# MOTION PLANNING AND FEEDBACK CONTROL FOR A UNICYCLE IN A WAY POINT FOLLOWING TASK: THE VFO APPROACH 

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#### Abstract

This paper is devoted to the way point following motion task of a unicycle where the motion planning and the closed-loop motion realization stage are considered. The way point following task is determined by the user-defined sequence of waypoints which have to be passed by the unicycle with the assumed finite precision. This sequence will take the vehicle from the initial state to the target state in finite time. The motion planning strategy proposed in the paper does not involve any interpolation of way-points leading to simplified task description and its subsequent realization. The motion planning as well as the motion realization stage are based on the Vector-Field-Orientation (VFO) approach applied here to a new task. The unique features of the resultant VFO control system, namely, predictable vehicle transients, fast error convergence, vehicle directing effect together with very simple controller parametric synthesis, may prove to be useful in practically motivated motion tasks.


Keywords: unicycle, way point following, motion planning, feedback control, vector fields.

## 1. Introduction

In the robotics literature, one usually distinguishes three basic and uniquely defined control tasks (de Luca et al., 1998): trajectory tracking, path following, and posture stabilization (set-point regulation). However, in practice, mobile robot control tasks cannot be easily and definitely classified into one of these types (Lawrence et al., 2008). For instance, one can find practical motion problems like tracking a leading vehicle where its instantaneous motion cannot be anticipated, or the task of motion with the target not defined in advance but determined by defining the desired azimuthal direction and longitudinal robot velocity, or finally the task of motion along the geometrical contour not known in advance, like the wall-following problem (Siegwart and Nourbakhsh, 2004). Moreover, sometimes even if a particular task belongs to one of the first two types mentioned above, its full description in terms of the reference full-state trajectory or geometrical path can be a nontrivial algorithmic problem, especially in the case of motion planning in a cluttered environment (Madi, 2004). Thus it seems to be desirable to propose an alternative method of motion task determination, which would combine two useful features, namely, the simplicity of task description characteristic for the set-point regulation prob-
lem together with the ability of shaping the robot's path which is intrinsic to the path following motion problem. One of the simplest methods consists in the determination of the set $\mathcal{S}_{t}$, which is made of the way-point sequence along with the initial (starting) state and the target (final) state. The set $\mathcal{S}_{t}$ can be treated as a simplified definition of the desired path assuming that the way-points are chosen sufficiently close to each other and the vehicle motion between them is predictable and sufficiently smooth. Simplified task determination might turn out to be computationally efficient and useful for simple motion re-planning by adding way-points to the set $\mathcal{S}_{t}$ even already during task realization.

The planning problem for motion described by the set of way-points can be solved in many ways. The shortest path with finite curvature connecting the way-points is made of the sequence of arcs and rectilinear segments (Reeds and Shepp, 1990). Planning the motion involves finding and combining the finite sequence of the mentioned primitives leading to the resultant curve with a discontinuous curvature. The smoothing procedure proposed in (Scheuer and Fraichard, 1997) and (Fleury et al., 1995), where clothoid segments were used in the neighborhoods of discontinuity points, can improve the quality of planned motion. However, the above approaches finally lead to
geometrical paths determined using an interpolation procedure between the way-points. As a consequence, the control task becomes the path following problem and the feature of simplified task definition may be lost. Moreover, as mentioned in (Samson, 1992), path following task realization usually involves the determination of the instantaneous perpendicular distance to the path, which is generally a non-trivial or even a non-unique mathematical problem. Thus, the issue of simplified task determination, which does not involve extending it to the full path, together with simplified motion realization seems to be an open research problem.

This paper proposes an alternative algorithm of motion planning in a free space using only the sequence of defined and recomputed way-points. The method does not involve an extension to any geometrical path preserving the simplicity of the task in a sense of its description as well as realization. The finite-time motion control problem (with the motion time-horizon being a function of controller parameters) is solved by a modified version of the original Vector-Field-Orientation (VFO) feedback controller presented by the authors in (Michałek and Kozłowski, 2009), but designed here with a strict connection to the newly proposed planning method. The motion task proposed and subsequently accomplished by the strategy introduced in this paper will be called way point following $\sqrt{1}$. The name emphasizes the combination of features characteristic for set-point regulation and partially for the path following problem, leading, however, to simplified version of the latter.

The paper is an extension of the preliminary work (Michałek and Kozłowski, 2008) and is organized as follows: Section 2 introduces basic assumptions, includes the system model and the task definition considered. A brief explanation of the original VFO control approach for posture stabilization is presented in Section 3 The motion planning algorithm and the VFO motion control strategy are the main topic of Sections 4 and 5, respectively. Section6illustrates simulation results. Conclusions are given in Section 7

## 2. Problem formulation

2.1. Unicycle model. The vehicle model taken into account in the paper is a unicycle with the following kinematics:

$$
\dot{\boldsymbol{q}}(\tau)=\left[\begin{array}{l}
1  \tag{1}\\
0 \\
0
\end{array}\right] u_{1}(\tau)+\left[\begin{array}{c}
0 \\
\cos \theta(\tau) \\
\sin \theta(\tau)
\end{array}\right] u_{2}(\tau)
$$

where $\boldsymbol{q} \triangleq\left[\begin{array}{lll}\theta & x & y\end{array}\right]^{T}=\left[\begin{array}{ll}\theta & \boldsymbol{q}^{* T}\end{array}\right]^{T} \in \mathbb{R}^{3}$ is a state vector describing the orientation angle and the position vector of a local frame attached to the unicycle (see Fig. (1). The

[^0]

Fig. 1. Unicycle in the global frame $\{G\}$.
control inputs $u_{1}, u_{2} \in \mathbb{R}$ can be interpreted as angular and longitudinal velocity, respectively.
2.2. Way point following motion task. Here we provide the mathematical background for the way point following motion task. Let us introduce the following set of finite number of way-points:

$$
\begin{equation*}
\mathcal{S}_{t} \triangleq\left\{\boldsymbol{q}_{t 0}, \boldsymbol{q}_{t 1}, \boldsymbol{q}_{t 2}, \ldots, \boldsymbol{q}_{t N}\right\} \tag{2}
\end{equation*}
$$

which have to be passed by the unicycle (with assumed precision) during the motion task considered, with $\boldsymbol{q}_{t 0}$ being an initial (starting) point, and $\boldsymbol{q}_{t N}$ being a target (final) point, where

$$
\boldsymbol{q}_{t i} \triangleq\left[\begin{array}{c}
\theta_{t i}  \tag{3}\\
\boldsymbol{q}_{t i}^{*}
\end{array}\right] \in \mathbb{R}^{3}, \quad \boldsymbol{q}_{t i}^{*}=\left[\begin{array}{c}
x_{t i} \\
y_{t i}
\end{array}\right] \in \mathbb{R}^{2} .
$$

We assume that

$$
\begin{equation*}
\boldsymbol{q}_{t 0} \triangleq \boldsymbol{q}(\tau=0) \tag{4}
\end{equation*}
$$

being an initial state of the system (11), and $\boldsymbol{q}_{t N}=$ $\left[\begin{array}{lll}\theta_{t N} & x_{t N} & y_{t N}\end{array}\right]^{T}$ is explicitly defined by the user. The determination of the remaining way-points $\boldsymbol{q}_{t i}, i=$ $1, \ldots, N-1$ from (2) will be described in the sequel as part of the motion planning stage. In this subsection we assume that they are given.

Next, introduce a term of the $i$-th motion segment denoting the motion stage associated to the transition from the $\boldsymbol{q}_{t i-1}$ to the $\boldsymbol{q}_{t i}$ way-point of (2). The two way-points $\boldsymbol{q}_{t i-1}$ and $\boldsymbol{q}_{t i}$ will be called the boundaries of the $i$-th motion segment. Now one can formulate the way point following motion task for the unicycle.

Definition 1. (Way point following task) Find the bounded control input functions $u_{1}(\cdot)$ and $u_{2}(\cdot)$ which take the state $q$ of the model (1) from the initial point (4) to the desired target point, $\boldsymbol{q}_{t N}$, passing according to index order through all the way-points $\boldsymbol{q}_{t i}$ for $i=1, \ldots, N-1$ with the assumed precision in sense that

$$
\begin{equation*}
\lim _{\tau \rightarrow \infty}\left\|\boldsymbol{q}_{t N}-\boldsymbol{q}(\tau)\right\| \leqslant \epsilon_{N} \tag{5}
\end{equation*}
$$

and

$$
\begin{equation*}
\forall i=1, \ldots, N \quad \exists \tau_{i}<\infty: \quad\left\|e_{i}^{*}\left(\tau_{i}\right)\right\| \leqslant \epsilon_{i} \tag{6}
\end{equation*}
$$

where

$$
\begin{equation*}
e_{i}^{*}(\tau) \triangleq \boldsymbol{q}_{t i}^{*}-\boldsymbol{q}^{*}(\tau) \tag{7}
\end{equation*}
$$

is a position error in the $i$-th motion segment, and $\tau_{i}$ denotes the time instant when the norm $\left\|e_{i}^{*}(\tau)\right\|$ enters the assumed vicinity $\epsilon_{i} \geqslant 0$.

Note that (5) together with (6) means finite-time convergence for the target position error $e_{N}^{*}(\tau)$ to the $\epsilon_{N^{-}}$ neighborhood and asymptotic convergence of the target orientation error $\left|\theta_{t N}-\theta(\tau)\right|$ to zero.

The above motion task can be accomplished in several ways. The proposition described below makes use of the specific geometrical features of the VFO stabilizer (see (Michałek and Kozłowski, 2009)), especially the so-called directing effect. The concept, however, involves conducting a simple motion planning stage before, which will subsequently allow smooth transition of the vehicle via motion segment boundaries. Moreover, it will be shown that the method allows free shaping of the longitudinal velocity profile $u_{2}=u_{2}(s)$, as a function of some parameter $s$, in the forward as well as in the backward motion strategy. Summarizing, the presented control concept consists of two main stages: (i) the motion planning stage relying on the determination of the way-points $\boldsymbol{q}_{t i}$ for $i=1$ to $i=N-1$, and (ii) the motion control stage accomplishing the way point following task formulated in Definition 1 . Both stages will be described in Sections 4 and 5, respectively. To make the concept clear enough, a brief recall concerning the VFO control approach is given first in the next section (for a detailed description, see (Michałek and Kozłowski, 2009)).

## 3. Background on the VFO stabilizer

The form of the VFO stabilizer results from the vector field orientation control method, which originates from a simple geometrical interpretation connected with the kinematics (11). In this interpretation, the input $u_{1}$ is treated as orienting control, which allows one to freely change the orientation of the vector field $\boldsymbol{g}_{2}^{*}(\theta)=[\cos \theta \sin \theta]^{T}$ driving directly the $\theta$ variable (orienting variable). The second input $u_{2}$ plays the role of pushing control, which drives (pushes) the rest of the state variables along the current direction of $\boldsymbol{g}_{2}^{*}(\theta)$. The VFO stabilizer is defined by the following equations:

$$
\begin{align*}
& u_{1}(\tau) \triangleq h_{1}(\tau)  \tag{8}\\
& u_{2}(\tau) \triangleq\left\|\boldsymbol{h}^{*}(\tau)\right\| \cos \alpha(\tau) \tag{9}
\end{align*}
$$

where $\alpha=\angle\left(\boldsymbol{g}_{2}^{*}(\theta), \boldsymbol{h}^{*}\right)$, and the so-called convergence vector field $\boldsymbol{h}=\left[\begin{array}{ll}h_{1} & \boldsymbol{h}^{* T}\end{array}\right]^{T}=\left[\begin{array}{lll}h_{1} & h_{2} & h_{3}\end{array}\right]^{T}$. Here $\boldsymbol{h}=$ $\boldsymbol{h}\left(\boldsymbol{q}_{t}, \boldsymbol{q}, \cdot\right) \in \mathbb{R}^{3}$ defines at every state point $\boldsymbol{q}$ the desired convergence direction and is also a function of an instantaneous distance to the reference point $\boldsymbol{q}_{t}=\left[\begin{array}{lll}\theta_{t} & x_{t} & y_{t}\end{array}\right]^{T}$. The particular form of this vector field is a second crucial
element of the whole VFO control strategy and in the case of the posture stabilization task it is defined as follows:

$$
\begin{align*}
& h_{1}(\tau) \triangleq k_{1} e_{a}(\tau)+\dot{\theta}_{a}(\tau)  \tag{10}\\
& \boldsymbol{h}^{*}(\tau) \triangleq k_{p} \boldsymbol{e}^{*}(\tau)+\boldsymbol{v}^{*}(\tau) \tag{11}
\end{align*}
$$

where

$$
\begin{align*}
& e_{a}(\tau) \triangleq \theta_{a}(\tau)-\theta(\tau)  \tag{12}\\
& \theta_{a}(\tau) \triangleq \operatorname{Atan2c}\left(\operatorname{sgn}(k) h_{3}(\tau), \operatorname{sgn}(k) h_{2}(\tau)\right)  \tag{13}\\
& \dot{\theta}_{a}(\tau)=\frac{\dot{h}_{3}(\tau) h_{2}(\tau)-\dot{h}_{2}(\tau) h_{3}(\tau)}{h_{2}^{2}(\tau)+h_{3}^{2}(\tau)}  \tag{14}\\
& \boldsymbol{e}^{*}(\tau) \triangleq \boldsymbol{q}_{t}^{*}-\boldsymbol{q}^{*}(\tau), \quad \boldsymbol{q}_{t}^{*}=\left[x_{t} y_{t}\right]^{T}  \tag{15}\\
& \boldsymbol{v}^{*}(\tau) \triangleq-\eta \operatorname{sgn}(k)\left\|\boldsymbol{e}^{*}(\tau)\right\| \boldsymbol{g}_{2 t}^{*}  \tag{16}\\
& \dot{\boldsymbol{h}}^{*}(\tau)=-k_{p} \dot{\boldsymbol{q}}^{*}(\tau)+\dot{\boldsymbol{v}}^{*}(\tau)  \tag{17}\\
& \dot{\boldsymbol{v}}^{*}(\tau)=-\eta \operatorname{sgn}(k) \frac{\boldsymbol{e}^{* T}(\tau) \dot{\boldsymbol{e}}^{*}(\tau)}{\left\|\boldsymbol{e}^{*}(\tau)\right\|} \boldsymbol{g}_{2 t}^{*} \tag{18}
\end{align*}
$$

$k_{1}, k_{p}>0$ and $0<\eta<k_{p}$ are the VFO design parameters, $\boldsymbol{g}_{2 t}^{*}=\left[\begin{array}{cc}\cos \theta_{t} & \sin \theta_{t}\end{array}\right]^{T}$, and $\operatorname{Atan} 2 \mathrm{c}(\cdot, \cdot): \mathbb{R} \times \mathbb{R} \mapsto$ $\mathbb{R}$ is a continuous version of the four-quadrant function $\operatorname{Atan} 2(\cdot, \cdot): \mathbb{R} \times \mathbb{R} \mapsto[-\pi, \pi)$.

Equations (8) to (18) reveal the VFO control strategy. Equation (13) defines the desired auxiliary orientation angle, expected to be followed by the vehicle, and computed according to the current direction of $\boldsymbol{h}^{*}$. The additional term $\operatorname{sgn}(k) \in\{+1,-1\}$ can be treated here as a decision variable, which allows choosing the desired motion strategy (forward/backward) of the vehicle along the direction of $\boldsymbol{h}^{*}$. According to (12), (10) and (8), the orienting control $u_{1}$ is responsible for reorienting the vehicle to make the auxiliary orientation error (12) tend to zero. A geometrical interpretation of the above follows: setting $e_{a}=0$ is equivalent to making the direction (and the orientation if $\operatorname{sgn}(k)=+1)$ of $\boldsymbol{g}_{2}^{*}(\theta)$ coincident with the direction determined by $\boldsymbol{h}^{*}$. In addition, with appropriate value selection for $\operatorname{sgn}(k)$ and the input $u_{2}$, it consequently implies that the longitudinal velocity vector $\dot{\boldsymbol{q}}^{*}$ can be aligned with the convergence vector $\boldsymbol{h}^{*}$ and the vehicle position can be driven to the reference point.

It is worth noting that the definition of $\boldsymbol{h}^{*}$ proposed in (11) is peculiar in the sense that at the limit for $\boldsymbol{e}^{*} \rightarrow \mathbf{0}$ the auxiliary variable (13) converges to the reference orientation $\theta_{t}$. This has a great importance for the convergence of the vehicle orientation to the reference one in a neighborhood of the reference position $\boldsymbol{q}_{t}^{*}$. The pushing control $u_{2}$ proposed in (9) drives the substate vector $\boldsymbol{q}^{*}$ along the current direction of $\boldsymbol{g}_{2}^{*}(\theta)$ with the intensity proportional to the current orthogonal projection of $\boldsymbol{h}^{*}$ onto $\boldsymbol{g}_{2}^{*}(\theta)$ realizing the so-called careful pushing strategy (Michałek and Kozłowski, 2009).

Summarizing, the control inputs of the VFO approach are designed in a way which guarantees that the


Fig. 2. Effect of directing the vehicle during set-point control with the VFO stabilizer ( $\boldsymbol{h}^{*}$ denoted for $k_{p}=1$ ).
state of the system (1) evolves in time along the convergence vector field towards the reference state. The transient stage of the vehicle motion can be effectively shaped by introducing the virtual reference velocity $\boldsymbol{v}^{*}$ defined in (16), see also (Michałek and Kozłowski, 2009). Using it in the definition (11) causes the vehicle directing effect, which turned out to be very useful in smoothing the vehicle motion when approaching the reference position and achieving the reference orientation $\theta_{t}$. The enhancement of the directing effect depends on the value of the design parameter ${ }^{[2} \eta$ (see (16)). Figure 2 illustrates the VFO stabilization strategy for the unicycle where the vehicle directing effect results from introduction of the virtual reference velocity vector $\boldsymbol{v}^{*}$. The proof of asymptotic convergence for the posture error $\boldsymbol{e}=\boldsymbol{q}_{t}-\boldsymbol{q}$ to zero, was presented in (Michałek and Kozłowski, 2009).

The VFO stabilizer is a discontinuous controller and belongs to the class of almost stabilizers (according to terminology proposed in (Astolfi, 1996)). The authors believe that the VFO methodology can be treated as a generalization of the control concept described in polar-coordinates. Simulations and experimental results presented in (Michałek and Kozłowski, 2009) revealed several practically important features of the closed-loop system with the VFO stabilizer, namely, fast and nonoscillatory posture convergence for any of the vehicle initial conditions and very simple controller parametric synthesis leading to the possibility of simple transient stage shaping. These features allow anticipating vehicle behavior during the convergence process and naturally motivate one to utilize the VFO concept for motion planning and control in the task of way point following.

## 4. VFO motion planning algorithm

The first stage of the proposed concept is a motion planning procedure, which involves determining the remaining way-points $\boldsymbol{q}_{t i}$ for $i=1,2, \ldots, N-1$ from the set (2). Let us assume (in addition to the assumptions made in

[^1]Subsection (2.2) that for all $i=1, \ldots, N-1$ the position components $\boldsymbol{q}_{t i}^{*}$ (see (3)) of the way-points are defined in advance, and only the orientation components $\theta_{t i}$ remain to be determined by the planning procedure. This assumption enables the user to shape the path of the robot in the task space. The aim is thus to describe how the way-point orientations are computed in the VFO planning stage.

The particular points from the set (2) divide the planned motion into $N$ segments. Motion in the $i$-th segment can be treated as a problem of set-point control between the initial point $\boldsymbol{q}_{t i-1}$ and the final one $\boldsymbol{q}_{t i}$. The crucial principle in the VFO motion planning stage is the computation of the way-point orientations $\theta_{t i}$ so that it guarantees the consistency of their values with the auxiliary orientation angles $\theta_{a}$ defined by the convergence vectors $\boldsymbol{h}^{*}$ (see (131) on the segment boundaries during the subsequent motion realization stage conducted with the VFO stabilizer. This principle comes from the requirement of continuity for time-evolution of the the auxiliary variable during the motion realization stage. It allows generating a smooth vehicle movement also when passing through segment boundaries. Computation details fulfilling the above principle are presented below. The motion planning computations are carried out for the nominal case, in which the vehicle is able to pass via all the waypoints accurately. ${ }^{3}$.

Our objective is to find the orientation $\theta_{t i-1}$ from the beginning of the $i$-th motion segment assuming that $\theta_{t i}$ for the end of the segment is already computed (the computational procedure starts from the target point $\boldsymbol{q}_{t N}$ to the first way-point $\boldsymbol{q}_{t 1}$ ). Let us describe the particular vectors obtained using the VFO motion strategy (presented in Section (3) for the $i$-th motion segment. The posture error for the $i$-th segment is defined in (7). The position error results from the equation

$$
\boldsymbol{e}_{i}^{*}(\tau) \triangleq \boldsymbol{q}_{t i}^{*}-\boldsymbol{q}^{*}(\tau)=\left[\begin{array}{c}
x_{t i}-x(\tau)  \tag{19}\\
y_{t i}-y(\tau)
\end{array}\right]
$$

where $\tau \in\left[\tau_{i-1}, \tau_{i}\right]$. Now, according to the notation introduced in Section 3, one can define

$$
\begin{align*}
& \boldsymbol{v}_{i}^{*}(\tau) \triangleq-\eta_{i} \operatorname{sgnU} 2_{i}\left\|\boldsymbol{e}_{i}^{*}(\tau)\right\| \boldsymbol{g}_{2 t i}^{*}  \tag{20}\\
& \boldsymbol{h}_{i}^{*}(\tau) \triangleq k_{p} \boldsymbol{e}_{i}^{*}(\tau)+\boldsymbol{v}_{i}^{*}(\tau)=\left[h_{2 i}(\tau) h_{3 i}(\tau)\right]^{T}  \tag{21}\\
& \theta_{a i}(\tau) \triangleq \operatorname{Atan} 2 \mathrm{c}\left(\operatorname{sgnU} 2_{i} \cdot h_{3 i}(\tau), \operatorname{sgnU}_{i} \cdot h_{2 i}(\tau)\right) \tag{22}
\end{align*}
$$

with

$$
\begin{equation*}
\boldsymbol{g}_{2 t i}^{*}=\left[\cos \theta_{t i} \sin \theta_{t i}\right]^{T} \tag{23}
\end{equation*}
$$

and where $0<\eta_{i}<k_{p}, k_{p}>0$ and $\operatorname{sgnU} 2_{i} \in\{+1,-1\}$ should be treated as control system design parameters

[^2]

Fig. 3. Description of particular vectors and the explanation of way-point orientation determination in subsequent motion segments for the VFO motion planning strategy ( $\boldsymbol{h}^{*}$ denoted for $k_{p}=1$ and $\operatorname{sgnU} 2_{i}=+1$ ).
chosen by the user. The terms in the above equations play in the $i$-th segment the same role as described in Section 3. The decision variable $\operatorname{sgnU} 2_{i}$ (introduced here instead of the $\operatorname{sgn}(k)$ function from (13)-(18)) allows choosing the motion strategy during approaching the $i$-th waypoint $\left(\operatorname{sgnU} 2_{i}=+1\right.$ for forward, $\operatorname{sgnU} 2_{i}=-1$ for backward motion).

The definitions (19)-(22) were written as functions of time, since they will be time-dependent during the motion realization stage. However, for the planning procedure, only their values at the initial time instant $\tau_{i-1}$ are important, since at this time instant the vehicle should arrive at the beginning of the $i$-th motion segment. Hence, for $\tau=\tau_{i-1}$, the auxiliary angle (22) takes the value

$$
\begin{equation*}
\theta_{a i}\left(\tau_{i-1}\right)=\arg \left(\boldsymbol{h}_{i}^{*}\left(\tau_{i-1}\right)\right) \tag{24}
\end{equation*}
$$

where for simplicity we introduced the notion $\arg \left(\boldsymbol{h}^{*}\right) \equiv$ Atan2c $\left(\operatorname{sgnU} 2_{i} \cdot h_{3}, \operatorname{sgnU} 2_{i} \cdot h_{2}\right)$. According to the VFO motion planning principle mentioned before, the desired orientation of the way-point $\boldsymbol{q}_{t i-1}$ results from the following substitution:

$$
\begin{equation*}
\theta_{t i-1}:=\theta_{a i}\left(\tau_{i-1}\right) \tag{25}
\end{equation*}
$$

Note that for the nominal case one has $e_{i}^{*}\left(\tau_{i-1}\right) \equiv$ $\boldsymbol{q}_{t i}^{*}-\boldsymbol{q}_{t i-1}^{*}$. Equation (25) expresses the VFO motion planning strategy. It means that the desired orientation in the $i$-th way-point should be consistent with the orientation of the convergence vector $\boldsymbol{h}^{*}$ computed for the point $\boldsymbol{q}_{t i}^{*}$ defining simultaneously the convergence direction to the next way-point from the set (2).

The VFO motion planning strategy is graphically explained in Fig. 3. The following algorithm summarizes the computations involved in the VFO motion planning stage:

S0. Initial data: $\boldsymbol{q}_{t 0} \equiv \boldsymbol{q}(0), \boldsymbol{q}_{t N}, k_{p}>0$
and $\boldsymbol{q}_{t i}^{*}, \eta_{i}, \operatorname{sgnU} 2_{i}$ for all $i=1, \ldots, N$;
S1. Counter initialization: $i:=N$;
S2. $\boldsymbol{e}_{i}^{*}\left(\tau_{i-1}\right)=\boldsymbol{q}_{t i}^{*}-\boldsymbol{q}_{t i-1}^{*}$;

S3. $\boldsymbol{v}_{i}^{*}\left(\tau_{i-1}\right)=-\eta_{i} \operatorname{sgnU} 2_{i}\left\|\boldsymbol{e}_{i}^{*}\left(\tau_{i-1}\right)\right\| \boldsymbol{g}_{2 t i}^{*} ;$
S4. $\boldsymbol{h}_{i}^{*}\left(\tau_{i-1}\right)=k_{p} \boldsymbol{e}_{i}^{*}\left(\tau_{i-1}\right)+\boldsymbol{v}_{i}^{*}\left(\tau_{i-1}\right) ;$
S5. $\theta_{a i}\left(\tau_{i-1}\right)=\arg \left(\boldsymbol{h}_{i}^{*}\left(\tau_{i-1}\right)\right)$;
S6. $\theta_{t i-1}:=\theta_{a i}\left(\tau_{i-1}\right)$;
S7. IF $(i==2)$ THEN STOP
ELSE $i:=i-1$ and GOTO S2.

## 5. VFO control for way point following

After the motion planning procedure, the second stage of the proposed concept, namely, motion realization, is considered. We propose to utilize at this stage the modified version of the VFO feedback controller presented in Section 3. This modification results from the following important issues. First, the original VFO stabilizer (8)-(9) is defined only for one motion segment determined by the initial and the final posture, guaranteeing asymptotic convergence for the posture error to zero. This means that the final position cannot be reached in finite time. Second, longitudinal velocity of the vehicle controlled by the original pushing control (9) evolves from relatively high value at the beginning of the transient stage to zero in the final stage. Hence, using the original definition of the controller for the way point following task would prevent the vehicle from passing smoothly and in finite time through particular segment boundaries and, as a consequence, from accomplishing the task considered.

According to the above, we propose to organize the motion realization stage as follows. To guarantee reaching the segment boundary in finite time, let us replace the asymptotic convergence demand of the original VFO stabilizer with practical convergence to the assumed non-zero vicinities $\epsilon_{1}, \ldots, \epsilon_{N}>0$ of the way-points $\boldsymbol{q}_{t 1}, \ldots, \boldsymbol{q}_{t N}$ in the particular motion segments. Relaxing the convergence demand remains consistent with Definition 1 and seems to be practically justified. We also propose, following the works of (Sasiadek and Duleba, 1995) and (Sordalen and de Wit, 1993), the switching procedure, which will be responsible for the activation of the next way-point from the set $\mathcal{S}_{t}$ as soon as the position error norm determined for the $i$-th motion segment reaches the assumed $\epsilon_{i}$-neighborhood of the $i$-th way-point. Activating the way-point should be understood as passing it on to the realization stage. The switching procedure can be described by the increasing condition for the index $i$ which indicates the currently active, i.e., being realized, motion segment:

$$
\begin{equation*}
\operatorname{IF}\left(\left(\left\|\boldsymbol{e}_{i}^{*}\right\| \leqslant \epsilon_{i}\right) \text { AND }(i<N)\right) \text { THEN } i:=i+1 \tag{26}
\end{equation*}
$$

assuming additionally that the initial index value $i:=1$ is set for $\tau=0$ (beginning of the motion realization stage).

Let us now define the VFO control inputs for the way point following task. The modified pushing control input for particular motion segments is defined as follows:
$u_{2}(\tau) \triangleq\left\{\begin{array}{lll}\rho_{i}\left\|\boldsymbol{h}_{i}^{*}(\tau)\right\| \cos \alpha_{i}(\tau) & \text { for } \tau \in\left[\tau_{i-1}, \tau_{i}\right), \\ 0 & \text { for } \tau \geqslant \tau_{N},\end{array}\right.$
where $i=1, \ldots, N, \boldsymbol{h}_{i}^{*}(\tau)$ is defined in (21), $\alpha_{i}(\tau)=$ $\angle\left(\boldsymbol{g}_{2}^{*}(\theta(\tau)), \boldsymbol{h}_{i}^{*}(\tau)\right)$, and the non-negative continuous scaling function $\rho_{i}=\rho_{i}(\cdot)$ is introduced in order to properly shape the longitudinal velocity profile along the motion segments. To make our discussion more general, we do not determine here any particular form of the function $\rho_{i}$. However, an example will be given in Subsection 5.1. The definition (27) indicates that the vehicle is stopped after reaching the assumed neighborhood $\epsilon_{N}>0$ of the target point $\boldsymbol{q}_{t N}$. In this way, the motion time horizon in the realization stage is finite, as required in Definition 1

The orienting control is defined as follows:

$$
u_{1}(\tau) \triangleq \begin{cases}k_{1} e_{a i}(\tau)+\dot{\theta}_{a i}(\tau) & \text { for } \quad \tau \in\left[\tau_{i-1}, \tau_{i}\right)  \tag{28}\\ k_{1}\left(\theta_{t N}-\theta(\tau)\right) & \text { for } \quad \tau \geqslant \tau_{N}\end{cases}
$$

where $i=1, \ldots, N, k_{1}>0$ is a design coefficient, and $e_{a i}(\tau)=\theta_{a i}(\tau)-\theta(\tau)$ with $\theta_{a i}(\tau)$ defined in (22). The feed-forward term $\dot{\theta}_{a i}$ comes from time-differentiation of (22) and has the form

$$
\begin{equation*}
\dot{\theta}_{a i}(\tau)=\frac{\dot{h}_{3 i}(\tau) h_{2 i}(\tau)-\dot{h}_{2 i}(\tau) h_{3 i}(\tau)}{h_{2 i}^{2}(\tau)+h_{3 i}^{2}(\tau)} \tag{29}
\end{equation*}
$$

with $\boldsymbol{h}_{i}^{*}(\tau)=\left[h_{2 i}(\tau) h_{3 i}(\tau)\right]^{T}$ defined by (20)-(21) and with

$$
\begin{equation*}
\dot{\boldsymbol{h}}_{i}^{*}(\tau)=-k_{p} \dot{\boldsymbol{q}}^{*}(\tau)+\dot{\boldsymbol{v}}_{i}^{*}(\tau) \tag{30}
\end{equation*}
$$

(compare 17), where

$$
\begin{equation*}
\dot{\boldsymbol{v}}_{i}^{*}(\tau)=-\eta_{i} \operatorname{sgn} U 2_{i} \frac{\boldsymbol{e}_{i}^{* T}(\tau) \dot{e}_{i}^{*}(\tau)}{\left\|e_{i}^{*}(\tau)\right\|} \boldsymbol{g}_{2 t i}^{*} \tag{31}
\end{equation*}
$$

with $\boldsymbol{g}_{2 t i}^{*}$ determined in (23), comes from timedifferentiation of (20).

Comparing (28) and (8), one can find that for all $N$ motion segments the orienting control for the way point following task is almost analogous to the original definition for the VFO stabilizer. The difference comes from the switching procedure applied here and from the last stage (for $\tau \geqslant \tau_{N}$ ), where the modified orienting control has to stabilize the orientation of the vehicle stopped by the pushing control (27) in the assumed non-zero neighborhood $\epsilon_{N}$ of the target point $\boldsymbol{q}_{t N}$.

Remark 1. The definitions (22) and (29) are well defined for $\left\|\boldsymbol{h}_{i}^{*}\right\| \neq 0$. Since the condition $\left\|\boldsymbol{h}_{i}^{*}\right\|=0$ can be met only for $\left\|e_{i}^{*}\right\|=0$ (see (21) and (20), one avoids this indeterminacy assuming that all vicinities $\epsilon_{1}, \ldots, \epsilon_{N}$ for the way-points are greater than zero. This implies that the
vehicle never reaches the current $i$-th way-point before the switching procedure activates the next way-point from the set $\mathcal{S}_{t}$, or before the pushing input (27) stops the vehicle (for $\tau \geqslant \tau_{N}$ ) in the neighborhood $\epsilon_{N}$ of the target point $\boldsymbol{q}_{t N}$.

The control input definitions (27) and (28) are proposed as a result of some heuristic approach based on the authors' experiences obtained so far during simulation and experimental tests with VFO controllers (see (Michałek and Kozłowski, 2009)). Stability and error convergence analysis in the closed-loop system with the proposed VFO controller for the way point following task is conducted in Subsection 5.2
5.1. Remarks on scaling function selection. The scaling function $\rho$ introduced in the definition (27) can be selected in many ways according to a particular application. The only constraint of its construction lies in positive semi-definiteness. The possible selection determines $\rho=\rho(s)$ as a function of some independent and normalized parameter $s \in[0,1]$. In a particular case, $s$ can be a time variable $(s \triangleq \tau)$. However, this selection would lead to the points-tracking task rather than to way point following, considered in this paper. For the latter, it seems to be more appropriate to define $s$ in terms of some geometrical terms related to a realized motion task (with analogy to the well-known proposition from (Samson, 1992)). One such proposition for the $i$-th motion segment can be defined as follows:

$$
\begin{equation*}
s_{i} \triangleq 1-\frac{\left\|\boldsymbol{h}_{i}^{*}(\tau)\right\|}{\left\|\boldsymbol{h}_{i}^{*}\left(\tau_{i-1}\right)\right\|}, \tag{32}
\end{equation*}
$$

where $\tau \in\left[\tau_{i-1}, \tau_{i}\right]$.
Since $\left\|\boldsymbol{h}_{i}^{*}(\tau)\right\|$ can evolve in time only when the vehicle moves with a non-zero longitudinal velocity $u_{2}$, the parameter $s_{i}$ evolves also in relation to vehicle motion. Now, the function $\rho_{i}=\rho_{i}\left(s_{i}\right)$ allows shaping longitudinal velocity taking, for instance,

$$
\begin{equation*}
\rho_{i}\left(s_{i}\right) \triangleq \frac{\bar{\rho}_{i}\left(s_{i}\right)}{\left\|\boldsymbol{h}_{i}^{*}(\tau)\right\|} \tag{33}
\end{equation*}
$$

where $\bar{\rho}_{i}\left(s_{i}\right)$ is a design function founded, for example, on the polynomial basis. The simplest example is the zeroorder polynomial $\bar{\rho}_{i} \triangleq U_{2}$ with $U_{2}>0$ being a constant denoting the desired driving velocity of the controlled vehicle5. Substituting (33) with the mentioned zero-order polynomial into the definition (27) gives the particular form of the VFO pushing control in the $i$-th motion segment as follows:

$$
\begin{equation*}
u_{2}(\tau)=U_{2} \cos \alpha_{i}(\tau) \tag{34}
\end{equation*}
$$

Note that the above proposition generally results in a piecewise continuous input signal, with the possible dis-

[^3]continuity points only in transition through the motion segment boundaries (at the time instants $\tau_{i}$ ).
5.2. Stability and convergence analysis. It will be shown that the VFO feedback controller defined in (27) and (28) applied to the vehicle model (1) together with the motion segment switching condition (26) guarantees accomplishing the task given in Definition 1. More specifically, one can show that

A1. The vehicle position error $\left\|e_{i}^{*}(\tau)\right\|$ in any $i$-th motion segment converges to the assumed non-zero $\epsilon_{i}-$ neighborhood of the $i$-th way-point position $\boldsymbol{q}_{t i}^{*}$ in finite time. As a consequence, the vehicle position converges from the initial point $\boldsymbol{q}^{*}(0)$ to the target one $\boldsymbol{q}_{t N}^{*}$ in finite time as well.

A2. The vehicle orientation $\theta(\tau)$ exponentially converges to the auxiliary orientation $\theta_{a i}(\tau)$ in any $i$-th motion segment.

A3. The vehicle orientation $\theta(\tau)$ asymptotically converges to the target orientation $\theta_{t N}$ after the vehicle reaches the assumed vicinity $\epsilon_{N}$ of the target position $\boldsymbol{q}_{t N}^{*}$.

Let us consider the $i$-th motion segment joining the current vehicle state $\boldsymbol{q}(\tau)$ with the $i$-th way-point $\boldsymbol{q}_{t i}$. The analysis starts by applying the orienting input (28) into (1), which gives

$$
\begin{equation*}
\dot{e}_{a i}(\tau)+k_{1} e_{a i}(\tau)=0 \quad \Rightarrow \quad \lim _{\tau \rightarrow \infty} e_{a i}(\tau)=0 \tag{35}
\end{equation*}
$$

The above equation allows concluding A2.
In the next step, let us recall the definition (7), which implies $\dot{\boldsymbol{e}}_{i}^{*}=-\dot{\boldsymbol{q}}^{*}$. This relation can be equivalently rewritten as (cf. (21))

$$
\dot{\boldsymbol{e}}_{i}^{*}=-\dot{\boldsymbol{q}}^{*}+\rho_{i}\left(\boldsymbol{h}_{i}^{*}-k_{p} \boldsymbol{e}_{i}^{*}-\boldsymbol{v}_{i}^{*}\right)
$$

where $\rho_{i}$ is the scaling function from (27). After reordering the above equation, one obtains

$$
\begin{equation*}
\dot{\boldsymbol{e}}_{i}^{*}+\rho_{i} k_{p} \boldsymbol{e}_{i}^{*}=\rho_{i} \boldsymbol{r}_{i}-\rho_{i} \boldsymbol{v}_{i}^{*} \tag{36}
\end{equation*}
$$

where

$$
\begin{equation*}
\boldsymbol{r}_{i}=\boldsymbol{h}_{i}^{*}-\boldsymbol{g}_{2}^{*}(\theta) \bar{u}_{2} \quad \text { with } \quad \bar{u}_{2}:=\left\|\boldsymbol{h}_{i}^{*}\right\| \cos \alpha_{i} . \tag{37}
\end{equation*}
$$

The latter formula can be easily obtained recalling that $\dot{\boldsymbol{q}}^{*}=\boldsymbol{g}_{2}^{*}(\theta) u_{2}$, where $\boldsymbol{g}_{2}^{*}=[\cos \theta \sin \theta]^{T}$ (see (1) and (27): $\dot{\boldsymbol{q}}^{*}=\boldsymbol{g}_{2}^{*}(\theta) \rho_{i}\left\|\boldsymbol{h}_{i}^{*}\right\| \cos \alpha_{i}=\rho_{i} \boldsymbol{g}_{2}^{*}(\theta) \bar{u}_{2}$. Additionally, it can be shown (see Appendix) that the following two relations hold:

$$
\begin{equation*}
\left\|\boldsymbol{r}_{i}\right\|=\left\|\boldsymbol{h}_{i}^{*}\right\| \gamma_{i}(\theta), \quad \lim _{\theta \rightarrow \theta_{a i}} \gamma_{i}(\theta)=0 \tag{38}
\end{equation*}
$$

where $\gamma_{i}(\theta)=\sqrt{1-\cos ^{2} \alpha_{i}(\theta)} \in[0,1]$ and $\alpha_{i}(\theta)=$ $\angle\left(\boldsymbol{g}_{2}^{*}(\theta), \boldsymbol{h}_{i}^{*}\right)$. Let us introduce the positive definite function $V_{i}\left(\boldsymbol{e}_{i}^{*}\right) \triangleq \frac{1}{2} \boldsymbol{e}_{i}^{* T} \boldsymbol{e}_{i}^{*}$. Its time-derivative along the solution of (36) can be estimated as follows:

$$
\begin{aligned}
\dot{V}_{i} & =\boldsymbol{e}_{i}^{* T} \dot{\boldsymbol{e}}_{i}^{*}=\boldsymbol{e}_{i}^{* T}\left[-\rho_{i} k_{p} \boldsymbol{e}_{i}^{*}+\rho_{i} \boldsymbol{r}_{i}-\rho_{i} \boldsymbol{v}_{i}^{*}\right] \\
& =-\rho_{i} k_{p}\left\|\boldsymbol{e}_{i}^{*}\right\|^{2}+\rho_{i} \boldsymbol{e}_{i}^{* T} \boldsymbol{r}_{i}-\rho_{i} \boldsymbol{e}_{i}^{* T} \boldsymbol{v}_{i}^{*} \\
& \leqslant-\rho_{i}\left[k_{p}\left\|\boldsymbol{e}_{i}^{*}\right\|^{2}-\left\|\boldsymbol{e}_{i}^{*}\right\|\left\|\boldsymbol{r}_{i}\right\|-\left\|\boldsymbol{e}_{i}^{*}\right\|\left\|\boldsymbol{v}_{i}^{*}\right\|\right] \\
& \stackrel{2038}{=}-\rho_{i}\left[k_{p}\left\|\boldsymbol{e}_{i}^{*}\right\|^{2}-\left\|\boldsymbol{e}_{i}^{*}\right\|\left\|\boldsymbol{h}_{i}^{*}\right\| \gamma_{i}-\eta_{i}\left\|\boldsymbol{e}_{i}^{*}\right\|^{2}\right] \\
& =-\rho_{i}\left[\left(k_{p}-\eta_{i}\right)\left\|\boldsymbol{e}_{i}^{*}\right\|^{2}-\gamma_{i}\left\|\boldsymbol{e}_{i}^{*}\right\|\left\|k_{p} \boldsymbol{e}_{i}^{*}+\boldsymbol{v}_{i}^{*}\right\|\right] \\
& \leqslant-\rho_{i}\left[\left(k_{p}-\eta_{i}-\gamma_{i} k_{p}\right)\left\|\boldsymbol{e}_{i}^{*}\right\|^{2}-\gamma_{i}\left\|\boldsymbol{e}_{i}^{*}\right\|\left\|\boldsymbol{v}_{i}^{*}\right\|\right] \\
& \stackrel{\text { 20 }}{=}-\rho_{i}\left[\left(k_{p}-\eta_{i}-\gamma_{i} k_{p}\right)\left\|\boldsymbol{e}_{i}^{*}\right\|^{2}-\gamma_{i} \eta_{i}\left\|\boldsymbol{e}_{i}^{*}\right\|^{2}\right] \\
& =-\rho_{i}\left[k_{p}-\eta_{i}-\gamma_{i}\left(k_{p}+\eta_{i}\right)\right]\left\|\boldsymbol{e}_{i}^{*}\right\|^{2} \\
& =-\rho_{i} \zeta\left(\gamma_{i}\right)\left\|\boldsymbol{e}_{i}^{*}\right\|^{2} .
\end{aligned}
$$

The above time-derivative is negative definite for the positive function $\rho_{i}$ if $\zeta\left(\gamma_{i}(\tau)\right)>0$. The last condition will be analyzed in the sequel, but first we focus our attention on the function $\rho_{i}$, which has crucial influence on the rate of position error time-evolution. Recalling Subsection 5.1 let us define $\rho_{i}$ as follows:

$$
\begin{equation*}
\rho_{i} \triangleq \frac{U_{2}}{\left\|\boldsymbol{h}_{i}^{*}(\tau)\right\|} \tag{39}
\end{equation*}
$$

where $U_{2}>0$ determines the user-defined longitudinal velocity value along the $i$-th motion segment. Using the definition (21), one gets $\left\|\boldsymbol{h}_{i}^{*}(\tau)\right\|=\left\|\boldsymbol{e}_{i}^{*}(\tau)\right\| \cdot\left\|\boldsymbol{\vartheta}_{i}(\tau)\right\|$, where $\boldsymbol{\vartheta}_{i}(\tau)=k_{p} \boldsymbol{\vartheta}_{e i}(\tau)-\eta_{i} \operatorname{sgnU} 2_{i} \boldsymbol{g}_{2 t i}^{*}$ and $\boldsymbol{\vartheta}_{e i}(\tau)$ is a unit vector of the position error $\boldsymbol{e}_{i}^{*}(\tau)$. Note that, since $\eta_{i}<k_{p}$, it is guaranteed that $\left\|\boldsymbol{\vartheta}_{i}(\tau)\right\| \neq 0$ for all $\tau \geqslant 0$. Now, an upper bound of $\dot{V}_{i}$ can be calculated as follows:

$$
\dot{V}_{i} \leqslant-\frac{U_{2} \cdot \zeta\left(\gamma_{i}\right)}{\left\|\boldsymbol{h}_{i}^{*}(\tau)\right\|}\left\|\boldsymbol{e}_{i}^{*}\right\|^{2}=-\frac{U_{2} \cdot \zeta\left(\gamma_{i}\right)}{\left\|\boldsymbol{\vartheta}_{i}(\tau)\right\|}\left\|\boldsymbol{e}_{i}^{*}\right\| .
$$

Let us recall and analyze the inequality $\zeta\left(\gamma_{i}(\tau)\right)>0$, which still remains the sufficient condition for the convergence of $\left\|e_{i}^{*}\right\|$. The convergence condition takes the following form:

$$
\begin{equation*}
\zeta\left(\gamma_{i}(\tau)\right)>0 \quad \Leftrightarrow \quad \gamma_{i}(\tau)<\frac{k_{p}-\eta_{i}}{k_{p}+\eta_{i}} \tag{40}
\end{equation*}
$$

By assumption, one has $\forall_{i=1, \ldots, N} 0<\eta_{i}<k_{p}$, hence the ratio $\left(k_{p}-\eta_{i}\right) /\left(k_{p}+\eta_{i}\right)<1$. Since $\gamma_{i}(\theta(\tau)) \in[0,1]$ for all $\tau \in \mathbb{R}$, and since (35) and (38) hold, one concludes that there exists a finite time instant $\tau_{\gamma i} \in\left[\tau_{i-1}, \infty\right)$ such that

$$
\begin{equation*}
\forall_{\tau \geqslant \tau_{\gamma i}} \quad \gamma_{i}(\tau)<\frac{k_{p}-\eta_{i}}{k_{p}+\eta_{i}} \tag{41}
\end{equation*}
$$

and the function $\zeta\left(\gamma_{i}(\tau)\right)$ becomes positive for all $\tau \geqslant$ $\tau_{\gamma i}$. For the finite time-interval $\left[\tau_{i-1}, \tau_{\gamma i}\right)$ we cannot, in general, guarantee that $\zeta\left(\gamma_{i}(\tau)\right)$ is positive and, consequently, that $V_{i}$ is non-increasing. However, we can show that the finite-time escape for $\left\|e_{i}^{*}\right\|$ is also not possible. Namely, in the worst case when $\gamma_{i}=1$ one obtains $\zeta\left(\gamma_{i}=1\right)=-2 \eta_{i}$ yielding $\dot{V}_{i}(\tau) \leqslant$ $\left(2 U_{2} \eta_{i}\left\|\boldsymbol{e}_{i}^{*}(\tau)\right\| /\left\|\boldsymbol{\vartheta}_{i}(\tau)\right\|\right)<\infty$ (we assume that $\left.\left\|e_{i}^{*}(0)\right\|<\infty\right)$. Since the last inequality may hold only for $\tau \in\left[\tau_{i-1}, \tau_{\gamma i}\right)$, where $\tau_{\gamma i}$ is finite, the norm $\left\|e_{i}^{*}\right\|$, the functions $V_{i}$ and $\dot{V}_{i}$ remain bounded also in the time interval $\left[\tau_{i-1}, \tau_{\gamma i}\right)$. Hence, let us consider timeevolution of $V_{i}(\tau)$ and $\left\|e_{i}^{*}(\tau)\right\|$ for $\tau \geqslant \tau_{\gamma i}$, accepting that $V_{i}\left(e_{i}^{*}\left(\tau_{\gamma i}\right)\right) \geqslant V_{i}\left(e_{i}^{*}(0)\right)$.

Now, an upper bound of the time-derivative of the function $V_{i}$ can be estimated as follows:

$$
\begin{equation*}
\dot{V}_{i} \leqslant-\frac{U_{2} \cdot \zeta\left(\gamma_{i}\right)}{\left\|\boldsymbol{\vartheta}_{i}(\tau)\right\|}\left\|\boldsymbol{e}_{i}^{*}\right\| \leqslant-c_{i} \sqrt{V_{i}} \tag{42}
\end{equation*}
$$

with

$$
\begin{equation*}
c_{i}=\frac{\sqrt{2} U_{2} \zeta\left(\gamma_{i m}\right)}{k_{p}+\eta_{i}}=\sqrt{2} U_{2}\left(\frac{k_{p}-\eta_{i}}{k_{p}+\eta_{i}}-\gamma_{i m}\right)>0 \tag{43}
\end{equation*}
$$

The bounding value $\gamma_{i m}$ from (43) can be estimated in two ways:

W1. as the initial value $\gamma_{i}\left(\tau_{i-1}\right)$ if it fulfills the condition (40) at the beginning of the $i$-th motion segment ( $\tau_{\gamma i}=\tau_{i-1}$ ), otherwise

W2. as a maximal value of function $\gamma_{i}$ which fulfills (40), i.e., equal to $\gamma_{i}\left(\tau_{\gamma i}\right)$.

According to the work of (Bhat and Bernstein, 2000), the result obtained in (42) allows concluding finite-time convergence for the position error $\boldsymbol{e}_{i}^{*}(\tau)$ to zero in the $i$-th motion segment. The convergence time interval $T_{i}=\tau_{i}-$ $\tau_{\gamma i}$ can be estimated as follows, cf. (Bhat and Bernstein, 2000)

$$
\begin{equation*}
T_{i} \leqslant \hat{T}_{i}, \quad \text { where } \quad \hat{T}_{i}=\frac{2}{c_{i}} \sqrt{V_{i}\left(e_{i}^{*}\left(\tau_{\gamma i}\right)\right)} \tag{44}
\end{equation*}
$$

Note that $\hat{T}_{i}$ depends on the coefficient $c_{i}$ estimated in (43), which in turn depends on the estimated value of $\gamma_{i m}$. When $\gamma_{i m}$ can be estimated as in W1, (44) may be useful in practice, as will be shown in Section 6. In the case of W2, however, (44) gives rather a theoretical solution, since it provides a very conservative estimate of the convergence time.

As a direct consequence of the finite-time convergence result, the time instant $\tau_{i}$ when the norm $\left\|e_{i}^{*}(\tau)\right\|$ enters into the nonzero $\epsilon_{i}$-neighborhood of the $i$-th waypoint must be finite and is less than $\tau_{\gamma i}+\hat{T}_{i}$. Using the
switching procedure (26), which activates the next waypoint in the time instant $\tau_{i}$, implies that the $N$ motion segments are completed in finite time:

$$
\tau_{N}<\sum_{i=1}^{N}\left(\left(\tau_{\gamma i}-\tau_{i-1}\right)+\hat{T}_{i}\right) \quad \text { with } \quad \tau_{0}=0
$$

Since the input $u_{2}(\tau)$ is zero for $\tau \geqslant \tau_{N}$ (compare (27), the vehicle stops in the $\epsilon_{N}$-neighborhood of the target position $\boldsymbol{q}_{t N}^{*}$. This completes A1.

To show A3, it suffices to substitute into the model (11) the orienting control input (28) for $\tau \geqslant \tau_{N}$ resulting in the following equation: $\dot{\theta}(\tau)+k_{1} \theta(\tau)=k_{1} \theta_{t N}$. It is evident that, after reaching the $\epsilon_{N}$-neighborhood of the target position, the vehicle orientation will converge exponentially to the target orientation $\theta_{t N}$ with the time constant equal to $1 / k_{1}$.

Next it is of interest to discuss two issues not treated explicitly in the preceding analysis and concerning control quality in the closed-loop system with the proposed VFO controller.

First, it is worth noting that the function $\rho_{i}$ introduced in (27) can take the zero value in a finite number of time instants not violating at the same time the finitetime convergence result obtained above for the position error $\boldsymbol{e}_{i}^{*}(\tau)$. It can be seen from (36) that for $\rho_{i}=0$ one has $\dot{e}_{i}^{*}=\mathbf{0}$ and the position error cannot diverge. This property gives great flexibility in shaping, using the function $\rho_{i}$, the longitudinal velocity profile for the vehicle in practical tasks. The second issue concerns time evolution of the vehicle orientation $\theta(\tau)$ in relation to the way-point orientations $\theta_{t i}$ computed in the planning stage for $i=1, \ldots, N-1$. According to the motion planning procedure presented in Section 4 the way-point orientation $\theta_{t i}$ is computed to keep the continuity in time evolution of the auxiliary angle $\theta_{a i}(\tau)$ (defined in (22)) during segment boundary transition in the motion realization stage.

The continuity issue can be explained as follows. For the case in which we assume that for all $i=1, \ldots, N-1$ $\lim _{\tau \rightarrow \tau_{i}}\left\|e_{i}^{*}(\tau)\right\|=0$ and $e_{a i}(\tau) \equiv 0 \Rightarrow \theta(\tau) \equiv$ $\theta_{a i}(\tau)$, one can show that $\lim _{\tau \rightarrow \tau_{i}} \theta_{a i}(\tau)=\theta_{t i}$. Since $\theta_{t i}:=\theta_{a i+1}\left(\tau_{i}\right)$ (according to the step S6 in Section 4), one concludes that $\lim _{\tau \rightarrow \tau_{i}} \theta_{a i}(\tau)=\theta_{a i+1}\left(\tau_{i}\right)$ and, as a consequence of the assumed equality $\theta(\tau) \equiv \theta_{a i}(\tau)$, that $\theta\left(\tau_{i}^{-}\right)=\theta\left(\tau_{i}^{+}\right)$, yielding continuous evolution also for the vehicle orientation during the segment boundary transition. However, due to the assumption about the nonzero $\epsilon_{i}$ values in the motion realization stage, one cannot generally guarantee that the auxiliary orientation variable $\theta_{a i}(\tau)$ (and also the vehicle orientation $\theta(\tau)$ ) will precisely converge to the planned way-point orientation $\theta_{t i}$ in the neighborhood of the segment boundary. The resultant discontinuity in the evolution of the auxiliary variable can be minimized by increasing the intensity of the direct-
ing effect, choosing higher values for the $\eta_{i}$ parameters, as mentioned in Section 3 It is worth noting that, in spite of mentioned discontinuity, the feed-forward term $\dot{\theta}_{a i}(\tau)$ present in (28) and computed according to (29) will be bounded, since the time-derivatives (30) and (31) are also bounded during the segment boundary transition for any non-zero value of $\epsilon_{i}$.

## 6. Simulation results

To show the effectiveness of the proposed concept, two numerical simulation tests, denoted as $\operatorname{SimA}$ and $\operatorname{SimB}$, are conducted for the unicycle model (11). In both cases the same set of the six way-points is defined with the position components $\left[x_{t i} y_{t i}\right]^{T}=\boldsymbol{q}_{t i}^{*}$ presented in Tables 1 and 2 The initial vehicle posture, being simultaneously the first way-point, is chosen as: $\boldsymbol{q}_{t 0}=\boldsymbol{q}(0)=\left[\begin{array}{lll}0 & -4 & 3.5\end{array}\right]^{T}$. The VFO parameters for all $i=1, \ldots, N$ are $\eta_{i}=3.5$, $\epsilon_{i}=0.005 \mathrm{~m}, k_{1}=10$, and $k_{p}=5$ (both in $[1 / \mathrm{s}] 7$. Tables 1 and 2 include also the way-point orientations $\theta_{t i}$. The orientations computed in the motion planning stage are denoted in bold. The initial and the target way-point orientations ( $\theta_{t 0}$ and $\theta_{t N}$ ) are defined in advance by the user and are not modified during the planning stage. For the simulations, the scaling function $\rho$ in the definition (27) is chosen as follows:

$$
\rho_{i}\left(\boldsymbol{h}_{i}^{*}\right) \triangleq \begin{cases}\frac{U_{2}}{\left\|\boldsymbol{h}_{i}^{*}(\tau)\right\|} & \text { for } \quad i=1, \ldots, N-1  \tag{45}\\ \frac{U_{2}}{\left\|\boldsymbol{h}_{N}^{*}\left(\tau_{N-1}\right)\right\|} & \text { for } \quad i=N\end{cases}
$$

with $U_{2}=0.4 \mathrm{~m} / \mathrm{s}$.
Note that for all the motion segments except the last one (the $N$-th one) the scaling function $\rho_{i}$ has the same form as in 39, which was used in the convergence analysis in Subsection 5.2 leading to the finite-time convergence result. The proposed definition of $\rho_{N}$ has been motivated only by the requirement of smooth decreasing of longitudinal vehicle velocity during approaching the $\epsilon_{N}$-neighborhood of the target position $\boldsymbol{q}_{N}^{*}$. Since in this case the Lyapunov analysis yields the inequality $\dot{V}_{N} \leqslant-\left(U_{2} \zeta\left(\gamma_{N}\right) /\left\|\boldsymbol{h}_{N}^{*}\left(\tau_{N-1}\right)\right\|\right)\left\|\boldsymbol{e}_{N}^{*}\right\|^{2}$, the convergence of the position error to zero is asymptotic. However, since the $\epsilon_{N}$-neighborhood is greater that zero (by assumption), the entering-time instant $\tau_{N}$ must be finite also in this case.

Using the definition (45) in (27) gives the pushing control input in the form of (34) for all the motion segments excluding the last one. In the $N$-th motion segment,

[^4]the pushing input takes the form
$$
u_{2}=U_{2} \frac{\left\|\boldsymbol{h}_{N}^{*}(\tau)\right\|}{\left\|\boldsymbol{h}_{N}^{*}\left(\tau_{N-1}\right)\right\|} \cos \alpha_{N}(\tau)
$$
which continuously converges to zero when the vehicle approaches the $\epsilon_{N}$-neighborhood of the target way-point position.

The simulation tests SimA and SimB differ only in motion strategy selection indicated by the values of the $\operatorname{sgnU} 22_{i}$ parameter $\left(\operatorname{sgnU} 2_{i}=+1\right.$ for the forward motion and $\operatorname{sgnU} 2_{i}=-1$ for the backward one) included in Tables 1 and 2 According to the presented values, the desired vehicle motion in the test $\operatorname{SimA}$ was set to the forward strategy for all motion segments. In the test $\operatorname{SimB}$, the way-points $\boldsymbol{q}_{t 2}$ and $\boldsymbol{q}_{t 3}$ should be approached in the backward manner. The obtained result 88 are illustrated in Figs. 4 -7. The way-point positions are indicated in Figs. 4 and by small circles; the way-point orientations are denoted by short straight lines.

It is worth noting, referring to Figs. 6 and 7 that time evolution of the posture errors $e_{\theta i}(\tau)=\theta_{t i}-\theta(\tau)$, $e_{x i}(\tau)=x_{t i}-x(\tau)$, and $e_{y i}(\tau)=y_{t i}-y(\tau)$ is fast and non-oscillatory towards the subsequent way-points. From the time plots on a logarithmic scale one can see that the evolution rate of errors in the first four motion segments is higher than exponential, as a consequence of finite-time convergence. Tables 3 and 4 present the estimated and the obtained time intervals for the three selected way-points in both simulation tests. The comparison of particular values reveals that the upper bound from (44) computed for the coefficient estimated in (43) is rather conservative in the cases considered. The task realization times $\tau_{N}$ for particular tests are $\tau_{N}=39.6 \mathrm{~s}$ for SimA and $\tau_{N}=39.8 \mathrm{~s}$ for SimB.

The plots in Figs. 4 and 5 indicate that vehicle behavior along the particular segments seems to be intuitively predictable in spite of the fact that the path between the way-points is not defined explicitly. Note also that time evolution of the auxiliary variable $\theta_{a i}(\tau)$ does not reveal any substantial discontinuity (caused by the nonzero $\epsilon_{i}$ vicinities), which, together with the characteristic directing effect, yields practically acceptable and smooth transition through the motion segment boundaries. Worth noting is the simplicity in motion strategy selection (forward/backward) involving only the appropriate value selection for the decision variable sgnU2 ${ }_{i}$.

## 7. Conclusions

The VFO motion planning and feedback control strategy presented in this paper allows treating the problem of driving the unicycle through the sequence of desired way-

[^5]

Fig. 4. SimA: vehicle motion in the global frame obtained in the simulation A.


Fig. 5. SimB: vehicle motion in the global frame obtained in the simulation B.

Table 1. Coordinates of the way-points and motion strategy used in the simulation A .

| $i$ | 0 | 1 | 2 | 3 | 4 | 5 |
| :---: | :---: | :---: | :---: | :---: | :---: | :---: |
| $\theta_{t i}[\mathrm{rad}]$ | 0.00 | $\mathbf{- 1 . 5 0}$ | $\mathbf{1 . 0 5}$ | $\mathbf{- 1 . 1 7}$ | $\mathbf{0 . 0 1}$ | 1.57 |
| $x_{t i}[\mathrm{~m}]$ | -4.0 | -2.0 | -1.0 | 0.0 | 1.0 | 1.5 |
| $y_{t i}[\mathrm{~m}]$ | 3.5 | 3.0 | 1.0 | 1.5 | 1.0 | 1.5 |
| sgnU $_{i}$ | - | +1 | +1 | +1 | +1 | +1 |

Table 2. Coordinates of the way-points and motion strategy used in the simulation B .

| $i$ | 0 | 1 | 2 | 3 | 4 | 5 |
| :---: | :---: | :---: | :---: | :---: | :---: | :---: |
| $\theta_{t i}[\mathrm{rad}]$ | 0.00 | $\mathbf{- 5 . 0 2}$ | $\mathbf{- 3 . 3 1}$ | $\mathbf{- 1 . 1 7}$ | $\mathbf{0 . 0 1}$ | 1.57 |
| $x_{t i}[\mathrm{~m}]$ | -4.0 | -2.0 | -1.0 | 0.0 | 1.0 | 1.5 |
| $y_{t i}[\mathrm{~m}]$ | 3.5 | 3.0 | 1.0 | 1.5 | 1.0 | 1.5 |
| $\operatorname{sgnU}_{i}$ | - | +1 | -1 | -1 | +1 | +1 |

points in a simplified and effective way. The simplification results from the fact that the motion planning stage does not involve any interpolating procedure between the
way-points with any geometrical path as opposed to many solutions proposed in the literature. Motion planning relies only on way-point orientation computations starting

Table 3. Selected values of the obtained and the estimated convergence time intervals for the test SimA.

| $i$ | 2 | 3 | 4 |
| :---: | :---: | :---: | :---: |
| $T_{i}[\mathrm{~s}]$ | 6.5 | 3.5 | 3.0 |
| $\hat{T}_{i}[\mathrm{~s}]$ | 31.8 | 15.9 | 16.3 |
| $\tau_{i}[\mathrm{~s}]$ | 12.9 | 16.4 | 19.4 |

Table 4. Selected values of the obtained and the estimated convergence time intervals for test SimB.

| $i$ | 2 | 3 | 4 |
| :---: | :---: | :---: | :---: |
| $T_{i}[\mathrm{~s}]$ | 6.7 | 3.5 | 3.0 |
| $\hat{T}_{i}[\mathrm{~s}]$ | 31.8 | 16.0 | 16.1 |
| $\tau_{i}[\mathrm{~s}]$ | 13.1 | 16.6 | 19.6 |

from the target posture and finishing on the first userdefined way-point from the set $\mathcal{S}_{t}$. The simplicity of the approach comes from the unique features of the VFO stabilizer which was adapted here to the way point following task. The main important features include predictable and non-oscillatory transients of the vehicle with the useful $d i$ recting effect, intuitive geometrical interpretation of VFO control inputs and, as a consequence, very simple parametric synthesis of the controller. The concept guarantees passing in finite time from the initial vehicle posture to the target one driving via all the way-points with the assumed finite precision, yielding finally practical stability of the closed-loop system 9 in the neighborhood of the target point. The desired motion strategy (forward/backward motion) in approaching the particular way-points can be freely and easily shaped by the bi-valued decision variable (in our case, $\operatorname{sgnU} 2_{i}$ ). The profile of longitudinal vehicle velocity can be easily shaped by introducing the scaling function in definition of the pushing input. The predictability of vehicle motion during transients between particular way-points together with the flexibility of motion strategy shaping allows designing the resulting vehicle path geometry in a simple and effective way.

Possible future extensions of the work may include automatic selection of the $\eta_{i}$ parameters responsible for the intensity of the directing effect, automatic motion safety estimation in the sense of the room size needed for task realization, and the adoption of the proposed concept to other classes of mobile robots. Experimental validation of the proposed method will be conducted in near future using a newly designed differentially-driven Leonardi transportation vehicle, being now under construction. The control system will be designed in a cascade form with two low-level PI velocity control loops for vehicle directdrives. The higher-level kinematic VFO controller pro-

[^6]

Fig. 6. SimA: time plots of signals obtained in the simulation A.
posed in this paper will be responsible for computing the desired wheel velocities for the low-level PI loops. Position feedback of the vehicle platform layer will be realized by the fusion of signals from wheel encoders, a laser scanner and a vision system mounted on board. Practical utilization of the proposed control method involves treating the control inputs limitations, which can be taken into ac-


Fig. 7. SimB: time plots of signals obtained in the simulation B.
count using the scaling procedure described in (Michałek and Kozłowski, 2009).

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## Appendix

Derivation of the left-hand side equation of (38). Recalling (37), one can write

$$
\begin{aligned}
\boldsymbol{r}_{i}=\boldsymbol{h}_{i}^{*}-\boldsymbol{g}_{2}^{*} \bar{u}_{2}=\left[\begin{array}{l}
h_{2 i} \\
h_{3 i}
\end{array}\right] & -\left[\begin{array}{l}
\bar{u}_{2} \mathrm{c} \theta \\
\bar{u}_{2} \mathrm{~s} \theta
\end{array}\right] \\
& =\left\|\boldsymbol{h}_{i}^{*}\right\|\left[\begin{array}{c}
\frac{h_{2 i}}{\left\|\boldsymbol{h}_{i}^{*}\right\|}-\mathrm{c} \alpha_{i} \mathrm{c} \theta \\
\frac{h_{3 i}}{\left\|\boldsymbol{h}_{i}^{*}\right\|}-\mathrm{c} \alpha_{i} \mathrm{~s} \theta
\end{array}\right],
\end{aligned}
$$

where the notation $\mathrm{c} \beta \equiv \cos \beta, \mathrm{s} \beta \equiv \sin \beta$ is used. Next, it is easy to calculate the norm of the vector $\boldsymbol{r}_{i}$ :

$$
\begin{aligned}
\left\|\boldsymbol{r}_{i}\right\|^{2}= & \left\|\boldsymbol{h}_{i}^{*}\right\|^{2}\left[\frac{h_{2 i}^{2}}{\left\|\boldsymbol{h}_{i}^{*}\right\|^{2}}-\frac{2 h_{2 i} \mathrm{c} \alpha_{i} \mathrm{c} \theta}{\left\|\boldsymbol{h}_{i}^{*}\right\|}+\mathrm{c}^{2} \alpha_{i} \mathrm{c}^{2} \theta\right. \\
& \left.+\frac{h_{3 i}^{2}}{\left\|\boldsymbol{h}_{i}^{*}\right\|^{2}}-\frac{2 h_{3 i} \mathrm{c} \alpha_{i} \mathrm{~s} \theta}{\left\|\boldsymbol{h}_{i}^{*}\right\|}+\mathrm{c}^{2} \alpha_{i} \mathrm{~s}^{2} \theta\right] \\
= & \left\|\boldsymbol{h}_{i}^{*}\right\|^{2}\left[1-2 \mathrm{c} \alpha_{i} \frac{h_{2 i} \mathrm{c} \theta+h_{3 i} \mathrm{~s} \theta}{\left\|\boldsymbol{h}_{i}^{*}\right\|}+\mathrm{c}^{2} \alpha_{i}\right] \\
= & \left\|\boldsymbol{h}_{i}^{*}\right\|^{2}\left(1-2 \mathrm{c} \alpha_{i} \mathrm{c} \alpha_{i}+\mathrm{c}^{2} \alpha_{i}\right) \\
= & \left\|\boldsymbol{h}_{i}^{*}\right\|^{2}\left(1-\mathrm{c}^{2} \alpha_{i}\right)
\end{aligned}
$$

and, finally,

$$
\left\|\boldsymbol{r}_{i}\right\|=\left\|\boldsymbol{h}_{i}^{*}\right\| \sqrt{1-\cos ^{2} \alpha_{i}(\theta)}=\left\|\boldsymbol{h}_{i}^{*}\right\| \gamma_{i}(\theta) .
$$

Calculation of the limit in (38) Knowing that $\cos \alpha_{i}=$ $\left(\boldsymbol{g}_{2}^{* T}(\theta) \boldsymbol{h}_{i}^{*}\right) /\left(\left\|\boldsymbol{g}_{2}^{*}(\theta)\right\|\left\|\boldsymbol{h}_{i}^{*}\right\|\right)$, one can obtain

$$
\begin{aligned}
\gamma_{i}^{2}(\theta) & =1-\mathrm{c}^{2} \alpha_{i}(\theta) \\
& =1-\frac{\left(h_{2 i} \mathrm{c} \theta+h_{3 i} \mathrm{~s} \theta\right)^{2}}{\left\|\boldsymbol{h}_{i}^{*}\right\|^{2}\left\|\boldsymbol{g}_{2}^{*}\right\|^{2}} \\
& =\frac{h_{2 i}^{2}+h_{3 i}^{2}-\left(h_{2 i} \mathrm{c} \theta+h_{3 i} \mathrm{~s} \theta\right)^{2}}{h_{2 i}^{2}+h_{3 i}^{2}} \\
& =\frac{\left(h_{2 i} \mathrm{~s} \theta-h_{3 i} \mathrm{c} \theta\right)^{2}}{h_{2 i}^{2}+h_{3 i}^{2}} .
\end{aligned}
$$

For $\theta(\tau) \rightarrow \theta_{a i}(\tau)$, according to (22) we have

$$
\lim _{\theta \rightarrow \theta_{a i}} \tan \theta=h_{3 i} / h_{2 i} \Rightarrow \lim _{\theta \rightarrow \theta_{a i}} \mathrm{~s} \theta=\left(h_{3 i} \mathrm{c} \theta\right) / h_{2 i}
$$

which, substituted into the preceding equation, allows concluding that $\lim _{\theta \rightarrow \theta_{a i}} \gamma_{i}(\theta)=0$.

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[^0]:    ${ }^{1}$ A similar terminology was proposed in (Lawrence et al., 2008).

[^1]:    ${ }^{2}$ Strictly speaking, the enhancement depends on the difference $k_{p}$ $\eta$, see (Michałek and Kozłowski, 2009) for more details.

[^2]:    ${ }^{3}$ This assumption will be further weakened for the motion realization stage.
    ${ }^{4}$ Note that $\theta_{t N}$ is defined in advance and does not involve any computations-see the prerequisites in Subsection 2.2

[^3]:    ${ }^{5}$ Understood as an absolute value

[^4]:    ${ }^{6}$ In relation to the chosen value of the $k_{p}$ coefficient (see (Michałek and Kozłowski, 2009)).
    ${ }^{7}$ Particular values have been selected here according to the general hints presented in (Michałek and Kozłowski, 2009) and in part by the trial-and-error method.

[^5]:    ${ }^{8}$ Simulations were conducted with the Matlab/Simulink software using the variable-step ode45 (Dormand-Prince) solver option.

[^6]:    ${ }^{9}$ Practical stability means ultimate boundedness of the stabilization error by its convergence to the assumed vicinity of the origin (Morin and Samson, 2003; Kozłowski and Pazderski, 2004).

