# Self-passivating tungsten alloys of the system W-Cr-Y for high temperature applications

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#### Abstract

Self-passivating tungsten based alloys for the first wall armor of future fusion reactors are expected to provide a major safety advantage compared to pure tungsten in case of a loss-of-coolant accident with simultaneous air ingress, due to the formation of a stable protective scale at high temperatures in presence of oxygen which prevents the formation of volatile and radioactive WO<sub>3</sub>. This work analyses the oxidation and thermo-shock resistance of W-Cr-Y alloys obtained by mechanical alloying followed by HIPing. Alloys with different Cr and Y contents are produced in fully dense form with nanocrystalline or ultrafinegrained microstructure and a dispersion of Y-rich oxide nanoparticles located mainly at the grain boundaries. Isothermal oxidation experiments confirm an excellent oxidation resistance due to the formation of protective oxide scales at the very surface. These layers mainly consist of  $Cr_2O_3$  and mixed Y-W and Cr-W oxides. The superior oxidation resistance of these alloys is confirmed by tests simulating accident-like conditions. The thermal conductivity of these alloys at 600-1000°C is 2-3 times higher than standard Nibase superalloys like Inconel-718. The material also exhibits outstanding thermal-shock resistance: 1000 pulses of 0.19 GW/m<sup>2</sup> power density and 1 ms duration at 400°C base temperature resulted in no damage, while an increased power density of 0.38 GW/m<sup>2</sup> resulted in the formation of a crack-network and slight

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surface roughening. An additional thermal treatment at 1550 °C improves slightly the oxidation and significantly the thermal shock resistance of the alloy.

## Keywords

Tungsten alloy, self-passivating alloy, oxidation resistance, thermal shock resistance, nuclear fusion, plasma-facing materials

## 1. Introduction

The highest melting point among all metals and a comparably high thermal conductivity make tungsten (W) very attractive for many high temperature applications such as power generation, concentrated solar power, target of spallation sources, and plasma-facing materials (PFM) of future fusion reactors among others. However, the poor oxidation resistance of W limits its use either to a vacuum environment or to temperatures below 500°C in oxidizing environments. In the application as PFM for future fusion reactors such as DEMO, W will be exposed to high steady state heat loads in the MW/m<sup>2</sup> regime, especially at the so-called divertor [1], but the high oxidation rate of W does not represent a concern in a deuterium-tritium plasma environment under normal operation conditions. However, a loss-of-coolant accident (LOCA) with simultaneous air ingress into the vacuum vessel would lead to a temperature rise of the in-vessel components between 1000 and 1200°C (depending on the reactor design) within a view to tens of days due to the nuclear afterheat [2]. Under this situation, the use of pure W in DEMO represents a safety risk since part of the W would oxidize forming volatile WO<sub>3</sub> with the potential release of significant amounts of highly radioactively activated species. A possible solution to this important safety issue is the addition of oxide forming alloying elements to pure W, resulting in the growth of a stable protective oxide scale that prevents W from oxidation at high temperatures in presence of oxygen. Under normal operation, the surface of these self-passivating alloys will consist of pure W due to preferential sputtering of the alloying elements by hydrogen isotopes coming from the fusion plasma. A proper selection of the alloying elements is crucial for fusion application since they have to exhibit sufficiently low activation by neutron irradiation from the plasma, in addition to other important properties such as low volume increase of the oxides, low vapor pressure, high melting point and no formation of undesirable intermetalics. For this reason, alloying elements forming stable oxides like AI, Nb or Ni are not permitted because of their high activation by neutrons.

During the last years significant efforts were undertaken to develop self-passivating W alloys for fusion application, starting from thin films of the systems W-Cr-Si and W-Cr-Ti [3,4] which served as "model materials" for the manufacturing of bulk alloys by powder metallurgical (PM) routes [5,6]. More recently, it has been demonstrated that W alloys of the system W-Cr-Y result in a significant reduction of the oxidation rate at temperatures up to 1000°C compared to previously studied ternary systems [7,8]. Besides, the addition of Y leads to the formation of a  $Y_2O_3$  nanodispersion which inhibits grain growth and acts as reinforcement, as reported also in [9,10] for pure W.

In this work the most recent developments on self-passivating alloys of the system W-Cr-Y manufactured by PM for application at the DEMO blanket first wall are presented. The purpose of this research is to develop a W-based material with good oxidation resistance due to self-passivation up to at least 1000°C,

together with as high as possible thermal conductivity and high mechanical strength. In this way, the material should be able to withstand not only the expected thermal load during normal operation, but also transient thermal loads like the so-called edge localized modes (ELMs), during which a large amount of energy is loaded on the PFMs in a very short time [11]. Results on W-Cr-Y alloys with different Cr and Y contents, manufactured by mechanical alloying (MA) and densified by Hot Isostatic Pressing (HIP), are presented. The alloys are subjected to a subsequent heat treatment (HT) at 1550°C, i.e. above the spinodal decomposition temperature of the W-Cr phase diagram, to dissolve the Cr-rich phase and achieve a single bcc phase [12]. By this means, thermal stresses induced by the second phase, which may appear during thermal shock loads, are avoided [7,13]. Microstructural investigations and microhardness of the as HIPed and HIP+ HTed materials are shown and compared to alloys of previous investigations. Thermal conductivity and flexural strength, results of oxidation tests under isothermal and accident-like conditions, as well as thermal shock tests at JUDITH-1 (Juelich Divertor Test Facility Hot Cells) [14] simulating loads under transient events are included.

## 2. Experimental

Elemental powders of pure W (99.95%, 15-30 µm), Cr (99.95%, 74 µm) and Y (99.9%, 20-30 µm) were used to produce samples of the W-Cr-Y system with Cr contents of 8 and 10 wt.% and Y contents of 0.3, 0.5 and 1 wt.%. The starting powders were mechanically alloyed under Ar atmosphere in a planetary ball mill using WC grinding jars and balls. The MA parameters were optimized to obtain a homogeneously alloyed powder at the minimum milling time to keep impurities from jars and balls as low as possible. Metallic capsules with the alloyed powder were evacuated, degassed, sealed and HIPed at 1250 °C for 2 h at 150 MPa. After HIP a heat treatment (HT) was performed at 1550 °C under H<sub>2</sub> atmosphere. The oxygen and nitrogen contents of powders and bulk materials were determined using the inert gas fusion method (ASTM E1569), and the carbon content by the combustion method (ASTM E1019). Powders and bulk samples were characterized by field emission scanning electron microscopy (FE-SEM) and energy dispersive X-ray spectroscopy (EDS). The relative density of the samples was determined from the geometrical and theoretical densities. The average grain size of the dense materials was determined by quantitative metallography. The Vickers microhardness was measured applying a load of 0.5 kg for 5 seconds. Three-point bending tests (3PBT) were performed on smooth and single edge laser-notched beams [15] of nominal dimensions 1.8 × 1.8 × 20 mm<sup>3</sup> over the temperature range 20 - 1100 °C under high vacuum. All tests were performed in displacement control at a fixed loading rate of 100 µm/min with a 16 mm span width. Flexural strength was computed by Euler–Bernoulli equations for slender beams up to failure. However, when yield stress was exceeded, the 0.2% strength offset was reported. Fracture toughness, i.e. the stress intensity factor for mode I stress, was then computed from the critical load and the beam section using the equation proposed by Guinea et al. [16]. The thermal conductivity was measured by the laser flash method.

Isothermal oxidation tests at 800°C and 1000°C for up to 60 h were performed by thermogravimetric analysis (TGA) using a mixture of 80% Ar and 20%  $O_2$  at atmospheric pressure. The mass gain of the samples due to oxide formation was measured by a thermobalance with a sensitivity of 0.025 µg. The procedure is described in detail in [4]. Besides, oxidation tests simulating accident-like conditions were also performed. These tests consist of a preheating in Ar 6.0 up to 600°C followed by oxidation in a mixture of 80 vol.% Ar and 20 vol.%  $O_2$  at linear increasing temperature from 600 to 1000°C during about 17 h, two isothermal oxidation steps in air at 1000°C for 1 h, each of them followed by isothermal steps in Ar 6.0

at 1000°C for 1 h, and cooling down in Ar. Surfaces and cross sections of the oxidized samples were analyzed by FE-SEM and Focused Ion Beam (FIB).

Thermal shock tests were performed at the electron beam facility JUDITH-1 [14] on samples of dimensions  $10 \times 10 \times 4 \text{ mm}^3$ , which were exposed at a base temperature of 400°C to loads consisting of 1000 pulses with power densities of 0.19 and 0.38 GW/m<sup>2</sup> for 1 ms simulating the conditions expected at the divertor under ELMs. To ensure a homogeneous loading, a small area (4×4 mm<sup>2</sup>) was scanned with a focused electron beam at very high scanning frequencies. After exposure, the surface and cross section of the samples were investigated by optical microscopy.

## 3. Results and discussion

Mechanically alloyed powders of compositions (in wt.%) W-8Cr-0.5Y, W-10Cr-0.3Y, W-10Cr-0.5Y and W-10Cr-1Y (corresponding to W-23Cr-0.9Y, W-28Cr-0.5Y, W-28Cr-0.8Y and W-28Cr-1.6Y in at.%, respectively) were can encapsulated and HIPed at 1250 °C for 2 h, achieving relative densities >99% (Table 1), i.e. the materials are fully dense within the experimental error. The contents of interstitial elements after MA amount to 1200 - 1500 ppm for oxygen, 100 - 160 ppm for nitrogen and 200 - 400 ppm for carbon, and remain the same after HIP.

Alloy composition	HT at 1550°C	Relative density (%)	Average grain size (μm)	Microhardness	
				Alloys (HV <sub>0.5</sub> )	Pure W (HV) [21]
W-8Cr-0.5Y	No	99.1	102 ± 3	1265 ± 4	1340
W-10Cr-0.3Y	No	99.8	84 ± 3	1145 ± 3	1441
W-10Cr-0.5Y	No	99.8	110 ± 4	1220 ± 8	1304
W-10Cr-1Y	No	99.6	228 ± 8	1190 ± 8	1012
W-10Cr-0.5Y	Yes	98.7	230 ± 11	980 ± 6	1009

Table 1: Relative density, average grain size and microhardness of W-Cr-Y alloys of different compositions after HIP at 1250°C and after subsequent HT at 1550°C (only for W-10Cr-0.5Y).

Besides, a HT after HIP was performed on alloy W-10Cr-0.5Y at 1550 °C, i.e. at a temperature above the spinodal decomposition of the W-Cr phase diagram, see Fig. 1. According to this phase diagram, after HIPing at 1250 °C two bcc phases are formed: a ( $\alpha$ W,Cr) main phase with minor presence of ( $\alpha$ Cr,W). The compositions of these two phases were estimated from the shift of the peaks obtained by X-ray diffraction (XRD) with respect to the pure W and Cr peaks. These compositions are indicated by the blue lines in the phase diagram of Fig. 1, and agree well with the predictions of the phase diagram at 1250°C. The purpose of this HT is to dissolve the Cr-rich phase, more prone to oxidation, and to produce a single phase material (see red cross and lines in Fig. 1) in which misfit strains and thermal stresses induced by the second phase are avoided. HT temperatures higher than 1550°C are avoided due to the presence of an eutectic W-Y<sub>2</sub>O<sub>3</sub> at 1560°C [17], which would lead to a significant grow of the Y<sub>2</sub>O<sub>3</sub> nanoparticles (described in section 3.1).

After the HT, the content of interstitials remains practically unchanged. A slight reduction of relative density is observed after HT, which may be related to residual Ar gas trapped in the powders during MA [18]; during HIPing the Ar containing micro-pores collapse and a nearly dense material is obtained due to the

high pressure applied. However, if a HT without pressure is performed at a temperature higher than the one of the HIP cycle, the small pores grow contributing to the formation of so-called temperature induced porosity [19].

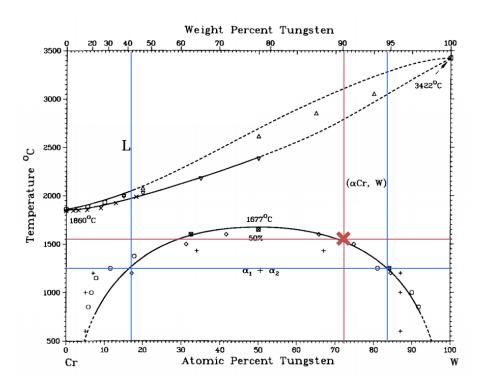


Figure 1: W-Cr phase diagram [12] indicating the composition in the W-10Cr-0.5Y alloy of the bcc ( αW,Cr) and ( αCr,W) phases after HIP deduced from the peak positions of the corresponding XRD peaks (blue lines). The composition of the single bcc phase expected after HT at 1550 °C is indicated (red cross and lines).

#### 3.1 Microstructure

After HIP, a very fine and homogeneous microstructure can be appreciated in all alloys (Fig 2 (a)-(c)), with two main phases identified by EDS as a W-rich phase with Cr in solution (( $\alpha$ W,Cr), bright grey majority phase) and a Cr-rich phase with W in solution (( $\alpha$ Cr,W), dark grey discontinuous phase). This microstructure and the composition of the phases determined by XRD are consistent with the predictions of the W-Cr phase diagram at 1250 °C, Fig. 1. For the alloys containing 0.3 and 0.5% Y the average grain size of the matrix is of the order of 100 nm (see Table 1), i.e. considerably finer than that of the binary system W-15Cr described in [7], indicating that Y acts as grain growth inhibitor, as reported in [9,10]. A dispersion of Y<sub>2</sub>O<sub>3</sub> nanoparticles of about 15 nm size is formed mainly at the grain boundaries (GB) as a result of the decomposition of less stable oxides during HIPing and the high oxygen affinity of Y, which reacts with the O introduced in the alloy during MA. This cleaning effect of GB from oxygen together with the grain refinement are supposed to be beneficial for improving the alloy's strength at high temperature [9,10,20]. For a Y content of 1% the amount of oxygen present in the material is not enough to form Y<sub>2</sub>O<sub>3</sub>, i.e. there is an excess of Y. This leads to coarsening of the Y-O-rich nanoparticles and thus, to a less efficient inhibition of grain growth: as can be seen in Table 1 and Fig 2 (c), the matrix grain size in this case is above 200 nm. The microstructure of the W-10Cr-0.5Y alloy after HT at 1550  $^{\circ}$ C is shown in Fig. 2 (d) and consists of a single ( $\alpha$ W,Cr) phase, as expected from the W-Cr phase diagram. Residual particles of ( $\alpha$ Cr,W) phase can be also found. As a consequence of the HT, grain growth up to 230 nm is observed in the matrix (Table 1), as well as a thickening of the Y-rich oxide nanoparticles, up to about 50 nm.

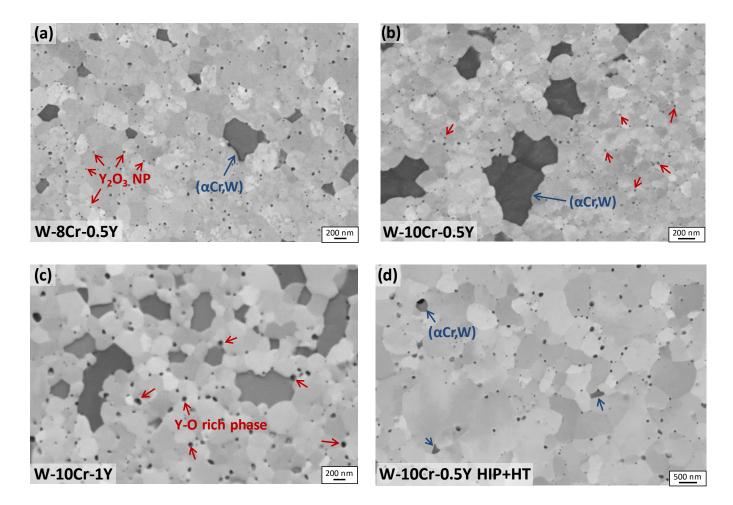


Figure 2: Microstructure after HIP of (a) W-8Cr-0.5Y alloy, (b) W-10Cr-0.5Y alloy, (c) W-10Cr-1Y alloy, and (d) of W-10Cr-0.5Y alloy after HIP + HT.

#### 3.2 Microhardness and thermal conductivity

The Vickers microhardness of all alloys after HIP and of W-10Cr-0.5Y alloy after HIP + HT is listed in Table 1. As reference, the micro-hardness of pure polycrystalline W at the corresponding average grain size is included, taking into account its Hall-Petch relation-ship of pure W [21]. For the alloys containing 0.3 and 0.5% Y both as-HIPed and HIP+HTed, these values are only slightly lower than those of pure W at the corresponding grain size. Even though a comparison with pure W is not straightforward because our alloys have a significant amount of Cr, and because other hardening mechanisms like solution and precipitation hardening may be also present, it can be nevertheless stated that there is a strong influence of grain size on hardening.

The thermal conductivity as a function of temperature of the as HIPed alloys W-8Cr-0.5Y, W-10Cr-0.5Y and W-10Cr-1Y is shown in Fig. 3 together with that of the binary alloy W-10Cr and of Inconel 718 for comparison. The thermal conductivity of all W-Cr-Y alloys is similar in the whole temperature range. The

slight decrease of thermal conductivity of the W-Cr-Y alloys compared to the binary system W-10Cr may indicate that a small amount of Y is in solid solution. Compared to Inconel 718, the thermal conductivity of the W-Cr-Y alloys at 600-900°C is about 2.5 times higher than Inconel 718. As a consequence, the resulting temperature increase for a thickness of 2 mm is < 100 K and thus, not of concern for the intended application as PFM of the DEMO blanket with expected power densities < 2 MW/m<sup>2</sup> [22].

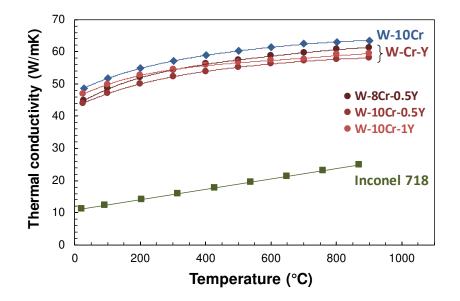


Figure 3: Thermal conductivity as a function of temperature of binary alloy W-10Cr, ternary alloys W-Cr-Y and Inconel 718 for comparison.

#### 3.3 Mechanical properties

Non-standard 3PBT tests were performed on both as-HIPed and after HT W-10Cr-0.5Y alloys from 25 °C to 1100 °C to explore their mechanical response, i.e. fracture toughness and flexural strength.

Toughness values are plotted as a function of temperature in Fig. 4. Both materials exhibited similar behavior at temperatures above 600 °C, with toughness values around 6 MPa.m<sup>1/2</sup>, which are almost maintained above this temperature. However, at lower temperatures, fracture values of the W-10Cr-0.5Y alloy after HT were 30 % higher than these of the as HIPed alloy. The reduction of the Cr-rich phase with W in solution ( $\alpha$ Cr,W), which is indeed the weakest phase of the system, together with the improved adhesion as a consequence of the grain growth could be the reason of this behavior.

From the flexural strength tests (Fig. 5 (a) and (b)) it can be observed that both specimens showe brittle behavior up to 900 °C while some plastic deformation takes place at 950 °C. Above 1000 °C, the material clearly exhibits plastic deformation. Thus, the ductile-to-brittle transition temperature (DBTT) lies around 950 °C. At 1100 °C the material deforms plastically without breaking. Although the W-Cr-Ti system presented a similar DBTT [5], some decrease was expected taking into account the grain size reduction [23], but there is no variation though. Most probably, the Y-rich nanoparticles contribute to reinforcement but not to ductilization because they are located mainly at the GB contributing to their weakening. In fact, the fracture is intergranular for all tested temperatures up to 1050°C across the W-rich phase. Above 900°C there is some transgranular fracture across the Cr-rich grains.

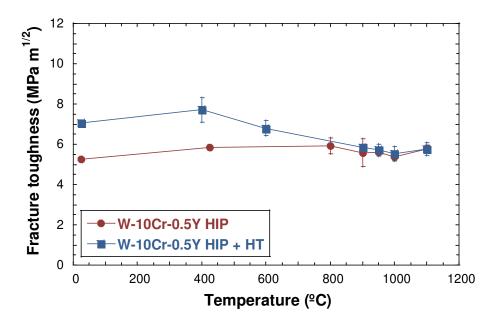


Figure 4: Average fracture toughness versus test temperature for W-10Cr-0.5Y materials. Mean values and standard error.

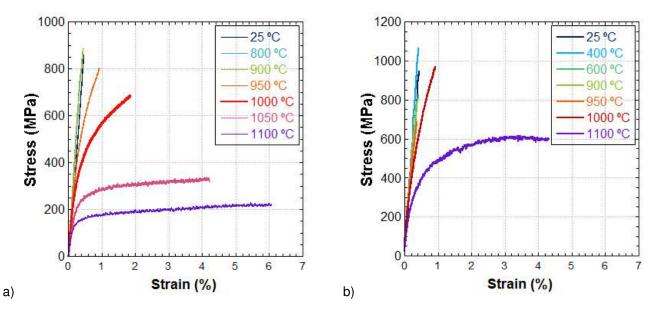


Figure 5: Flexural strength vs strain from 25 °C up to 1100 °C of (a) W-10Cr-0.5Y alloy after HIP, and (b) after HIP + HT.

The proof flexural strength of both alloys can be observed in Fig. 6, where ultimate flexural strength is plotted as a function of temperature. Above the DBTT, the 0.2% yield strength offset is used and represented with dashed lines and open symbols. Both materials maintained the strength between RT and 900 °C, but then the strength decreases slightly. As observed for the fracture toughness, the effect of the HT is noticeable just at temperatures below the DBTT. Above this temperature its strength decreases slightly compared to the as HIPed material.

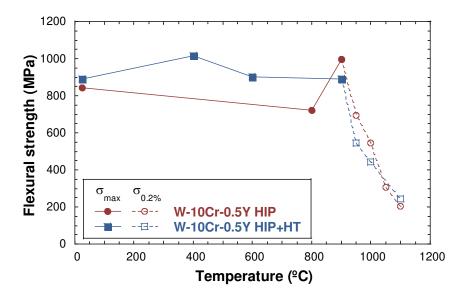
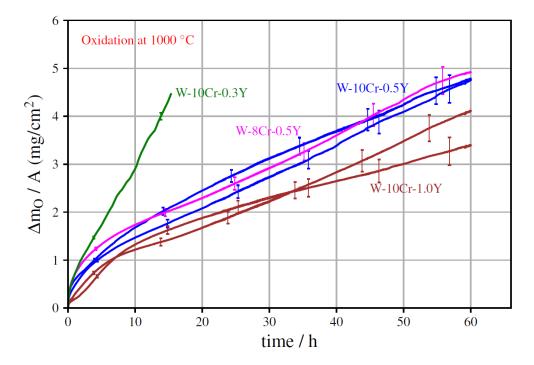


Figure 6: Proof flexural strength as a function of temperature for the as HIPed W-10Cr-0.5Y alloy and the alloy after after HIP + HT. Mean values and standard error. Full symbols and continuous lines illustrate brittle behavior and hence, ultimate flexural strength instead of yield strength.



#### 3.4 Oxidation tests

Figure 7: Mass gain per unit area during isothermal oxidation of different as-HIPed W-Cr-Y alloys at 1000°C. The error bars correspond to the drift of the thermos-balance provided by the TGA manufacturer. Lines of the same colors correspond to repetition of the measurement with a new sample.

Isothermal oxidation tests at 800  $^{\circ}$ C and 1000  $^{\circ}$ C for up to 60 h were performed on W-Cr-Y alloys of different Cr and Y contents. The mass gain per unit area of each alloy during isothermal oxidation at 1000  $^{\circ}$ C is shown in Fig. 7. The error bars corresponds to the drift of the TGA thermo-balance of 5 µg/h provided by the manufacturer (Netzsch) and increase, thus, with increasing exposure time. Lines of the

same color in Fig. 7 correspond to a second measurement with a new sample to assure the reproducibility, which appears to be satisfactory. For alloys containing 0.5 and 1% Y, the initially parabolic kinetics transform into linear for longer exposure times. Alloy of composition W-10Cr-1Y exhibit the lowest oxidation rate of all investigated alloys so far [6,7], while an Y content of 0.3% Y results in a significantly higher oxidation rate, indicating that 0.3 Y does not provide sufficient protection against oxidation. The difference between the oxidation rates of alloys W-8Cr-0.5Y and W-10Cr-0.5Y does not seems relevant.

In the Arrhenius plot of Fig. 8 the linear oxidation rates of pure W, W-10Cr-0.5Y, W-10Cr-1Y (measured at 1000 and 1100°C) and W-12Cr-0.5Y (the latter from [7] and only for 800°C) are shown vs the inverse of temperature. At 800°C the oxidation rates of W-10Cr-0.5Y alloy are four orders of magnitude lower than pure W, and at 1000°C the ones of W-10Cr-0.5Y and W-10Cr-1Y are about three orders of magnitude lower than pure W.

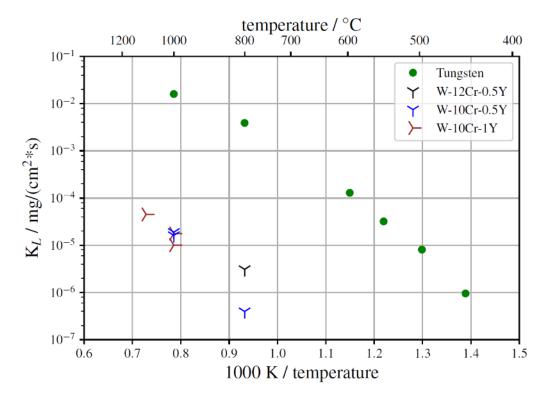


Figure 8: Arrhenius plot of linear oxidation rates of as-HIPed W-Cr-Y alloys and pure W for comparison. Data for W-12Cr-0.5Y are taken form [7].

In Fig. 9 the mass gain per unit area during tests simulating accident-like conditions up to 1000 °C is shown for as-HIPed W-Cr-Y alloys of different composition as well as for the W-10Cr-2Ti alloy of previous work [7] for comparison. Alloys of the W-Cr-Y system exhibit the lowest mass gain of all investigated alloys, except the one with 0.3% Y, which starts failing at about 800°C after about 9 h exposure to synthetic air. The W-10Cr-1Y alloy exhibit the lowest oxidation rate, in agreement with the isothermal oxidation tests (Fig. 7). Under accident-like conditions the W-10Cr-0.5Y alloy results in slightly better oxidation resistance than the W-8Cr-0.5Y one, and in clearly lower mass gains than the W-12Cr-0.5Y one, indicating that 10% Cr provides better oxidation protection than 8 and 12% Cr. This is only partly in agreement with the findings of [8], where the optimum Cr and Y concentration for lowest oxidation rate in isothermal oxidation tests at 1000°C was investigated for thin films produced by magnetron sputtering. In this work lowest oxidation rate were found for 11.4% Cr with a very slight increase for concentrations beyond 12% and significantly

higher rates for about 10% Cr. One has to take into account that a comparison between bulk materials and thin films is not straightforward because of the many microstructural differences between the two systems. Samples of alloys W-10Cr-0.5Y and W-10Cr-2Ti subjected to HT were studied in [13] and [7], respectively, showing only a slight decrease in mass gain with respect to the as-HIPed material. In Fig. 9 it can be observed that the mass increase of all alloys W-Cr-Y alloys (except the W-10Cr-0.3Y one) follows a parabolic law at the second isothermal segments. Another important observation is the fact that in the isothermal segments without oxygen there is no mass loss, indicating that there is no evaporation of tungsten oxide, previously formed during the oxidation segment.

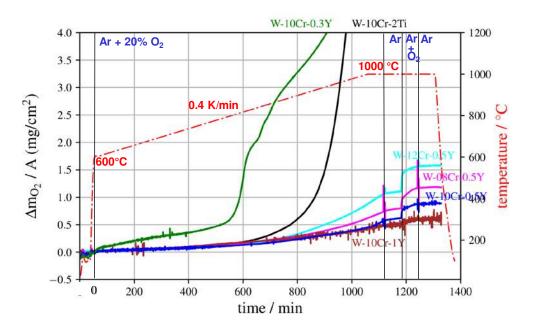


Figure 9: Mass increase per unit area during tests simulating accident-like conditions on different as HIPed W-Cr-Y alloys, and comparison with previous works [6, 7]. After preheating in Ar up to 600°C, synthetic air is introduced and temperature is increased linearly from 600 to 1000°C, followed by 1 h isothermal oxidation in air at 1000°C, an isothermal step in pure Ar for 1 h, repetition of the two isothermal steps and cooling down in Ar.

The surface and cross-sections of the formed scales both after isothermal and accident-like oxidation were analyzed by FE-SEM, FIB and EDS. In Fig. 10 the cross-sections of the W-10Cr-0.5Y and W-10Cr-1Y alloys after accident-like oxidation are presented. In both cases, three layers can be distinguished: a thin  $Cr_2O_3$  scale and a thin  $Cr_2WO_6$  layer just below, followed by a thicker WO\_3 layer. Partial oxidation of the ( $\alpha Cr$ ,W) phase located below and near the scales can be also observed, as well as a dark phase along the GB identified as  $Cr_2O_3$ , indicating diffusion of cations and anions along the GB. After isothermal oxidation with longer exposure times than the accident-like tests, W-Y mixed oxides can be appreciated above and inside the thin  $Cr_2O_3$  layer. Analysis of these mixed oxides by grazing incidence XRD and EDS allowed their identification as  $Y_2W_3O_{12}$ , in agreement with the findings of Telu et al. in the system W-Cr- $Y_2O_3$  [24]. This mixed oxide has been also found to a much lesser extent after 60 h isothermal oxidation at 800°C.

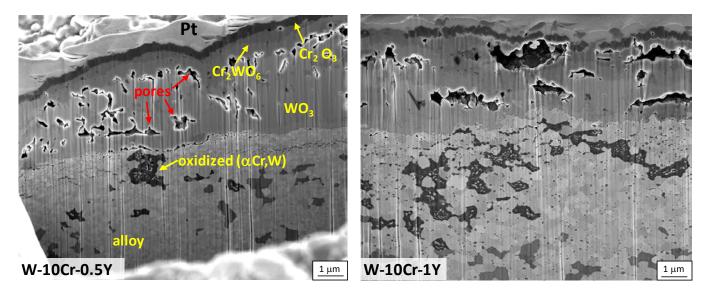


Figure 10: Cross-section of (left) as-HIPed W-10Cr-0.5Y and (right) W-10Cr-1Y alloys after the accident-like oxidation tests.

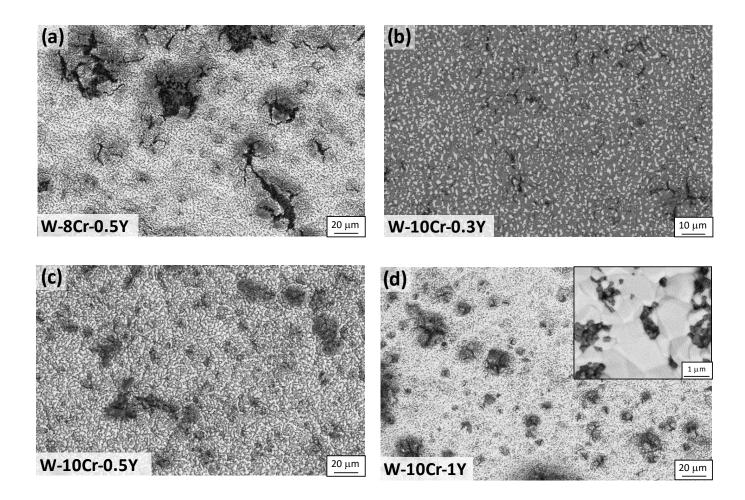


Figure 11: Surface of alloys (a) W-8Cr-0.5Y, (b) W-10Cr-0.3Y, (c) W-10Cr-0.5Y, and (d) W-10Cr-1Y alloy after 24 h isothermal oxidation at 1000°C.

The surface of the alloys W-8Cr-0.5Y, W-10Cr-0.3Y alloy, W-10Cr-0.5Y and W-10Cr-1Y after 24 h isothermal oxidation at 1000°C is shown in Fig. 11. A network of white particles corresponding to the  $Y_2W_3O_{12}$  oxide can be appreciated. The alloy with 1% Y exhibits the highest  $Y_2W_3O_{12}$  particle density, almost covering the thin  $Cr_2O_3$  layer, which has the aspect of small black particles (s. insert in Fig. 11 (d)). In the alloy with 0.3% Y the  $Y_2W_3O_{12}$  particles are very dispersed above the  $Cr_2O_3$  layer (Fig. 11 (b), while in the alloys with 0.5% Y (Fig. 11 (a) and (c)) these particles start to cover the  $Cr_2O_3$  layer. In the alloy with 8% Cr the  $Cr_2O_3$  layer exhibits cracks, probably because there is not enough Cr to form a continuous and stable protecting scale, which explains the higher oxidation rate of this alloy compared to W-10Cr-0.5Y. Taking into account the high oxidation rate of the W-10Cr-0.3Y alloy (Fig. 7 and 9), it is evident that the  $Y_2W_3O_{12}$  phase plays an important role in the passivation effect of W-Cr-Y alloys. It seems that the  $Y_2W_3O_{12}$  particles reinforce the protecting effect of the  $Cr_2O_3$  scale.

Taking into account the much lower amount of this phase after isothermal oxidation at 800°C, and the fact that  $WO_3(s)$  starts to sublimate at 750°C [25], it is plausible to assume that  $Y_2W_3O_{12}$  forms by the reaction of Y2O3 with  $WO_3(g)$ , as proposed also in [24].

### 3.5 Thermal shock tests

The as HIPed and the HIP+HTed W-10Cr-0.5Y alloys were exposed to 1000 heat pulses of 0.19 GW/m<sup>2</sup> for 1 ms at 400 °C and subsequently to similar 1000 heat pulses of 0.38 GW/m<sup>2</sup>. The latter pulses simulate the conditions expected during ELMs in the divertor, while the load at the blanket first wall during ELMs will be significantly lower. Nevertheless, in a DEMO-like device massive gas injection will be used to mitigate or avoid disruptions; this gas injection will produce photon flashes, which result in very similar loads at the first wall to those applied here in a single pulse. Thus, this test simulates the load at the first wall after 1000 mitigated disruptions.

Taking into account the absorbed load of  $0.38 \text{ GW/m}^2$ , the pulse duration of 1 ms and the thermo-physical properties of the materials at the base temperature of 400 °C, a temperature increase of about 1000 K is expected. The fast increase in surface temperature during thermal shock loading and the formation of a steep temperature gradient causes high compressive stresses in the small heat affected zone. At temperatures above the DBTT the ductility of the material should be able to compensate the plastic deformation and the induced stresses resulting in an increased surface roughness. In contrast, at temperatures below the DBTT the risk of brittle crack formation increases due to fast cooling across the DBTT. In the present case, after the pulses of 0.19 GW/m<sup>2</sup> no damage was detected at the surface of the samples. After 1000 pulses of 0.38 GW/m<sup>2</sup> the surface of the as-HIPed material was chipped whereas the sample after HT exhibited just a crack network with no chipping, as can be observed in Fig. 12. The surface roughness of the HIP+HTed material was reduced with respect to pure W grade reference material IGP W [26] under the same conditions from 0.52  $\mu$ m to 0.26  $\mu$ m. The cracks in the alloy after HT are clearly narrower than those of the as-HIPed material. Compared to the pure W reference material, the HTed W-10Cr-0.5Y alloy exhibits a larger crack density, but the crack length and width is significantly lower for the alloy than for pure W.

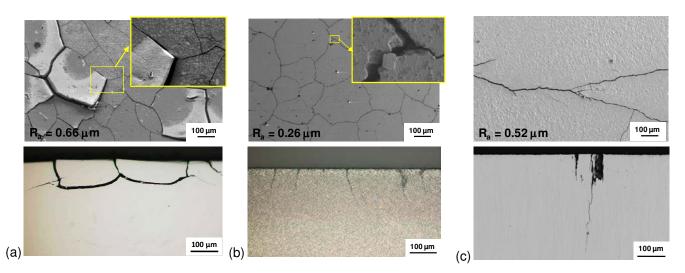


Figure 12: Surface and cross section after loading at JUDITH-1 with 1000 ELM-like pulses of 0.38 GW/m<sup>2</sup> for 1 ms on (a) as-HIPed W-10Cr-0.5Y, (b) HIP+HTed W-10Cr-0.5Yand (c) pure tungsten reference material IGP W [26]

The W-10Cr-0.5Y alloy presents an improved performance to thermal shock loading compared to previously tested alloys of the system W-Cr-Ti [7], which is likely due to the reinforcement attained in both strength and toughness at high temperatures. The very fine microstructure and the presence of Y-rich oxide nanoparticles together with the slightly higher thermal conductivity might be responsible for this improvement. The HT clearly contributes to improve the thermal shock resistance since only a single phase is present in the material, which leads to stress relief.

## 4. Conclusions

Fully dense self-passivating alloys of compositions W-8Cr-0.5Y, W-10Cr-0.3Y, W-10Cr-0.5Y and W-10Cr-1Y (in wt.%) were produced by MA and HIP. Alloys containing 0.3 and 0.5% Y are nanocrystalline and exhibit a  $Y_2O_3$  nanoparticle dispersion mainly at the GB, inhibiting grain growth and contributing to increase the high temperature strength compared to previous alloys of the W-Cr-Ti system. For the alloy containing 1% Y there is not enough oxygen to form  $Y_2O_3$ , resulting in coarsening of the Y-O-rich nanoparticles and thus, to a less efficient inhibition of grain growth.

The thermal conductivity of the alloys is about 2.5 times higher than Inconel 718 at 600-900°, resulting in a temperature increase < 100 K for the conditions expected at the PFMs of the DEMO blanket.

The W-10Cr-1Y alloy exhibit the lowest oxidation rates of all investigated alloys both under isothermal and accident-like conditions. The  $Y_2W_3O_{12}$  phase formed at the surface ad inside the  $Cr_2O_3$  scales seems to contribute substantially to the oxidation reduction of these alloys by improving and stabilizing the self-passivating effect of the  $Cr_2O_3$  scale.

A final HT at 1550°C improves slightly the behavior of the W-10Cr-0.5Y alloy under oxidation and significantly under thermal shock conditions, where this alloy exhibits comparable or better performance than a pure W reference material.

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