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	SOME MEASUREMENTS AND POWER SPECTRA
	OF RUNWAY ROUGHNESS
	By James H. Walls, John C. Houbolt, and Harry Press
	Langley Aeronautical Laboratory Langley Field, Va.
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## TECHNICAL NOTE 3305

# SOME MEASUREMENTS AND POWER SPECTRA

# OF RUNWAY ROUGHNESS

By James H. Walls, John C. Houbolt, and Harry Press

### SUMMARY

Measurements of actual runway roughness obtained by a profile-survey method (engineer's level) are presented. Data were obtained from a survey of a relatively rough runway and a smooth runway. The results of this study are presented as roughness profiles of the runways surveyed and in the form of power spectra.

#### INTRODUCTION

The frequency of occurrence of large load applications in routine ground airplane operations has caused a growing concern in regard to the roughness of landing and taxiing surfaces. In order to obtain information on this problem, it was thought desirable to make detailed measurements of the roughness characteristics of actual runway surfaces. As one of the initial steps in this study, measurements were made of two runways available to the National Advisory Committee for Aeronautics at Langley Field, Va. The two runways selected were known to be of very different degrees of roughness; one runway was considered relatively smooth whereas the other was considered rather rough - possibly rough enough to preclude active use. The measurements made are presented here directly as elevation profiles. In addition, the power spectra of the runway elevations were also determined and are presented in order to permit a description of the frequency characteristics of the runway roughness.

#### SYMBOLS

D distance over which moving average is taken

- L wavelength, ft
- m number of uniformly spaced points over the frequency range at which power estimates are desired

**1**S

n number of equally spaced elevations taken over the runway

x distance, ft

2

X arbitrary value of x, ft

 $\Delta x$  space interval, ft

y(x) random function of distance (runway height)

- R(X) autocorrelation function
- $\Phi(\Omega)$  power-spectral density function, defined by equation (1),  $\frac{\text{ft}^2}{\text{radian/ft}}$
- X displacement distance, ft

 $\Omega$  reduced frequency,  $2\pi/L$ , radians/ft

 $\sigma$  root-mean-square value of y;  $\sqrt{y^2(x)}$ 

#### SURVEY OF VARIATIONS IN RUNWAY HEIGHT

A diagram of the landing and parking strips that presently exist at Langley Field is presented in figure 1 and shows the extent of the two surveys made. Runway 17-35 was chosen because it is considered representative of a satisfactorily smooth runway; the other runway chosen, 12-30, is considered rough and is used only for parking. Both runways are of standard concrete construction.

The roughness measurements were made by means of a surveyor's level, rod, and tape. This means was selected because it could be applied directly without the delays attendant on the development of special instrumentation. In using this technique it was necessary to select an interval at which elevations would be obtained. It was thought that the frequency range between 0.5 and 35 cycles per second would be the region of principal concern for most airplanes. At a landing speed of 100 miles per hour this would correspond to wavelengths between about 300 and 4 feet. As a consequence, a reading interval of 2 feet was selected. This choice was dictated by two considerations: first, it was expected that there would be little variation in runway height at wavelengths less than 4 feet and, second, the communications sampling theorem (ref. 1), which states that sampling a disturbance at intervals of one-half the shortest wavelength present completely specifies the disturbance. Only 1,400 feet at the south end of runway 35 was surveyed so as not to interfere with ground traffic from runway 7. Visual observation suggested that except for the initial 100 feet, which is a sloping macadam overrun, this 1,400 feet is fairly representative of the remainder of the runway. Three thousand feet or nearly all of runway 12 was covered. The two surveys were conducted with greater ease than was originally expected. Approximately 6 hours was spent on runway 35 and slightly more than twice that time was spent on runway 12. This means that on an average about 115 readings per hour were obtained. The pace was steady but not hurried. Detailed runway elevations are given in table I, and figure 2 shows the runway profiles. These elevations are plotted about a zero arithmetic mean.

## POWER SPECTRA OF RUNWAY HEIGHT

#### Definition of the Power Spectrum

In addition to the actual runway-height profiles, it appeared desirable, because of the random character of the height fluctuations, to determine the power spectra of runway height. These power spectra would provide a description of the frequency content of the runway height variations and be directly applicable to the calculation of airplane responses in the frequency plane.

Since the runway roughness under consideration here is a space disturbance rather than a disturbance in time, it is desirable to define the power-spectral density function in terms of the frequency argument  $\Omega$ in radians per foot rather than the conventional argument  $\omega$  in radians per second. In terms of this frequency argument, the power-spectral density function of the disturbance y(x) is defined in the following manner:

$$\left| \begin{array}{c} \phi(\Omega) = \lim_{X \longrightarrow \infty} \frac{1}{2\pi X} \left| \int_{-X}^{X} y(x) e^{-i\Omega x} dx \right|^2 \end{array} \right|$$
(1)

where the bars indicate the modulus of the complex quantity. Equation (1) may be used to evaluate the power-spectral density function from observed data, but, in practice, the power-spectral density function may be determined more conveniently and less tediously through use of a related function, the autocorrelation function R(X), defined by

$$R(X) = \lim_{X \longrightarrow \infty} \frac{1}{2X} \int_{-X}^{X} y(x)y(x + X)dx$$
(2)

The autocorrelation function has the symmetrical property R(X) = R(-X)and is reciprocally related to the power-spectral density function by the Fourier cosine transformation in the following manner:

$$\Phi(\Omega) = \frac{2}{\pi} \int_{0}^{\infty} R(X) \cos \Omega X \, dX$$

$$R(X) = \int_{0}^{\infty} \Phi(\Omega) \cos \Omega X \, d\Omega$$
(3)

From the foregoing it is seen that

$$R(0) = \overline{y^2(x)} = \sigma^2 = \int_0^\infty \Phi(\Omega) d\Omega \qquad (4)$$

where  $\sigma$  is the root-mean-square value (or standard deviation) of the disturbance and is a convenient measure for a comparison of the overall roughness of the two runways surveyed.

### Evaluation of Power Spectra

The actual evaluations of the power spectra were made by means of the numerical-calculation procedure derived by Tukey in reference 2. This procedure is also described and discussed in reference 3.

As a preliminary to these power-spectrum calculations, it appeared desirable to make some modifications to the actual measured profiles. Examination of these profiles (fig. 2) indicated that the runway height exhibited large changes in elevation at very long wavelengths. These large changes at low frequencies have a tendency to complicate and distort power estimates at the higher frequencies because of the effective filter characteristics of the numerical estimators. Since, in the present study, there was little interest in the longer wavelengths, it was decided to avoid these adverse affects by removing the longer wavelengths. One simple and convenient way of prefiltering a disturbance is by the use of moving averages. This means was used in the present study to filter out some of the longer wavelengths. A moving average of the runway height for a 300-foot distance, as defined by

$$y_{m}(x) = \frac{1}{150} \sum_{K=-75}^{K=75} y(x + K \Delta x)$$

was determined for each of the runways and is shown as the dashed curve in figure 2. The variations of runway height about this moving average were then determined and are shown in figure 3. As can be seen roughly from a comparison of figures 2 and 3, the main effect of this operation is the removal of the low-frequency components of the height variations.

The actual filtering effects of this operation on the spectrum estimates are derived in detail in the appendix. The attenuation function for the spectrum introduced by the moving average is given by

$$\left(1 - \frac{\sin \frac{\Omega D}{2}}{\frac{\Omega D}{2}}\right)^2$$

where D is the distance over which the average is taken. The attenuation function is shown as a function of  $\Omega D/2$  in figure 4. A separate scale for wavelength L is also shown for the present value of D = 300 feet. It can be seen from the figure that the principal effect of the moving average in the present case is to attenuate the effects of the longer wavelengths, with the attenuation factor decreasing from 1 at 300 feet to 0.490 at 400 feet and 0.132 at 600 feet.

The actual steps involved in the numerical estimation of the spectra (ref. 2) are as follows:

1. The autocorrelation coefficients were determined from the successive values of elevation  $y_1, y_2, \ldots, y_n$  according to the following numerical form of equation (2):

$$R_{r} = \frac{1}{n-r} \sum_{q=1}^{n-r} y_{q} y_{q+r} \qquad (r = 0, 1, 2, ... m) \qquad (5)$$

where  $R_r = R(\chi)$ ;  $\chi = r \Delta x$ .

2. Initial or "raw" estimates of the power density were then determined by use of the following numerical form of equation (3):

$$L_{r} = \frac{\Delta x}{\pi} \left( R_{0} + 2 \sum_{q=1}^{m-1} R_{q} \cos \frac{qr\pi}{m} + R_{m} \cos r\pi \right)$$
(6)

where  $L_r = L(\Omega)$ ;  $\Omega = \frac{r\pi}{m \Delta x}$ . These estimates have an effective filter which has the undesirable character of appreciable side-band areas and thus permits a wide diffusion of power.

3. Final or "smoothed" estimates of power density, which are estimates based on a more desirable and sharper filter, were then determined from

$$\Phi_{0} = 0.54L_{0} + 0.46L_{1}$$

$$\Phi_{r} = 0.23L_{r-1} + 0.54L_{r} + 0.23L_{r+1}$$

$$\Phi_{m} = 0.46L_{m-1} + 0.54L_{m}$$
(7)

The values of  $\Phi_r$  obtained in the foregoing manner can roughly be considered to be estimates of the average power over the frequency interval

 $\frac{(r-1)\pi}{m\Delta x} < \Omega < \frac{(r+1)\pi}{m\Delta x}$ . The distances of 1,400 and 3,000 feet covered in the survey and the interval  $\Delta x$  of 2 feet lead to values of n of 701 and 1,501, respectively, for the two runways. Evaluation was made with m = 40. The autocorrelation coefficients obtained are shown in figure 5 and the resulting power estimates are plotted as a function of

the reduced frequency  $\Omega$  in figure 6. Each point in the present case represents the average power in a frequency interval  $\pm \frac{\pi}{80}$  about the

value plotted. For clarity in presentation, some of the power estimates at the higher frequency are not shown but were used in obtaining the faired curves. A scale of wavelength L is also shown in figure 6. The root mean square of runway height  $\sigma$ , which is a convenient measure of the average roughness power, is listed in figure 6 for each of the runways in the figure.

It may be recalled that the spectra of figure 6 represent the spectra of the runway height variations about the moving average. The effects of the moving-average operation on the derived spectra have been shown in the appendix to be equivalent to multiplying the actual spectra by the attenuation function  $\left(1 - \frac{\sin \frac{\Omega D}{2}}{\frac{\Omega D}{2}}\right)^2$ . In principle, then, it would appear

that the effects of the moving-average operation could be removed by dividing the spectra of figure 6 by this attenuation function. However, in practice this cannot be done explicitly, since the spectrum estimates are averages of the power over finite band widths. A good approximation of the effect may be obtained, however, by dividing the faired spectrum estimates of figure 6 by the attenuation function and then averaging the resultant spectra over each of the band widths. It may be seen that these operations would have a negligible effect for the range of frequencies covered in figure 6.

#### DISCUSSION

Examination of figures 2 and 3 indicates that the runway heights fluctuate in a random manner, with variations of as much as several inches for each runway. The profile for runway 12 on both figures shows wider fluctuations, particularly at the longer wavelengths. The overall variations in height at the wavelengths of principal interest are best compared in figure 3, where the effects of the very long wavelengths (greater than 300 feet) have been largely removed by the moving average. It is clear from this figure that runway 12 is appreciably rougher than runway 35, as was to be expected. The height profiles of figures 2 and 3, representing what may be considered as a satisfactory and an unsatisfactory runway, provide representative runway inputs for response calculations.

A more detailed comparison of the characteristics of the runway height variations is possible from the spectra of figure 6. The overall height fluctuations as represented by the root-mean-square values of 0.057 feet for runway 12 and 0.021 feet for runway 35 indicate that runway 12 is almost three times as rough as runway 35. By comparing the spectra it can be seen that runway 12 has 10 times the power of runway 35 at the longer wavelengths (300 to 500 feet) and about twice the power at the shorter wavelengths (below 50 feet). The rapid decrease of power with increasing frequency displayed by both spectra is perhaps generally typical of runway height spectra and provides a guide to a representative spectrum shape. Inasmuch as the spectra of figure 6 represent satisfactory and unsatisfactory levels of runway roughness, the heights of the spectra provide an initial guide toward the establishment of criteria for runway roughness.

# CONCLUDING REMARKS

As an initial step in the study of runway roughness, data were obtained from a survey of a relatively smooth runway and a runway which is considered rough. A surveyor's level, rod, and tape were used to obtain these data. In general, this method was found to be quick and relatively inexpensive. The results obtained are presented as profiles of runway height. The power spectra of runway height were also determined and are presented. These results provide an initial guide toward establishment of criteria for runway roughnesses and are suitable for airplane response calculations.

Langley Aeronautical Laboratory,

National Advisory Committee for Aeronautics, Langley Field, Va., August 13, 1954. NACA TN 3305

### APPENDIX

# EFFECT OF MOVING AVERAGES ON THE POWER SPECTRUM OF A RANDOM FUNCTION

Consider a random function f(x). The centered moving average of f(x) over the interval D is defined by

$$g(x) = \frac{1}{D} \int_{x-\frac{D}{2}}^{x+\frac{D}{2}} f(x_1) dx_1$$
 (A1)

The deviation of the disturbance about its moving average is then given by

$$h(x) = f(x) - g(x) = f(x) - \frac{1}{D} \int_{x-\frac{D}{2}}^{x+\frac{D}{2}} f(x_1) dx_1$$
(A2)

The spectral properties of h(x) as compared with those of f(x) are of concern; specifically, the effects on the spectrum of f(x) introduced by the operation of equation (A2).

The "finite" Fourier transform of equation (A2) may be written

$$H(\Omega) = \int_{-X}^{X} h(x)e^{-i\Omega x}dx = \int_{-X}^{X} f(x)e^{-i\Omega x}dx - \frac{1}{D} \int_{-X}^{X} \int_{x-\frac{D}{2}}^{x+\frac{D}{2}} f(\eta)e^{-i\Omega x}d\eta dx$$
(A3)

If the order of integration of the double integral term is interchanged, there results

$$H(\Omega) = \int_{-X}^{X} f(x) e^{-i\Omega x} dx - \frac{1}{D} \int_{\eta - \frac{D}{2}}^{\eta + \frac{D}{2}} \int_{-X}^{X} f(\eta) e^{-i\Omega x} d\eta dx \quad (A4)$$

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The x-integral of the second term may now be evaluated to give

$$H(\Omega) = \int_{-X}^{X} f(x) e^{-i\Omega x} dx - \frac{\sin \frac{\Omega D}{2}}{\frac{\Omega D}{2}} \int_{-X}^{X} f(\eta) e^{-i\Omega \eta} d\eta = \left(1 - \frac{\sin \frac{\Omega D}{2}}{\frac{\Omega D}{2}}\right) F(\Omega)$$
(A5)

where  $F(\Omega)$  is the "finite" Fourier transform of f(x)

$$F(\Omega) = \int_{-X}^{X} f(x) e^{-i\Omega x} dx \qquad (A6)$$

The power spectrum of h(x) follows directly from equation (A5) in accordance with the following equation:

$$\Phi_{h}(\Omega) = \lim_{X \longrightarrow \infty} \frac{1}{2\pi X} |H(\Omega)|^{2} = \lim_{X \longrightarrow \infty} \frac{1}{2\pi X} H(\Omega) H(-\Omega)$$
(A7)

Substitution of equation (A5) into this equation yields the following simple result:

$$\Phi_{\rm h}(\Omega) = \left(1 - \frac{\sin\frac{\Omega D}{2}}{\frac{\Omega D}{2}}\right)^2 \lim_{X \longrightarrow \infty} \frac{1}{2\pi X} F(\Omega)F(-\Omega) = \left(1 - \frac{\sin\frac{\Omega D}{2}}{\frac{\Omega D}{2}}\right)^2 \Phi_{\rm f}(\Omega)$$
(A8)

The squared term on the right side of this equation is the effective spectrum attenuation function or transfer function introduced by the moving-average process defined by equation (A2). It is shown plotted as a function of  $\Omega D/2$  in figure 4. Also shown in the figure is a scale of wavelength for D = 300 feet, the value used in the present study.

As a matter of added interest, an alternate derivation of equation (A8) may be given which is slightly more involved but which is more general in the sense that the function g(x) does not have to be functionally related to f(x). Equation (A1) may also be written

$$g(x) = \frac{1}{D} \int_{-\infty}^{\infty} f(x_1)k(x - x_1)dx_1 \qquad (A9)$$

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where

• 12.1

$$k(x) = 1$$
  $(-D/2 \le x \le D/2)$ 

$$k(x) = 0 \qquad (elsewhere)$$

Now, from equation (A2) the power spectra of h(x) and g(x) can be shown to be related in the following manner (ref. 4):

$$\Phi_{h}(\Omega) = \Phi_{f}(\Omega) + \Phi_{g}(\Omega) - \Phi_{fg}(\Omega) - \Phi_{gf}(\Omega)$$
(Alo)

where  $\Phi_{f}(\Omega)$  and  $\Phi_{g}(\Omega)$  are the power spectra of f(x) and g(x), respectively, and  $\Phi_{fg}(\Omega)$  and  $\Phi_{gf}(\Omega)$  are the cross spectra of f(x)and g(x) defined in the following manner:

$$\Phi_{fg}(\Omega) = \lim_{X \longrightarrow \infty} \frac{F(-\Omega)G(\Omega)}{2\pi X}$$

$$\Phi_{gf}(\Omega) = \lim_{X \longrightarrow \infty} \frac{F(\Omega)G(-\Omega)}{2\pi X}$$
(All)

where  $F(\Omega)$  is defined by equation (A6) and

$$G(\Omega) = \int_{-X}^{X} g(x) e^{-i\Omega x} dx \qquad (A12)$$

The cross spectra in general have both real and imaginary terms, in contrast to the simple power spectra which are always real.

In order to determine the spectrum of h(x), it is necessary to evaluate each of the spectra on the right side of equation (AlO). Considering  $\Phi_{g}(\Omega)$  first, it can be seen from equation (A9) that g(x) is related to f(x) by the conventional convolution or Duhamel integral and is thus obtained by a linear operation on f(x). Using the simple relation between spectra of an input disturbance and an output response for linear systems, there is immediately obtained the following relation between g(x) and f(x):

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$$\Phi_{g}(\Omega) = \Phi_{f}(\Omega) \left( \frac{\sin \frac{\Omega D}{2}}{\frac{\Omega D}{2}} \right)^{2}$$
(A13)

where the squared term represents the amplitude of the frequencyresponse function for the operation defined by equation (A9).

The cross-spectra terms may be evaluated in the following manner. From equations (A9), (All), and (Al2), for example,

$$\Phi_{fg}(\Omega) = \lim_{X \to \infty} \frac{F(-\Omega) \int_{-X}^{X} \left[ \frac{1}{D} \int_{-\infty}^{\infty} f(x_1)k(x - x_1)dx_1 \right] e^{-i\Omega x} dx}{2\pi X}$$
(A14)

In the limit, this equation reduces to

$$\Phi_{fg}(\Omega) = \Phi_{f}(\Omega) \frac{\sin \frac{\Omega D}{2}}{\frac{\Omega D}{2}}$$
(A15)

The cross spectrum in this case has no imaginary term reflecting the lack of any introduction of a phase shift by the linear operation defined by equation (A2). In the present case

$$\Phi_{\mathbf{f}g}(\Omega) = \Phi_{g\mathbf{f}}(\Omega) \tag{A16}$$

Substituting the results of equations (A13), (A15), and (A16) into equation (A10) yields the same result as given by equation (A8).

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- 4. James, Hubert M., Nichols, Nathaniel B., and Phillips, Ralph S.: Theory of Servomechanisms. McGraw-Hill Book Co., Inc., 1947.

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	Runway elevation, ft		800 410 1020 1020 1025			120'- 120'- 890'-	420 570 770	470 270 270	870 870 870 770	080 180	060°- 680°- 7 - 160°-	095 101 101 111
	Distance along runway, ft	1,100 1,100 1,100 1,100 1,100	1,1,1,1,1,1,1,1,1,1,1,1,1,1,1,1,1,1,1,	277 777 777 777 777 777 777 777 777 777	11111 11111 11111 11111 11111 11111 1111	1,142 1,144 1,148 1,148	, , , , , , , , , , , , , , , , , , ,	1,168 1,168	1,172 1,174 1,176 1,178	1,1,4,4,4,4,4,4,4,4,4,4,4,4,4,4,4,4,4,4	200 196 198 198 198 198 198 198 198 198 198 198	1,202 1,204 1,206 1,208
ort irned	Runway elevation, ft	0.104 0.108 0.088 0.081 0.021 0.072	50000000000000000000000000000000000000	88888	88886	68888	669 9057 9057 9050	240. 200. 200. 200. 200.	520.0 250.0 850.0 850.0 850.0	720. 720. 720.	200- 210- 210- 210- 210- 210-	-012 -010 -110 -100-
REIGHT DATA - Co attrued	Distance along runway, ft	1, 398 998 998 998 998 998 998 998 998 998	, 1, 1, 000 1, 0	1,1,01,01,00 00,00,00,00,00,00,00,00,00,00,00,00,	0058 0054 0058 0058 0058 0058 0058	, , , , , , , , , , , , , , , , , , ,	100 100 100 100 100 100 100 100 100 100	747544 8888 8688 868	400 7088 7088 7088 707 707 707 707 707 70	н, 072 1, 074 1, 076 1, 036	880 4880 1111111111111111111111111111111	1,092 1,094 1,096 1,098
- BASIC RUNNAY	Runway elevation, ft	भारत स्वान भूम	<i>ត់ដូ</i> ដូដូដ	ม่า่า่อ่อ่	891 211 211 211	<b>गंगंगं</b> विवेद्यां	1111 2019 2019 2019 2019 2019	89999999999999999999999999999999999999	102 102 103 103 103 103 103 103 103 103 103 103		211 201 200 200 200 200 200 200 200 200	.085 .086 .089 .098 .098
TARLE I	Distance along runway, ft	****	82888	88883	8888 8888 8888 8	25 55 55 55 27 75 58 58 56 28 75 58 58 56	88883 88883	¥\$\$\$&	8888 8888 8888	8888£	972 976 978 978 978	988 988 9886 9886 9886 9886
i	Runway elevation, ft	690.0 897.5 898.5 898.5 898.5 898.5 898.5 898.5 898.5	8.5.5.5.8.8 8.19.6.888	88. 860 860 860 860 860 860	88. 879. 879. 888. 899. 899. 899. 899. 8	866. 2001 2001 2001 2001 2001	866 866 867 867 867 867 867 867 867 867	999 999 999 101	511. 811. 521. 821.	721- 130 251- 251-		.106 .097 .006 .001 .102
	Distance along runway, ft	772 774 776 778 80 780	88888 88888 8	82828	88888 9888 19888 19888 19888 1988 1988	814 816 816 828 820	88888 8888 99888 99888 99888 99888 99888 99888 99888 99888 99888 9988 9988 9988 9988 9988 9988 9988 9988 9988 9988 9988 9988 9988 9988 9988 9988 9988 9988 998 998 998 998 998 998 998 998 998 998 998 998 998 999 909 90 90	£ 5,45,85,85 5,45,85,85 5,45 8,55 8,55 8,	8 8 8 8 8 7 1 9 8 8 7 2 8 8 8 9 8 9 8 9 8 9 8 9 8 9 8 9 8 9 8 9	8989895 258989 258989 26989 26989 26989 2699 2699 2699 26	862 8664 870 870	872 874 876 878 880
	Runway elevation, ft	140.0 940. 250.	640 740 740 740 740 740 740 740 740 740 7	0.00 0.00 0.00 0.00 0.00 0.00 0.00 0.0		170. 270. 270.	960 960 960 960 960 960 960	040 045 045 045 045 045 045 045 045 045	050 150 940 940 940	.055 .056 .059 .051 .051	050 047 045 048 049 050	.052 .059 .062 .065
	Distance along runway, ft	665 666 666 70 866 70 866 866 866 866 866 866 866 866 866 86	672 6674 866 674 86 86 80 86 80 80 80 80 80 80 80 80 80 80 80 80 80	888 988 988 988 988 988 988 989 989 989	695 669 700 700 700 700 700 700 700 700 700 70	702 706 7708 7708	2021 116 128 128 128	722 728 728 728 728 728	277777 2777777	33222 22222 2	8777888	22222E

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RUNWAY	(a) (a)
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ft									
Runway elevation,	-0.238	230	612	215	<b>1</b> 1 1	602 -			
Distance along runway, ft	1, 382 1. 384	1,386 1,388	1, 390	1, 392	366	1,398 1,400			
ť									
Rumway elevation,	-0.247	245	250	243		241	542 142	012	239
Distance along runway, ft	1, 352 1. 354	1,356	1,360	1,362 1,362	295 7 - 1 98 7 - 1	1,368	1, 372 1, 374	1, 376	- 200 
Rumway elevation, ft	-0.221	<u>हि</u>	112	-219	627-	सूर सूर -	231	238	142
Distance along runway, ft	1,322		1,330	1, 532		1, 338 1, 340	1,342	1,346	1, 348 1, 350

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	Runway elevation, ft	-0.487 -0.487 480 480 475 475	254 4 5 5 5 7 7 7 7 7 7 7 7 7 7 7 7 7 7 7	1432 1428 1428 1428	- 410 424 423 423 423	807	895 1 1 2 89 89 89 89 89 89 89 89 89 89 89 89 89 8		356 344 328 317	906 		
	Distance along runway, ft	87878 8787 777	¥ <b>\$</b> \$\$	577 577 577 577 577 577 577 577 577 577	ጟቒጟቜቜዿ	87877	88883 89883	615 616 618 618 628 628	6.66 6.66 6.66 6.66 6.66 6.66 6.66 6.6	99999 99999	33338S	848888 86888
	Runnay elevation, ft		885. 885. 175. 885.	855 1855 1888 1888 1988 1988 1988 1988 1	88845 , , , , , , ,	377 372 374 368 368	574 572 570 579 390	. 621 201 201 201	0994 9944 1944	1462 1462 1494 1494		528 522 522
	Distance along runway, ft	244 244 244 244 244 244 244 244 244 244	222888 222888	4988645 4986655	4774 176 176 178 178 178 178 178 178 178 178 178 178	8787 8787 8787 8787 8797 8797 8797 8797	\$\$\$\$ <b>\$</b> \$	8888 8888 8888 8888 8888 8888 8888 8888 8888	271 271 271 271 271 271 271 271 271 271	255 255 255 255 255 255 255 255 255 255	877.282 877.282	3.555 2.5555 2.5555 2.5555 2.5555 2.5555 2.5555 2.5555 2.5555 2.5555 2.5555 2.5555 2.5555 2.5555 2.5555 2.55555 2.5555 2.5555 2.5555 2.5555 2.5555 2.55555 2.55555 2.55555 2.55555 2.55555 2.55555 2.55555 2.55555 2.55555 2.5
Contrinued	Runway elevation, ft	-0.210 -0.220 238 238 241 241		- 235 - 232 - 235 - 240		238 254 254 274 274		555 555 555 555 555 555 555 555 555 55	22.5.5.5.5 22.5.5.5.5 23.5.5.5.5 23.5.5.5.5 23.5.5.5.5 24.5.5.5 25.5.5.5.5 25.5.5.5.5 25.5.5.5.5.	106 216 215 215 215 215	125 125 125 125 125 125 125 125 125 125	2027 2027 2027 2027 2027 2027 2027 2027
N REIGHT DATA -	Distance along runway, ft	828883 828883	¥\$\$\$8	<u>K</u> &&&&	፝ጞ፝፞፞፞ <u>ቖ</u> ፞፞፞፞፞ቖ፞፞፞፞ቘ	777 777 777 777 777 777 777 777 777 77	**	ኟቘኟቘቜ	22222	고극고 문문 문문 문 문 문 문	ቔቘቘቘቒ	ያጃጃጀ <del>3</del>
L BASIC RUNW	Runway elevation, ft	850.0- 1047 - 1041 - 1050 - 1050 -		108 105 115 118	811 011 121 121	041 241 241	841 191 841	-178 -127 -127 205 205	- 218 - 226 - 226	22 22 22 22 22 22 22 22 22 22 22 22 22	122 8122 6021 6021	- 1081 
TARE	Distance along runway, ft	50 556 558 558 558 558 558 558 558 558 558	8888 8888 8888 8888 8888 8888 8888 8888 8888	2440 5589 55555	X7XXX	38888 8888	272 276 278 278 278 280 280	88 88 88 88 88 88 88 88 88 88 88 88 88	8, 8, 9, 8, 8, 8, 8, 8, 8, 8, 8, 8, 8, 8, 8, 8,	200 200 200 200 200 200 200 200 200 200	24 4 8 8 8 8 8 8 8 8 8 8 8 8 8 8 8 8 8 8	<u> አ</u> ቋቋቋጅ
	Runway elevation, ft	400. 200. 200. 200. 200.	840. 0.09. 140. 140.	9,6,6,6,6,6,	110. 1410. 250. 240. 240.	89999	010, 410, 000, 10,000, 10,000, 10,000, 10,000, 10,000, 10,000, 10,000, 10,00,000, 10,00,000, 10,00,000, 10,00,000, 10,00,000, 10,00,000,0	700 200 200 700	018 020 030 028 028			
	Distance along runway, ft	211 111 121 121 121 121 121 121 121 121	27 47 X2 82 X1	ያጃጽጽਤ	212211	84889 84889	233 233 233 233 233 233 233 233 233 233	172 174 176 178 178	89 89 89 89 89 89 89 80 80 80 80 80 80 80 80 80 80 80 80 80	265 266 266 266 266 266 266 266 266 266	200 200 200 200 200 210 200 210 200 210 200 20	212 214 218 218 218 228
	Runway elevation, ft	177. 177. 177. 177. 177. 177. 177. 177.	& & ਟੋਟੋ ਨੋਟੋਟੋਟੋਟੋਟੋਟੋਟੋਟੋਟੋਟੋਟੋਟੋਟੋਟੋਟੋਟੋਟ	2 1 2 2 2 2 2 2 2 2 2 2 2 2 2 2 2 2 2 2	125 1991 1995 171	431. 833. 834.4	941 941 941 941 941 941 941 941 941 941	11111 848.841	22222	સંસંગ્લુ	401. 101. 102. 1080. 009.	580. 1480. 1680. 1770.
	Distance along runway, ft	ဝလာက္ကစ	24258	<u>አ</u> ቋጽይ	ይዳጽጽያ	33338	X7X88	8488F	64668	හිදු.සි.සි හි	83888	2010 2010 2010 2010 2010 2010 2010 2010

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	Runway elevation, ft	0.061 .055 .055 .072 .072	-005 -015 -015 -016 -016	750. 1200. 0200. 0200.	240. 240. 1220.	900 10 10 10 10 10 10 10		600 100 100 100 100 100 100 100 100 100	016 022 022 022 022 022 022 022 022	200 200 200 200 200 200 200 200 200 200	80. 280 280 280 280 280 280 280	.078 .087 .115 .130 .341
	Distance along runway, ft	1,212 1,214 1,216 1,220	1,222 1,224 1,226 1,228 1,230	1,232 1,234 1,238 1,240	1,244 2442 1,244 2,246 1,244 2,246 1,246	1,258 255 1,256 1,256	1,262 1,266 1,266	1,272 1,274 1,276 1,278	288 1988 1988 1988 1988 1988 1988	24528888 2688888 1111111	1,302 1,304 1,306 1,308 1,10 1,10 1,10 1,00 1,10 1,10 1,10 1,	1, 312 1, 316 1, 316 1, 328
	Runway elevation, ft	801.0 2006 212 215 222 222	<i>ई हे हे हे हे</i>	ភិដុងរំងំតំ	8511 8821 8831 8831	889979	172 177 168 166	991 521 531 531 531 531 531 531 531 531 531 53	1.1.1.1.1. 8.1.1.1.1 8.1.2.1.8	8211 2010 2000 2000	.078 .072 .075 .075 .070	-072 -050 -050 -050 -050
	Distance along runway, ft	201 201 1001 1001 1001 1001 1001 1001 1	2114 9114 9214 8214	1,128 1,128	ትትትት የአያያያ	444444 1444 1444 1588 1588 1588 1544 1544	34888 11111 11111	1,168 164 168 168 168 168	1,172 1,174 1,176 1,178	1,182 1,182 1,186 1,188 1,188 1,198	1,192 1,194 1,196 1,198	1,202 1,204 1,206 1,208 1,208
	Runway elevation, ft	0.080 470 775 770- 770-	866 866 866 866 866 867 867 867 867 867	\$44 <u>1</u> 4	11111111 1111111	888544 888544	221 221 221	12 24 25 25 25 25 25 25 25 25 25 25 25 25 25	84980 84980	611 611 811 811 811		8.8.8.9.9 8.9.8
	Distance along runway, ft	т 388 888 886 886 886 886 886 886 886 886	1,000 1,000 1,000 1,000 1,000 1,000	1,012 1,012 1,016 1,018 020 1,020	444 008 008 008 008 008 008 008 008 008	200 020 100 10 10 10 10 10 10 10 10 10 10 10 1	1,042 1,044 1,046 1,048 1,050	200 700 700 700 700 70 70 70 70 70 70 70	1,062 1,062 1,066 1,068 0,070	1,072 1,074 1,076 1,078 1,080	1,082 1,084 1,086 1,086 1,098	1,092 1,094 1,096 1,098 1,100
	Rummery elevation, ft	-0 -0 -0 -0 -0 -0 -0 -0 -0 -0 -0 -0 -0 -		588 588 588 588 588 588 588 588 588 588	910. 810. 710. 710. 720.	880. 880. 890.	820 760 920	540. 340. 170. 770.	0.250 0.250 0.250 0.250 0.250 0.250	660 660 660 660	0.05 0.05 0.05 0.05 0.05 0.05 0.05 0.05	880. 1980. 1980. 1980. 170.
	Distance along runway, ft	\$\$ \$\$ \$\$ \$\$ \$ \$ \$ \$ \$ \$ \$ \$ \$ \$ \$ \$ \$	88888888888888888888888888888888888888	88888	2176 200 2176 200 200 200 200 200 200 200 200 200 200	255 258 258 258 258 258 258 258 258 258	88883	2456 2456 2666 2666 2666 2666 2666 2666	266 976 980 980 980	388 888 888 888 888 888 888 888 888 888	22 22 22 22 22 22 22 22 22 22 22 22 22	ZZZZZZZZZZZZZZZZZZZZZZZZZZZZZZZZZZZZZZ
	Runway elevation, ft	0,00,0- 1,078 070 1,078 870	960	100 100 088 082 077	470		640 740 840	046 038 460 200 200	120** 600** 500**	028 150 120 280	023 028 030 022	10 10 10 10 10 10 10 10 10 10 10 10 10 1
ı	Distance along runway, ft	241 116 117 117 117 117 117 117 117 117 11	84888 848888 84888 84888 84888 84888 84888 84888 84888 84888 84888 8488	8 <b>4</b> 88	888889	812 814 818 818 826 828	828 826 928 928 928 928 928 928 928 928 928 928	9-9-9-9-9-9-9-9-9-9-9-9-9-9-9-9-9-9-9-	9 17 17 17 17 17 17 17 17 17 17 17 17 17	869 874 860 860 860 860	88888 88888 888888	8772 8774 8776 878 878 878
	Rumway elevation, ft	-0.275 288 288 284 284 284	88. 88. 81. 81. 81. 81. 81. 81. 81. 81.	8.9.7.7.7.7. 8.7.7.7.7.7.7.7.7.7.7.7.7.7.7.		- 188 - 182 - 182 - 182 - 182	781 691 731 771	-1186 -1181 -1175 -1160	-158 -159 -150 -1160 -1160	148 128 055	201- 201- 201-	27. 27. 27.
-	Distance along runway, ft	388888 388888 2888888	606 678 80 80 80 80 80 80 80 80 80 80 80 80 80	<u>8</u> 8888	8888 8888 8888	202 407 807 108 807 110	778 116 728 728	722 724 726 728	877777 8777282	22222 22222 22222	877888 877888	8498 P

19. 1. - BASIC RUMMY REIGHT DATA - Continued (b) Continued .

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	Rumway elevation,	0.259 .264 .256 .256	253 253 253 253 253 253 253 253 253 253	68, 68, 68, 89, 89, 89, 80, 80, 80, 80, 80, 80, 80, 80, 80, 80	ភូនភូន	861 861 802 912	210 200 200 200 200	261. 181. 181. 181. 181. 181. 181. 181.	1911 1911 1911 1911 1911 1911 1911 191	181. 1801 1901 1901 1901	211 211 211 211	4444 48844
	Distance along runway, ft	1, 872 1, 874 1, 876 1, 878 1, 880	1, 1, 1, 1, 1, 1, 1, 1, 1, 1, 1, 1, 1, 1	4,4,4,4,4,4,4,4,4,4,4,4,4,4,4,4,4,4,4,	1,902 1,904 1,906 1,910	1,912 1,914 1,918 1,918	1,922 1,924 1,928 1,928	4,938 459,1 936,1 938,0 940,1 938	ਕ <b>16</b> 176 16 16 16 16 16 16 16 16 16 16 16 16 16	952 1,958 1,958 1,958	886 1986 1986 1986 10 10 10 10 10 10 10 10 10 10 10 10 10	1,972 1,974 1,976 1,978 1,980
	Runway elevation, ft	555. 555. 545. 555.	265 87 87 87 87 87 87 87 87 87 87 87 87 87	225 2715 7155 1155 1155	¥ġġġţţ	¥ĔŔŔĔŔ	8.	अंस्ट्रे इस्ट्रेड्रे	¥÷÷¥¥	ġĔġġŖ	755 925 915 915 915 915 915 915 915 915 915 91	ૡ૽ૹૻૡ૾ૼૡૼૡૼ
	Distance along runway, ft	1,762 1,762 1,768 1,768 1,768	1,772 1,774 1,776 1,778 1,780	788 784 784 788 788 798	44444 877 888 888	808 404 408 408 408 410 410 410	280 486 988 988 988 988 988 988 988 988 988 9	425 986 986 986 986 986 986 986 986 986 986	449 89 1998 1998 1998 1998 1998 1998 199	278 1918 1918 1918 1 1 1 1 1 1 1 1 1 1 1 1	958 958 958 958 958 958	1, 864 1, 864 1, 866 1, 870 1, 870
ut i mued	Runway elevation, ft	0.474 0.480 1497 774	*****	911 911 2611 611 611 611	254 2514 418 2114 2114	888891 88881	104 200 200 200 200 200 200 200 200 200 2	हेह्रेह्रेह्रेह्रेह्र्	& <u>&amp;</u> &&&&	8.E.E.	<u> </u>	
HEIGHE DATA - Co stimued	Distance along runway, ft	465 969 969 969 969 969 969 969 969 969 9	668 999 9664 9666 9668 9699	,672 676 776 676 776 776 776 776	, 1, 1, 1, 1, 1, 1, 1, 1, 1, 1, 1, 1, 1,	469 869 969 969 969 960	444444 802 802 802 802 802 802 802 802 802 802	1,17 1,17 1,17 1,17 1,17 1,17 1,17 1,17	172 1724 1728 1728 1728 1728	ጚጚጚጚጚ ዾ፝፝፝፞፞ዾ፟ፚ፝ፚ፝፟ቜ	1111111 111111 111111	747 777 777 777 777 777 777 777 777 777
- BASIC RUNWAY (b) COU	Runway elevation, ft	0 5555 5555 5555 5555 5555 5555 5555 5	222 222 222 222 222 222 222	1 1 1 1 1 1 1 1 1 1 1 1 1 1 1 1 1 1 1	855-5-5-5-5-5-5-5-5-5-5-5-5-5-5-5-5-5-5	4444 8844 8844	12222	इन् <u>र</u> ्ट्युव इन्द्र्युव	1997 1997 1997 1997 1997 1997 1997 1997	4.44 4.72 70 70 70 70 70	4244 4844 2894 2844	7.7.988 7.888 885
TABLE L	Distance along runway, ft	2, 2, 2, 2, 2, 2, 2, 2, 2, 2, 2, 2, 2, 2	1, 558 1,	1111 288 288 288 288 298 298 298 298 299 299	1112 244 244 244 244 244 244 244 244 244	1,1,1,1,1,2,2,4,2,5,8,5,5,5,5,5,5,5,5,5,5,5,5,5,5,5,5,5	44444 888888 8888888	44444 6666668 6098609 6008 6008 6008 6008 600	1 1 1 1 1 1 1 1 1 1 1 1 1 1 1 1 1 1 1	248888 68888 68888 6888 6888 6888 6888 6	1111 899 899 899 909 909	1,642 1,642 1,642 1,6466 1,6466 1,646 1,646 1,646 1,646 1,646 1,646 1,646 1,646 1,64
	Rummay elevation, ft	ક્ષે છે. કે છે છે છે છે છે.	र्ट्ट्ट्र्यू इट्ट्र्यू इट्ट्र्यू	8 2 2 2 2 2 2 2 2 2 2 2 2 2 2 2 2 2 2 2	2,2,4,4,4,4,4,4,4,4,4,4,4,4,4,4,4,4,4,4	891 891 891 891 891 891	224 84 84 84 84 84 84 84 84 84 84 84 84 84	8544 8544 85544	22234 222823	889	49444	1959 985 885
:	Distance along runway, ft				44444 88854 88754 877575757575757575757575757575757575757	1 - - - - - - - - - - - - -	1,1,1,1,1, 8,4,8,8,8 8,4,8,8,8,8,9		444444 88888 888888	11111 224 228 228 228 228 227 227 227 227 227 227	111111 888 888 888 888 888 888 888 888	1,1,1,1,1,1,1,1,2,2,2,2,2,2,2,2,2,2,2,2
	Runway elevation, ft	661 291 7510	12 222 222 222 222 222 222 222 222 222	98. 272 286 286	82.5 82.5 83.5 83.5 83.5 83.5 83.5 83.5 83.5 83		886 886 895 895 895 895 895 895 895 895 895 895	99898989	¥.	194 194 194 194 194 194	देवंबंदर्ध	449944 449944
i	Distance along runvay, ft	1, 322 1, 324 1, 326 1, 326 1, 330	4,4,4,4,4,4,4,5,5,5,5,5,5,5,5,5,5,5,5,5	11111 222 232 232 242 252 252 252 252 252 252 252 252 25	2498899 2492 1111111	44444 888 888 888 888 888 888 888 888 8	1, 572 1, 574 1, 576 1, 576 1, 578	888888 1111111	7388 7386 7398 7398 7398 747 747 747 747 747 747 747 747 747 74	201 201 1 1 1 1 1 1 1 1 1 1 1 1 1 1 1 1	टाक्ष 981 भाषक मेने मेने मेने	224 224 1 1 1 1 1 1 1 1 1 1 1 1 1 1 1 1

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Runway elevation, ft		132 126 136		152 841 841 841 841 841 841	941	411 121 821 821		691 191 791 315	8-2	222 222 222 222 222 222 222 222 222 22	6955 1955 1955 1955 1955 1955 1955 1955
Distance along runway, ft	2,532 2,534 2,536 2,536 2,536 2,536 2,536 2,536 2,536 2,536 2,536 2,536 2,536 2,5377 2,5377 2,5377 2,5377 2,5377 2,5377 2,5377 2,5377 2,5377 2,5377 2,	2,542 2,544 2,546 2,546 2,550 2,550	9,9,9,9,9,9,9,9,9,9,9,9,9,9,9,9,9,9,9,	0.0.0.0.0.0 28,8,8,8,0 28,8,8,0 28,8,8,0 28,8,0,0 28,8,0,0 28,8,0,0 28,8,0,0,0 28,8,0,0,0,0,0,0,0,0,0,0,0,0,0,0,0,0,0,0	2,572 2,576 2,576 2,580	~~~~~~~~~~~~~~~~~~~~~~~~~~~~~~~~~~~~~~	9,9,9,9,9,9, 8,8,8,8,8,8,8,8,8,8,8,8,8,8	0,000 2,0000 2,0000 2,000 2,000 2,0	2,5,5,5,5,5,5,5,5,5,5,5,5,5,5,5,5,5,5,5	0288854 029 029 029 029 029 029 029 029 029 029	2,658 665 665 665 665 665 665 665 665 665
Runway elevation, ft		821 821 221	811 901 811 811 811	821 821 821 941	441 146 841 841 841		841. 141. 141.	041 1410- 1411 1411 841		4417 0411 8411 8411	841. 841. 851. 041.
Distance along runvay, ft	2,422 2,422 2,425 2,45 2,4	2222 2222 2222 2222 2222 222 222 222 2	2, 144 144 2, 144 2, 148 148 2, 148 2, 148 2	242889 24444 2020 2020 2020	2399 239 299 299 299 299 299 299 299 299	3,472 2,474 2,476 2,478 2,478 2,478	484 484 484 0 484 484 4 884 4 0 6 4	2,492 2,496 2,496 2,498	20 10 10 10 10 10 10 10 10 10 10 10 10 10	2,512 2,514 2,516 2,518 2,528 2,520	2, 522 2, 524 2, 528 2, 528 2, 530
Runway elevation, ft		2,11 611 22,1,-	991. 991. 991. 991. 991. 991.	217. 217. 217. 217. 217.	168 168 148 145	441 241 121 121	851	567 547 	141 821 141	871 671 671	711 451 851 851
Distance along runway, ft	2, 312 2, 314 2, 316 2, 318 2, 320	22 72 72 72 72 72 72 72 72 72 72 72 72 7	25522 2552 2552 25522 25552 25552 25552	2355 555 56 5 5 5 5 5 5 5 5 5 5 5 5 5 5 5	24980 5555 337 337 337 337 337 337 337 337 33	2352 2555 2555 2555 2555 2555 2555 2555	2,372 2,374 2,376 2,378 2,378	ૡ૾ૡૺૡૺૡૼૡૺ ૹૢૺૢૢૢૢૢૢૢૢૢૢૢૢૢૢૢૢૢૢૢૢૢૢૢૢૢૢૢૢૢૢૢૢૢૢ	2,392 2,396 2,398 2,398 2,398 2,500	2,402 2,404 2,406 2,408 2,408	2,412 414 2,412 2,413 8,418 2,420
Rumay elevation, ft		141 172 178 1188 1188	178 801 178 801	200 100 100 100 100 100 100 100 100 100		238 248 248 316	¥. 1. 1. 1. 1. 1. 1. 1. 1. 1. 1. 1. 1. 1.				
Distance along runway, ft	2,200 2,200	212 217 217 217 217 217 217 217 217 217	222 252 252 252 252 252 252 252 252 252	2,2,2,2,2,2,2,2,2,2,2,2,2,2,2,2,2,2,2,	2,242 2,2442 2,2448 2,248 2,248 2,248 2,250 2,250	۲ ۲ ۲ ۲ ۲ ۲ ۲ ۲ ۲ ۲ ۲ ۲ ۲ ۲ ۲ ۲ ۲ ۲ ۲	20 20 20 20 20 20 20 20 20 20 20 20 20 2	2,272 2,274 2,276 2,290	28888888 39886 39886 39886 39886 39886 39886 39886 39886 39886 39886 39886 39866 399666 399666 399666 399666 399666 399666 3996666 39966666666	2,20 2,20 2,00 2,00 2,00 2,00 2,00 2,00	2,308 2,304 2,306 2,308 2,308 2,308
Runway elevation, ft	0.005 .005 066 	220 100 250 250	450 456 549 549 650	046 046 045 048	048 050 057 055 055	840 740 740	6£0 940 8£0	025 029 051 051 056 058	050 050 069 078	076 085 095 097 100	111- 611- 721- 821-
Distance along runway, ft	2,099 1008 1008 1008 1008 1008 1008 1008 1	% 2010 2010 2010 2010 2010 2010 2010 201	2 4 1 1 8 1 1 8 1 6 7 1 8 1 7 8 1 7 8 1 7 8 1 7 8 1 7 8 1 8 1	27,28 124 27,28 821,2 821,2 7,2 7,2 7,2 7,2 7,2 7,2 7,2 7,2 7,2 7	2,2,2,2,2,2,2,2,2,2,2,2,2,2,2,2,2,2,2,	2,1442 2,1445 2,146 2,148 2,150	2,158 2,158 2,158 2,158 2,158 2,158	291,9 151,9 191,9 191,9 199,10	2,172 2,174 2,176 2,178 2,178	2,2,2,2,2,2,2,2,2,2,2,2,2,2,2,2,2,2,2,	2,192 2,194 2,198 2,200
Runway elevation, ft	34.0 84.1 82.1 87.1 87.1 87.1 87.1 87.1 87.1 87.1 87	sitti	1021 2010 2017 2017 2017	260. 270. 270. 470.	.092 .091 .080 .072	270- 170- 850- 850-	-045 -045 -040 -040	-036 -034 -030 -030 -005 -005		500. 770. 120. 810.	910- 1100- 1000- 1000-
Distance slong runway, ft	4448888 888888888888888888888888888888	4,4,4,4,4,5,888,000 4,998,4,4,4,5,998,000 4,998,8,6,4,4,5,5,5,5,5,5,5,5,5,5,5,5,5,5,5,5,5	0000 0000 0000 0000 0000 0000 0000 0000 0000	2,014 2,016 2,018 2,018 2,018	050 700 2005 2005 2005 2005 2005 2005 20	2,032 2,032 2,036 2,000 2,036	2049 0440 0446 040 040 040 0 040 0 0 0 0 0 0	4,4,5,6,6, 2,5,0,6,6,6,6,7,6,6,6,6,7,6,6,6,7,6,6,7,6,7	000 0000 0000 0000 0000 0000 0000 0000 0000	2,072 2,074 2,076 2,078 2,080	2,084 2,084 2,0866 2,0866 2,0866 2,0866 2,0866 2,0866 2,0866 2,0866 2,0866 2,0866 2,0866 2,0866 2,086 2,082 2,082 2,082 2,082 2,082 2,082 2,082 2,082 2,082 2,082 2,082 2,082 2,082 2,082 2,086
	Distance along Runway the runway ft elevation, ft runway, ft elevation, ft runway, ft elevation, ft runway, ft elevation, ft runway, ft elevation, ft	Interact along runwy, ft         Runwy runwy, ft         R	Interact along runwy, ft         Runwy runwy, ft         Runwy, ft         Runw	Interact allow runwy, ft         Runwy runwy, ft         Runwy, ft	Intense along transv, ft         Runwy, runwy, ft         Runwy, runwy, ft         Runwy, runwy, ft         Runwy, runwy, ft         Runwy, runwy, ft         Runwy, ft         Ru	Difference rules         New yre         Dirfwerer rules         Dirfwerer rules <thdirfwerer rules<="" th="">         Dirfwerer rules         <th< th=""><th>Tuttuber, Line         Tuttuber, Line         Tuttube</th><th>Luttore Line         Intrase trans, rol         Intrase, rol         Intrase trans, rol         Intrase, rol         Interest, rol         <th< th=""><th>Turkur, I., Man, I., Man,</th><th>Turner, Ind         Turner, Ind</th><th>Material         Material         Material</th></th<></th></th<></thdirfwerer>	Tuttuber, Line         Tuttube	Luttore Line         Intrase trans, rol         Intrase, rol         Intrase trans, rol         Intrase, rol         Interest, rol <th< th=""><th>Turkur, I., Man, I., Man,</th><th>Turner, Ind         Turner, Ind</th><th>Material         Material         Material</th></th<>	Turkur, I., Man,	Turner, Ind         Turner, Ind	Material

TANKE I.- BASIC NUMMAY HEIGHT DATA - Continued (b) Continued

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	Runway elevation, ft	121 121 951.0-	128	821 1921 1921 1921	4 9 9 1 1 9 8 1 1 5 8 1 1 1 5 8 1 1 1 1 1 1 1 1 1 1	न्द्राः	-0.115 - 1004 - 1008 - 1088 - 117	
	Distance along runway, ft	276'2 1776'2 276'2	2,9 <del>1</del> 8 2,950	~~~~~~~~~~~~~~~~~~~~~~~~~~~~~~~~~~~~~~	° 896 896 8966 8966 8966 8966 8966 8966	2,972 2,976 2,976 2,980	2,988 986,5,988 9988,6,5,998 9988,6,5,5,5,5,5,5,5,5,5,5,5,5,5,5,5,5,5,5	2,992 2,996 2,998 3,009
	Runway elevation, ft	-0.168 071 168	091 041	147 147 147 147 147 147 147 147	721 151 851	871	961 221 121	051 821 821 221
	Distance along runway, ft	2,882 1986 2,882 2,886 2,986 2,996 2	2,888 2,890	v, v, v, v, v, 8,8,8,8,8 2,4,8,88,8 2,4,8,88,8 8,4,8,88,8 8,4,8,88,8 8,4,8,8,8 8,4,8,8,8 8,4,8,8,8 8,4,8,8,8,8	2,200 2,900 1,9000 1,9000 1,9000 1,9000 1,9000 1,9000 1,9000 1,9000 1,9000 1,9	2,912 2,914 2,918 2,918	2,922 2,928	2,932 934 936 936 949 949 949 949
-	Runway elevation, ft	-0.179 -1.196 -1.188	061 181	891 221 221 221		2211- 9211- 6211- 6211-	091 122 121	155 131 130 135 162
-	Distance along runway, ft	2,822 2,824 2,824	2,828 2,830	9,9,9,9,9 9,9,9,9,9 9,9,9,9,9,9,9,9,9,9	9, 9, 9, 9, 9, 9, 9, 9, 9, 9, 9, 9, 9, 9		962 9664 9864 9866 978 978 978 978	2, 872 2, 874 2, 876 2, 880 2, 880
	Runway elevation, ft	-0.166 -168 -162	158		9911-1- 9811-1- 9811-1-1-	9999 <b>1</b> 5	797 1997 1997 1997 1997 1997 1997 1997	511 891 871
	Distance along runway, ft	2,762 2,764	2,768 2,776	2,772 2,774 2,776 2,778 2,778	282 194 196 198 198 198 198 198 198 198 198 198 198	2,792 1992 1998 1988 1988 1988	0,0,0,0,0,0,0,0,0,0,0,0,0,0,0,0,0,0,0,	2, 9, 9, 9, 9, 9, 9, 9, 9, 9, 9, 9, 9, 9,
and a second	Runway elevation, ft	-0.246 242 249	235	214 201 185 174	841 421 781 781	220	217*- 627*- 987*- 902*-	221 241 161 621
	Distance along runway, ft	2, 702 2, 704 2, 706	2,710	2,712 2,714 2,716 2,718 2,718	2,722 7724 9,728 9,729 9,720 9,729 9	ດູດູດູດູດ 271 280 280 290 290 20 20 20 20 20 20 20 20 20 20 20 20 20	2,742 2,744 2,744 2,748 2,750	8,778 1778 1788 1788 1788 1788 1788 1788
	Runway elevation, ft		292			582 582 582 582 582 582 582	- 278 278 222 222	249 2,
	Distance along runvay, ft	2,642 2,644	20 20 20 20 20 20 20 20 20 20 20 20 20 2	2,658 2,658 2,658 2,658 2,658 2,658 2,658 2,658 2,758	998889 9889 9889 9899 9999 9999 9999 9	2,678 2,676 2,678 2,678 2,678 2,678	ૡૢૢૢૢૢૢૢૢૢૢૢૢૢૢૢૢૢૢૢૢૢૢૢૢૢૢૢૢૢૢૢૢૢૢૢૢ	89898 8999 8999 8999 8999 8999 8999 89

TARUE I.- BASIC RUNNAY HEIGHT LATA - Concluded (b) Concluded

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Portion of runways surveyed

Figure 1.- Diagram of runways at Langley Field, Va.

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Figure 2.- 1

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Figure 2.- Variation of runway elevation about an a





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Figure 3.- Var

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Figure 3.- Variation of runway elevation about a 300-foot moving mean

F163-2

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-foot moving mean.

Fig 3-3





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Figure 5.- Autocorrelation coefficients for runway height.  $X = r \Delta x$ .

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Figure 6.- Power-spectral density functions for the two runways.

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