



SAND92-1381 Unlimited Release Printed February 1993 Distribution Category UC - 235

# Testing and Evaluation of Large-Area Heliostats for Solar Thermal Applications

John W. Strachan Richard M. Houser Solar Thermal Test Department

Sandia National Laboratories Albuquerque, New Mexico 87185

## Abstract

Two heliostats representing the state-of-the-art in glass-metal designs for central receiver (and photovoltaic tracking) applications were tested and evaluated at the National Solar Thermal Test Facility in Albuquerque, New Mexico from 1986 to 1992. These heliostats have collection areas of 148 and 200 m<sup>2</sup> and represent low-cost designs for heliostats that employ glass-metal mirrors. The evaluation encompassed the performance and operational characteristics of the heliostats, and examined heliostat beam quality, the effect of elevated winds on beam quality, heliostat drives and controls, mirror module reflectance and durability, and the overall operational and maintenance characteristics of the two heliostats. A comprehensive presentation of the results of these and other tests is presented. The results are prefaced by a review of the development (in the United States) of heliostat technology.



## ACKNOWLEDGMENTS

The authors are indebted to Jake VanDerGeest for his enthusiastic and tireless assistance in the testing and evaluation of the prototype heliostats. We are grateful to Daniel Alpert who guided our efforts and provided invaluable assistance in the interpretation of our test results, including the performance of HELIOS code simulations of collector performance. Our thanks also go to Bob Edgar, Sam Dunkin, "JJ" Kelton, Larry Yellowhorse, and Winn Erdman for their assistance in numerous technical tasks. The manufacturers of the large-area heliostats provided valuable technical support and many useful suggestions throughout the evaluation period; we thank David Gorman (Advanced Thermal Systems, Inc.) and Daryl Bielenberg (Solar Power Engineering Company, Inc.).

# CONTENTS

1.	Introduction	. 1
	Background on Heliostat Technology Development	. 1
	First-Generation Heliostats	. 1
	Second-Generation Heliostats	. 3
	Large-Area Heliostats	. 4
	Heliostat Costs	. 6
	Objectives and Scope of the Evaluation of the Large-Area Heliostats	. 7
2.	Description of the Large-Area Heliostats	
	Second-Generation Heliostat Design Specifications	. 9
	The ATS Heliostat	. 9
	The SPECO Heliostat	10
	Mirror Focal Length and Heliostat Canting Range	11
3.	Testing and Evaluation of the Large-Area Heliostats	.13
	Overview	
	Tools	. 13
	The Beam Characterization System	.13
	Meteorological Data	. 15
	Reflectivity Measurements	. 16
	Mirror Module Evaluation	. 16
	Mirror Reflectivity	. 16
	Mirror Module Durability	. 17
	Beam Characterization	. 19
	Beam Quality	. 19
	Wind Effects	. 30
	Heliostat Drive and Control Systems	. 35
	Operational Characteristics of the Drive Systems	
	Life-Cycle Testing	
	Tracking Accuracy Tests	
	Failure and Remanufacture of the SPECO Drive	
	Failure and Modification of the Low-Cost Heliostat Drive	.40
4	Summary and Conclusion	.42
	Heliostat Costs	. 42
	Mirrors	.42
	Mirror Module Durability	.42
	Beam Characterization	.42
	ATS Heliostat Beam Quality	.43
	SPECO Heliostat Beam Quality	
	Wind Effects	
	Operational Characteristics of the Drive Systems	
	Drive Control Repeatability	
	Life-Cycle Testing	
	· •	

All-Day Tracking Failure and Remanufacture of the SPECO Drive Failure and Redesign of the Low-Cost Heliostat Drive					
5. References		46			
6. Appendices		47			
	ATS All-Day Beam Quality Test Data				
Appendix B	SPECO All-Day Beam Quality Test Data				
	Normalization Procedure for Comparison of Beam Quality				
	Test Results				
Appendix D	: Analysis of Measurement Error in the Beam Characterization				
••	System				
Appendix E	Determining the Solar Average Specular Reflectivity of a				
••	Collector Mirror	54			

-

# FIGURES

Figure 1. Front View of the Large-Area Heliostats 1
Figure 2. Heliostat Cost Trends
Figure 3. Rear View of ATS Heliostat
Figure 4. Rear View of SPECO Heliostat
Figure 5. Schematic Of Beam Characterization System
Figure 6. Long-Term Reflectivity of Large-Area Heliostat Mirrors
Figure 7. Effects of Soiling on the Reflectivity of the ATS Heliostat
Figure 8. Contour Plots of ATS Heliostat Beam on August 26, 1991
Figure 9. All-Day Performance of the ATS Heliostat
Figure 10. Contour Plots of SPECO Heliostat Beam
Figure 11. All-Day Performance of the SPECO Heliostat
Figure 12. Comparison of the Beam Power of the ATS and SPECO
Figure 13. Comparison of Peak Power and Beam Diameters of the ATS and SPECO25
Figure 14. ATS and SPECO Beam Power Normalized in Terms of Mirror Reflectivity
Figure 15. Comparing Measured and Predicted Beam Profiles for the ATS Heliostat
Figure 16. Comparing Measured and Predicted Beam Profiles for the SPECO Heliostat
Figure 17. Polar Plot of Beam Centroid Movement
Figure 18. Effect of Wind on ATS Heliostat Beam Position
Figure 19. Effect of Wind on SPECO Heliostat Beam Position
Figure 20. Comparison of Linear Curve Fits of Wind Effects Results
Figure 21. Tracking Control Error

# TABLES

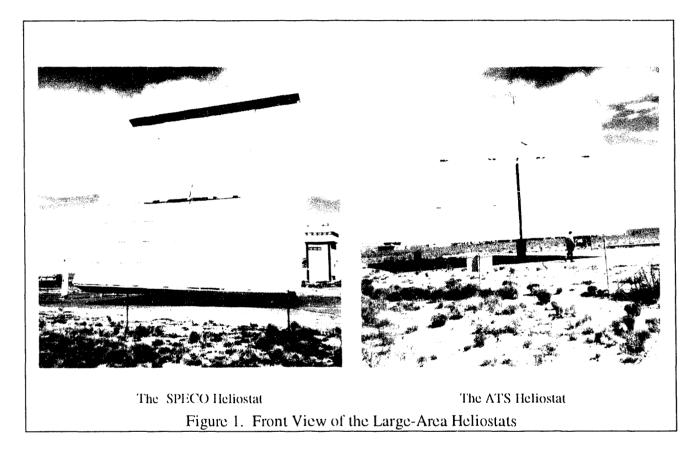
Table 1. Second-Generation Heliostat Design Requirements	4
Table 2. Large-Area Heliostat Specifications	11
Table 3. Measured Beam Characteristics of ATS Heliostat	22
Table 4. Measured Beam Characteristics of SPECO Heliostat	24
Table 5. Summary of Wind Effects Test Results	33
Table 6. Large-Area Heliostat Slew Rates	36
Table 7. Power Utilization of the Large-Area Heliostats	36
Table 8. Large-Area Heliostat Drive Control Repeatability	37
Table 9. Summary of Life Cycling Data for Large-Area Heliostats	38
Table 10. ATS All-Day Beam Quality Test Results	47
Table 11. SPECO All-Day Beam Quality Test Results	48
Table 12. Estimates of BCS Error for Heliostats and Dish Concentrators Measurements	53

=

## 1. INTRODUCTION

## Background on Heliostat Technology Development

Large-area, glass-metal heliostats have collection areas in the range of 100 to 200 m<sup>2</sup>; they are the culmination of more than 10 years of development of commercial heliostats for solar thermal central receiver technology and represent the lowest cost glass-metal heliostat technology developed to date (see Fig. 1).



## FIRST-GENERATION HELIOSTATS

The recent history [1] of heliostat development under the sponsorship of the United States Department of Energy (DOE) began in 1975 with the Pilot Plant System Research Experiment. The heliostat design effort, which was carried out by three US contractor teams led by Martin Marietta Corporation, McDonnell Douglas Astronautics Company, and Honeywell, Inc.,<sup>1</sup> produced the first-generation prototype heliostats

<sup>&</sup>lt;sup>1</sup>Although not detailed here, this research effort explored the development of dome-enclosed heliostats using mirror modules made of aluminized or silvered plastic membranes. Teams led by Boeing Engineering and Construction Co., as well as General Electric, developed designs, and Boeing built four 48-m<sup>2</sup> prototypes.

ranging in size from 37 to 41  $m^2$  of mirrored collection area. The basic heliostat design features that emerged from this effort and were carried forward into succeeding designs included:

- Mirrors made from laminated glass or from a composite (sandwich) of glass and a styrofoam (honeycomb) core. (The laminated glass design proved itself over time to be the better choice.)
- A metal framework upon which one or more mirrors were mounted to form a mirror module.
- An overall metal structure or frame to support the mirror modules. This frame attached to the heliostat drive system.
- A two-axis drive that permitted rotation of the solar collector in elevation (up-down) and in azimuth (east-to-west) combined in one unit.
- A cylindrical pedestal to support the mirror modules, their support structure, and the heliostat drive system.

In a top-to-bottom descriptive sequence 1) mirrors are joined to form mirror modules, 2) mirror modules are bolted or bonded to a support structure, 3) the support structure is welded or bolted to an elevation drive assembly, 4) the elevation drive assembly is joined to the azimuth drive, 5) the azimuth drive is set atop the heliostat pedestal, and 6) the pedestal is set into or onto a concrete foundation.

In 1977, Martin Marietta was awarded a contract to manufacture and install 220 heliostats to serve as the heliostat field for the Central Receiver Test Facility (CRTF) in Albuquerque, New Mexico.<sup>2</sup> The design of these heliostats was customized to meet the changing needs of a test facility: the mirrors they employed were adjustable and smaller than those of Martin Marietta's first-generation design, and the total heliostat area was also smaller ( $37.2 \text{ m}^2$  versus 40 m<sup>2</sup>).

In mid-1978, Martin Marietta and McDonnell Douglas were contracted to develop competitive designs for Solar I, a 10-MW<sub>e</sub> pilot solar thermal (central receiver) electric power plant built in Barstow, California. Prototype heliostats were evaluated at Sandia's CRTF, and Martin Marietta was subsequently (1980-81) awarded the contract to manufacture 1,911 first-generation heliostats and install 1,818 of them in the Solar I plant. The remaining 93 heliostats were for installation at the European solar thermal test facility in Almeria, Spain. The design of these heliostats incorporated improvements in both the mirror modules and the drive mechanisms, the need for which had been identified in the test and evaluation phase of the DOE Pilot Plant System Research Experiment.

<sup>&</sup>lt;sup>2</sup>The facility is known today as the National Solar Thermal Test Facility (NSTTF).

## SECOND-GENERATION HELIOSTATS

A second round of DOE-sponsored heliostat development took place between 1977-1981. First, with the goal of reducing the cost of the first-generation heliostats, a series of component design studies was carried out by four US companies (Boeing, General Electric, McDonnell Douglas, and Solaramics) [2]. An important result of that study was the identification of superior heliostat design approaches, approaches that were mature, well analyzed and tested, and carried a minimum risk.

Beginning in 1979, contracts were issued to design and build prototype heliostats that would be tested by a national laboratory (Sandia National Laboratories, hereafter referred to as Sandia) and to develop detailed plans and cost estimates for the low-cost, mass manufacture, assembly, installation, and maintenance of heliostats. Five US companies (ARCO, Boeing, Martin Marietta, McDonnell Douglas, and Westinghouse<sup>3</sup>) participated in this effort. The resulting second-generation glass-metal heliostats were a step larger than the first generation and ranged in size from 44 to 57 m<sup>2</sup>. These heliostats were subjected to thorough testing at the CRTF. An important technical result of this effort was the definition of heliostat design specifications, which include a) operational modes, b) optical performance, c) survival, and d) a 30-year life [1]. These specifications are reprinted in Table 1.

A follow-on effort was carried out by Martin Marietta and McDonnell Douglas, which sought to further reduce heliostat costs by identifying justifiable changes to either the heliostat design specifications or the specified environmental conditions. Some of the significant findings/recommendations [3, 4] resulting from this effort were that:

- Heliostats should be designed for strength (against high wind) rather than for stiffness in operational winds.
- They should be designed against standard design code wind speeds rather for the 100- to 200-year winds.
- Because of their proximity to the receiver, 50 to 60% of the heliostats in a field that are closest to the receiver tower can have lower pointing accuracy and poorer beam quality without degrading heliostat field performance.
- Heliostats that are located to the inside of the heliostat field will experience a reduction of their wind loads due to shielding from the outlying heliostats and can therefore be designed to lower wind loads.

<sup>&</sup>lt;sup>3</sup>Westinghouse participated in the design portion of this effort but did not build a prototype.

Table 1. Second-Generation Heliostat Design Requirements [1]				
Category: Requirements:				
Operational Modes:	• Normal modes (track, standby, wire walk, stow)			
	Track in up to 35-mph wind			
	• Slew in 50-mph wind			
	Resolve tracking singularity in 15 minutes			
	Reposition in 15 minutes			
	Emergency defocus in 3 minutes			
	Electrical transients (operate through a 3-cycle dropout)			
Optical Performance:	• Beam pointing (1.5 mr RMS maximum, reflected beam error for each axis)			
	<ul> <li>Beam quality (theoretical beam shape plus 1.4 mr fringe, 32°F - 122°F)</li> </ul>			
	• Wind load deflection (3.6-mr RMS maximum reflective surface			
	deflection in 27-mph wind, discounting foundation)			
	• Foundation deflection (0.45-mr maximum set after survival),			
	1.5-mr maximum twist or tilt in 27 mph wind)			
Survival:	• 90-mph wind, heliostat stowed			
	• 50-mph wind, heliostat in any orientation			
	• Temperature, -20°F - 122°F			
	• Hail, 3/4 in. at 65 ft/sec, any orientation; 1 in. at 75 ft/sec,			
	heliostat stowed			
	Cold water shock			
30-Year Life:	• Life of all components must be cost effective for 30 years			
	Mirror and drive mechanism are critical components			

• Depending on the site of use, a narrower operational ambient temperature range may be possible and could benefit mirror modules that defocus with decreasing temperatures.

One of the more important consequences of these results was the movement in commercial heliostat design toward increased heliostat size.

## LARGE-AREA HELIOSTATS

Large-area heliostats represent the continued evolution toward lower cost of the second-generation heliostats, and private industry has been responsible for that evolution. After the second-generation heliostat effort, ARCO built thirty 53-m<sup>2</sup> heliostats for a central receiver plant that provided process heat for enhanced oil recovery [5]. Later, both McDonnell Douglas and ARCO built heliostats in the 85- to 95-m<sup>2</sup> size range. ARCO manufactured 865 units of this size: 864 as photovoltaic (PV) trackers and 1 as a heliostat. 756 of the PV trackers employed flat mirrors to double the direct sunlight incident upon their photovoltaic modules. ARCO further increased the heliostat size to 148 m<sup>2</sup> and built 45 such units, two of

them as heliostats and 43 as PV trackers equipped with nearly 150  $m^2$  of mirrors to reflect sublight onto the PV arrays.<sup>4</sup>

In addition to increased collection area, the post-second-generation heliostats underwent several noteworthy design improvements [5]. The ARCO heliostat drive system was made simpler and less costly by the replacement of its stepper motors and their costly electronic controllers with DC motors and Hall-effect encoders. The ARCO second-generation mirror module also underwent a complete transformation. In the former design, modules were formed from mirrors mounted on a box structure substrate made of formed sheet metal ribbing. This design was replaced with laminated glass-on-glass mirrors glued to rolled-metal hat sections. The result was a lighter, less costly, and more durable mirror module. SPECO developed a  $200-m^2$  heliostat with a wind spoiler designed to reduce heliostat loads induced by high winds in the stow position. This SPECO heliostat also explored the use of second surface mirrors protected only by paint (applied to the back side).

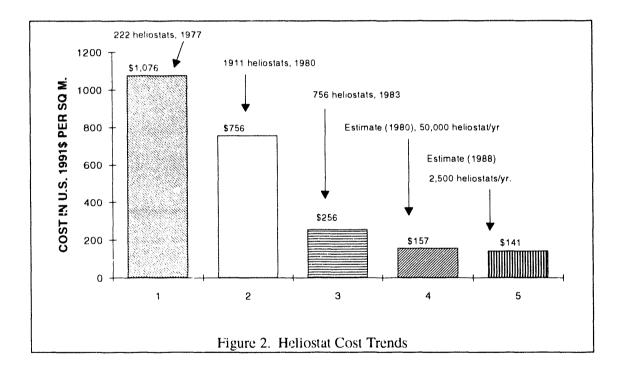
Large-area heliostats have the potential for lower cost per unit collection area. While the cost per unit area of the mirrors and for the support structure remains fairly constant with increasing heliostat size, the cost of the remainder of the heliostat can be spread over a larger area. In other words, a heliostat field with large-area heliostats would have fewer pedestals and drive systems, and less attending controls hardware and field wiring. However, the size of a heliostat drive system and the magnitude of pedestal loads increase as the cube of the heliostat's nominal diameter. For this reason, the cost of the heliostat drive and the survival of the heliostat under dynamic wind loads are major concerns, especially large-area heliostats because of their greater mass and lower natural frequency. Another concern is the increased beam size due to off-axis aberration and the possibility of reduced optical performance due to gravity-induced sagging of the mirror support structure.

Given the concern over the cost and wind survival of heliostat drives, an effort to develop a low-cost drive for heliostats was started in 1986. Sandia placed a contract with Peerless-Winsmith, Inc. to design a prototype low-cost drive [6]. The most promising design was to be built and tested in the lab and on a NSTTF heliostat. This low-cost drive replaced the drive originally supplied by ATS (manufactured by A.G. Flender), and was the ATS's drive during its test and evaluation period.

<sup>&</sup>lt;sup>4</sup>When ARCO withdrew from solar thermal R & D, a number of ARCO employees purchased the rights to ARCO's heliostat technology and formed Advanced Thermal System, Inc. (ATS). ATS supplied one of the two large-area heliostats that are the subject of this report.

#### **HELIOSTAT COSTS**

Cost information for heliostat manufacture and installation in the US is limited because they have been manufactured here only in small quantities. There were 222 heliostats built and installed at the Central Receiver Test Facility (CRTF) in Albuquerque, New Mexico in 1977, and these had an installed cost of (1977) per square meter of energy collection area  $(/m^2)$ . The most manufactured were the 1,911 heliostats built for Solar I (and the Almeria solar plant) in 1980; these heliostats were built at a cost of (1970), which represents a cost reduction of about 30 % over the CRTF heliostats. In 1983, 756 mirror-enhanced PV trackers were manufactured and installed at Carrissa Plains in 1983 by ARCO; these were of post-second-generation design. As heliostats, they would have had collection areas of 95 m<sup>2</sup> each; their cost (estimated by adjusting the cost of the trackers for the material differences) was  $214/m^2$  (1983). Only recurring costs (and no developmental costs) were included in Fig. 2.



Several manufacturing studies conducted during the second-generation heliostat development effort include cost estimates for the mass-manufacture of heliostats [2,7]. These studies were conducted prior to the completion of the Solar I heliostats and do not reflect the lessons learned from the manufacture of even a

<sup>&</sup>lt;sup>5</sup>Based on the Producer Price Index for durable (non-food) consumer goods, Table B-61, page 867, *Economic Report to the President Transmitted to the Congress, February*, 1992, U.S. Government Printing Office, 1992.

modest quantity of heliostats. The manufacturing cost studies undertaken by ARCO, Martin Marietta, and McDonnell Douglas as part of their second-generation development effort produced estimates that ranged from  $100/m^2$  to  $116/m^2$  (1980\$) for installed (second-generation) heliostats [7].<sup>6</sup> These estimates were based on assumed annual production rates of 50,000 units.

A more recent heliostat cost study provides perhaps the most accurate available estimate for large-area heliostats [8], albeit for relatively small production quantities. A study performed in 1988 for PG&E estimated the installed cost for large-area heliostats based on the manufacture of 25,000 units over a tenyear period at a rate of 2,500 per year. Under these conditions the estimated (installed) capital cost was  $131/m^2$  (\$1988). The results of this study are more meaningful because the study benefited from the cumulative experience in heliostat manufacture up to 1988. However, the production rate of 2,500 heliostats per year is low, and one would expect additional cost-reductions for larger quantity heliostat manufacture.

Figure 2 provides a graphical summary of heliostat cost trends, all costs adjusted to 1991\$. The cost value used for the second-generation heliostat is the higher value (116  $m^2$  in 1980\$) cited in the *Second Generation Heliostat Evaluation Executive Summary* [7]. It should be not<sup>a</sup>d that this estimate was made in the absence of any actual industry experience at that time in volume manufacture of heliostats. The estimate was based on an assumed production level of 50,000 heliostats per year (a rate 20 times greater than that assumed for the large-area heliostat manufacturing cost estimate made in 1988). All costs shown in the figure have been adjusted for inflation and converted into 1991 dollars using the Producer Price Index for durable, finished (nonfood) consumer goods (see footnote #6).

## **Objectives and Scope of the Evaluation of the Large-Area Heliostats**

In the mid-1980s, Sandia purchased two large-area heliostats with the intent of evaluating their performance characteristics. The heliostats, which had solar energy collection areas of 148 and 200 m<sup>2</sup>, were built respectively by Advanced Thermal Systems, Inc. (ATS) and Solar Power Engineering Company (SPECO). The heliostats were installed for testing and evaluation at Sandia's NSTTF, the ATS heliostat in 1986 and the SPECO in 1987.

The purpose of the test effort was to evaluate performance and durability of the devices and to seek answers to a number of related questions: would the heliostats perform according to their specifications? How much power would they be capable of delivering to a receiver? What would be the overall size of the heliostat beam? What performance penalty, if any, would there be because of the increased beam size associated

<sup>&</sup>lt;sup>6</sup>These cost estimates were adjusted to set equal the cost of identical materials in the three studies.

with a larger collector? Would the heliostats perform within specifications under moderately high winds? Would they be vulnerable to wind damage?

The results of that test and evaluation effort, which was conducted from 1987 to 1992, are presented here. The effort focused on heliostat beam quality, wind effects, heliostat drives and controls, mirror module performance and durability, and overall operational and maintenance characteristics.

## 2. DESCRIPTION OF THE LARGE-AREA HELIOSTATS

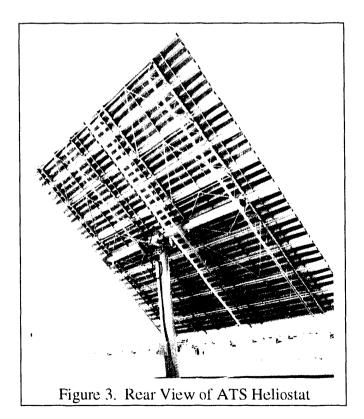
## **Second-Generation Heliostat Design Specifications**

As mentioned in the background discussion of Section 1, one of the important products of the DOEsponsored heliostat development effort was the definition of a set of specifications or design/performance guidelines for commercial heliostats. These specifications reflect the experience gained through two generations of heliostat design, manufacture, and testing. Known as the "Second-Generation Heliostat Design Requirements," they are reproduced [1] in Table 1 as a point of reference.

## The ATS Heliostat

The mirror sections of the ATS heliostat are  $1.22 \text{-m}^2$ , 1-mm-thick silvered glass bonded to 3-mm glass. Its mirror modules are formed by adhering the mirrors side-by-side to four parallel, sheet-metal hat sections. The four hat sections are fastened together and stiffened by three cross members to form a mirror module.

The rectangular mirror modules are supported on the heliostat by a rack constructed from two trusses welded to the heliostat's torque tube (see Fig. 3). There are 10 mirror modules bolted to each of the heliostat's 2 racks for a total of 20 mirror modules. The torque tube itself is bolted to either side of the heliostat's elevation drive.



The original ATS drive system was manufactured by A.G. Flender. Its azimuth and elevation drives were both two-stage (two gears in series) worm gears with gear ratios of approximately 18,600:1. Both drives employed 90 VDC, 1/4-hp motors.

As part of the DOE effort to develop heliostat technology, a US gear manufacturer, Peerless-Winsmith, Inc. was contracted in 1986 to develop a low-cost drive for two-axis tracking solar collectors. The product of that effort [6] was installed on the ATS heliostat in 1988 so that it could be evaluated. As a result, all the tracking accuracy, wind effects, and other drive-systemrelated tests were performed on the Winsmith low-cost drive system (not the A.G. Flender drive system). Its elevation drive employs a jack screw, while the azimuth drive is a planocentric drive. Initially the Winsmith drive was equipped with 1/2-hp motors, but these were found to coast (had no inherent braking). This produced excessive updating by the heliostat controller and led to their replacement with 1/4-hp motors. The motors on both drives were equipped with Hall-effect encoders for position feedback.

The heliostat drive motors are controlled and driven by a local control board (mounted in a weather-tight box on the heliostat pedestal) equipped with an on-board microprocessor and power supply. Operator control of the heliostat is via a computer program running on a personal computer (PC) that is connected by its I/O (input/output) port to the local control board. Manual control is also an option with a manual control box that plugs directly into the local control board at the base of the heliostat pedestal.

The essential specifications and component descriptions of the two large-area heliostats are summarized in Table 2.

## The SPECO Heliostat

The SPECO heliostat employs 3.2-mm-thick glass/mirror facets measuring  $1.5 \times 1.8 \text{ m}$ . The glass is second surface silvered, double strength, plain float glass; the mirrored back surface is protected with a coat of white paint. Bonded to the backside of each mirror are nine threaded studs to facilitate mounting to the mirror module. The mirror modules are formed by securing two mirrors with their studs to a rectangular, hollow-tube steel frame. The 36 mirror modules are bolted to a K-frame rack structure, as shown in Fig. 4 with four rows and nine modules per row.

The K-frame rack to which all the mirror modules are secured is bolted to the elevation gear housing. The elevation drive gear in turn is mounted atop the azimuth gear box, which itself is affixed to the heliostat's (0.61 m (2 ft) diameter steel pedestal).

The SPECO heliostat has two wind spoilers perpendicular to the mirrored surfaces and attached to the top and bottom edges of the heliostat (visible in Fig. 4).

The SPECO's azimuth and elevation drives are worm gears (with 29,200:1 ratios). They are actuated by 1/4-hp DC motors with Hall-effect encoders for position feedback. The drive was manufactured by Hub City.

The designers of the SPECO heliostat opted to use the same heliostat control system developed by ATS and employed in their heliostat. It is described in the preceding section.

Table 2.         Large-Area Heliostat Specifications					
	SPECO	ATS			
Configuration:	Glass-Metal Dual Axis Tracking	Glass-Metal Dual Axis Tracking			
Area:	$200.3 \text{ m}^2$ (2,156 ft <sup>2</sup> )	$146.9 \text{ m}^2 (1,581 \text{ ft}^2)$			
Height	14.75 m (48.39 ft)	12.56 m (41.21 ft)			
Width	14.52 m (47.64 ft)	12.34 m (40.49 ft)			
Stow Position:	Normal: vertical	Normal: vertical			
	High wind: horizontal, face up	High wind: horizontal, face up			
Mirror Modules:					
Size					
Area	• $5.57 \text{ m}^2 (60 \text{ fi}^2)$	• $7.44 \text{ m}^2 (80 \text{ ft}^2)$			
Height	• 5.52 m (5.00 ft)	• 1.22 m (4.00 ft)			
Width	• 3.66 m (12.01 ft)	• 6.1 m (20.0 ft)			
Reflectivity <sup>7</sup>	• 83%	• 94%			
Mirror	3.2-mm silvered glass (1.5 m x 1.8 m	1-mm silvered glass bonded to			
	facet); its mirrored backside is	3-mm glass. Module formed by			
Backing &	protected with paint.	bonding 5 mirrors to an aluminized			
Substrate:	9 threaded studs attached to backside	sheet metal hat section with cross			
	of facet. 2 facets per module.	members attached by 4 threaded studs.			
	The steel frame module is bolted at 3	These are positioned at four points to 2			
Mounting	locations to a "K-frame" rack	truss racks which are welded to a			
Structure:	structure designed with integral	torque tube. The tube is bolted to both			
	wind spoilers.	sides of the elevation drive.			
Drive System:	Dual axis drive <sup>8</sup> mounted at top of	Low-cost heliostat drive mounted to			
·	pedestal	top of pedestal			
Azimuth	• 29,200:1 ratio worm gear	Planocentric drive			
Elevation	• 29,200:1 ratio worm gear	Jack screw			
Motors	• 1/4-hp DC, with Hall-effect	• 1/4-hp DC, with Hall-effect			
	position indicators	position indicators.			
Electronics	Heliostat controller via	Heliostat controller via			
	microprocessor and power supply.	microprocessor and power			
	<ul> <li>Computer with I/O heliostat</li> </ul>	supply.			
Control	controller.	Computer with I/O heliostat			
		controller.			
Pedestal:	0.61-m-(2 ft) diameter steel pipe	0.61-m-(2 ft) diameter steel pipe			
	pedestal placed in augured hole with	pedestal placed in augured hole with			
	concrete backfill.	concrete backfill.			

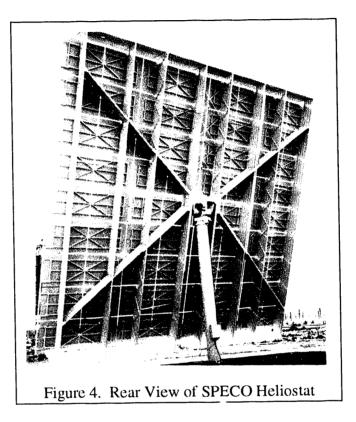
## **Mirror Focal Length and Heliostat Canting Range**

The focal length of the individual mirrors in a module is set during their manufacture by giving them a slight curvature. The ATS mirror modules were manufactured with a nominal focal length of 305 m (1000 ft). The SPECO mirror modules were manufactured with a nominal focal length of 244 m (800 ft).

-

<sup>&</sup>lt;sup>7</sup>Solar-averaged hemispherical reflectivity of clean mirror following installation of the heliostat. <sup>8</sup>Manufactured by Hub City.

When mounted to the module support structure, the individual mirror modules were aimed (or canted) to a common aim point corresponding to the slant range of the heliostat (to the intended receiver). The design slant range for the ATS and SPECO was 235 m (771 ft).



## 3. TESTING AND EVALUATION OF THE LARGE-AREA HELIOSTATS

## **Overview**

The ATS heliostat was installed at Sandia's NSTTF in Albuquerque, New Mexico, in 1986; the SPECO in 1987. The test program covered the six-year period from 1986 through 1992, and focused on the issues of heliostat beam quality, mirror module performance and durability, tracking accuracy, dynamic wind effects, and overall operational and maintenance characteristics. A number of evaluation tools was employed in this effort, the most important of which are described here.

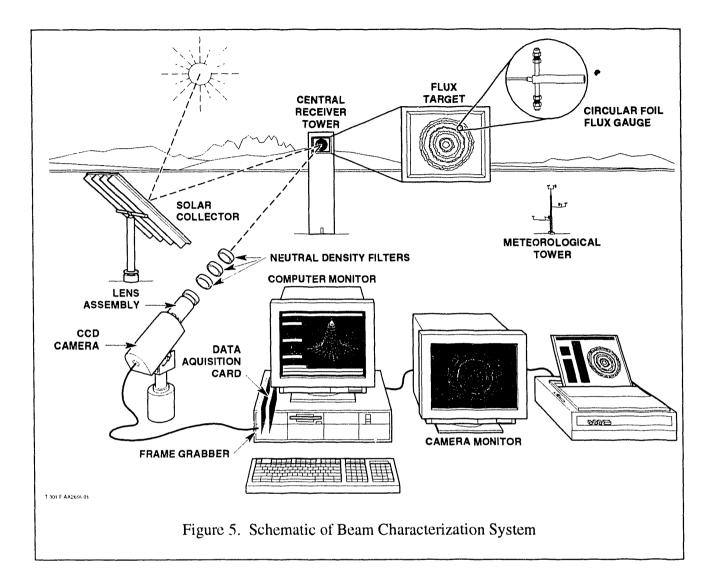
## Tools

## THE BEAM CHARACTERIZATION SYSTEM

The Beam Characterization System [9] (BCS) is a primary tool used at the NSTTF for evaluating the performance characteristics of solar concentrators and their mirror modules or facets. It was used in numerous tests of the large-area heliostats including beam quality, dynamic wind effects, and tracking accuracy and repeatability tests.

In tests such as these, the BCS is employed to image the beam of concentrated solar energy as it is tracked by the heliostat on a flux target. The result of the BCS process is a flux map of that beam. The information obtained includes 1) the location of the beam centroid and peak, 2) the flux densities over the entire beam, 3) the peak flux, 4) the total beam power, and 5) the nominal diameter of a circle that contains all flux equal to or greater than 10% (or any specified percentage) of the peak flux. Other standard image analysis functions can also be performed on the BCS image.

The components of the BCS system and their interrelation are depicted in Fig. 5. The flux target used by the BCS is a white, nonspecular reflective surface with unique optical properties. The relative position of the target approximates the position of a solar receiver in a central receiver plant. (Because of practical considerations, it is actually positioned somewhat lower than a receiver would be, approximately at the midpoint of Sandia's solar tower.) Images of the heliostat beam on the target are captured using a video-type camera positioned at a convenient location with a normal or near-normal view of the target. The target surface is painted with a high-temperature titanium-oxide paint. Because of the diffuse (Lambertian) reflectance characteristics of the target, the intensity of the reflected light reaching the camera from each point on the surface of the target is directly (linearly) proportional to the intensity of the light reaching that same point on the target from the concentrator. This characteristic of the BCS target is essential to the measurement technique and ensures the desired result: the image of the flux beam captured by the camera is a scaled version of the actual flux incident upon the target.



Because of the extreme brightness of the light from the target, the BCS camera is equipped with neutral density filters to attenuate the light to within the camera's working dynamic range. The lens system may or may not include a zoom lens, depending on the distance from the camera to the BCS target. In any case, the target and the collector beam are imaged on the camera's sensor, digitized by a commercial frame-grabber, and processed by image analysis software that is resident on the BCS's high-end personal computer.

A flux gauge mounted on the surface of the BCS target provides a single absolute measurement of flux at one point in the BCS image. Together with the value of the picture element (pixel) intensity at that location in the image, the flux gauge reading is used to establish a conversion factor, which can be applied to the BCS image. Using this factor, the intensity value of every pixel in the image is converted into an absolute flux density value. In this manner the BCS image becomes a flux map.

The analysis functions of the commercial image analysis system (IAS) produce most of the required BCS information, including the locations of the beam centroid and peak flux, and the diameter of the beam.<sup>9</sup> When multiple images are analyzed, such as in tracking error or dynamic wind effect tests, the IAS can compute and display the movement of the beam centroid from image to image.

As stated before, the magnitude of the peak power and the total beam power is obtained by applying the conversion factor (pixel-level-to-flux-density) to the IAS's results (which are in relativistic, pixel-intensity units).

The BCS also has analog data acquisition capability, which is employed to measure the flux gauge mentioned above and relevant environmental conditions including wind speed, wind direction, and normal incident insolation. These instrument readings are acquired at the same time as the beam image data. Meteorological data acquisition is described below.

The results from the analysis of image and non-image data are typically transferred to a spreadsheet for final analysis. Based on a preliminary error analysis of the BCS, the expected measurement error (i.e., the standard deviation for measurements) of that system for the tests described in this report is 6 to 10%. Measurement resolution is approximately 1% of the full-scale flux value measured. An analysis of the BCS error sources is provided in Appendix D.

#### **METEOROLOGICAL DATA**

Environmental conditions including wind speed, wind direction, and direct normal solar insolation can be measured each time a BCS image is acquired. A meteorological tower located 100 m west of the ATS heliostat<sup>10</sup> is equipped with wind speed (3-cup anemometers) and wind direction (lightweight vanes) sensors at three heights: 10.0, 6.1, and 3.1 m (32, 20, and 10 ft) above the ground. A normal incidence pyroheliometer (NIP) is mounted on a separate platform located 75 m north of the two large-area heliostats. Cabling from these sensors as well as from the flux gauge is routed to a data acquisition board in the PC, which is the platform for the BCS. Each time a camera image is acquired, the BCS is able to sample each of the sensors and digitize the reading; the meteorological data is then stored with or separate from the acquired beam image, as desired.

<sup>&</sup>lt;sup>9</sup>Typically, the diameter of a circle containing all flux that is greater or equal to 10% of the peak flux.

<sup>&</sup>lt;sup>10</sup>The SPECO is located 50 m due east of the ATS; the two heliostats are both 240 m north of the NSTTF's Solar Tower upon which is located the BCS target at a height of about 30 m.

#### **REFLECTIVITY MEASUREMENTS**

Mirror reflectivity was measured (both clean and dirty) at regular intervals throughout the six-year test period. The specular reflectivity of the mirrors was measured with a Device and Services (D & S) Portable Specular Reflectometer Model 15R (Serial #013), using a receiver aperture set to an effective acceptance angle of 15 mr. This instrument has an accuracy of 0.01, a repeatability of 0.005, and a resolution of 0.001, all in absolute reflectance units [10]. Because the D & S device uses 660-nm-wavelength light as its light source, the measured reflectance values require an adjustment to obtain a solar average specular reflectivity value for the heliostat. Information regarding this procedure is provided in Appendix E.

## **Mirror Module Evaluation**

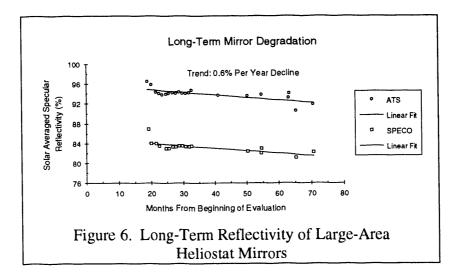
### MIRROR REFLECTIVITY

The ATS and the SPECO mirrors employ silver mirroring material (mirror module design and composition described in Section 2). The mirrors employed for the SPECO heliostat were known in advance to have low reflectivity, but Sandia accepted their use as an expedient. This type of mirror, however, would normally not be selected for a working (commercial) collector system.

During the evaluation period at Sandia, the heliostats were not cleaned except by the natural effect of rain and snowfall. Exceptions to this were made on two or three occasions when, for the purpose of investigating cleaning techniques, the mirror modules were sprayed with a high-pressure solution of water and cleaning agent. Reflectivity measurements were made of the clean and of the dirty or "as is" condition of the mirrors. The clean mirror measurements were made after wetting a portion of a mirror modules with deionized water and then wiping dry that area with an absorptive paper towel. Each (clean or dirty) specular reflectance value was obtained from 20 measurements at randomly selected points on the mirror modules on the bottom row of the heliostat (the row closest to the ground when the heliostat is in the vertical position). The average of the 20 measurements was then computed and adjusted to obtain the solar average value.

#### Long-Term Mirror Degradation

The reflectivity of the mirror modules of the two heliostats suffered an average degradation rate of 0.6 to 0.7% per year during the evaluation period. These trends in clean mirror reflectivity are plotted in Fig. 6. The trend in degradation is fairly linear: there is no indication from the data of a leveling off of the degradation.



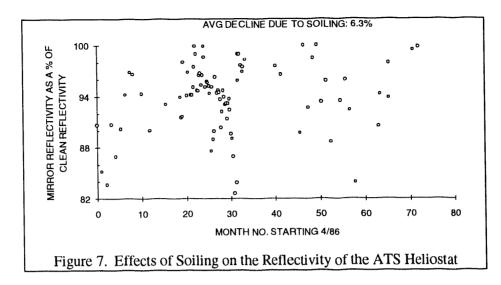
### **Mirror Soiling**

The "as is" reflectivity of the mirror modules of both heliostats varied considerably over the short term, due to soiling by the interaction of airborne soil, water condensation on mirrors, and rain and snow. To demonstrate the effect of mirror soiling, the "as is" (dirty) mirror reflectivity of the ATS heliostat is plotted in Fig. 7 as a percentage of the **current clean-mirror reflectivity value**. The SPECO heliostat data exhibited a very similar trend. Note that on this plot a reflectivity value of 100% only indicates that the dirty and clean mirror reflectivity was the same at that time (probably due to the cleaning action of a recent rain or snowfall). This plot illustrates the magnitude of the negative effect soiling will have on heliostat field power production. On the average, soiling reduced mirror reflectivity (and therefore thermal power output) of the large-area heliostats by 6.3 and 8.8% for the ATS and the SPECO, respectively.

### MIRROR MODULE DURABILITY

#### **Mirror Corrosion**

A detailed inspection of the mirror modules of both heliostats was performed in August 1992 to determine the condition of both mirrors and modules. Several mirrors on the SPECO heliostat were broken by snow accumulation caused by interference of the lower wind spoiler with the bottom row of mirrors. (More details are provided in "Mirror Breakage and SPECO Wind Spoiler Design"). Excepting that, the condition of the SPECO mirrors and the mirror modules was good. A minor amount of flaking of the protective paint on the rear edges of some of the mirrors was observed, however. (The SPECO mirrors are 3.2-mm backsurface silvered mirrors with a protective coat of white paint.) The nominal dimensions of the dried flakes of protective paint observed along the mirror edges were approximately 1/3 x 1/8 in. Slight corrosion of the silver mirroring material was observed at the sites where peeling had occurred. There was <u>no</u> indication that



the peeling process was advancing inward to the interior surface areas of the mirrors. This peeling, and the associated mirror corrosion, was observed on 6 of the 24 SPECO mirrors that were inspected.

SPECO mirror modules, the immediate support structure for the mirrors, were also in good condition. The pads to which the threaded studs are attached and which are bonded to the back surface of the mirrors showed no evidence of debonding or deterioration of their adhesive.

The ATS mirrors and modules were also found to be in good condition, with no evidence of corrosion or debonding of any kind. Some broken mirrors were observed; the breaking occurred at the time of failure of the (Winsmith) low-cost heliostat drive. (More details are provided in "Failure and Modification of the Low-Cost Heliostat Drive".) There was some indication of slight mirror corrosion along some of the broken edges of these mirrors.

## Mirror Breakage and SPECO Wind Spoiler Design

Breakage of several SPECO mirrors took place during the winter of 1990. Snow had accumulated on the mirror modules while the heliostat was in its stowed horizontal position.<sup>11</sup> To remove the snow, the heliostat was moved to a vertical position so the snow could slide off the mirror modules. When the snow slid off the mirror modules it landed on the lower wind spoiler where it became packed and trapped. When the heliostat was returned to the heliostat stow position, the weight of the accumulated packed snow was sufficient to break the lower edges of two of the eight mirrors on the bottom row.

<sup>&</sup>lt;sup>11</sup>The horizontal stow position is favored because it is the position of least vulnerability during high winds.

There was no indication from the evaluation that the SPECO mirrors were substandard. Rather, the lesson learned was that this particular arrangement of the lower wind-spoiler facilitates the accumulation of ice or snow on mirrors and can result in breakage. A redesign of the lower wind spoiler is suggested, as well as attention to any structural elements at the boundaries of the mirror modules. The experience also suggests that snow removal should be a well-planned maintenance activity in systems that have large solar collection areas.

## **Beam Characterization**

## **BEAM QUALITY**

## **Purpose of Beam Quality Testing**

The optical performance of the large-area heliostats was measured in order to characterize the concentrated solar energy they can deliver to the central receiver. The objectives of the beam quality tests included a) characterization of the heliostat beam, including total beam power, peak power, and overall beam shape and b) evaluation of changes in beam quality throughout the day (due to gravity effects as the heliostat elevation angle varied, changes in insolation and sunshape, and changes in heliostat-to-receiver orientation.

## **Description of the Beam Quality Test**

The primary tool used for beam quality tests was the BCS. The BCS is described in some detail in the subsection "Tools" in the introduction to this section.

For the beam quality tests, an image of the heliostat beam was captured with the BCS as the beam was tracked on a flux target. The image was then scaled into flux density units by applying to each pixel in the digitized image a (pixel-intensity-to-flux-density) conversion factor, based on the measure of a single flux gauge in the BCS target. The processed image reveals the peak power density, the locations of the peak as well as the beam centroid, the total power, the cross-sectional area containing the power, and the overall distribution of the power in the heliostat beam.

On several representative days of the year, beam quality measurements of the two large-area heliostats were made with the BCS over the course of the day. These tests were performed under clear-sky and low wind speed conditions (less than 10 mph).

The results of the beam quality test are a good measure of the flux distribution that these heliostats can deliver to a central receiver. Given that the flux measurements are made on the BCS target, the specific spatial relationship of the heliostats to the BCS target was as follows:

- the slant range from both beliostats to the BCS target was 240 m (800 ft).
- the elevation angle from the heliostats to the target was 5°.
- the target was south of both heliostats to within 5°.

#### **Results of the Beam Quality Test**

The all-day beam quality tests that were performed on the ATS and SPECO heliostats included tests performed on August 26 and 27, 1991, respectively. The results on these two days are presented because they are typical as well as representative of all test results obtained. On both test days, the beam quality measurements were made at 30- to 45-minute intervals over the course of the day using the BCS. Each beam quality measurement produced an estimate of the heliostats' total beam power (in kW), the peak flux (kW/m<sup>2</sup>), and the beam diameter (m). The direct normal insolation (kW/m<sup>2</sup>) and the wind speed and direction were also measured at each of these times. For each test, an estimate was made as to the theoretical maximum total beam power, based on the measured insolation, the total reflective area of the heliostat, its reflectivity (as measured at the time of the test), and the angle between the heliostat normal and the sun (to calculate the resulting cosine loss) at the time of the measurement. Given the measurement uncertainty inherent in the BCS, the total power and peak flux values reported here have an expected measurement uncertainty (standard deviation) of 6 to 10%. Significant differences occurred between the measurements), but they are within the expected (10%) error range. An error analysis of the BCS is presented in Appendix D.

#### ATS Heliostat Beam Quality Measurements

The all-day beam quality test of the ATS consisted of 20 BCS measurements made at 30- to 45-minute intervals throughout the day. Contour plots for a selection of nine of these measurements are provided in Fig. 8. The effects of cosine angle and insolation level on the beam shape are easily appreciated from the plots: at times distant fro—solar noon, the reduced insolation and the larger cosine loss result in reduced power and a larger beam diameter.

Table 3 provides a listing of the actual measured values for total beam power, peak flux, and beam diameter associated with the nine contour plots.

08:30 a.m.	10:03 a.m.	10:32 a.m.				
Ø						
11:28 a.m.	12:30 p.m.	2:10 p.m.				
3:12 p.m.	4:12 p.m.	5:45 p.m.				
Figure 8. Contour Plots of ATS Heliostat Beam on August 26, 1991. The dimensions (h x w) of each plot window are 9.0 x 10.5 m						

It is evident from the contour plots that outside 3 hours from solar noon the ATS beam size begins to increase significantly, and at 4.5 hours or more from solar noon the beam begins to take on a bimodal distribution, an evident divergence of the solar energy reflected from the left and right halves of the collector. This may be caused by gravity-induced deflection of the left and right sides of the mirror module support structure relative to one another. (See section titled "Comments on Off-Noon Performance and Modeling.")

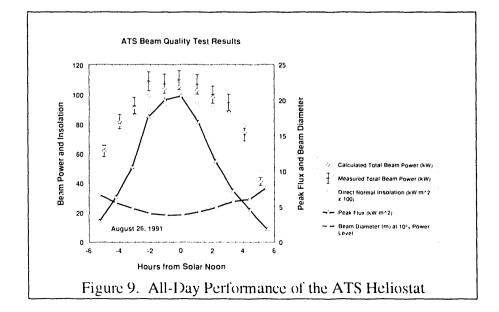
The measured values for beam power, peak flux, beam diameter, etc. for all 20 BCS measurements (of August 26) are plotted in Fig. 9. A table containing those data values is provided in Appendix B.

Table 3. Measured Beam Characteristics of ATS Heliostat							
	(Corresponds to Figure 8)						
	Hours from	Insolation	Beam	Peak	Beam		
Time	Solar Noon	$(kW/m^2)$	Power	Flux	Diameter (m)		
			(kW)	(kW/m <sup>2</sup> )			
08:30	-4.63	0.72	71	4.59	6.22		
10:03	-3.08	0.85	92	10.60	4.84		
10:32	-2.60	0.88	101	13.87	4.49		
11:28	-1.67	0.91	101	18.49	3.99		
12:30	-0.63	0.94	111	21.02	3.89		
14:10	1.03	0.93	106	16.81	4.30		
15:12	2.07	0.91	99	11.36	4.93		
16:12	3.07	0.87	94	7.25	5.69		
17:45	4.62	0.72	67	3.24	7.21		

When normalized to a one-sun level (defined as an insolation level of 1 kW/m<sup>2</sup>) the maximum total power observed for the ATS on that date was 119 kW. The highest peak flux level observed was 21.0 kW/m<sup>2</sup>, and the smallest beam diameter was 3.9 m.

## SPECO Heliostat Beam Quality Measurements

The contour plots of selected images of the SPECO heliostat beam shown in Fig. 10 reveal the changing shape of the beam over the course of the day. The increase in beam size at times away from solar noon is more pronounced for the SPECO than for the ATS. This is due only in part to the larger size ( $200 \text{ m}^2$ ) of the SPECO. Table 4 lists the actual measured parameters (beam power, peak flux, etc.) corresponding to each of the contour plots.



08:30 a.m.	10:0 <b>1</b> a.m.	11:03 a.m.				
12:04 p.m.	1:00 p.m.	2:00 p.m.				
	P					
3:00 p.m.	4:30 p.m.	6:00 p.m.				
	Figure 10. Contour Plots of SPECO Heliostat Beam. The dimensions (h x w) of each plot window are 9.0 x 10.5 m					

The increase in the beam diameter and its splitting in half is quite substantial at time distant from solar noon, and may be an indication of deflection of the SPECO's mirror module support structure (see discussion in the "Comments on Off-Noon Performance and Modeling").

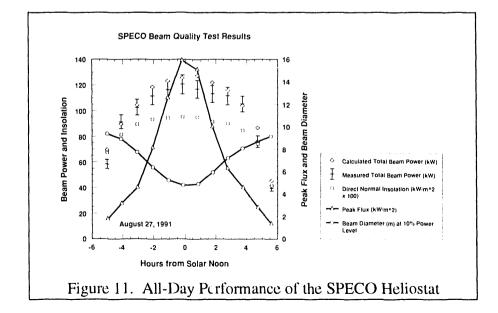
The SPECO's performance over the entire test day is depicted in a multiple-overlay plot in Fig. 11, while the corresponding data values are tabled in Appendix A.

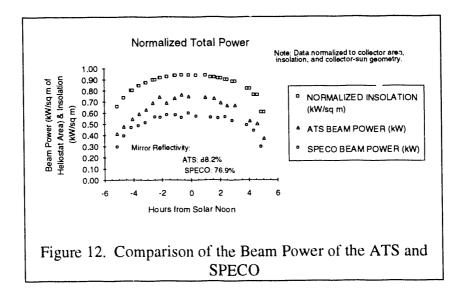
The normalized noontime total beam power of the SPECO was measured at 129 kW, and the highest peak flux (actual, not normalized) was  $16.7 \text{ kW/m}^2$ . The smallest observed beam diameter was 4.7 m.

Tal	Table 4. Measured Beam Characteristics of SPECOHeliostat (Corresponds to Fig. 11)						
Time	Time from SN	NIP (kW/m)	Beam Power (kW)	Peak Flux (kW/m <sup>2</sup> )	Beam Diameter (m)		
8:30	-4.63	0.748	76.9	2.54	9.18		
10:01	-3.12	0.888	102.2	4.60	7.72		
11:03	-2.08	0.929	111.2	8.14	6.35		
12:04	-1.07	0.941	115.9	12.52	5.22		
13:00	-0.13	0.945	119.9	15.92	4.75		
14:00	0.87	0.946	116.3	15.03	4.86		
15:00	1.87	0.912	112.3	9.91	5.90		
<u>16:30</u>	3.37	0.894	105.9	5.25	7.61		
18:00	4.87	0.736	75.2	2.79	8.56		

### Comparing the Beam Quality of the Two Heliostats

In order to compare the optical performance of the two heliostats, the test data was normalized in a manner intended to eliminate differences due to collector areas, insolation, and collector-to-sun-to-target geometry. The resulting performance data of the two heliostats is plotted together in Figs. 12 and 13. In the plots, the beam power test data has been normalized a) to  $1 \text{ m}^2$  of heliostat reflective area, b) to an average insolation level for the time of each measurement, and c) to an average cosine loss (i.e., to an average collector-to-sun-

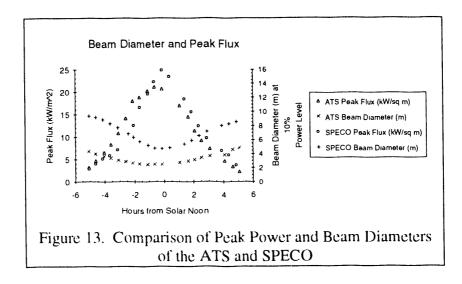




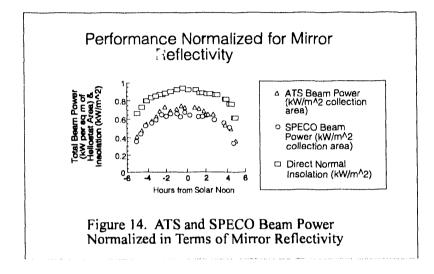
to-BCS target geometry) for the time of each measurement. The direct normal insolation (NIP) values are the time-local average of NIP measurements made on the days of the ATS and SPECO beam quality tests. The measured peak flux values were normalized in terms of those average insolation values. Finally, the beam diameter values have been plotted unchanged.

A precise statement of the normalization or treatment of the data shown in these plots is provided in Appendix E.

The intent of the normalization process was to preserve the verisimilitude of the data by retaining the effect on power collection of the reduced insolation, the higher air mass, and the poorer sun-heliostat-target geometry in the early morning and late afternoon while eliminating the effects of different heliostat collection areas, insolation levels, and heliostat-to-sun orientations.



The SPECO heliostat's optical performance was less than one might have expected for a number of reasons. First, the use of lower quality glass mirrors resulted in a solar averaged reflectivity that was 11 percentage points lower than that of the ATS mirrors (initial solar averaged reflectivity of the SPECO mirrors was 83%; ATS mirrors were 94%). To examine the performance of the heliostats in the absence of this effect, the total beam power data plotted in Fig. 12 was further normalized by dividing each heliostat's performance values by the heliostat reflectivity measured at the time of the test. The results are plotted in Fig. 14.



Second, the canting (aiming) of the SPECO mirror modules was not optimized (as were those of the ATS heliostat); a realignment of the modules would almost certainly reduce the nominal beam diameter.

Finally, another likely contributor to overall slope error in the SPECO mirror modules relates to the position of the module-mounting studs on the mirrors. These studs attach the mirror to the metal frame of the mirror module; they are adjusted to set the curvature of the mirror. There are nine studs on each mirror, spaced evenly in a rectangular pattern approximately 30 cm from the edges of the mirror. This means that about 50% of the mirror area lies to the outside of the stud pattern. A consequence of this is that the curvature obtained in the mirror surface areas to the outside of the stud pattern is less than that obtained for the interior area.

2

## Modeling the Beam Quality of the Large-Area Heliostats

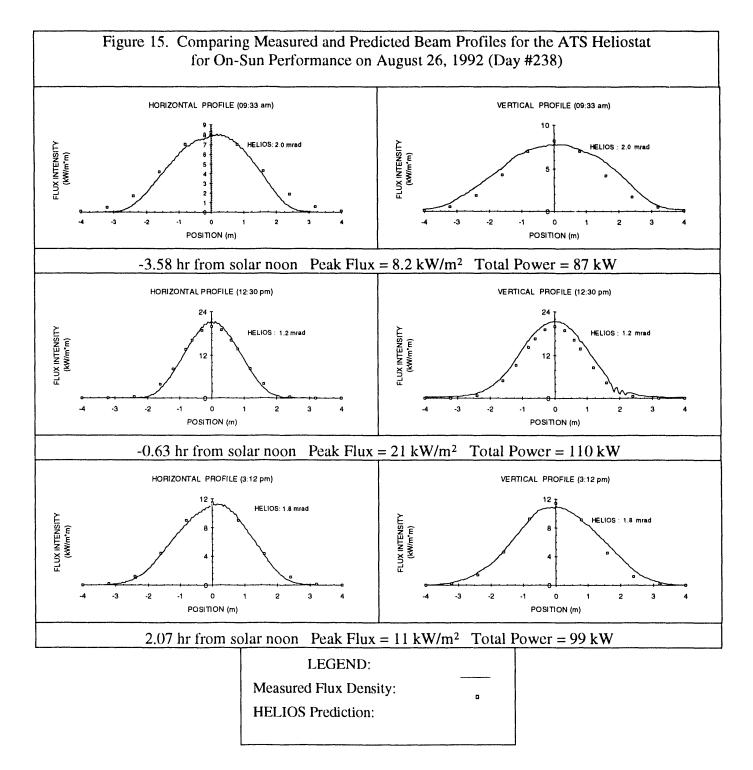
The measured performance of the large-area heliostats was compared to performance predictions generated by HELIOS[11], a computer code for modeling the optical performance of concentrators. The comparison was obtained by varying the input slope error<sup>12</sup> to the HELIOS model until the predicted measurement had been made. The results of the computer runs are plotted together with the measured beam profiles in Figs. 15 and 16. In the case of the ATS heliostat, an overall heliostat slope error of 1.2 mr (as an input parameter for the HELIOS run) was found to provide a reasonable match between the predicted and measured beam centroid profiles <u>at or near solar noon</u>. An input slope error of 2.0 mr was required in the HELIOS model to obtain a good match for the morning measurement; this is an increase in effective overall slope error over the noon time of about 0.8 mr.

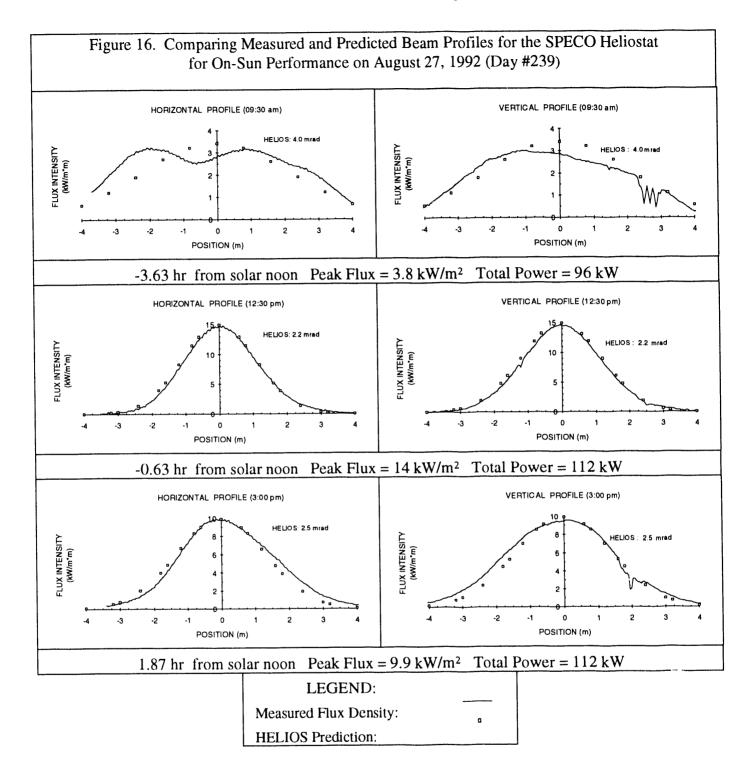
Modeling of the SPECO's performance (Fig. 16) indicated that an overall slope error of 2.2 to 2.5 mr was a reasonable estimate for the SPECO heliostat at or near the solar noon-time. Matching of the HELIOS model to the measured performance outside of solar noon is required for an input slope error of 4 mr.

## **Comments on Off-Noon Performance and Modeling**

Gravity-induced deflection is a reasonable explanation for the observed flux distribution of the heliostats (see ()8;3() a.m. contour in Fig. 8). The bimodal nature of the distribution outside of solar noon is evidence that the heliostat has effectively developed two separate aim points. Moreover, the construction of the mirror module support structures and their means of attachment to the elevation drive (see Figs. 3 and 4) suggests that gravity-induced deflection ("clam-shelling" or "opening up") of the left and right sides of the heliostat is favored when the mirror modules are in the horizontal orientation. Such deflection would result in a divergence of the beams from the left and right sides of the collector. In other words, as a heliostat's reflective surface is rotated from a vertical to a horizontal position, the left and right sides of the module support sag and the structure "opens up"; as the heliostat is lowered in elevation, the structure recovers from that deflection and "closes up." Mirror modules are canted to obtain the best beam characteristics at noontime when the heliostat's average elevation angle is approximately 34° from horizontal (for Albuquerque, New Mexico). At this elevation angle, the left and right sides of the mirror module support structure have some deflection or sagging. When the heliostat is rotated toward the vertical orientation at times outside of noon, the module's support structure recovers from the gravity-induced deflection, and the aim points of the left and right sides of the heliostat cross over one another. This explanation is consistent with the observed beam quality measurements, but tests have not been performed to verify or refute the hypothesis.

<sup>&</sup>lt;sup>12</sup>The overall slope error includes the effects of waviness (deviations of the surface on a small-scale), surface slope errors, and canting errors.





-

#### Summary/Conclusions on the Beam Quality Test Results

- On the test days discussed here, under noontime, normalized (1,000 kW/m<sup>2</sup>) conditions, the ATS and SPECO respectively delivered up to 119 and 127 kW of thermal energy. On a unit basis this is 0.79 and 0.63 kW/m<sup>2</sup> of collection area for the ATS and the SPECO, respectively.
- The smallest observed beam diameter (10% power contour line) for the ATS and the SPECO was 3.85 and 4.75 m, respectively, and the highest measured peak flux levels of the two heliostats were 21.0 and 15.9 kW/m<sup>2</sup>.
- The ATS's noontime overall slope error of 1.2 mr (as determined by matching actual performance profiles to those generated using the HELIOS computer code) meets the second-generation heliostat design requirements[1] for beam shape ("theoretical beam shape plus 1.4 mr fringe"). Its slope errors outside of solar noon (1.7 to 2.0 mr) just exceed the requirements.
- The SPECO heliostat's noontime performance of 2.2 mr falls short of the design requirement, and outside of noon its performance of 4.0 mr exceeds the design limit by a large margin.
- The SPECO heliostat's performance would be improved by further optimizing the alignment of the mirror modules.
- The positioning of the SPECO mirror-to-module mounting studs may incur additional surface slope errors, and could be evaluated as a means to improve collector performance.
- The beam quality measurements outside of solar noon for both heliostats indicate the emergence of two separate aim points. This effect is most evident for the SPECO heliostat, which for times away from solar noon substantially exceeds the specified maximum slope error. The effect may be caused by gravity-induced deflection of the left and right portions of the mirror module support structure. The acceptability of the optical performance of either heliostat would depend on the heliostat's location (slant range and orientation) as well as the dimensions of the receiver.

#### WIND EFFECTS

#### **Background and Description of Test**

The large-area heliostats were designed (to the second-generation heliostat specifications) for normal operation under windy conditions up to 27 mph [1]. In the event wind gusts exceed 27 mph, the heliostats are to be sent to a stow position that minimizes wind drag on the collector. The purpose of the wind effects tests was to evaluate the effect of the wind on beam quality for high velocities within the operational limit,

and to seek answers to a number of questions: At what wind speed levels does significant heliostat movement begin to occur? How much movement of the beam results at elevated wind speeds? What is the relationship of the angle of attack of the wind on heliostat movement? At what wind speed and for what angles of attack does the beam quality become unacceptable from an operational standpoint? Do the wind spoilers on the SPECO heliostat reduce wind drag?

The tools employed in the wind effects tests were the BCS and the meteorological tower. The BCS was used to observe and capture successive images of the heliostat beam on the BCS target while simultaneously measuring and recording wind speed and direction. The test was performed at times when there was both sunshine<sup>13</sup> and wind speeds were above 10 to 15 mph. The simultaneous occurrence of high winds and sunny conditions is fairly infrequent at the NSTTF, and, in addition, winds are predominantly from the west and southwest, which limited the recorded wind "events" to a fairly narrow range of angles of attack. Nonetheless, successful tests were carried out on several occasions, and the results answer some questions regarding wind effects.

The BCS is employed in the wind effects test to obtain a measure of heliostat beam centroid movement during the wind event.<sup>14</sup> For a 27-mph wind, the (second-generation) specifications (see Table 1) permit a maximum mirror surface deflection of 3.6 mr and a maximum pedestal twist or tilt of 1.5 mr. But the actual source of the beam deflection (be it pedestal twist or bend, deflection of the mirror support structure, or bending of the mirrors themselves) cannot be determined from heliostat beam movement alone. The wind effects test only provided a measure of overall beam deflection due to the wind. Since the expected combined effect of two independent error sources is the square root of the sum of the squares of the two errors, an approximate combined specification for wind-induced beam deflection would be 3.9 mr.<sup>15</sup>

In the wind effects test, the BCS images are used to measure the relative movement of the beam centroid and the general shape of the heliostat beam during the observed wind event. For the purposes of the test, the average position of the heliostat beam during the test is taken to be the desired aim point. This is not precisely the case, but is a reasonable approach since wind events cannot be predicted in advance, and in most cases the "actual" heliostat aim point on the BCS target has not been established prior to the onset of

<sup>&</sup>lt;sup>13</sup>Since the wind effect was measured by observing heliostat beam movement with the BCS, sunshine is a required condition for the test. Fully clear conditions, however are not essential.

<sup>&</sup>lt;sup>14</sup>A valid flux map of the heliostat beam is not possible since the wind-induced movement of the beam "smears" the flux measurements obtained from the gauge on the BCS target.

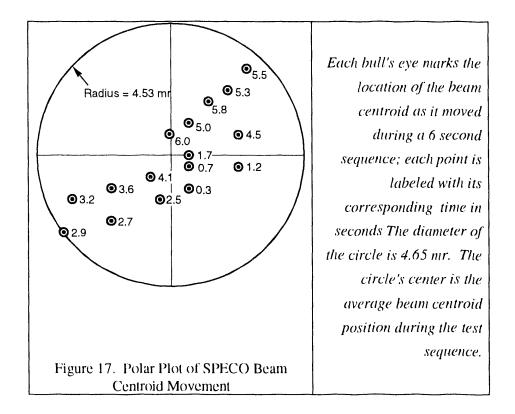
 $<sup>153.9 = \</sup>sqrt{(3.6)^2 + (1.5)^2}$ 

the test. A consequence, however, is that this test does not measure any constant offset of the average beam position during the wind event (constant offset compared to the average beam position when there is no wind).

The wind effects test was performed using the image analysis system's fastest image-capture rate (0.1 seconds per image) and taking the greatest number of images possible. Some image series were taken at full camera resolution, which resulted in 48 images and a 4.8 second time series; others were taken at half resolution, resulting in 192 images and a 19.2 second series. The digitized images of the beam and the associated wind data were then saved for later analysis.

#### **Results and Discussion**

To illustrate the wind-induced beam movement that was observed during the wind effects tests, Fig. 17 provides a polar plot of the (SPECO) heliostat beam's centroid movement. In the plot, the successive locations of the beam centroid are marked with a cross; each cross is labeled with its associated image-capture time in seconds. The beam centroid movement during a representative interval of the 20-second period of BCS image data capture can in this way be observed. The movement observed during this particular wind event was the maximum beam movement observed in all the tests. The average wind speed during the event was 18 mph, and the wind's angle of attack (AOA, the angle between the wind direction and the vector normal to the surface of the heliostat) was 150° (i.e., the wind was coming from the back of the



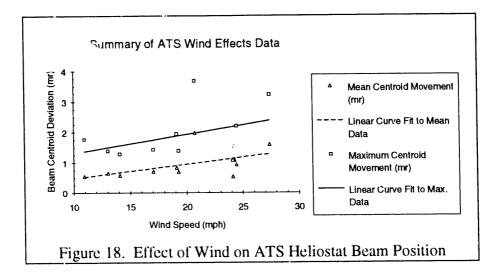
heliostat). The average movement of the beam centroid was 1.8 mr during the 20-second time interval recorded, and the maximum was 4.65 mr. (4.65 is also the radius of the circle in the polar plot.)

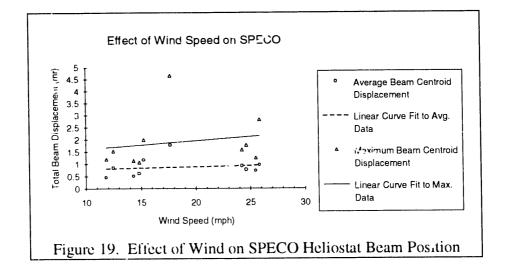
Because of the predominance of west to southwesterly winds at the test site, the wind effects tests were performed in a fairly narrow range of AOA: for the ATS heliostat, the AOA for all tests was between  $92^{\circ}$  and  $111^{\circ}$ ; for the SPECO, it was between  $104^{\circ}$  and  $150^{\circ}$ . The absence of test data for "head on" AOA (an AOA of or near  $0^{\circ}$  or  $180^{\circ}$ ) is unfortunate since one would expect to observe maximum wind drag at these angles. A description of the data obtained and a summary of the results are provided in Table 5.

Although similar trends were observed for both heliostats, the test data obtained is probably insufficient to draw firm conclusions. The wind effects data for the two heliostats is presented in Figs. 18 and 19. There is considerable scatter in the data, and although clear trends are difficult to infer, linear curve fits of the data are also presented in the figures.

Most of the wind effects data obtained were for AOA close to 90°, i.e., with the wind parallel or nearly parallel to the heliostat surface, i.e., beam deflection under "head on" AOA was not measured. The observations that were made were at angles closer to perpendicular to the collector surface. The moment coefficients for shallow angles of attack are expected to be greater and should produce a greater dynamic response. The results suggest that beam movement will remain within heliostat specifications. In an 18-mph wind, with an AOA of 150°, the SPECO heliostat's maximum beam movement of 4.7 mr did exceed the 3.9-mr (second-generation design) specification. The average beam deflection of 1.8 mr, however, was well within specified limits. In a 27-mph wind, with an AOA of 111°, the ATS heliostat's maximum beam deflection was 3.2 mr, while its average was 1.6 mr. Thus, the data that was obtained, though perhaps

Table 5. Summary of Wind Effects Test Results				
	ATS Heliostat	SPECO Heliostat		
Number of recorded events	11	10		
Range of wind speeds:	11 to 27 mph	12 to 26 mph		
Range of angles of attack:	92° to 111°	104° to 150°		
Maximum observed beam deviation:	3.7 mr (at 20.7 mph & 92° AOA)	4.7 mr (at 17.7 mph & 150° AOA)		
Average deviation of beam centroid deviation for all observations	0.92 mr	0.86		
Average maximum beam centroid deviation for all observations	1.9 mr	1.9		

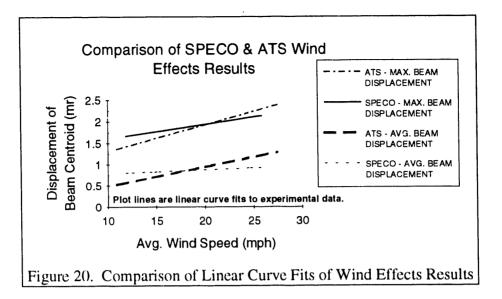




insufficient, suggests that the large-area heliostats will perform within second-generation specifications in terms of their average wind-induced beam deflection.

Figure 20 provides a side-by-side comparison of the linear curve fits of beam movement for the two heliostats. Given the great degree of scatter in the original data, the linear curve fits are not particularly meaningful, but the plot does illustrate the similarity of the performance of the two heliostats. The near-horizontal slope of the SPECO curves are intriguing given the fact that the heliostat's wind spoilers were intended to dampen out the structure's response to the wind! Although the number of observations made was limited, they suggest that the wind spoilers do dampen heliostat response at higher wind speeds.

ر س



## Conclusions

- For both large-area heliostats, the observed deflection of the collector beam in winds from 12 to 27 mph remained within specifications for the second-generation heliostat design. The average beam centroid movement was 0.9 mr, and the average maximum movement was 1.9 mr.
- The observations that were <u>made</u> suggest that the heliostats will perform within second-generation design specifications in terms of their **average** wind-induced beam deflection.

#### **Heliostat Drive and Control Systems**

## **OPERATIONAL CHARACTERISTICS OF THE DRIVE SYSTEMS**

The drive systems of the two heliostats were evaluated in terms of standard operation, power consumption, slew rates, and drive control repeatability. During all the tests, ambient temperatures were in the range of  $95^{\circ}$  to  $98^{\circ}$  F, and wind speeds were 5 to 8 mph. All of the ATS results presented here pertain to the low-cost drive.<sup>16</sup>

#### **Slew Rates**

The average slew rate (results summarized in Table 6) for the two heliostats was around .21° per second. The rate varied only slightly for the heliostats in different modes of operation (such as driving upward in elevation or combined driving in azimuth and elevation).

<sup>&</sup>lt;sup>16</sup>The original drive manufactured by A.G. Flender was replaced with the Winsmith Low-Cost drive as explained above.

Table 6. Large-Area Heliostat Slew Rates				
Slew Rates in degrees per second	ATS	SPECO		
Azimuth slew rate:	.20	.22		
Elevation down rate:	.21	.22		
Elevation up rate:	.22	.18		
Combined rates, elev. drive down:				
Azimuth:	.21	.22		
Elevation:	.21	.22		
Combined rates, elev. drive up:				
Azimuth:	.20	.21		
Elevation	.21	.18		

#### **Power Consumption**

Table 7 summarizes the results of tests to evaluate both the instantaneous power draw of the drives as well as the power consumption during a typical (10-hour) day of heliostat operation. The measurements were made with an AC watt-hour (WH) meter.<sup>17</sup> The all-day power consumption of the ATS and SPECO, respectively, was measured at 292 and 385 watt-hours. This included heliostat movement from horizontal stow to on-sun tracking of a target, tracking of target for 10 hr, and then returning to the horizontal stow position. The test demonstrates the inefficiency of worm gear drives (employed by the SPECO heliostat); however, they have the advantage of high strength, large gear reduction in a single stage, anti-backdriving capabilities, and low cost.

## **Control Drive Repeatability**

A group of tests was performed to evaluate the repeatability of the drive's control system; the results are summarized in Table 8. The first test consisted of commanding the heliostat to drive 45° in azimuth and 30° in elevation away from a starting position and then to return to the original starting position. To determine control drive repeatability, an independent measurement of the heliostat's position before and after the

Table 7. Power Utilization of the Large-Area Heliostats			
	ATS:	SPECO:	
Quiescent Power:	10 W	14 W	
Avg. Power Draw (both drives)	35 - 75 W	200 - 300 W	
Avg. Power Draw (elev. drive)	22 - 35 W	120	
Typical peak draw (both drives)	175 W	470 W	
On-sun tracking:	80 W (average)	140 W (average)	
All-day power consumption	292 WH	385 WH	

<sup>&</sup>lt;sup>17</sup>Model WH3-14, manufactured by Ohio Semitronics, Inc.

Table 8. Large-Area Heliostat Drive Control Repeatability			
	ATS:	SPECO:	
Move away & return to original position:	1.3 +/- 0.1 mr	0.5 +/- 0.1 mr	
Lose power, re-calibrate encoders, & return to original position:	0.97 +/- 0.1 mr	0.64 +/- 0.1 mr	

commanded movement was made using a laser beam. With the heliostat in its starting position, a laser mounted on the mirror support structure was aimed on a paper target positioned about 50 ft from the heliostat. The initial location of the laser beam on the paper target was marked at the outset of the test and after each "round trip" of the heliostat. Differences in the laser beam position represent repeatability error of the heliostat drive. A variation of this test involved commanding the heliostat to drive past the home position and then return to it.

A second test involved repositioning the heliostat after executing a sequence of commands that simulated power loss to the heliostats. After the simulated power loss, the heliostats were commanded to go to the "wake" position, which causes a resetting of the drive motor encoders, and then to return to the "home" or reference position. The same laser mounted on the heliostat (as described above) was used to measure the difference between the original laser beam position on a target and its position in repeated iterations of the test sequence. For this test, an average repeatability error of 0.97 mr and 0.64 mr was observed for the ATS and SPECO heliostats, respectively.

The results of these tests indicate that the SPECO's drive system achieved better repeatability, although the performance of both heliostats is acceptable in terms of second-generation design specifications.

#### LIFE-CYCLE TESTING

41114411

The drives of the two large-area heliostats were exercised in order to give them accelerated use and wear. This practice was initiated after the installation of the low-cost drive on the ATS heliostat, and the accelerated life-cycling information given here is pertinent to the low-cost drive and not to the A.G. Flender drive (see related comment on the ATS drive). The accelerated life cycling was carried out by the use of batch command files executed by the heliostat computer control program. This feature of the computer-based control system permitted the sequential, unattended execution of drive commands. A typical cycle in a batch command file would drive the heliostat through the range of movement in azimuth and elevation that it would normally experience in a day of operation at a central receiver plant. The drive would command the heliostat from its normal morning sun-tracking position all the way to its normal evening position, and then pause for several minutes to allow the drive motor to cool before returning again to the morning position.

The batch command file consisted of multiple repetitions of this sequence; by this means, a number of equivalent years of heliostat operation was obtained.

The azimuth and elevation drive-cycling was performed separately. The elevation drive was cycled during the nighttime only so as to avoid tracking the potentially hazardous heliostat beams up and down through the heliostat field during the day. The azimuth cycling was carried out at any convenient time; a result of this was that more azimuth than elevation drive life-cycles were carried out.

During the period from April 1990 to August 1991, the ATS (Winsmith low-cost drive) received an equivalent of 4.22 years<sup>18</sup> of operation of the azimuth drive and 2.98 years of operation of the elevation drive; the SPECO's (Hub City) azimuth drive received 3.36 years and its elevation drive received 2.58 years of operation.<sup>19</sup> The life-cycling information is reproduced in Table 9.

No failures of either drive system were experienced during this time.

Table 9.         Summary of Life Cycling Data for Large-Area Heliostats			
	ATS	SPECO	
Elevation Drive Cycles**	1086	943	
Azimuth Drive Cycles	1540	1226	

\*\* 1 cycle = 1 day of normal operation

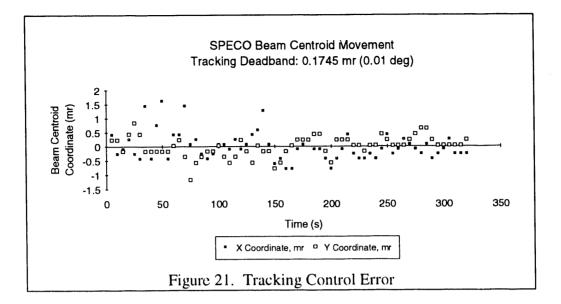
## TRACKING ACCURACY TESTS

Both the SPECO and ATS heliostats employed a prototype heliostat control system<sup>20</sup> which has a tracking deadband that is set in the software by the operator. The control system initiates movement of the heliostat drives only when the positional error exceeds the specified deadband. In an evaluation of this aspect of the tracking controls, the RMS tracking error of the azimuth and elevation drives, respectively, was measured at 0.50 and 0.35 (+/- 0.1) mr when the tracking deadband was set at its minimum value, 0.1745 mr. The test data is plotted in Fig.21.

All-day tracking error tests were performed on the two heliostats in order to measure the overall tracking error over a period of several days. Larger than expected tracking errors with the ATS prompted

<sup>&</sup>lt;sup>18</sup>A year of operation defined as 365 cycles.

<sup>&</sup>lt;sup>19</sup>This cycling of the SPECO occurred after the failure and repair of the drive unit, which is discussed below.
<sup>20</sup>See description of the prototype heliostat control system in Section 2, "Description of the Large-Area Heliostats."



investigation and led to the discovery of an intermittent malfunctioning of the ATS's elevation drive encoder. Because the tracking tests were concluded prior to the repair of the ATS encoder, the results presented here are for the SPECO heliostat.

For these tests, the BCS was employed in the test to observe and record the location of the SPECO heliostat's beam centroid at 20- to 30-minute intervals as it tracked the BCS target over a 6-day period. Differences between the observed location of the beam centroid and the coordinates of the aim-point were the measure of overall tracking error.

The all-day tracking error tests revealed systematic tracking errors that increased linearly as a function of time from solar noon. There was evidence of substantial pedestal tilt errors<sup>21</sup> which were probably a consequence of the earlier failure of the SPECO drive (described in the next section). The average magnitude of the tracking error of the SPECO during the all-day tracking tests was 1.65 (+/- 0.25) mr; a maximum error of 3.9 (+/- 0.25) mr was observed 4 hours after solar noon.

Because the prototype heliostat control system is capable of compensating for pedestal tilt error, an analysis of the all-day tracking error test data was performed in order to differentiate the linearly occurring tracking error, which could have been eliminated by inputting pedestal tilt data into the control system. When the systematic linear error of the test data was subtracted out, what remained was an RMS tracking error of  $(0.83 \text{ and } 0.28 (+/- 0.1) \text{ mr}, \text{ respectively, for the azimuth and elevation drives. This result, though not exhaustive, provides a reasonable representation of tracking error for the SPECO heliostat. The test results$ 

<sup>&</sup>lt;sup>21</sup>Pedestal tilt errors are those resulting from the pedestal not being perfectly vertical.

compare favorably with the Second Generation Heliostat Design Requirements (Table 1), which specify a maximum RMS tracking error of 1.5 mr for each drive.

#### FAILURE AND REMANUFACTURE OF THE SPECO DRIVE

In a winter storm in early 1988, the SPECO's drive unit, which was manufactured by Hub City, Inc., was damaged by winds [12]. Because of general power outages that occurred at the time of the failure, the exact time of failure and the wind speed at which the failure occurred are unknown. Likewise, it could not be determined if the failure occurred at winds above the design-specified 90-mph (stow position) survival wind.

Upon inspection, several broken teeth were discovered in the (elevation) bronze worm gear that had been cast onto the steel hub of the drive. Moreover, the bronze casting was found to have significant surface imperfections due to the casting process.

A second hub manufactured at the same time as the original was installed in the drive was then returned to Sandia. X-rays taken of the worm gear through the housing revealed similar surface imperfections in the bronze worm gear casting, but showed no large voids or other problems. During torsional load tests, one of the worm gear teeth fractured and a second one cracked. The subsequent analysis resulted in several design changes to the gear and its recasting. The worm gear was recast (at a different foundry) and the remanufactured gear passed subsequent X-ray examination and load testing, and was re-installed on the SPECO heliostat. The drive performed satisfactorily through the remainder of the test period.

## FAILURE AND MODIFICATION OF THE LOW-COST HELIOSTAT DRIVE [13]

In June 1988, a low-cost heliostat drive specifically developed for heliostats under DOE sponsorship was installed for evaluation on the ATS heliosta (this drive is described in Table 2). On January 9, 1989, during a period of severe cold, the azimuth drive unit failed structurally, resulting in the detachment of the heliostat from the pedestal.

The failure of the azimuth drive was caused by water, which seeped into the azimuth drive cavity. The water, together with the drive's lubricating grease, completely filled the cavity and then froze. The freezing process began at the outside surfaces and traveled inward, sealing off any escape route for fluid. The expansion of the water as it froze increased the internal pressure; eventually the hydraulic pressure was sufficient to shatter the outer housing of the azimuth drive, and with it the support for the outer race for the main support bearing. The entire mirror structure then tilted and caused further damage to the azimuth drive.

Careful inspection of the failed drive led to the conclusion that water penetrated the cavity around the bolts that fasten the elevation drive unit to the top of the azimuth unit. Several of these bolts were found to be loose, with evidence of associated water corrosion in the bolt holes. These bolt holes are located radially to the inside of an outer sealing o-ring. As a result, any water leaking through them would have a direct path down into the azimuth drive cavity.

A design modification was made to provide an additional sealing ring radially to the inside of that bolt circle. The use of Loctite was also proposed to insure secure tightening of the bolts. (Lockwiring of the bolts was suggested as an alternative.) A drive unit with these modification was re-installed on the ATS heliostat and operated satisfactorily (and with no failures) for the balance of the evaluation period.

# 4. SUMMARY AND CONCLUSION

## **Heliostat Costs**

• A 1988 cost estimate for the manufacture of large-area heliostats (at a rate of 2,500 per year) set the cost (1991\$) at \$141/m<sup>2</sup> of collector area.

## Mirrors

- Mirror Module Reflectivity: slight degradation of mirror reflectivity was observed over a period of 6 years in both the ATS and the SPECO mirror modules; the average degradation rate for both heliostats was 0.6 to 0.7% per year. Stabilization of the reflectivity during this time period was not observed.
- Mirror Soiling: in the environment of Albuquerque, New Mexico, mirror reflectivity was observed on the average to decrease from 6 to 9% due to module soiling when no mirror cleaning was performed and the only cleaning action was that provided naturally by rain, snow, and wind action.

## **Mirror Module Durability**

- Both heliostats' mirror modules have demonstrated durability. After five years at the New Mexico test site, small flakes of the protective paint at the edges of approximately one-quarter of the SPECO mirrors were peeling. Corrosion of the mirroring was evident at these very small and localized edge areas.
- After five years, the mechanical integrity and condition of the mirror modules were excellent: the adhesives used to bond the glass and metal parts of the module showed no signs of debonding or decomposing, and the threaded studs, hat sections, and other metal parts remained free of rust or corrosion.<sup>22</sup>
- Heliostat designers should consider the issue of snow removal from heliostat collection surfaces. Structural elements having a physical proximity to the outer edges mirror array have the potential to trap snow on the mirrors, leading to possible mirror breakage.

# **Beam Characterization**

• At times outside of solar noon, the optical performance of both collectors is hampered by the divergence of the beams from the left and right sides of the heliostat. This effect is most evident with

<sup>&</sup>lt;sup>22</sup>In fairness, it should be pointed out that Albuquerque, New Mexico is a fairly dry climate; outdoor metal structures do not in general suffer speedy rust and corrosion.

the SPECO heliostat. It may be caused by gravity-induced deflection of the heliostats, which would indicate insufficient stiffness of the mirror module support structures. The BCS test results alone are insufficient to confirm this hypothesis.

## **ATS Heliostat Beam Quality**

- During a three-hour period centered around solar noon, the average total beam power of the SPECO (normalized to an insolation level of sun or 1 kW/m<sup>2</sup>) was 115 kW; this amounts to 0.77 kW/m<sup>2</sup> of collection area. The average peak flux and beam diameter (actual measured values) during this same interval were 18.5 kW/m<sup>2</sup> and 4.1 m, respectively, with an uncertainty of +/- 6 to 10%.
- For noontime performance, the estimated overall slope error of the ATS collector (based on comparisons of the actual beam profiles with those predicted by the HELIOS code) is 1.2 mr. The expected error in this comparison of measured and modeled performance is approximately 0.5 mr. This value meets the second-generation heliostat design specification of 1.4 mr (see Table 1). During off-noon conditions the slope error increases to approximately 1.8 to 2.0 mr, a value slightly in excess of the specification. The expected error in this comparison of measured and modeled performance is approximately 0.5 mr.

## **SPECO Heliostat Beam Quality**

- During a three-hour period centered around solar noon, the average total beam power of the SPECO (normalized to an insolation level of sun or 1 kW/m<sup>2</sup>) was 124 kW; this amounts to 0.62 kW per square meter of collection area. The average peak flux and beam diameter (actual measured values) during this same interval were 13.8 kW/m<sup>2</sup> and 5.0 m, respectively, with an uncertainty of +/- 6 to 10%.
- For noontime performance, the estimated overall slope error of the SPECO collector (based on the HELIOS code) was 2.2 mr. The expected error in this comparison of measured and modeled performance is approximately 0.5 mr. This value exceeds the second-generation heliostat design specification of 1.4 mr (see Table 1). During off-noon conditions the slope error increases to approximately 4 mr, a value considerably in excess of the specification.
- The SPECO heliostat's performance would be improved by realignment of the mirror modules.
- The positioning of the SPECO mirror-to-module mounting studs may incur additional surface slope errors, and could be evaluated as a means to improve collector performance.

# Wind Effects

- In terms of their **average** wind-induced beam deflection, both heliostats appear to be structurally stiff enough to perform within specifications under design wind conditions (27 mph or less).
- For both heliostats, the average measured heliostat beam deflection in winds from 12 to 27 mph was (0.9 (+/- 0.25) mr while the average maximum movement was 1.9 (+/- 0.25) mr. Some wind-induced beam movements did exceed the 3.9-mr beam pointing error specification.
- No observations were made at wind angles of attack of (or close to) ()° or 180° (wind perpendicular to face or back of collector). This leaves unexplored an important region of the design space.

# **Operational Characteristics of the Drive Systems**

- Slew Rates: the average slew rate for both heliostats was around .21° per second, and proved to be adequate. The rate varied only slightly during different modes of operation.
- ATS Power Utilization: the ATS heliostat consumed 292 watt-hours during a typical day of operation involving 10 hours of on-sun tracking. The average power draw of the drives was found to be 35 to 75 W for simultaneous operation of the drives, and 22 to 35 W for operation of the elevation drive only.
- SPECO Power Utilization: the SPECO's all-day power consumption was 385 W; the average power for operation of both drives was 200 W, and about 120 W for operation of the elevation drive only.

# **Drive Control Repeatability**

- In several tests measuring the ability of the heliostat drive control system to drive off of and then return to a specific position, an average repeatability error of 0.5 and 1.3 mr was measured for the SPECO and ATS, respectively. (Measurement uncertainty was +/- 0.1 mr.)
- Tests involving the return of the heliostat to a specified position following the (simulated) loss of power resulted in repeatability measures of 0.97 and 0.64 mr for SPECO and ATS, respectively. (Measurement uncertainty was +/- 0.1 mr.)

# Life-Cycle Testing

• The drive systems of both heliostats survived life-cycle testing without failures. The elevation drives of the ATS and SPECO, respectively, received 3.0 and 2.6 years of life-cycling, while their azimuth drives received 4.2 and 3.4 years, respectively.

## **All-Day Tracking**

- Tracking error data was obtained for the SPECO heliostat, but not for the ATS heliostat (which was equipped with the low-cost drive).
- The tracking error of the prototype heliostat control system was determined experimentally to be 0.50 and 0.35 (+/-0.1) mr, respectively for the azimuth and elevation drives when the control system's tracking deadband was set at the minimum permissible value, 0.1745 mr.
- Based on analysis of all-day tracking test data, the overall RMS tracking error of the azimuth and elevation drives, respectively, was found to be 0.83 and 0.28 (+/-0.15) mr, and compares favorably with the Second Generation Heliostat Design Requirements (Table 1) which specify a maximum RMS tracking error of 1.5 mr for each drive.

## Failure and Remanufacture of the SPECO Drive

- The failure of the SPECO drive led to the discovery of design and manufacturing faults in the bronzecast worm gear. A detailed report on this is given by Grossman [12].
- After its redesign and manufacture, the SPECO drive performed well and experienced no further failures.

## Failure and Redesign of the Low-Cost Heliostat Drive

• A failure of the Peerless-Winsmith drive in June 1988 led to the discovery of a design flaw that permitted the penetration of water into the azimuth drive cavity. After modifications to the design, a new drive unit was manufactured and operated satisfactorily for the balance of the evaluation period.

## 5. REFERENCES

- 1. C.L. Mavis, A Description and Assessment of Heliostat Technology, SAND87-8025 Sandia National Laboratories, California, 1987.
- 2. Sandia National Laboratories California, Prototype/Second Generation Heliostat Evaluation and Recommendations, Executive Summary, 1978.
- 3. Martin Marietta Corporation, Second Generation Heliostat Optimization Studies, SAND82-8175, Sandia National Laboratories, 1982.
- 4. McDonnell Douglas Astronautics Company, *Optimization of the Second Generation Heliostat and Specification*, SAND82-8181, Sandia National Laboratories, 1982.
- 5. Personal communication from D. Gorman Advanced Thermal Systems, Inc., to J.W. Strachan and R.M. Houser, June-August, 1992.
- 6. Winsmith, Development of a Low-Cost Drive Tracking Mechanism for Solar-Heliostats or PV Arrays, contract report for Sandia National Laboratories, Document No. 90-5753, 1989.
- 7. Heliostat Development Division, *Second Generation Heliostat Evaluation Executive Summary*, SAND81-8033, Sandia National Laboratories, 1982.
- 8. PG&E, RD&D Solar Central Receiver Technology Advancement for Electric Utility Applications, Phase 1 Topical Report, Vol. 1, Report #007.2-88.2, 1988.
- 9. J. W. Strachan, *Revisiting the BCS, a Measurement System for Characterizing the Optics of Solar Collectors*, ISA/93 Conference Proceedings, Albuquerque, NM, 1993.
- R. B. Pettit, J. M. Freese, A. R. Mahoney, "The Development of a Portable Specular Reflectometer for Monitoring Solar Mirror Materials, Optical Materials and Process Technology for Energy Efficiency and Solar Applications," *International Society for Optical Engineering*, Vol. 428, pp. 125-134, 1983.
- 11. F. Biggs and C.N. Vittitoe, *The HELIOS Model for the Optical Behavior of Reflecting Solar Concentrators*, SAND76-0347, Sandia National Laboratories, 1979.
- 12. J.W. Grossman, *Static Load Testing of a Heliostat Drive*, SAND90-2624, Sandia National Laboratories, 1991.
- 13. Correspondence dated February 27, 1989, from Werner H. Heller, Peerless-Winsmith, Inc. to J.W. Grossman, Sandia National Laboratories, Albuquerque, included a three-page report, "Evaluation of Failed Azimuth Drive Unit."
- 14. Bevington, P.R., *Data Reduction and Error Analysis for the Physical Sciences*, McGraw-Hill Book Company, New York, NY, 1965, p. 61.

## 6. APPENDICES

# Appendix A: ATS All-Day Beam Quality Test Data

Table 10 contains the original measured test parameters obtained during the all-day beam quality test of the ATS heliostat on August 26, 1992. The Beam Characterization System was employed in the tests, and the approximate measurement uncertainty (average measurement error) of that system is 6 to 8% for the total beam power and peak flux measures, 2% for the beam diameter and for the relative flux distribution, and 1% for the normal incidence pyroheliometer (NIP) measure.

Table 10. ATS All-Day Beam Quality Test Results						
		enseenteene	ay: August 2		I	
	Hr from	NIP	Total Bean	n Pwr (kW):	Effective	Peak
Time	Solar Noon	(kW/m <sup>2</sup> )	Cale.	Meas'd.	Beam Dia.	Flux
					<u>(P:)</u>	(kW/m <sup>2</sup> )
08:02 a.m.	-5.10	0.645	61.9	61.7	6.78	3.11
08:30 a.m.	-4.63	0.723	71.2	71.1	6.22	4.59
09:02 a.m.	-4.10	0.784	81.2	81.5	5.66	6.46
09:30 a.m.	-3.58	0.828	86.9	87.9	5.27	8.19
10:01 a.m.	-3.08	0.852	9(),9	92.4	4.84	10.60
10:30 a.m.	-2.60	0.881	95.9	101.1	4.49	13.87
11:00 a.m.	-2.10	0.902	99.3	108.8	4.16	17.36
11:28 a.m.	-1.67	0.910	101.4	100.6	3.99	18.49
12:00 p.m.	-1.1()	0.928	103.4	106.8	3.85	20.04
12:30 p.m.	-0.63	0.937	104.4	111.3	3.89	21.02
1:00 p.m.	-(),1()	0.936	104.3	109.3	3.85	20.59
2:08 p.m.	1.03	0.929	102.3	106.3	4.30	16.81
2:43 p.m.	1.62	0.906	98.6	103.7	4.63	14.20
3:09 p.m.	2.07	0.910	97.9	99.4	4.93	11.36
3:39 p.m.	2.55	0.891	93.5	94.4	5.35	9,09
4:09 p.m.	3.07	().87()	86.6	93.8	5.69	7.25
5:11 p.m.	4.08	().792	73.9	72.2	6.01	4.52
5:41 p.m.	4.62	().72()	64.6	66.7	7.21	3.24
6:09 p.m.	5.07	().479	41.2	40.7	7.54	1.79
6:39 p.m.	5.55	0.533	44.5	45.7	7.95	1.81

## Appendix B: SPECO Ali-Day Beam Quality Test Data

- 44

Table 11 contains the original measured test parameters obtained during the all-day beam quality test of the SPECO heliostat on August 27, 1992. The Beam Characterization System was employed in the tests and the approximate measurement uncertainty (average measurement error) of that system is 6 to 8% for the total beam power and peak flux measures, 2% for the beam diameter and for the relative flux distribution, and 1% for the normal incidence pyroheliometer (NIP) measure.

Table 11. SPECO All-Day Beam Quality Test Results						
Test Day: August 27, 1991						
		NIP	Total Beam	Pwr (kW):	Effective	Peak
Time	Solar Noon	(kW/sq m)	Calc.	Meas'd.	Beam Dia.	Flux
					(m)	(kW/m <sup>2</sup> )
8:03	-5.08	().674	69.4	58.3	9.38	1.82
8:30	-4.63	0.748	79.1	76.9	9.18	2.54
9:00	-4.1.	0.811	89.2	91.2	8.89	3.25
9:30	-3.63	0.856	96.9	95.8	8.33	3.75
10:01	-3.12	0.888	1()4.()	102.2	7.72	4.60
10:42	-2.43	0.913	109.3	112.4	6.85	6.93
11:03	-2.08	().929	118.1	111.2	6.35	8.14
11:30	-1.63	0.938	121.5	116.0	5.76	10.26
12:04	-1.07	0.941	123.2	115.9	5.22	12.52
12:30	-0.63	0.944	125.8	112.1	4.79	14.30
13:00	-0.13	0.945	125.3	119.9	4.75	15.92
13:30	0.37	0.949	127.3	122.7	4.7()	16.72
14:00	0.87	0.946	126.8	116.3	4.86	15.03
14:30	1.37	0.934	123.9	114.8	5.40	11.93
15:00	1.87	().912	120.9	112.3	5.90	9.91
15:30	2.37	0.911	119.4	116.6	6.55	7.96
16:01	2.88	().897	115.1	110.7	7.16	6.23
16:30	3.37	0,894	112.5	105.9	7.61	5.25
17:01	3.88	0.842	103.4	104.5	8.04	4.61
17:30	4.37	0.8	95.7	96.7	8.29	4.00
18:00	4.87	0.736	85.6	75.2	8.56	2.79
18:30	5.37	0.599	67.3	63.0	8.85	2.12
18:54	5.77	0.435	47.3	40.1	8.71	1.34

# Appendix C: Normalization Procedure for Comparison of Beam Quality Test Results

Figures 12 and 13 of this report provide a comparison of the all-day beam quality performance of the two heliostats. The plots are based on the original test data, but the data has been normalized and otherwise treated in order to eliminate performance differences due to differences in insolation and heliostat-to-sun geometry at the time of each separate measurement as well as the difference in collection area of the two heliostats.

As stated in the text of the report, the intent of these normalizations was to preserve the verisimilitude of the data by retaining the effect on power collection of the reduced insolation, the higher air mass, and the poorer sun-heliostat-target geometry in the early morning and late afternoon while eliminating the effects of different heliostat collection areas, insolation levels, and heliostat-to-sun orientations.

The total beam power values plotted in Fig. 12 adjusted the original power values of each heliostat as follows:

PN = (PH / AH) (NIPAVG / NIPH) (COSAVG / COSH).(1)

In Eq. (1),  $P_N$  is the normalized power (the value to be plotted in the comparison chart),  $P_H$  is the actual measured power, and  $A_H$  is the collector area. NIP<sub>AVG</sub> and NIP<sub>H</sub> are, respectively, the insolation at the time of the BCS measurement and the average insolation for BCS measurements made during that (10-15 minute) interval of the day. The term COS refers to the cosine loss, which is merely the cosine of the angle between the normal to the heliostat surface and the sun vector.  $COS_H$  is the actual cosine loss for the heliostat in question at the time of the BCS measurement, while  $COS_{AVG}$  is the average of the cosine losses for those measurement made during that interval of the day.

The peak flux measurements for the two heliostats were adjusted in terms of input insolation by normalizing them to an average insolation value for that time of the day:

$$PF_{N} = PF_{H} (NIP_{AVG} / NIP_{H}).$$
<sup>(2)</sup>

 $PF_{H}$  and  $PF_{N}$  are the heliostat's measured and normalized peak flux values, respectively, and  $NIP_{AVG}$  and  $NIP_{H}$  have the same meaning as in Eq. (1).

In order to eliminate the performance difference because of differences in the reflectivity of the ATS and SPECO mirror modules, the beam quality data was normalized in terms of mirror reflectivity. The resulting data are compared in a plot in Fig. 14; the beam power data in that plot were normalized as follows:

$$P_{N}' = P_{N} / R_{H}$$
(3)

In Eq. (3)  $P_N'$  is the total beam power normalized for reflectivity, while  $P_N$  is the beam power previously normalized as described in Eq. (1).  $R_H$  is the reflectivity of the heliostat in question, while  $P_N'$  is the beam power further normalized to the ATS reflectivity. The reflectivity of the ATS measured at the time of this test is  $R_{ATS}$ . That of the SPECO is  $R_{SPECO}$ .

.

# Appendix D: Analysis of Measurement Error in the Beam Characterization System

The individual elements of the BCS flux map array may be in error either in terms of the flux density value or the physical dimension and/or position of the pixel. Pixel geometry (pixel height and width) in the BCS image is easily corrected for the lengthening or foreshortening due to the non-normal flux incident angle on the target or the capture angle of the camera relative to the same. But based on experience, the use of high-quality lenses ensures that any physical distortion of the image due to the lens assembly is small enough to be ignored.

Given the BCS components currently in use, the resolution of the flux measurement is 1 part in 256 (8 computer bits) or about 0.5% of the full-scale intensity level in the image. The system's geometric resolution (the size and number of picture elements) is dependent on the camera sensor and the digitizer, and, of course, the distance from the camera to the target. The picture array size for common digital imaging systems is in the range from a 250 x 250 to a 1000 x 1000 array. The uncertainties associated with the flux density measurement are not small enough to ignore, and their analysis is the subject of this section.

Consider a small area of the flux target corresponding exactly to a single picture element of the BCS image. Let  $\mathbf{F}$  be the average flux density over that area; we'll also refer to the "flux bundle" from the collector that is incident on that area at the moment of the measurement. The aim of the analysis is to establish the uncertainty associated with the BCS measure of  $\mathbf{F}$ .

**F** is processed as an input signal by the measurement system, and the output (from the image digitizer) is a pixel level which we'll call **P**. In a manner of speaking, during its passage through the BCS, **F** is transformed by each successive element: the target, the neutral density (ND) filters, the lens assembly, the camera sensor, and finally, the digitizer (frame grabber). The validity of the measurement process requires that each successive transformation be linear and constant over both the full flux range and over the entire target. Thus, the collector flux reflected from a pixel-sized area on the target toward the camera must be proportional to the collector flux incident on that area. Let us call this constant (output/input) ratio  $\mathbf{k}_{\rm T}$  (T for target!);  $\sigma_{\mathbf{k}_{\rm T}}$  is the standard deviation (for all the pixels in that image) of  $\mathbf{k}_{\rm T}$ , and is the expected error or deviation of the ratio from pixel to pixel. We define similar terms for each succeeding "in-line" element in the measurement:  $\mathbf{k}_{\rm ND}$  for the ND filters,  $\mathbf{k}_{\rm L}$  for the lens assembly,  $\mathbf{k}_{\rm C}$  for the camera sensor, and  $\mathbf{k}_{\rm D}$  for the population of pixels in the image. We can employ these terms to express the relationship between the flux density, **F**, and the pixel level, **P**:

$$\mathbf{P} = \mathbf{k}_{\mathrm{D}} \cdot \mathbf{k}_{\mathrm{C}} \cdot \mathbf{k}_{\mathrm{L}} \cdot \mathbf{k}_{\mathrm{ND}} \cdot \mathbf{k}_{\mathrm{T}} \cdot \mathbf{F}.$$
 (1)

And, assuming that the errors in each of the  $\mathbf{k}$  terms are independent, we can express the error [14] associated with  $\mathbf{P}$  as

$$\left(\sigma_{\rm P}/{\rm P}\right)^2 = \left(\sigma_{\rm k_{\rm D}}/{\rm k_{\rm D}}\right)^2 + \left(\sigma_{\rm k_{\rm C}}/{\rm k_{\rm C}}\right)^2 + \left(\sigma_{\rm k_{\rm L}}/{\rm k_{\rm L}}\right)^2 + \left(\sigma_{\rm k_{\rm ND}}/{\rm k_{\rm ND}}\right)^2 + \left(\sigma_{\rm k_{\rm T}}/{\rm k_{\rm T}}\right)^2 \tag{2}$$

In this expression for measurement uncertainty, the ratio  $(\sigma_P/P)$  gives the expected error as a percent (or fraction) of the full-scale value of **P**.

Determining pixel intensity error by this means requires information about the error of each of the BCS components. In practice, an alternative, experimental measure of relative pixel intensity error is possible by evaluating the uniformity of a BCS image of the flux target while it is illuminated uniformly. If the target is uniformly illuminated, any non-uniformities in the resulting digitized image are a measure of system measurement error.

The pixel-level-to-flux-density conversion factor for the BCS image is the ratio of the flux gauge measurement ( $\mathbf{F}_{FG}$ ) and the pixel level,  $\mathbf{P}_{FG}$ , at the location of the flux gauge

$$\mathbf{F} = (\mathbf{F}_{\mathrm{FG}} / \mathbf{P}_{\mathrm{FG}}) \cdot \mathbf{P}. \tag{3}$$

These two values also introduce error into the measurement. Since these errors are independent of one another, the uncertainty in  $\mathbf{F}$  can be expressed in terms of  $\mathbf{F}_{FG}$ ,  $\mathbf{P}_{FG}$ , and  $\mathbf{P}$ :

$$\left(\sigma_{\rm F}/{\rm F}\right)^2 = \left(\sigma_{\rm F_{\rm BS}}/{\rm F_{\rm FG}}\right)^2 + \left(\sigma_{\rm P_{\rm BS}}/{\rm P_{\rm FG}}\right)^2 + \left(\sigma_{\rm P}/{\rm P}\right)^2. \tag{4}$$

The accuracies of the BCS flux measurement are listed in Table 12. The estimated error for individual heliostat measurements is 10% (of the peak flux level in the flux map), and for groups of heliostats as well as for point-focus collectors it is 6%. In each case, the flux gauge is the source of largest error and lies with the large uncertainty associated with the calibration of the gauges. The calibration uncertainty is greatest for the gauges used for single heliostat measurements and is the focus of attention in current efforts to improve the BCS. The larger error associated with estimating the level of the flux gauge pixel for the dish concentrator BCS is due to the fact that because of the small size of the collector beam the flux gauge produces a larger perturbation in the image. In addition, the flux gradient across the gauge is also usually fairly steep, which increases the uncertainty of the flux measurement itself.

Table 12. Estimates of BCS Error for Heliostats and Dish Concentrators Measurements			
Source of Measurement Uncertainty	BCS System and Component Measurement Uncertainty		
	Dish Concentrators	Single Heliostats:	Heliostat Groups:
Pixel Level, $(\sigma_P/P)$	< 1%	2%	1 %
Flux gauge, $(\sigma_{F_{RG}}/F_{FG})$	5%	10%	5 %
Estimate of Pixel Level at Flux Gauge: $(\sigma_{P_{Ri}}/P_{FG})$	3%	1%	1 %
Overall Uncertainty, $(\sigma_F/F)$	6 %	10 %	6 %

\_

\_

-

# Appendix E: Determining the Solar Average Specular Reflectivity of a Collector Mirror

The Device & Services (D & S) Portable Specular Reflectometer employs 660-nm wavelength light as its light source for measurements, so collector mirror measurements made with the device must be adjusted to obtain an estimate of the mirror's reflectivity over the solar spectrum.

To do this requires prior knowledge of the following:

- 1. The solar average total hemispherical reflectivity,  $R_{2\pi,S.A.}$  of the mirror material which is its average reflectivity over the 300- to 2500-nm range of the spectrum, a range containing more than 90% of the available energy. This is determined from a laboratory test or from the literature.
- 2. The hemispherical reflectivity of the same sample for 660-nm light,  $R_{2\pi,660nm}$ .

These two measures provide a means of determining the difference between the mirror sample's reflectivity over the solar spectrum to its reflectivity at 660 nm.

The specular reflectivity measurements,  $R_{15nr,660nm}$ , which are obtained (in the field) with the portable D&S reflectometer and using a 15-mr aperture are then adjusted as follows to obtain the solar averaged specular (15 mr aperture) reflectivity,  $R_{15nr,S,A}$ , of the mirror module:

$$R_{15mr,S,A} = R_{2p,S,A} - (R_{2p,660mr} - R_{15mr,660mr})$$
(1)

#### **DISTRIBUTION:**

U.S. Department of Energy (5) Forrestal Building Code CE-132 1000 Independence Avenue, SW Washington, DC 20585 Attn: G. Burch S. Gronich

U.S. Department of Energy (2) Forrestal Building Code CE-13 1000 Independence Avenue, SW Washington, DC 20585 Attn: R. Annan

U.S. Department of Energy (2) Albuquerque Operations Office P.O. Box 5400 Albuquerque, NM 87115 Attn: G. Tennyson N. Lackey

U.S. Department of Energy San Francisco Operations Office 1333 Broadway Oakland, CA 94612 Attn: R. Hughey

Acurex Corporation (2) 555 Clyde Avenue Mountain View, CA 94039 Attn: J. Schaefer H. Dehne

Advanced Thermal Systems 7600 East Arapahoe Suite 319 Englewood, CO 80112 Attn: D. Gorman

Agua y Energia Electrica Sociedad del Estado Attn: Grupo - Fuentes Alternas Gabato 3713 1826 Remedios de Escalada Buenos Aires, ARGENTINA Allegheny Ludlum Steel (2) Market and Product Development Alabama and Pacific Avenues Brackenridge, PA 15014 Attn: J.M. Hunt J.P. Ziemianski

Allegheny Ludlum Steel 80 Valley St. Wallingford, CT 06492 Attn: J.J. Halpin

Analysis Review & Critique 6503 81st St. Cabin John, MD 20818 Attn: C. LaPorta

Arizona Public Service Company P.O. Box 53999 Phoenix, AZ 85072 Attn: J. McGuirk

Arizona Solar Energy Office Dept. of Commerce 3800 N. Central Phoenix, AZ 85012 Attn: F. Mancini

Australian National University Department of Engineering Physics P.O. Box 4 Canberra ACT 2600 AUSTRALIA Attn: S. Kaneff

Barber-Nichols Engineering 6325 West 55th Avenue Arvada, CO 80002 Attn: R. Barber Battelle Pacific Northwest Laboratory (2) P.O. Box 999 Richland, WA 99352 Attn: D. Brown

BDM Coporation 1801 Randolph Road, S.E. Albuquerque, NM 87106 Attn: W. Schwinkendorf

Bechtel National, Inc. (4) 50 Beale Street 50/15 D8 P.O. Box 3965 San Francisco, CA 94106 Attn: P. DeLaquil B. Kelly S. Nickovich J. Egan

Bomin Solar Industriestr. 8 D-7850 Lorrach GERMANY Attn: Dr. Hans Jurgen Kleinwachter

Black & Veatch Consulting Engineers (2) P.O. Box 8405 Kansas City, MO 64114 Attn: J. C. Grosskreutz

Tom Brumleve 1512 Northgate Road Walnut Creek, CA 94598

California Energy Commission 1516 Ninth Street, M-S 43 Sacramento, CA 95814 Attn: A. Jenkins

California Polytechnic University Dept. of Mechanical Engineering 3801 West Temple Avenue Pomona, CA 91768 Attn: W. Stine California Public Utilities Com. Resource Branch, Room 5198 455 Golden Gate Avenue San Francisco, CA 94102 Attn: T. Thompson

Central and Southwest Services Mail Stop 7RES 1616 Woodall Rogers Freeway Dallas, TX 75202 Attn: E.L. Gastineau

Centro Investigaciones Energeticas (4) Medioambientales y Technologicas Instituto de Energias Renovables Avda. Complutense, 22 28040 Madrid, SPAIN Attn: M. Macias M. Romero E. Conejero J.M. Figarola

Clever Fellows Innovation Consortium, Inc. R.D. 1, Box 410, River Road Melrose, NY 12121 Attn: J. A. Corey, P.E.

Cummins Power Generation (2) MC 60125 P.O. Box 3005 Columbus, IN 47202-3005 Attn: R. Kubo

Cummins Power Generation South 150 Tannehill Drive Abilene, TX 79602 Attn: M. McGlaun

Dan Ka Products, Inc. 3905 South Mariposa Englewood, CO 80110 Attn: D. Sallis

Dynatherm Corporation 1 Beaver Court P.O. Box 398 Cockeysville, MD 21030 Attn: Dr. W. Bienert D. Wolf DLR Pfaffenwaldring 38-40 7000 Stuttgart 80 GERMANY Attn: R. Buck

DLR (3) Linder Hohe 5000 Koln 90 GERMANY Attn: M. Becker W. Meinecke M. Bohmer

Electric Power Research Institute P.O. Box 10412 Palo Alto, CA 94303 Attn: J. Schaeffer

El Paso Electric Company 303 N. Oregon, 10th Floor P.O. Box 982 El Paso, TX 79960 Attn: D. Gutierrez

Energy Research Centre (2) R.S. Phy. Sc. Australian National University Canberra ACT 2601 AUSTRALIA Attn: K. Inall

Energy Technology Engr. Center Rockwell International Corp. P.O. Box 1449 Canoga Park, CA 91304 Attn: W. Bigelow

Engineering Perspectives 20 19th Ave. San Francisco, CA 94121 Attn: J. Doyle

ENTECH, Inc. P.O. Box 612246 DFW Airport, TX 75261 Attn: R. Walters Flachglas Solartechnik GmbH Muhlengasse 7 D-5000 Koln 1 FEDERAL REPUBLIC OF GERMANY Attn: J. Benemann

Flachglas Solartechnik GmbH (2) Theodor-Heuss-Ring 1 5000 Koln 1 Germany Attn: Dr. M. Geyer I. Susemihl

Florida Solar Energy Center 300 State Road, Suite 401 Cape Canaveral, FL 32920 Attn: Library

Ford Aerospace Ford Road Newport Beach, CA 92663 Attn: R. Babbe

Foster Wheeler Solar Development Corporation (2) 12 Peach Tree Hill Road Livingston, NJ 07039 Attn: M. Garber R. Zoschak

Garrett Turbine Engine Co. 111 South 34th Street P.O. Box 5217 Phoenix, AZ 85010 Attn: E. Strain

LUZ International (2) 924 Westwood Blvd. Los Angeles, CA 90024 Attn: D. Kearney

Georgia Institute of Technology GTRI/EMSL Solar Site Atlanta, GA 30332 Attn: T. Brown Georgia Power (2) 7 Solar Circle Shenandoah, GA 30265 Attn: W. King

Harris Corporation (2) Government Aerospace Systems Division P.O. Box 9400 Melbourne, FL 32902 Attn: K. Schumacher

Industrial Solar Technologies (4) 5775 West 52nd Avenue Denver, CO 80212 Attn: R. Gee K. May

Institute of Gas Technology 34245 State Street Chicago, IL 60616 Attn: Library

Interatom GmbH (2) P.O. Box D-5060 Bergisch-Gladbach FEDERAL REPUBLIC OF GERMANY Attn: M. Kiera

Jet Propulsion Laboratory 4800 Oak Grove Drive Pasadena, CA 91109 Attn: M. Alper

Lawrence Berkeley Laboratory MS 90-2024 One Cyclotron Road Berkeley, CA 94720 Attn: A. Hunt

Leo Gutierrez 434 School Street Livermore, CA 94550 Los Angeles Dept. of Water and Power Alternate Energy Systems 111 North Hope Street, Room 661A Los Angeles, CA 90012 Attn: B. Engels

McDonnell-Douglas Astronautics Company (3) 5301 Bolsa Avenue Huntington Beach, CA 92647 Attn: R. L. Gervais D. Steinmeyer J. Rogan

Mechanical Technology, Inc. (2) 968 Albany Shaker Road Latham, NY 12110 Attn: G. Dochat J. Wagner

Meridian Corporation 4300 King Street, Suite 400 Alexandria, VA 22302-1508 Attn: D. Kumar

NASA Lewis Research Center (4) 21000 Brook Park Road Cleveland, OH 44135 Attn: R. Shaltens J. Schrieber

National Renewable(6)Energy Laboratory1617 Cole BoulevardGolden, CO 80401Attn:T. WilliamsG. JorgensenT. WendelinA. LewandowskiM. Bohn

Nevada Power Co. P.O. Box 230 Las Vegas, NV 89151 Attn: Mark Shank

New Mexico Solar Energy Institute New Mexico State University Box 3SOL Las Cruces, NM 88003 Northern Research and Engineering Corp. 39 Olympia Avenue Woburn, MA 01801-2073 Attn: J. Kesseli

Pacific Gas and Electric Company (2) 3400 Crow Canyon Road San Ramon, CA 94526 Attn: G. Braun J. Iannucci

Power Kinetics, Inc. (2) 415 River Street Troy, NY 12180-2822 Attn: W. E. Rogers

Renewable Energy Institute 1001 Connecticut Ave. NW Suite 719 Washington, DC 20036 Attn: K. Porter

Rocketdyne Division 6633 Canoga Park Ave. Canoga Park, CA 91304 Attn: W. Marlatt

Sacramento Municipal Utility District Generation Systems Planning Power Systems Department 6201 S Street P.O. Box 15830 Sacramento, CA 96852-1830 Attn: R.P. Wichert, P.E.

San Diego Gas and Electric Company P.O. Box 1831 San Diego, CA 92112 Attn: R. Figueroa

Schlaich, Bergermann & Partner Hohenzollernstr. 1 D - 7000 Stuttgart 1 GERMANY Attn: W. Schiel Science Applications (2) International Corporation 10343 Roselle Street, Suite G San Diego, CA 92121 Attn: K. Beninga

Science Applications International Corporation Mail Stop 32 10260 Campus Point Court San Diego, CA 92121 Attn: B. Butler

Sci-Tech International Advanced Alternative Energy Solutions 5673 W. Las Positas Blvd. Suite 205 P.O. Box 5246 Pleasanton, CA 84566 Attn: U. Ortabasi

Solar Energy Industries Assoc. (2) 777 North Capitol St. NE, Suite 805 Washington, DC 20002 Attn: S. Sklar K. Sheinkopf

Solar Kinetics, Inc. (2) P.O. Box 540636 Dallas, TX 75354-0636 Attn: J.A. Hutchison P. Schertz

Solar Power Engineering Company (10) P.O. Box 91 Morrison, CO 80465 Attn: H. Wroton D. Bielenberg (9)

Solar Steam P.O. Box 32 Fox Island, WA 98333 Attn: D. Wood

South Coast AQMD 9150 Flair Drive El Monte, CA 91731 Attn: L. Watkins Southern California Edison (3) P.O. Box 800 Rosemead, CA 91770 Attn: P. Skvarna

SRS Technologies 990 Explorer Blvd. NW Huntsville, AL 35806 Attn: R. Bradford

Stearns Catalytic Corporation P.O. Box 5888 Denver, CO 80217 Attn: T.E. Olson

Stirling Machine World 1823 Hummingbird Court West Richland, WA 99352-9542 Attn: B. Ross

Stirling Technology Company (2) 2952 George Washington Way Richland, WA 99352 Attn: M.A. White

Stirling Thermal Motors (2) 275 Metty Drive Ann Arbor, MI 48103 Attn: L. Johansson

Stone and Webster Engineering Corp. P.O. Box 1214 Boston, MA 02107 Attn: R.W. Kuhr

Sulzer Bros, Ltd. New Technologies CH-8401 Winterthur SWITZERLAND Attn: H. Fricker, Manager Sunpower, Inc. 6 Byard Street Athens, OH 45701 Attn: W. Beale

Thermacore, Inc. (2) 780 Eden Road Lancaster, PA 17601 Attn: D. Ernst

Tom Tracey 6922 South Adams Way Littleton, CO 80122

United Solar Tech, Inc. 3434 Martin Way Olympia, WA 98506 Attn: R. Kelley

University of Arizona Engineering Experimental Station Harvil Bldg., Room 151-D Tucson, AZ 85721 Attn: D. Osborne

University of Chicago Enrico Fermi Institute 5640 Ellis Avenue Chicago, IL 60637 Attn: J. O'Gallagher

University of Houston Solar Energy Laboratory 4800 Calhoun Houston, TX 77704 Attn: L. Vant-Hull J. Richardson

University of Minnesota (2) Dept. of Mechanical Engineering 111 Church St., SE Minneapolis, MN 55455 Attn: E. A. Fletcher T. Simmon University of New Mexico (2) Dept. of Mechanical Engineering Albuquerque, NM 87131 Attn: M. W. Wilden W. A. Gross

University of Nevada at Las Vegas Mechanical Engineering Dept. Las Vegas, NV 89154 Attn: B. Boehm

WG Associates 6607 Stonebrook Circle Dallas, TX 75240 Attn: V. Goldberg

SANDIA INTERNAL:

1513	D.R. Adkins
1708	R.E. Loehman
1846	D.H. Doughty
1846	C.S. Ashley
2402	J.L. Ledman
2476	F.P. Gerstle
2476	S.T. Reed
6200	D.L. Hartley
6200	B.W. Marshall
6201	P.C. Klimas
6202	G.L. Jones
6213	A.R. Mahoney
6215	C.P. Cameron
6215	W.W. Erdman
6215	R.M. Edgar
6215	R.M. Houser (10)
6215	J.W. Strachan (20)
6215	L. Yellowhorse
6215	Library (5)

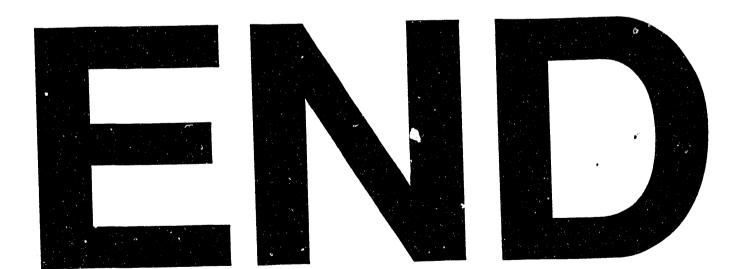
Eric Weber 302 Caribbean Lane Phoenix, AZ 85022

David White 3915 Frontier Lane Dallas, TX 95214

3M-Energy Control Products (2) 207-1W 3M Center St. Paul, MN 55144 Attn: R. Dahlen

- C.E. Tyner 6216 D.J. Alpert (4) 6216 6216 J.R. Anderson 6216 C.E. Andraka 6216 J.M. Chavez 6216 R.B. Diver J.W. Grossman 6216 6216 G.J. Kolb 6216 T.R. Mancini D.F. Menicucci 6216 6216 J.B. Moreno T.A. Moss 6216 M.R. Prairie 6216 6216 M.A. Powell 7141 Technical Library (5) 7613-2 Document Processing for DOE/OSTI (8) **Technical Publications** 7151
- 8523-2 Central Technical Files

# DATE FILMED 10/04/93



۰Ĵ

5