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1 **Oblique reflection of large internal solitary waves in a two-layer fluid**

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11
12 **Abstract**

13 The oblique reflection of an incident internal solitary wave is investigated using a fully-nonlinear
14 and strongly-dispersive internal wave model. The 3rd order theoretical solution for an internal
15 solitary wave in a two-layer system is used for the incident solitary wave. Two different incident
16 wave amplitude cases are investigated, in which nine and eleven different incident angles are used
17 for the small and large incident amplitude cases respectively. Under both amplitudes, at least for the
18 cases investigated here, relatively smaller incident angles result in Mach reflection while relatively
19 larger incident angles result in regular reflection. Under Mach-like reflection generation of a ‘stem’
20 is observed for a certain range of incident angles, in addition to the reflected wave. The stem is

21 found to have, in a certain sense, the characteristics of an internal solitary wave, though the
22 maximum stem wave amplitude is less than four times as large as the original incident internal
23 solitary wave. The stem length is confirmed to increase faster for the larger incident wave amplitude.
24 The maximum amplification factor for the small incident wave is the same as in previous studies.
25 However, the maximum amplification factor for the large incident wave is less than that for the
26 small wave. The results of these calculations are compared with those of the corresponding KP
27 theory and it is found that a lower amplification factor may be a significant characteristic of internal
28 solitary waves.

29

30 **Keywords:** variational principle; solitary wave; interaction; Mach stem; two-layer system; KP
31 theory

32

33

34 **1. Introduction**

35 The mechanism of occurrence of large amplitude surface waves in shallow water regions has
36 been discussed [Kharif and Pelinovsky, 2003], along with similar kinds of problems related to
37 “Freak waves” in deep water. Kharif and Pelinovsky [2003] suggested that one of the significant
38 causals of “Freak waves” is soliton resonance, which occurs due to the interaction of two solitary
39 waves. In contrast to surface waves, previous studies have revealed that large-amplitude internal
40 solitary waves may exist in the ocean based on images taken from the aircraft and satellites [Wang
41 and Pawlowicz, 2012] [Xue et al., 2013]. For instance, Helfrich and Melville [2006] provided
42 images of the interaction of internal solitary waves. In a recent study, Shimizu and Nakayama [2017]
43 provided the occurrence of large-amplitude internal solitary waves due to resonance in the Andaman
44 Sea by using a three-dimensional MITgcm simulations [Marshall et al., 1997] [Adcroft et al., 1997].
45 Shimizu and Nakayama [2017] demonstrated that the theoretical and numerical studies are required
46 to clarify how such a large-amplitude internal solitary wave occur. However, the interaction of
47 internal solitary waves has not been adequately investigated in previous studies. For example, Yuan
48 et al. [2018] demonstrated the importance of nonlinear interaction of soliton resonance of internal
49 waves with the topographic effect. In particular, the study of internal solitary waves, as steady
50 progressive nonlinear waves, should be a promising avenue for not only clarifying the phenomenon
51 itself but also for understanding the behavior of nonlinear internal waves. Therefore, this study aims
52 to investigate the two-dimensional interactions of internal solitary waves due to soliton resonance by
53 using numerical simulation.

54 For surface waves, Miles [1977] theoretically proposed the concept of “resonance”, which is
55 the interaction of three solitary waves with different incident angles in the two-dimensional weakly
56 nonlinear interaction of shallow water solitary waves. Miles applied this concept to the phenomenon
57 called Mach reflection in which the third solitary wave (stem), together with usual reflected wave, is
58 generated around the wall during the reflection process. The theory insists that the stem amplitude is
59 4 times the amplitude of the incident solitary wave at the critical incident angle defined as the angle
60 when the maximum amplitude occurs and the angle between Mach and regular reflection under
61 weakly nonlinear condition [Melville, 1980]. Mach reflection occurs when an incident angle is less
62 than the critical incident angle. Funakoshi [1980] numerically computed the reflection problem using
63 the Boussinesq equations for shallow water in which an incident solitary wave propagating along
64 one straight wall was reflected due to another straight wall oblique to the straight wall, roughly
65 supporting Miles’ results under weakly nonlinear conditions (**Fig. 1**). However, regarding the
66 maximum amplitude, it seems that the critical incident angle in Funakoshi’s result was somewhat
67 smaller than that of Miles. For an incident angle sufficiently larger than the critical incident angle,
68 Funakoshi’s result is in better agreement with the other result of Miles (perturbation solutions in
69 weak interaction) [Miles 1977]. However, Tanaka [1993] investigated the oblique reflection of a
70 large amplitude solitary wave by numerically solving the inviscid water wave equations using a
71 spectral method, finding that the maximum amplitude is about three times the amplitude of the
72 incident solitary wave. The critical incident angle in Tanaka’s numerical result was much smaller
73 than the value predicted by Miles’ theory. Yeh et al. [2010] and Li et al. [2011] analytically and

74 experimentally studied the reflection of an obliquely incident solitary wave, finding that the
75 maximum fourfold amplification predicted by Miles was not realized in a laboratory experiment
76 under strong nonlinear condition. Gidel et al. [2017] also showed the slightly small amplification
77 factor compared to Miles. These studies suggest that there are some differences between the weakly
78 nonlinear theory of Miles and other numerical and experimental results. Though Kodama [2010] and
79 Kodama et al. [2016] improve theoretical result by detailed analysis for the Kadomtsev-Petviashvili
80 (KP) equation which is a horizontally two-dimensional version of the KdV equation, the reason for
81 the quantitative differences of amplification in the neighborhood of the margin between regular and
82 Mach reflection is still unclear. It has been suggested that fully-nonlinear and strongly-dispersive
83 wave equations are needed for analyzing the deformations of solitary waves.

84 For internal waves, Maxworthy [1980] carried out laboratory experiments that showed the
85 occurrence of a Mach stem in the interaction of two internal solitary waves. However, there are few
86 experimental studies regarding the occurrence of a Mach stem in stratified flow fields. Theoretically,
87 Tsuji and Oikawa [2007] demonstrated that the importance of “critical depth” which may suppress
88 amplification rate due to soliton resonance by using the Extended Kadomtsev-Petviashvili equation.
89 Critical depth is obtained from weakly nonlinear analysis for a two-layer system where internal
90 solitary waves do not exist and corresponds to a conjugate flow [Lamb, 1998] [Tsuji and Oikawa,
91 2007] [Nakayama et al.,2012]. Oikawa and Tsuji [2006] showed that as the amplification of internal
92 solitary waves in the region where two internal solitary waves propagate in different directions and
93 cross each other decreases, the critical depth corresponds to the depth where a conjugate flow

94 appears. To analyze such a strongly nonlinear effect, higher order equations for internal waves are
95 required. For example, Lamb [1998], Nakayama [2006] and Nakayama and Imberger [2010]
96 demonstrated that a three-dimensional numerical model using a high-resolution mesh is useful for
97 the deformation of internal solitary waves. However, the computational cost is too expensive to
98 analyze the interaction of two internal solitary waves. Therefore, vertically integrated model may be
99 applied to solve internal solitary wave interactions [Choi and Camassa, 1999] [Horn et al., 2000]
100 [Horn et al., 2002]. For example, Choi and Camassa [1999] introduced higher order equations for
101 internal waves, but it is needed to be extended to a horizontally two-dimensional system, such as a
102 Kadomtsev-Petviashvili equation.

103 Nakayama and Kakinuma [2010] developed the Fully-nonlinear and strongly-Dispersive
104 Internal wave equations in a 2 layer system (FDI-2s equations), which can be applied to a
105 horizontally two-dimensional system without assuming a weak-nonlinearity along the perpendicular
106 direction to the progress direction, such as a Kadomtsev-Petviashvili equation. Thus, we apply the
107 FDI-2s equations to investigate the interaction of two internal solitary waves due to soliton
108 resonance in a two-layer system. Firstly, we investigate the applicability of the FDI-2s equations for
109 large amplitude internal solitary waves by comparing with laboratory experiments by Koop and
110 Butler [1981]. Also, the FDI-2s equations are applied to reproduce deformation of internal waves by
111 using laboratory experiments [Horn et al., 2001] [Horn et al., 2002]. Finally, we investigated the
112 interaction of two internal solitary waves by giving the total 20 different conditions regarding an
113 initial amplitude and an incident angle (**Fig. 1** and **Table 1**).

114

115 **2. Methods**

116 **2.1 Fully-nonlinear and strongly-dispersive internal wave equations in a two-layer system**

117 We consider waves propagating in a stable two-layer inviscid fluid at rest as shown in **Fig. 2**
118 where two-layers are indicated as $i = 1$ and $i = 2$ from top to bottom. The flow is assumed to be
119 incompressible. The depth and density of each layer is indicated by h_i and ρ_i , respectively, with $\rho_1 <$
120 ρ_2 . By assuming irrotational flow, the velocity potential ϕ_i is introduced as;

$$121 \quad \mathbf{u}_i = \nabla\phi_i \quad \text{and} \quad w_i = \frac{\partial\phi_i}{\partial z}, \quad (1)$$

$$122 \quad \nabla = \left(\frac{\partial}{\partial x}, \frac{\partial}{\partial y} \right), \quad (2)$$

123 where, \mathbf{u}_i is the horizontal velocity vector for the layer i , and w_i is the vertical velocity for the layer i .

124 The functional for the variational problem is obtained by adding terms for interfacial pressure
125 into the variational method by Luke [1967] and disregarding vorticity terms

$$126 \quad F_i[\phi_i, \zeta_{i,j}] = \int_{t_0}^{t_1} \iint_A \int_{\zeta_{i,0}}^{\zeta_{i,1}} \left\{ \frac{\partial\phi_i}{\partial t} + \frac{1}{2}(\nabla\phi_i)^2 + \frac{1}{2}\left(\frac{\partial\phi_i}{\partial z}\right)^2 + gz + \frac{P_{i-j} + P_i}{\rho_i} \right\} dz dA dt, \quad (3)$$

$$127 \quad P_i = \sum_{k=1}^{i-1} (\rho_i - \rho_k) g h_k, \quad (4)$$

128 where g is the gravitational acceleration, A is the orthogonal projection of the volume occupied by
129 the fluid onto the xy plane, and $\zeta_{i,1}$ and $\zeta_{i,0}$ are the interfacial displacement of the upper and lower
130 interface for layer i .

131 In order to derive a set of two-dimensional horizontal equations, the velocity potential is
 132 expanded into the sum of $Z_{i,\alpha}$ multiplied by their weightings $f_{i,\alpha}$ by following Isobe [1995].

$$133 \quad \phi_i(x, y, z, t) = \sum_{\alpha=0}^{N-1} Z_{i,\alpha}(z, h_i(x, y, t)) f_{i,\alpha}(x, y, t). \quad (5)$$

134 After substituting (5) into (3) and integrating (3) vertically, the Euler-Lagrange equations are
 135 obtained by applying the variational principle [Isobe, 1995]. For our study of internal waves, we
 136 assume the displacement of the water surface is zero which simplifies the model. Following Isobe
 137 [1995], the vertically distributed function, $Z_{i,\alpha}$, is determined by

$$138 \quad Z_{i,\alpha} = z^\alpha. \quad (6)$$

139 Finally, the equations for fully-nonlinear and strongly-dispersive internal wave equations are:

140 [1-layer]

$$141 \quad \zeta^\alpha \frac{\partial \zeta}{\partial t} + \frac{1}{\alpha + \beta + 1} \nabla \left(\zeta^{\alpha + \beta + 1} \nabla f_{1,\beta} \right) - \frac{\alpha \beta}{\alpha + \beta - 1} \zeta^{\alpha + \beta - 1} f_{1,\beta} = 0, \quad (7)$$

$$142 \quad \zeta^\beta \frac{\partial f_{1,\beta}}{\partial t} + \frac{1}{2} \zeta^{\beta + \gamma} \nabla f_{1,\beta} \nabla f_{1,\gamma} + \frac{\beta \gamma}{2} \zeta^{\beta + \gamma - 2} f_{1,\beta} f_{1,\gamma} + g \zeta + \frac{p_1}{\rho_1} = 0, \quad (8)$$

143 [2-layer]

$$144 \quad \zeta^\alpha \frac{\partial \zeta}{\partial t} + \frac{1}{\alpha + \beta + 1} \nabla \left\{ \left(\zeta^{\alpha + \beta + 1} - b^{\alpha + \beta + 1} \right) \nabla f_{2,\beta} \right\} - \frac{\alpha \beta}{\alpha + \beta - 1} \left(\zeta^{\alpha + \beta - 1} - b^{\alpha + \beta - 1} \right) f_{2,\beta} = 0, \quad (9)$$

$$145 \quad \zeta^\beta \frac{\partial f_{2,\beta}}{\partial t} + \frac{1}{2} \zeta^{\beta + \gamma} \nabla f_{2,\beta} \nabla f_{2,\gamma} + \frac{\beta \gamma}{2} \zeta^{\beta + \gamma - 2} f_{2,\beta} f_{2,\gamma} + g \zeta + \frac{p_1 + (\rho_2 - \rho_1) g h_1}{\rho_1} = 0, \quad (10)$$

146 where, for α and β the summation convention is applied.

147 From now on, we call the Fully-nonlinear and strongly-Dispersive Internal wave equations (7) to
148 (10), the FDI-2s equations. It should be noted that the FDI-2s can be extended to a multi-layer
149 system based on the Euler-Lagrange equations, the Fully-nonlinear and strongly-Dispersive Internal
150 wave equations in a multi-layer system (FDI-MLS equations) [Nakayama and Kakinuma, 2010].

151

152 **2.2 The 3rd order theoretical solution for an internal solitary wave and performance** 153 **evaluation**

154 In the numerical simulations it was necessary to specify a large amplitude internal solitary wave,
155 which progresses with little deformation, as an initial condition. When the KdV theoretical solution
156 is used as an initial large amplitude internal solitary waves and the FDI-2s equations are used in the
157 computation, small-amplitude high-frequency internal waves are likely to occur due to the
158 adjustment of the initial approximate wave which results in a decrease in the amplitude of the
159 internal solitary wave [Lamb, 2002] [Nakayama, 2006]. Therefore, as an initial condition we used a
160 3rd order theoretical solution for the internal solitary wave in a two-layer fluid, which was obtained
161 by simplifying the 9th order solution of Mirie and Pennell [1989] (see APPENDIX A).

162 We investigated the characteristics of the 3rd order theoretical solutions and the FDI-2s
163 equations for the analysis of internal solitary waves based on the laboratory experiments of Koop
164 and Butler [1981], which showed the relationship between the amplitude, a_0 , and effective
165 wavelength, λ_l , of internal solitary waves (**Fig. 3**). The amplitude and effective wavelength of the

166 FDI-2s equations were obtained by conducting numerical computations with the initial condition of
167 the 3rd order theoretical solutions. We applied the two-layer shallow water configuration used by
168 Koop and Butler which used $h_1 = 0.06948$ m, $h_2 = 0.01366$ m and $\rho_1 / \rho_2 = 0.63$. The mesh
169 grid interval, $\Delta x = 0.004$ m, the time step, $\Delta t = 0.00005$ s were used in the numerical computations.
170 The thin solid lines envelope the measurement plots by Koop and Butler [1981] for the laboratory
171 experiments of $\rho_1 / \rho_2 = 0.63$ and $h_1 / h_2 = 5.09$. The 3rd order theoretical solutions are found to
172 agree better with the laboratory results than the KdV theoretical solutions. The FDI-2s equations
173 agree with the 3rd order theoretical solutions up to $a_0 / h_2 = 0.25$, and then λ_I / h_2 tends to be larger
174 than the 3rd order theoretical solutions, which agrees with the fully-nonlinear solutions obtained
175 using numerical computations by Grue *et al.* [1997] in FIGURE 5 of Choi and Camassa [1999].

176 In order to demonstrate the applicability of the FDI-2s equations for the deformation of internal
177 solitary waves, the FDI-2s equations were applied to the laboratory experiment of Horn *et al.* [2000]
178 [2002]. The length, width and height of their tilting tank were 6.0 m, 0.30 m and 0.29 m,
179 respectively. For the laboratory experiment, $h_1 = 0.232$ m, $h_2 = 0.058$ m, $\rho_2 - \rho_1 = 20.0$ kg m⁻³,
180 and a tilting angle = 0.5° were chosen (**Fig. 4(a)**). The interfacial thickness was less than 0.01 m,
181 which provides a two-layer-like system. The total mechanical energy (kinetic + potential) due to a
182 tilting density interface decreases during the deformation from the internal seiche to a train of
183 internal solitary waves due to viscous losses [Horn *et al.*, 2000 and 2002]. Therefore, energy
184 dissipation due to viscous losses at the interface and the boundaries was added to the FDI-2s
185 equations. As a result, the interfacial displacement from the FDI-2s equations agrees very well with

186 Horn *et al.*'s laboratory experiment, thereby confirming the robustness of the FDI-2s equations for
187 analyzing the excitation of internal solitary waves (**Fig. 4(b)**).

188 **2.3 Boundary conditions**

189 To analyze the two-dimensional interaction of internal solitary waves, we adopt a finite
190 difference method for a type of domain used by Funakoshi [1980] (**Fig. 1**). It is necessary to resolve
191 the zero momentum boundary condition for the oblique boundary condition in the computational
192 domain (**Fig. 1**). Simanjuntak *et al.* [2009] introduced a numerical computation technique whereby
193 zero normal velocity boundary conditions can be successfully applied to reproduce internal wave
194 reflections in a stratified flow field by comparing with analytical solutions. Since a velocity potential
195 is used in this study, it is not possible to directly apply the zero normal velocity boundary condition.
196 Therefore, in this study we propose a new technique for zero normal velocity boundary conditions
197 by following Simanjuntak *et al.* [2009]. The intersection point of the oblique boundary line and the
198 perpendicular line to the oblique boundary from a node outside of the computational domain is
199 defined as (x', y') (**Fig. 5**). The unknown velocity potential, $\phi_{i,j}$, is determined by applying the
200 Galerkin method using the known velocity potentials, $\phi_{i-1,j}$, $\phi_{i,j+1}$ and $\phi_{i-1,j+1}$, so as to satisfy the zero
201 normal velocity boundary conditions shown in (11) on the oblique boundary (see APPENDIX B).

$$202 \quad \frac{\partial \phi'_{i,j}}{\partial \mathbf{n}} = 0. \quad (11)$$

203

204 3. Results

205 3.1 Computational conditions and critical incident angle of an internal solitary wave

206 Two different amplitudes of initial internal solitary waves A and B were used in the simulations.
207 The common computational conditions of the numerical analysis were that the upper layer depth was
208 0.20 m, the lower layer depth was 0.80 m, and the ratio of the density between the upper and lower
209 layers was 0.5 ($h_2 / h_1 = 4.00$), which is a similar set up to that of Koop and Butler [1981], (ρ_1 / ρ_2
210 $= 0.63$ and $h_1 / h_2 = 5.09$), though the depth of the layers is reversed. The small and large
211 normalized amplitudes of the initial internal solitary waves, a_0 / h_2 , were 0.01 and 0.05, are 4 % and
212 20 % of the upper layer depth, respectively (**Fig. 6**). In the large 0.04 m ($a_0 / h_2 = 0.05$) amplitude
213 case, the 3rd order internal solitary wave solution was confirmed to have wider wavelength compared
214 to the KdV theory, which has been confirmed in a previous study [Nakayama, 2006] (also see **Fig.**
215 **3**).

216 To predict an amplification factor, the definition of parameter κ comes from Yeh *et al.* [2010]
217 who modified Miles' result.

$$218 \quad \kappa = \frac{\tan(\varphi)}{\sqrt{3p \frac{a_0}{h_2} \cos(\varphi)}} = \frac{\tan(\varphi)}{\tan(\varphi_c) \cos(\varphi)} \quad (12)$$

219 where a_0 is the amplitude of incident internal solitary wave (see APPENDIX C for details, including
220 the definition of parameter, p , (C2)) and φ_c is the critical incident angle by Miles [1977].

221 Finally, the amplification factor, a_f , can be obtained as

$$222 \quad a_f = \begin{cases} (1+\kappa)^2 & \kappa < 1 \quad \text{Mach reflection} \\ \frac{4}{1+\sqrt{1-\kappa^{-2}}} & \kappa > 1 \quad \text{regular reflection} \end{cases} \quad (13)$$

223 where a_f is the ratio of the amplitude at the oblique boundary to the amplitude of the incident internal
224 solitary wave.

225 The critical incident angle, φ_c , corresponds to the boundary between Mach and regular reflection.

226 According to Miles [1977], the ratio of the maximum amplitude to the amplitude of the incident

227 internal solitary wave (we call this the maximum amplification factor) occurs when the angle of the

228 incident internal solitary wave is equal to φ_c . As the critical incident angle obtained from the

229 numerical computations, φ_c , is expected to be different from the modified Miles prediction, the

230 critical incident angle obtained from the modified Miles prediction, φ_{c_kp} , is given by

$$231 \quad \frac{\tan(\varphi_{c_kp})}{\cos(\varphi_{c_kp})} = \sqrt{3p \frac{a_0}{h_2}}. \quad (14)$$

232 In the small amplitude case, φ_{c_kp} obtained from (14) was 14.4 degrees, while it was 27.7

233 degrees for the large amplitude case. Therefore, for the small amplitude case, nine different incident

234 angles were given corresponding to cases A1 to A9: 10, 11, 12, 12.5, 13, 14, 15, 20 and 30 degrees

235 (**Table 1**). On the other hand, for the large amplitude case eleven different incident angles were

236 given corresponding to cases B1 to B11: 10, 12, 14, 16, 18, 20, 23, 26, 28, 30 and 40 degrees (**Table**

237 1). Since the very small time step is required due to the use of the variational principle, linear theory
238 shows that the celerity corresponds to the Courant-Friedrichs-Lewy condition, 0.00221. Although
239 parallel computation was conducted using 12 CPUs using openMP, it took about 2500 s for the
240 amplitude of a stem to reach the maximum amplitude in case A4, which was the most expensive
241 runtime cost case, and it is necessary to prepare $7,200 \times 1,500 = 10,800,000$ meshes in the direction
242 of progress, leading to a runtime cost that was too expensive. Therefore, we carried out actual
243 computations only in the effective computational domains in order to reduce the runtime cost (**Fig.**
244 **7**). The left boundary of the effective computational domain had a sponge layer to reduce the internal
245 wave energy, and perfect reflection conditions were specified at the top boundary in order to sustain
246 the internal solitary wave energy during its progression. **Fig. 7** demonstrates that the stem was
247 formed due to reflection from the oblique boundary.

248

249 **3.2 Stem length and wave amplitude**

250 Stem formation was investigated for all cases in order to clarify the influence of the incident
251 angle on the development of the stem length (**Fig. 8**). Since φ_{c_kp} were 14.4 and 27.7 degrees for
252 case A and case B, it was expected that a stem would be formed in cases A1 to A6 (incident angles
253 between 10 and 14 degrees) and in cases B1 to B8 (incident angles between 10 and 26 degrees). A
254 stem was formed in cases A1 to A6 and cases B1 to B8. However, the stem length reached a steady
255 state in cases A5, A6, B7 and B8, although the stem length should keep increasing if the stem is due

256 to Mach reflection. Therefore, the crests in cases A5, A6, B7 and B8 are considered to occur due to
257 regular reflection. In contrast, the cases from A1 to A4 and cases from B1 to B6 are considered to
258 have a ‘stem’ due to Mach reflection because the stem length increased linearly in time.

259 Previous studies have found that the larger the amplitude of an incident internal solitary wave,
260 the faster the stem extends, which was investigated here by comparison of the same incident angle
261 between cases A1 and B1, and cases A3 and B2. The extension speed of a stem under case B1 with
262 an incident angle of 10 degrees was 2.52 times as fast as under case A1 when $t / (h_2 / c_0) = 553$. The
263 stem extension speed under case B2 was 2.91 times as fast as under case A3 for an incident angle of
264 12 degrees over $t / (h_2 / c_0) = 553$. Therefore, we confirmed that the stem extension speed increases
265 with increasing amplitude of the incident internal solitary wave when other conditions are the same.

266 The time taken to reach the maximum amplitude due to the internal solitary wave interaction
267 was plotted against each incident angle of the internal solitary wave (**Fig. 9**). The duration for the
268 large amplitude case was shorter than the small amplitude case. The maximum duration for each
269 small and large amplitude case appeared in case A4 and B6 when φ_c was less than φ_{c_kp} shown in
270 (14). Interestingly, when a stem exists, cases A4 and B6 correspond to the maximum incident angle
271 cases, and the smaller the incident angle, the faster the stem extension speed. In contrast, when
272 incident angles are closer to φ_c , amplification factors and durations become larger and longer until
273 the stem reaches a stationary state. Therefore, cases A4 and B6 took the longest to reach the
274 maximum amplitude for the small and large amplitude cases, respectively.

275 The amplification factor obtained from numerical simulations were investigated using the KP
276 theory ((14) and **Fig. 10**). The small amplitude case showed a maximum amplification factor of
277 about 3.4, which agrees with previous studies [Funakoshi, 1980] and is smaller than the maximum
278 value (= 4.0). For the large amplitude case the maximum amplification was found to be about 3.0,
279 which is similar to previous studies [Tanaka, 1993] [Yeh et al., 2010] related to the interaction of
280 large amplitude surface solitary waves. For smaller incident angles the amplification factor agrees
281 well with the predictions of KP theory (**Fig. 10**). Interestingly, cases A5, A6, B7 and B8, in which
282 the crest reaches steady state, are found to be located between the maximum amplification factor
283 case and φ_{c_kp} , which is categorized as a regular reflection. From **Figs. 8-10**, it can be seen that the
284 maximum amplification occurred when the incident angles were less than φ_{c_kp} (cases A4 and B6)
285 and when the time taken to reach the maximum amplitude was longest for the small and large
286 amplitude cases, respectively.

287

288 **4. Discussion**

289 **4.1 Limiting wave amplitude of soliton resonance**

290 Li *et al.* [2011] and Tanaka [1993] found from experimental and numerical results for surface
291 waves that amplification was suppressed when the amplitude of an initial surface wave was
292 relatively large, which corresponds to a large φ_{c_kp} . In particular, when the amplitude of an incident
293 surface wave is large, the maximum amplification factor has been found to be less than 3.0, based on

294 numerical computations by Tanaka [1993]. Therefore, although this suppression close to φ_{c_kp} has
295 been seen in surface waves in previous studies, the effect of large amplitudes may also be inherent to
296 internal waves.

297 There may be another possible explanation for the suppression of amplification. Weakly
298 nonlinear analysis for a two-layer system yields a critical depth where internal solitary waves do not
299 exist because the nonlinearity vanishes and dispersion prevails [Lamb, 1998] [Tsuji and Oikawa,
300 2007] [Nakayama et al., 2012]. Therefore, in a two-layer system, the critical depth from the water
301 surface is given by (15).

$$302 \quad h_c = h_1 + a_c = \frac{\sqrt{\rho_1}(h_1 + h_2)}{\sqrt{\rho_1} + \sqrt{\rho_2}}. \quad (15)$$

303 where h_c is the critical depth from the water surface, and a_c is the maximum possible wave
304 amplitude.

305 Tsuji and Oikawa [2007] demonstrated that resonance is suppressed when the initial density
306 interfacial level is close to the critical depth by using the Extended Kadomtsev-Petviashvili equation.
307 If the height of the stem of resonance is equal to the distance between the critical depth and the
308 interface at rest, the corresponding amplification factors are 26.75 and 5.35 for the small and large
309 amplitude cases, respectively. Therefore, since the density interface of the amplified internal solitary
310 wave was closer to the critical depth for the large amplitude case compared to the small amplitude

311 case, the amplification due to resonance of two internal solitary waves may be suppressed for the
312 large amplitude case.

313

314 **4.2 Characteristics of stem**

315 In previous studies, the stem induced by resonance was investigated by assuming that the stem is
316 an internal solitary wave. To confirm whether the stem is an internal solitary wave or not, the shape
317 of the stem was compared to the 3rd order theoretical solutions (**Fig. 11**). The largest amplitude case,
318 case B6, was selected and compared to the 3rd order theoretical solutions, showing very good
319 agreement with slightly larger effective wavelength of the FDI-2s equations, which shows the same
320 tendency when a_0 / h_2 is about 0.6 in **Fig. 3**. The normalized celerity by linear theory was 1.573
321 while the celerity of the incident internal solitary wave was 1.356. Therefore, a stem has the
322 potential to be an internal solitary wave from the perspective of the shape of the density interface. If
323 a stem is an internal solitary wave, it progresses without having any decay or deformation. We thus
324 made an attempt to carry out one-dimensional numerical computations using the FDI-2s equations
325 specifying the shape of the density interface displacement and the velocity potential of the stem from
326 cases A1, A5, B1 and B6 (**Fig. 12**). For case B6 there was a slight decrease in amplitude due to the
327 formation of high-frequency internal waves, which follow the stem. However, the decrease in
328 amplitude was negligible, and all cases kept the shape of the original stem and waves propagated

329 with speeds of an internal solitary wave, which demonstrates that stems induced by the resonance of
330 internal solitary waves have the characteristics of an internal solitary wave.

331

332 **5. Conclusion**

333 The oblique reflection of an internal solitary wave in a two-layer system has been studied using
334 the FDI-2s equations. For the small amplitude case: $a_0/h_2 = 0.01$, the maximum amplification factor
335 was found to be about 3.4. The amplification factor followed (14) in the region where a Mach stem
336 occurred and the amplification factor was less than the Miles prediction, 4. The critical incident
337 angle obtained from the numerical computations was confirmed to be equal to the critical incident
338 angle obtained from the modified Miles prediction, φ_{c_kp} . The maximum amplification factor
339 reached about 3.0 when the amplitude of the initial internal solitary wave was large (large amplitude
340 case: $a_0/h_2 = 0.05$). It may thus be expected that the larger φ_{c_kp} is, the smaller the amplification
341 factor, which is expected based on the experimental and numerical results by Li *et al.* [2011] and
342 Tanaka [1993] for surface waves. However, there is the possibility that a maximum possible wave
343 amplitude exists when the density interfacial level is close to the critical depth.

344

345 **6. Author contribution**

346 K. Nakayama designed the all numerical computations and wrote most of the paper and
347 performed theoretical analysis. T. Kakinuma and H. Tsuji discussed about the numerical

348 computational results with K. Nakayama. All authors read and commented on drafts of this paper.

349

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356 Urban Safety and Security, Kobe University.

357

358 **APPENDIX A**

359 The 3rd order equations for an internal solitary wave are obtained by using the 9th order internal
 360 solitary wave equations [Mirie and Pennell, 1989].

361 $h = -h_1 + \zeta$ (A1)

362 $\zeta / h_2 = \varepsilon A_{11} S + \varepsilon^2 (A_{21} S + A_{22} S^2) + \varepsilon^3 (A_{31} S + A_{32} S^2 + A_{33} S^3)$ (A2)

363 $S = \text{sech}^2 X$ (A3)

364 $X = \frac{\sqrt{3K\varepsilon}}{2h_2} (x - x_0 - C_R t)$ (A4)

365 $C_R = \sqrt{gh_0} C_0 (1 + \varepsilon C_1 + \varepsilon^2 C_2 + \varepsilon^3 C_3)$ (A5)

366 $C_0 = \sqrt{\frac{1 - \sigma}{1 + \sigma / \gamma}}$ (A6)

367 $\varepsilon = a_0 / h_2$ (A7)

368 $A_{11} = \begin{cases} 1 & \sigma < \gamma^2 \\ -1 & \sigma > \gamma^2 \end{cases}$ (A8)

369 $A_{21} = \frac{1}{K(1 + \sigma\gamma)} \left[-\frac{1}{2} K^2 A_{11} (1 + \sigma\gamma^3) - K(1 - \sigma) + 2A_{11} (1 + \sigma / \gamma^3) \right]$ (A9)

370 $A_{22} = \frac{1}{2K(1 + \sigma\gamma)} \left[\frac{3}{2} K^2 A_{11} (1 + \sigma\gamma^3) + 2K(1 - \sigma) - 2A_{11} (1 + \sigma / \gamma^3) \right]$ (A10)

371 $A_{31} = \frac{1}{K(1 + \sigma\gamma)} \left[-\frac{K^2}{5} (1 - \sigma\gamma^2) + \frac{2}{3} K A_{11} (1 + \sigma / \gamma) - \frac{8}{3} (1 - \sigma / \gamma^4) - K A_{21} A_{11} (1 - \sigma) + 4A_{21} (1 + \sigma / \gamma^3) \right]$

$$\begin{aligned}
372 \quad & -2KA_{21}C_1(1+\sigma\gamma) - \frac{K^2}{30}A_{22}(1+\sigma\gamma^3) - \frac{2}{3}KA_{22}A_{11}(1-\sigma) + 2A_{22}(1+\sigma/\gamma^3) - KA_{22}C_1(1+\sigma\gamma) \\
373 \quad & + \frac{4}{3}A_{22}C_2(1+\sigma/\gamma) - K^2A_{11}C_1(1+\sigma\gamma^3) - 2KC_1(1-\sigma) + 4A_{11}C_1(1+\sigma/\gamma^3) \Big] \quad (A11)
\end{aligned}$$

$$\begin{aligned}
374 \quad & A_{32} = \frac{1}{K(1+\sigma\gamma)} \left[\frac{3}{5}K^3A_{11}(1+\sigma\gamma^5) - \frac{K^2}{5}(1-\sigma\gamma^2) - \frac{K}{3}A_{11}(1+\sigma/\gamma) + \frac{3}{4}K^2A_{21}(1+\sigma\gamma^3) + 2KA_{21}A_{11}(1-\sigma) \right. \\
375 \quad & \left. - 3A_{21}(1+\sigma/\gamma^3) - \frac{21}{20}K^2A_{22}(1+\sigma\gamma^3) + \frac{K}{3}A_{22}A_{11}(1-\sigma) - A_{22}(1+\sigma/\gamma^3) + \frac{3}{2}K^2A_{11}C_1(1+\sigma\gamma^3) + 2KC_1(1-\sigma) \right. \\
376 \quad & \left. - 2A_{11}C_1(1+\sigma/\gamma^3) - 2KA_{22}C_1(1+\sigma\gamma) + \frac{4}{3}(1-\sigma/\gamma^4) \right] \quad (A12)
\end{aligned}$$

$$\begin{aligned}
377 \quad & A_{33} = \frac{1}{K(1+\sigma\gamma)} \left[-\frac{3}{5}K^3A_{11}(1+\sigma\gamma^5) + \frac{9}{20}K^2(1-\sigma\gamma^2) - \frac{K}{3}A_{11}(1+\sigma/\gamma) + \frac{1}{3}(1-\sigma/\gamma^4) + \frac{31}{20}K^2A_{22}(1+\sigma\gamma^3) \right. \\
378 \quad & \left. + \frac{4}{3}KA_{22}A_{11}(1-\sigma) - A_{22}(1+\sigma/\gamma^3) \right] \quad (A13)
\end{aligned}$$

$$379 \quad K = \frac{1-\sigma/\gamma^2}{1+\sigma\gamma} A_{11} \quad (A14)$$

$$380 \quad C_1 = \frac{1-\sigma/\gamma^2}{2(1+\sigma/\gamma)} A_{11} \quad (A15)$$

$$381 \quad C_2 = \frac{1}{2(1+\sigma/\gamma)} \left[\frac{K^2}{5}(1+\sigma\gamma^3) + 3(1+\sigma/\gamma)C_1^2 \right] \quad (A16)$$

$$\begin{aligned}
382 \quad & C_3 = \frac{1}{2(1+\sigma/\gamma)} \left[\frac{2}{35}K^3(1+\sigma\gamma^5) + \left(\frac{K^2}{5}A_{21}A_{11} + \frac{2K^2}{5}C_1 \right) (1+\sigma\gamma^3) + (2KA_{21}A_{11}C_1 + KC_1^2 + 2KC_2) \right. \\
383 \quad & \left. (1+\sigma\gamma) - (A_{21}A_{11}C_1^2 + 2A_{21}A_{11}C_2 + 2C_1C_2)(1+\sigma/\gamma) \right] \quad (A17)
\end{aligned}$$

384 where, a_0 is the amplitude of incident internal wave.

385

386 **APPENDIX B**

387 Velocity potential inside of a mesh is given by (B1) using the Galerkin method.

$$\begin{aligned}
 388 \quad \phi = & \phi_{i-1,j} \frac{x-x_{i-1,j}}{\Delta x} \frac{y-y_{i-1,j}}{\Delta y} + \phi_{i,j} \frac{x_{i,j}-x}{\Delta x} \frac{y-y_{i,j}}{\Delta y} \\
 389 \quad & + \phi_{i-1,j+1} \frac{x-x_{i-1,j+1}}{\Delta x} \frac{y_{i-1,j+1}-y}{\Delta y} + \phi_{i,j+1} \frac{x_{i,j+1}-x}{\Delta x} \frac{y_{i,j+1}-y}{\Delta y} . \quad (B1)
 \end{aligned}$$

390 Zero normal velocity boundary condition is given as (B2).

$$391 \quad \frac{\partial \phi'_{i,j}}{\partial \mathbf{n}} = \frac{\partial x}{\partial \mathbf{n}} \frac{\partial \phi'_{i,j}}{\partial x} + \frac{\partial y}{\partial \mathbf{n}} \frac{\partial \phi'_{i,j}}{\partial y} = 0, \quad (B2)$$

$$392 \quad \frac{\partial \phi'_{i,j}}{\partial x} = \phi_{i-1,j} \frac{1}{\Delta x} \frac{y'-y_{i-1,j}}{\Delta y} - \phi_{i,j} \frac{1}{\Delta x} \frac{y'-y_{i,j}}{\Delta y} + \phi_{i-1,j+1} \frac{1}{\Delta x} \frac{y_{i-1,j+1}-y'}{\Delta y} - \phi_{i,j+1} \frac{1}{\Delta x} \frac{y_{i,j+1}-y'}{\Delta y}, \quad (B3)$$

$$393 \quad \frac{\partial \phi'_{i,j}}{\partial y} = \phi_{i-1,j} \frac{x'-x_{i-1,j}}{\Delta x} \frac{1}{\Delta y} + \phi_{i,j} \frac{x_{i,j}-x'}{\Delta x} \frac{1}{\Delta y} - \phi_{i-1,j+1} \frac{x'-x_{i-1,j+1}}{\Delta x} \frac{1}{\Delta y} - \phi_{i,j+1} \frac{x_{i,j+1}-x'}{\Delta x} \frac{1}{\Delta y}. \quad (B4)$$

394 Therefore, zero normal velocity boundary condition (B2) is rewritten as (B5).

$$\begin{aligned}
 395 \quad \frac{\partial \phi'_{i,j}}{\partial \mathbf{n}} = & \frac{\partial x}{\partial \mathbf{n}} \left[\phi_{i-1,j} \frac{1}{\Delta x} \frac{y'-y_{i-1,j}}{\Delta y} - \phi_{i,j} \frac{1}{\Delta x} \frac{y'-y_{i,j}}{\Delta y} + \phi_{i-1,j+1} \frac{1}{\Delta x} \frac{y_{i-1,j+1}-y'}{\Delta y} - \phi_{i,j+1} \frac{1}{\Delta x} \frac{y_{i,j+1}-y'}{\Delta y} \right] \\
 396 \quad & + \frac{\partial y}{\partial \mathbf{n}} \left[\phi_{i-1,j} \frac{x'-x_{i-1,j}}{\Delta x} \frac{1}{\Delta y} + \phi_{i,j} \frac{x_{i,j}-x'}{\Delta x} \frac{1}{\Delta y} - \phi_{i-1,j+1} \frac{x'-x_{i-1,j+1}}{\Delta x} \frac{1}{\Delta y} - \phi_{i,j+1} \frac{x_{i,j+1}-x'}{\Delta x} \frac{1}{\Delta y} \right] = 0. \quad (B5)
 \end{aligned}$$

397 Finally, the unknown velocity potential, $\phi_{i,j}$, is determined from the known velocity potentials,

398 $\phi_{i-1,j}$, $\phi_{i,j+1}$ and $\phi_{i-1,j+1}$.

$$\begin{aligned}
 399 \quad \phi_{i,j} = & \left\{ \phi_{i-1,j} \left[(y'-y_{i-1,j}) \frac{\partial x}{\partial \mathbf{n}} + (x'-x_{i-1,j}) \frac{\partial y}{\partial \mathbf{n}} \right] + \phi_{i-1,j+1} \left[(y_{i-1,j+1}-y') \frac{\partial x}{\partial \mathbf{n}} - (x'-x_{i-1,j+1}) \frac{\partial y}{\partial \mathbf{n}} \right] \right. \\
 400 \quad & \left. - \phi_{i,j+1} \left[(y_{i,j+1}-y') \frac{\partial x}{\partial \mathbf{n}} + (x_{i,j+1}-x') \frac{\partial y}{\partial \mathbf{n}} \right] \right\} / \left[(y'-y_{i,j}) \frac{\partial x}{\partial \mathbf{n}} - (x'-x_{i,j}) \frac{\partial y}{\partial \mathbf{n}} \right]. \quad (B6)
 \end{aligned}$$

401

402

403 **APPENDIX C**

404 Here we describe the results of the KP equation and their modification for comparison to our
 405 numerical results for internal waves in a two-layer fluid system with a rigid lid shown in **Fig. 2**. It is
 406 also assumed that the interface is not near the critical depth. Details of derivation of the equations are
 407 omitted and only the results are described.

408 The KdV equation for waves propagating in the direction $\mathbf{n} = (\cos \varphi, \sin \varphi)$ in this system is
 409 written in the physical coordinates as

$$410 \quad \frac{\partial \zeta}{\partial t} + V \frac{\partial \zeta}{\partial \chi} - \frac{3Vp}{2h_2} \zeta \frac{\partial \zeta}{\partial \chi} + \frac{Vh_2^2 q}{6} \frac{\partial^3 \zeta}{\partial \chi^3} = 0 \quad (C1)$$

411 where, ζ is the displacement of the interface, $\chi = \mathbf{n} \cdot \mathbf{x} = x \cos \varphi + y \sin \varphi$ ($\mathbf{x} = (x, y)$ the position
 412 vector in a horizontal plane), t the time. The constants p and q are given by

$$413 \quad p = \frac{\sigma - \gamma^2}{\gamma(\gamma + \sigma)} \quad (C2)$$

$$414 \quad q = \frac{\gamma(1 + \gamma\sigma)}{\gamma + \sigma} \quad (C3)$$

$$415 \quad \gamma = \frac{h_1}{h_2} \quad (C4)$$

$$416 \quad \sigma = \frac{\rho_1}{\rho_2} \quad (C5)$$

417 The solitary wave solution of the KdV equation (C1) is given by

$$418 \quad \zeta = -a_0 \operatorname{sech}^2 \left\{ \sqrt{\frac{3pa_0}{4qh_2^3}} \left[\chi - \left(1 + \frac{pa_0}{2h_2} \right) Vt - \chi_0 \right] \right\} \quad (C6)$$

$$419 \quad V = \sqrt{gh_2 \gamma \frac{1 - \sigma}{\gamma + \sigma}} \quad (C7)$$

420 where, χ_0 is an arbitrary constant and we consider the case $p > 0$.

421 The KP equation for waves propagating almost in the x direction is in the physical
422 coordinates

$$423 \quad \frac{\partial}{\partial x} \left(\frac{\partial \zeta}{\partial t} + V \frac{\partial \zeta}{\partial x} - \frac{3Vp}{2h^2} \zeta \frac{\partial \zeta}{\partial x} + \frac{Vq}{6} h^2 \frac{\partial^3 \zeta}{\partial x^3} \right) + \frac{V}{2} \frac{\partial^2 \zeta}{\partial y^2} = 0 \quad (C8)$$

424 The solitary wave solution to this equation is

$$425 \quad \zeta = -a_1 \operatorname{sech}^2 \left\{ \sqrt{\frac{3pa_1}{4qh_2^3}} \left[x + y \tan \varphi - V \left(1 + \frac{pa_1}{2h_2} + \frac{1}{2} \tan^2 \varphi \right) t - x_0 \right] \right\} \quad (C9)$$

426 where a_1 is an amplitude and x_0 is an arbitrary constant.

427 Now, transformation of the variables yields

$$428 \quad u = \frac{3p}{2h_2} \zeta \quad (C10)$$

$$429 \quad X = \frac{x - Vt}{\sqrt{qh_2}} \quad (C11)$$

$$430 \quad Y = \frac{y}{\sqrt{qh_2}} \quad (C12)$$

$$431 \quad T = \frac{2Vt}{3\sqrt{qh_2}} \quad (C13)$$

432 The KP equation (C8) is written as

$$433 \quad \frac{\partial}{\partial x} \left(4 \frac{\partial u}{\partial T} - 6u \frac{\partial u}{\partial X} + \frac{\partial^3 u}{\partial X^3} \right) + 3 \frac{\partial^2 u}{\partial Y^2} = 0 \quad (C14)$$

434 As the incident angle decreases in the KP equation, the oblique reflection of the solitary wave
435 due to a rigid wall changes from a regular type to a Mach type at the critical angle. The asymptotic
436 value of the factor of the maximum wave amplitude at the wall to the amplitude of the incident
437 solitary wave approaches four at the critical incident angle. For the KP equation (C14), it is known

438 that the amplification factor [29] can be given by

$$439 \quad \text{amplification factor} = \begin{cases} (1+\kappa)^2 & \kappa < 1 \\ \frac{4}{1+\sqrt{1-\kappa^{-2}}} & \kappa > 1 \end{cases} \quad (\text{C15})$$

$$440 \quad \kappa = \frac{\tan(\varphi)}{\sqrt{3p\frac{a_0}{h_2}}} = \frac{\tan(\varphi)}{\tan(\varphi_c)} \quad (\text{C16})$$

441 For shallow water waves, the equations corresponding to (C14) and (C15) may be called the
 442 Miles' prediction. The Miles' prediction does not agree with the numerical computations of
 443 Funakoshi [1980] and Tanaka [1993], or the experiments of Li *et al.* [2011]. However, Yeh, Li, and
 444 Kodama [2010] [2016] considered an explanation as follows: the KP equation is derived under the
 445 assumption of quasi-two-dimensionality, in which $\alpha = O(\epsilon^{1/2})$ and $\epsilon = O(a_0/h_2) \ll 1$. The solution
 446 (C9) of the KP equation (C8) is rewritten as follows by the use of χ :

$$447 \quad \zeta = -a_1 \text{sech}^2 \left\{ \sqrt{\frac{3pa_1}{4qh_2^3 \cos^2 \varphi}} \left[\chi - V \cos \varphi \left(1 + \frac{pa_1}{2h_2} + \frac{1}{2} \tan^2 \varphi \right) t - \chi_0 \right] \right\} \quad (\text{C17})$$

448 If $\alpha = O(\epsilon^{1/2})$, $\cos \alpha = 1 - (1/2) \tan^2 \varphi + O(\epsilon^2)$ and the velocity of the above solitary wave
 449 solution becomes

$$450 \quad V \cos \varphi \left(1 + \frac{pa_1}{2h_2} + \frac{1}{2} \tan^2 \varphi \right) = V \left(1 + \frac{pa_1}{2h_2} + O(\epsilon^2) \right) \quad (\text{C18})$$

451 Thus, if we define (C19), the solution (C17) approximates to the KdV solution as (C20)

$$452 \quad a_0 = \frac{a_1}{\cos^2 \varphi} = a_1 (1 + \tan^2 \varphi) = a_1 (1 + O(\epsilon)) \quad (\text{C19})$$

$$453 \quad \zeta = -a_0 \text{sech}^2 \left\{ \sqrt{\frac{3pa_0}{4qh_2^3}} \left[\chi - V \left(1 + \frac{pa_0}{2h_2} \right) t - \chi_0 \right] \right\} + O(\epsilon) \quad (\text{C20})$$

454 Therefore, the simulations and experiments should be compared with the Miles' prediction

455 (C15) with

456
$$\kappa = \frac{\tan(\varphi)}{\sqrt{3p \frac{a_0}{h_2} \cos \varphi}} \quad (C21)$$

457 Let us call the Miles' prediction (C15) with (C21) as the modified Miles' prediction. The

458 modified Miles' prediction for the shallow water waves agrees well with the numerical computations

459 and experiments except near the critical incident angle.

460

461

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538

539

540

541 **Figure captions**

542

543 Fig. 1. Schematic horizontal plane view of a computation corresponding to the occurrence of Mach
544 and regular reflection of a soliton with an incident angle of φ . θ indicates the reflection angle.

545

546 Fig. 2. Schematic diagram of a two-layer system. Upper and lower boundaries are considered as
547 rigid walls.

548

549 Fig. 3 Comparisons with the laboratory experiments by Koop and Butler [22]. Thin solid lines
550 indicate the region of the experiments' plots. Dashed and thick solid lines indicate the KdV
551 theoretical solutions and the 3rd order theoretical solutions, respectively. Circles indicate numerical
552 computation results by using the FDI-2s equations.

553

554 Fig. 4 Comparisons with the laboratory experiments by Horn *et al.* [17] [46] [47]. (a) Initial set up.
555 (b) Comparisons of interfacial displacement at the Wavegauge B between the laboratory experiments
556 and the FDI-2s equations.

557

558 Fig. 5. Schematic diagram for satisfying boundary conditions of momentum. Normal velocity to
559 an oblique boundary should be zero.

560

561 Fig. 6. Initial waves. The solid lines shows the 3rd order theoretical solution and dashed line shows the
562 KdV solution. (a) case A. (b) case B.

563

564 Fig. 7. Interfacial displacement of case B5. Each solid line square in the bottom figure indicates the
565 computational region at the times indicated. Top three figures show enlarged progress of internal solitary
566 wave at $t / (h_2 / c_0) = 0, 277$ and 498 .

567

568 Fig. 8. Time series of the length of a stem for case A and case B. Length is normalized by the
569 lower layer depth. (a) cases A1 to A6. (b) cases B1 to B8.

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576 Incident angle and amplification factor vs incident angle.

577

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579 thick solid lines show the interfacial displacement of a stem. The thin solid lines show the 3rd order

580 theoretical solution.

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582 Fig. 12. Progress of internal solitary waves by using the interfacial displacement and velocity
583 potential around a stem. (a) case A1. (b) case A5. (c) case B1. (d) case B6.

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587

588 **Table captions**

589

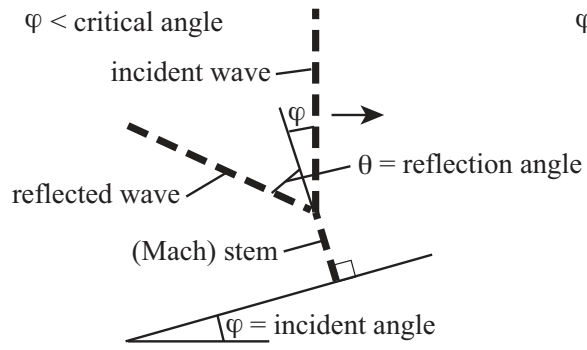
590 Table 1. Computational conditions for small and large amplitude cases.

591

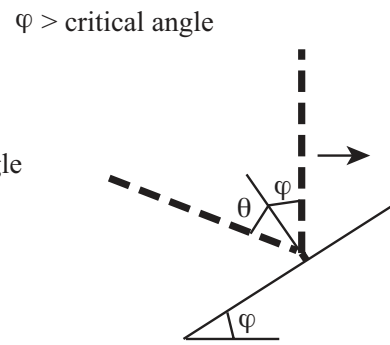
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593

Mach reflection



Regular reflection



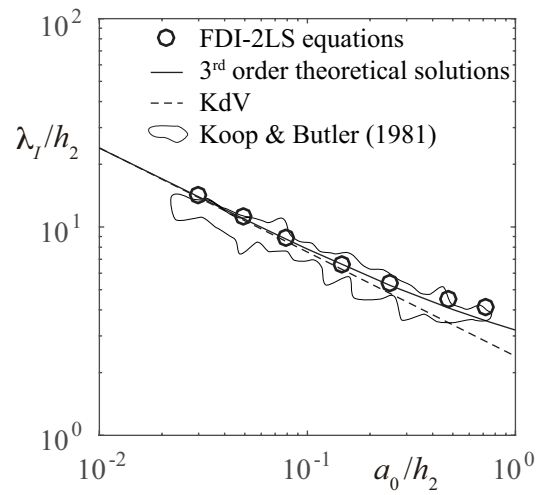
594

595 Fig. 1. Schematic horizontal plane view of a computation corresponding to the occurrence of Mach and

596 regular reflection of a soliton with an incident angle of φ . θ indicates the reflection angle.

597

598

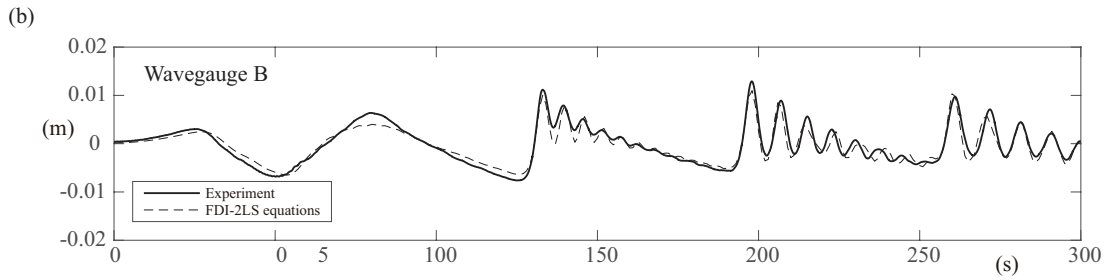
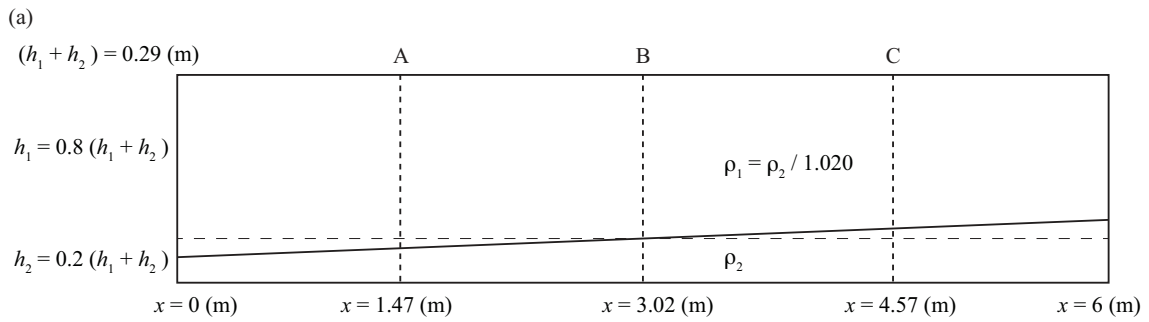


604

605 Fig. 3 Comparisons with the laboratory experiments by Koop and Butler [22]. Thin solid lines
 606 indicate the region of the experiments' plots. Dashed and thick solid lines indicate the KdV
 607 theoretical solutions and the 3rd order theoretical solutions, respectively. Circles indicate numerical
 608 computation results by using the FDI-2s equations.

609

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611

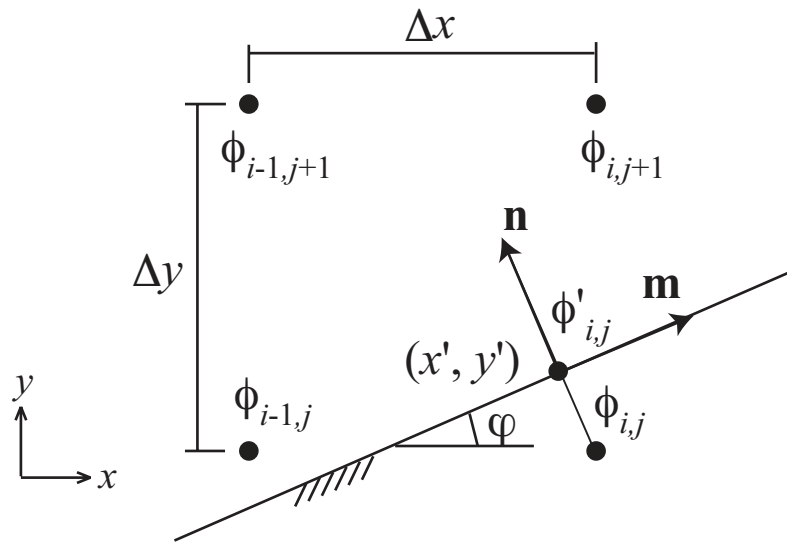
612 Fig. 4 Comparisons with the laboratory experiments by Horn *et al.* [17] [46] [47]. (a) Initial set up.

613 (b) Comparisons of interfacial displacement at the Wavegauge B between the laboratory experiments

614 and the FDI-2s equations.

615

616



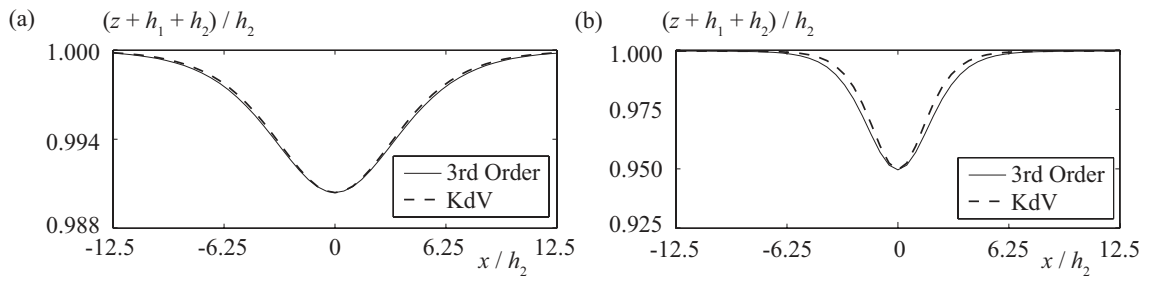
617

618 Fig. 5. Schematic diagram for satisfying boundary conditions of momentum. Normal velocity to

619 an oblique boundary should be zero.

620

621

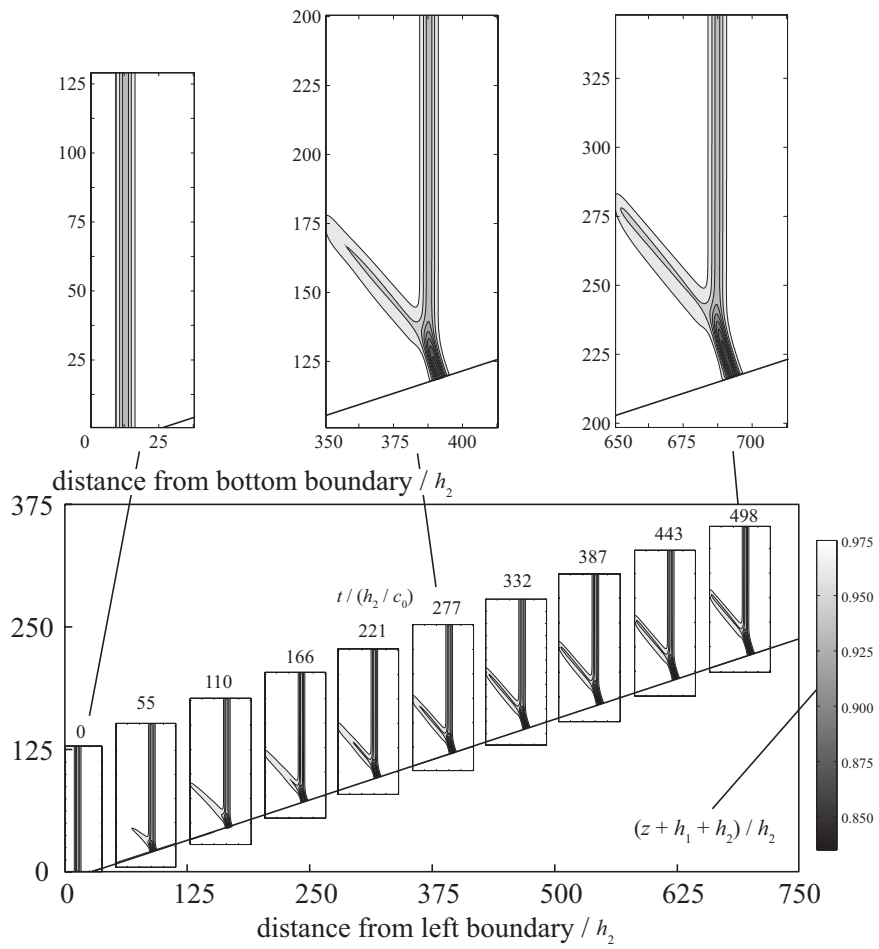


622

623 Fig. 6. Initial waves. The solid lines shows the 3rd order theoretical solution and dashed line shows
 624 the KdV solution. (a) case A. (b) case B.

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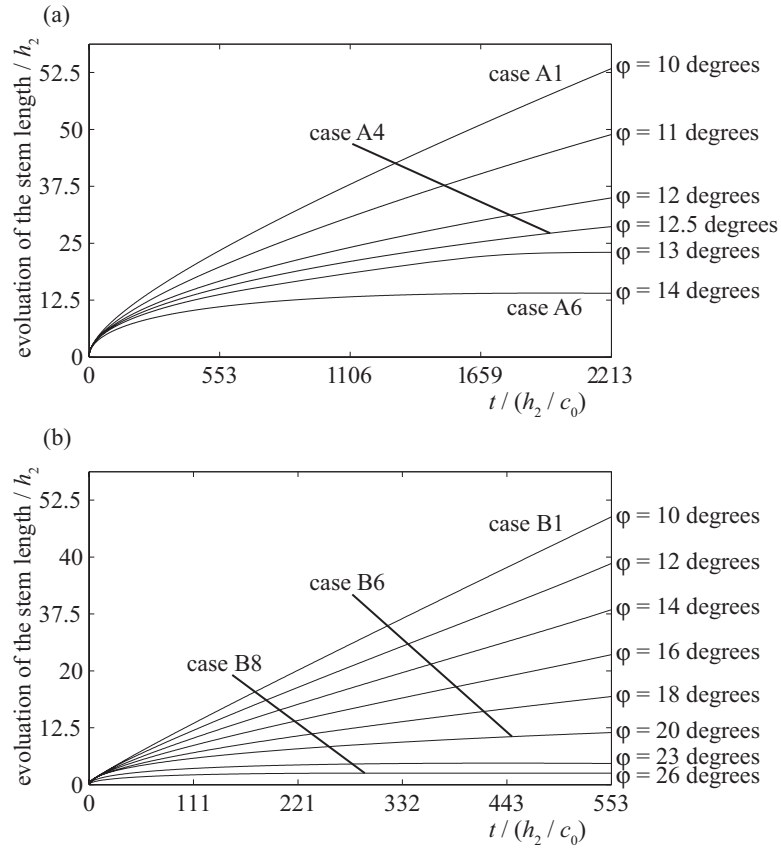
628 Fig. 7. Interfacial displacement of case B5. Each solid line square in the bottom figure indicates the

629 computational region at the times indicated. Top three figures show enlarged progress of internal

630 solitary wave at $t = 0$ s, $t = 250$ and $t = 450$ s.

631

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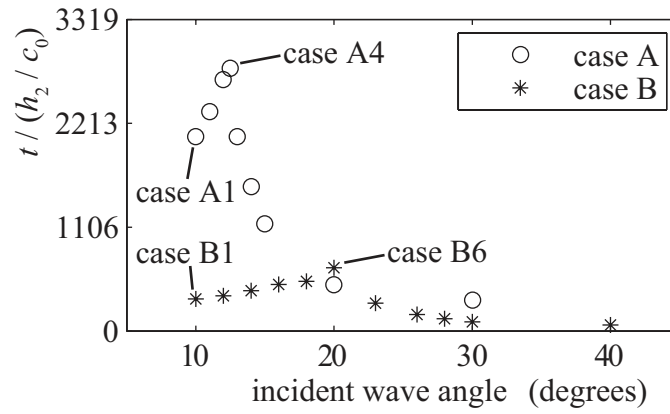
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635 layer depth. (a) cases A1 to A6. (b) cases B1 to B8.

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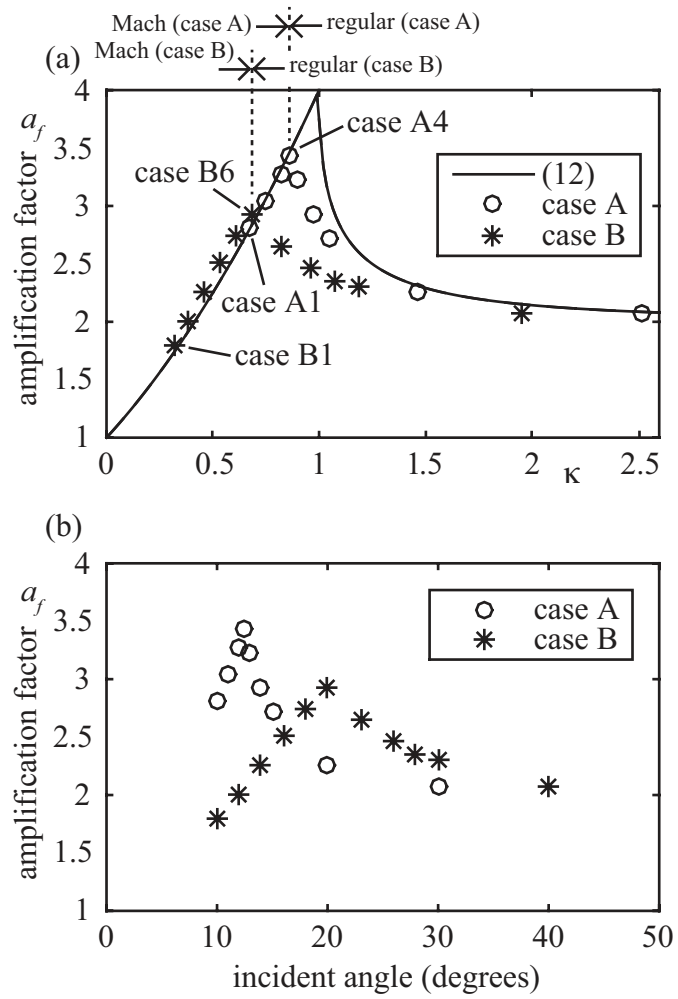
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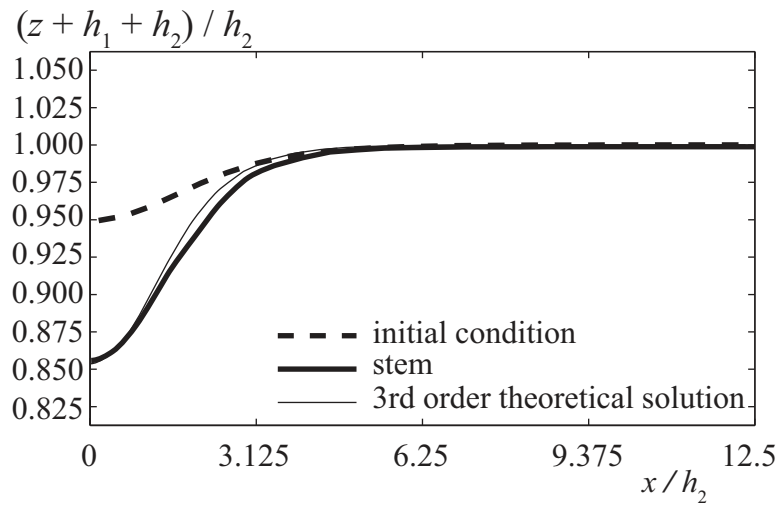
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646 angle and amplification factor vs incident angle.

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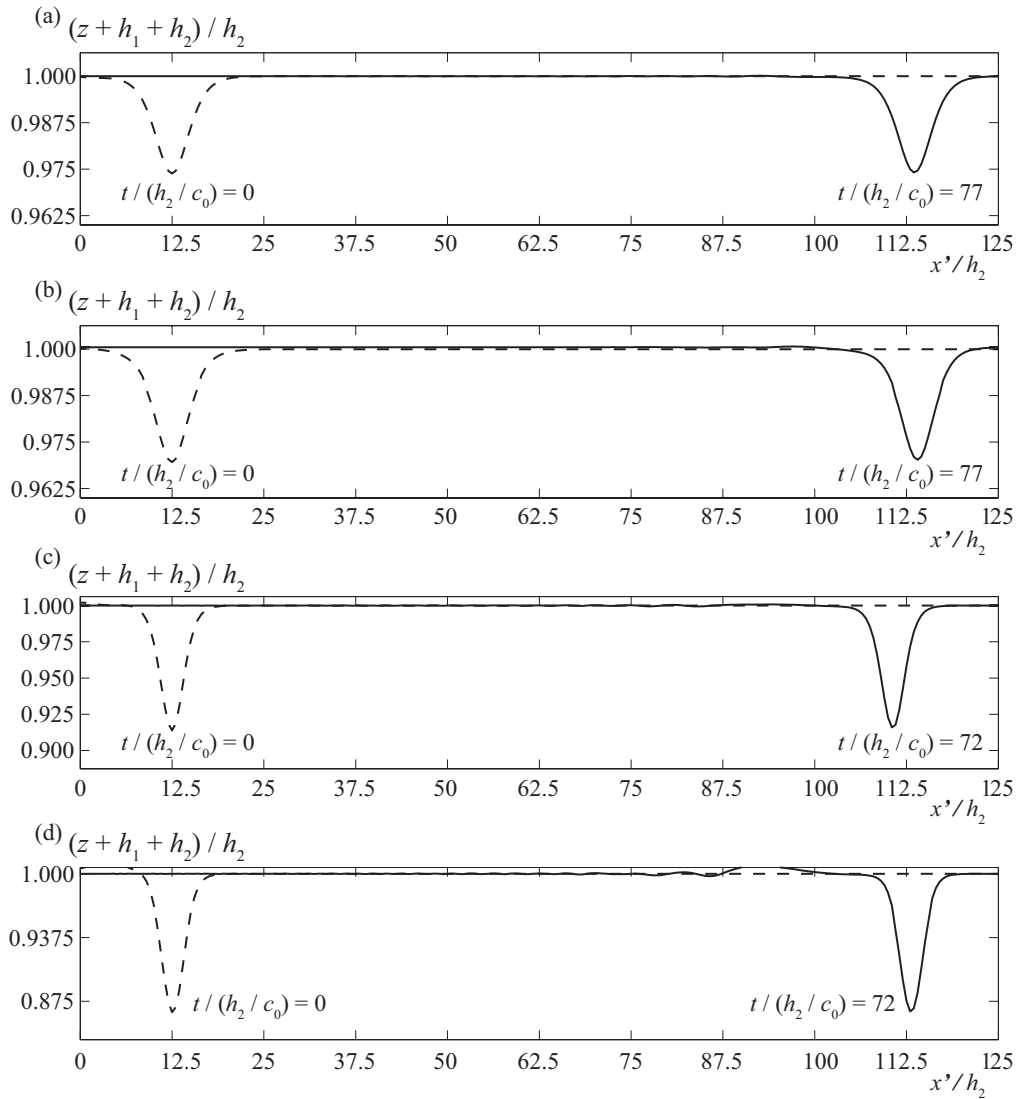
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651 The thick solid lines show the interfacial displacement of a stem. The thin solid lines show the 3rd

652 order theoretical solution.

653

654



655

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657 potential around a stem. (a) case A1. (b) case A5. (c) case B1. (d) case B6.

658

659

660 Table 1. Computational conditions for small and large amplitude cases.

case	a_0 / h_2	α (degree)	$\varphi_{c_{kp}}$ (degree)	κ	amplification factor
A1	0.01	10	14.4	0.68	2.81
A2	0.01	11	14.4	0.75	3.05
A3	0.01	12	14.4	0.82	3.27
A4	0.01	12.5	14.4	0.86	3.43
A5	0.01	13	14.4	0.90	3.23
A6	0.01	14	14.4	0.97	2.92
A7	0.01	15	14.4	1.05	2.72
A8	0.01	20	14.4	1.46	2.26
A9	0.01	30	14.4	2.52	2.08
B1	0.05	10	27.7	0.32	1.79
B2	0.05	12	27.7	0.39	2.01
B3	0.05	14	27.7	0.46	2.27
B4	0.05	16	27.7	0.53	2.50
B5	0.05	18	27.7	0.61	2.74
B6	0.05	20	27.7	0.69	2.94
B7	0.05	23	27.7	0.82	2.65
B8	0.05	26	27.7	0.97	2.46
B9	0.05	28	27.7	1.07	2.35
B10	0.05	30	27.7	1.19	2.29
B11	0.05	40	27.7	1.95	2.08

661 For all cases, $h_1 = 0.2$ m, $h_2 = 0.8$ m, and $\rho_1/\rho_2 = 1.0/2.0$, respectively.

662