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Solar energy optimization in solar-HVAC using Sutterby hybrid nanofluid with Smoluchowski temperature conditions: a solar thermal application

Wasim Jamshed¹, Mohamed R. Eid^{2,3}, Rabia Safdar⁴, Faraj Ali Pasha⁵, Siti Suzilliana Putri Mohamed Isa⁶, Mohammad Ali^{7,8}, Zulfiqar Rehman⁹ & Wajaree Weera¹⁰✉

In solar heating, ventilation, and air conditioning (HVAC), communications are designed to create new 3D mathematical models that address the flow of rotating Sutterby hybrid nanofluids exposed to slippery and expandable surfaces. The heat transmission investigation included effects such as copper and graphene oxide nanoparticles, as well as thermal radiative fluxing. The activation energy effect was used to investigate mass transfer with fluid concentration. The boundary constraints utilized were Maxwell speed and Smoluchowski temperature slippage. With the utilization of fitting changes, partial differential equations (PDEs) for impetus, energy, and concentricity can be decreased to ordinary differential equations (ODEs). To address dimensionless ODEs, MATLAB's Keller box numerical technique was employed. Graphene oxide Copper/engine oil (GO-Cu/EO) is taken into consideration to address the performance analysis of the current study. Physical attributes, for example, surface drag coefficient, heat flux, and mass exchange are mathematically processed and shown as tables and figures. Numerous diverse factors are varied. The temperature field is enhanced by an increase in the volume fraction of copper and graphene oxide nanoparticles, while the mass fraction field is enhanced by an increase in activation energy.

List of symbols

T_∞	Ambient temperature (K)
Re	Reynold's number
Ω	Angular velocity
C_∞	Ambient concentration ($\frac{\text{mol}}{\text{m}^3}$)
$\dot{\gamma}$	Second invariant strain tensor
e	Consistency index

¹Department of Mathematics, Capital University of Science and Technology (CUST), Islamabad 44000, Pakistan. ²Department of Mathematics, Faculty of Science, New Valley University, Al-Kharga, Al-Wadi Al-Gadid 72511, Egypt. ³Department of Mathematics, Faculty of Science, Northern Border University, Arar 1321, Saudi Arabia. ⁴Department of Mathematics, Lahore College Women University, Lahore, Pakistan. ⁵Aerospace Engineering Department, King Abdulaziz University, Jeddah 21589, Saudi Arabia. ⁶Centre of Foundation Studies for Agricultural Science, Universiti Putra Malaysia, Seri Kembangan, Malaysia. ⁷Mechanical Engineering Program, Physical Science and Engineering Division, King Abdullah University of Science and Technology, Thuwal 23955-6900, Saudi Arabia. ⁸KAUST Clean Combustion Research Center, King Abdullah University of Science and Technology, Thuwal 23955-6900, Saudi Arabia. ⁹Department of Mathematics, Air University, Islamabad 44000, Pakistan. ¹⁰Department of Mathematics, Faculty of Science, Khon Kaen University, Khon Kaen 40002, Thailand. ✉email: wajawe@kku.ac.th

R_δ	Thermal radiation
U_w	Stretching velocity along x -axis ($\frac{m}{s}$)
Pr	Prandtl number
σ	Reaction rate constant ($\frac{mol}{lit \cdot s}$)
σ_v	Velocity accommodation coefficient
Γ_1	Temperature slip (K)
n	Fitted rate constant
T_w	Temperature at the wall (K)
μ_0	Zero shear fee viscosity
D_η	Deborah number
λ	Rotation parameter
E	Material time constant
T	Temperature
S	Extra stress tensor
N	Power-law behavior index
Γ_1	Velocity slip ($\frac{m}{s}$)
q_r	Radiative heat flux ($\frac{W}{m^2}$)
E	Activation energy ($\frac{J}{mol}$)
σ_T	Temperature accommodation coefficient
Sc	Schmidt number
A_1	Rivilian-Erikson tensor
C_w	Concentration at the wall ($\frac{mol}{m^3}$)

Researchers have concentrated on new energy measuring to meet the requirements and needs of companies in this period. Researchers are interested in developing a few devices with the highest rate of heating and cooling. These might save and maintain optimal energy efficiency. Furthermore, poor heat transmission and flowing base liquid conducting have an impact on the performance and operation of solar collectors. Many efforts have been made in this respect to improve the thermal characteristics of base liquids. Solar energy is the renewable energy source from the sun for industrial applications such as electricity generation¹⁻³, heating⁴⁻⁶, cooling⁷⁻⁹, and desalination¹⁰⁻¹². The benefits of solar energy technology are that this type of energy is limitless, clean, and has no fuel to burn. The most common types of solar energy are photovoltaic (PV) systems¹³⁻¹⁵, thin-film solar cells¹⁶⁻¹⁸, solar power plants^{19,20}, and passive solar heating^{21,22}. The Photovoltaic applications were reported in the field of telecommunications²³, agriculture²⁴, used with livestock/cattle²⁵, street lighting²⁶, and rural electrification²⁷. The usage of thin-film solar cells was in rooftops at the institutional and commercial buildings²⁸, solar farms²⁹, power traffics³⁰, and solar steam generation³¹. Passive solar heating is implemented in circulation spaces such as lobbies, hallways, and break rooms that allow occupants to avoid the sun.

HVAC stands for heating, ventilation, and air conditioning, whereas AC is defined as conditioning. AC is designed to cool the air and control humidity in the house and was invented by Willis Carrier in 1902³². Besides, the primary purpose of HVAC system for residential^{33,34} and commercial buildings^{35,36} is to provide a heating mode in the winter and cooling mode in the summer. This system also filters smoke, odors, dust, airborne bacteria, carbon dioxide, and other harmful gases to improve air indoors^{37,38}. In addition, HVAC system acts as a humidity controller of air indoors^{39,40}. Meanwhile, the HVAC system powered by solar energy is known as solar-HVAC (S-HVAC), where it is installed by PV panels to capture the sunlight and convert it into electricity. John Hollick is one of S-HVAC innovators, and he patented the method and apparatus for cooling ventilation air for a building⁴¹. The solar PV panel is connected to the HVAC to convert the solar energy into electricity to power all the parts responsible for the heating or cooling mode in the HVAC. The benefits of the S-HVAC systems instead of traditional HVAC, are lower utility bills, preserve the environment, and ease of installation. HVAC systems have moving parts such as fans and vibrating coils that often break, whereas S-HVAC have fewer moving parts and these systems have fewer breakage risks.

Among the several renewable resources that may be put practically anywhere in the globe, solar power promises to be the major technology for the transition to a decarbonized energy supply. The efficacy of a photovoltaic (PV) system is directly proportional to the amount of solar energy available. Many governments see renewables and energy conservation measures as a viable method to reduce coal consumption. The primary solar devices that can convert sunlight into electricity are PV system and concentrated solar power (CSP). CSP concentrates sun radiation to increase the temperature of a working fluid, and this fluid drives a heat engine and electric generator. CSP generates alternating current (AC), which has a high distribution rate on the power network. Besides, PV collects sunlight through the photoelectric effect to generate electricity in the form of a direct electric current (DC). The DC generated by the PV system is then transformed to AC through the inverters to ensure that the electricity is distributed on the power network. CSP stores energy by using Thermal Energy Storage technologies (TES), and it is not subjected to weather restrictions: This means that CSP can be used at all times (cloudy day, overnight, low sunlight, etc.) to generate electricity. On the other hand, PV system only stores low thermal energy compared to CSP, since it only uses a battery instead of the storage technology like TES. Therefore, CSP has more qualities over PV by performing more noteworthy efficiencies, lower speculation costs, gives warm capacity limit, and a superior mixture activity ability with different energizes to satisfy baseload need around evening time⁴².

Parabolic trough solar collector (PTSC) is one type of CSP system that has been used proficiently in water heating^{43,44}, air-conditioning^{45,46}, and solar-aircraft⁴⁷⁻⁵¹. PTSC consists of a reflector with a reflecting surface (parabolic-shaped mirror) and a receiver. The reflector collects the incident solar radiation and reflects it onto a receiver located in the focal line of the parabola. The working fluid inside the receiver absorbs the heat from the

solar radiation, causing the fluid temperature to increase. Finally, high-pressure superheated steam is generated from this working fluid in a conventional reheat steam turbine-generator to produce electricity. The running fluid in PTSC should have those features: (a) excessive thermal potential and thermal conductivity, (b) low thermal growth and occasional viscosity, (c) strong charge of thermal and chemical properties, (d) minimal charge of corrosive interest and (e) low toxicity⁵². One of the simplest operating fluids in PTSC is innovated nanofluid referred to as hybrid nanofluid and is ready via way of means of submerging specific nanoparticles withinside the equal base fluid. Therefore, there are recent studies regarding the hybrid nanofluid as a working fluid in PTSC installed in solar aircraft^{47–51}, and when PTSC is equipped with turbulators^{53–58}. The following types of hybridizing nanofluid were implemented in the PTSC solar aircraft: Casson hybrid nanofluid⁴⁷, Reiner Philippoff hybrid nanofluid^{48,49}, and tangent hyperbolic hybrid nanofluid^{50,51}. Meanwhile, A turbulator is a tool that transforms a laminar boundary layer right into a turbulent boundary layer to optimize heat transfer. Hence, various patterns of turbulators inserted in PTSC were reported, such as single twisted turbulator⁵³, obstacles act as turbulator⁵⁴, finned rod turbulator⁵⁵, two twisted tape acts as turbulator⁵⁶, inner helical axial fins as turbulator⁵⁷, and conical turbulator⁵⁸.

When it comes to thermodynamic rules, the second law of thermodynamics is far more dependable than the first law due to its limits of efficiency in heat transmission in industrial applications. The second law is applied to reduce the irreversibility of thermal constructions. Irreversibility is observed in a variety of thermofluidic apparatuses, including thermal solar, air separators, and reactors, and that competence loss is entirely inter-related with it. This generated irreversibility is determined by the rate of entropy production. The extinction of functional energy is measured by entropy generating. Any system's generated irreversibility creates continuous entropy, which eviscerates the functional energy required to execute the job. Such energy loss might be produced by heat transport by convective, conductive, and radiative fluxing. Furthermore, magnetic fields, buoyancy, and fluid friction all contribute to the generation of entropy. As a result, entropy generation minimization is required for diverse thermal equipment to acquire an optimal quantity of energy. The degree of entropy generating in crossbreed nanofluid is impacted by the expansion of twofold nanomaterials into the base liquid. The non-Newtonian cross breed nanofluid heavily influenced by entropy age have been examined, where this type of nanofluid contains the following double nanomaterial and base-fluid: Cu-Al₂O₃/H₂O^{59–65}, Cu-Al₂O₃/EG⁶⁶, Cu-Ag/EG^{67,68}, Cu-TiO₂/H₂O^{69,70}, Cu-Ag/H₂O⁷¹, Cu-Go/H₂O⁷², Cu-Ti/H₂O, CuO-TiO₂/H₂O and C71500-Ti₆Al₄V/H₂O⁷³, Cu-Fe₃O₄/EG⁷⁴, Cu-CuO/blood⁷⁵, Ag-MgO/H₂O⁷⁶, Ag-Gr/H₂O⁷⁷, CuO-TiO₂/EG⁷⁸, Fe₃O₄-Co/kerosene⁷⁹, MWCNT-Fe₃O₄/H₂O⁸⁰, and MWCNT-MgO/H₂O⁸¹. The thermal properties of hybrid nanofluid over an elastic curved surface⁵⁹, stretching sheet^{61,63,70,78}, disk⁶⁴, stretching disk⁶², and wedge⁷⁹ were reported. In addition, the flow of a hybrid nanofluid in a cavity was investigated under the following conditions: square cavity⁶⁸, porous open cavity⁶⁹, and various complex shape cavity⁸¹. The investigation of a hybrid nanofluid flow through a channel⁶⁶ and microchannel⁷⁵ have been performed, where these channels are rotating⁶⁶, placed vertically⁷³, and recharging⁷⁷. The flow of a hybrid nanofluid in an enclosure was studied by Alsabery et al.⁶⁰, Ghalambaz et al.⁶⁵, and Abu-Libdeh et al.⁷⁶. Alsabery et al.⁶⁰ implemented the wavy enclosure containing the inner solid blocks, whereas Ghalambaz et al.⁶⁵ considered an enclosed cavity with vertical and horizontal parts in their fluid model. On the other hand, Abu-Libdeh et al.⁷⁶ selected a porous enclosure with a trapezoid geometry where this type of geometry is used for cooling purposes on the hybrid nanofluid. Meanwhile, Xia et al.⁶⁷ and Khan et al.⁷² developed the fluid flow model bounded by two rotating parallel frames. The heat analysis of the peristaltic flow of hybrid nanofluid internal a duct become studied through McCash et al.⁷¹. The electroosmotic pump is involved in the hybrid nanofluid flow studied by Munawar and Saleem⁷⁵, with ohmic heating. Shah et al.⁸⁰ chose a porous annulus to study the characteristics of a hybrid nanofluid model.

Non-Newtonian fluid models are much more different than those of Newtonianism fluids. The stress values for non-Newtonian fluid are nonlinear functions against strain, yield stress, or time-dependent viscosity. Examples of this type of fluid are Casson fluid^{82–86}, Maxwell fluid^{87–91}, nanofluid (also including hybrid case)^{47–81}, etc. Sutterby fluid model is one type of non-Newtonianism fluid⁹², and it describes the viscosity of dilute polymer solutions⁹³. Polymer solutions have been applied in related industrial phenomena or products, such as turbulent pipe flows^{94,95}, stability of polymer jets^{96,97}, and oil recovery enhancement^{98,99}. The heat and mass transfer withinside the flow of magnetohydrodynamics (MHD) Sutterby nanofluid over a stretching cylinder, with the impact of temperature-structured thermal conductivity have been explored by Sohail et al.¹⁰⁰ and Raza et al.¹⁰¹. The bioconvection of Sutterby fluid flow was reported when this fluid flows across the wedge¹⁰² and between two rotating disks¹⁰³. Gowda et al.¹⁰⁴, Yahya et al.¹⁰⁵, and Khan et al.¹⁰⁶ incorporated the Cattaneo-Christov heat flux model in their mathematical Sutterby fluid model to archive effective thermal properties. The Cattaneo-Christov heat flux model was developed when the fluid was bounded by a rotating disk¹⁰⁴, flat surface¹⁰⁵, and wedge¹⁰⁶. The effect of entropy generation and activation energy were considered by Hayat et al.¹⁰⁷. In contrast, El-Dabe et al.¹⁰⁸ incorporated the boundaries of the attractive field, compound response, permeable media, heat radiation, gooev dissemination, and couple pressure. Parveen et al.¹⁰⁹, Arif et al.¹¹⁰, Jayadevamurthy et al.¹¹¹, Nawaz¹¹², and Waqas et al.¹¹³ investigated the thermal performance of the Sutterby fluid model with the presence of various hybrid nanoparticles. The base fluid that has become selected was blood^{109,110}, water¹¹¹, and ethylene glycol^{112,113}. These researchers^{109–113} implemented the dual nanoparticles in their Sutterby hybrid nanofluid, namely as: (i) Au and Al₂O₃¹⁰⁹, (ii) CuO and Al₂O₃¹¹⁰, (iii) Cu and SiO₂¹¹¹, (iv) MoS₂ and SiO₂¹¹², and (v) first fluid contained SiO₂ and SWCNT, and second fluid used MoS₂ and MWCNT¹¹³.

Motivation

The goal of this study is to look at a Sutterby hybrid fluid traveling along a stretchy surface with copper and graphene oxide nanoparticles. The following are the main points of the current study:

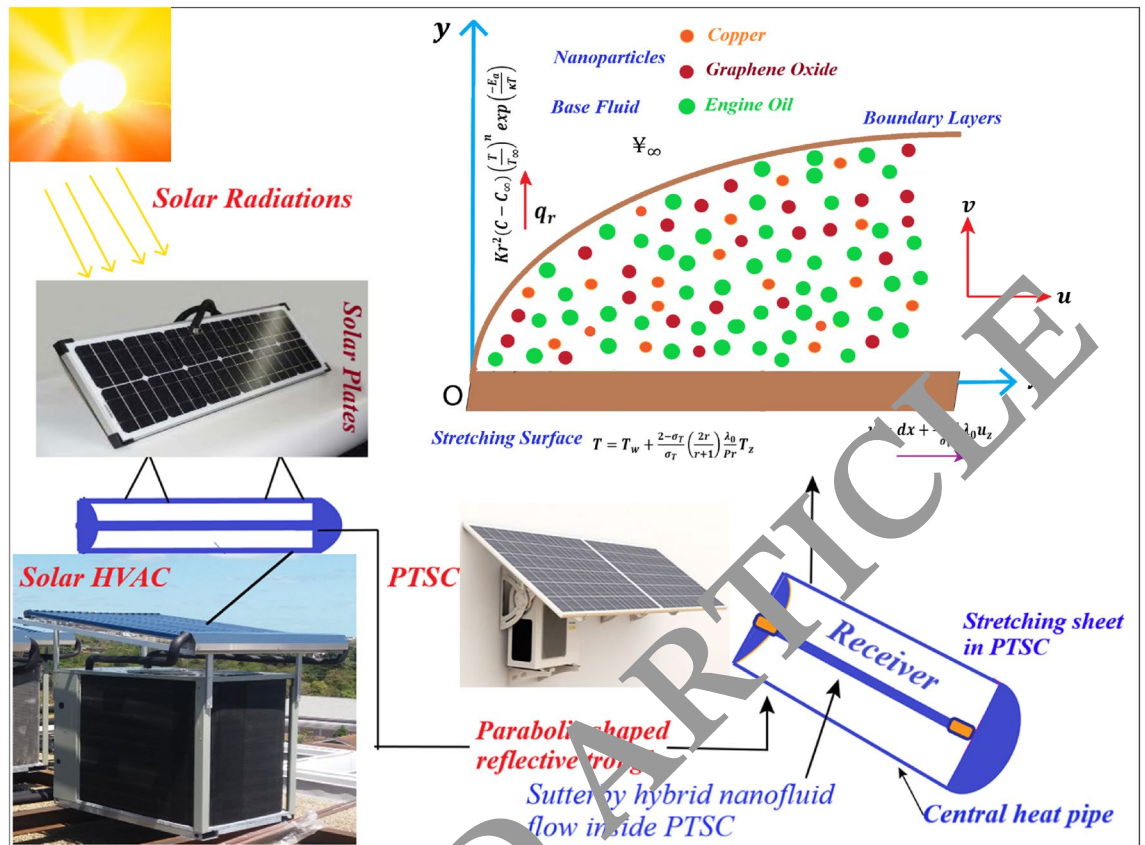


Figure 1. The graphical model of the current problem.

- The effect of ultrafine strong nanoparticles (copper and graphene oxide) at the Sutterby hybrid fluid has yet to be contemplated.
- In the extant literature, the 3D kind of Sutterby nanofluid has been built and explored.
- The results of Maxwell speed slipperly and Smoluchowski heat slipperly boulder situations on hybrid nanofluid impacting on an extensible floor are but to be investigated.

The paper's structure

The following is a summary of the paper's structure.

- The governing model was created on the premise of a boundary layer.
- Controlling PDEs are converted into ODEs using appropriate similarity transformation.
- The ODEs are adapted to 1st-ordered and resolved a usage of the Keller container numerical method included in MATLAB.
- Physical portions along with the pores and drag force factor and Nusselt number are mathematically decided and demonstrated in tables.
- Mathematical model's velocity, temperature, and awareness elements are numerically calculated and represented withinside the shape of figures.

Proposed mathematical model. The graphical model is presented in Fig. 1, and the characteristics of the proposed mathematical model are as below:

- 3D model (as in Fig. 2), where x - and y - axes contain planes, where z -axis fluid flow region is at the third axis $z \geq 0$.
- The fluid rotates along z -axis, showing that this axis acts as the axis of rotation for the rotating fluid. This fluid has an angular velocity Ω .
- The involved fluid in this model is incompressible Sutterby fluid, flowing on an extendable surface. This surface is located at xy -plane.
- The Maxwell velocity slip¹¹⁴ effect is investigated, by adding the component of stretching $u_w = dx$, together with the slip length $\frac{2-\sigma_v}{\sigma_v} \lambda_0 U_z$.
- The Smoluchowski temperature slip¹¹⁵ is added, by implementing the term $\frac{2-\sigma_T}{\sigma_T} \left(\frac{2r}{r+1} \right) \frac{\lambda_0}{P_r} T_z$.

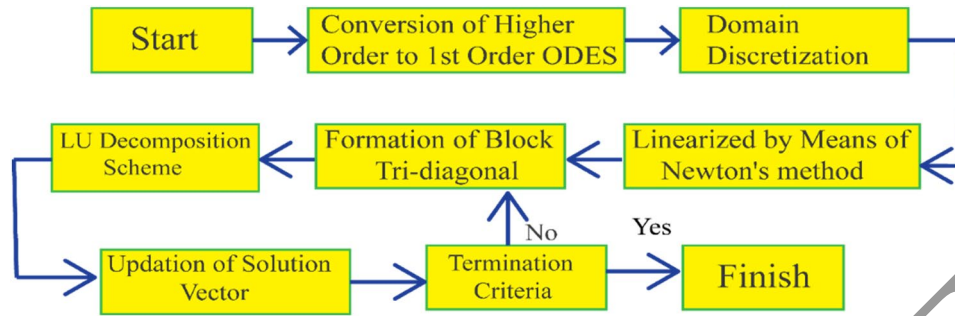


Figure 2. Schematic chat of KBM procedure.

- Surface temperature and concentration are denoted by T_w and C_w , respectively. Mean while, T_∞ and C_∞ represent the ambient temperature as well as concentration.

The physical properties of Sutterby hybrid nanofluid are presented in Eq. (1). The dynamics viscosity, density, precise heat and thermal conductivity of hybrid nanofluid are indicated by μ_{hnf} , ρ_{hnf} , α_{hnf} , $(\rho C_p)_{hnf}$ and k_{hnf} , respectively.

$$\left. \begin{aligned} \mu_{hnf} &= \mu_f(1 - \phi_{Cu})^{-2.5}(1 - \phi_{GO})^{-2.5}, \\ \rho_{hnf} &= [(1 - \phi_{GO})\{(1 - \phi_{Cu})\rho_f + \phi_{Cu}\rho_{p1}\} + \phi_{GO}\rho_{p2}], \\ (\rho C_p)_{hnf} &= [(1 - \phi_{GO})\{(1 - \phi_{Cu})(\rho C_p)_f + \phi_{Cu}(\rho C_p)_{p1}\} + \phi_{GO}(\rho C_p)_{p2}], \\ \frac{\kappa_{hnf}}{\kappa_{gf}} &= \left[\frac{(\kappa_{p2} + 2\kappa_{gf}) - 2\phi_{GO}(\kappa_{gf} - \kappa_{p2})}{(\kappa_{p2} + 2\kappa_{gf}) + \phi_{GO}(\kappa_{gf} - \kappa_{p2})} \right] \left[\frac{(\kappa_{p1} + 2\kappa_f) - 2\phi_{Cu}(\kappa_f - \kappa_{p1})}{(\kappa_{p1} + 2\kappa_f) + \phi_{Cu}(\kappa_f - \kappa_{p1})} \right]. \end{aligned} \right\} \quad (1)$$

A Cauchy tensor of tension for Sutterby liquid is presented as¹¹⁶

$$T = -pI + S, \quad (2)$$

in which p , I and S constitute pressure, identification tensor, and further strain tensor, respectively. Subsequently, S in Eq. (2) is given as

$$S = \mu_0 \left[\frac{\sinh^{-1}(E\dot{\gamma})}{E\dot{\gamma}} \right]^m A_1, \quad (3)$$

where in μ_0 is 0 shear free viscosity, and E is a material time constant. In Eq. (3), the second one invariant stress tensor $\dot{\gamma}$ and primary order Rivilian-Erikson tensor A_1 were interpreted in Eqs. (4) and (5), respectively.

$$\dot{\gamma} = \sqrt{\frac{tr(A_1)^2}{2}}, \quad (4)$$

$$A_1 = (gradV) + (gradV)^T. \quad (5)$$

The m values determine the fluid categories, where Newtonian fluid when $m = 0$, pseudo-plastic (shear thinning) when $m > 0$, and dilatant (shear thickening) when $m < 0$. In addition, the velocity field of the fluid is taken as $V = [u(x, y, z), v(x, y, z), w(x, y, z)]$.

Under the restriction as stated above, the modeled equations are premeditated by¹¹⁷:

$$uu_x + vv_y + ww_z = 0, \quad (6)$$

$$uu_x + vv_y + ww_z - 2\Omega v = \frac{\mu_{hnf}}{\rho_{hnf}} \frac{v}{2} u_{zz} \left(1 - \frac{Ne^2}{2} (u_z)^2 \right), \quad (7)$$

$$uv_x + vv_y + ww_z + 2\Omega u = \frac{\mu_{hnf}}{\rho_{hnf}} \frac{v}{2} v_{zz} \left(1 - \frac{Ne^2}{2} (v_z)^2 \right), \quad (8)$$

$$uT_x + vT_y + wT_z = \alpha_{hnf} T_{zz} - \frac{1}{(\rho C_p)_{hnf}} (q_r)_z, \quad (9)$$

$$uC_x + vC_y + wC_z = DC_{zz} - Kr^2(C - C_\infty) \left(\frac{T}{T_\infty}\right)^n \exp\left(\frac{-E_a}{\kappa T}\right). \tag{10}$$

Equations (6)–(10) are controlled by the following boundary conditions:

$$\left. \begin{aligned} y = 0 : u = dx + \frac{2-\sigma_v}{\sigma_v} \lambda_0 u_z, v = 0, w = 0, C = C_w, \\ T = T_w + \frac{2-\sigma_T}{\sigma_T} \left(\frac{2r}{r+1}\right) \frac{\lambda_0}{Pr} T_z. \\ y \rightarrow \infty : u \rightarrow 0, v \rightarrow 0, T \rightarrow T_\infty, C \rightarrow C_\infty. \end{aligned} \right\} \tag{11}$$

In Eq. (9), Rosseland approximation¹¹⁸ is added:

$$q_r = -\frac{4\sigma^*}{3\kappa^*} T_z^4 = -\frac{4\sigma^*}{3\kappa^*} T^3 \frac{\partial^2 T}{\partial z^2}. \tag{12}$$

where in σ^* and κ^* stand for Stefan-Boltzmann constant and imply absorption coefficient, respectively.

The appropriate transformations¹¹⁹ have been selected, as shown in (13):

$$\left. \begin{aligned} u = dx f'(\beta), \quad v = dx g(\beta), \quad w = -\sqrt{d} v f(\beta), \quad \theta(\beta) = \frac{T - T_\infty}{T_w - T_\infty}, \\ \beta = \sqrt{\frac{d}{v}} z, \quad \phi(\beta) = \frac{C - C_\infty}{C_w - C_\infty}. \end{aligned} \right\} \tag{13}$$

The transformations (13) are implemented to dimensionless the early mathematical model (6)–(10), together with (12). As a result, the following forms have occurred:

$$f''' \left(1 - \frac{N}{2} R_\eta D_\eta f''^2\right) - 2B_1 B_2 f'^2 - 2B_3 B_2 f f'' + 4B_1 B_2 \lambda g = 0, \tag{14}$$

$$g'' \left(1 - \frac{N}{2} R_\eta D_\eta f'^2\right) - 2B_1 B_2 g' + 2B_1 B_2 f g' - 4B_1 B_2 \lambda f' = 0, \tag{15}$$

$$\left(B_3 + \frac{4}{3} R_\delta\right) \theta'' + B_4 P_r f \theta' = 0, \tag{16}$$

$$S_\delta f'' + S_\delta f \phi' - \sigma S_\delta (1 + \Gamma \theta)^n \exp\left(\frac{-E}{1 + \Gamma \theta}\right) \phi = 0, \tag{17}$$

After implementing (13) in (11), the dimensionless BCs are:

$$\left. \begin{aligned} \beta \rightarrow 0 : f'(0) = \frac{d}{v} \Gamma_1 f''(0), \quad g(0) = 0, \quad f(0) = 0, \quad \theta(0) = 1 + \Gamma_2 \theta', \quad \phi' = 1. \\ \beta \rightarrow \infty : f' \rightarrow 0, \quad g \rightarrow 0, \quad \theta \rightarrow 0, \quad \phi \rightarrow 0. \end{aligned} \right\} \tag{18}$$

The final dimensionless governing parameters in (14)–(17) have been derived as

$$\left. \begin{aligned} E = \left(\frac{E_a}{\kappa T_\infty}\right), \quad \Gamma = \frac{T_w - T_\infty}{T_\infty}, \quad R_\eta = \frac{dx^2}{v}, \quad D_\eta = e^2 d^2, \quad S_\delta = \frac{v}{D}, \\ \sigma = \frac{k_z}{a}, \quad \lambda = \frac{\Omega}{d}, \quad \Gamma_1 = \frac{2-\sigma_v}{\sigma_v} \lambda_0 \sqrt{\frac{d}{v}}, \quad \Gamma_2 = \frac{2-\sigma_T}{\sigma_T} \frac{\lambda_0}{Pr} \frac{2r}{r+1} \sqrt{\frac{d}{v}}, \\ R_\delta = \frac{4\sigma T_\infty^3}{\kappa^* \kappa_\infty} \end{aligned} \right\} \tag{19}$$

where B_1, B_2, B_3 and B_4 are constants¹²⁰ as below:

$$\left. \begin{aligned} B_1 &= \frac{1}{(1-\phi_{Cu})^{2.5} (1-\phi_{GO})^{2.5}}, \\ B_2 &= \frac{1}{(1-\phi_{GO}) \{ (1-\phi_{Cu}) + \phi_1 \frac{\rho p_1}{\rho_f} \} + \phi_{GO} \frac{\rho p_2}{\rho_f}}, \\ B_3 &= \left[\frac{(\kappa_{p_2} + 2\kappa_{gf}) - 2\phi_{GO}(\kappa_{gf} - \kappa_{p_2})}{(\kappa_{p_2} + 2\kappa_{gf}) + \phi_{GO}(\kappa_{gf} - \kappa_{p_2})} \right] \left[\frac{(\kappa_{p_1} + 2\kappa_f) + \phi_{Cu}(\kappa_f - \kappa_{p_1})}{(\kappa_{p_1} + 2\kappa_f) - 2\phi_{Cu}(\kappa_f - \kappa_{p_1})} \right], \\ B_4 &= (1 - \phi_{GO}) \{ (1 - \phi_{Cu}) + \phi_{Cu} \frac{(\rho C_p)_{p_1}}{(\rho C_p)_f} \} + \phi_{GO} \frac{(\rho C_p)_{p_2}}{(\rho C_p)_f}. \end{aligned} \right\} \tag{20}$$

Thermophysical properties of copper and graphene oxide nanoparticles^{120,121} have been tabulated in Table 1. The skin friction coefficients in horizontal x - and vertical axes y - are shown in Eq. (21). From Eq. (21) also, τ_{xz} and τ_{yz} ¹²² are expressed in Eq. (22).

$$Cf_x = \frac{\tau_{xz}}{\rho_f U_w^2}, \quad Cf_y = \frac{\tau_{yz}}{\rho_f U_w^2}, \tag{21}$$

$$\tau_{xz} = -\mu_{mf} \left[u_z + \frac{Ne^2}{3} (u_z)^3 \right], \quad \tau_{yz} = -\mu_{mf} \left[v_z + \frac{Ne^2}{3} (v_z)^3 \right], \tag{22}$$

Thermophysical	ρ (kg/m ³)	c_p (J/kgK)	k (W/mK)
Copper (Cu)	8933	385.0	401.00
Engine oil (EO)	884	1910	0.144
Graphene oxide (GO)	1800	717	5000

Table 1. Thermophysical properties.

Finally, surface drag coefficients are derived as:

$$Cf_x Re_x^{1/2} = \frac{f'' + \frac{N}{3} R_\eta D_\eta f''^3}{B_1}, \quad Cf_y Re_x^{1/2} = \frac{g' + \frac{N}{3} R_\eta D_\eta g'^3}{B_1}. \tag{23}$$

The dimensional heat transfer coefficient¹²² is expressed in Eq. (24), where the heat flux q_w is shown in Eq. (25).

$$Nu_x = \frac{xq_w}{(T_f - T_\infty)|_{z=0}} + \frac{xq_r}{k(T_f - T_\infty)|_{z=0}}, \tag{24}$$

$$q_w = -k T_z|_{z=0}. \tag{25}$$

From Eqs. (24), (25), the dimensionless Nusselt number is obtained:

$$Nu_x Re_x^{-1/2} = -\left(B_3 + \frac{4}{3} R_\delta\right) \theta'(0). \tag{26}$$

The Sherwood number and the mass flux are given in Eqs. (27) and (28), respectively.

$$Sh_x = \frac{xq_w}{D(C_w - C_\infty)} C_z|_{z=0}, \tag{27}$$

$$q_w = -D C_z|_{z=0}, \tag{28}$$

After manipulation of Eq. (28) into Eq. (27), The dimensionless shape of the mass transfer coefficient is

$$Sh_x Re_x^{-1/2} = -\phi'(0). \tag{29}$$

Numerical scheme

Keller box method (KBM)¹²³ is selected as the current numerical technique to perform the solutions for the ODEs (14)–(17), together with BCs (18). The coding of KBM is built in MATLAB software, wherein the flow chart of KBM technique is depicted in Fig. 2. The present-day numerical method applies a finite distinction scheme, which is a combination technique of order 4 and it runs in the back of KBM MATLAB. The above-mentioned nonlinear differential problem, i.e., Eqs. (14)–(17) followed by the end point condition supplied by Eq. (18) is solved using the Keller box approach.

Step 1 Conversion of ODEs

The aforementioned equations are fairly turned into a new sophisticated first order coupled system:

$$\left. \begin{aligned} y'_1 &= y_2, & y_1(0) &= 0, \\ y'_2 &= y_3, & y_2(0) &= 1 + \Gamma_1 s, \\ y'_3 &= \frac{2B_1 B_2 [y_2 - 2\lambda y_4 - y_1 y_3]}{1 - \frac{N}{2} R_\eta D_\eta y_3^2}, & y_3(0) &= s, \\ y'_4 &= y_5, & y_4(0) &= 0, \\ y'_5 &= \frac{2B_1 B_2 [2\lambda y_2 - y_1 y_5 + y_4 y_2]}{1 - \frac{N}{2} R_\eta D_\eta y_5^2}, & y_5(0) &= t, \\ y'_6 &= y_7, & y_6(0) &= 1 + \Gamma_2 u, \\ y'_7 &= \frac{P_f b_4 y_1 y_7}{\left(B_3 + \frac{4}{3} R_\delta\right)}, & y_7(0) &= u, \\ y'_8 &= y_9, & y_8(0) &= 1, \\ y'_9 &= \left(\sigma S_\delta (1 + \Gamma y_6)^n \exp\left(\frac{-E}{1 + \Gamma y_6}\right)\right) y_9 - S_\delta y_1 y_9, & y_9(0) &= v. \end{aligned} \right\} \tag{30}$$

Step 2 Domain discretization & difference equations

Likewise, domain discretization in $x - \beta$ plane is signified. In view of this web, net points are $\beta_0 = 0, \beta_j = \beta_{j-1} + h_j, j = 0, 1, 2, 3, \dots, J, \beta_J = 1$ where, h_j is the step-size. Relating central difference formulation at midpoint $\beta_{j-1/2}$

$$\left(\frac{(y_1)_j - (y_1)_{j-1}}{h_j}\right) = \left(\frac{(y_2)_j + (y_2)_{j-1}}{2}\right), \tag{31}$$

$$\left(\frac{(y_2)_j - (y_2)_{j-1}}{h_j}\right) = \left(\frac{(y_3)_j + (y_3)_{j-1}}{2}\right), \tag{32}$$

$$\left(\frac{(y_4)_j - (y_4)_{j-1}}{h_j}\right) = \left(\frac{(y_5)_j + (y_5)_{j-1}}{2}\right), \tag{33}$$

$$\left(\frac{(y_6)_j - (y_6)_{j-1}}{h_j}\right) = \left(\frac{(y_7)_j + (y_7)_{j-1}}{2}\right), \tag{34}$$

$$\left(\frac{(y_8)_j - (y_8)_{j-1}}{h_j}\right) = \left(\frac{(y_9)_j + (y_9)_{j-1}}{2}\right) \tag{35}$$

$$\left\{ \begin{aligned} &\left(1 - \frac{N}{2} R_\eta D_\eta \left(\frac{(y_3)_j + (y_3)_{j-1}}{2}\right)^2\right) \left(\frac{(y_3)_j - (y_3)_{j-1}}{h_j}\right) - 2B_1 B_2 \left(\frac{(y_2)_j + (y_2)_{j-1}}{2}\right) \\ &- 2B_1 B_2 \left[-2\lambda \left(\frac{(y_4)_j + (y_4)_{j-1}}{2}\right) - \left(\frac{(y_1)_j + (y_1)_{j-1}}{2}\right) \left(\frac{(y_3)_j + (y_3)_{j-1}}{2}\right)\right] = 0 \end{aligned} \right\} \tag{36}$$

$$\left\{ \begin{aligned} &\left(1 - \frac{N}{2} R_\eta D_\eta \left(\frac{(y_5)_j + (y_5)_{j-1}}{2}\right)^2\right) \left(\frac{(y_5)_j - (y_5)_{j-1}}{h_j}\right) - 4B_1 B_2 \lambda \left(\frac{(y_2)_j + (y_2)_{j-1}}{2}\right) \\ &- 2B_1 B_2 \left[-\left(\frac{(y_1)_j + (y_1)_{j-1}}{2}\right) \left(\frac{(y_5)_j + (y_5)_{j-1}}{2}\right) + \left(\frac{(y_4)_j + (y_4)_{j-1}}{2}\right) \left(\frac{(y_2)_j + (y_2)_{j-1}}{2}\right)\right] = 0 \end{aligned} \right\} \tag{37}$$

$$\left(B_3 + \frac{4}{3} R_\delta\right) \left(\frac{(y_7)_j - (y_7)_{j-1}}{h_j}\right) - P_r f b_4 \left(\frac{(y_1)_j + (y_1)_{j-1}}{2}\right) \left(\frac{(y_7)_j + (y_7)_{j-1}}{2}\right) = 0 \tag{38}$$

$$\left\{ \begin{aligned} &\left(\frac{(y_9)_j - (y_9)_{j-1}}{h_j}\right) + S_\delta \left(\frac{(y_1)_j + (y_1)_{j-1}}{2}\right) \left(\frac{(y_9)_j + (y_9)_{j-1}}{2}\right) \\ &- \sigma S_\delta \left(\frac{(y_1)_j + (y_1)_{j-1}}{2}\right) + \Gamma \left(\frac{(y_6)_j + (y_6)_{j-1}}{2}\right) \left(1 - E \left(1 - \Gamma \left(\frac{(y_6)_j + (y_6)_{j-1}}{2}\right)\right)\right) \left(\frac{(y_9)_j + (y_9)_{j-1}}{2}\right) = 0 \end{aligned} \right\} \tag{39}$$

Step 3 Newton method

Equations (29) through (37) are linearized using Newton's linearization technique

$$\left. \begin{aligned} (y_1)_j^{n+1} &= (y_1)_j^n + (\delta y_1)_j^n, (y_2)_j^{n+1} = (y_2)_j^n + (\delta y_2)_j^n, \\ (y_3)_j^{n+1} &= (y_3)_j^n + (\delta y_3)_j^n, (y_4)_j^{n+1} = (y_4)_j^n + (\delta y_4)_j^n, \\ (y_5)_j^{n+1} &= (y_5)_j^n + (\delta y_5)_j^n, (y_6)_j^{n+1} = (y_6)_j^n + (\delta y_6)_j^n, \\ (y_7)_j^{n+1} &= (y_7)_j^n + (\delta y_7)_j^n, (y_8)_j^{n+1} = (y_8)_j^n + (\delta y_8)_j^n, \\ (y_9)_j^{n+1} &= (y_9)_j^n + (\delta y_9)_j^n. \end{aligned} \right\} \tag{40}$$

Step 4 Block tridiagonal structure

The linear mathematical model now has the block tridiagonal shape, written

$$A \Delta = S, \tag{41}$$

where

λ	$-f''(0)$		Present results
	Ref. ¹²⁴	Ref. ¹¹⁷	
0.2	-1.0331	-1.0330	-1.0330
0.4	-1.1009	-1.1009	-1.1009
0.5	-1.1384	-1.1384	-1.1384
0.6	-1.1764	-1.1763	-1.1763

Table 2. Assessment of $-f''(0)$ with^{117,124}.

N	R_η	D_η	λ	Γ_1	$Cf_x Re_x^{\frac{1}{2}}$	$Cf_y Re_x^{\frac{1}{2}}$
0.5	0.5	0.5	0.1	0.5	-1.57353	-0.32626
1.0					-1.72214	-0.32554
2.0					-1.80911	-0.33423
	1.0				-1.72314	-0.33522
	1.5				-1.79635	-0.33434
	2.0				-1.80135	-0.33312
		1.0			-1.73416	-0.33432
		3.0			-2.03212	-0.33122
		5.0			-2.42080	-0.33075
			0.3		-1.76581	-0.81728
			0.5		-1.86153	-1.17724
			0.7		-1.95245	-1.39521
				0.7	-1.35680	-0.31952
				0.9	-1.12517	-0.28862
				1.1	-0.97046	-0.26568

Table 3. Behaviors of diverse factors on the wall frictional factors.

$$A = \begin{bmatrix} [L_1] & [N_1] \\ & [L_2] & [N_2] \\ & & \ddots & \ddots \\ & & & [M_{j-1}] & [L_{j-1}] & [N_{j-1}] \\ & & & & [M_j] & [L_j] \end{bmatrix}, \Delta = \begin{bmatrix} [\Delta_1] \\ \vdots \\ [\Delta_{j-1}] \\ [\Delta_j] \end{bmatrix} \text{ and } S = \begin{bmatrix} [S_1] \\ \vdots \\ [S_{j-1}] \\ [S_j] \end{bmatrix}. \quad (42)$$

where the overall size of the block-triangle matrix A is $J \times J$ and the supvector's block size is 9×9 . LU decomposition method implementation for solving Δ . A mesh size of $h_j = 0.01$ is regarded adequate for mathematical assessment, and the difference between the current and previous iterations for the needed precision has been set at 10^{-6} .

Result's verification

The comparative analysis of the numerical values skin friction coefficient values $-f''(0)$, are tabulated in Table 2. The comparison is made with the previous researchers^{117,124}, with the various values of rotating parameter λ . However, other parameters have remained zero such as consistency parameter, Reynolds, Deborah numbers, and speed slippage ($N = R_\eta = D_\eta = \Gamma_1 = 0$). Besides, $B_1 = B_2$ is fixed to obtain this comparative analysis. From Table 2, it is clear that the accuracy of the current results is quite high. Therefore, the current numerical scheme KBS is quite reliable, authentic, and acceptable for subsequent calculations.

Result and discussion

This segment shows and discusses the impact of diverse parameters at the floor frictional factor, Nusselt value, speed, energy, and concentricity outlines with the use of tables and figures. In the case of separated boundaries, Table 3 is intended to mirror the effect of wall frictional factors Cf_x and Cf_y consistent with the table, changes inside the power-regulation conduct list N , Reynolds number R_η , Deborah D_η , pivot boundary and speed slippage cause a decline inside the surface coefficient of drag along the x - orientation, however an expansion when the speed slippage boundary δ_1 is gotten to the next level. This is physically since both the Reynolds number

R_δ	P_r	Γ_2	σ	S_δ	δ	E	n	$Nu_x Re_x^{-1}$	$Sh_x Re_x^{-1}$
0.5	6450	0.5	0.7	0.7	0.5	0.5	0.5	1.98441	0.75801
1.0								2.04137	0.77854
2.0								2.15052	0.78670
	6.3							1.95437	0.70766
	6.5							1.99442	0.72530
	6.7							2.04212	0.74416
		0.7						1.77233	0.79807
		0.9						1.69735	0.77125
		1.1						1.62549	0.73238
			0.9					1.88597	0.85053
			1.1					1.88597	0.96349
			1.3					1.88597	0.99955
				0.9				1.88597	0.89820
				1.1				1.88597	0.95739
				1.3				1.88597	1.06865
					0.7			1.88597	0.82853
					0.9			1.88597	0.88564
					1.1			1.88597	0.92008
						0.7		1.88597	0.65371
						0.9		1.88597	0.57571
						1.1		1.88597	0.43371
							-0.5	1.88597	0.73871
							0	1.88597	0.76621
							0.9	1.88597	0.80589

Table 4. Diverse factors influence on Nusselt and Sherwood numbers.

$R_\eta = \frac{dx^2}{\nu}$ and the Deborah number $D_\eta = \frac{a^2 d^2}{\nu}$ depend on the viscosity of the nanofluid and follows the frictional force is diminished. C_f ascends because of expansions in N , and Γ_1 but falls in light of an increment in its values. This is because increasing the rapidity slippage $\Gamma_1 = \frac{2-\sigma_v}{\sigma_v} \lambda_0 \sqrt{\frac{d}{\nu}}$ increases the reaction rate, and this effect occurs. Table 4 is expected to examine fitness and mass exchange rates for dimensionless various variables. It is found that when the radiation boundary R_δ and Prandtl number P_r are changed, Nusselt number improves, however, devalues as the temperature slips Γ_2 . This is because the presence of heat radiation boosts the stored thermal energy and then begins to release it through the nanofluid molecules, which improves the rate of mutual rate of heat transfer, which in turn grows the number of Nusselt. The mass exchange rate increments when R_δ , substance response rate, Schmidt number Sc , temperature contrast boundary, and fixed value steady n increment, yet reduces as P_r , heat slippery Γ_2 , and enactment energy E decline.

The impact of R_η on $f'(\eta)$ is portrayed in Fig. 3. R_η decides if the conduct is laminar or tempestuous at the act level. The Reynolds number is the ratio of inertial power to goeey power. It is worth noting that the higher the Reynolds number, the greater the inertial power over the goeey power, the thicker the consistency, and the smaller the motion field. Indeed, increasing the volume fraction of nanoparticles reduces liquid fixation, diminishes liquid thickness, and boosts idleness. Finally, a significant component in the lowering of the rapidity field. Figure 4 shows the impact of D_η on $f'(\beta)$. Physically, smaller Deborah values make the material to operate more freely, resulting in a flow of Newtonian viscosity. With increasing Deborah quantities, the effectual enters the non-Newtonianism zone, with increased elasticity ratings and solid-like behavior. The bigger the Deborah quantity, the stronger the viscidness effect. Deborah values distinguish amongst liquid solids and fluid properties on a physical level. As D_η increases, the fluid changes from a fluid to a solid. The substance behaves like a liquid for lesser D_η and such as a solid for greater D_η . As D_η increases, fluid behaviour such as shear thickening becomes more difficult to flow through the surface, lowering $f'(\beta)$. The behavior of the power law exponent M at $f'(\beta)$ (Fig. 5). When shear force is applied, N affects the viscosity of the nanofluid. The letters N stand for fluid shear thinning and Newtonianism. Positive variations in N boost viscidness (shearing thicker) and decrease the velocity of fluid flowing through a ductile surface, thus use caution. Physically, shearing thicker occurs as a result of a larger volume fraction of nanomolecules, a rise in fluid viscosity, and a reduction in fluid rapidity $f'(\beta)$. The relationship between rotational parameter and $f'(\beta)$ is shown in Fig. 6. The fractional size of gold nanomolecules is magnified, which reduces $f'(\beta)$ and the thickness of the momentum boundary layer. An alteration in $f'(\beta)$, acts like shear thickening. When the torque increases, this cause to incremental changes in the viscosity of the fluid to develop, the nanofluid rapidity decreases. The effect of R_η on $g(\beta)$ is depicted in Fig. 7. In opposite of the viscidness influence, R_η emphasizes the relevance of the inertia effect. The consistency of the liquid is decreased, and the liquid speed $g(\beta)$ is diminished when R_η is expanded. The motivation behind Fig. 8 is to stress the feature of D_η on $g(\beta)$. Higher thick powers that lull the liquid speed led to an expansion in D_η . The liquid behaves exactly

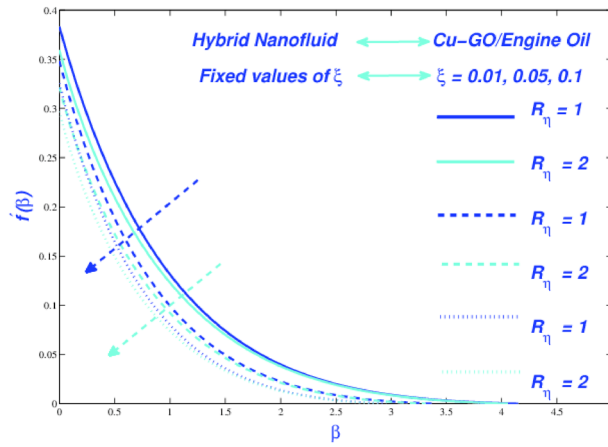


Figure 3. Influence of R_η on f' .

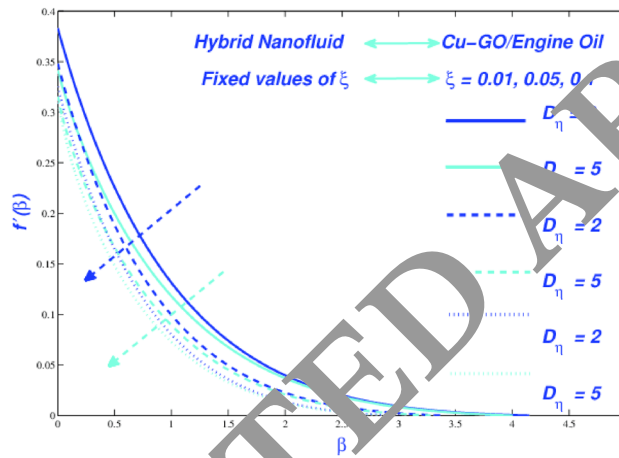


Figure 4. Impact of D_η on f' .

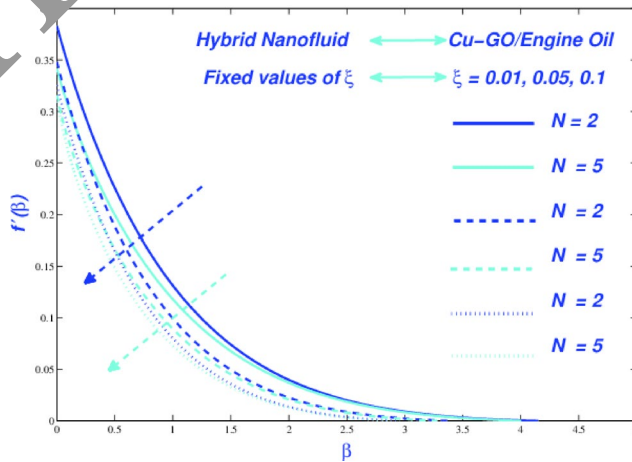


Figure 5. Impact of N on f' .

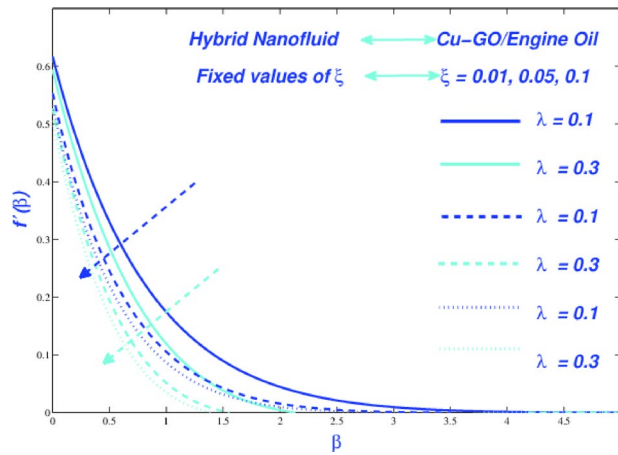


Figure 6. Effect of λ on f' .

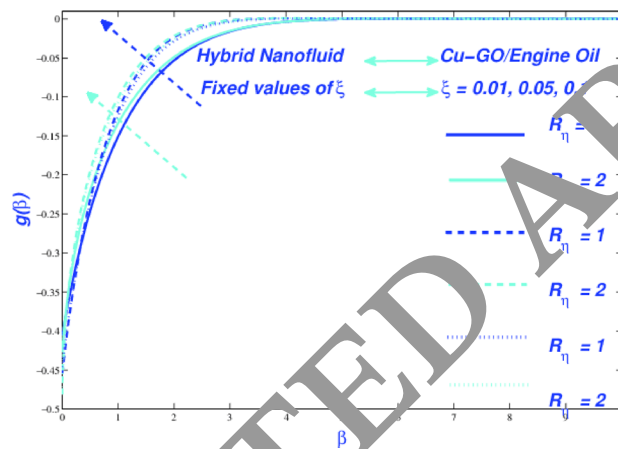


Figure 7. Impact of R_η on g .

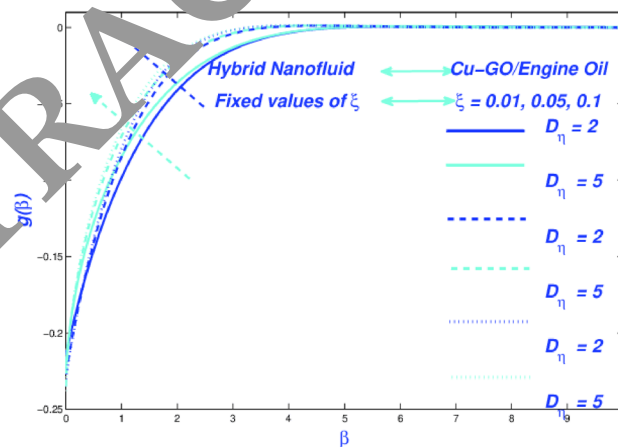


Figure 8. Effect of D_η on f' .

like shearing dilatation due to a consistent change in D_η . It's intriguing to see how increasing the quantity of nanomolecules influences liquid thickness while lowering it. Physically, boosting the amount of nanostructure particles enhances liquid consistency, lowering liquid speed and $g(\beta)$. Figure 9 shows the impact of Γ_1 on $f'(\beta)$. An amplification of Γ_1 lessens the worth of $f'(\beta)$. In the status of slippery limit restrictions, the speed of the plate and the liquid are not equivalent at the plate, bringing about a decrease in liquid speed and a diminishing

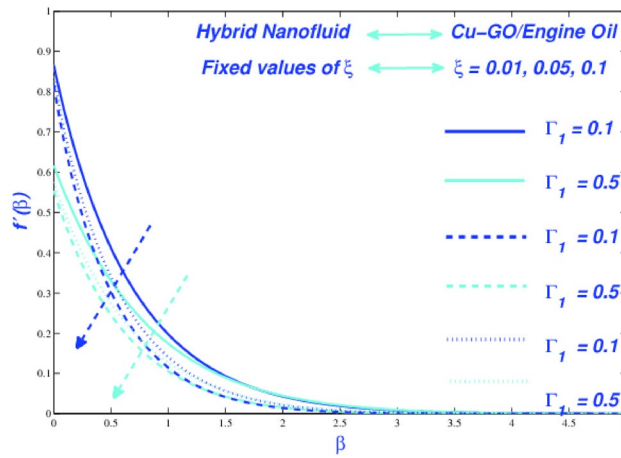


Figure 9. Effect of Γ_1 on f' .

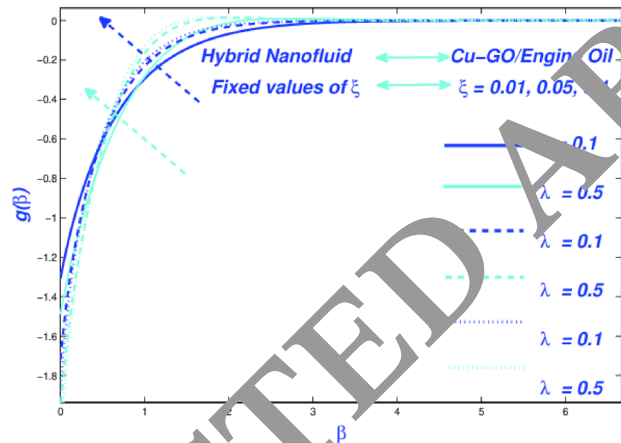


Figure 10. Impact of λ on g .

speed. Figure 9 shows a portrayal of $g(\beta)$. This is physically because the liquid near the boundary layer is more viscous due to the accumulation of particles close to the surface, which reduces the velocity and increases the further away from the boundary layer. Another important concept is that as the percentage of nanoparticles in the liquid grows, the thickness of the liquid reduces, making it simpler to travel across an extensible plate. Magnification in the volume part of nanomolecules builds a liquid and diminishes the liquid speed and $g(\beta)$.

Figure 11 is intended to depict R_δ performing on $\theta(\beta)$. R_δ is the most thing of a heat transfer rules in terms of physics. It is commonly known that amplification in R_δ causes the heat transfer rate to increase. It is because of an improvement in R_δ lowers the average absorbing factor, resulting in amplification in $\theta(\beta)$. Practically, an increase in the size of the nanomolecules paired with R_δ enhances the thermal conducting of the fluid, boosting $\theta(\beta)$. The effect of P_r on $\theta(\beta)$ is depicted in Fig. 12. When P_r is small, heat diffuses quickly in comparison to velocity (momentum), and vice versa when P_r is large. Furthermore, because of amplification in P_r , the thickness of the thermal boundary layer declines $\theta(\beta)$. This is physically due to the inverse relationship between the Prandtl number and the thermal diffusivity, as the lack of thermal diffusivity occurs as a result of the low thermal conducting and thus enhances the Prandtl number, which works to increase the temperature inside the nanoliquid. The link between Γ_1 and temperature is seen in Fig. 13. A magnification of Γ_1 reduces the space among the surface and surrounding heat, transporting less temperature from a plate to a liquid and, due to the lowering a fluid heat.

Figure 14 emphasizes the effect of chemically response charge σ at the awareness area $\phi(\beta)$. The physical interpretation refers to the amount $\sigma(1 + \delta\theta)^n \exp\left(\frac{-E}{1+\delta\theta}\right)$ magnifies at the likewise of improvement in σ or n which inspires the destructive chemically reactive action which diminishes the mass size range. The exponential part in the formula means that when the active energy diminishes, the rate constant of a reaction grows exponentially. Because the rate of a reaction is directly proportionate to its rate constant, the rate also grows exponentially¹²⁵. The impact of S_δ at the mass area $\phi(\beta)$ is defined in Fig. 15. The Schmidt quantity is the ratio of momentum to mass diffusivity. It's well worth noting that a high-quality alternate in S_δ reduces mass diffusivity.

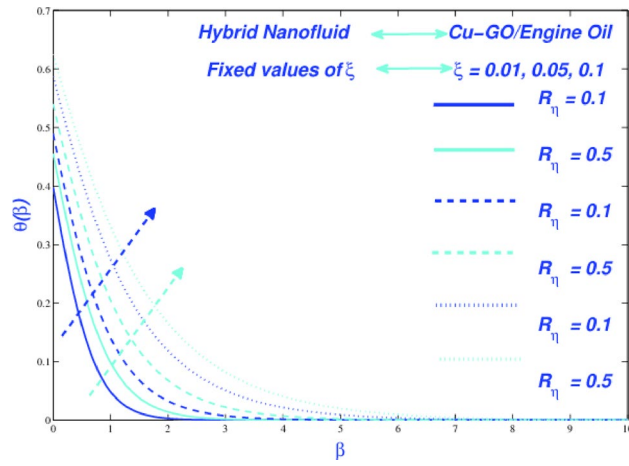


Figure 11. Influence of R_η on θ .

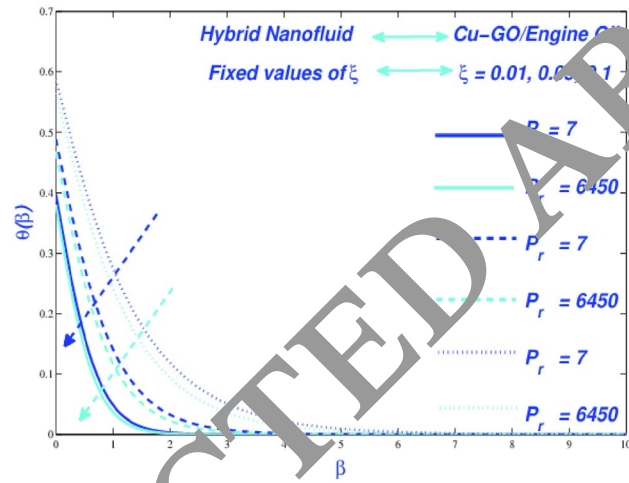


Figure 12. Influence of P_r on θ .

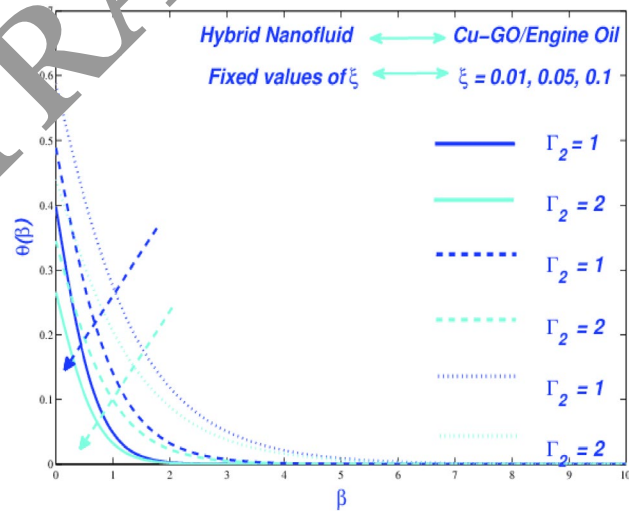


Figure 13. Impact of Γ_2 on θ .

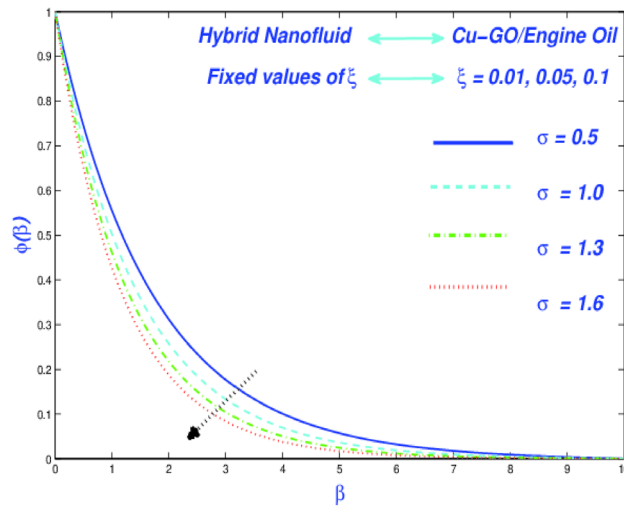


Figure 14. Effect of σ on ϕ .

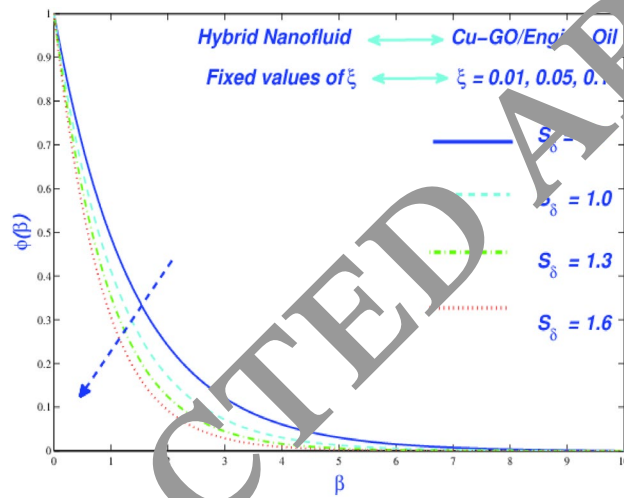


Figure 15. Effect of S_δ on ϕ .

Physically, the fluid viscosity falls because of a growth in $S_\delta = \frac{\nu}{D}$, which reduces mass diffusion and will increase momentum diffusivity. The presence of S_δ maximum possibly reduces the fluid viscosity and $\phi(\beta)$.

Conclusions

3-D rotating Sutterby hybrid fluid with copper-Graphene oxide nanomolecules, active energy, impetus, heat slippery boundary constraints, and radiative heat flow is defined in this paper. The numerical solution to the simulated problem was achieved using the MATLAB KBM built-in technique. The following are some of the most important aspects of the results:

- The profile $f'(\eta)$ denigrates at the behalf of extension in R_η , D_η , and N .
- Magnification within the factors λ and N monitors to an extension in $g(\beta)$.
- Intensification in θ_w boosts $\theta(\beta)$ however a decline in $\theta(\beta)$ occurs due to an enhancement in R_δ .
- The value of the Nusselt wide variety decreases below amplification in Γ_1 .
- It is essential that $\phi(\beta)$ will increase within the case of extension in ξ .
- A positive variant in Γ_2 will increase $\phi(\beta)$.
- The mass fractional size discipline outline reduces for the chemically response factor Γ .

The Keller-box method could be applied to a variety of physical and technical challenges in the future^{126–139}.

Data availability

The results of this study are available only within the paper to support the data.

Received: 21 March 2022; Accepted: 28 June 2022

Published online: 07 July 2022

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Author contributions

W.J. formulated the problem. W.J. and M.R.E. solved the problem. W.J., M.R.E., R.S., A.A.P., M.A., Z.R., S.S.P.M.I., and W.W. computed and scrutinized the results. All the authors equally contributed in writing and proof reading of the paper. All authors reviewed the manuscript.

Competing interests

The authors declare no competing interests.

Additional information

Correspondence and requests for materials should be addressed to W.W.

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