Pitchfork and octopus bifurcations in a hyperelastic tube subjected to compression: Analytical post-bifurcation solutions and imperfection sensitivity

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Abstract

This paper studies the axisymmetric deformations of a nonlinearly hyperelastic tube subjected to axial compression. We aim at investigating the critical buckling stresses and modes, deriving the analytical solutions for the post-bifurcation deformations and studying the imperfection sensitivity. For a general isotropic hyperelastic tube, a coupled series-asymptotic method is utilized to derive two simplified model equations with specified constraints on the tube geometry. Then, we specialize to the Blatz–Ko material. With greased end conditions, through linear bifurcation analysis, we obtain the critical stress values and the corresponding mode numbers. The analytical solutions for the post-bifurcation states are constructed by the multiple scales method. By examining the solution behavior in the post-bifurcation regime, it is found that a thick tube could be considerably softer than a thin one. The singularities theory is used to consider the imperfection sensitivity, which reveals the mechanism is the existence of two modes corresponding to the same critical stress. Numerical solutions are also obtained which confirm the validity of the analytical solutions.

Keywords

Tube, post-bifurcation solutions, hyperelasticity, asymptotic analysis, imperfection sensitivity

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Dedicated to Professor KR Rajagopal.

I. Introduction

Buckling of elastic shell structures has been studied extensively. For example, Koiter [1] (see also [2]) developed an asymptotic approach for studying the post-bifurcation behavior and imperfection sensitivity of an elastic cylindrical shell under compression. Hutchinson [3] conducted a post-bifurcation analysis for externally pressurized spherical shells. Hunt et al. [4] used the von Karman equations of shells to characterize localization and periodicity in cylindrical shells. Usually, in most analytical studies, only geometrical nonlinearity is taken into account. Relatively speaking, few works deal with the postbifurcation behavior of shell structures within the framework of finite elasticity, which takes into account both geometrical and material nonlinearity. One major reason is probably the complexity of the governing field equations. As mentioned in a review article [5], 'the equations of motion...for an isotropic hyperelastic solid comprise a formidable system of nonlinear, partial differential equations (PDEs). The complexity of this system and its potential for generating non-unique solutions for even the simplest boundary-value problems,..., overwhelm our ability to solve them generally.' As a consequence, most theoretical studies on hyperelastic cylindrical shells in literature were carried out within the framework of linearized theories. By assuming the disturbances superimposed on a finite deformation are small, governing PDEs can be linearized. Then, it is possible to deduce the bifurcation modes and critical loads.

The first theoretical study on the bifurcation of an incompressible hyperelastic thick-walled cylindrical tube was given in [6]. More than forty years later, this problem was revisited in [7–9] and some new results for both axisymmetric and asymmetric buckling instabilities were obtained. Based on the threedimensional incremental equations for both thin-walled and thick-walled tubes in [10, 11], numerical studies were carried out in [12] on deformations of a thick-walled tube under axial loading and external pressure. For a compressible hyperelastic tube under compression, a linear bifurcation analysis was carried out in [13]. More recently, by using the Stroh formalism, Goriely et al. [14] studied the compression of a tube composed of an incompressible hyperelastic material. Both buckling modes and barreling modes were considered by them. In particular, they found the values of the geometric parameters at which there is a transition between buckling and barreling.

While linear bifurcation analysis can provide many insights, it cannot provide information on the post-bifurcation behavior (e.g. the answer for the very basic question of the bifurcation type). Actually, it was mentioned in [14]: 'It is also enticing to consider the possibility of performing an analytical post-buckling analysis of the solutions ... We leave this daunting task for another day.' In this paper, we shall conduct an analytical study on the axisymmetrical post-bifurcation states of a *compressible* hyperelastic tube under compression.

It is well known that it is difficult to obtain exact solutions for post-bifurcation states from the field equations. Usually, one has to resort to the approximate analytical or numerical solutions. Recently, a methodology of coupled series-asymptotic expansions, which was first developed to study nonlinear elastic waves (see [15, 16]), has been successfully used in constructing approximate analytical solutions within the framework of field equations for thin/slender structures composed of hyperelastic materials (see [17-21]). In this paper, this methodology will be adopted to study the axisymmetric deformations of a compressible hyperelastic tube subjected to uniaxial compression. One of our motivations is to obtain the analytical post-bifurcation solutions, which have not been obtained before in the current setting, to the best of the authors' knowledge. As we shall see, the analytical results reveal many insights. For example, for the single-mode case it is found that the bifurcation is of the *supercritical pitchfork* bifurcations type. For the case of two modes corresponding to one critical stress, there are eight bifurcated branches (four are supercritical and four are subcritical), which is called an *octopus* bifurcation. It appears that such a bifurcation has not been reported before in the context of cylindrical shells. Another important feature of the present work is that the post-bifurcation solutions are deduced from the field equations without ad hoc assumptions, which are often used in literature on postbuckling analysis.

An important issue in shell structures is imperfection sensitivity. With the explicit amplitude equations, it becomes possible to use the singularities theory to study imperfection sensitivity (see [22, 23]). One of the important advantages of this analytical approach (it appears that it has not been widely adopted in engineering literature) is that the qualitative information can be deduced without the need to make assumptions about the particular form of the imperfection (which is often done in the asymptotic approach for studying imperfection sensitivity). The analysis reveals that the mechanism for imperfection sensitivity is that there are two modes corresponding to a critical load. It is then deduced that usually a thin tube is imperfection-sensitive while in general a thick tube is not, which is in agreement with some experimental observations as mentioned in [24].

Experiments on the compression of tubes have also been extensively carried out. One of the interesting experiments was done by Willis [25] (see also [5] for a description). Motivated by this experimental work, Dorfmann and Haughton [13] studied the incremental equations for a compressible hyperelastic cylindrical tube compressed at two frictionless ends. It is worth mentioning that a number of experiments on compression of plastic cylindrical tubes has been carried out (see [24, 26, 27]). The present study is carried out within the framework of three-dimensional field equations of finite elasticity without an assumption about the form of the displacement field. Since a plastic material may be regarded as a nonlinear elastic one in a pure loading process, the results in this paper may also provide some qualitative information on the compression of plastic tubes. We also notice that the first bifurcation mode is axisymmetric for not very thin tubes according to the experiments in [26] (for diameter-to-thickness ratios of 20, 40, 59.7) and [24] (for diameter-to-thickness ratios between 23 and 52). This is one reason why we study axisymmetric deformations in this paper.

Cylindrical tubes are commonly used as energy absorbers due to their high stiffness and strength combined with low weight when subjected to axial load (see [28]). Metallic cylindrical tubes in automobiles can absorb the kinetic crash energy to limit the damage in a crash. The capability of absorbing energy depends on the type of buckling mode, which is related to the geometry of the tube and the boundary conditions. In this paper, another motivation is to study the effect of the geometry of the tube on the post-bifurcation states. With analytical solutions, this effect can be examined easily. For energy absorption purposes, one needs the axial stress to be small in the post-bifurcation regime. Usually, people would think that a thin tube is softer than a thick one. While this is true up to the bifurcation, we find that in the post-bifurcation regime a thick tube could be softer than a thin tube.

The structure of this paper is organized as follows. In Section 2, we set up the three-dimensional axisymmetric field equations and the traction-free boundary conditions for a general hyperelastic material. In Section 3, we derive two coupled nonlinear ordinary differential equations (ODEs) by the combined series-asymptotic expansions method. In Section 4, the tube is further assumed to be composed of a Blatz–Ko material. In Section 5, with sliding or greased conditions at two ends, linear bifurcation analysis is made to obtain the critical stress values and the corresponding mode numbers. The influences of the geometrical parameters are also examined in detail in this section. In Section 6, we derive the approximate analytical solutions for the post-bifurcation states. It turns out that this nonlinear bifurcation analysis can recover the critical stress values and the mode numbers. By using the singularities theory, in Section 7 the imperfection sensitivity is analyzed. In Section 8, numerical solutions are compared with the approximate analytical solutions and very good agreement is found. Finally, some concluding remarks are presented.

2. Field equations

We consider the axisymmetric deformation of a three-dimensional hyperelastic cylindrical tube subjected to a static axial force at one end with zero axial displacement at the other end. The inner and outer surfaces of the tube are traction-free and the end boundary conditions will be considered later. In the stressfree configuration, the inner and outer radii of the tube are, respectively, *a* and *b*, and the length is *l*. We take the cylindrical polar coordinates system and denote by (R, Θ, Z) and (r, θ, z) the coordinates of a material point of the tube in the reference and current configurations, respectively. The displacements of a material point can be represented as

$$U(R,Z) = r(R,Z) - R, \quad \Theta = \theta, \quad W(R,Z) = z(R,Z) - Z. \tag{2.1}$$

We introduce the orthonormal bases associated with the cylindrical coordinates and denote these by $\{\mathbf{E}_R, \mathbf{E}_{\Theta}, \mathbf{E}_Z\}$ and $\{\mathbf{e}_r, \mathbf{e}_{\theta}, \mathbf{e}_z\}$ in the reference and current configurations, respectively. Then the deformation gradient tensor **F** with respect to the bases $\{\mathbf{e}_i \otimes \mathbf{E}_i\}$ is given by

$$\mathbf{F} = \begin{pmatrix} 1 + U_R & 0 & U_Z \\ 0 & 1 + \frac{U}{R} & 0 \\ W_R & 0 & 1 + W_Z \end{pmatrix}.$$
 (2.2)

For an isotropic hyperelastic material, the strain-energy function Φ is a function of the three principle stretches e_1 , e_2 and e_3 of the Lagrangian strain tensor $\mathbf{E} = (1/2)(\mathbf{F}^T\mathbf{F} - \mathbf{I})$, that is, $\Phi = \Phi(e_1, e_2, e_3)$, where \mathbf{I} is the identity tensor. The conjugate stress tensor is given by

$$\mathbf{T} = \frac{\partial \Phi}{\partial \mathbf{E}} \tag{2.3}$$

with components $T_{ij} = \partial \Phi / \partial E_{ji}$. If the strains are small, it is possible to expand T_{ij} in terms of the strains up to any order. The formula containing terms up to the third-order material nonlinearity is

$$T_{ij} = A_{ijkl}^{(1)} E_{lk} + \frac{1}{2} A_{ijklmn}^{(2)} E_{lk} E_{nm} + \frac{1}{6} A_{ijklmnpq}^{(3)} E_{lk} E_{nm} E_{qp} + O(|E_{st}|^4),$$
(2.4)

where

$$A_{ijkln}^{(1)} = \frac{\partial^2 \Phi}{\partial E_{ji} \partial E_{lk}} |_{\mathbf{E}=\mathbf{0}}, \quad A_{ijklmn}^{(2)} = \frac{\partial^3 \Phi}{\partial E_{ji} \partial E_{lk} \partial E_{nm}} |_{\mathbf{E}=\mathbf{0}},$$

$$A_{ijklmnpq}^{(3)} = \frac{\partial^4 \Phi}{\partial E_{ji} \partial E_{lk} \partial E_{nm} \partial E_{qp}} |_{\mathbf{E}=\mathbf{0}}$$
(2.5)

are the incremental elastic moduli, which can be calculated once a specific form of the strain energy function is given. The formulae for $A_{ijkl}^{(1)}$ and $A_{ijklmn}^{(2)}$ can be found in [29]. By a similar procedure, the formula for $A_{ijklmnpq}^{(3)}$ can also be deduced. The non-zero elastic moduli are given in the Appendix. From the conjugate stress tensor **T**, the nominal stress tensor **S** is given by $\mathbf{S} = \mathbf{TF}^{T}$.

For a static equilibrium state of the tube (neglecting the body force), we have the following two field equations:

$$\frac{\partial S_{Rr}}{\partial R} + \frac{\partial S_{Zr}}{\partial Z} + \frac{S_{Rr} - S_{\Theta\theta}}{R} = 0, \qquad (2.6a)$$

$$\frac{\partial S_{Rz}}{\partial R} + \frac{\partial S_{Zz}}{\partial Z} + \frac{S_{Rz}}{R} = 0.$$
(2.6b)

We consider the case where the inner and outer surfaces of the tube are traction-free. Thus, we have the traction-free boundary conditions

$$S_{Rr} = 0, \quad S_{Rz} = 0, \quad \text{at} \quad R = a, b.$$
 (2.7)

Equations (2.6a) and (2.6b) together with (2.7) provide the governing equations and the boundary conditions for two unknowns U and W, which compose a complicated coupled nonlinear system of PDEs. It is almost formidable to construct exact solutions from those PDEs for the postbifurcation states. The aim is to construct explicit approximate analytical solutions (which is also a very difficult task). For this purpose, we shall adopt an approach involving combined series-asymptotic expansions to tackle this complicated problem. A similar methodology has been utilized to study non-linear elastic waves and phase transitions in incompressible materials (see [15, 16, 30]). In the following section, we shall derive the reduced model equations by this method of coupled series-asymptotic expansions.

3. Nondimensionalization and series-asymptotic reductions

According to [15], for this kind of axisymmetric problem it is convenient to introduce the transformation

$$U = Ru, \qquad s = R^2. \tag{3.1}$$

Equation (3.1)₁ is introduced based on the observation that $U = R\lambda$ (λ is a constant) for a homogeneous deformation induced by axial compression, while (3.1)₂ is introduced to eliminate 1/R in (2.6a) and (2.6b). The length *l* of the tube is a natural choice to scale the spatial variables. To be consistent with (2.4), we consider the case where under compression one end of the tube moves a distance up to *h* with the constraint of $\varepsilon = h/l$ being small (*h* is then a measure of the axial displacement). Thus, we introduce the following scalings:

$$s = l^2 \tilde{s}, \quad Z = l\tilde{z}, \quad u = \frac{h}{l} \tilde{u}, \quad W = h\tilde{w}.$$
 (3.2)

After substituting (3.2) into (2.6a), (2.6b) and (2.7), we obtain a dimensionless system, in which two geometric parameters $\nu_0 = a^2/l^2$ and $\nu_1 = b^2/l^2$ arise. To use expansion methods to proceed forward, some restrictions on the geometric parameters have to be imposed. We consider the case where $\delta = \nu_1 - \nu_0 = (b+a)(b-a)/l^2$ is small such that $O(\delta^2)$ terms can be omitted. We point out that this implies a small thickness-to-length ratio, but the thickness-to-radius ratio does not need to be small at this stage.

For convenience of notation, we hereafter drop the tildes (i.e. *s* means \tilde{s}). We note that since $0 \le s - \nu_0 \le \delta$, $s - \nu_0$ is a small variable. If we assume u(s, z) and w(s, z) are sufficiently smooth in *s*, we can take the following series expansions of *u* and *w* in the neighborhood of $s = \nu_0$:

$$\begin{cases} u(s,z) = u_0(z) + (s - \nu_0)u_1(z) + (s - \nu_0)^2 u_2(z) + (s - \nu_0)^3 u_3(z) + \dots, \\ w(s,z) = w_0(z) + (s - \nu_0)w_1(z) + (s - \nu_0)^2 w_2(z) + (s - \nu_0)^3 w_3(z) + \dots. \end{cases}$$
(3.3)

Remark: Since $R/l - \sqrt{\nu_0} = \sqrt{\nu_0 + s - \nu_0} - \sqrt{\nu_0}$, which can be expanded as a power series of $s - \nu_0$ for $\delta/\nu_0 < 1$, there is no need for the above expansions to contain $(R/l - \sqrt{\nu_0})^n$ (n = 1, 2, ...) terms.

Substituting (3.3) into the nondimensionalized traction-free boundary conditions (2.7), we obtain the following four equations:

$$\nu_{0}\mathscr{D}_{1}(\mathbf{v}_{0})u_{1} + \varepsilon\nu_{0}\mathscr{D}_{2}(\mathbf{v}_{0})w_{1} + (\zeta_{1} + \zeta_{2})u_{0} + \zeta_{2}w_{0z} + \varepsilon^{2}\nu_{0}\mathscr{D}_{3}(\mathbf{v}_{0})u_{1}^{2} + \varepsilon\nu_{0}\mathscr{D}_{4}(\mathbf{v}_{0})u_{1}w_{1} + \varepsilon\nu_{0}\mathscr{D}_{5}(\mathbf{v}_{0})w_{1}^{2} + \varepsilon^{2}\nu_{0}^{3}(4\zeta_{1} + 8\eta_{1} + \frac{4}{3}\kappa_{1})u_{1}^{3} + \varepsilon^{2}\nu_{0}^{2}4(\zeta_{1} + \eta_{1} + \eta_{4} + \kappa_{5})u_{1}w_{1}^{2} + \varepsilon\mathscr{D}_{6}(\mathbf{v}_{0})$$
(3.4a)

$$+\varepsilon^2 \mathscr{D}_7(\mathbf{v_0}) = 0,$$

 $\varepsilon\nu_0\mathscr{D}_8(\mathbf{v_0})u_1 + \mathscr{D}_9(\mathbf{v_0})w_1 + \zeta_3u_{0z} + \varepsilon\nu_0\mathscr{D}_{10}(\mathbf{v_0})u_1^2 + \varepsilon\nu_0\mathscr{D}_{11}(\mathbf{v_0})u_1w_1 + \varepsilon\nu_0\mathscr{D}_{12}(\mathbf{v_0})w_1^2$

$$+\varepsilon^{2}\nu_{0}^{2}4(\zeta_{1}+\eta_{1}+\eta_{4}+\kappa_{5})u_{1}^{2}w_{1}+\varepsilon^{2}\nu_{0}\left(4\zeta_{1}+8\eta_{4}+\frac{4}{3}\kappa_{9}\right)w_{1}^{3}+\varepsilon\mathscr{D}_{13}(\mathbf{v_{0}})+\varepsilon^{2}\mathscr{D}_{14}(\mathbf{v_{0}})=0,$$
(3.4b)

$$((3\zeta_{1}+\zeta_{2})u_{1}+4\zeta_{1}\nu_{0}u_{2}+\zeta_{2}w_{1z}) + \varepsilon \mathscr{D}_{15}(\mathbf{v}_{2}) + \varepsilon^{2}\mathscr{D}_{16}(\mathbf{v}_{2}) + \delta((5\zeta_{1}+\zeta_{2})u_{2} + 6\zeta_{1}\nu_{0}u_{3}+\zeta_{2}w_{2z} + \varepsilon \mathscr{D}_{17}(\mathbf{v}_{3}) + \varepsilon^{2}\mathscr{D}_{18}(\mathbf{v}_{3})) + O(\delta^{2}) = 0,$$
(3.4c)

$$(2\zeta_{3}w_{1} + \zeta_{3}u_{0z} + \nu_{0}(4\zeta_{3}w_{2} + \zeta_{3}u_{1z})) + \varepsilon \mathscr{D}_{19}(\mathbf{v}_{2}) + \varepsilon^{2}\mathscr{D}_{20}(\mathbf{v}_{2}) + \delta(4\zeta_{3}w_{2} + \zeta_{3}u_{1z}) + \nu_{0}(6\zeta_{3}w_{3} + \zeta_{3}u_{2z}) + \varepsilon \mathscr{D}_{21}(\mathbf{v}_{3}) + \varepsilon^{2}\mathscr{D}_{22}(\mathbf{v}_{3})) + O(\delta^{2}) = 0,$$
(3.4d)

where $\mathbf{v_i}$ (i = 0, 1, 2, 3) are the four vectors defined below:

$$\begin{cases} \mathbf{v_0} = (u_0, w_0), \ \mathbf{v_1} = (u_0, u_1, w_0, w_1), \\ \mathbf{v_2} = (u_0, u_1, u_2, w_0, w_1, w_2), \ \mathbf{v_3} = (u_0, u_1, u_2, u_3, w_0, w_1, w_2, w_3), \end{cases}$$

 η_j , ζ_j , κ_j (j = 1,...) are the elastic moduli given in the Appendix and \mathcal{D}_i (i = 1, ..., 22) are operators on the arguments whose long expressions are omitted. In obtaining (3.4c) and (3.4d), we have neglected terms higher than $O(\delta^2)$. In this system, there are eight unknowns u_i and w_i (i = 0, 1, 2, 3) if the abovementioned higher-order terms are omitted. The other four equations can be obtained from the field equations.

Substituting (3.3) into (2.6a) and (2.6b), the left-hand sides of the equations become two series in $s-\nu_0$ and all the coefficients of $(s-\nu_0)^n$ (n = 0, 1, 2,...) should be zero. From the coefficients of $(s-\nu_0)^0$ and $(s-\nu_0)^1$, four equations with the same eight unknowns are obtained:

$$\nu_0(8\zeta_1u_1 + 2(\zeta_2 + \zeta_3)w_{1z} + \zeta_3u_{0zz}) + 8\zeta_1\nu_0^2u_2 + \varepsilon\mathscr{D}_{23}(\mathbf{v}_1) + \varepsilon^2\mathscr{D}_{24}(\mathbf{v}_1) = 0,$$
(3.5a)

$$8\zeta_1 u_1 + 2(\zeta_2 + \zeta_3) w_{1z} + \zeta_3 u_{0zz} + \nu_0 (32\zeta_1 u_2 + 4(\zeta_2 + \zeta_3) w_{2z} + \zeta_3 u_{1zz}) + 24\zeta_1 \nu_0^2 u_3 + \varepsilon \mathcal{D}_{25}(\mathbf{v_2}) + \varepsilon^2 \mathcal{D}_{26}(\mathbf{v_2}) = 0,$$
(3.5b)

$$4\zeta_{3}w_{1} + 2(\zeta_{2} + \zeta_{3})u_{0z} + \zeta_{1}w_{0zz} + \nu_{0}(8\zeta_{3}w_{2} + 2(\zeta_{2} + \zeta_{3})u_{1z}) + \varepsilon \mathscr{D}_{27}(\mathbf{v}_{1}) + \varepsilon^{2}\mathscr{D}_{28}(\mathbf{v}_{1}) = 0,$$
(3.5c)

$$16\zeta_{3}w_{2} + 4(\zeta_{2} + \zeta_{3})u_{1z} + \zeta_{1}w_{1zz} + \nu_{0}(24\zeta_{3}w_{3} + 4(\zeta_{2} + \zeta_{3})u_{2z}) + \varepsilon \mathscr{D}_{29}(\mathbf{v}_{2}) + \varepsilon^{2}\mathscr{D}_{30}(\mathbf{v}_{2}) = 0.$$
(3.5d)

Now the governing system (2.6a) to (2.7) has been transformed into a closed system of eight equations (3.4) and (3.5) with eight unknowns u_i and w_i (i = 0, 1, 2, 3). A key that the present methodology works is that a closed finite system can be obtained!

We find that (3.4a) and (3.4b) are two algebraic equations for u_1 and w_1 . By perturbation expansions in terms of the small parameter ε , we can obtain the asymptotic expressions of u_1 and w_1 . Also, it can be seen that (3.5a) and (3.5c) are two linear algebraic equations for u_2 and w_2 and (3.5b) and (3.5d) are two linear algebraic equations for u_3 and w_3 . Another key for the present methodology is that among the closed system of eight equations six are algebraic equations! As a result, u_i and w_i (i = 1, 2, 3) can be expressed in terms of u_0 and w_0 (their long expressions are omitted here). Substituting these expressions into (3.4c) and (3.4d) and neglecting $O(\varepsilon^3, \varepsilon^2 v_0, \delta v_0, (\delta/v_0)^2)$ terms, we can obtain two equations with two unknowns u_0 and w_0 . It turns out that one of the two equations can be integrated once. As a result, we have

$$\begin{aligned} \frac{\zeta_{1}^{2} - \zeta_{2}^{2}}{2\zeta_{1}} u_{0} + \frac{(\zeta_{1} - \zeta_{2})\zeta_{2}}{2\zeta_{1}} w_{0z} + \varepsilon(a_{1}u_{0}^{2} + a_{2}u_{0}w_{0z} + a_{3}w_{0z}^{2}) + \varepsilon^{2}(a_{4}u_{0}^{3} + a_{5}u_{0}^{2}w_{0z} \\ &+ a_{6}u_{0}w_{0z}^{2} + a_{7}w_{0z}^{3}) + \varepsilon\nu_{0}\left(\frac{\zeta_{2}(-\zeta_{1} + \zeta_{2})}{4\zeta_{1}}u_{0z}^{2} + \frac{\zeta_{2}(-\zeta_{1} + \zeta_{2})}{2\zeta_{1}}u_{0}u_{0zz} + \frac{-\zeta_{1}^{2} + \zeta_{2}^{2}}{2\zeta_{1}}u_{0zz}w_{0z} \\ &+ \frac{-\zeta_{1}^{2} + \zeta_{2}^{2}}{2\zeta_{1}}u_{0z}w_{0zz}\right) + \delta\left(\frac{\zeta_{1}^{2} - \zeta_{2}^{2}}{8\zeta_{1}}w_{0zzz} + \varepsilon\left(a_{8}u_{0z}^{2} - \frac{(\zeta_{1} - \zeta_{2})^{2}}{8\zeta_{1}}u_{0}u_{0zz}\right) \\ &+ \frac{2\zeta_{1}^{2} - \zeta_{1}\zeta_{2} - \zeta_{2}^{2}}{8\zeta_{1}}u_{0zz}w_{0z} + a_{9}u_{0z}w_{0zz} + a_{10}w_{0zz}^{2} + a_{11}u_{0}w_{0zzz} + a_{12}w_{0z}w_{0zzz}\right) \\ &+ \frac{1}{\nu_{0}}\left(\frac{-\zeta_{1}^{2} + \zeta_{2}^{2}}{2\zeta_{1}}u_{0} + \frac{\zeta_{2}(-\zeta_{1} + \zeta_{2})}{2\zeta_{1}}w_{0z} + \varepsilon(a_{13}u_{0}^{2} + a_{14}u_{0}w_{0z} + a_{15}w_{0z}^{2}\right) \\ &+ \varepsilon^{2}(a_{16}u_{0}^{3} + a_{17}u_{0}^{2}w_{0z} + a_{18}u_{0}w_{0z}^{2} + a_{19}w_{0z}^{3})\right) = 0, \end{aligned}$$

$$(3.6a)$$

$$A + \frac{\zeta_{2}(-\zeta_{1}+\zeta_{2})}{2\zeta_{1}}u_{0} + \frac{-\zeta_{1}^{2}+\zeta_{2}^{2}}{2\zeta_{1}}w_{0z} + \varepsilon(a_{20}u_{0}^{2}+a_{21}u_{0}w_{0z}+a_{22}w_{0z}^{2}) + \varepsilon^{2}(a_{23}u_{0}^{3}) + a_{24}u_{0}^{2}w_{0z} + a_{25}u_{0}w_{0z}^{2} + a_{26}w_{0z}^{3}) + \varepsilon\nu_{0}\frac{-\zeta_{1}^{2}+\zeta_{2}^{2}}{4\zeta_{1}}u_{0z}^{2} + \delta\left(\frac{\zeta_{1}^{2}-\zeta_{2}^{2}}{8\zeta_{1}}u_{0zz}\right) + \varepsilon\left(\frac{\zeta_{2}(-\zeta_{1}+\zeta_{2})}{16\zeta_{1}}u_{0z}^{2} + a_{34}u_{0}u_{0zz} + a_{35}u_{0zz}w_{0z} + \frac{-\zeta_{1}^{2}+\zeta_{2}^{2}}{4\zeta_{1}}u_{0z}w_{0zz}\right) + \frac{1}{\nu_{0}}(\varepsilon(a_{27}u_{0}^{2}+a_{28}u_{0}w_{0z}+a_{29}w_{0z}^{2}) + \varepsilon^{2}(a_{30}u_{0}^{3}+a_{31}u_{0}^{2}w_{0z}+a_{32}u_{0}w_{0z}^{2} + a_{33}w_{0z}^{3})) = 0,$$
(3.6b)

where *A* is an integration constant and a_i (i = 1, 2, ..., 35) are constants dependent on material constants. By considering the resultant force $T = \int_0^{2\pi} \int_a^b S_{Zz} R \, dR \, d\Theta$ acting on a cross-section that is planar and perpendicular to the axial direction of the tube, we find that

$$A = \frac{T}{2\pi(b^2 - a^2)\varepsilon} = \frac{\gamma(\zeta_1^2 + \zeta_1\zeta_2 - 2\zeta_2^2)}{2(\zeta_1 + \zeta_2)\varepsilon},$$
(3.7)

where γ is the dimensionless engineering stress.

We point out that in deriving the above two asymptotic model equations $O(\varepsilon^3, \varepsilon^2 \nu_0, \delta^2, \delta \nu_0, (\delta/\nu_0)^2)$ terms have been neglected. For them to yield reliable results, we impose that these omitted terms should be less than 5.5%. Now, we examine the restrictions in terms of the loading and geometrical parameters.

For $\delta^2 < 0.055$, $(\delta/\nu_0)^2 < 0.055$ and $\delta\nu_0 < 0.055$, we need, respectively, $(b/l)^2 - (a/l)^2 < 0.2234$, b/l < 1.106(a/l) and $((b/l)^2 - (a/l)^2)(a/l)^2 < 0.055$. Also, suppose that we impose $\varepsilon < 0.2$ (ε can be regarded as a loading parameter). Then, $\varepsilon^3 < 0.055$ automatically. Therefore, $\varepsilon^2\nu_0 < 0.055$ implies that (a/l) < 1.173. Based on the above analysis, we plot the valid domain of a/l and b/l in Figure 1.

We notice that within this domain we do not need the tube to be very thin (the diameter-to-thickness ratio can be as small as 20) or slender (the diameter-to-length ratio can be as large as 2.34). For most of the tubes used in the experiments by [24, 26, 27] the geometrical parameters were in this domain.

4. Governing equations for the Blatz-Ko material

In Section 3, we obtained the governing equations for the axisymmetric deformations of a hyperelastic tube subjected to uniaxial compression. Once the strain-energy function of a specific hyperelastic material is given, we can calculate the corresponding elastic moduli through the formulae presented in the Appendix. Substituting the elastic moduli into the coefficients of (3.6a) and (3.6b), the model equations for the specific material can be determined.

In the following discussion, we shall take the Blatz–Ko material as an example. The strain-energy function of the Blatz–Ko material is



Figure I. The valid domain of a/l and b/l for the model equations is the one bounded by curves 2, 3 and 4. Curves 1, 2, 3 and 4 are determined by $(b/l)^2 - (a/l)^2 = 0.2234$, b/l = 1.106(a/l), $((b/l)^2 - (a/l)^2)(a/l)^2 = 0.055$ and b/l = a/l, respectively.

$$\Phi(e_1, e_2, e_3) = \frac{\mu_0}{2} \left(\frac{1}{1+2e_1} + \frac{1}{1+2e_2} + \frac{1}{1+2e_3} + 2\sqrt{(1+2e_1)(1+2e_2)(1+2e_3)} - 5 \right), \tag{4.1}$$

where μ_0 is the shear modulus for infinitesimal deformations. By using the formulae in the Appendix, the non-zero elastic moduli (defined in (2.5)) of the Blatz–Ko material are

$$\begin{aligned} \zeta_1 &= 3\mu_0, \quad \zeta_2 = \mu_0, \quad \eta_1 = -21\mu_0, \quad \eta_2 = -\mu_0, \quad \eta_3 = \mu_0, \\ \kappa_1 &= 177\mu_0, \quad \kappa_2 = 3\mu_0, \quad \kappa_3 = \mu_0, \quad \kappa_4 = -\mu_0. \end{aligned}$$
(4.2)

Substituting (4.2) into (3.6a) and (3.6b) and retaining the original dimensional variables by letting $U = \varepsilon u_0$ (*aU* represents the radial displacement of a material point on the inner surface) and $G = \varepsilon w_{0z}$ (which represents the axial strain of a material point on the inner surface), we can obtain the following governing equations for the Blatz-Ko material:

$$\begin{aligned} \frac{4}{3}U - \frac{26}{9}U^2 + \frac{398}{81}U^3 + \left(\frac{1}{3} - \frac{1}{9}G + \frac{2}{27}U^2\right)G + \left(-\frac{1}{18} + \frac{1}{54}U\right)G^2 + \frac{2}{81}G^3 \\ + \nu_0 \left(-\frac{1}{6}U_Z^2 - \frac{1}{3}UU_{ZZ} - \frac{4}{3}U_{ZZ}G - \frac{4}{3}U_ZG_Z\right) + \delta\left(\frac{7}{12}U_{ZZ}G - \frac{19}{36}U_Z^2 - \frac{1}{6}UU_{ZZ} \\ + \frac{19}{36}U_ZG_Z - \frac{16}{9}G_Z^2 + \left(\frac{1}{3} - \frac{1}{9}U - \frac{16}{9}G\right)G_{ZZ} + \frac{1}{\nu_0}\left(-\frac{4}{3}U + \frac{34}{9}U^2 - \frac{218}{27}U^3 + \left(-\frac{1}{3} + \frac{5}{9}U - \frac{4}{3}U^2\right)G + \left(\frac{1}{9} - \frac{1}{3}U\right)G^2 - \frac{2}{27}G^3\right)\right) = 0, \end{aligned}$$

$$\begin{aligned} & \left(4.3a\right) \\ & + \left(-\frac{1}{3} + \frac{5}{9}U - \frac{4}{3}U^2\right)G + \left(-\frac{4}{3} + \frac{1}{9}U - \frac{1}{54}U^2\right)G + \left(\frac{26}{9} - \frac{2}{27}U\right)G^2 - \frac{398}{81}G^3 \\ & - \frac{2\nu_0}{3}U_Z^2 + \delta\left(-\frac{1}{24}U_Z^2 + \left(\frac{1}{3} - \frac{1}{9}U\right)U_{ZZ} - \frac{16}{9}U_{ZZ}G - \frac{2}{3}U_ZG_Z + \frac{1}{\nu_0}\left(\frac{2}{9}U^2 - \frac{4.3b}{10}U_Z^2\right)G + \left(\frac{1}{9}U - \frac{19}{54}U^2\right)G + \left(\frac{1}{72} - \frac{17}{108}U\right)G^2 - \frac{2}{81}G^3\right) = 0. \end{aligned}$$

$$(4.3a)$$

From (4.3a) and (4.3b), we find that the coefficients of these two equations do not depend upon the material constant μ_0 and only depend upon the geometrical parameters δ and ν_0 .

In the following sections, we study the bifurcations of the governing system of ODEs (4.3a) and (4.3b) subjected to some proper end boundary conditions.

5. Linear bifurcation analysis with sliding boundary conditions

In this section, we will conduct a linear bifurcation analysis for the coupled ODE system (4.3a) and (4.3b) with sliding boundary conditions. Without loss of generality, we suppose the length of the tube l to be 1 in the following discussions.

We assume sliding boundary conditions at the two ends of the tube, which means the shear stresses on the two end surfaces of the tube are zeros, that is,

$$S_{Zr} = 0$$
 at $Z = 0, 1.$ (5.1)

We also assume that the two end surfaces of the tube keep flat during the deformation, which leads to

$$W_R(R,Z) = 0$$
 at $Z = 0, 1,$ (5.2)

where W is the axial displacement defined in (2.1). Based on the above two equations and adopting the manipulations introduced in Section 3, we obtain the following reduced boundary conditions:

$$U_Z = 0, \quad G_Z = 0 \quad \text{at} \quad Z = 0, 1.$$
 (5.3)

We first consider the homogeneous deformation of the elastic tube, that is,

$$U = U_h, \quad G = G_h \quad \text{for} \quad 0 \le Z \le 1, \tag{5.4}$$

where U_h and G_h are two constants. It is clear that (5.4) satisfy the end boundary conditions (5.3). To determine U_h and G_h , we substitute (5.4) into (4.3a) and (4.3b) to arrive at two algebraic equations:

$$\frac{4}{3}U_{h} - \frac{26}{9}U_{h}^{2} + \frac{398}{81}U_{h}^{3} + \left(\frac{1}{3} - \frac{1}{9}U_{h} + \frac{2}{27}U_{h}^{2}\right)G_{h} + \left(-\frac{1}{18} + \frac{1}{54}U_{h}\right)G_{h}^{2} + \frac{2}{81}G_{h}^{3} + \rho\left(-\frac{4}{3}U_{h} + \frac{34}{9}U_{h}^{2} - \frac{218}{27}U_{h}^{3} + \left(-\frac{1}{3} + \frac{5}{9}U_{h} - \frac{4}{3}U_{h}^{2}\right)G_{h} + \left(\frac{1}{9} - \frac{1}{3}U_{h}\right)G_{h}^{2}$$
(5.5a)
$$-\frac{2}{27}G_{h}^{3} = 0,$$

$$\frac{1}{2}U_{h} + \frac{1}{2}U_{h}^{2} - \frac{2}{2}U_{h}^{3} + \left(-\frac{4}{3} + \frac{1}{2}U_{h} - \frac{1}{3}U_{h}^{2}\right)G_{h} + \left(\frac{26}{3} - \frac{2}{2}U_{h}\right)G_{h}^{2} - \frac{398}{6}G_{h}^{3}$$

$$\frac{5}{4}\gamma - \frac{1}{3}U_{h} + \frac{1}{18}U_{h}^{2} - \frac{2}{81}U_{h}^{3} + \left(-\frac{4}{3} + \frac{1}{9}U_{h} - \frac{1}{54}U_{h}^{2}\right)G_{h} + \left(\frac{26}{9} - \frac{2}{27}U_{h}\right)G_{h}^{2} - \frac{398}{81}G_{h}^{3} + \rho\left(\frac{2}{9}U_{h}^{2} - \frac{38}{81}U_{h}^{3} + \left(\frac{1}{9}U_{h} - \frac{19}{54}U_{h}^{2}\right)G_{h} + \left(\frac{1}{72} - \frac{17}{108}U_{h}\right)G_{h}^{2} - \frac{2}{81}G_{h}^{3}\right) = 0,$$
(5.5b)

where $\rho = \delta/\nu_0$. Note that (5.5a) and (5.5b) are both third-order polynomial equations. Thus obtaining the explicit expressions of U_h and G_h from these two equations may be very complicated. Obviously, U_h and G_h should be functions of the engineering stress γ . As γ is a parameter of $O(\varepsilon)$ (see (3.2) and (5.5b)), we consider the asymptotic expressions of U_h and G_h in terms of γ . By using the regular perturbation method, we can obtain the following asymptotic expressions for U_h and G_h :

$$U_h = -\frac{1}{4}\gamma - \frac{13}{32}\gamma^2 - \frac{125}{128}\gamma^3, \qquad G_h = \gamma + \frac{9}{4}\gamma^2 + \frac{25}{4}\gamma^3.$$
(5.6)

We note that the asymptotic solutions do not depend on ρ (as should be the case for a homogeneous deformation induced by uniaxial compression; thus this can also be regarded as a consistency check for our approach).

In Figure 2, we show the comparisons of the solutions of (5.5a) and (5.5b) with the homogeneous solutions of the original exact equations (2.6a), (2.6b) and (2.7). It can be seen that when $-0.2 < \gamma < 0.1$, they agree with each other very well. Also, through the comparisons of the asymptotic solutions (5.6) with the solutions of (5.5a) and (5.5b), we find that when $|\gamma| < 0.16$, the asymptotic solutions (5.6) are good approximations of the solutions of (5.5a) and (5.5b).

In addition to the constant solutions, there may also exist nontrivial solutions satisfying (4.3a) and (4.3b) and the boundary conditions (5.3) when the external force exceeds a critical value. In the following, we will determine the critical stress values at which there exist nontrivial solutions. First, we set

$$U = U_h + \Delta U, \qquad G = G_h + \Delta G, \tag{5.7}$$



Figure 2. Comparisons of the solutions (circles) of (5.5a) and (5.5b) with the homogeneous solutions (triangles) of (2.6a) to (2.7).

where ΔU and ΔG are two small variations. Substituting (5.7) into (4.3a) and (4.3b) and further dropping the nonlinear terms, we have

$$c_1 \Delta G_{ZZ} + c_2 \Delta G + c_3 \Delta U_{ZZ} + c_4 \Delta U = 0, \qquad (5.8a)$$

$$c_1 \Delta U_{ZZ} + c_5 \Delta U + c_6 \Delta G = 0, \tag{5.8b}$$

where c_i (i = 1, 2, ..., 26) in the above equations and the following equations are constants dependent on U_h , G_h , ρ and ν_0 .

From (5.3), ΔU and ΔG should satisfy the following boundary conditions:

$$\Delta U_Z = 0, \quad \Delta G_Z = 0 \quad \text{at} \quad Z = 0, 1.$$
 (5.9)

To find the critical stress values at which there exist nontrivial solutions of U and G satisfying (4.3a), (4.3b) and (5.3) is to find the critical stress values at which there exist nontrivial solutions of ΔU and ΔG for the linear system (5.8a) and (5.8b) with the boundary conditions (5.9). We suppose that ΔU and ΔG have the following general forms:

$$\Delta U(Z) = Ae^{\lambda Z}, \qquad \Delta G(Z) = Be^{\lambda Z}, \tag{5.10}$$

where A, B and λ are constants. Substituting (5.10) into (5.8a) and (5.8b), we have the following linear system:

$$M_1\begin{pmatrix}A\\B\end{pmatrix} = \begin{pmatrix}m_{11} & m_{12}\\m_{21} & m_{22}\end{pmatrix}\begin{pmatrix}A\\B\end{pmatrix} = 0,$$
(5.11)

where m_{ii} (*i*, *j* = 1, 2) are in terms of λ , U_h , G_h , ρ and ν_0 which are not shown for brevity.

To find nontrivial solutions of (5.8a) and (5.8b), we need Det $M_1 = 0$, which yields the characteristic equation of the ODE system of (5.8a) and (5.8b),

$$\psi_3 \lambda^4 + \psi_2 \lambda^2 + \psi_1 = 0, \tag{5.12}$$

where ψ_i (i = 1, 2, 3) are constants dependent on c_i (i = 1, ..., 6). Equation (5.12) is a fourth-order polynomial equation, thus there are four roots. As (5.12) only depends on the even orders of λ , the four roots can be divided into two pairs. The two roots in each pair are negatives of each other. We can classify the four roots of equation (5.12) into the following three different cases: (1) there are four real roots; (2) there are four purely imaginary roots; (3) there are four single complex roots (both the real parts and the imaginary parts of these complex roots are non-zero). In cases (1) and (2), there may exist twofold repeated roots at some special points.

Substituting (5.6) into (5.12), we find that the coefficients of (5.12) depend only on the stress value γ and two parameters ρ and ν_0 . In derivations of (3.6a) and (3.6b), some restrictions on the geometrical parameters have been imposed (see Figure 1). On the other hand, to ensure the validity of the asymptotic expressions of homogeneous solutions (5.6), we require $|\gamma| < 0.16$. Under these constraints and through some calculations, we find that there are two particular stress values $\gamma_m(< 0)$ and $\gamma_p(> 0)$ (dependent on ρ). According to these two values, as γ varies, types of roots of (5.12) can be divided into five cases, listed in Table 1.

Table I. Types of roots of (5.12) varying with γ .

(i) $-0.16 < \gamma < \gamma_m$ (ii) $\gamma = \gamma_m$	Two different pairs of purely imaginary conjugate roots
$\begin{array}{l} \text{(iii)} \gamma_m < \gamma < \gamma_p \end{array}$	Four single complex roots
(iv) $\gamma = \gamma_p$ (v) $\gamma_p < \gamma < 0.16$	Iwo pairs of repeated real roots Four single real roots

In this paper, we study the compression problem (which implies $\gamma < 0$). Then both cases (i) and (ii) are possible. We first study case (ii) and we will show that this case will not happen in general. In this case $\gamma = \gamma_m$, we let the two pairs of repeated roots be $\pm \lambda_1 i$ and the solution expressions for (5.8a) and (5.8b) are given by

$$\begin{cases} \Delta U = (A_1 + A_3 Z) \cos(\lambda_1 Z) + (A_2 + A_4 Z) \sin(\lambda_1 Z), \\ \Delta G = (B_1 + B_3 Z) \cos(\lambda_1 Z) + (B_2 + B_4 Z) \sin(\lambda_1 Z), \end{cases}$$
(5.13)

where A_i and B_i (i = 1, 2, 3, 4) are real constants.

By substituting (5.13) into (5.8a) and (5.8b) and comparing the coefficients of $cos(\lambda Z)$, $sin(\lambda Z)$, $Zcos(\lambda Z)$ and $Zsin(\lambda Z)$, we have

$$A_1 = b_2 B_4, \quad A_2 = -b_2 B_3, \quad A_3 = b_1 B_3, \quad A_4 = b_1 B_4, \\B_1 = b_3 B_4, \quad B_2 = -b_3 B_3,$$
(5.14)

where b_i (i = 1, 2, 3) are in terms of U_h , G_h , ρ , ν_0 and λ_1 whose expressions are omitted for brevity. If we further substitute (5.13) and (5.14) into the boundary condition (5.9), we get a linear system for B_3 and B_4 and it is easy to find that there are only trivial solutions $B_3 = B_4 = 0$ except when $b_2 = b_1 b_3$. However, this only occurs in a very special circumstance (one needs $\gamma = \gamma_m$ but also $b_2 = b_1 b_3$).

In this paper, we shall concentrate on case (i) and in this case we denote the four purely imaginary roots of (5.12) as

$$\pm \lambda_1 \mathbf{i} = \pm \left(\frac{\psi_2 + \sqrt{\psi_2^2 - 4\psi_1\psi_3}}{2\psi_3}\right)^{\frac{1}{2}} \mathbf{i}, \quad \pm \lambda_2 \mathbf{i} = \pm \left(\frac{\psi_2 - \sqrt{\psi_2^2 - 4\psi_1\psi_3}}{2\psi_3}\right)^{\frac{1}{2}} \mathbf{i}. \tag{5.15}$$

Without loss of generality, we take both λ_1 and λ_2 to be positive. For this case, the solution expressions for ΔU and ΔG can be written as

$$\begin{cases} \Delta U = A_1 \cos(\lambda_1 Z) + A_2 \sin(\lambda_1 Z) + A_3 \cos(\lambda_2 Z) + A_4 \sin(\lambda_2 Z), \\ \Delta G = B_1 \cos(\lambda_1 Z) + B_2 \sin(\lambda_1 Z) + B_3 \cos(\lambda_2 Z) + B_4 \sin(\lambda_2 Z), \end{cases}$$
(5.16)

where A_i and B_i (i = 1, 2, 3, 4) are real constants.

By substituting (5.16) into (5.8a) and (5.8b) and comparing the coefficients of $cos(\lambda_i Z)$ and $sin(\lambda_i Z)$ (i = 1, 2), we have

$$B_1 = f(\lambda_1)A_1, \quad B_2 = f(\lambda_1)A_2, \quad B_3 = f(\lambda_2)A_3, \quad B_4 = f(\lambda_2)A_4, \tag{5.17}$$

where $f(\lambda) = (c_1 \lambda^2 - c_5)/c_6$.

Substituting (5.16) into the boundary condition (5.9), and from (5.17), we have

$$M_2(A_1 \quad A_2 \quad A_3 \quad A_4)^{\rm T} = 0, \tag{5.18}$$

where

$$M_2 = \begin{pmatrix} 0 & \lambda_1 & 0 & \lambda_2 \\ -\lambda_1 \sin \lambda_1 & \lambda_1 \cos \lambda_1 & -\lambda_2 \sin \lambda_2 & \lambda_2 \cos \lambda_2 \\ 0 & f(\lambda_1)\lambda_1 & 0 & f(\lambda_2)\lambda_2 \\ -f(\lambda_1)\lambda_1 \sin \lambda_1 & f(\lambda_1)\lambda_1 \cos \lambda_1 & -f(\lambda_2)\lambda_2 \sin \lambda_2 & f(\lambda_2)\lambda_2 \cos \lambda_2 \end{pmatrix}$$

To find nontrivial solutions of ΔU and ΔG , we need

$$\operatorname{Det} M_2 = (f(\lambda_1) - f(\lambda_2))^2 \lambda_1^2 \lambda_2^2 \sin \lambda_1 \sin \lambda_2 = 0.$$
(5.19)

It is easy to see that (5.19) holds in two cases: (i) $f(\lambda_1) = f(\lambda_2)$; (ii) $\lambda_1 = n\pi$ or $\lambda_2 = n\pi$ (where *n* is a positive integer).

We have tested some parameter values and find that in general $f(\lambda_1) = f(\lambda_2)$ together with (5.12) cannot be satisfied. Therefore, in the following we consider the bifurcation points corresponding to $\lambda = n\pi$ i. In the case of $\lambda_1 = n\pi$ or $\lambda_2 = n\pi$, from (5.18), it can be deduced that the solution (5.16) has the following reduced form:

$$\Delta U = C_1 \cos(n\pi Z), \qquad \Delta G = f(n\pi)C_1 \cos(n\pi Z), \qquad (5.20)$$

where C_1 is a constant.

Substituting $\lambda = n\pi i$ into (5.12), we have

$$\psi_3 \pi^4 n^4 - \psi_2 \pi^2 n^2 + \psi_1 = 0, \tag{5.21}$$

which provides the relationship between the critical stress value γ_c , the mode number *n* and the geometrical parameters ρ and ν_0 . Actually, substituting the asymptotic expressions of U_h and G_h (see (5.6)) into the above equation, we have

$$g_3\pi^4 n^4 - g_2\pi^2 n^2 + g_1 = 0, (5.22)$$

where

$$\begin{cases} g_1 = \frac{135}{(\rho\nu_0)^2} - \frac{135}{\rho\nu_0^2} + \left(-\frac{945}{2(\rho\nu_0)^2} + \frac{945}{2\rho\nu_0^2} \right) \gamma_c + \left(-\frac{405}{4(\rho\nu_0)^2} + \frac{405}{4\rho\nu_0^2} \right) \gamma_c^2 + \left(\frac{46,665}{16(\rho\nu_0)^2} - \frac{75}{32\nu_0^2} - \frac{46,365}{16\rho\nu_0^2} \right) \gamma_c^3, \\ g_2 = -\frac{9}{\nu_0} + \left(\frac{135}{2\rho\nu_0} + \frac{99}{2\nu_0} - \frac{135}{\rho^2\nu_0} \right) \gamma_c + \left(-\frac{4545}{32\rho\nu_0} + \frac{789}{8\nu_0} + \frac{4365}{16\rho^2\nu_0} \right) \gamma_c^2 \\ + \left(-\frac{36,585}{256\rho\nu_0} + \frac{1023}{4\nu_0} + \frac{37,395}{128\rho^2\nu_0} \right) \gamma_c^3, \\ g_3 = 9 - \frac{189}{2} \gamma_c + \frac{69}{2} \gamma_c^2 + \frac{8433}{16} \gamma_c^3. \end{cases}$$

Equation (5.22) explicitly provides the relation between the critical stress value γ_c , the mode number n and the geometric parameters ρ and ν_0 . For given mode number n and ν_0 , it determines the relation between γ_c and ρ . For example, for $\nu_0 = 0.1$ and 0.22, we show the dependence of these two quantities for some different mode numbers n in Figure 3. We find that once mode number n is known for a given ν_0 , the critical stress γ_c is always a decreasing function of ρ as can be seen from Figure 3.



Figure 3. The plots of γ_c vs ρ (a measure of the tube thickness) for mode numbers (a) $3 \le n \le 8$ and (b) $2 \le n \le 7$. The arrow indicates the direction of increasing of *n*.



Figure 4. The plots of γ_c vs ν_0 (a measure of the tube slenderness) for mode numbers $2 \le n \le 7$. The arrow indicates the direction of increasing of *n*.

Also, from Figure 3, we see that two curves can intersect at a particular value of ρ , in other words, two bifurcation mode numbers can correspond to the same critical stress value. However, it is not found that more than two modes can correspond to the same critical stress value. Denote the intersection point of the *n*th-mode and *k*th-mode curves as $(\rho^{(n, k)}, \gamma_c^{(n, k)})$; we can observe the following pattern in Figure 3.

For a geometrical parameter ρ , suppose that the (k + 1) th mode has the smallest critical stress value $|\gamma_c^s|$ (i.e. the first mode number is k + 1). If for a particular ρ it happens that (γ_c^s, ρ) is an intersection point, then γ_c^s has to be $\gamma_c^{(k,k+1)}$, in other words, it has to be the intersection point of the neighboring-mode curves. For example, for $\nu_0 = 0.1$ and $\rho = 0.0784001$, the smallest $|\gamma_c^s|$ is attained at the seventh-mode curve. It happens that (γ_c^s, ρ) is an intersection point for this particular ρ , and we can see it is the intersection point of the seventh-mode and sixth-mode curves. When this happens, from (5.18) we can deduce that the solution (5.16) has the following reduced form:

$$\begin{cases} \Delta U = C_1 \cos(k\pi Z) + C_2 \cos((k+1)\pi Z), \\ \Delta G = f(n\pi)C_1 \cos(k\pi Z) + f((k+1)\pi)C_2 \cos((k+1)\pi Z), \end{cases}$$
(5.23)

where C_i (i = 1, 2) are constants.

This observation is very useful, since some interesting phenomena can arise when there are two modes for the same critical stress. These will be addressed in later sections.

In Figure 3, for a given ν_0 , we can deduce the first mode number from the value of ρ . Once ν_0 is given, we can calculate all $\rho^{(k, k+1)}$ (k = 1, 2, 3, ...). One observation from Figure 3 is that for a given ρ lying between $\rho^{(k, k+1)}$ and $\rho^{(k+1, k+2)}$, $|\gamma_c^s|$ is always attained at the (k + 1) th-mode curve. So, the first bifurcation mode is determined. We also observe that the difference between $\rho^{(k, k+1)}$ and $\rho^{(k+1, k+2)}$ becomes smaller as ρ decreases. Thus, for a thin tube it easily happens that two bifurcation modes correspond to the same critical stress.

Similarly, when the mode number *n* and ρ are given, we can determine the relation between γ_c and ν_0 from (5.22). This relation is shown in Figure 4 for some different mode numbers *n*. This figure can be used to find the first mode number from the value of ν_0 . Denote the intersection point of the *n*th-mode and *k*th-mode curves as $(\nu_0^{(n, k)}, \gamma_c^{(n, k)})$ and suppose the (k + 1)th mode has the smallest critical stress value $|\gamma_c^s|$. We can observe the following pattern in Figure 4. For a given ν_0 lying between $\nu_0^{(k, k+1)}$ and $\nu_0^{(k+1, k+2)}$, $|\gamma_c^s|$ is attained at the (k + 1)th-mode curve. We also observe that the smallest values of $|\gamma_c^s|$ for each mode number are almost the same (see the horizontal dashed lines in Figure 4). This implies that for a given ρ the critical stress has a uniform lower bound γ_{cl} for all ν_0 ; see Figure 4. In particular, if both the inner and outer radii of the tube are fixed (then ρ is fixed), the critical stress has a uniform lower bound for all different tube lengths. This implies that, when the compression stress is below γ_{ch} no bifurcation to axisymmetrical modes can take place no matter what the length of the tube is.



Figure 5. The curve of the critical stress γ_c as a continuous function of *n* when $\nu_0 = 0.1$ and $\rho = 0.13$.

Of course, all the conclusions are based on the constraints of the geometrical parameters imposed at the end of Section 3.

When both ν_0 and ρ are given, one can find the first mode number from the curve of the critical stress by regarding it as a continuous function of the mode number, which is shown in Figure 5 for $\nu_0 = 0.1$ and $\rho = 0.13$. From this figure, we find that the first mode number should be n = 5, which can also be observed from Figure 3(a) or Figure 4(a).

In the above linear analysis, we have presented some relations between the critical stress value, mode number and geometrical parameters. In the following section, we will obtain the approximate postbifurcation solutions by using the method of multiple scales.

6. Analytical post-bifurcation solutions and discussions

A linear bifurcation analysis can yield the critical loads for bifurcation and the eigenmodes. However, one has to conduct a nonlinear analysis to understand the post-bifurcation behavior. However, it seems impossible to find the exact analytical solutions for the complicated coupled nonlinear ODE system (4.3a), (4.3b) and (5.3). When the external stress is close to the critical stress value γ_c , as the buckling amplitude is small, it is possible to find the approximate analytical solutions by perturbation methods. In this section, some nonlinear bifurcation analyses will be carried out in the vicinity of the bifurcation states governed by the coupled system (4.3a) and (4.3b) with the boundary conditions (5.3) at near-critical loads.

To proceed with the perturbation method, we first need to introduce a small parameter. Let U_h and G_h denote the solutions for a homogeneous deformation of (4.3a) and (4.3b) when $\gamma = \gamma_c$; in other words, U_h and G_h satisfy (5.5a) and (5.5b) with $\gamma = \gamma_c$. For the present problem, we seek the solutions in the form of the following perturbation expansions:

$$U = U_h + \zeta U^{(0)} + \zeta^2 U^{(1)} + \zeta^3 U^{(2)} + \dots,$$

$$G = G_h + \zeta G^{(0)} + \zeta^2 G^{(1)} + \zeta^3 G^{(2)} + \dots,$$

$$\gamma = \gamma_c + \Delta \gamma, \qquad \Delta \gamma = \zeta^2 \hat{\gamma},$$

(6.1)

where ζ is a small positive parameter representing the amplitude of the difference $U - U_h$ and $\Delta \gamma = \zeta^2 \hat{\gamma}$ is the incremental stress value on γ_c . The above assumptions imply that the amplitude of $U - U_h$ is of the same order as $G - G_h$ and is of order $\sqrt{|\Delta \gamma|}$. This order relation is determined by requiring that $\hat{\gamma}$ should appear in the amplitude equation for $U^{(0)}$, which will become clear in the following derivations.

In the following, we use the method of multiple scales with two variables (see [31]) to find the approximate analytical solutions of (4.3a) and (4.3b). We first introduce

$$Z_0 = Z, \qquad Z_1 = \zeta^2 Z.$$
 (6.2)

Substituting (6.1) into (4.3a) and (4.3b) and equating the coefficients of the like powers of ζ to zero, we obtain a system of equations. As expected, the coefficients of ζ^0 yield (5.5a) and (5.5b) with $\gamma = \gamma_c$, which are automatically satisfied.

(1) From the coefficients of ζ , we have

$$c_1 G_{Z_0 Z_0}^{(0)} + c_2 G^{(0)} + c_3 U_{Z_0 Z_0}^{(0)} + c_4 U^{(0)} = 0, (6.3a)$$

$$c_1 U_{Z_0 Z_0}^{(0)} + c_5 U^{(0)} + c_6 G^{(0)} = 0.$$
(6.3b)

Note that the above equations are similar to (5.8a) and (5.8b).

One can solve $G^{(0)}$ from the second equation, which is then substituted into the first equation to give

$$\psi_3 U_{Z_0 Z_0 Z_0 Z_0}^{(0)} + \psi_2 U_{Z_0 Z_0}^{(0)} + \psi_1 U^{(0)} = 0.$$
(6.4)

The characteristic equation of (6.4) is simply (5.12), as expected. Therefore, the roots are given by (5.15). Then the solution expressions of $U^{(0)}$ and $G^{(0)}$ are

$$U^{(0)} = K_1(Z_1)e^{i\lambda_1 Z_0} + K_2(Z_1)e^{i\lambda_2 Z_0} + CC,$$
(6.5a)

$$G^{(0)} = f(\lambda_1)K_1(Z_1)e^{i\lambda_1 Z_0} + f(\lambda_2)K_2(Z_1)e^{i\lambda_2 Z_0} + CC,$$
(6.5b)

where *CC* denotes the complex conjugate of the previous terms, and $K_1(Z_1)$ and $K_2(Z_1)$ are two functions to be determined.

(2) From the coefficients of ζ^2 , we have

$$c_{1}G_{Z_{0}Z_{0}}^{(1)} + c_{2}G^{(1)} + c_{3}U_{Z_{0}Z_{0}}^{(1)} + c_{4}U^{(1)} - \frac{16}{9}\rho\nu_{0}(G_{Z_{0}}^{(0)})^{2} - \frac{16}{9}\rho\nu_{0}G^{(0)}G_{Z_{0}Z_{0}}^{(0)} - \frac{1}{9}\rho\nu_{0}G_{Z_{0}Z_{0}}^{(0)}U^{(0)} - \frac{19}{36}\rho\nu_{0}(U_{Z_{0}}^{(0)})^{2} + c_{7}(U^{(0)})^{2} + c_{8}G^{(0)}U^{(0)} + c_{9}(G^{(0)})^{2} + c_{10}G_{Z_{0}}^{(0)}U_{Z_{0}}^{(0)} + c_{11}G^{(0)}U_{Z_{0}Z_{0}}^{(0)} + c_{12}U^{(0)}U_{Z_{0}Z_{0}}^{(0)} - \frac{1}{6}\nu_{0}(U_{Z_{0}}^{(0)})^{2} = 0,$$
(6.6a)

$$c_{1}U_{Z_{0}Z_{0}}^{(1)} + c_{5}U^{(1)} + c_{6}G^{(1)} - \frac{2}{3}\rho\nu_{0}G_{Z_{0}}^{(0)}U_{Z_{0}}^{(0)} - \frac{1}{24}\rho\nu_{0}(U_{Z_{0}}^{(0)})^{2} - \frac{16}{9}\rho\nu_{0}G^{(0)}U_{Z_{0}Z_{0}}^{(0)} - \frac{2}{3}\nu_{0}(U_{Z_{0}}^{(0)})^{2} - \frac{1}{9}\rho\nu_{0}U^{(0)}U_{Z_{0}Z_{0}}^{(0)} + c_{13}(U^{(0)})^{2} + c_{14}G^{(0)}U^{(0)} + c_{15}(G^{(0)})^{2} = -\frac{5}{4}\hat{\gamma}.$$
(6.6b)

From the second equation, $G^{(1)}$ can be expressed in terms of $U^{(1)}$. We substitute the expression into the first equation to yield an equation for $U^{(1)}$ which gives

$$U^{(1)} = M_{1}(Z_{1})e^{i\lambda_{1}Z_{0}} + M_{2}(Z_{1})e^{i\lambda_{2}Z_{0}} + \alpha_{1}K_{1}^{2}(Z_{1})e^{2i\lambda_{1}Z_{0}} + \alpha_{2}K_{2}^{2}(Z_{1})e^{2i\lambda_{2}Z_{0}} + \alpha_{3}K_{1}(Z_{1})K_{2}(Z_{1})e^{i(\lambda_{1}+\lambda_{2})Z_{0}} + \alpha_{4}K_{1}^{*}(Z_{1})K_{2}(Z_{1})e^{i(-\lambda_{1}+\lambda_{2})Z_{0}} + CC + \alpha_{0}\hat{\gamma} + \alpha_{5}K_{1}(Z_{1})K_{1}^{*}(Z_{1}) + \alpha_{6}K_{2}(Z_{1})K_{2}^{*}(Z_{1}),$$
(6.7)

where $M_1(Z_1)$ and $M_2(Z_1)$ are two unknown functions, and α_i (i = 0, 1, ..., 6) are constants related to U_h , G_h , ρ and ν_0 . Also, we can obtain the solution expression of $G^{(1)}$:

$$G^{(1)} = f(\lambda_1)M_1(Z_1)e^{i\lambda_1 Z_0} + f(\lambda_2)M_2(Z_1)e^{i\lambda_2 Z_0} + \beta_1 K_1^2(Z_1)e^{2i\lambda_1 Z_0} + \beta_2 K_2^2(Z_1)e^{2i\lambda_2 Z_0} + \beta_3 K_1(Z_1)K_2(Z_1)e^{i(\lambda_1 + \lambda_2) Z_0} + \beta_4 K_1(Z_1)K_2^*(Z_1)e^{i(\lambda_1 - \lambda_2) Z_0} + CC + \beta_0 \hat{\gamma} + \beta_5 K_1(Z_1)K_1^*(Z_1) + \beta_6 K_2(Z_1)K_2^*(Z_1),$$
(6.8)

where $\beta_i (i = 0, 1, ..., 6)$ are constants related to U_h , G_h , ρ and ν_0 .

(3) From the coefficients of ζ^3 , we have

$$c_{1}G_{Z_{0}Z_{0}}^{(2)} + c_{2}G^{(2)} + c_{3}U_{Z_{0}Z_{0}}^{(2)} + c_{4}U^{(2)} - \frac{32}{9}\rho\nu_{0}G_{Z_{0}}^{(0)}G_{Z_{0}}^{(1)} + 2c_{1}G_{Z_{0}Z_{1}}^{(0)} - \frac{16}{9}\rho\nu_{0}G_{Z_{0}Z_{0}}^{(1)}G^{(0)} - \frac{1}{9}\rho\nu_{0}G_{Z_{0}Z_{0}}^{(1)}G^{(0)} + c_{1}U^{(1)} + U_{Z_{0}Z_{0}}^{(1)}(c_{11}G^{(0)} + c_{12}U^{(0)}) + c_{21}(U^{(0)})^{3} + c_{22}G^{(0)}(U^{(0)})^{2} + c_{23}(G^{(0)})^{2}U^{(0)} + c_{24}(G^{(0)})^{3} + 2c_{7}U^{(0)}U^{(1)} + c_{8}G^{(1)}U^{(0)} + c_{8}G^{(0)}U^{(1)} + 2c_{9}G^{(0)}G^{(1)} + c_{10}G_{Z_{0}}^{(1)}U_{Z_{0}}^{(0)} + c_{10}G_{Z_{0}}^{(0)}U_{Z_{0}}^{(1)} + c_{10}G_{Z_{0}}^{(0)}U_{Z_{0}}^{(1)} + c_{25}U_{Z_{0}}^{(0)}U_{Z_{0}}^{(1)} + 2c_{3}U_{Z_{0}Z_{1}}^{(0)} + U_{Z_{0}Z_{0}}^{(0)}U_{Z_{0}}^{(1)} + c_{10}G_{Z_{0}}^{(0)}U_{Z_{0}}^{(1)} + c_{25}U_{Z_{0}}^{(0)}U_{Z_{0}}^{(1)} + 2c_{3}U_{Z_{0}}^{(0)}U_{Z_{0}}^{(1)} + U_{Z_{0}Z_{0}}^{(1)}U_{Z_{0}Z_{0}}^{(0)} + c_{1}G_{Z_{0}}^{(0)}U_{Z_{0}}^{(1)} + c$$

We can express $G^{(2)}$ in terms of $U^{(2)}$ from the second equation and then substitute the expression into the first equation to yield an equation for $U^{(2)}$. To remove secular terms, we set $M_1(Z_1) = M_2(Z_1) = 0$ and

$$\begin{cases} s_0 \hat{\gamma} K_1(Z_1) + s_1 K_1^2(Z_1) K_1^*(Z_1) + s_2 K_1(Z_1) K_2(Z_1) K_2^*(Z_1) + \mathbf{i} \, s_3 \frac{dK_1}{dZ_1} = 0, \\ t_0 \hat{\gamma} K_2(Z_1) + t_1 K_2^2(Z_1) K_2^*(Z_1) + t_2 K_2(Z_1) K_1(Z_1) K_1^*(Z_1) + \mathbf{i} \, t_3 \frac{dK_2}{dZ_1} = 0, \end{cases}$$
(6.10)

where s_i and t_i (i = 0, 1, 2, 3) are constants related to U_h , G_h , ρ and ν_0 . Substituting $K_1(Z_1) = R_1(Z_1)e^{i\theta_1(Z_1)}$ and $K_2(Z_1) = R_2(Z_1)e^{i\theta_2(Z_1)}$ into (6.10), we obtain

$$R_1(Z_1) = R_1$$
(constant), $R_2(Z_1) = R_2$ (constant) (6.11)

and

$$R_1\left(s_0\hat{\gamma} + s_1R_1^2 + s_2R_2^2 - s_3\frac{d\theta_1}{dZ_1}\right) = 0, \quad R_2\left(t_0\hat{\gamma} + t_1R_2^2 + t_2R_1^2 - t_3\frac{d\theta_2}{dZ_1}\right) = 0, \quad (6.12)$$

which provide the equations for the amplitudes of post-bifurcation solutions when θ_1 and θ_2 are determined. From (6.12), we also have

$$\theta_1(Z_1) = \frac{1}{s_3} \left(s_0 \hat{\gamma} + s_1 R_1^2 + s_2 R_2^2 \right) Z_1 + \phi_1, \quad \text{if} \quad R_1 \neq 0, \tag{6.13}$$

$$\theta_2(Z_1) = \frac{1}{t_3} \left(t_0 \hat{\gamma} + t_1 R_2^2 + t_2 R_1^2 \right) Z_1 + \phi_2, \quad \text{if} \quad R_2 \neq 0, \tag{6.14}$$

where ϕ_1 and ϕ_2 are two integration constants.

To summarize, after the above calculations we have obtained the expressions for the first two terms:

$$\begin{cases} U^{(0)} = 2R_1 \cos(\lambda_1 Z_0 + \theta_1(Z_1)) + 2R_2 \cos(\lambda_2 Z_0 + \theta_2(Z_1)), \\ G^{(0)} = 2f(\lambda_1)R_1 \cos(\lambda_1 Z_0 + \theta_1(Z_1)) + 2f(\lambda_2)R_2 \cos(\lambda_2 Z_0 + \theta_2(Z_1)) \end{cases}$$
(6.15)

and

$$\begin{cases} U^{(1)} = \alpha_0 \hat{\gamma} + (2\alpha_1 \cos(2\lambda_1 Z_0) + \alpha_5) R_1^2 + (2\alpha_2 \cos(2\lambda_2 Z_0) + \alpha_6) R_2^2 \\ + 2(\alpha_3 \cos((\lambda_1 + \lambda_2) Z_0 + \phi_1 + \phi_2) + \alpha_4 \cos((-\lambda_1 + \lambda_2) Z_0 - \phi_1 + \phi_2)) R_1 R_2, \\ G^{(1)} = \beta_0 \hat{\gamma} + (2\beta_1 \cos(2\lambda_1 Z_0 + 2\theta_1 (Z_1)) + \beta_5) R_1^2 + (2\beta_2 \cos(2\lambda_2 Z_0 + 2\theta_2 (Z_1)) + \beta_6) R_2^2 \\ + 2(\beta_3 \cos((\lambda_1 + \lambda_2) Z_0 + \theta_1 (Z_1) + \theta_2 (Z_1)) \\ + \beta_4 \cos((\lambda_1 - \lambda_2) Z_0 + \theta_1 (Z_1) - \theta_2 (Z_1))) R_1 R_2, \end{cases}$$
(6.16)

In the above obtained solutions, there are four unknown constants R_1 , R_2 , ϕ_1 and ϕ_2 , which can be determined by the end boundary conditions.

Obviously, at O(1) the end boundary conditions (5.3) are automatically satisfied. At $O(\zeta)$, we need

$$U_{Z_0}^{(0)}(0,0) = 0, \quad G_{Z_0}^{(0)}(0,0) = 0, \quad U_{Z_0}^{(0)}(1,\zeta^2) = 0, \quad G_{Z_0}^{(0)}(1,\zeta^2) = 0.$$
 (6.17)

To find the nontrivial solutions, we require that at least one of R_1 and R_2 is non-zero; in other words, we will discuss the following two cases: (1) one of R_1 and R_2 is zero; (2) both R_1 and R_2 are non-zero.

6.1. $R_1 \neq 0$ and $R_2 = 0$

In this subsection, we discuss the case where one of R_1 and R_2 equals zero. Without loss of generality, we let $R_1 \neq 0$ and $R_2 = 0$. For this case, there are two unknown constants R_1 and ϕ_1 , which can be determined by the end conditions (6.17). To satisfy (6.17), from (6.15) and (6.13), we have

$$\begin{cases} \sin \phi_1 = 0, \\ f(\lambda_1) \sin \phi_1 = 0, \\ \sin \left(\lambda_1 + \frac{1}{s_3} \left(s_0 \hat{\gamma} + s_1 R_1^2 \right) \zeta^2 + \phi_1 \right) = 0, \\ f(\lambda_1) \sin \left(\lambda_1 + \frac{1}{s_3} \left(s_0 \hat{\gamma} + s_1 R_1^2 \right) \zeta^2 + \phi_1 \right) = 0. \end{cases}$$
(6.18)

We note that when $(6.18)_1$ is satisfied, $(6.18)_2$ is automatically satisfied. From $(6.18)_1$, we have

$$\phi_1 = m_1 \pi$$
 with $m_1 = 0, \pm 1, \pm 2, \dots$ (6.19)

Also, when $(6.18)_3$ is satisfied, $(6.18)_4$ is automatically satisfied. We note that $(6.18)_3$ should be satisfied for any $\hat{\gamma}$, and in particular we take $\hat{\gamma} \to 0$ (which implies that $R_1 \to 0$) to obtain $\sin(\lambda_1 + \phi_1) = 0$ which yields that

$$\lambda_1 = n_1 \pi$$
 with $n_1 = 1, 2, \dots$ (6.20)

(we have imposed that $\lambda_1 > 0$). Substituting (6.19) and (6.20) into (6.18)₃, we obtain

$$R_1^2 = r_1 \hat{\gamma},\tag{6.21}$$

where $r_1 = -s_0/s_1$. This amplitude equation for R_1 can also be obtained by substituting $\theta_1 = \phi_1$ and $R_2 = 0$ into (6.12).

We recall that $\lambda_1 i = n_1 \pi i$ is a root of the characteristic equation of (6.4) (i.e. it satisfies (5.12)). Letting $\lambda = n\pi i$ in (5.12), we recover (5.22), which can be used to determine the relationship between the critical stress value and the mode number. Thus, the nonlinear bifurcation analysis presented in this section can recover the results obtained by the linear bifurcation analysis.

Substituting the expressions of θ_1 , ϕ_1 , λ_1 , R_1 into (6.15) and (6.16), we obtain

$$\begin{cases} U \approx U_h + 2(-1)^{m_1} (r_1 \Delta \gamma)^{1/2} \cos(n_1 \pi Z) + (\alpha_0 + (2\alpha_1 \cos(2n_1 \pi Z) + \alpha_5) r_1) \Delta \gamma, \\ G \approx G_h + 2f(n_1 \pi) (-1)^{m_1} (r_1 \Delta \gamma)^{1/2} \cos(n_1 \pi Z) + (\beta_0 + (2\beta_1 \cos(2n_1 \pi Z) + \beta_5) r_1) \Delta \gamma, \end{cases}$$
(6.22)

which are the analytical expressions for describing the post-bifurcation states when $\Delta \gamma$ is small.

Similarly, when $R_1 = 0$ and $R_2 \neq 0$, the approximate solutions of U and G are given by

$$\begin{cases} U \approx U_h + 2(-1)^{m_2} (r_2 \Delta \gamma)^{1/2} \cos(n_2 \pi Z) + (\alpha_0 + (2\alpha_1 \cos(2n_2 \pi Z) + \alpha_6) r_2) \Delta \gamma, \\ G \approx G_h + 2f(n_2 \pi) (-1)^{m_2} (r_2 \Delta \gamma)^{1/2} \cos(n_2 \pi Z) + (\beta_0 + (2\beta_2 \cos(2n_2 \pi Z) + \beta_6) r_2) \Delta \gamma, \end{cases}$$
(6.23)

where $r_2 = -t_0/t_1$, $m_2 = 0, \pm 1, \pm 2, \dots$ and $n_2 = 1, 2, \dots$

The simple analytical expressions (6.22) and (6.23) can provide both qualitative and quantitative information on the post-bifurcation states (in general, it is very difficult, if even possible, to deduce qualitative information from numerical solutions). Now, we give some discussions. It should be noted that the information below cannot be deduced from a linear bifurcation analysis.

First, from the expression (6.21) for the amplitude, we see that the sign of r_1 , which depends on U_h , G_h , ρ and ν_0 , determines the nature of the bifurcations. If $r_1 < 0$, we need $\Delta \gamma = \zeta^2 \hat{\gamma} < 0$ to ensure the positivity of R_1^2 . Thus the bifurcation occurs only when $|\gamma| > |\gamma_c|$, which is a supercritical bifurcation. Conversely, if $r_1 > 0$, we need $\Delta \gamma = \zeta^2 \hat{\gamma} > 0$ and the bifurcation occurs only when $|\gamma| < |\gamma_c|$, which is a supercritical bifurcation. For the geometrical parameters considered here, we find that r_1 is always negative, and thus all bifurcations in the case of $R_1 \neq 0$ and $R_2 = 0$ are supercritical.

Also, in the present case we see that the solutions (6.22) and (6.23) contain m_1 or m_2 . Since m_1 and m_2 can be even or odd, there are two nontrivial solutions for a given mode number. This implies that it is a supercritical pitchfork bifurcation (see Figure 12).

It is expected that the geometrical parameters have important influences on the post-bifurcation states. For the energy absorption purpose, usually one would like the structure to be as 'soft' as possible in the post-bifurcation regime. Here, we say that a structure is 'soft' if for a reduction of the length of the tube the end stress is small. The usual thinking is that a thin structure is softer than a thick structure. However, it may not be true in the post-bifurcation regime. As we shall see below, increasing the thickness of the tube slightly can reduce the stress significantly.

We denote the leading-order amplitudes of U and G as d_1 and d_2 respectively. It can be seen from (6.22) and (6.23) that d_1 can be described through $2|r_1|^{1/2}$ or $2|r_2|^{1/2}$ and d_2 through $2f(n_1\pi)|r_1|^{1/2}$ or $2f(n_2\pi)|r_2|^{1/2}$ (which are dependent on the geometrical parameters ρ and ν_0 and the mode number; please note that U_h and G_h are dependent on γ_c , which, in turn, is determined by ρ and ν_0 and the mode number (see (5.22)).

In Figure 6, we plot the curves of d_1 and d_2 for $\nu_0 = 0.1$ (corresponding to Figure 3(a)). Please note that for $\rho^{(k+1,k+2)} < \rho < \rho^{(k,k+1)}$ (k = 3, ..., 8 for $\nu_0 = 0.1$) the first mode number is k (see the discussions above (5.23)). As a result, we only draw the curves corresponding to the kth mode for ρ in this interval. From this figure, we find that both d_1 and d_2 are increasing functions of ρ for $\rho^{(k+1,k+2)} < \rho < \rho^{(k,k+1)}$. Then, corresponding to the same mode number, it is expected that as the engineering stress increases the current length of the thicker tube after bifurcation will decrease faster than that of the thinner tube.



Figure 6. Plots of d_1 and d_2 vs ρ for $\nu_0 = 0.1$.



Figure 7. Plot of the length of the tube *l* vs the engineering stress γ when $\nu_0 = 0.1$ (full lines are for $\rho = 0.105$ and dashed lines are for $\rho = 0.135$).

In Figure 7, we give the plots of the current length *l* of the tube after bifurcation versus the engineering stress γ for two slightly different tube thicknesses (however, the first bifurcation mode is still the same). In Figure 7, curves are plotted for $\rho = 0.105$ and $\rho = 0.135$ both of which give the same first bifurcation mode n = 5 as we can see from Figure 6(a). We note that the difference between the two thicknesses is about 15%. The thinner tube will bulge first at a smaller critical stress $|\gamma_c^{(1)}|$ than that of the thicker tube (i.e. $|\gamma_c^{(2)}|$) as expected. After the bifurcations, the two curves intersect at (γ^*, l^*) . We observe that when $|\gamma_c^{(1)}| < |\gamma| < |\gamma^*|$, the engineering stress of the thicker tube is larger than that of the thinner tube for the same length of the tube, so the thinner tube is 'softer'. When $|\gamma| > |\gamma^*|$, for the same tube length (or the same reduction of its original length) the engineering stress in the thicker tube is significantly smaller than that in the thinner tube. The difference in stress can be large when *l* is getting smaller. Thus, in this post-bifurcation regime the thicker tube is softer than the thinner tube. We can see that the geometry (thickness) of the tube can affect the softness of the tube greatly, and this effect may be useful in designing a tube as an energy absorber.

Remarks:

1. For this case, the amplitude equation obtained from (6.12) is

$$s_0 \hat{\gamma} R_1 + s_1 R_1^3 = 0.$$

For this kind of amplitude equation, it has been shown by the singularities theory in [32] that for a supercritical bifurcation the structure is imperfection-insensitive and for a subcritical bifurcation it is imperfection-sensitive. Since we have a supercritical pitchfork bifurcation here, the tube is imperfection-insensitive in the case discussed in this section.

2. Since there are multiple solutions when the external stress exceeds the critical stress value, we need to calculate the total elastic potential energy to determine which solution has the smallest value, that is, which is the energetically preferred solution. For the current force-controlled problem, the asymptotic expression of the total elastic potential energy of the tube is given by

$$\begin{split} J &= \pi \delta \mu_0 \int_0^1 \left(\frac{4}{3} U^2 - \frac{52}{27} U^3 + \frac{199}{81} U^4 + G\left(\frac{2}{3} U - \frac{1}{9} U^2 + \frac{4}{81} U^3 \right) \\ &+ G^2 \left(\frac{4}{3} - \frac{1}{9} U + \frac{1}{54} U^2 \right) + G^3 \left(-\frac{52}{27} + \frac{4U}{81} \right) + \frac{199}{81} G^4 + \nu_0 \left(\frac{1}{3} U U_Z^2 + \frac{4}{3} U_Z^2 G \right) \\ &+ \delta \left(-\frac{5}{9} U U_Z^2 - \frac{1}{6} U U_{ZZ} + \frac{1}{12} U^2 U_{ZZ} + G \left(-\frac{5}{36} U_Z^2 - \frac{2}{3} U_{ZZ} + \frac{1}{2} U U_{ZZ} \right) \right) \\ &+ \frac{7}{3} U_{ZZ} G^2 + \frac{1}{6} U U_Z G_Z + \frac{2}{3} U_Z G G_Z + \frac{1}{\nu_0} \left(-\frac{8}{9} U^2 + \frac{56}{27} U^3 - \frac{902}{243} U^4 \right) \\ &+ G \left(-\frac{4}{9} U + \frac{2}{3} U^2 - \frac{272}{243} U^3 \right) + G^2 \left(-\frac{1}{18} + \frac{2}{9} U - \frac{34}{81} U^2 \right) \\ &+ G^3 \left(\frac{5}{108} - \frac{79}{486} U \right) - \frac{19}{486} G^4 \right) \bigg) dZ - \frac{5}{2} \pi \delta \mu_0 \gamma \int_0^1 G \, dZ. \end{split}$$

By using (6.24), the approximate analytical solutions (6.22) and (6.23) for the post-bifurcation solutions and the algebraic equations (5.5a) and (5.5b) for the trivial solutions, we can calculate the total potential energy values for nontrivial solutions and the corresponding trivial solutions. For some parameter values, the results are listed in Table 2.

In Table 2 we find that for each case two nontrivial solutions have the same energy value, which is less than that of the trivial solution. Then two nontrivial solutions are both the preferred solution. An unfortunate side is that the present theory is unable to determine whether the solution with $m_1 = 1$ or that with $m_1 = 2$ is more preferred, although the solution profiles are very different.

6.2. $R_1 \neq 0$ and $R_2 \neq 0$

In this subsection, we discuss the case where both R_1 and R_2 do not equal zero. For this case, there are four unknown constants R_1 , R_2 , ϕ_1 and ϕ_2 , and they can also be determined by the end conditions (6.17). By using an analysis similar to that in the above section, we can obtain

$$\phi_1 = m_1 \pi, \quad \phi_2 = m_2 \pi \quad \text{with} \quad m_1, m_2 = 0, \pm 1, \pm 2, \dots,$$
 (6.24)

$$\lambda_1 = n_1 \pi, \quad \lambda_2 = n_2 \pi \quad \text{with} \quad n_1, n_2 = 1, 2, \dots (n_1 \neq n_2)$$
 (6.25)

and

$$s_0\hat{\gamma} + s_1R_1^2 + s_2R_2^2 = 0, \qquad t_0\hat{\gamma} + t_1R_2^2 + t_2R_1^2 = 0.$$
 (6.26)

From (6.13), (6.14) and (6.26), we have $\theta_1 = \phi_1$ and $\theta_2 = \phi_2$.

Solving the amplitudes equations (6.26), we have

$$R_1^2 = r_1 \hat{\gamma}, \qquad R_2^2 = r_2 \hat{\gamma},$$
 (6.27)

where $r_1 = (t_0 s_2 - t_1 s_0)/(s_1 t_1 - s_2 t_2)$ and $r_2 = (s_0 t_2 - s_1 t_0)/(s_1 t_1 - s_2 t_2)$.

We note that both $\lambda_1 i = n_1 \pi i$ and $\lambda_2 i = n_2 \pi i$ $(n_1 \neq n_2)$ should satisfy the characteristic equation of (6.4) simultaneously. That is, both n_1 and n_2 satisfy (5.22). Thus, when both $R_1 \neq 0$ and $R_2 \neq 0$, there

Table 2.	Total	potential	energy va	alues	scaled	by	$\pi\delta\mu_0$
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	Nontrivial solutions	Trivial solution	
(1) $\nu_0 = 0.1, \rho = 0.16, \gamma = -0.13$ (2) $\nu_0 = 0.1, \rho = 0.13, \gamma = -0.096$ (3) $\nu_0 = 0.1, \rho = 0.1, \gamma = -0.07$	$\begin{array}{c} -0.0211568 \ (m_1 = 1, 2) \\ -0.0112424 \ (m_2 = 1, 2) \\ -0.0075429 \ (m_2 = 1, 2) \end{array}$	-0.0178934 -0.010141 -0.0055647	

are two different mode numbers corresponding to the same critical stress value. It should be noted that this case is a special one, as for it to happen there is a relation between the two geometrical parameters ν_0 and ρ (since both n_1 and n_2 satisfy (5.22)).

After substituting the expressions of θ_i , ϕ_i , λ_i and R_i (i = 1, 2) into (6.15) and (6.16), we obtain

$$U \approx U_{h} + 2(-1)^{m_{1}}(r_{1}\Delta\gamma)^{1/2}\cos(n_{1}\pi Z) + 2(-1)^{m_{2}}(r_{2}\Delta\gamma)^{1/2}\cos(n_{2}\pi Z) + (\alpha_{0} + (2\alpha_{1}\cos(2n_{1}\pi Z) + \alpha_{5})r_{1} + (2\alpha_{2}\cos(2n_{2}\pi Z) + \alpha_{6})r_{2})\Delta\gamma$$
(6.28a)
$$-2(-1)^{m_{1}+m_{2}}(\alpha_{3}\cos((n_{1}+n_{2})\pi Z) + \alpha_{4}\cos((n_{1}-n_{2})\pi Z))(r_{1}r_{2})^{1/2}\Delta\gamma,$$

$$G_{h} + 2f(n_{1}\pi)(-1)^{m_{1}}(r_{1}\Delta\gamma)^{1/2}\cos(n_{1}\pi Z) + 2f(n_{2}\pi)(-1)^{m_{2}}(r_{2}\Delta\gamma)^{1/2}\cos(n_{2}\pi Z)$$

$$G \approx G_{h} + 2f'(n_{1}\pi)(-1)^{m_{1}}(r_{1}\Delta\gamma)^{1/2}\cos(n_{1}\pi Z) + 2f'(n_{2}\pi)(-1)^{m_{2}}(r_{2}\Delta\gamma)^{1/2}\cos(n_{2}\pi Z) + (\beta_{0} + (2\beta_{1}\cos(2n_{1}\pi Z) + \beta_{5})r_{1} + (2\beta_{2}\cos(2n_{2}\pi Z) + \beta_{6})r_{2})\Delta\gamma$$
(6.28b)
$$-2(-1)^{m_{1}+m_{2}}(\beta_{3}\cos((n_{1}+n_{2})\pi Z) + \beta_{4}\cos((n_{1}-n_{2})\pi Z))(r_{1}r_{2})^{1/2}\Delta\gamma,$$

which are the analytical solutions for describing four ($m_1 = 1, 2$ and $m_2 = 1, 2$) post-bifurcation states for small $\Delta \gamma$ when both n_1 and n_2 satisfy (5.22).

Now, we discuss some bifurcation features for this case.

First, from (6.27) we see that we need both $r_1 < 0$ and $r_2 < 0$ or both $r_1 > 0$ and $r_2 > 0$, which correspond to a supercritical bifurcation and a subcritical bifurcation, respectively. For the present problem we find that within the imposed constraint on the geometrical parameters we always have $r_1 > 0$ and $r_2 > 0$. Thus, those solutions given in (6.28) represent subcritical branches.

Also, we note that when n_1 and n_2 satisfy (5.22) there are also another four post-bifurcation states corresponding to $R_1 \neq 0$ and $R_2 = 0$, and $R_1 = 0$ and $R_2 \neq 0$, since (6.17) is also satisfied for those cases. The solution expressions are provided by (6.22) ($m_1 = 1, 2$) and (6.23) ($m_2 = 1, 2$). Each of these four branches is supercritical, as discussed in the previous section. Thus, in the case of two different modes corresponding to the same critical stress there are in total eight post-bifurcation states around the bifurcation point, four subcritical branches and four supercritical branches. We call such a bifurcation an *octopus* bifurcation (see Figure 15).

The study on the imperfection sensitivity for this case, which needs certain tools in singularities theory, is presented in the next section.

Remark: In obtaining the analytical solutions for the above two different cases, we only use the sliding boundary conditions at $O(\zeta)$. It is easy to verify that the obtained solutions also satisfy the conditions at $O(\zeta^2)$.

7. Imperfection sensitivity for the case of two modes corresponding to a critical stress

Subcritical bifurcation branches occur when two modes correspond to the same critical stress value. The important feature of a subcritical bifurcation is the imperfection sensitivity. In this section, we will use singularities theory to study this important issue for this case. We take $\nu_0 = 0.1$ and $\rho = 0.103995$ for example (correspondingly the two modes are $n_1 = 5$, $n_2 = 6$ and the critical stress is $\gamma_c = -0.0722086$) and follow the discussions in [22, 23].

In Section 6 we obtained the amplitudes equations (6.21) when $R_1 \neq 0$ and $R_2 = 0$ and (6.26) when $R_1 \neq 0$ and $R_2 \neq 0$. The amplitudes equations, which include both the trivial solutions and the nontrivial solutions of R_1 and R_2 , are given by (see (6.12))

$$s_0 \hat{\gamma} R_1 + s_1 R_1^3 + s_2 R_1 R_2^2 = 0, \qquad t_0 \hat{\gamma} R_2 + t_1 R_2^3 + t_2 R_2 R_1^2 = 0.$$
(7.1)

For the parameters chosen above, we find that the non-degeneracy conditions defined in Definition 2.2 [23, p. 423] are satisfied. Then according to Proposition 2.3 in [23, p. 424], the bifurcation problem of (7.1) with two state variables has the normal form

$$\varepsilon_1 R_1^3 + p R_1 R_2^2 + \varepsilon_2 (-\hat{\gamma}) R_1 = 0, \qquad q R_2 R_1^2 + \varepsilon_3 R_2^3 + \varepsilon_4 (-\hat{\gamma}) R_2 = 0, \tag{7.2}$$

where

$$\varepsilon_1 = \operatorname{sgn}(s_1), \quad \varepsilon_2 = \operatorname{sgn}(-s_0), \quad \varepsilon_3 = \operatorname{sgn}(t_1), \quad \varepsilon_4 = \operatorname{sgn}(-t_0),$$
$$p = \left| \frac{t_0}{t_1 s_0} \right| s_2, \quad q = \left| \frac{s_0}{s_1 t_0} \right| t_2, \quad p \neq \varepsilon_2 \varepsilon_3 \varepsilon_4, \quad q \neq \varepsilon_1 \varepsilon_2 \varepsilon_4, \quad pq \neq \varepsilon_1 \varepsilon_3.$$

From Theorem 2.4 in [23, p. 425], the normal form (7.2) has the following universal unfolding:

$$\varepsilon_1 R_1^3 + p R_1 R_2^2 + \varepsilon_2 (-\hat{\gamma}) R_1 = 0, \qquad q R_2 R_1^2 + \varepsilon_3 R_2^3 + \varepsilon_4 (-\hat{\gamma} - \sigma) R_2 = 0, \tag{7.3}$$

where σ is a small parameter representing the imperfection magnitude.

For the chosen ρ and ν_0 , we find that

$$\varepsilon_1 = \varepsilon_3 = 1, \qquad \varepsilon_2 = \varepsilon_4 = -1. \tag{7.4}$$

For the above case, a detailed discussion on the bifurcation diagrams with different values of p and q has been given in [23]. For $\nu_0 = 0.1$ and $\rho = 0.103995$, we find that p = -15.12 and q = -4.71, and the corresponding unperturbed and perturbed bifurcation diagrams are given in Figures 8 and 9. The solid lines represent the stable solutions and the dashed lines represent the unstable solutions.

From Figure 9, we find that when $\sigma > 0$, the real critical stress value is not changed, and so this value is not imperfection-sensitive. In this case, the stable branch is the supercritical R_1 -mode. However, with a further increase of the load (an $O(\sigma)$ amount), this mode also becomes unstable. Then, the tube has to jump to a state corresponding to a remote stable solution (which cannot be determined by the present theory). In this sense, we may say that the structure is imperfection-sensitive.

When $\sigma < 0$, the real critical stress value becomes

$$\gamma^* = \gamma_c - \sigma, \tag{7.5}$$

whose absolute value is smaller than that of the unperturbed critical stress value. So, this value is imperfection-sensitive. Afterwards, when the load increases, the solution goes along the supercritical R_2 -mode. However, with a further increase of the load (an $O(\sigma)$ amount), this branch becomes unstable. Then, the tube has to jump to a state corresponding to a remote stable solution (which cannot be determined by the present theory). Thus, the tube is imperfection-sensitive.

As discussed in Section 5, for a thin tube it very easily happens that two modes correspond to the same critical stress. The theory presented here gives the explanation as to why a thin tube is usually imperfection-sensitive. For a thick tube, only if the two geometrical parameters ρ and ν_0 satisfy a



Figure 8. Unperturbed schematic bifurcation diagram.



Figure 9. Perturbed schematic bifurcation diagrams.

further relation are there two modes corresponding to the same critical stress. Then, in general, a thick tube is imperfection-insensitive.

We remark that imperfection sensitivity may not be a bad thing when the tube is used for energy absorption purposes, since, corresponding to the remote solutions mentioned above, the structure may be softer. Of course, one has to examine the remote solutions to draw such a conclusion, which will be left for future investigation.

8. Numerical solutions and graphic results

In Sections 5 and 6, we have determined the critical stress values of γ_c analytically and obtained the analytical post-bifurcation solutions by the multiple scales method. To assess the validity of the analytical results, in this section we use the bifurcation analysis software AUTO (see [33]) to do numerical experiments, and comparisons of the results obtained by two approaches will be made. Some graphic results will also be presented.

It is relatively straightforward to use AUTO to find the bifurcation points of a system of ODEs with specific boundary conditions, which can then yield the critical stress values for the present problem. Some critical stress values γ_c for the first mode number obtained in Section 5 are compared with those obtained by AUTO in Table 3. We find that the values obtained from the analytical expressions are very close to the numerical values (the largest relative error among three cases is about 3%). Thus, (5.22) provides reliable results for the critical stress values.

To use AUTO to compute post-bifurcation solutions of boundary-value problems could be a nontrivial matter. In fact, we intended to get the numerical solutions by AUTO first and then use them to help us construct the analytical solutions. However, the outcome was not satisfactory since we were not sure whether AUTO missed a branch of solutions. It turned out the other way around. We obtained the analytical solutions first to find all the bifurcated branches, which then helped us define the proper amplitude of the solutions in AUTO programs. And finally, AUTO can yield all the post-bifurcation solutions. The numerical solutions of the third case in Table 3 are plotted in Figure 10.

For the cases listed in Table 3, there is only one mode number corresponding to a critical stress value, and the analytical solutions of U and G are given by (6.22) when $R_2 = 0$ and (6.23) when $R_1 = 0$. The solution curves of the third case in Table 3 are plotted in Figure 10. One can see that there is very good agreement between the numerical solutions obtained by AUTO and the analytical solutions.

			n		γ_c (Nume	erical)		γ_c (Analytical)	
(1) $\nu_0 = 0.1, \rho = 0.16$ (2) $\nu_0 = 0.1, \rho = 0.13$ (3) $\nu_0 = 0.1, \rho = 0.1$		4 5 6			-0.125478 -0.0948662 -0.0680846			-0.121826 -0.0935275 -0.06772	
	0.06	Â			0.06		A		

Table 3.	Comparisons	of the	critical	stress	values.



Figure 10. Comparisons of the analytical solutions (solid curves) of U with the numerical solutions (dots) for $\nu_0 = 0.1$, $\rho = 0.1$, $n_2 = 6$ and $\gamma = -0.07$.



Figure 11. Shapes of the buckled tubes corresponding to the case in Figure 10.



Figure 12. Pitchfork bifurcation diagrams for $\nu_0 = 0.1$ and $\rho = 0.1$. Full lines are the analytical solutions and circles are the numerical solutions.

Table 4.	The c	ritical	stress	values	when	two	modes	exist

	n	γς	
(1) $\nu_0 = 0.1, \rho = 0.103995$	$n_1 = 5, n_2 = 6$	-0.0722086	
(2) $\nu_0 = 0.22, \rho = 0.115096$	$n_1 = 3, n_2 = 4$	-0.0847745	

By using (2.1) and the analytical expressions of U and G, we can plot the three-dimensional shapes of the buckled tubes in the current configuration. The buckled shapes corresponding to the cases in Figure 10 are given in Figure 11.

For given ν_0 , ρ and mode numbers *n* listed in Table 3, we find that these bifurcations are pitchfork and supercritical bifurcations. This type of bifurcation is illustrated in Figure 12, which gives the bifurcation diagrams for case (3) in Table 3. In this figure, the bifurcation diagrams plotted by using the analytical solutions of *U* are compared with the numerical results obtained by AUTO and the amplitude $\varphi(U)$ is defined by

$$\varphi(U) = U(0) - U_h. \tag{8.1}$$

Once again, very good agreement between the analytical solutions and the numerical ones is found.

Those comparisons demonstrate that the simple analytical expressions (6.22) and (6.23) provide reliable results for the post-bifurcation states. We note that when we further increase the external stress it is found that the tube intends to deform with folds.

As we discussed for Figure 3 in Section 5, for the particular value of ρ at the intersection point, two modes correspond to the same critical stress value. Two examples are listed in Table 4.



Figure 13. Plots of the approximate analytical solutions of U when $\nu_0 = 0.1$, $\gamma = -0.066$ and $\rho = 0.103995$ ($\gamma_c = -0.0722086$, $n_1 = 5$ and $n_2 = 6$).



(a) $m_1 = 1, m_2 = 1$ (b) $m_1 = 1, m_2 = 2$ (c) $m_1 = 2, m_2 = 1$ (d) $m_1 = 2, m_2 = 2$

Figure 14. Shapes of the buckled tubes corresponding to the cases in Figure 13 when $\gamma = -0.066$.

For the two cases in Table 4, there are two different modes n_1 and n_2 corresponding to the same critical stress value. Unfortunately, AUTO fails to get the bifurcation point and any branch of solutions for this case (on the other hand, the failure of the numerical approach demonstrates the indispensability of analytical solutions). Therefore, in the following we shall present some graphic results based on the analytical solutions.

According to the analytical results in Section 6, there exist eight nontrivial solution branches for the particular ρ and ν_0 when there are two different modes corresponding to the same critical stress value (two branches corresponding to the unique mode n_1 , two branches corresponding to the unique mode n_2 and four branches corresponding to two modes n_1 and n_2). In Figure 13, we plot the four subcritical post-bifurcation solutions of U by using the corresponding approximate analytical solutions.

Also, by using (2.1) and the analytical expressions of U and G, the shapes of the buckled tubes corresponding to the cases in Figure 13 are given in Figure 14.

As discussed before, the bifurcation corresponding to a unique mode number is a supercritical one and the bifurcation corresponding to two mode numbers is a subcritical one. Thus, the supercritical and the subcritical bifurcations can occur simultaneously when two modes correspond to the same critical



Figure 15. Octopus bifurcation diagram for $\nu_0 = 0.1$ and $\rho = 0.103995$.

stress value. The bifurcation diagrams in Figure 15 are plotted by using the approximate analytical solutions of U. This type of bifurcation was called an octopus bifurcation at the end of Section 6 since there are eight 'legs' (branches) around the bifurcation point.

9. Concluding remarks

Linear and nonlinear bifurcation analysis is conducted for the axisymmetric deformations of a hyperelastic tube subjected to axial compression. The analytical results reveal many fresh insights, including:

- (a) The critical stress has a uniform lower bound (independent of the slenderness) for fixed inner and outer radii.
- (b) For the single-mode case the bifurcation is a supercritical pitchfork bifurcation.
- (c) For the case of two modes corresponding to a critical stress the bifurcation is an octopus bifurcation, which is the mechanism for imperfection sensitivity.
- (d) An explanation is given as to why a thin tube is usually imperfection-sensitive while a thick tube is not in general.
- (e) In the post-bifurcation regime a thick tube could be considerably softer than a thin one (which may be useful in designing a tube as an energy absorber).
- (f) Within the framework of hyperelasticity fold-like profiles can develop in the post-bifurcation regime.

The methodology is systematic without ad hoc assumptions and is based on the three-dimensional field equations of finite deformations (taking into account both geometrical and material nonlinearity). The method of coupled series-asymptotic expansions is adopted to derive the asymptotic model equations in a mathematically consistent manner. Then, the method of multiple scales is used to construct the post-bifurcation solutions at near-critical loads. With the obtained amplitude equations, the singularities theory is used to study the imperfection sensitivity. We believe that the approach presented here is general and can be used to study other instability phenomena in structures composed of nonlinearly elastic materials (e.g. the buckling of a thin nonlinearly elastic film on a nonlinearly elastic compliant substrate). We shall leave these for future investigations.

Declaration of conflicting interests

None declared.

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Appendix: Incremental elastic moduli

For an isotropic hyperelastic material, the strain-energy function Φ only depends on the three principle stretches of the Lagrangian strain tensor **E**, that is, $\Phi = \Phi(e_1, e_2, e_3)$. Denote $\Phi_j = \partial \Phi / \partial e_j|_{e_1 = e_2 = e_3 = 0}$: then in the case where there are no prestresses, Φ_1 , Φ_2 and Φ_3 should vanish.

The non-zero first-order incremental elastic moduli can be written as

$$\zeta_1 = A_{1111}^{(1)} = \Phi_{11}, \quad \zeta_2 = A_{1122}^{(1)} = \Phi_{12}, \quad \zeta_3 = A_{1212}^{(1)} = \frac{1}{2}(\zeta_1 - \zeta_2).$$

There are only two independent constants among ζ_i .

The non-zero second-order incremental elastic moduli can be written as

$$\begin{split} \eta_1 &= A_{111111}^{(2)} = \Phi_{111}, \quad \eta_2 = A_{111122}^{(2)} = \Phi_{112}, \quad \eta_3 = A_{112233}^{(2)} = \Phi_{123}, \\ \eta_4 &= A_{111212}^{(2)} = \frac{1}{4} (\eta_1 - \eta_2), \quad \eta_5 = A_{331212}^{(2)} = \frac{1}{2} (\eta_2 - \eta_3), \\ \eta_6 &= A_{121323}^{(2)} = \frac{1}{8} (\eta_1 - 3\eta_2 + 2\eta_3). \end{split}$$

There are only three additional non-zero constants among η_i .

The non-zero third-order incremental elastic moduli can be written as

$$\begin{aligned} \kappa_{1} &= A_{1111111}^{(3)} = \Phi_{1111}, \quad \kappa_{2} = A_{1111122}^{(3)} = \Phi_{1112}, \quad \kappa_{3} = A_{11112222}^{(3)} = \Phi_{1122}, \\ \kappa_{4} &= A_{11112233}^{(3)} = \Phi_{1123}, \quad \kappa_{5} = A_{11111212}^{(3)} = \frac{1}{6} (\kappa_{1} - \kappa_{2}), \quad \kappa_{6} = A_{11112323}^{(3)} = \frac{1}{2} (\kappa_{3} - \kappa_{4}), \\ \kappa_{7} &= A_{11221212}^{(3)} = \frac{1}{12} (\kappa_{1} + 2\kappa_{2} - 3\kappa_{3}), \quad \kappa_{8} = A_{11221313}^{(3)} = \frac{1}{4} (\kappa_{2} - \kappa_{4}), \\ \kappa_{9} &= A_{1221212}^{(3)} = \frac{1}{8} (\kappa_{1} - 4\kappa_{2} + 3\kappa_{3}), \quad \kappa_{10} = A_{12121313}^{(3)} = \frac{1}{24} (\kappa_{1} - 4\kappa_{2} + 3\kappa_{3}), \\ \kappa_{11} &= A_{11121323}^{(3)} = \frac{1}{24} (\kappa_{1} - \kappa_{2} - 3\kappa_{3} + 3\kappa_{4}). \end{aligned}$$

There are only four additional non-zero constants among κ_i .