelkefi et al., 2012a; Daqaq, 2012; Abdelkefi et al., 2012b, 2013). These
35 systems vary from simple beam and beam-mass systems to more complex
36 structures, such as zigzag and spiral systems (Karami and Inman, 2011).
37 The purpose for these variations is to enable harvesting energy at specific
38 frequencies. One advantage for using unimorph piezoelectric beam-mass sys-
39 tems for energy harvesting from ambient vibrations is their simple config-
40 uration and design and optimal performance. An issue, however, has been
41 the accurate modeling of these simple devices which is required for efficient
42 design. Such modeling is required because when operating in their linear
43 regime, piezoelectric beam-mass harvesters will only be able to efficiently
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8 harvest energy over a narrow range of frequencies. Missing this range of fre-
9 quencies will result in significantly reduced harvested power. Roundy and
10 Wright (2005) and DuToil et al. (2005) modeled a piezoelectric cantilever
11 beam as a mass-spring-damper system. Such a model is limited to the fun-
12 damental frequency of the structure and does not account for the effects of
13 the dynamic mode shapes on the electrical response of the harvester. The
14 effect of the spring mass (distributed mass) was not considered in the forcing
15 amplitude. This assumption fails in cases where the proof mass is small.
16 Erturk and Inman (2008) showed that the use of the traditional form of the
17 lumped-parameter model leads to erroneous results for both transverse and
18 longitudinal cantilevered beams under base vibrations. They showed that
19 the predicted response can be underestimated when using the conventional
20 effective mass of cantilevered beams or bars. This is particularly true when
21 there is a small or no tip mass. As such, they introduced correction factors
22 to improve the prediction capability of lumped models for harmonic base
23 excitation for both transverse and longitudinal vibrations. Improved models
24 that are based on the Galerkin discretization were used in different stud-
25 ies (Erturk and Inman, 2008, 2009; Abdelkefi et al., 2011, 2012c; Masana
26 and Daqaq, 2011; Ben Ayed et al., 2013). This method is more accurate
27 in comparison to the lumped-parameter models. This approach takes into
28 consideration the effects of the dynamic mode shapes, strain distribution and
29 higher vibration modes on the electromechanical response of the harvester.
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49 A common problem, when basing a reduced-order model for piezoelectric
50 energy harvesters on the Galerkin approximation, is the determination of
51 the mode shapes and natural frequencies. The classical mode shapes of a
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8 fully-covered piezoelectric cantilever beam is usually assumed for experiments
9 where the piezoelectric material does not cover the whole substrate beam
10 (Song et al., 2009, 2010; Masana and Daqaq, 2011; Alamin et al., 2012;
11 Hobeck and Inman, 2012). In this work, we improve the prediction capability
12 by deriving an analytical electromechanical model of a beam-mass energy
13 harvester with a piezoelectric patch that does not cover the whole substrate
14 beam. The exact mode shapes and natural frequencies using the derived
15 analytical model, approximate analytical model (determined by assuming a
16 fully-covered beam), and finite element analysis results are compared with
17 experimental measurements. Furthermore, linear analysis of the analytical
18 models is performed to investigate the effects of the load resistance on the
19 fundamental global frequency and the harvester's response. The validation
20 of the different methods is performed through comparison with experimental
21 measurements. To design enhanced harvesters that can generate energy at
22 low frequency excitations, a parametric study is conducted to investigate the
23 effects of varying the length of the piezoelectric material on the mode shapes,
24 natural frequencies, and the performance of the harvester.
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41 **2. Representation of the used models and experimental setup**

42 *2.1. Analytical models*

43 *2.1.1. Global electromechanical modeling*

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46 We consider the problem of harvesting energy from a directly excited uni-
47 morph piezoelectric cantilever beam with a tip mass. The cantilever beams
48 consist of aluminum and piezoelectric layers and is subjected to direct ex-
49 citation, as shown in Figure 1. The piezoelectric layers are bounded by
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two-in-plane electrodes of negligible thicknesses connected to an electrical load resistance. These electrodes are assumed to be perfectly conductive and cover the entire piezoelectric surface. We assume that the thickness of the beam is small compared to its length so that the shear deformation and rotary inertia can be neglected. The clamped end of the beam is subjected to a transverse harmonic displacement $Y(t) = Y_0 \cos(\Omega t)$.

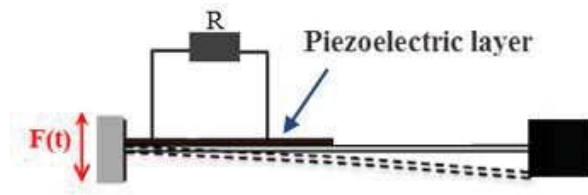


Figure 1: A schematic of the piezoelectric energy harvester under direct excitation

By modeling this bi-layered cantilever beam as an Euler-Bernoulli beam, the partial differential equation governing its relative transverse vibration $v = v(x, t)$ when subjected to direct excitation is written as

$$\frac{\partial^2 M(x, t)}{\partial x^2} + c_a \frac{\partial v(x, t)}{\partial t} + m \frac{\partial^2 v(x, t)}{\partial t^2} = -[m + M_t \delta(x - L)] \frac{\partial^2 Y(t)}{\partial t^2} + M_t L_c \frac{d\delta(x-L)}{dx} \frac{\partial^2 Y(t)}{\partial t^2} \quad (1)$$

where $\delta(x)$ is the Dirac delta function, L is the length of the substrate beam, L_c is half the the length of the tip mass, c_a is the viscous air damping coefficient, m is the mass of the beam per unit length, M_t is the tip mass, and $M(x, t)$ is the internal moment which has three components (Erturk and Inman, 2008). The first of these components is the resistance to bending and

is given by $EI \frac{\partial^2 v(x,t)}{\partial x^2}$. The second component is due to strain rate damping effect and is represented by $c_s I \frac{\partial^3 v(x,t)}{\partial x^2 \partial t}$. The third component is the contribution of the unimorph piezoelectric layer. This contribution is represented by $\vartheta_p (H(x - L_1) - H(x - L_2))V(t)$ where $H(x)$ is the Heaviside step function, $V(t)$ is the generated voltage, L_1 is the distance from the left end of cantilever beam to the starting location of the piezoelectric layer, L_2 is the distance from the left end of cantilever beam to the ending location of the piezoelectric layer and ϑ_p is the piezoelectric coupling term. This term is given by (Erturk and Inman, 2008; Abdelkefi et al., 2011, 2012c)

$$\vartheta_p = -e_{31} b_2 \frac{(y_1 + y_2)}{2} \quad (2)$$

where $e_{31} = E_p d_{31}$ is the piezoelectric stress coefficient, b_2 is the width of the piezoelectric layer and y_1 and y_2 are the positions of the layers with respect to the neutral axis, as shown in Figure 2, $\bar{y} = \frac{(h_p + h_s) E_p h_p}{2(E_p h_p + E_s h_s)} + \frac{h_s}{2}$ and are related as follows:

$$y_0 = -\bar{y}, y_1 = h_s - \bar{y}, y_2 = (h_s + h_p) - \bar{y}$$

where h_s and h_p are the thicknesses of the aluminum and piezoelectric layers, respectively. E_s and E_p are the respective Young's Modulus of these layers.

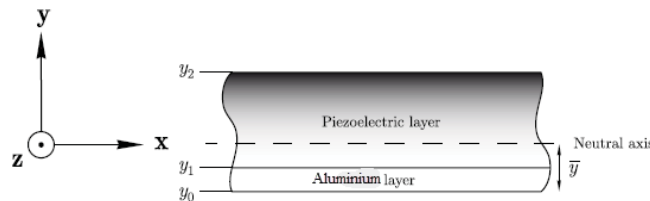


Figure 2: Neutral axis position.

Substituting the moment $M(x, t)$ in equation (1) by its three components, the governing equation of motion of the electromechanical system is written as

$$EI \frac{\partial^4 v(x,t)}{\partial x^4} + c_s I \frac{\partial^5 v(x,t)}{\partial x^4 \partial t} + c_a \frac{\partial v(x,t)}{\partial t} + m \frac{\partial^2 v(x,t)}{\partial t^2} + \left(\frac{d\delta(x-L_1)}{dx} - \frac{d\delta(x-L_2)}{dx} \right) \vartheta_p V(t) \quad (3)$$

$$= -[m + M_t \delta(x-L)] \frac{\partial^2 Y(t)}{\partial t^2} + M_t L_c \frac{d\delta(x-L)}{dx} \frac{\partial^2 Y(t)}{\partial t^2}$$

where the stiffness EI and mass of the beam per unit length m are given by:

$$EI = EI_1 = \frac{1}{12} b_1 E_s h_s^3 \quad \text{and} \quad m = m_1 = b_1 \rho_s h_s \quad \text{for } 0 \leq x < L_1 \quad \text{or} \\ L_2 < x \leq L$$

and

$$EI = EI_2 = \frac{1}{3} E_s b (y_1^3 - y_0^3) + \frac{1}{3} E_p b (y_2^3 - y_1^3) \quad \text{and} \quad m = m_2 = b_1 \rho_s h_s + \\ b_2 \rho_p h_p \quad \text{for } L_1 \leq x \leq L_2$$

where ρ_s and ρ_p are the densities of the aluminum and piezoelectric layers, respectively.

To relate the mechanical and electrical components, we use the Gauss law (IEEE, 1987)

$$\frac{d}{dt} \int_A \mathbf{D} \cdot \mathbf{n} \, dA = \frac{d}{dt} \int_A D_3 \, dA = \frac{V}{R} \quad (4)$$

where \mathbf{D} is the electric displacement vector and \mathbf{n} is the normal vector to the plane of the beam. The electric displacement component D_3 is given by the following relation (Erturk and Inman, 2009):

$$D_3(x, t) = e_{31} \varepsilon_{11}(x, t) + \epsilon_{33}^s E_3 \quad (5)$$

where ε_{11} is the axial strain component in the aluminum and piezoelectric layers and is given by $\varepsilon_{11}(x, y, t) = -y \frac{\partial^2 v(x,t)}{\partial x^2}$, ϵ_{33}^s is the permittivity component at constant strain. Substituting (5) into (4), we obtain the equation governing the strain-voltage relation (Erturk and Inman, 2008):

$$-e_{31} \frac{(y_1 + y_2)}{2} b_2 \int_{L_1}^{L_2} \frac{\partial^3 v(x, t)}{\partial t \partial x^2} dx - \frac{\epsilon_{33}^s b_2 (L_2 - L_1)}{h_p} \frac{dV(t)}{dt} = \frac{V(t)}{R} \quad (6)$$

2.1.2. Eigenvalue problem analysis

To perform the linear analysis, we discretize the system using the Galerkin procedure which requires the exact mode shapes of the structure. These mode shapes are determined by dropping the damping, forcing, and polarization from equation (3), and letting $v(x, t) = \phi(x)e^{i\omega t}$. Because the piezoelectric layer does not cover the whole cantilever beam, we divide the mode shape into three different regions:

$$\phi(x) = \phi_1(x) \text{ for } 0 \leq x < L_1$$

$$\phi(x) = \phi_2(x) \text{ for } L_1 \leq x \leq L_2$$

$$\phi(x) = \phi_3(x) \text{ for } L_2 < x \leq L$$

The resulting eigenvalue problem for each region is given by

$$EI_1 \phi_1^{iv} - m_1 \omega^2 \phi_1 = 0 \quad (7)$$

$$EI_2 \phi_2^{iv} - m_2 \omega^2 \phi_2 = 0 \quad (8)$$

$$EI_1 \phi_3^{iv} - m_1 \omega^2 \phi_3 = 0 \quad (9)$$

with the boundary conditions

$$\phi_1(0) = 0, \phi_1'(0) = 0, \phi_1(L_1) = \phi_2(L_1); \quad (10)$$

$$\phi_1'(L_1) = \phi_2'(L_1), EI_1\phi_1''(L_1) = EI_2\phi_2''(L_1) \quad (11)$$

$$EI_1\phi_1'''(L_1) = EI_2\phi_2'''(L_1), \phi_2(L_2) = \phi_3(L_2), \phi_2'(L_2) = \phi_3'(L_2) \quad (12)$$

$$EI_2\phi_2''(L_2) = EI_1\phi_3''(L_2), EI_2\phi_2'''(L_2) = EI_1\phi_3'''(L_2) \quad (13)$$

$$EI_1\phi_3''(L) - \omega^2 M_t L_c \phi_3(L) - \omega^2 (I_t + M_t L_c^2) \phi_3'(L) = 0 \quad (14)$$

$$EI_1\phi_3'''(L) + \omega^2 M_t \phi_3(L) + \omega^2 M_t L_c \phi_3'(L) = 0 \quad (15)$$

where I_t is the rotary inertia of the tip mass M_t at its center and L_c is half of the length of the tip mass. The mode shapes for the three different regions are then written as

$$\phi_1(x) = A_1 \sin \beta_1 x + B_1 \cos \beta_1 x + C_1 \sinh \beta_1 x + D_1 \cosh \beta_1 x \quad (16)$$

$$\phi_2(x) = A_2 \sin \beta_2 x + B_2 \cos \beta_2 x + C_2 \sinh \beta_2 x + D_2 \cosh \beta_2 x \quad (17)$$

$$\phi_3(x) = A_3 \sin \beta_1 x + B_3 \cos \beta_1 x + C_3 \sinh \beta_1 x + D_3 \cosh \beta_1 x \quad (18)$$

where the coefficients of β_1 and β_2 are related by $\beta_1 = \sqrt[4]{\frac{EI_2 m_1}{EI_1 m_2}} \beta_2$. Normalizing the eigenfunctions using the following orthogonality conditions yields the relation between the different coefficients in (10)-(15):

$$\int_0^{L_1} \phi_{1s}(x)m_1\phi_{1r}(x)dx + \int_{L_1}^{L_2} \phi_{2s}(x)m_2\phi_{2r}(x)dx \quad (19)$$

$$+ \int_{L_2}^{L_3} \phi_{3s}(x)m_1\phi_{3r}(x)dx + \phi_{3s}(L)M_t\phi_{3r}(L) + \phi_{3s}'(L)(I_t + M_tL_c^2)\phi_{3r}'(L)$$

$$+ \phi_{3s}(L)M_tL_c\phi_{3r}'(L) + \phi_{3s}'(L)M_tL_c\phi_{3r}(L) = \delta_{rs}$$

$$\int_0^{L_1} \frac{d^2\phi_{1s}(x)}{dx^2} EI_1 \frac{d^2\phi_{1r}(x)}{dx^2} dx + \int_{L_1}^{L_2} \frac{d^2\phi_{2s}(x)}{dx^2} EI_2 \frac{d^2\phi_{2r}(x)}{dx^2} dx \quad (20)$$

$$+ \int_{L_2}^L \frac{d^2\phi_{3s}(x)}{dx^2} EI_1 \frac{d^2\phi_{3r}(x)}{dx^2} dx = \delta_{rs}\omega_r^2$$

where s and r are used to represent the vibration modes and δ_{rs} is the Kronecker delta, defined as unity when s is equal r and zero otherwise.

To derive a model of the considered energy harvester, we express the displacement $v(x, t)$ using the Galerkin procedure in the form

$$v(x, t) = \sum_{i=1}^{\infty} \phi_{ji}(x)q_i(t) \quad (21)$$

where $j=1,2,3$ depending on the value of x , $q_i(t)$ are the modal coordinates and $\phi_{ji}(x)$ are the mode shapes. Substituting equation (21) into equations (3) and (6) and considering one mode in the Galerkin approach, we obtain the following coupled equations of motions:

$$\ddot{q}(t) + 2\xi\omega\dot{q}(t) + \omega^2q(t) + \chi V(t) = f(t) \quad (22)$$

$$C_p\dot{V}(t) + \frac{V(t)}{R} - \chi\dot{q}(t) = 0 \quad (23)$$

where ξ is the mechanical damping ratio (measured experimentally), ω is the fundamental natural frequency of the structure, the coefficients χ and C_p are the piezoelectric coupling term and the capacitance of the harvester

which are given by $\chi = (\phi'_2(L_2) - \phi'_2(L_1))\vartheta_p$ and $C_p = \frac{\epsilon_{33}^s b_2(L_2 - L_1)}{h_p}$. $f(t)$ is the forcing term of the first mode which is given by: $f(t) = a[m_1 \int_0^{L_1} \phi_1(x) dx + m_2 \int_{L_1}^{L_2} \phi_2(x) dx + m_1 \int_{L_2}^L \phi_3(x) dx + M_t \phi_3(L) + M_t L_c \phi_3'(L)] \cos(\Omega t) = F \cos(\Omega t)$, where $a = Y_0 \Omega^2$ is the base excitation and $F = m_1 \int_0^{L_1} \phi_1(x) dx + m_2 \int_{L_1}^{L_2} \phi_2(x) dx + m_1 \int_{L_2}^L \phi_3(x) dx + M_t \phi_3(L) + M_t L_c \phi_3'(L)$.

To determine closed form expressions for the tip deflection and the generated voltage, we assume that the base excitation can be expressed by $Y = Y_0 e^{i\Omega t}$, where Ω is the excitation frequency, and that $q = Q e^{i\Omega t}$ and $V = V_0 e^{i\Omega t}$. Equations (22) and (23) can then be written in the form

$$\begin{bmatrix} \omega^2 - \Omega^2 + 2i\xi\Omega & \chi \\ -i\chi\Omega & \frac{1}{R} + i\Omega C_p \end{bmatrix} \begin{bmatrix} Q \\ V_0 \end{bmatrix} = \Omega^2 Y_0 \begin{bmatrix} F \\ 0 \end{bmatrix}, \quad (24)$$

The solution is obtained for the tip displacement $v(L, t) = \phi_3(L) \text{Re}(Q e^{i\Omega t})$ and the generated voltage V is determined by taking only the real part of the solution of Equation (24) ($V = \text{Re}(V_0 e^{i\Omega t})$) since we are assuming that $Y(t) = Y_0 \cos(\Omega t)$ which is the real part of the considered excitation. Finally, the electrical harvested power is computed as $P = \frac{V^2}{R}$.

To demonstrate the importance of considering different regions in the determination of the mode shapes, we consider another analytical model that is based on the classical mode shapes of a unimorph piezoelectric cantilever beam (fully covered). In this analysis, we determine the mode shapes of a fully covered piezoelectric cantilever beam and then we use the orthogonality conditions presented in equations (19) and (20) to normalize the equations of motion. We will refer to this model as approximate model.

2.2. Finite element electromechanical modeling

A finite element model was also developed to investigate the performance of the proposed piezoelectric energy harvester. The commercial software ANSYS was used to analyze this model as shown in Figure 3. In this electromechanical model, 3D 20-node structural solid element SOLID186 was applied for the aluminum beam and proof mass. Moreover, 3D 20-node coupled-field solid element SOLID226 was applied for the piezoelectric sheet. The voltage degree of freedom on the top and bottom surfaces were coupled to provide uniform electrical potentials and, thus, to emulate the electrodes of the piezoelectric sheet. First, the electrodes of the piezoelectric sheet were disconnected from the resistor and modal analysis was performed. The electrical potential of the bottom electrode of the piezoelectric sheet was set to zero. This analysis gives the open global frequency. Then, both electrodes were set to zero. This analysis provides the short global frequency. Subsequently, the two electrodes of the piezoelectric sheet were connected to the resistor by coupling the voltage degree of freedom of the electrodes and the two node of the resistor element, as shown in Figure 3. Instead of applying the displacement at the clamped boundary, a corresponding acceleration field due to base excitation was applied to the harvester. Harmonic analysis was then performed to obtain the steady-state power output.

2.3. Experimental setup

A prototype of the piezoelectric energy harvester was devised and tested. The experimental setup is shown in Figure 4. The harvester was composed of an aluminum cantilever beam bonded with a piezoelectric Macro Fiber Composite (MFC) at its root and a proof mass was attached to its free end. The

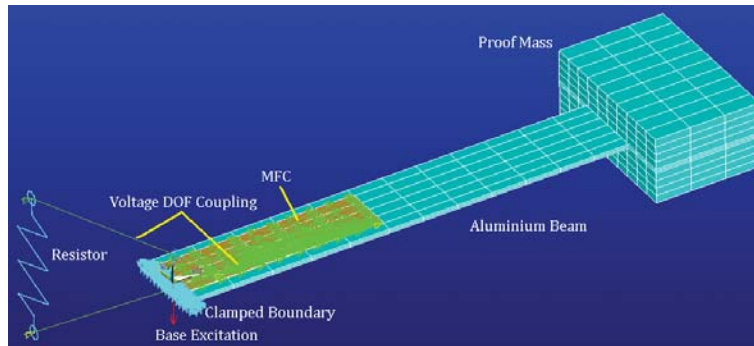


Figure 3: Finite element representation of the piezoelectric energy harvester

center of the proof mass was exactly located at the free end of the cantilever. The parameters of the devised piezoelectric energy harvester are given in Table 1. The sinusoidal drive signal was generated from a function generator and amplified before it was fed to the seismic shaker. An accelerometer was used to monitor the acceleration of the shaker, which was kept constant at $1m/s^2$ during the slow sinusoidal sweep performed by tuning the amplifier. The harvester delivers its power to a resistor R . A current DAQ card was used to log the root mean square value of current, I_{rms} . The average power delivered to the resistor was calculated by:

$$P_{avg} = I_{rms}^2 R \quad (25)$$

Table 1: Physical and geometric properties of the cantilever beam and the tip body

E_s	Aluminum Young's Modulus (GN/m^2)	69.5
E_p	Piezoelectric material (MFC) Young's Modulus (GN/m^2)	30.336
ρ_s	Aluminum density (kg/m^3)	2700
ρ_p	Piezoelectric material density (kg/m^3)	5440
L	Length of the beam (mm)	62.5
L_1	Left of the beam to starting of the piezoelectric layer (mm)	0
L_2	Left of the beam to ending of the piezoelectric layer (mm)	28
b_1	Width of the aluminum layer (mm)	10
b_2	Width of the piezoelectric layer (mm)	7
h_s	Aluminum layer thickness (mm)	0.6
h_p	Piezoelectric layer thickness (mm)	0.2
M_t	Tip mass (g)	4.64
L_{struc}	Length of the tip mass (mm)	15
b_{struc}	Thickness of the tip mass (mm)	9.4
d_{31}	Strain coefficient of piezoelectric layer (pC/N)	-170
ϵ_{33}^s	Permittivity component at constant strain (nF/m)	12.653
ξ	Mechanical damping ratio	0.00878

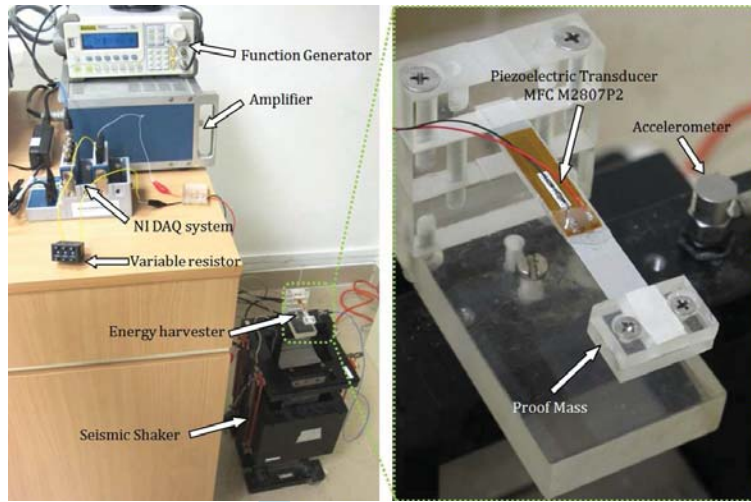


Figure 4: Experimental setup

3. Linear analysis and determination of the optimum load resistance

3.1. Effects of the load resistance on the natural frequency and damping

The effects of the electrical load resistance on the natural frequency and damping of the harvester are determined from a linear analysis of the coupled electromechanical problem. Introducing the following state variables:

$$\mathbf{X} = \begin{bmatrix} X_1 \\ X_2 \\ X_3 \end{bmatrix} = \begin{bmatrix} q \\ \dot{q} \\ V \end{bmatrix}, \quad (26)$$

the equations of motion are rewritten as

$$\dot{X}_1 = X_2 \quad (27)$$

$$\dot{X}_2 = -2\xi\omega X_2 - \omega^2 X_1 - \chi X_3 \quad (28)$$

$$\dot{X}_3 = -\frac{1}{RC_p}X_3 + \frac{\chi}{C_p}X_2 \quad (29)$$

Clearly, these equations have the form

$$\dot{\mathbf{X}} = B\mathbf{X} \quad (30)$$

where

$$B = \begin{bmatrix} 0 & 1 & 0 \\ -\omega^2 & -2\xi\omega & -\chi \\ 0 & \frac{\chi}{C_p} & -\frac{1}{RC_p} \end{bmatrix}.$$

The matrix B has a set of three eigenvalues λ_i , $i = 1, 2, 3$. The first two eigenvalues are similar to those of a pure beam-mass system in the absence of the piezoelectricity effect. The third eigenvalue λ_3 is a result of the electromechanical coupling and is always real and negative. The first two eigenvalues are complex conjugates ($\lambda_2 = \bar{\lambda}_1$). The real part of these eigenvalues represents the electromechanical damping coefficient and the positive imaginary part corresponds to the global frequency of the coupled system.

Inspecting the matrix B , we note that the electrical load resistance has an effect on the overall damping and frequency of the system. Figure 5(a) shows the variation of the global frequency with the electrical resistance. The global frequency is approximately equal to 175.2 rad/s (27.88 Hz) when the load resistance is set equal to $10^2 \Omega$; we refer to this frequency as the short global frequency. Increasing the load resistance results in an increase in the global frequency to a value near 176.4 rad/s (28.07 Hz) when the load resistance is near $R = 10^8 \Omega$; we refer to this frequency as the open global frequency. The significant increase from 175.2 rad/s to 176.4 rad/s takes place when the load resistance is increased from near $10^5 \Omega$ to near

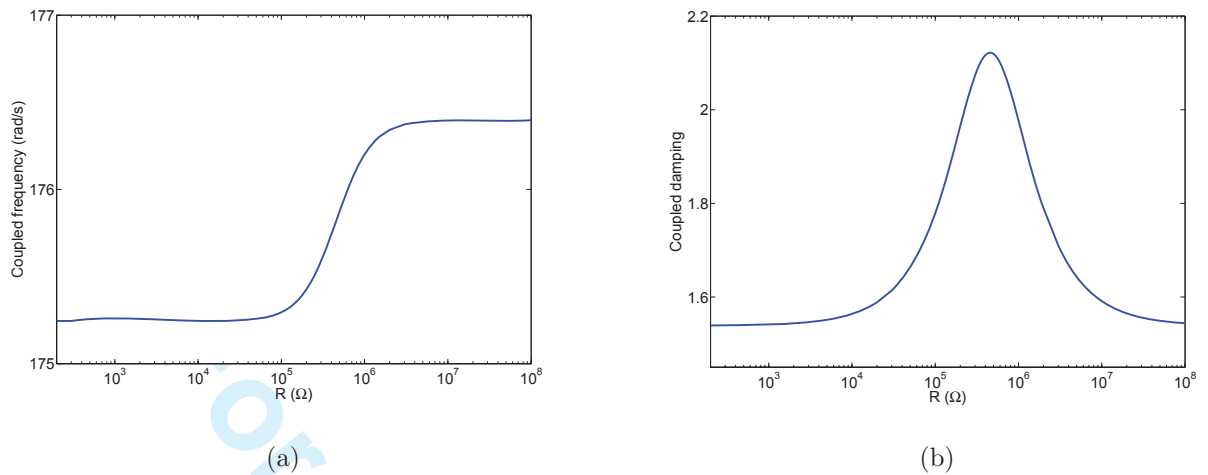


Figure 5: Variations of the (a) global natural frequency and (b) global coupled damping with the electrical load resistance when $L_2=28$ mm (experiment prototype).

$10^6 \Omega$. As for the electromechanical damping, inspecting Figure 5(b), we note that the electromechanical damping is maximum for specific values of the load resistance. The region of load resistance over which the electromechanical damping is relatively high coincides with the region over which a significant increase in the global frequency occurs. Away from this region, the electromechanical damping coefficient is relatively smaller.

3.2. Determination of the optimum load resistance

Based on the derived analytical model, we plot respectively in Figures 6(a), (b), and (c) the frequency-response curves of the displacement, generated voltage, and harvested power when varying the load resistance. The variations of the tip displacement with the load resistance is negligible with minimum values obtained when the load resistance is in the range between $10^5 \Omega$ and $10^6 \Omega$, as shown in Figure 6(a). This is expected because the

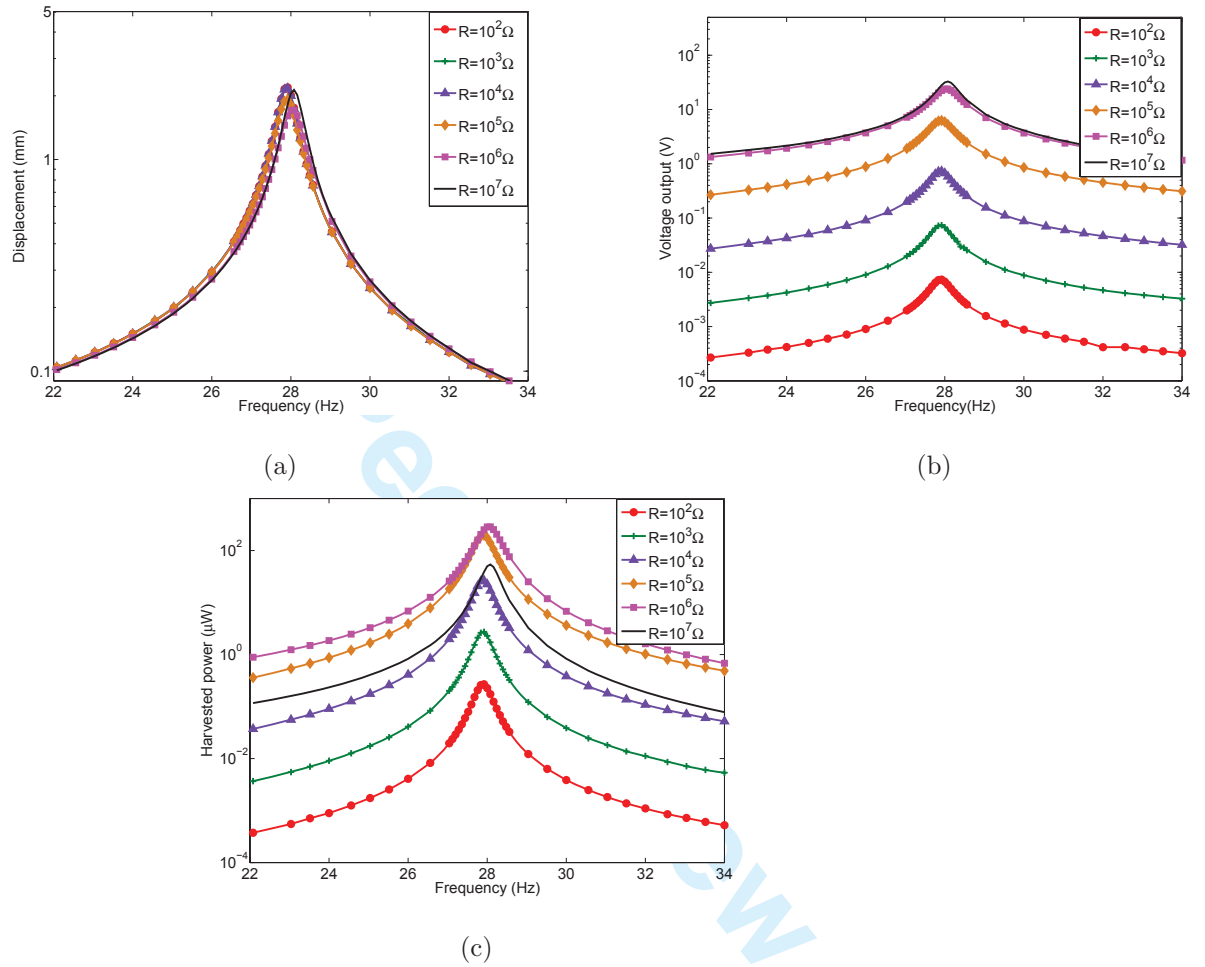


Figure 6: Frequency-response curves of the (a) tip displacement, (b) generated voltage, and (c) harvested power for different values of the electrical load resistance when $L_2=28$ mm and when $a_{rms} = 1 \text{ m/s}^2$.

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8 global electromechanical damping is maximum in this range. Furthermore,
9 inspecting Figure 6(b), we note that increasing the load resistance is accom-
10 panied by an increase in the generated voltage and a slight shift in the global
11 frequency. On the other hand, increasing the load resistance is not accom-
12 panied with an increase in the harvested power, as shown in Figure 6(c).
13 These results are more clear in the plotted curves of Figure 7 which show
14 the tip displacement, generated voltage, and harvested power for the short-
15 and open-circuit configurations. The short- and open-circuit configurations
16 are defined by matching the excitation frequency Ω with the short and open
17 global frequencies, respectively. As mentioned above, minimum values of the
18 tip displacement are obtained when the global damping is maximized, as
19 shown in Figure 7(a). These minimum displacement values are obtained for
20 load resistance values between $10^5 \Omega$ and $10^6 \Omega$ for both short- and open-
21 circuit configurations. In the lower range ($R < 10^4 \Omega$) and higher range
22 ($R > 10^7 \Omega$), the variation of the tip displacement with the load resistance is
23 relatively small. It follows from Figure 7(b) that the generated voltage always
24 increases when the load resistance is increased and then reaches a constant
25 value for both configurations. However, there is an optimum value of the
26 load resistance for which the harvested power is maximized. This optimum
27 value depends on the considered configuration and is larger in the case of the
28 open-circuit configuration. We also note that maximum levels of harvested
29 power are accompanied with minimum levels of the tip displacement which
30 occur when the load resistance is between $10^5 \Omega$ and $10^6 \Omega$. More accurately,
31 maximum levels of harvested power are obtained when the electrical load
32 resistance is almost equal to $4 \times 10^5 \Omega$.
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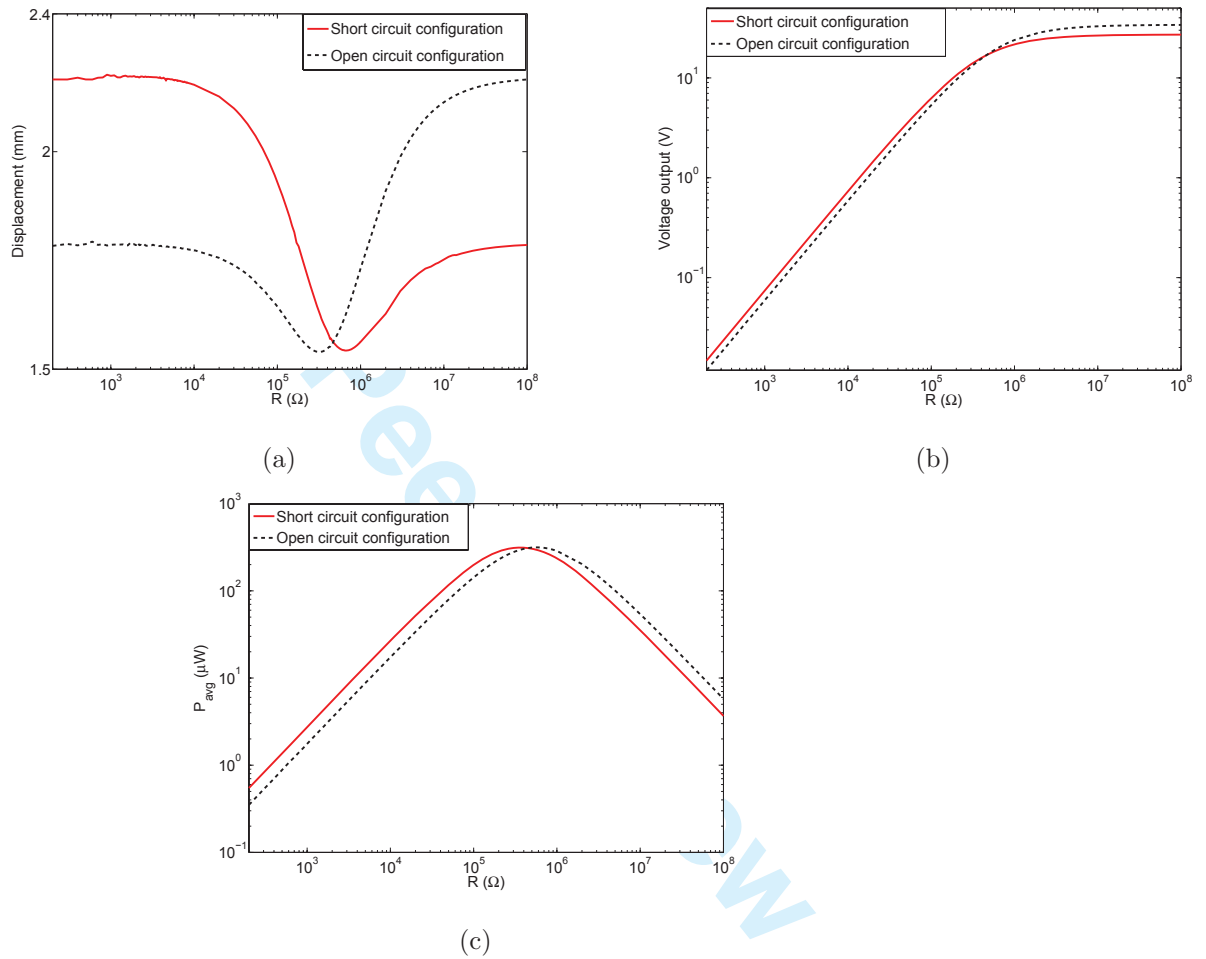


Figure 7: (a) Tip displacement, (b) generated voltage, and (c) harvested power when $L_2=28$ mm and when $a_{rms} = 1m/s^2$ for the short- and open-circuit configurations.

4. Experimental measurements and models validation

A comparison of the short and open global frequencies obtained from the analytical and finite element analysis with the experimental measurements is presented in Table 2. The values show that the short and open global frequencies of the derived analytical model and finite element electromechanical model are in good agreement ($< 3\%$) with the experimental results. On the other hand, there is a discrepancy in the short and open frequencies obtained by using the approximate model and experimental measurements. In fact, the approximated model overestimates the values of these two global frequencies. This overestimation can lead to erroneous results when performing the frequency-response analysis and short and open-circuit configurations or if an approximate model is used to design an energy harvester.

Table 2: Short and open global frequencies: comparisons between different models and experimental measurements.

	Short frequency (Hz)	% difference	Open frequency (Hz)	% difference
Experiment	27.8	–	28	–
Derived model	27.88	0.29	28.07	0.25
Approximated model	28.31	1.8	29	3.6
FEA (ANSYS)	27.79	0.035	27.97	0.11

In Figure 8, we plot the frequency-response curves of the average harvested power as obtained from the different models and the experimental measurements when the load resistance is set equal to the optimal value ($R=4 \times 10^5 \Omega$) and when the root mean square of the base acceleration is set equal to $1 m/s^2$. The plots show that the derived analytical model ac-

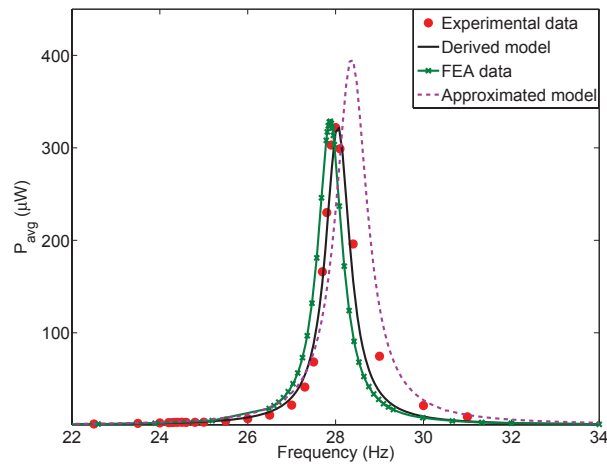


Figure 8: Comparisons of the frequency-response curves between the used models and experimental measurements when $a_{rms} = 1m/s^2$.

curately predicts the experimental measurements. The finite element results are generally in good agreement. There is a small discrepancy in the global frequency of the harvester and slight overestimation of the value of the harvested power at resonance. On the other hand, the approximated analytical model overestimates both the global frequency and the level of harvested power.

Figure 9 shows variations of the average harvested power with the load resistance as predicted from the analytical models and finite element model, and measured experimentally. The analytical derived model and finite electromechanical model accurately predict the response of the harvester for both the short- and open-circuit configurations. On the other hand, the analytical approximate model significantly underestimates or overestimates the power level over a broad range of the resistance values. The above comparisons and validations show that the analytically derived and finite element electrome-

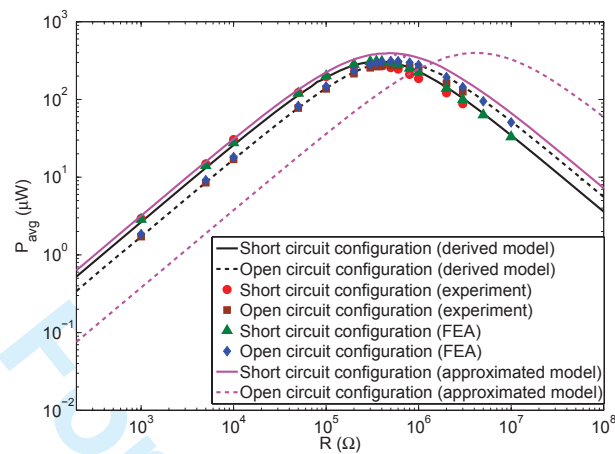


Figure 9: Comparisons of the short- and open-circuit configurations between the used models and experimental measurements when $a_{rms} = 1 \text{ m/s}^2$.

chanical models give more accurate results when compared to the analytical approximate model. Because the use of the analytical derived model is more flexible and computationally inexpensive than the finite element model, we will use it in the following section to design enhanced piezoelectric energy harvesters with smaller volumes that have higher power densities.

5. Piezoelectric material length effects on the behavior of the harvester: Tunability

5.1. Effects of the piezoelectric material length on the natural frequencies and mode shapes

One of the most interesting parameters that can affect the performance of the harvester is the length of the attached piezoelectric material. Changing this length results in a variation in the capacitance of the harvester, the piezoelectric coupling, the natural frequency, the mode shape, and the forcing

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8 term. Consequently, all associated terms in equations (22) and (23) will be
9 changed and new analyses must be performed. We start by determining
10 the effects of varying the length of the piezoelectric material on the natural
11 frequency and associated mode shape. This investigation is performed for
12 two different system parameters. The first one has the same parameters as
13 those of the experimental prototype except for the length of the piezoelectric
14 sheet (L_2) which is varied systematically. We refer to this configuration as
15 the first energy harvesting system. In the second configuration, we change
16 the length of the aluminum beam to 70 mm, the thickness and width of the
17 piezoelectric sheet to 0.356 mm and 1 cm, respectively, and the tip mass to
18 4.52 g. We refer to this configuration as the second energy harvesting system.

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28 Figure 10(a) shows variations of the natural frequency of these two systems
29 with the length of the piezoelectric sheet. The plots show that decreasing
30 the length of the piezoelectric material results in a decrease in the
31 value of the natural frequency for both systems. This is beneficial in terms of
32 managing low frequency excitations for piezoelectric energy harvesters and
33 enhancing their power densities. Furthermore, there is an optimum value
34 of the piezoelectric length beyond which the natural frequency could not be
35 increased significantly. This value is near 53 mm for the first system and 60
36 mm for the second system. However, for small length values of the piezoelectric
37 sheet, the rate of variation of the natural frequency with L_2 is important.
38 Based on this analysis and depending on the available excitation frequency,
39 the harvester can be passively tuned to match its natural frequency to the
40 available excitation frequency. Figures 10(b) and 10(c) show variations of the
41 first mode shape with the length of the piezoelectric sheet for both systems,
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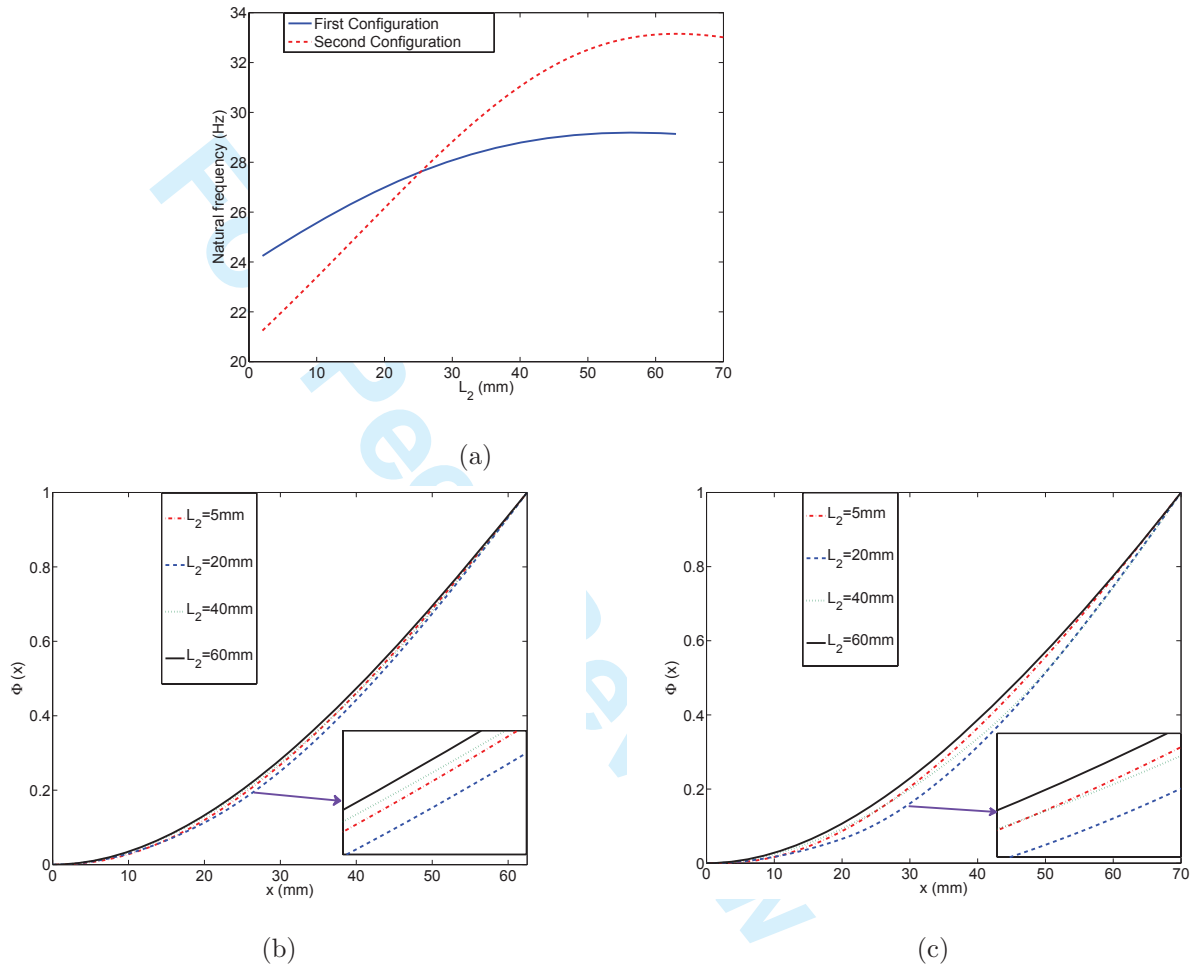


Figure 10: Variations of the (a) structural natural frequency and (b,c) mode shapes with the length of the piezoelectric material for the first and second systems.

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8 respectively. The plots show that there is a significant change in the mode
9 shape when varying the length of the piezoelectric sheet. The difference
10 between the mode shapes varies depending on this length.
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13 14 5.2. Effects of the piezoelectric material length on the performance of the 15 harvester 16 17

18 The plots in Figures 11 and 12 show the frequency-response curves of
19 the harvested power when varying the length of the piezoelectric material
20 for different values of the electrical load resistance. The plots show that
21 the level of the harvested power is significantly affected by the length of the
22 piezoelectric sheet as well as the load resistance. The frequency-response
23 curves of the first system when the length of the piezoelectric sheet is set
24 equal to 50 mm and 60 mm are very close to each other. In addition, for
25 the same system and when $R = 10^5 \Omega$ and $R = 10^6 \Omega$, the maximum
26 (resonant) values of the harvested power are very close when the length
27 of the piezoelectric sheet is set equal to 30 mm, 40 mm, 50 mm, and 60
28 mm. In the second system, the same behavior is observed. Furthermore, for
29 both systems, there is a compromise between the length of the piezoelectric
30 material, the electrical load resistance, and the available excitation frequency
31 that leads to maximum levels of harvested energy.
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44 To investigate more this compromise, we plot in Figures 13 and 14 vari-
45 ations of the resonant average harvested power with the load resistance and
46 length of the piezoelectric material and for both systems, respectively. It
47 follows from Figures 13(a) and 14(a) that there is an optimum value of the
48 load resistance for which the resonant value of the harvested power is maxi-
49 mized for all considered piezoelectric lengths. We note also that this region is
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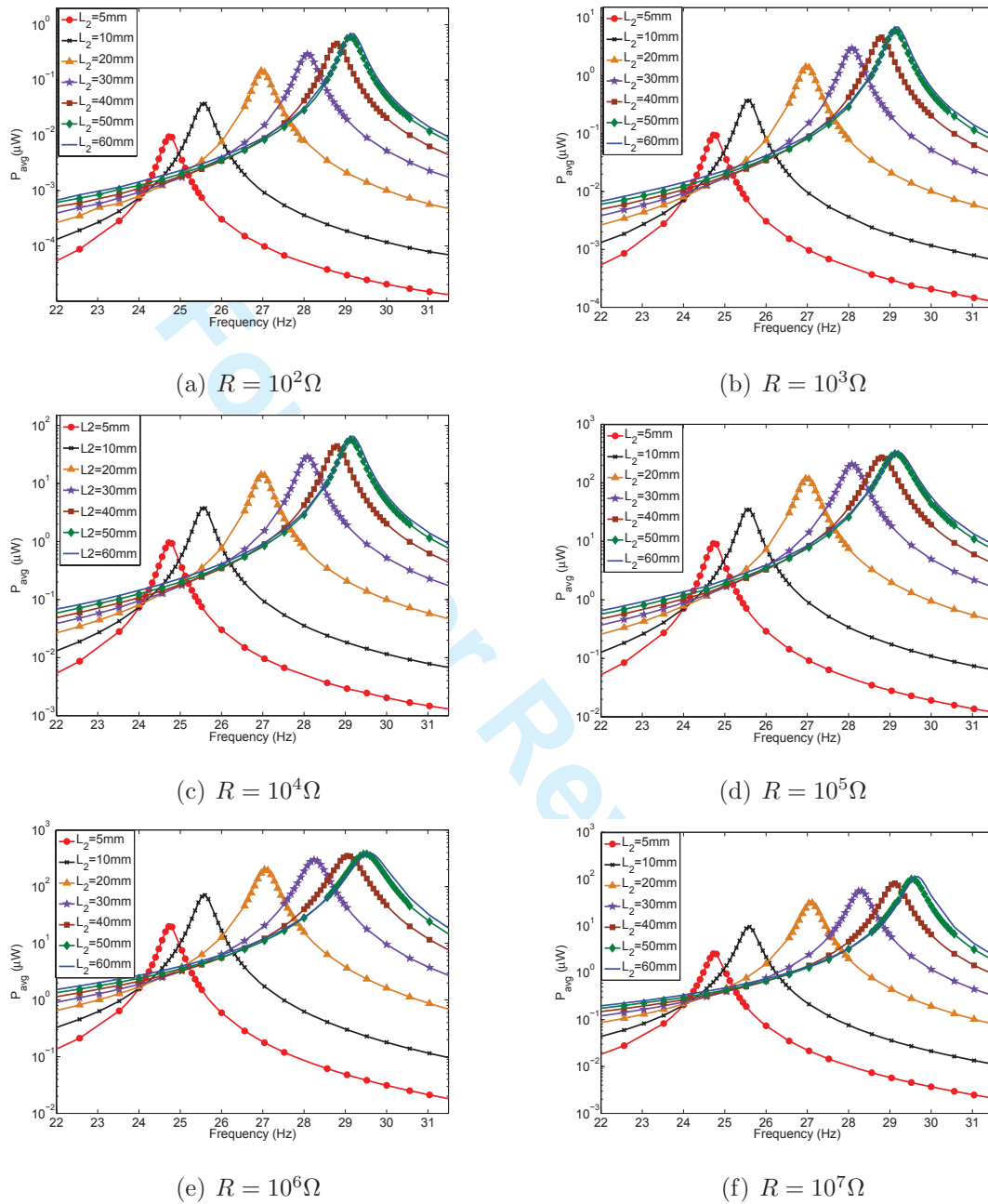


Figure 11: Variations of the average harvested power with the length of the piezoelectric material for different load resistances and for the first system when $a_{rms} = 1 \text{ m/s}^2$.

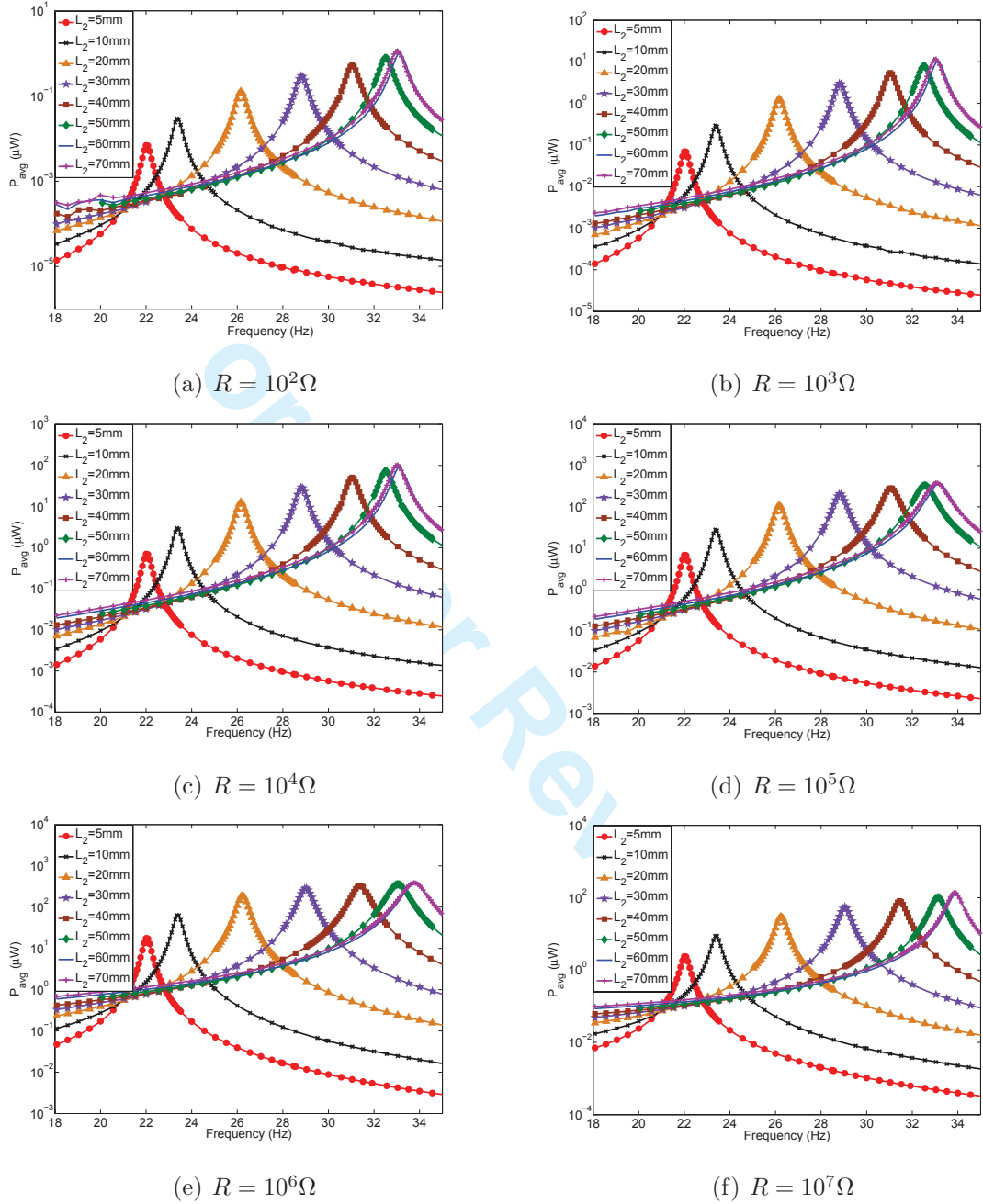


Figure 12: Variations of the harvested power with the length of the piezoelectric material for different load resistances and for the second system when $a_{rms} = 1 \text{ m/s}^2$.

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8 almost the same for both systems. In the first system, increasing the length
9 of the piezoelectric material is accompanied with an increase in the value of
10 the resonant harvested power. The rate of increase of the harvested power is
11 important when the length of the piezoelectric sheet is increased from 5 mm
12 to 30 mm. Beyond 30 mm, this rate becomes very small, as shown in Figure
13 13(b). In the second system, increasing the length of the piezoelectric sheet
14 is followed by an increase in the average harvested power and then stabilizes
15 at higher piezoelectric length values. In addition, the rate of increase of the
16 harvested power is significantly affected by the length of the piezoelectric
17 sheet when its length is between 5 mm and 20 mm. At higher values, the
18 variation rate of the harvested power becomes very small. Consequently, we
19 can conclude that depending on the excitation frequency, an enhanced har-
20 vester can be designed by changing the length of the piezoelectric material
21 and the electrical load resistance.
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34 35 *5.3. Effects of the length of the piezoelectric sheet on the short- and open-* 36 *circuit configurations* 37 38

39 As mentioned above, the length of the piezoelectric sheets affects the
40 natural frequency, the mode shape, the capacitance of the harvester, and
41 the piezoelectric coupling. Because the values of the short and open global
42 frequencies depend on the harvester's parameters, we present in Table 3
43 the values of these frequencies for both systems and for different lengths of
44 the piezoelectric sheet. We note that the difference between the open and
45 short frequencies is negligible for small values of the piezoelectric material
46 length. Furthermore, increasing the length of the piezoelectric material is
47 accompanied with an increase in the difference between these frequencies.
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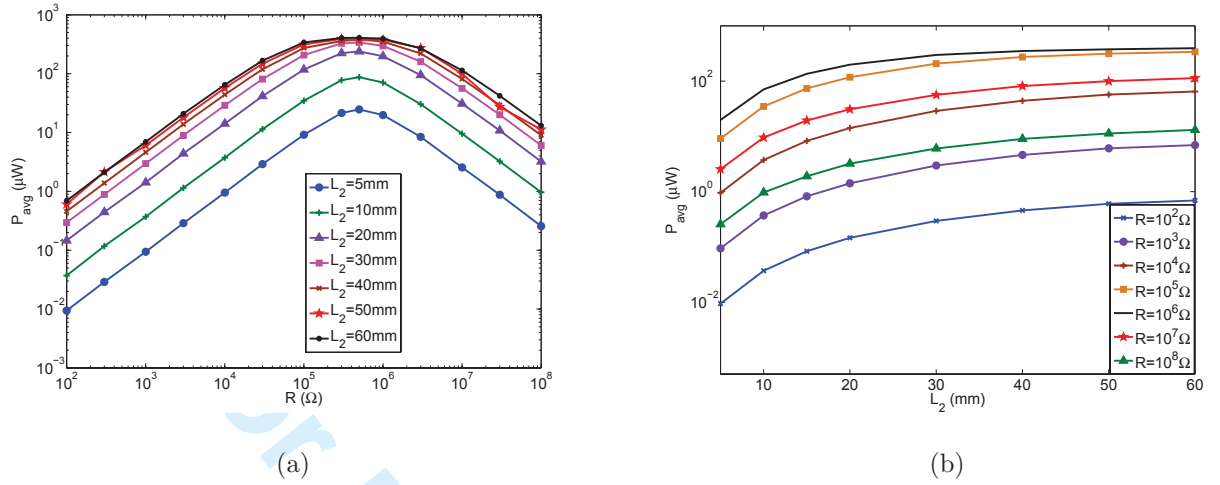


Figure 13: Variations of the resonant average harvested power with (a) the load resistance and for different piezoelectric lengths and (b) the length of the piezoelectric material for different load resistances when $a_{rms} = 1 \text{ m/s}^2$ and for the first system.

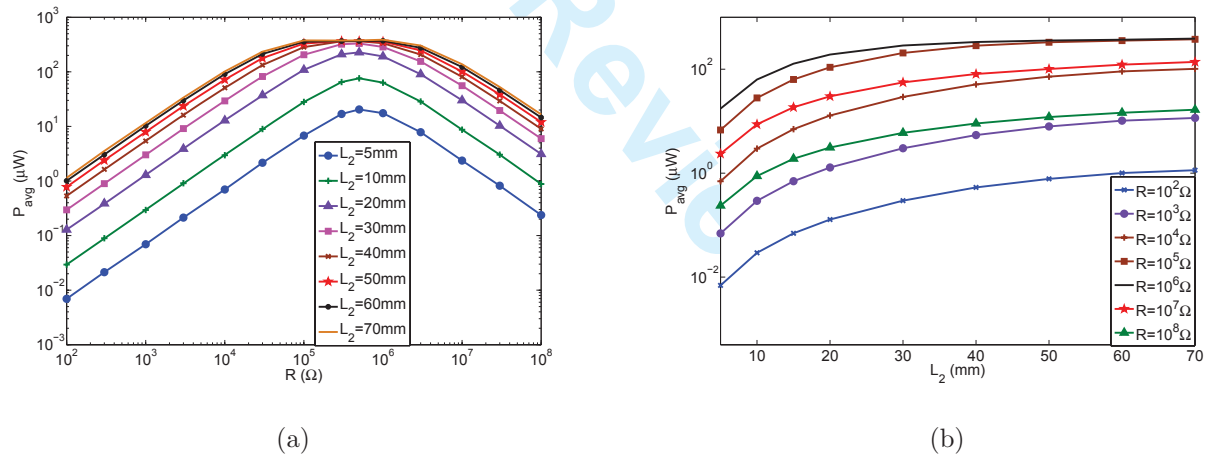


Figure 14: Variations of the resonant average harvested power with (a) the load resistance and for different piezoelectric lengths and (b) the length of the piezoelectric material for different load resistances when $a_{rms} = 1 \text{ m/s}^2$ and for the second system.

Table 3: Short and open global frequencies for different lengths of the piezoelectric sheet and for both systems.

$L_2(mm)$	$w_s(Hz)$ (first)	$w_o(Hz)$ (first)	$w_s(Hz)$ (second)	$w_o(Hz)$ (second)
5	24.75	24.756	22.033	22.037
10	25.561	25.586	23.385	23.402
20	26.989	27.086	26.161	26.247
30	28.079	28.286	28.822	29.042
40	28.785	29.112	31.039	31.453
50	29.124	29.552	32.509	33.129
60	29.165	29.656	33.12	33.897
70	–	–	33.011	33.863

We plot in Figures 15(a) and (b) the average harvested power using short- and open-circuit configurations of both the first and second systems. We note that the short- and open-circuit configurations when $L_2 = 10 \text{ mm}$ are the same. This is expected because the short and open frequencies are almost the same, as shown in Table 3. In addition, increasing the length of the piezoelectric sheet is accompanied with a significant distinction between the short- and open-circuit configurations. For $L_2 = 40 \text{ mm}$ and $L_2 = 60 \text{ mm}$, we note that, depending on the region of the considered load resistance and the short- or open-circuit configuration, the average harvested power can be higher when $L_2 = 40 \text{ mm}$ or higher when $L_2 = 60 \text{ mm}$. For example, when the load resistance is set equal to $10^3 \Omega$ and for the short-circuit configuration, the average value of the harvested power is higher when $L_2 = 60 \text{ mm}$ than $L_2 = 40 \text{ mm}$. On the other hand, when changing the value of the load resistance to $10^7 \Omega$, the average value of the harvested power is higher when

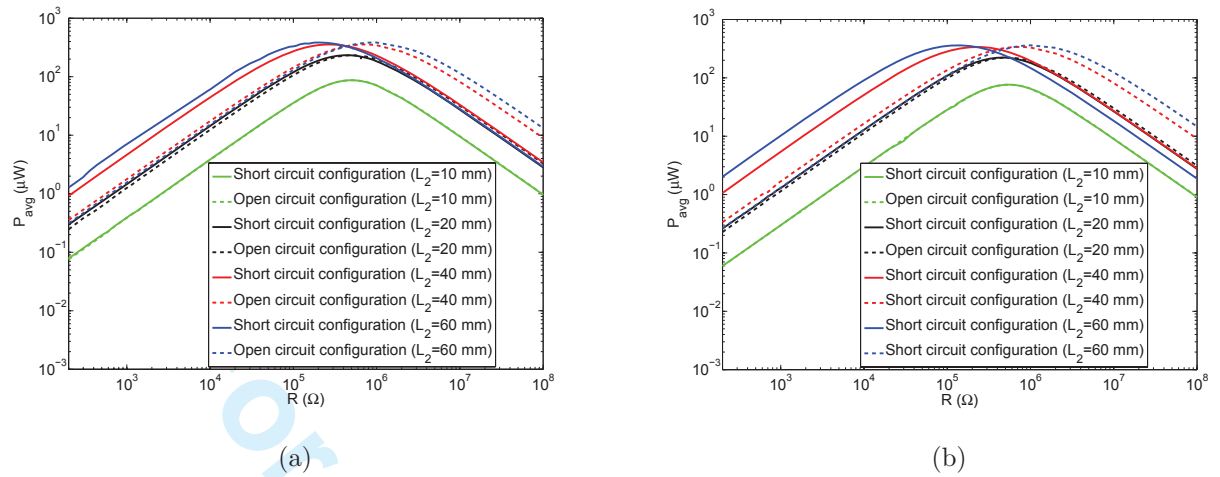


Figure 15: Variations in the average harvested power using short- and open-circuit configurations for different length of the piezoelectric material and for (a) the first system and (b) the second system and when $a_{rms} = 1 \text{ m/s}^2$.

$L_2 = 40 \text{ mm}$ than $L_2 = 60 \text{ mm}$. We conclude that depending on the available excitation frequency, there is a compromise between the load resistance and the length of the piezoelectric sheet to get enhanced levels of harvested power.

6. Conclusions

We have developed analytical and finite element electromechanical models that take into account the fact that the piezoelectric sheet does not cover the whole substrate beam of a beam-mass energy harvester. In addition, we used the approximate electromechanical model that is based on the classical mode shapes (fully-covered beam) in the Galerkin discretization. A linear analysis of the derived analytical model was performed to determine the optimal load resistance. By comparing results from the derived analytical

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8 models and finite element analysis results with experimental measurements,
9 we determined that a model which uses mode shapes that are based on the
10 length of the piezoelectric sheet are better suited than approximate models
11 that are based on a fully covered beam to predict the performance of the
12 harvester. The results showed that the finite element results gives a slight
13 discrepancy in the global frequency of the harvester and harvested power at
14 resonance. In orde to design enhanced piezoelectric energy harvesters that
15 can generate energy at low frequency excitations, a parametric study based
16 on the analytical derived model was then performed to investigate the effects
17 of the length of the piezoelectric sheet on the natural frequency and level
18 of harvested power. The results showed that depending on the available
19 low excitation frequency an enhanced piezoelectric energy harvester can be
20 tuned and optimized by changing the length of the piezoelectric sheet and
21 load resistance.
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