

 Open access • Journal Article • DOI:10.1002/PREP.19970220408

Size effect and detonation front curvature — [Source link](#)

P. Clark Souers

Institutions: Lawrence Livermore National Laboratory

Published on: 01 Sep 1997 - Propellants, Explosives, Pyrotechnics (WILEY-VCH Verlag GmbH)

Topics: Detonation, Curvature, TATB, Heat flux and Thermal conductivity

Related papers:

- [Experimental study on the nonideal detonation for JB-9014 rate sticks](#)
- [A study of hydrogen azide detonation with heat transfer at the wall](#)
- [Detonation propagation characteristics for CH₄-2H₂-3O₂ mixtures in a tube filled with orifice plates](#)
- [Detonation velocity deficits of H₂/O₂/Ar mixture in round tube and annular channels](#)
- [Spinning detonation and velocity deficit in small diameter tubes](#)

Share this paper:    

View more about this paper here: <https://typeset.io/papers/size-effect-and-detonation-front-curvature-1fn35yyihm>

231581

UCRL-JC-123194 Rev 3
PREPRINT

Size Effect and Detonation Front Curvature

P.C. Souers

This paper was prepared for submittal to the
1997 Topical Conference on Shock Compression of Condensed Matter
Amherst, MA
July 27 - August 1, 1997

July 1997



Lawrence
Livermore
National
Laboratory

This is a preprint of a paper intended for publication in a journal or proceedings. Since changes may be made before publication, this preprint is made available with the understanding that it will not be cited or reproduced without the permission of the author.

P. C. Souers

Energetic Materials Center, Lawrence Livermore National Laboratory, Livermore, California 94550 (USA)

Heat flow in a cylinder with internal heating is used as a basis for deriving a simple theory of detonation front curvature, leading to the prediction of quadratic curve shapes. A thermal conductivity of 50 MW/mm² is found for TATB samples.

We first consider the size effect for CHNO explosives, where the detonation velocity declines with decreasing radius. If energy is lost out the side of the cylinder, we have:¹

$$\left(\frac{\langle E_o \rangle}{E_o} \right)^{1/2} = \frac{U_s}{D} = 1 - \frac{\langle x_e \rangle}{\sigma R_o} \quad (1)$$

where E_o is the total energy of detonation, U_s and D are the detonation velocities in cylinders of radius R_o and infinite size, and $\langle x_e \rangle$ is the sonic reaction zone length. Also, σ is a wall expansion function, empirically set for unconfined samples to

$$\sigma = 11 \exp \left(-8 \frac{\langle x_e \rangle}{R_o} \right) + 2 \quad (2)$$

As the reaction zone increases, the skin layer increases exponentially to a limiting value. Eq. 2 was created empirically and is designed to enlarge the reaction zone length as the radius increases.

We now consider the detonation front, using the mathematics for uniform heat flow in a cylinder with an internal heat source.² We replace

temperature with the detonation front lag, L , as the cause of the energy flow from the cylinder center to the edge. The thermal conductivity becomes a heat flow constant, K , with the units W/mm². If R is the radius, we have

$$\frac{1}{R} \frac{\partial}{\partial R} \left(R \frac{\partial L}{\partial R} \right) = \frac{A_o}{K}, \quad (3)$$

where A_o is the energy lost per unit volume out the side of the cylinder, because this is replaced from farther inside the cylinder. The energy must be divided by the time to cross the reaction zone ($\langle x_e \rangle / U_s$) to get power. We obtain the lost energy from Eq. 1:

$$A_o \approx \frac{2U_s E_o}{\sigma R_o} \quad (4)$$

We integrate Eq. 3 and substitute Eq. 4 to get

$$L = \kappa R^2 = \frac{A_o R^2}{4K} \approx \frac{U_s E_o}{2K \sigma R_o} R^2 \quad (5)$$

The lag is quadratic with radius, a result that fits fairly well for most explosives, as seen for Forbes' PBXN-111 shots in Figure 1.³ The constant κ in Eq. 5 is the curvature and $1/\kappa$ is the radius of curvature. The quadratic relation has a 15-20% deviation at the edge. Table 1³⁻⁹ lists some of the limited data. We see that κ is not constant for a given group of the same explosive but increases with radius. At the cylinder edge:

$$L_o = \frac{U_s E_o}{2K\sigma} R_o \quad (6)$$

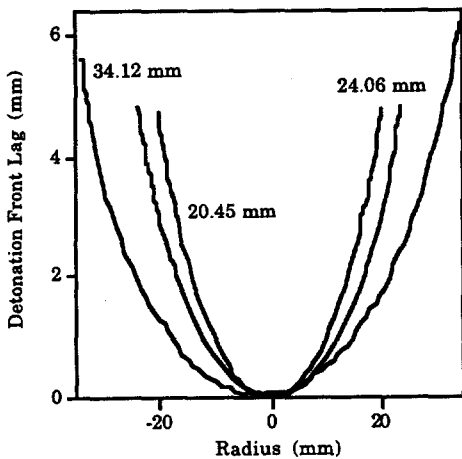


FIGURE 1. Quadratic detonation front shapes for PBXN-111 at three radii.

The thermal conductivity may be found from

$$K = \frac{U_s E_o}{2\kappa\sigma R_o} \quad (7)$$

The best data in Table 1 is for the TATB explosives where $K \approx 50 \text{ MW/mm}^2$. We also see that

$$\langle x_e \rangle \approx L_o \quad (8)$$

The explosive uses the lag to move energy to the edge but needs to keep the front as short as possible so that Eq. 8 seems reasonable in this regard. In Eq. 6, $U_s E_o/K \approx 1$ and $R_o/2\sigma \approx \langle x_e \rangle$, so that Eq. 8 is a coincidence from our few examples.

From Eqs. 1, 5 and 8, we obtain

$$U_s = D - \left(\frac{R_o^2}{\sigma} \right) \kappa \quad (9)$$

This is the starting equation in Detonation Shock Dynamics and Whithams Shock Dynamics, where we see some of the structure of the constant that goes with the curvature.^{10,11}

Table 1 and the above theory applies to most explosive data, type 1, where voids are present to create hot spots and the reaction zone is longer than the void size. The type 2 curvature is the 1.74 g/cc PETN curve from our laboratory shown in Figure 2.

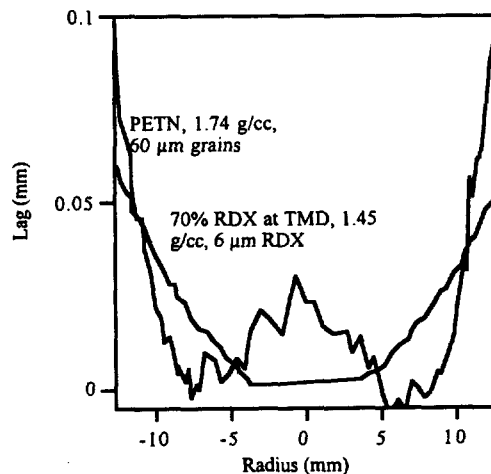


FIGURE 2. Two different types of detonation front curvatures where the edge lag is only 0.1 mm. The PETN is not quadratic in shape: the RDX paste is quadratic but does not follow the theory in the text.

The edge lag is only 0.1 mm and the curvature is rough and not quadratic. Here, we believe the intergrain voids are of the size of the reaction zone so that energy flow to the edges is scattered, producing a ragged front.

The rare type Ia quadratic curvature of the fine-grained 70% RDX explosive, which has had all voids pressed out that no hot spots occur, is shown in Figure 2.^{12,13} Although the edge lag is 0.1 mm, the size effect of Eq. 6 predicts a reaction zone of 1.0 mm. This discrepancy is caused by the added difficulty of getting the binder-enclosed grains to burn. This does not appear in pure liquids (like NM) because the liquid is continuous. The RDX shape is quadratic because the 6 μm grains are smaller than the reaction zone.

ACKNOWLEDGEMENTS

We wish to thank John Bdzil, Jerry Forbes, and Damian Swift for their kind assistance in supplying curvature data from their laboratories. This work was performed under the auspices of the US Department of Energy by the Lawrence Livermore National Laboratory under contract number W-7405-ENG-48.

REFERENCES

1. P. C. Souers, "Size Effect and Detonation Front Curvature," *Propellants, Explosives, Pyrotechnics*, to be published.
2. H. S. Carslaw and J. C. Jaeger, *Conduction of Heat in Solids*, 2nd ed. (Clarendon Press, Oxford, 1959), pp. 130-132, 191, 232.
3. J. W. Forbes, E. R. Lemar, G. T. Sutherland and R. N. Baker, Detonation Wave Curvature, *Corner Turning and Unreacted Hugoniot of PBXN-111*, Naval Surface Warfare Center Report NSWCDD/TR-92/164, Silver Spring, MD, 1992.
4. LLNL Cylinder Test.
5. Los Alamos National Laboratory, Los Alamos, NM, unpublished data, courtesy John Bdzil, private communication, 1996.
6. D. Swift, AWE, Aldermaston, Reading, Great Britain, private communication, 1996.
7. F. Chaisse' and J. N. Oeconomos, "The Shape Analysis of a Steady Detonation Front in Right Circular Cylinders of High Density Explosive. Some Theoretical and Numerical Aspects," *Proceedings Tenth Symposium (International) on Detonation, Boston, MA, July 12-16, 1993*, pp. 50-57.
8. J. B. Bdzil, *J. Fluid Mech.* **108**, 195-226 (1981).
9. R. Engleke and J. B. Bdzil, *Phys. Fluids* **26**, 1210 (1983).
10. D. S. Stewart and J. D. Bdzil, "Examples of Detonation Shock Dynamics for Detonation Wave Spread Applications," *Proceedings Ninth Symposium (International) on Detonation, Portland, OR, August 28-September 1, 1989*, pp. 773-783.
11. B. D. Lambourn and D. C. Swift, "Applications of Whitham's Shock Dynamics Theory to the Propagation of Divergent Detonation Waves," ref. 10, pp. 784-797.
12. H. Moulard, J. W. Kury and A. Delclos, "The Effect of RDX Particle Size on the Shock Sensitivity of Cast PBX Formulations," *Proceedings Eighth Symposium (International) on Detonation, Albuquerque, NM, July 15-19, 1985*, pp. 902-913.
13. H. Moulard, "Particular Aspect of the Explosive Particle Size Effect on Shock Sensitivity of Cast PBX Formulations," ref. 10, pp. 18-24.

TABLE 1. Summary of detonation front curvature and size effect data for various explosives in cylinders. "U" means unconfined; "C" is metal-confined.

Explosive	ρ_0 (g/cc)	R_0 (mm)	L_0 (mm)	$\langle x_e \rangle$ (mm)	κ (mm ⁻¹)	U_s (mm/ μ s)	D (mm/ μ s)	K (MW/ mm ²)		ref
PBX-9502	1.89	4.99	0.77	0.9	0.02400	7.46	7.78	54	U	5
EDC-35	1.90	5.00		0.9	0.02830	7.44	7.73	45	U	6
EDC-35	1.90	5.00		0.9	0.02880	7.44	7.73	45	U	6
EDC-35	1.90	5.00		0.9	0.03070	7.44	7.73	42	U	6
PBX-9502	1.89	5.00	0.74	0.9	0.02370	7.46	7.78	54	U	5
PBX-9502	1.89	6.00	0.78	1.0	0.01705	7.50	7.78	60	U	5
PBX-9502	1.89	8.98	1.03	1.4	0.00995	7.55	7.78	66	U	5
PBX-9502	1.89	24.99	2.18	2.5	0.00276	7.67	7.78	64	U	5
T2	1.86	25.00	1.92	2.5	0.00271	7.62	7.65	65	U	7
PBX-9502	1.89	25.01	2.08	2.5	0.00283	7.68	7.78	62	U	5
EDC-35	1.90	25.40		2.5	0.00261	7.67	7.73	66	U	6
EDC-35	1.90	25.40		2.5	0.00267	7.67	7.73	64	U	6
EDC-35	1.90	25.40		2.5	0.00284	7.67	7.73	60	U	6
LX-17	1.91	25.40	2.07	3.1	0.00293	7.63	7.72	39	C	4
T2	1.86	50.00	2.92	3.0	0.00107	7.63	7.65	65	U	7
NM	1.12	6.35	0.21	0.3	0.00490	6.20	6.21	33	C	8
NM	1.12	9.57	0.86	0.3	0.00800	6.21	6.24	22	U	9
NM	1.12	13.78	0.80	0.2	0.00330	6.23	6.24	33	U	9
NM	1.12	18.42	0.85	0.1	0.00186	6.23	6.24	42	U	9
NM-guar	1.17	5.26	0.98	1.2	0.03350	5.80	6.55	24	U	9
NM-guar	1.17	6.80	0.95	0.9	0.01810	6.09	6.55	23	U	9
NM-guar	1.17	9.57	0.99	1.1	0.00866	6.13	6.55	32	U	9
NM-guar	1.17	18.59	1.13	1.9	0.00215	6.14	6.55	62	U	9
PBXN-111	1.79	20.45	4.81	6	0.01120	5.15	5.81	7	U	3
PBXN-111	1.79	20.52	4.29	6	0.00981	5.16	5.81	8	U	3
PBXN-111	1.79	24.01	4.75	6	0.00870	5.31	5.81	7	U	3
PBXN-111	1.79	24.06	4.87	6	0.00797	5.31	5.81	8	U	3
PBXN-111	1.79	34.12	5.74	7	0.00457	5.57	5.81	9	U	3

Technical Information Department • Lawrence Livermore National Laboratory
University of California • Livermore, California 94551

